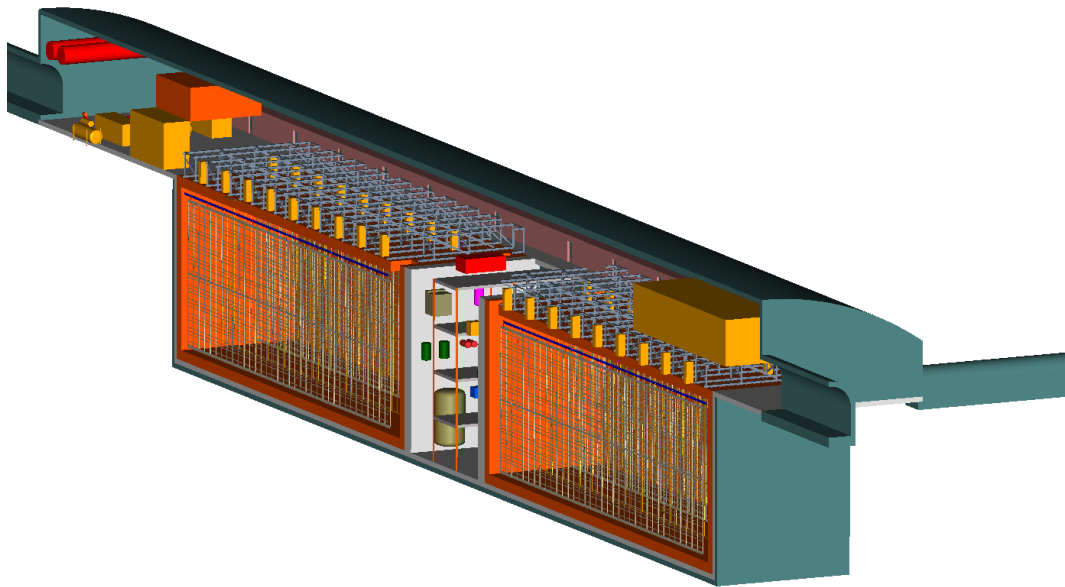


Long-Baseline Neutrino Experiment (LBNE) Project

Conceptual Design Report

Volume 4: The Liquid Argon Detector at the Far Site

March 15, 2012



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Acronyms, Abbreviations and Units

| | |
|----------|--|
| A | ampere (also mA, kA) |
| ADC | analog-to-digital converter |
| APA | anode plane assembly |
| APD | avalanche photodiodes |
| ArgoNeuT | Mini LArTPC Exposure to Fermilab's NuMI Beam |
| ASIC | application-specific integrated circuit |
| atm | atmosphere |
| B | byte (also MB, GB, etc.) |
| b | barn, a measure of cross section; bit (also Mb, Gb, etc.) |
| bar | bar (also mbar, etc.) |
| barg | bar gauge (deprecated per Wikipedia) |
| BGR | band-gap reference |
| Bis-MSB | 1,4-bis[2-(2-methylphenyl)ethenyl]-benzene, a wavelength-shifting chemical |
| Bq | becquerel |
| C | coulomb |
| CAD | computer-aided design |
| CDF | one of two decommissioned collider detectors at Fermilab's Tevatron, along with D-Zero |
| CDR | Conceptual Design Report |

| | |
|--------|--|
| CF | conventional facilities |
| cf | cubic foot (also ft ³) |
| CFD | computerized fluid dynamics |
| cfm | cubic feet per meter (also ft ³ /m) |
| Ci | curie |
| CP | charge parity |
| CPA | cathode plane assembly |
| D-Zero | one of two decommissioned collider detectors at Fermilab's Tevatron, along with CDF |
| DAQ | data acquisition |
| DC | direct current |
| DCM | Data concentrator modules |
| DCM | data concentrator module |
| DOE | Department of Energy |
| DUSEL | Deep Underground Science and Engineering Laboratory |
| EM | electromagnetic |
| ENC | equivalent noise charge |
| ESH | Environment, Safety and Health |
| eV | electron-volt (also keV, MeV, GeV) |
| F | farad (also pF, nF) |
| FEM | front-end module |
| FESHM | Fermilab's ES&H Manual |
| FFT | Fast Fourier Transform |
| FIRUS | Fire and Utilities; Fermilab site-wide, high-reliability, remote monitoring system used to monitor building fire panels and various utilities throughout the lab |
| FLARE | Fermilab Liquid Argon Experiments |
| FPGA | field-programmable gate array |

