



Overview of the Suomi National Polar-orbiting Partnership (NPP) Sensor Data Records from CrIS, ATMS, VIIRS and OMPS

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> With Contributions from JPSS SDR Team Leads and many other SDR Science Team Members

2014 STAR JPSS Science Teams Annual Meeting May 12-16, 2014 5830 University Research Court, College Park, MD 20740



SDR/LTM Scope Defined in JPSS Algorithm Management Plan



- The SDR teams have the expertise to implement the SDR sensor calibration and to plan, manage and carry out sensor algorithm activities that will be required during prelaunch, operations, and sustainment to meet mission goals and requirements.
- The SDR Senior Lead (SDR Chair) defines the resources required for the science teams as well as coordinates activities. The SDR teams perform data analysis, produce on-orbit look-up tables for SDR algorithms, generate or validate operational SDR algorithms and maintain instrument SDRs.
- The SDR teams conduct monitoring and analysis of sensor parameters to determine modifications in both the ground processing and flight tables to maintain accuracy and stability of the SDRs.
- The teams work with the evolution of the S-NPP algorithms to JPSS-1 and JPSS-2 requirements as described in the JPSS L1RD, JPSS L1RD Supplement, and the JPSS System Requirements Specification.
- The four SDR teams are each assigned to one of the four sensors: Visible-Infrared Imaging Radiometer Suite (VIIRS), Cross-Track Infrared Sounder (CrIS), Advanced Technology Microwave Sounder (ATMS), Ozone Mapping Profiler Suite (OMPS), and any other sensor used to satisfy the JPSS L1RD
- The SDR teams identify and develop corrections for existing SDR algorithms, define requirements for the sensor test program, monitor long-term instrument performance and sensor trending, and provide re-analysis of sensor performance over the sensor lifetime & across satellite platforms.
- The SDR teams will establish the operational criteria and thresholds and maintain them through coordination with the STAR Long-Term Monitoring Team.
- The SDR team coordinates with instrument and flight projects to assure essential project elements are implemented efficiently.
- The STAR LTMS team will track and maintain sensor health and data product quality over the life of the mission by leveraging tools and collaborations between STAR and OSPO already in place. The LTMS tools and findings will support the operational flight and ground segments and will continue to support ongoing collaborations with Office of Satellite Products and Operations (OSPO) and NDE operational teams. The LTMS functions are provided both before and after the transition of the ground system to NOAA operations.







Attendees for SUOMI NPP SDR Product Review Meeting in NOAA Center for Weather and Climate Prediction Auditorium

Review Outcomes: SNPP SDR Products Review Meeting was held on Dec. 18-20, 2013. NESDIS Senior Management Leads: Ms. Mary Kicza and Dr. Al Powell attended the review. The Cal/Val team scientists presented the results on their specific calval tasks and NWP and other users NWS/NOS offered their independent assessments of data product quality based on their intensive cal/val analyses. The review panel recommended that the CrIS, ATMS and VIIRS SDR products be ready to be declared validated scientifically. And three remaining issues were recommended to resolve before OMPS EV SDR goes to the validated stage: cross-track effects in NM need to be addressed; Stray-light improvements still needed in NP SDR; Artificial separation between EV SDR and Cal SDR should be eliminated

Significance: Suomi NPP CrIS and ATMS SDR products are continuing NOAA afternoon orbits sounding data for NWS NWP radiance assimilation. It is shown from CEP global forecast system (GFS) and ECMWF global models that uses of CrIS and ATMS data have similar or slightly better impacts on the global medium-range forecasts



Suomi NPP TDR/SDR Algorithm Schedule



Sensor	Beta	Provisional	Validated
CrIS	February 10, 2012	February 6, 2013	March 18, 2014
ATMS	May 2, 2012	February 12, 2013	March 18, 2014
OMPS	March 7, 2012	March 12, 2013	June, 2014
VIIRS	May 2, 2012	March 13, 2013	April 16, 2014

Beta

- Early release product.
- Initial calibration applied
- Minimally validated and may still contain significant errors (rapid changes can be expected. Version changes will not be identified as errors are corrected as on-orbit baseline is not established)
- Available to allow users to gain familiarity with data formats and parameters
- Product is not appropriate as the basis for quantitative scientific publications studies and applications

Provisional

- Product quality may not be optimal
- Incremental product improvements are still occurring as calibration parameters are adjusted with sensor on-orbit characterization (versions will be tracked)
- General research community is encouraged to participate in the QA and validation of the product, but need to be aware that product validation and QA are ongoing
- Users are urged to consult the SDR product status document prior to use of the data in publications
- Ready for operational evaluation

Validated

- On-orbit sensor performance characterized and calibration parameters adjusted accordingly
- Ready for use in applications and scientific publications
- There may be later improved versions
- There will be strong versioning with documentation





34 papers have been accepted in AGU Journal Geophysical Research Special Issue on Suomi NPP satellite calibration, validation and applications. *Guest Editor: Fuzhong Weng*

Suomi National Polar-Orbiting Partnership Satellite Calibration, Validation and Applications

JOURNAL OF GEOPHYSICAL RESEARCH SPECIAL ISSUE OF THE

Ushering in a New Era of Satellite Remote Sensing to Benefit Society





Suomi NPP Calibration/Validation Schedule

• Four Phases of Cal/Val:

- 1. Pre-Launch; all time prior to launch Algorithm verification, sensor testing, and validation preparation
- 2. Early Orbit Check-out (first 30-90 days) System Calibration & Characterization
- 3. Intensive Cal/Val (ICV); extending to approximately 24 months post-launch xDR Validation
- 4. Long-Term Monitoring (LTM); through life of sensors after ICV
- For each phase:
 - Exit Criteria established
 - Activities summarized
 - Products mature through phases independently



STAR ICVS-LTM for SNPP/JPSS

http://www.star.nesdis.noaa.gov/icvs/status_NPP_ATMS.php







JPSS STAR Science Team Annual Meeting ATMS SDR Team Report

Fuzhong Weng ATMS SDR Lead May 12, 2014

2014 STAR JPSS Science Teams Annual Meeting May 12-16, 2014 5830 University Research Court, College Park, MD 20740











- Overview
 - Products, Requirements, Team Members, Users, Accomplishments
- SNPP Algorithm Evaluation
 - Algorithm Description, Validation Approach and Datasets, Performance vs. Requirements, Risks/Issues/Challenges, Quality Monitoring, Recommendations
- Future Plans
 - Plan for JPSS-1 Algorithm Updates and Validation Strategies, Schedule and Milestones
- Summary





- ATMS is a new generation of microwave sounding instrument. Compared to AMSU-A, and MHS, it has
 - a higher spatial resolution for better detection of severe weather features
 - more channels at WG bands to better delineate atmospheric water vapor
 - overlapping field of views that can be used for resampling and noise reduction
- Calibration requirements for ATMS are much more stringent than for AMSU, and include prelaunch data analysis and post-launch characterization of
 - instrument noise behavior including striping index, power spectrum and NEDT
 - calibration accuracy, nonlinearity and gain stability
 - detection and correction of lunar intrusion in cold target observations
 - scan angle dependent bias from antenna emission and polarization
 - generation of three SDR products: TDR, SDR, and RSDR



- One federal scientist, Tsan Mo who retired in March 31, 2014, was in charge of all the operational calibration of AMSU/MHS instruments
- Other projects supported through NOAA climate data program and led by Chengzhi Zou on cross calibration of MSU and AMSU for climate data record
- STAR-based CalVal supported one contract scientist, Ninghai Sun, to develop the Integrated CalVal System (ICVS) for microwave applications
- Interactions with OSPO and EUMETSAT on operational upgrades of SDR or L1B algorithms were effective and efficient
- But, advanced calibration sciences have been generally lacking due to the resource limitation





- Builds a strong SDR science team which is participated by the key stakeholders
- Works closely with NASA on all the instrument related issues
- Develops innovative theory, analysis and methodology in ATMS calibration
- Utilizes unique SNPP and JPSS mission opportunities to learn new science
- Enhances STAR ICVS for real-time monitoring of SNPP instruments
- Works with NWP user community for timely feedbacks on ATMS SDR data quality
- Outreaches to the broad communities through peer-review papers
- Actively organizing JPSS meeting and attending various conferences (e.g. ITSC, IGARSS, AMS)





PI Name	Organizatio n	Team Members	Primary Role and Responsibility
Fuzhong Weng/Ninghai Sun	NOAA	T. Yang, M. Tian	Budget, Coordination, TVAC analysis, SDR sciences & algorithm, SRF, Long-term monitoring
Lin Lin/Andrew Collard	JCSDA/NCEP	Y. Chen	SRF analysis, LBLRTM, bias characterization, coordination with NWP users
Edward Kim	NASA	J. Lyu	NASA ATMS instrument scientist, TVAC data, instrument anomaly investigation
William Blackwell	MIT/LL	V. Leslie	Support NPP/J1 Calval, SDR sciences, PCT/LUT, prelaunch TVAC data analysis
Xiaolei Zou	NGI/FSU	Z. Qin, Y. Ma	Striping analysis and mitigation, cross calibration
Kent Anderson	NGES	M. Landrum	NGES ATMS instrument engineer
Degui Gu	NGAS	A. Foo	Algorithm test and integration for IDPS operations
Wael Ibrahim	Raytheon		IDPS operations
Kris Robinson	USU/SDL		ATMS geolocation error characterization

ASA JPSS Science POCs and Leads at NOAA/NASA





	MSU				AMSU/MHS			ATMS	
	Ch	GHz	Pol	Ch	GHz	Pol	Ch	GHz	Pol
				1	23.8	QV	1	23.8	QV
				2	31.399	QV	2	31.4	QV
	1	50.299	QV	3	50.299	QV	3	50.3	QH
							4	51.76	QH
				4	52.8	QV	5	52.8	QH
	2	53.74	QH	5	53.595 ± 0.115	QH	6	53.596 ± 0.115	QH
				6	54.4	QH	7	54.4	QH
	3	54.96	QH	7	54.94	QV	8	54.94	QH
				8	55.5	QH	9	55.5	QH
	4	57.95	QH	9	fo = 57.29	QH	10	fo = 57.29	QH
				10	fo \pm 0.217	QH	11	fo±0.3222±0.217	QH
				11	fo±0.3222±0.048	QH	12	$fo \pm 0.3222 \pm 0.048$	QH
				12	fo ±0.3222±0.022	QH	13	fo±0.3222±0.022	QH
				13	fo± 0.3222±0.010	QH	14	fo±0.3222 ±0.010	QH
				14	fo±0.3222±0.0045	QH	15	fo± 0.3222±0.0045	QH
				15	89.0	QV			
				16	89.0	QV	16	88.2	QV
				17	157.0	QV	17	165.5	QH
							18	183.31 ± 7	QH
Exact match to A	Exact match to AMSU/MHS					19	183.31 ± 4.5	QH	
Only Polarization different		19	183.31 ± 3	QH	20	183.31 ± 3	QH		
Unique Passband	l 			20	191.31	QV	21	183.31 ± 1.8	QH
Unique Passband, and Pol. different from closest AMSU/MHS channels		18	183.31 ± 1	QH	22	183.31 ± 1	QH		

ATMS Channel Weighting Functions



Three Generations of Microwave Sounding Instruments from MSU to AMSU/MHS to ATMS

ATMS Field of View Size for the beam width of 2.2° – black line

ATMS Resample to the Field of View Size for the beam width of 3.3°- blue line



ATMS Radiometric Calibration Flow Chart



ATMS Two-Point Calibration with Non-linearity Correction in Brightness Temperature

$$T_{b,ch} = T_{b,ch}^{w} + \frac{C_{ch}^{s} - C_{ch}^{w}}{C_{ch}^{w} - \overline{C_{ch}^{c}}} (T_{b,ch}^{w} - T_{b,ch}^{c}) + 4T_{NL}x(1-x)$$

$$\overline{C_{ch}^{w}}(i) = \sum_{\substack{k=i-N_{s} \ j=1 \ i+N_{s}}}^{i+N_{s}} \sum_{j=1}^{4} W_{k-i}C_{ch}^{w}(k,j)$$

$$\overline{C_{ch}^{c}}(i) = \sum_{\substack{k=i-N_{s} \ j=1 \ CW}}^{i+N_{s}} \sum_{j=1}^{4} W_{k-i}C_{ch}^{c}(k,j)$$
(Y)

 $x = \frac{I_{b,l}}{1}$

$$\overline{G_{ch}}(i) = \frac{C_{ch}^{w}(i) - C_{ch}^{c}(i)}{\overline{T_{b,ch}^{w}}(i) - \overline{T_{b,ch}^{c}}}$$
Nonline
cold pla
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measure
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earity of ATMS channel 1, calculated for ate (CP) at 5°C for redundancy ration 1 (RC1). Blue dots represent the ed scene temperatures. Black solid curve nts the regression curve. Dashed line nts the peak nonlinearity.

A dramatic difference from AMSU calibration is the treatment of nonlinearity term which is derived from the medium theorem and x is a parameter derived from the linear term.

Analysis of ATMS TVAC Test Data



Preliminary TVAC data analysis shows J1 ATMS is much cleaner than SNPP, except channel 16 and 17.

Uses of SNPP ATMS Pitch Maneuver Data February 20, 2012



ATMS Down Track Scan

courtesy of Vince Leslie, MITLL

ATMS Cross Track Spot

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SNPP Pitch-Over Maneuver for ATMS Calibration

- Calibrated space view scene brightness temperature from IDPS are not equal to 2.7K cosmic background
- Strange scan angle dependent feature from IDPS TDR products



New ATMS SDR Algorithm Including Spill-over and Side-lobe Corrections

For Quasi-V(TDR):

$$T_{a}^{Qv} = \eta_{me}^{vv} T_{b}^{Qv} + \eta_{me}^{hv} T_{b}^{Qh} + \eta_{se}^{vv} T_{b,se}^{Qv} + \eta_{se}^{hv} T_{b,se}^{Qh} + (\eta_{sc}^{vv} + \eta_{sc}^{hv}) T_{c,RJ} + S_{a}^{Qv}$$

For Quasi-H (TDR)

$$T_{a}^{Qh} = \eta_{me}^{hh} T_{b}^{Qh} + \eta_{me}^{vh} T_{b}^{Qv} + \eta_{se}^{hh} T_{b,se}^{Qh} + \eta_{se}^{vh} T_{b,se}^{Qv} + (\eta_{sc}^{hh} + \eta_{sc}^{vh}) T_{c,RJ} + S_{a}^{Qh}$$

Weng, F., X. Zou, M. Tian, W.J. Blackwell, N. Sun, H. Yang, X. Wang, L. Lin, and K. Anderson, 2013, Calibration of Suomi National Polar-Orbiting Partnership (NPP) Advanced Technology Microwave Sounder (ATMS), J. Geophys. Res, **118**, 1–14, doi:10.1002/jgrd.50840,

ATMS Polarization vs. Scan Angle



The brightness temperature with pure (dashed curve) and quasi- (solid curve) horizontal polarization (circle) and vertical (star) polarization states using the US standard atmospheric profile with sea surface wind speed being 5 m/s and sea surface temperature being 290 K.

ATMS SDR Biases due to the 3rd Stokes Component



Eh vector is defined as the electronic vector perpendicular to wave propagation plane

$$\begin{bmatrix} T_B^{QV} \\ T_B^{QH} \\ T_B^{Q3} \\ T_B^{Q4} \end{bmatrix} = \begin{bmatrix} \cos^2\theta & \sin^2\theta & 0.5\sin2\theta & 0 \\ \sin^2\theta & \cos^2\theta & -0.5\sin2\theta & 0 \\ -\sin2\theta & \sin2\theta & \cos2\theta & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} T_B^V \\ T_B^H \\ T_B^3 \\ T_B^4 \end{bmatrix} \qquad T_B^{QV} = T_B^H \sin^2\theta + T_B^V \cos^2\theta + T_b^3 \frac{1}{2}\sin2\theta \\ T_B^{QH} = T_b^H \cos^2\theta + T_b^V \sin^2\theta - T_b^3 \frac{1}{2}\sin2\theta \end{bmatrix}$$

ATMS Calibration Accuracy Assessment Using COSMIC Data

• Time period of data search:

January, 2012

• Collocation of ATMS and COSMIC data:

Time difference < 0.5 hour

Spatial distance < 30 km

(GPS geolocation at 10km altitude is used for spatial collocation)





3056 collocated measurements

Slide Courtesy of Lin Lin

ATMS Noise Equivalent Temperature (NEDT)

For a time series with a stable mean, the standard deviation of the measurements can be used as NEDT:

$$\sigma_{ch} = \left[\frac{1}{4N} \sum_{i=1}^{N} \sum_{j=1}^{4} \left(\frac{C_{ch}^{w}(i,j) - \overline{C_{ch}^{w}}(i)}{\overline{G_{ch}}(i)}\right)^{2}\right]^{1/2}$$

For a non-steady mean such as ATMS warm count from blackbody target, Allan deviation is recommended for NEDT:

$$\sigma^{Allan}(m) = \sqrt{\frac{1}{2m^2(N-2m)}} \sum_{j=1}^{N-2m} \left(\sum_{i=j}^{j+m-1} \left(C_{ch}^w(i+m) - C_{ch}^w(i) \right) \right)^2$$



ATMS channel 1 warm count mean (blue, y-axis on the right), the standard deviation (red, y-axis on the left) and the overlapping Allan deviation (green, y-axis on the left) of the 17-scanline (m) average as a function of the total sample size (N).

ATMS: Striping

Weak cross-track striping effect, especially for stratospheric temperature-sounding channels.





ATMS Resampling Algorithm Using the Backus-Gilbert (BG) Method



An effective AMSU-A target FOV: output of BG remap (shaded in gray) ATMS effective FOVs: Circles with colors indicating the magnitude of BG coefficients

Major Accomplishment Highlights

- ATMS TDR and SDR products have been declared a validated maturity level
- All the channels have noises much lower than specification
- ATMS processing coefficient table (PCT) were updated with nominal values
- Geolocation errors for all the channels are quantified and are smaller than specification
- On-orbit absolute calibration was explored using GPS RO data, LBLRTM and ATMS SRF. The biases at the upper-air sounding channels are characterized
- Remap SDR (RSDR) coefficients were optimally set and RSDR biases are assessed
- Complete dthe first cycle data analysis of J1 ATMS TVAC data

ATMS Channel Noise Characterization



All Channels are within Specifications (Weng et al., 2012, JGR)

ATMS Noise Equivalent Temperature (NEDT)



Channel Number

Channel Number

ATMS standard deviation (blue) and Allan deviation (red) with channel number. The sample size (N) is 150 and the averaging factor (m) for the warm counts is 17. The standard deviation is much higher than Allan deviation. On-orbit ATMS noise from the standard deviation is lower than specification but is higher than AMSU/MHS. ATMS resample algorithm can further reduce the noise comparable to AMSU/MHS

J1 NEDT v.s. Allan Variance at 300K



SNPP ATMS Pre-launch Calibration Accuracy through TVAC Data



Red – Calibration accuracy from nominal Thermal Vacuum (TVAC) data, Green – values obtained from the best TVAC data and Blue – specification

Prelauncht ATMS calibration accuracy is quantified from six redundant configuration (RC) thermal vacuum (TVAC) data and exceeds/is better than the specification

ATMS Post-launch Characterization of Calibration Accuracy through O-B



On-orbit ATMS calibration accuracy is characterized using GPSRO and ECMWF data as input to RT model and is better than specification for most of sounding channels.

SNPP ATMS Has Stable Noise


Microwave Radiometry Striping Noise

SNPP ATMS Ch 22

Striping noises are found in ATMS, MHS, and AMSU-B. The magnitudes of ATMS temperature and water vapor sounding channels are about ± 0.3 K and ± 1.0 K, respectively NOAA-18 MHS Ch3 NOAA-16 AMSU-B Ch3 Κ -1.0 -0.8 -0.6 -0.4 -0.2 0.0 0.2 0.4 0.6 0.8 1.0 See *Qin et al.*, 2013 JGR

T_b at Channel 1 within Sandy before and after Remap (0600 UTC October 28, 2012)

35N 990 34N 980 33N- T_{b} 32N 970 260 NCEP GFS SLP 31N 255 (original) (contour interval: 10hPa) 30N-250 980 29N-245 28N-990 240 27N 235 78W 76W 74W 72W 70W 68W 230 35N 225 34N T_b^{BG} 220 33N-215 $\Delta T_b = T_b^{BG} - T_b$ -1 32N (after BG) (K) 31N (contour interval: 1K) 30N 29N-28N-27N

72W

74W

78W

76W

70W

68W

Biases in the Tropics (NOAA-15, MetOp-A, SNPP)



NOAA-18 is subtracted. The pentad data set within $\pm 30^{\circ}$ latitudinal band.

ATMS Lunar Intrusion Correction Algorithm

Brightness temperature increment arising from lunar contamination can be expressed as a function of lunar solid angle, antenna response and radiation from the Moon

Space view Tb or radiance increment:

$$\Delta T_{moon} = G * \Omega * T_{moon}$$

Antenna response function:

$$\mathbf{G} = e^{\frac{-(\beta' - \alpha_0)^2}{2\delta^2}}, \text{ with } \delta = \frac{0.5 \cdot \theta_{3dB}}{\sqrt{2 \cdot \log 2}}$$

Weights of the Moon in antenna pattern:

$$\Omega_{moon} = \frac{\pi \left(\frac{r_{moon}}{D_{moon}}\right)^2}{\iint G(\theta, \varphi) d\theta d\varphi}$$

Brightness temperature of the Moon:

 $T_{moon} = 95.21 + 104.63 \cdot (1 - \cos\Theta) + 11.62 \cdot (1 + \cos2\Theta)$



With LI correction

ATMS SDR Scan Angle Dependent Bias

- Methodology:
 - SDR angular dependent biases are assessed using ECMWF and CRTM simulations
 - Cloud-affected radiances are removed with cloud liquid water algorithm (Weng et al., 2003)
 - Also, the measurements with the surface wind speeds are less than 10m/s are used
- Results:
 - ATMS SDR sounding channels have small bias but less angular dependent
 - But window channels have some significant biases



ATMS SDR Maturity Level – Validated

• Requirements

- Instrument & SDR performances exceeded requirements since Provisional status declaration 1/31/2013
- SDR software
 - Stable & free of errors since 11/14/2013 (Mx8.0)

• Documentation

- 6 presentations in this meeting
- 7 Journal papers
- SDR ATBD (revised)
- SDR user guide (new)
- SDR error budgets



IDPS ATMS SDR CalVal Milestones



Major Issues

- From 19th ITSC, NWP community requests NOAA to develop and share the software on ATMS de-striping and to make available 30 days of TDR and SDR data
- The ATMS brightness temperatures from IDPS are peculiar and show angular dependent pattern when its antenna scans over the cold space during the pitch maneuver period
- Updating the ATMS PCT/LUT at IDPS is very complicated and slow. One simple PCT value update took more than two weeks. It may become faster since PCT update is now approved as fast track
- J1 ATMS TVAC instrument noise at channel 17 is out of specification and some of channels continue showing striping pattern, though the J1 striping magnitude is smaller than SNPP

Path Forward

- Suomi NPP
 - Refine ATMS scan bias corrections for TDR to SDR conversion with better characterization of xpol spill-over, W/G band slope (note intercept has been updated)
 - Develop ATMS radiometric calibration in full radiance to make the SDR data consistent with NOAA heritage AMSU-A/MHS
 - Refine striping mitigation algorithm for WG bands
- JPSS -1 and -2
 - Support of and participation in pre-launch testing, instrument characterization and calibration data development
 - Software update/improvement (implementations of new calibration algorithms, full resolution SDR and computation efficiency schemes), delivering the SDR code in January 2015.
 - Work with NGES to better characterize ATMS antenna (side-lobe, xpol spill-over, polarization twist angle) for J1/J2 mission
 - A comprehensive test data set derived from SNPP and J1 TVAC tests for J1 algorithm and software development and test
 - Support J1 and J2 waiver studies

ATMS SDR Data Sets

- IDPS
 - SDRs produced by IDPS with versions up to Mx8.3
 - Calibration PCT/LUT: Updated with beam efficiency and scan bias correction
 - Lunar correction DR was submitted and will be in Mx8.3 or high version
 - Striping correction DR was submitted and will be implemented in MX8.6 or high
- ARTS (ATMS Radiance Transformation System)
 - Use for reprocessing ATMS in radiance
 - Replace the current IDPS processing for J1 and J2 mission
 - B-G resample SDR will be in 2.2 degree for channel 1 to 16







- ATMS TDR/SDR data has reached a validated maturity *level* (*definition: on-orbit performance is characterized and calibration parameters are adjusted accordingly. The data is ready for use by the operational center and scientific publications*)
- ATMS SDR team made following major calval accomplishments:
 - On-orbit NEDT is well characterized in standard and Allan variance and both way shows the instrument meets specification
 - Bias (accuracy) is well characterized with GPSRO data and ECMWF model outputs
 - All the important quality flags are checked and updated
 - Calibration coefficients from TDR to SDR are updated
 - Lunar intrusion correction was in operation since March 18, 2014
 - ATMS and AMSU-A inter-sensor biases are well characterized and ATMS TDR data are now within AMSU-A family
 - STAR ICVS can provide long-term monitoring of ATMS instruments
 - All the calval sciences have been published through peer-reviewed process
 - Work on J1 TAC test and data analysis is progressing well





CrIS SDR Team Report

Yong Han, CrIS SDR Team Lead

2014 STAR JPSS Science Teams Annual Meeting NOAA Center for Weather and Climate Prediction (NCWCP) 5830 University Research Park, College Park, Maryland May 12-16, 2014









- Team Membership
- Overview of last year's Cal/Val activities and achievements
- Ongoing calibration algorithm/code improvements
- Challenges and risks
- Next year's activities
- Summary





PI Name	Organization	
Yong Han	NOAA/STAR	
Deron Scott	SDL	
Hank Revercomb	UW	
Larrabee Strow	UMBC	
Dan Mooney	MIT/LL	
Degui Gu	NGAS	
Joe Predina	Logistikos Engineering LLC.	
Mike Cromp	Exelis	
Dave Johnson	NASA	
Wael Ibrahim	Raytheon	
Carrie Root	JPSS/DPA	





• S-NPP

- Intensive Cal/Val (ICV) activities (ended in Dec 2013) great success; SDR product reached Validated status
- Long-Term Monitoring (LTM) activities, covering all areas that are significant to the data quality
- Preparation for processing full spectral resolution data
- JPSS-1
 - Calibration algorithm/code improvements
 - J1 test data analysis
 - Proxy data development for Ops software tests





Accomplishment Highlights



Contraction of Contraction of Contraction

SDR validated in three stages: Beta, Provisional, and Validated



- Spectral CalVal
- Geolocation CalVal
- CrIS instrument and SDR trending and monitoring





CrIS SDR uncertainties (blue) vs. specifications (black)

Band	NEdN @287K BB mW/m²/sr/cm ⁻¹	Radiometric Uncertainty @287K BB (%)	Frequency Uncertainty (ppm)	Geolocation Uncertainty (km) *
LW	0.098 (0.14)	0.12 (0.45)	<mark>3</mark> (10)	1.2 (1.5)
MW	0.036 (0.06)	<mark>0.15</mark> (0.58)	<mark>3</mark> (10)	1.2 (1.5)
SW	0.003 (0.007)	0.2 (0.77)	3 (10)	1.2 (1.5)

- Requirements
 - Instrument & SDR performances exceed requirements by large margins
- SDR software
 - Stable & free of errors that could impact data quality since 11/14/2013 (Mx8.0)
- Documentation
 - SDR User's Guide (55 pages)
 - Revised ATBD
 - Peer-review Journal papers



Stable Instrument Performance

AND ATMOSA

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Daily occurrence of Good SDR spectra

LW	99.9817%
MW	99.9817%
SW	99.9816%

- No ice contamination on detector so far
- No significant South Atlantic Anomaly (SAA) impact
- No Fringe Count Error (FCE) so far

Mainly due to sun-glint saturation









NPP CrlS Brightness Temperature, 11 µm (900 cm⁻¹), Mapped, Descending, 12/02/2013



Overall SDR quality flag (Blue - good)

NPP CrIS Mid Wave SDR Overall Quality Flag, Mapped, Ascending, 12/02/2013 (Blue: Good; Green: Degraded; Red: Invalid)



NPP CrIS Mid Wave SDR Overall Quality Flag, Mapped, Descending, 12/02/2013



CrIS data monitoring website:

http://www.star.nesdis.noaa.gov/icvs/status NPP CrIS.php

RDR Truncation Module Implemented and Validated for IDPS to Process Full Spectral Resolution RDRs



- Activities
 - IDPS RDR truncation module development
 - IDPS SDR evaluation/validation for 2 onorbit full resolution tests
 - Bit trim mask evaluation/adjustment to meet data rate
 - Full resolution SDR processing experiments
 - 25 telecon meeting presentations
- Results
 - IDPS RDR truncation module was implemented & validated (Mx7.1)
 - Proposed Bit trim mask meets the data rate requirement
 - The noise impulse masks need to be lifted by 1 bit (no impact to the data rate)

IDPS CrIS SDR code is ready to process full resolution RDRs and produce normal mode SDRs

The Software truncation module works as expected: Obs – Calc results showing no difference before and during 8/27 FSR test





Preliminary J1 NL Correction Coefficients Derived from Bench DM Data





- The preliminary DM results for J1 are qualitatively similar to FM1 (SW is linear, some linear MW FOVs, all LW FOVs are nonlinear) and the same type of NL correction and TVAC and on-orbit a₂ analysis techniques will be needed for J1.
- Compared to FM-1, the J1 LW FOVs are more linear (except FOV5), and 8 of the J1 MW FOVs are very linear.
- Results are very similar to results found by Exelis (Lawrence S.)
- The difference between the June and Sept DM results (e.g. FOV5) are similar to inconsistent results seen for FM1 DM data analysis, which is still under investigation.

Preliminary Analysis of J1 Gas Cell Bench Test

Test results show good agreement with calculated data



Observed and calculated transmittance for all FOVs



Observed minus calculated transmittance spectra for all FOVs

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Ongoing Calibration Algorithm and Software Improvements





- Recent progress in the investigation of spectral ringing artifacts indicates the current IDPS CrIS SDR calibration algorithms may not be optimal, especially for full spectral resolution SDR processing
- The NWP/Sounding community is interested in using unapodized CrIS data. However, the ringing artifacts in the unapodized data are not negligible
- The current implementation of the spectral Correction Matrix Operator (CMO) is not optimal and may be difficult to apply for some of the calibration algorithms under considerations

Calibration Algorithms under Evaluations

ASA



Item	Member	Calibration	CMO Principals	Calibration Order
1	IDPS	$N = (SA_u^{-1} \cdot F_{s \to u} \cdot f_{ATBD}) \cdot \left\{ \frac{S_E - S_{SP}}{S_{ICT} - S_{SP}} \cdot ICT(T, u_{sensor*(1+delta)}) \right\}$		
2	ADL/CSPP	$N = (SA_u^{-1} \cdot F_{s \to u} \cdot f_{ATBD}) \cdot \left\{ \frac{S_E - S_{SP}}{S_{ICT} - S_{SP}} \cdot ICT(T, u_{senvor*(1+delta)}) \right\}$	$SA_u^{-1} \cdot F_{s \to u}$	
3	Exelis (old)	$N = \left(SA_u^{-1} \cdot F_{s \to u} \cdot f_{ATBD}\right) \cdot \left\{\frac{S_E - S_{SP}}{S_{ICT} - S_{SP}} f_{BH} \cdot \left[SA_u^{-1} \cdot F_{s \to u}\right]^{-1} \cdot ICT(T, u_{sensor})\right\}$		
4	UMBC/UW** option A	$N = F_{s \to u} \cdot f \cdot SA_s^{-1} \cdot \left\{ f \cdot \frac{FIR^{-1} \cdot \left(S_E - S_{SP}\right)}{FIR^{-1} \cdot \left(S_{ICT} - S_{SP}\right)} \cdot ICT(T, u_{sensor_off_axis}) \right\}$		Calibration first, then CMO
5	CCAST Cal mode 1	$N = F_{s \rightarrow u} \cdot f \cdot SA_s^{-1} \cdot \left\{ \frac{FIR^{-1} \cdot \left(S_E - S_{SP}\right)}{FIR^{-1} \cdot \left(S_{ICT} - S_{SP}\right)} \cdot ICT(T, u_{sensor_off_axis}) \right\}$		
6	UMBC/UW** option B	$N = F_{s \to w} \cdot \left\{ ICT(T, u_{sensor}) \cdot f \cdot SA_s^{-1} \cdot \left\{ f \cdot \frac{FIR^{-1} \cdot \left(S_E - S_{SP}\right)}{FIR^{-1} \cdot \left(S_{ICT} - S_{SP}\right)} \right\} \right\}$		
7	CCAST Cal mode 2	$N = F_{s \rightarrow u} \cdot f \cdot \left\{ ICT(T, u_{source}) \cdot SA_s^{-1} \cdot \left\{ Re\left[\frac{FIR^{-1} \cdot (S_E - S_{SP})}{FIR^{-1} \cdot (S_{ET} - S_{SP})} \right] \right\} \right\}$	$F_{s \to u} \cdot SA_s^{-1}$	
8	LL(old)*	$N = \left\{ \frac{M \cdot \left(FIR^{-1} \cdot \left(S_{E} - S_{SP} \right) \right)}{M \cdot \left(FIR^{-1} \cdot \left(S_{RT} - S_{SP} \right) \right)} \right\} \cdot ICT(T, u_{user})$		
9	Proposed(1)	$N = F_{s \rightarrow u} \cdot f_{ATBD} \cdot \left\{ \frac{SA_s^{-1} \cdot \left(FIR^{-1} \cdot \left(S_E - S_{SP}\right)\right)}{SA_s^{-1} \cdot \left(FIR^{-1} \cdot \left(S_{ICT} - S_{SP}\right)\right)} \cdot ICT(T, u_{sensor}) \right\}$		CMO first,
10	Proposed(2)	$N = ICT(T, u_{user}) \cdot \left\{ \frac{F_{s \to u} \cdot SA_s^{-1} \cdot f_{ATBD} \cdot \left(FIR^{-1} \cdot \left(S_E - S_{SP}\right)\right)}{F_{s \to u} \cdot SA_s^{-1} \cdot f_{ATBD} \cdot \left(FIR^{-1} \cdot \left(S_{RCT} - S_{SP}\right)\right)} \right\}$		then Calibration
11	Exelis(new)	$N = \left\{ \frac{\left(SA_u^{-1} \cdot F_{s \to u} \cdot \left(S_E - S_{SP}\right)\right)}{\left(SA_u^{-1} \cdot F_{s \to u} \cdot \left(S_{ICT} - S_{SP}\right)\right)} \right\} \cdot ICT(T, u_{uver})$	$SA_u^{-1} \cdot F_{s \to u}$	











A Proposed SDR Algorithm











Algorithms are implemented in ADL and then compared





SW Band (FOR1, full resolution)



Significant difference (ringing) seen in all three bands (unapodized) 0.1 - 0.5 K



Spectral Interpolation before/after the Calibration Ratio Has Big Difference





Note: Ref does interpolation before ratio



Ringing Artifact Reduction by Normalizing FIR Gain before Truncation of IGM



Significant ringing if spectrum is not normalized with FIR gain before interferogram (IGM) truncation and spectral interpolation

Ringing artifacts are largely reduced with the algorithm that normalizes S with the FIR gain

ND ATMOSE

NOAA

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- Recent investigation indicated the current IDPS SA⁻¹ is not optimal and may introduce significant ringing artifacts in full spectral resolution SDR processing
- New algorithms are proposed and are being evaluated
 - Use periodic Sinc function instead of the current Sinc function
 - Double the size of the SA⁻¹ matrix in computation
 - Derive the matrix SA⁻¹ through minimization



Radiance Difference due to the Difference in SA⁻¹



-20

2200

2300

2400

wavenumber cm-1

2500

Radiance

24





- The delivery of CrIS SDR software is scheduled on Jan 15, 2015. However, we still have a large amount of work to do in both algorithm and code changes
- Implementation of the proposed calibration algorithm requires a lot of code changes, which normally start after the algorithm investigations. However, the delivery schedule is pushing us to start working on the code changes before the conclusion of the investigations.
- Current IDPS does not support a dynamic switch between the normal mode and full spectral resolution mode SDR processing; in other words, the switch requires recompiling the software




- Suomi NPP
 - Continuation of RDR and SDR monitoring
 - Fine adjustment of spectral and radiometric calibration parameters and geolocation mapping parameters, if needed.
 - Continuation of Full Spectral Resolution work, if required.
 - SDR algorithm improvement to address the potential issues (e.g. FCE detection/correction, reduction of ringing artifacts and polarization effect correction)
 - Continuation of SDR software improvements to address the remaining and future issues
- JPSS J1
 - Support of and participation in pre-launch testing and instrument characterization
 - Calibration data (LUTs and coefficients) development
 - Algorithm/software development and improvements (full resolution SDR capability, calibration algorithms and FCE detection/correction module), delivering the SDR code in January 2015
 - Development of a comprehensive test data set derived from NPP observations and J1 TVAC tests for J1 algorithm and software development







- The team has successfully completed the CrIS SDR ICV process and achieved the Validated status for the S-NPP CrIS SDR product
- LTM activities are being routinely carried out to ensure the data product quality
- Work has been successfully completed to add a truncation module to the IDPS CrIS SDR software: the software is ready for handling full spectral resolution RDRs
- The team is making efforts to improve the calibration algorithms and processing software. Progress has been made. However, it is challenging to meet the software delivery schedule.
- Preliminary analysis of the bench test data was performed and the results are within the expectation
- The team has a clear path moving forward for both NPP and J1 missions





JPSS STAR Science Team Annual Meeting VIIRS SDR Team

Changyong Cao VIIRS SDR Lead May 12-16, 2014







Outline



- Overview
 - Products, Requirements, Team Members, Users, Accomplishments
- SNPP Algorithms Evaluation:
 - Algorithm Description, Validation Approach and Datasets, Performance vs. Requirements, Risks/Issues/Challenges, Quality Monitoring, Recommendations
- Future Plans
 - Plan for JPSS-1 Algorithm Updates and Validation Strategies, Schedule and Milestones
- Summary



VIIRS SDR Team



Leads	Organization	Members
Changyong Cao	NOAA/NESDIS/STAR	Slawomir Blonski, Frank Padula, Wenhui Wang, Jason Choi, Sirish Uprety, Sean Shao, Yan Bai, Vicky Lin
Frank Deluccia	The Aerospace Corp.	David Moyer, Kameron Rausch, others
J. Xiong/R. Wolfe	NASA/VCST	Hassan Oudrari, Vincent Chang, Aisheng Wu, John Fulbright, Jeff McIntire, Boriana Efrmova, Ning Lei, Gary Lin, Masahiro Nishihama, others
Lushalan Liao	NGAS	Ronsan Chu, Stephnie Weiss, Tahru Ohnuki, Frank Sun, others
Chris Moeller	U. Wisc.	others

Products:

22 SDRs

Users:

VIIRS EDR with more than 20 products





Major Achievements Since Provisional

- VIIRS on-orbit performance is well characterized & meets specifications
- RSBAutoCal being tested and independently validated by NOAA
- VIIRS DNB Straylight Correction implemented (Aug. 2013); tool kit has been evaluated by NOAA
- Geo-location uncertainties for I-/Mbands are ~ 70 m at nadir, meeting specifications at nadir and edge-ofscan (DNB terrain corrected geolocation product is expected in Mx8.3 in March 2014)



DNB Straylight Correction Implemented



RSBAutoCal Testing







Since the validated maturity workshop in December 2013:

- VIIRS SDR achieved validated maturity
- Validation time series developed for ~30 sites worldwide (W. Wang)
- DCC time series since launch established (W. Wang)
- Lunar band ratio time series developed (S. Shao & J. Choi)
- Calibration coefficient changes (c0=0) implemented (May 2014)
- I3/M10 bias studies (new results from Lunar band ratio analysis, see X. Shao in breakout session)
- Sun vector error findings (NASA)
- DNB terrain corrected geolocation (March 18, 2014 with MX8.4)
- Single Board Computer Lockup(SBC) #6 (or 7), aka "Petulant mode" on Feb. 4, 2014
- Flattening in the degradation shown in H and F factors
- VIIRS J1 polarization studies

On-going work:

- Continued updating the calibration knowledge base, with new events analyzed and documented
- Continued bias time series analysis between VIIRS and MODIS
- Continued longterm trending and monitoring



VIIRS Radiometric Validation Time Series at thirty validation sites world-wide





More details will be presented by W. Wang in the VIIRS Breakout session



VIIRS Event Log Database

An important part of the Calibration Knowledge base





For more details, see poster by Y. Bai et al.





 Milestone: Successfully completed the VIIRS SDR Validated Maturity Workshop, and achieved validated status in March 2014.

Accomplishments

- STAR held a three-day Suomi NPP SDR Science and Validated Product Maturity Review (December 18-20, 2013) at the NOAA NCWCP to assess the readiness of the VIIRS SDR data product maturity
- The VIIRS SDR team members and EDR users reported on the progress made since the Provisional Maturity Review demonstrating the VIIRS SDR maturity level
- Concluding the Workshop the review panel members reached consensus that overall the VIIRS SDR product has reached the validated status and therefore is recommended to be approved by the Algorithm Executive Review Board (AERB)
- The AERB approved the recommended validated status in March 2014.
- Significance: VIIRS SDR has achieved the validated maturity





VIIRS SDR Accuracy



	Requirement (absolute	Prelaunch and	Validation: Relative to	Note
	uncertainty for uniform	onboard calibration	MODIS/CrIS/IASI/other	
	scenes)		thru Inter-comparisons	
VIIRS RSB	2% typical reflectance;	1.2% for M1-M7;	2% (±1%) for matching	Except bands with very low signal
	0.3% stability;	1.5% for M8&9	bands	(ex. M11); sub-percent accuracy
	0.1% desirable for Ocean	1.4% for M10		for OC is very challenging.
	Color Applications	1.3% for I1&I2		Geolocation error: expectation is
		1.6% for I3		half I-band pixel; achieved better
				than quarter I-band pixel (1- σ)
VIIRS TEB	M12/M13: 0.7%(0.13K)	Better than 0.13K	0.1K based on statistical	M15 at 190K requirement is 2.1%
	@270K	for all M bands	comparison with	radiance or 0.56K
	M14: 0.6% (0.26K)	except M13 (0.14);	MODIS and CrIS	Geolocation uncertainty:
	@ 270K	0.47K for I4;	ER-2/SHIS Aircraft	expectation was half I-band pixel;
	M15/M16: 0.4%	0.23K for I5	underflight shows	achieved better than quarter I-
	(0.22K/0.24K) @270K		excellent agreement	band pixel (1- σ)
	I4: 5% (0.97K) @270K		M15 0.4 K bias relative to	
	I5: 2.5% (1.5K) @270K		CrIS at 200K (in	
			spec.)	
VIIRS DNB	• 5%, 10%,30% L _{min}	3.5%, 7.8%, and	• 4%, 7.7%, 11.8%	Geolocation error is a ~10th of a
	(LGS,MGS,HGS)	11% (LGS, MGS,	(LGS, MGS, HGS)	pixel (1- σ) on the ellipsoid earth
		HGS)		but can exceed 1km (up to 24 km
				at the edges of scan) without
				terrain correction



Recent RSB H&F factor trends





Recent F-factors show significant trend change which suggests that degradation has stopped or even reversed
Is this real or artificial?
How can we tell through validation?
Is this due to issues in the H-factor calculations?



H-factors in the above plot do not show major recent trend change due to smooth?
The unsmoothed version does show trend

- change (such as those produced by Autocal)
- What's the impact on the F-factor calibrations?

What's the impact on EDR products?







ODet1

412 nm

2.0×10³

o Det2

445 nm

Det3

488 nm

4.0×105

555 nm

6.0×10³

Orbit #





Courtesv of N. Lei, VCST

Det8

926 nm

1.2×104

Det7

865 nm

1.0×104

oDet4 Det5 Det6

672 nm

8.0×10³

745 nm





VIIRS J1 Status Update





- •Ambient testing: Jan. 2014
- •Pre-Environment Review (PER): Feb. 3-6, 2014
- Polarization issue (discussed later)
- •Electromagnetic Interference (EMI) testing completed May 2014 -Sync loss issue resolved for J1
 - -Single Board Computer (SBC) Lockup (aka Petulant Mode)
 - issueresolved for J1 (per Gleason and Raytheon)
- •Thermal Vacuum testing: Jun.-Oct. 2014





- VIIRS J1 polarization sensitivity is significantly out of spec for several bands due to filter coating changes
- The VIIRS SDR team is working closely with the flight and vendor to study mitigation strategies
 - Better characterization through additional prelaunch tests
 - Measure at more scan angles, and T-SIRCUS spectral measurements
 - Better quantification of the polarization phenomenon and VIIRS on-orbit performance
 - Better understanding of impacts on EDR products
- Suomi NPP VIIRS polarization meets the polarization sensitvity specification. VIIRS J2 is expected to meet the specification



3/17/2014: Initiated working groups to study the impacts of polarization on products, with several actions from the first telecon on March 17 (M. Goldberg).

4/2/2014: VIIRS SDR special telecon on VIIRS J1 detector level polarization study shows large variation across detectors (presentation by J. McIntire, NASA/VCST)

4/16/2014: MODIS Terra/Aqua prelaunch and on-orbit polarization studies show large increase over the life time of the Terra/MODIS instrument (presentation by J. Xiong, NASA/VCST)

4/24/2014: Recommendations for additional prelaunch testing (telecon): More measurement angles, monochromatic characterization using T-SIRCUS.

Other progress:

- GOME Polarization Measurement Device (PMD) on MetOp A and B
 - Sample data have been analyzed and a preliminary global map of DoLP map generated.
- Prototype polarization spectroradiometer developed leveraging the ASD spectrometer, with sample in-situ measurements



7

Polarization factors (combining all byonir configurations) – HAM A byonir in: M1-M3; byonir out: I1-I2, M4-M7 Factors above specification for M1-M4



Courtesy of J. McIntire





DOLP for Wavelength: (PP)413.82 and (PS)413.46

Time: 2014-04-15







DOLP for Wavelength: (PP)556.21 and (PS)555.06

Time: 2014-04-15





Ground-Based Polarization Spectroradiometer for Validating VIIRS Polarization Sensitivity (Prototype)







(Protractor will be replaced with 3D- printed piece)



See poster by A. Pearlman et al for details

Location: M Square parking lot at 5:38 to 6:00pm (April 17, 2014)

Took measurements of a highly polarized sky:

Pointed sensor at ~90^o to sun

Mostly clear with cirrus clouds covering ~75% of sky Measurement time: 5 minutes



Future plan: Lunar polarization measurements at UMD observatory

800 1000

1200 1400 1600 Wavelength (nm) 1800

14000

3000 Saw County 4000





130 160





- Achieving better calibration accuracy for Ocean Color applications
- Further improve onboard calibration
 - RSB autocal, solar vector, etc.
- Enhance vicarious monitoring capability to ensure high accuracy

- •Striping in both SST bands and RSB
- •Detector level RSR performance issues
- •Polarization effects
- •Single Board Computer Lockup (SBC), aka "Petulant mode"
- •Sync loss
- •J1 VIIRS support



Summary



- VIIRS SDR has achieved calibrated/validated Maturity Status in both radiometry and geolocation
- Continue improving the radiometric accuracy to meet Ocean Color application needs
 - Fine tune calibration coefficients (e.g.: c0=0)
 - RSB autocal
 - Closely monitoring trend changes
 - Lunar band ratio analysis
- Future work focus on:
 - J1 calibration support, such as polarization studies (observations and RTM)
 - Further enhancements in instrument performance through research (such as striping, detector level processing, improved accuracy, etc)
 - Long term monitoring



Backup slides





VIIRS On-orbit Performance Table



- SDRs = L1b = calibrated, geolocated radiance, reflectance and brightness temperature
- 22 types of SDRs -16 moderate resolution (MOD),
- 11 Reflective Solar Bands (RSB)
- 5 Thermal Emissive Bands (TEB)
- -5 imaging resolution (IMG),
- 3 RSB; 2 TEB -1 Day Night Band (DNB) imaging, broadband
- 6 non-gridded geolocation products
- -DNB, IMG, IMG terrain corrected, MOD, MOD terrain corrected, MOD unaggregated
- 2 gridded geolocation products -MOD, IMG

				Specification					Prelunch On Orbit		Drbit		
		Band No.	Driving EDR(s)	Spectral Range (um)	Horiz Sample (track Nadir	Sample Interval (km) (track x Scan) r End of Scan		Ltyp or Ttyp (Spec)	Lmax or Tmax	Spec SNR or NEdT (K)	Measured SNR or NEdT (K) (2)	Measured SNR or NEdT (K) (1)	Measured SNR or NEdT (K) (2)
					riduir	Child of Scan	Histo	44.0	125	25.2	616.9	579	599.0
		M1	Ocean Color Aerosol	0.402 - 0.422	0.742 - 0.259	1.60 x 1.58	Low	155	615	316	1092	974	1045.78
							High	40	127	380	622.4	564	572.02
		M2	Ocean Color Aerosol	0.436 - 0.454	0.742 - 0.259	1.60 × 1.58	Low	146	687	409	1118	975	1010.76
		мз					High	32	107	416	690	611	628.46
			Ocean Color Aerosol	0.478 - 0.498	0.742 - 0.259	1.60 x 1.58	Low	123	702	414	1111	1003	988.54
	~						High	21	78	362	581.1	522	534.96
	I	M4	Ocean Color Aerosol	0.545 -0.565	0.742 - 0.259	1.60 × 1.58	Low	90	667	315	963.2	846	856.51
8	5	11	Imagery EDR	0.600 - 0.680	0.371 - 0.387	371 - 0.387 0.80 × 0.789	Single	22	718	119	240.7	215	214.07
		ME	Orean Color Assess	0.662 - 0.692	0.742 - 0.250		High	10	59	242	366.6	321	336.13
		MD	Ocean Color Aerosol	0.002 - 0.082	0.742 - 0.239	1.00 × 1.58	Low	68	651	360	827.9	673	631.26
8		M6	Atmosph. Correct.	0.739 - 0.754	0.742 - 0.776	1.60 x 1.58	Single	9.6	41	199	415.2	355	368.4
Re		12	NDVI	0.846 - 0.885	0.371 - 0.387	0.80 x 0.789	Single	25	349	150	304.1	251	264.01
		M7	Ocean Color Aerosol	0.846 - 0.885	0.743 0.350	1.60 × 1.58	High	6.4	29	215	519.8	435	457.54
					0.742 - 0.233		Low	33.4	349	340	845.6	636	631.24
		M8	Cloud Particle Size	1.230 - 1.250	0.742 × 0.776	1.60 × 1.58	Single	5.4	165	74	273	233	221
		M9	Cirrius/Cloud Cover	1.371 - 1.386	0.742 × 0.776	1.60 × 1.58	Single	6	77.1	83	253	231	227
		13	Binary Snow Map	1.580 - 1.640	0.371 × 0.387	0.80 × 0.789	Single	7.3	72.5	6	172	149	149
	۲,	M10	Snow Fraction	1.580 - 1.640	0.742 × 0.776	1.60 x 1.58	Single	7.3	71.2	342	714	550	586
	ş.	M11	Clouds	2.225 - 2.275	0.742 × 0.776	1.60 × 1.58	Single	0.12	31.8	10	25	21.8	22
	s/	14	Imagery Clouds	3.550 - 3.930	0.371 × 0.387	0.80 × 0.789	Single	270	353	2.5	0.4	0.4	0.4
		M12	SST	3.660 - 3.840	0.742 x 0.776	1.60 x 1.58	Single	270	353	0.396	0.13	0.13	0.13
		M13	SST	3.973 - 4.128	0.742 x 0.259	1.60 × 1.58	High	300	343	0.107	0.04	0.042	0.04
Bar			Fires				Low	380	634	0.423			
alve													
ži l		M14	Cloud Top Properties	8.400 - 8.700	0.742 x 0.776	1.60 x 1.58	Single	270	336	0.091	0.06	0.06	0.05
	MIR	M15	SST	10.263 - 11.263	0.742 x 0.776	1.60 × 1.58	Single	300	343	0.07	0.03	0.03	0.03
	5	15	Cloud Imagery	10.500 - 12.400	0.371 x 0.387	0.80 x 0.789	Single	210	340	1.5	0.4	0.4	0.4
		M16	SST	11.538 - 12.488	0.742 × 0.776	1.60 x 1.58	Single	300	340	0.072	0.04	0.03	0.03

(1) The Aerospace Corporation (2) NASA NICSE

HSI uses 3 in-scan pixels aggregation at Nadir

Source: VIIRS user's guide. On orbit values (last two columns for March 8, 2012) are updated based on the Murphy table for RSB, provided by Aerospace; TEB values are provided by STAR and NASA.



VIIRS Sensor Specification - RSB sensitivity



rubier eritetett i Sensitivity requirements for									
Band	Center	Gain Type	Single	e Gain		Dual Gain			
	Wavelength								
	(nm)								
					High	Gain	Low	Gain	
			Ltyp	SNR	Ltyp	SNR	Ltyp	SNR	
M1	412	Dual	-	-	44.9	352	155	316	
M2	445	Dual	-	-	40	380	146	409	
M3	488	Dual	-	-	32	416	123	414	
M4	555	Dual	-	-	21	362	90	315	
M5	672	Dual	-	-	10	242	68	360	
M6	746	Single	9.6	199	-	-	-	-	
M7	865	Dual	-	-	6.4	215	33.4	340	
M8	1240	Single	5.4	74	-	-	-	-	
M9	1378	Single	6	83	-	-	-	-	
M10	1610	Single	7.3	342	-	-	-	-	
M11	2250	Single	0.12	10	-	-	-	-	
I1	640	Single	22	119	-	-	-	-	
I2	865	Single	25	150	-	-	-	-	
13	1610	Single	7.3	6	-	-	-	-	

Table: 3.1.5.6.1-1 Sensitivity requirements for VIIRS Sensor reflective bands

Notes:

The units of spectral radiance for Ltyp are watt m⁻² sr⁻¹ µm⁻¹.

The SNR column shows the minimum required (worst-case) SNR that applies at the end-of-scan. Elsewhere in the scan, aggregation will yield a larger SNR.

Within the same gain setting, at radiances larger than Ltyp, the SNR will be larger than what is specified in this table.

Absolute radiometric calibration uncertainty for uniform scenes: < 2%



- TEB sensitivity



Table: 3.1.5.6.2-1 Sensitivity requirements for VIIRS Sensor emissive bands

B and	Center	Gain Type	Single Gain		Dual Gain			
	Wavelength							
	(nm)							
					High	Gain	Low	Gain
			Ttyp	NEdT	Ttyp	NEdT	Ttyp	NEdT
M12	3700	Single	270	0.396	-	-	-	-
M13	4050	Dual	-	-	300	0.107	380	0.423
M14	8550	Single	270	0.091	-	-	-	-
M15	10763	Single	300	0.070	-	-	-	-
M16	12013	Single	300	0.072	-	-	-	-
I4	3740	Single	270	2.500	-	-	-	-
15	11450	Single	210	1.500	-	-	-	-

Notes:

The NEdT column corresponds to the minimum required (worst-case) SNR that applies at the end-of-scan. Elsewhere in the scan, aggregation will yield a larger SNR.

Within the same gain setting, at scene temperatures larger than Ttyp, the SNR will be larger than at Ttyp.

For reference, the NEdT values in Table 15 are related to the noise equivalent spectral radiance (NEdL) by the following formula:





Table: 3.1.5.9.2.3-1 Absolute radiometric calibration uncertainty of spectral radiance for moderate resolution emissive bands

Band	λc (µm)	Scene Temperature				
		190K	230K	270K	310K	340K
M12	3.7	N/A	7.0%	0.7%	0.7%	0.7%
M13	4.05	N/A	5.7%	0.7%	0.7%	0.7%
M14	8.55	12.3%	2.4%	0.6%	0.4%	0.5%
M15	10.763	2.1%	0.6%	0.4%	0.4%	0.4%
M16	12.013	1.6%	0.6%	0.4%	0.4%	0.4%

Table: 3.1.5.9.2.4-1	Radiometric calibration uncertaint	y for imaging emissive ba	nds

Band	Center Wavelength (nm)	Calibration Uncertainty
I4	3740	5.0%
15	11450	2.5%

Source: JPSS VIIRS Performance Requirement Document Code 472 472-00124



VIIRS On-orbit Performance

-SNR and NEDT















One stop shop for VIIRS SDR information



NCC

You are here: Foswiki > NCC Web > VIIRS (21 Nov 2013, ChangyongCao)

🏠 Home

- (i) Terms of Reference
- Publication Database
- About

🔅 GOES-R

- NPP/JPSS/VIIRS
- NPP/JPSS/OMPS
- NOAA/AVHRR
- 🌼 NOAA/SSU
- 🔅 MetOp
- 🔅 JASON
- 🌼 DSCOVR
- 🌼 Space Weather
- 🗽 Standards

🔣 Lunar Calibration

Selibration Sites

Visible Infrared Imaging Radiometer Suite (VIIRS)

The Visible Infrared Imaging Radiometer Suite (VIIRS) is one of the key instruments onboard the Suomi National Polar-Orbiting Partnership (Suomi NPP) spacecraft, which was opened on November 21, 2011, which enables a new generation of operational moderate resolution-imaging capabilities following the legacy of the AVHRR on NOAA an operational environmental monitoring and numerical weather forecasting, with 22 imaging and radiometric bands covering wavelengths from 0.41 to 12.5 microns, providing the records including clouds, sea surface temperature, ocean color, polar wind, vegetation fraction, aerosol, fire, snow and ice, vegetation, , and other applications. Results from calibration and validation have shown that VIIRS is performing very well. **VIIRS paper:** Cao, C., F. DeLuccia, X. Xiong, R. Wolfe, F. Weng, Early On-orbit Performance of the

News and Documents	VIIRS Performance and Monitoring	Data and Software
(I) News ⊡+	WIRS Longterm Monitoring □+	NIRS Image Gallery
Publication Database 🗈	K VIIRS On-orbit Performance Table ⊕	⑧ VIIRS data on CLASS →
VIIRS Users Guide	Standardized Calibration Parameters	() VIIRS data on ftp site (90 days) □+
NIRS Calibration ATBD	K VIIRS Spectral Response Functions	Data on GRAVITE
Conference Presentations	K VIIRS Event Log Database (experimental) □+	HIRS Software Tools
VIIRS Novel Applications	NPP/AQUA SNO Predictions	Planck Calculator for Infrared Remote Sensing
KIRS SDR Data Format	Radiometric Intercomparison with MODIS	VIIRS Line Spread Function along scan
& VIIRS SDR Meetings	VIIRS at Cal/Val Sites	Je VIIRS Cloud Mask (VCM)
U VIIRS FAQ	kon Lunar Calendar for DNB ⊡+	A SDR/EDR Team
About VIIRS	Moon in Space View Events □→	Standard Radiometric Test Scenes





- Cao, C., F. Deluccia, X. Xiong, R. Wolfe, and F. Weng, 2013a, Early On-orbit Performance of the Visible Infrared Imaging Radiometer Suite (VIIRS) onboard the Suomi National Polarorbiting Partnership (Suomi-NPP) Satellite, IEEE Transaction on Geoscience and Remote Sensing, DOI:10.1109/ TGRS.2013.2247768, in press (available online at IEEEXplore).
- Cao, C., X. Xiong, S. Blonski, Q. Liu, S. Uprety, X. Shao, Y. Bai, F. Weng, 2013, Suomi NPP VIIRS sensor data record verification, validation, and long-term performance monitoring, Journal of Geophysical Research: Atmospheres, DOI: 10.1002/2013JD020418
- Cao, C., X. Shao, S. Uprety, (2013b), Detecting Light Outages After Severe Storms Using the Suomi-NPP/VIIRS Day Night Band Radiances, IEEE Geoscience and Remote Sensing Letters, DOI: 10.1109/LGRS.2013.2262258, in press.
- Liu, Q., C. Cao, and F. Weng, 2013, Assessment of Suomi National Polar-Orbiting Partnership VIIRS Emissive Band Calibration and Inter-Sensor Comparisons, IEEE JSTAR, 10.1109/JSTARS.2013.2263197.
- Liao. L.B., S. Weiss, S. Mills, B. Hauss (2013), Suomi NPP VIIRS Day-Night-Band (DNB) On-Orbit Performance, Journal of Geophysical Research-Atmosphere, DOI: 10 1002/2013 D020475

- Wolfe, R., G. Lin, M. Nishihama, K. P. Tewari, J. C. Tilton, A. R. Isaacman et al., 2013, Suomi NPP VIIRS prelaunch and on-orbit geometric calibration and characterization, DOI: 10.1002/jgrd.50873, JGR special issue , in press.
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- Xiong, X., J. Butler, K. Chiang, B. Efremova, J. Fulbright, N. Lei, J. McIntire, H. Oudrari, J. Sun, Z. Wang, A. Wu (2013), VIIRS Onorbit Calibration Methodology and Performance, Journal of Geophysical Research-Atmosphere, DOI: 10.1002/2013JD020423.
- Uprety, S., C. Cao, X. Xiong, S. Blonski, A. Wu, and X. Shao, 2013, Radiometric Inter-comparison between Suomi NPP VIIRS and Aqua MODIS Reflective Solar Bands using Simultaneous Nadir Overpass in the Low Latitudes, JTech , doi: http://dx.doi.org/10.1175/JTECH-D-13-00071.1.







Courtesy of N. Lei, VCST





- In the case of IDPS algorithms, we want the algorithm leads to provide 1 of 3 recommendations:
 - 1. NPOESS algorithm has evolved into the NOAA-endorsed JPSS algorithm and any needed improvements should continue.
 - 2. NPOESS (or evolved) algorithm will not meet requirements or effort is too large, replace with NOAA-endorsed JPSS algorithm
 - 3. NOAA-endorsed algorithm should be used even if NPOESS (or evolved) algorithm meets performance because of legacy, enterprise, blended products, and other considerations.
- For 2 or 3, present the alternative algorithm methodology description, algorithm performance against the level 2 supplement specification and any user assessments.





JPSS STAR Science Team Annual Meeting OMPS SDR Team

Xiangqian Wu OMSP SDR Lead May. 12, 2014










- OMPS SDR Team
- Products and Users
- Requirements and Performance
- Accomplishments
- Algorithms Evaluation
- Future Plans for J1
- Summary





PI Name	Organization	Primary Roles
Fred Wu	NOAA/STAR	Budget and coordination; Instrument and product performance monitoring; J1 code development; TVAC data analysis; SDR algorithm.
Glen Jaross	NASA	Instrument scientist; TVAC data acquisition and analysis; SDR algorithm.
Bhaswar Sen	NGAS	G-ADA test for IDPS operations; TVAC data analysis; SDR algorithm.
Maria Caponi	Aerospace	Algorithm changes coordination; DR and issues tracking
Daniel Cumpton	Raytheon	IDPS operations





- Products:
 - OMPS nadir mapper (NM) and nadir profiler (NP) earth view (EV) and calibration (CAL) SDR in both nominal and diagnostic mode.
- Users:
 - OMPS EDR Team
 - Wider and future users via CLASS



Requirements and Performance



Parameters	Specification/Prediction	On-Orbit Performance
	Value	
NT 10 0	- 20/ C 11 - 11	-0.460/
Non-linearity	< 2% full well	< 0.46%
Non-linearity Accuracy	< 0.2%	±0.2%
On-orbit Wavelength Calibration	< 0.01 nm	0.15-0.25 nm
Stray Light NM Out-of- Band + Out-of-Field Response	For $NM \le 2$	average < 2%
Intra-Orbit Wavelength	Allocation (flow down from	~ 0.02 nm
Stability	EDR error budget) = 0.02 nm	
SNR	1000	> 1000
Inter-Orbital Thermal	Allocation (flow down from	~0.02 nm
Wavelength Shift	EDR error budget) = 0.02 nm	
CCD Read Noise	60 –е RMS	< 25 —е RMS
Detector Gain	43 (for NP)	47 (for NP)
	46 (for NM)	51 (for NM)
Absolute Irradiance	< 7%	< 3%
Calibration Accuracy		in 300-310 nm: up to ~10 % for both NM and NP
Absolute Radiance	< 8%	< 5%
Calibration Accuracy		in 300-310 nm: up to ~6 % for NM and NP
Normalized radiance Calibration Accuracy	< 1%	< 1%





- Beta maturity March 2012
- Provisional maturity March 2013
- Validated maturity
 - Primary review Dec 2013
 - Delta review planned for June 2014
 - Improved stray light correction and wavelength registration, for both NM & NP.
 - CAL SDR transition to GRAVITE is on schedule.





- Algorithm Description:
 - OMPS has three sensors. NOAA is responsible for SDR of two sensors (NM & NP).
 - Each sensor is configured to acquire earth view (EV) or calibration (CAL) data, in either nominal or diagnostic mode.
 - IDPS processes nominal EV data only
 - Transition is underway to process CAL SDR at GRAVITE
 - To automate the use of CAL SDR in EV SDR processing at IDPS
 - To archive the CAL SDR at CLASS





- Validation Approach and Datasets
 - Primary validation by examination of SDR characteristics such as dark, linearity, SNR.
 - Further validation:
 - Characteristics of EV SDR
 - Characteristics of EDR
 - Comparison with other measurements (GOME-2, SBUV/2)
 - Comparison with RTM (CRTM, MLS)
- Performance vs. Requirements
 - See earlier slide





- Risks/Issues/Challenges
 - Develop modifications to accommodate J1 upper
 - Produce CAL SDR in Ground System
- Quality Monitoring:
 - In place, and being continuously improved.
- Recommendations: NPOESS algorithm has evolved into the NOAA-endorsed JPSS algorithm and any needed improvements should continue. Substantial changes are expected for J1.





- JPSS-1 Algorithm Milestones
 - May: Unit test for decompressor and aggregator
 - July: Integration of pre-processor into IDPS
 - Aug: functional test of LUTs
 - Sept: Accommodate sparse LUTs
 - Oct: integration test of LUTs with J1 code
 - Nov: delivery to STAR AIT
 - Dec: delivery to DPA





- Validation Strategies
 - Pre-launch
 - Functional verification of LUT from SCDB
 - Integration tests of new LUTs and the modified code
 - Post-launch
 - Examination of SDR characteristics such as dark, linearity, SNR.
 - Characteristics of EV SDR
 - Characteristics of EDR
 - Comparison with other measurements (GOME-2, SBUV/2)
 - Comparison with RTM (CRTM, MLS)







- OMPS EV SDR is expected to reach the Validated maturity in June
- OMPS CAL SDR transition to GRAVITE is on schedule despite the setbacks
- Tasks and schedule for J1 preparation are well defined. Risk is low for performance but moderate for schedule and cost.





Instrument Performance and Sensor Data Quality Long Term Monitoring (LTM) in STAR Integrated Cal/Val System (ICVS)

Ninghai Sun, Fuzhong Weng, Michael Grotenhuis, Xin Jin, Jason Choi, Wanchun Chen

Satellite Meteorology and Climatology Division Center for Satellite Applications and Research National Environmental Satellite, Data and Information Service









- Introduction to Instrument Performance and Sensor Data Quality Long-Term Monitoring (LTM) in STAR Integrated Cal/Val System (ICVS)
- STAR ICVS-LTM Modules and Anomaly Samples
 - S-NPP Spacecraft LTM
 - S-NPP ATMS LTM
 - S-NPP VIIRS LTM
 - S-NPP CrIS LTM
 - S-NPP OMPS LTM
- Path Forward

NOAA/NESDIS/STAR ICVS-LTM System











STAR ICVS-LTM Website



http://www.star.nesdis.noaa.gov/icvs



755

5



S-NPP Spacecraft LTM Parameters





108 Parameters Provided in Real Time/LTM for S-NPP Spacecraft



S-NPP Spacecraft Customized Datasets



STAR ICVS Long-Term Monitoring

5/12/2014 15:00 UTC

Instrument Status > NPP > Spacecraft

Slide Show of All Charts for Selected Date

Displaying the last 24 hours of instrument status, updated every three hours.

	Select a parameter:	Spacecraft Monitoring Parameters	Select a Date:
	Spacecraft Monitoring Parameters	< 🛛 Daily Scan Level Spacecraft Monitoring - Diary 🛛 🔽 >	< 05-12-2014 🔢 >
	S/C CDH-RF Comm Telemetry	Daily Scan Level Spacecraft Monitoring - Diary	
	S/C HBD Transmitter Telemetry	Daily Scan Level Spacecraft Monitoring - Telemetry	
	S/C Bus Critical Telemetry	Level Spacecraft Monitoring - Diary - 05-12-2014	
	S/C ADCS Housekeeping High Bate Telemetry		
_	S/C Orbit State Telemetry		
	P/C DEED Instrument Boular Control Telemetry		
	S/C DSEP Instrument Power Control Telemetry		
	S/C POMA Configuration Telemetry	28 Command RX 1 Basenlate Temperature	
	S/C Temperature Telemetry	29 Command RX 2 Baseplate Temperature	
	ATMS S/C Telemetry Temperature	30 Command Receiver 1 Signal Strength Derived	
	CrIS S/C Telemetry Temperature	31 Command Receiver 2 Signal Strength, Derived	
	VIIRS S/C Telemetry Temperature	32.Command Receiver 1 Loop Stress, Derived	
	OMPS S/C Telemetry Temperature	28.Command RX 1 Baseplate Temperature	
	CERES S/C Telemetry Temperature	29.Command RX 2 Baseplate Temperature	
Spacecr	Concerned Manifesting Descentation	30.Command Receiver 1 Signal Strength, Derived	
	Spacecrait Monitoring Parameters	31.Command Receiver 2 Signal Strength, Derived	
	1. Scan Year	32.Command Receiver 1 Loop Stress, Derived	
	2. Scan Julian Day of the year	33.Command Receiver 2 Loop Stress, Derived	
	3. Scan UTC second from the midnig	nt 34.Star Tracker Maximum residual	
	4. Scan Orbit number	35.Total System Momentum 1	
	5. PUMA Bus Voltage	36.Total System Momentum 2	
	6. PUMA Total Bus Current	37.Total System Momentum 3	
	7. PUMA Battery 1 Voltage	FORMAT=(I4,1X,I3,1X,I5,1X,I5,33(1X,F10.4))	
	8 PUMA Battery 2 Voltage	***************************************	***

2013 68 83560 7074 32.4800 26.2400 32.3418 32.4018 952.2000 970.9200 378.9400 -549.7661 -399.8575 -549.7661 0.2960 0.2880 0.2920 0.1620 -0.6480 -0.3240 -0.4860 370.7336 13.5563 15.9864 -0.0006 -0.0590 0.0000 10.3897 9.3147 -106.9608 -103.9399 2890.5425 2886.5098 0.0001 -1.3824 -1.8048 0.0064 2013 68 83561 7074 32.4800 26.2400 32.3418 32.4018 952.2000 970.9200 375.3022 -546.2799 -400.3122 550.3724 0.2960 0.2880 0.2920 0.0000 -0.6480 -0.4860 -0.4860 370.7336 13.5563 15.9864 0.0006 0.0596 0.0000 10.3897 9.3147 -106.9608 -103.9399 2890.5425 2886.5098 0.0001 -1.3696 -1.7984 0.0064 2013 68 83562 7074 32.4800 25.5800 32.3418 32.4018 952.2000 970.9200 375.6053 -546.8862 -404.8595 -555.2229 0.2960 0.2880 0.2920 0.1620 -0.6480 -0.4860 -0.4860 370.7336 13.5563 15.9864 -0.0006 -0.0590 0.0000 10.3897 9.3147 -106.9608 -103.9399 2890.5425 2886.5098 0.0001 -1.4144 -1.8048 -0.0192 2013 68 83563 7074 32.3200 26.2400 32.3418 32.4018 952.2000 970.9200 376.6664 -552.4945 -401.8280 552.4945 0.2960 0.2880 0.2920 0.0000 -0.8100 -0.3240 -0.6480 370.7336 13.5563 15.9864 0.0006 -0.0596 0.0000 10.3897 9.3147 -106.9608 -103.9399 2890.5425 2886.5098 0.0000 -1.3376 -1.8432 0.0640 2013 68 83564 7074 32.4800 26.2400 32.3418 32.4018 952.2000 970.9200 373.9380 -550.2209 -403.9500 -0.2960 0.2880 0.2920 0.0000 -0.8100 -0.4860 -0.4860 370.7336 13.5563 15.9864 -0.0006 -550.2209 0.0596 0.0000 10.3897 9.3147 -106.9608 -103.9399 2890.5425 2886.5098 0.0000 -1.4016 -1.7984 -0.0192 201368 83565 7074 32.3200 26.2400 32.3418 32.4018 952.2000 970.9200 375.3022 -550.5240 -404.5563 -550.5240 0.2960 0.2880 0.2920 0.0000 -0.6480 -0.4860 -0.4860 370.7336 13.5563 15.9864 0.0006 0.0000 10.3897 9.3147 -106.9608 -103.9399 2890.5425 2886.5098 0.0001 -1.3312 -1.7728 0.0000 0.0596

8. PUIVIA Battery 2 Voltage 9. PUMA Battery 1 Pressure 10.PUMA Battery 2 Pressure 11.Wheel 1 Speed/Direction 12.Wheel 2 Speed/Direction 13.Wheel 3 Speed/Direction 14.Wheel 4 Speed/Direction 15.Gyro (TARA) 1 Motor Current 16.Gyro (TARA) 2 Motor Current 17.Gyro (TARA) 3 Motor Current 18.Reaction Wheel 1 Motor Current 19.Reaction Wheel 2 Motor Current 20.Reaction Wheel 3 Motor Current 21.Reaction Wheel 4 Motor Current 22.System Pressure 23. Propulsion Deck -Z Temperature 24. Propulsion Tank Temperature (Gas Side) 25.Control Frame Rate X 26.Control Frame Rate Y 27.Control Frame Rate Z



S-NPP ATMS LTM Parameters







S-NPP ATMS Quality Flag 20







S-NPP ATMS Quality Flag 20





Normal

Gap

QF On

Lunar Intrusion in Space View - Channel 6 (15 Orbits)	Normal	Flag On
Gain Error - Channel 6 (15 Orbits)	Normal	Flag On
Calibration With Fewer Samples - Channel 6 (15 Orbits	;) Normal	Flag On
Insufficient Space View Samples - Channel 6 (15 Orbits	s) Normal	Flag On
Insufficient Blackbody View Samples - Channel 6 (15 C	O <mark>rbits) Normal</mark>	Flag Or
Spare - Channel 6 (15 Orbits)	Normal	Flag Or
Spare - Channel 6 (15 Orbits)	Normal	Flag On
Spare - Channel 6 (15 Orbits)	Normal	Flag Or

TDR Before/After PCT Update





O ATMOS

NORE

MENT OF



Descending



90 N 75 N 60 N 45 N 30 N 15 N EQ 15 S 30 S 45 S 60 S 75 S 90 S 90 W 30 E 120 E 150 E 180 E 180 W 150 W 120 W 60 W 30 W 0 60 E 90 E Κ

250

260

270

280

Gap

240

230

220









S-NPP ATMS LTM Parameters































S-NPP VIIRS Solar Eclipse

Annular

Saros 148



2014 Apr 29

06:05 TD

- Solar eclipse The northern edge of the shadow first touches down in Antarctica at 05:57:35 UTC. The instant of greatest eclipse occurs just six minutes later at 06:03:25 UTC.
- VIIRS global true color image is darken by the solar eclipse;
- VIIRS solar diffuser count for M12 to 16, I4 and I5 bands decreased at about 5:00 7:00 UTC;





S-NPP VIIRS NO_SYNC LTM













350+ CrIS RDR/SDR LTM Parameters Provided in STAR-ICVS





• Sketchy jump of re-sampling laser wavelength on Feb 03.



Laser drifted about 1.23 ppm < 2 ppm



 Unexpected updating of re-sampling laser wavelength on Feb 05.

NPP CrIS Laser Wavelength: Measured/Monitored/Resampling,02/05/2014



Created at 02/10/2014 - 20:32:25 UTC



Laser drifted about 1.23 ppm < 2 ppm





- IDPS SDR algorithm is updated from Mx8.0 to Mx8.2 on Feb 20
- The engineering packet is updated from v35 to v36 on the same day

NPP CrIS Laser Wavelength: Measured/Monitored/Resampling,02/20/2014



Created at 02/24/2014 - 15:28:38 UTC



S-NPP CrIS Channel Performance

ND ATMOS

NOAR



NESD



S-NPP CrIS SDR Bias



NORA NPP CrIS 670 cm⁻¹ BT (K) Observ. – CRTM Calc., 05/04/2014 NPP CrIS BT Observ. - Calc., 14.93 µm (670 cm⁻¹), Mapped, Ascending, 05/04/2014 Created at 05/06/2014 - 02:30:08 UTC 90N FOV1 75N 60N 45N 30N FOV2 15N EQ 15S FOV3 30S 45S 60S FOV4 75S 90S 150E 180E 180W 60W 30W 120E 150W 120W 90W 30F 90E NPP CrIS BT Observ. - Calc., 14.93 µm (670 cm⁴), Mapped, Descending, 05/04/2014 FOV5 90N 75N FOV6 60N 45N 30N 15N FOV7 EQ 15S 30S FOV8 45S 60S -2 75S FOV9 90S 120E 150E 180E 180W 150W 120W 90W 60W 30W 30F 60F 90E -2.00 -1.00 0.00 1.00 2.00 -3.00 3.00 0 10 12 Time, UTC 14 16 18 20 22 2 4 6 8


S-NPP OMPS LTM Parameters





»OMPS Product

Generated from PEATE² Data



• The OMPS calibration monitoring system detected anomalies in the NM and NP standard deviation of dark current from dark calibration images acquired on Feb. 9, 2014





S-NPP OMPS NM/NP Anomaly





NP Anomalous Calibration Image





Particles from solar activity? Cosmic rays?





Count

Counts

11/10/201 Orbit# 18 05/07/2012 Orbit# 2726 11/02/2012 Orbit# 5268

Date – UTC

04/30/2013 Orbit# 7807 10/26/2013 Orbit# 10352 04/23/2014 Orbit# 12887

NM Anomaly in PEATE Calibration SDR data on April 20th: "spike" for one column of CCD1 smear signal in calibration dark image Suomi NPP OMPS Nadir Mapper near Counts Standard Deviation. Left Side Transient found in two columns of High standard deviation for ed: 04/29/2014 - 02:17:28 UTC 35th NM nominal dark image from the smears corresponding to Most Recent 10 days Available the image with the transient April 20th Diagnostic Data Nominal Data 738 736 04/15/2014 Orbit# 12774 04/17/2014 Orbit# 12802 04/19/2014 Orbit# 12835 04/21/2014 Orbit# 12859 04/23/2014 Orbit# 12887 04/13/2014 Orbit# 12745 50 100 150 200 250 300 Column Date – UTC Smear transients do affect Earthview (EV) data Entire Record Effect is small Caused 0.13% and 0.07% error in dark current rates, which amounts to 0.39 and 0.21 EV count (out of ~230,000). OMPS team is working to fix the necessary filters.



Instrument Anomaly Notification



	□ From Ninghai Sun <nsun@orbit082l.orbit2.nesdis.noaa.gov> ☆</nsun@orbit082l.orbit2.nesdis.noaa.gov>	
 Notify through e-mails 	Subject ICVS warning message (2012-07-08 04:28:08 UTC)	12:28 AM
Hothy through e mans	To Ninghai Sun 😭	
 Provide instrument status 	NOAA-19 AMSU-A Channel 8 NSS.AMAX.NP.D12189.S2346.E0137.B1759495 HEdT out of spe Current = 0.736	▲ ecification !!!! Spec. = 0.2500
 Provide data availability 	NSS.AMAX.NP.D12189.S2204.E2352.B1759394 NEdT out of spe Current = 0.738	cification !!!! Spec. = 0.2500
	NSS.AMAX.NP.D12189.S2021.E2209.B1759293 IEdT out of spe Current = 0.743	cification !!!! Spec. = 0.2500
 Provide data quality 	Current = 0.738 NISS.AMAX.INP.D12189.51652.E1837.B1758991 HEdT out of spe	cification !!!! Spec. = 0.2500
	Current = 0.718 NSS.AMAX.NP.D12189.S1526.E1648.B1758989 HEdT out of spe Current = 0.677	cification !!!! Spec. = 0.2500
	NSS.AMAX.NP.D12189.S1346.E1532.B1758889 NEdT out of spe Current = 0.711	cification !!!! Spec. = 0.2500
□ From Ninghai Sun <nsun@rhw1049.star1.nesdis.noaa.gov>☆</nsun@rhw1049.star1.nesdis.noaa.gov>	INSS.AMAX.NP.D12189.S1205.E1352.B1758788 NEdT out of spectrum Current = 0.659 Current = 0.659	cification !!!! Spec. = 0.2500
Subject NPP ATMS Lunar Intrusion Detected (2014-05-09) 5/10/2014 8:5 To Ninghai.Sun@noaa.gov 😭	55 AM 155.4MAX.107.012189.51024.E1211.B1/58087 1Edit out of spe Current = 0.663 NSS.4MAX.10P.012189.50849.E1030.B1758586 1Edit out of spe	cification !!!! Spec. = 0.2500
NPP ATMS Space View Lunar Intrusion is Detected on 2014-05-09	Current = 0.686 NSS.AMAX.NP.D12189.S0706.E0854.B1758485 HEdT out of spe	ecification !!!! Spec. = 0.2500
Channel 2 Lunar Intrusion Affected Scan Numbers = 2854 Channel 3 Lunar Intrusion Affected Scan Numbers = 1251	NSS.AMAX.NP.D12189.S0519.E0712.B1758384 NEdT out of spe Current = 0.719	cification !!!! Spec. = 0.2500
Channel 4 Lunar Intrusion Affected Scan Numbers = 1259 Channel 5 Lunar Intrusion Affected Scan Numbers = 1260	NSS.AMAX.NP.D12189.S0330.E0525.B1758283 NEdT out of spe Current = 0.696	cification !!!! Spec. = 0.2500
Channel 6 Lunar Intrusion Affected Scan Numbers = 1264 Channel 7 Lunar Intrusion Affected Scan Numbers = 1260	Current = 0.689	cification :::: spec. = 0.2500
Channel 8 Lunar Intrusion Affected Scan Numbers = 1256	NOAA-19 AMSU-A Channel 7 NSS AMAX NP D12189 S2346 E0137 B1759495 UEdT within spe	
Channel 10 Lunar Intrusion Affected Scan Numbers = 1250 Channel 10 Lunar Intrusion Affected Scan Numbers = 1246	NSS.AMAX.IN.D.D12189.52204.E2352.B1759394 HEdT out of spe Current = 0.253	cification !!!! Spec. = 0.2500
Channel 12 Lunar Intrusion Affected Scan Numbers = 1228 Channel 12 Lunar Intrusion Affected Scan Numbers = 1228	NSS.AMAX.NP.D12189.S2021.E2209.B1759293 NEdT out of spe Current = 0.255	cification !!!! Spec. = 0.2500
Channel 15 Lunar Intrusion Affected Scan Numbers = 1218 Channel 15 Lunar Intrusion Affected Scan Numbers = 1117	NSS.AMAX.NP.D12189.S1832.E2027.B1/59092 HEdT out of spe Current = 0.266 NSS.AMAX.NP.D12189.S1652.E1837.B1758901 HEdT within spe	controation !!!! Spec. = 0.2500
Channel 15 Lunar Intrusion Affected Scan Numbers = 1147 Channel 17 Lunar Intrusion Affected Scan Numbers = 1234	NSS.AMAX.NP.D12189.S1526.E1648.B1758989 IEdT within spe NSS.AMAX.NP.D12189.S1346.E1532.B1758889 IEdT out of spe	cification . ccification !!!! Spec. = 0.2500
Channel 19 Lunar Intrusion Affected Scan Numbers = 606 Channel 18 Lunar Intrusion Affected Scan Numbers = 606	Current = 0.253 NSS.AMAX.NP.D12189.S1205.E1352.B1758788 HEdT within spe	ecification .
Channel 20 Lunar Intrusion Affected Scan Numbers = 601	Current = 0.250	cification !!!! Spec. = 0.2500
Channel 21 Lunar Intrusion Affected Scan Numbers = 607 Channel 22 Lunar Intrusion Affected Scan Numbers = 598	NSS.AMAX.NP.D12189.50849.E1030.B1758586 NEdT out of spe Current = 0.254	cification !!!! Spec. = 0.2500
!!! Please go to <u>http://www.star.nesdis.noaa.gov/icvs/status NPP ATMS.php</u> to check Lunar Intrusion geophysical location and duration time.	INSS.AMAX.INP.D12189.S0/00.E0054.81/58485 ILEdT Within Spe NISS.AMAX.NP.D12189.S0519.E0712.81758284 IEdT within spe IISS.AMAX.NP.D12189.S0330.E0525.B1758283 IEdT out of spe	ecification . ecification . ecification !!!! Spec. = 0.2500
	Current = 0.252 NSS.AMAX.NP.D12189.S0142.E0335.B1758182 NEdT out of spe Current = 0.252	cification !!!! Spec. = 0.2500
	NOAA-19 MHS Channel 3	

Current = 3.0732

NSS.MHSX.NP.D12189.S2346.E0137.B1759495 NEdT out of specification !!!! Spec. = 1.0000





- Support SDR teams to maintain high quality satellite data products for S-NPP
 - Improve anomaly notification function
 - Provide flexible customized datasets for analysis
 - Improve ICVS-LTM system execution efficiency
 - Provide support for more sensors and more parameters
 - Improve sensor absolute bias monitoring module
 - Improve function of current ICVS-LTM website
- Extend current ICVS-LTM to support future NOAA and collaborative satellite programs
 - JPSS
 - GOES-R
 - GCOM