| | |
|---------|---|
| FR-4 | flame resistant 4 |
| FSE | field-shaping electrode |
| ft | foot or feet |
| G | gauss (also mG) or gradient (in magnets) |
| g | gram (also mg, kg) |
| G10/FR4 | a fire rated electrical-grade dielectric made with an epoxy material reinforced with a woven fiberglass mat |
| gal | gallon |
| GAr | gaseous argon |
| GLACIER | NASA's General Laboratory Active Cryogenic International Space Station Experiment Refrigerator |
| gpm | gallons per minute (also gal/min) |
| GTT | Gaztransport & Technigaz |
| h | hour |
| HSSD | High Sensitivity Smoke Detection |
| HV | high voltage |
| HVAC | heating, ventilation and air conditioning |
| Hz | hertz (s ⁻¹) |
| ICARUS | Imaging Cosmic and Rare Underground Signals, experiment at the LNGS |
| IHI | Ishikawajima-Harima Heavy Industries |
| in | inch |
| INFN | Istituto Nazionale della Fisica Nucleare |
| IT | integration prototype |
| IU | Indiana University |
| K | kelvin |
| l | liter |
| LANNDD | Liquid Argon Neutrino and Nucleon Decay Detector |

| | |
|------------|--|
| LAPD | Liquid Argon Purity Demonstrator |
| LAr | liquid argon |
| LAr-FD | LBNE's Liquid Argon Far Detector |
| LAr1 | one-kiloton LAr prototype for LBNE's LAr-FD |
| LArSoft | a reconstruction software package for LAr detectors |
| LArTPC | liquid argon time projection chamber |
| LBNE | Long-Baseline Neutrino Experiment |
| LED | light-emitting diode |
| LEM | large electron multiplier |
| LEMO | a push-pull connector made by the LEMO company in Switzerland |
| LHe | liquid helium |
| LN | liquid nitrogen, also written LN_2 and LN2 |
| LNG | liquefied natural gas |
| LNGS | Gran Sasso National Laboratory |
| LOTO | lockout/tagout; an OSHA safety practice |
| LPG | liquefied petroleum gas |
| LVDS | low-voltage differential signaling |
| m | meter (also nm, micron, mm, cm, km) |
| MCT | membrane cryostat test |
| MICA | type of dielectric material used in capacitors |
| MicroBooNE | A 100-ton LArTPC located along Fermilab's Booster neutrino beamline |
| min | minute |
| MINERvA | A neutrino-scattering experiment that uses the NuMI beamline at Fermilab |
| MINOS | Main Injector Neutrino Oscillation Search, a Fermilab experiment |
| MIP | minimum ionizing particle |
| MIT | Massachusetts Institute of Technology |

| | |
|---------|--|
| MOS | metal-oxide semiconductor |
| MOSFET | metal-oxide-semiconductor field-effect transistor |
| MS | mega samples |
| MTS | Materials Test Stand |
| MTU | master timing unit |
| MUX | multiplex |
| N | newton |
| N or N2 | nitrogen |
| NASA | National Aeronautics and Space Administration |
| NBTI | negative bias temperature instability |
| NC | neutral current |
| NICADD | Northern Illinois Center for Accelerator and Detector Development NOvA |
| NIM | Nuclear Instruments and Methods (journal) |
| NIST | National Institute of Standards and Technology |
| NOvA | NuMI Off-Axis Neutrino Appearance experiment at Fermilab |
| NPO | type of dielectric material used in capacitors |
| NSF | National Science Foundation |
| OD | outer diameter |
| ODH | oxygen deficiency hazard |
| OPERA | Oscillation Project with Emulsion-Racking Apparatus, at CERN and LNGS |
| OSHA | Occupational Safety and Health Administration |
| P5 | Particle Physics Project Prioritization Panel |
| Pa | pascal |
| PC | personal computer |
| PC-4 | Fermilab building, Proton Center building number 4 |
| PCI | peripheral component interconnect |

| | |
|--------|--|
| PDA | photon detection assembly |
| PDE | photon-detection efficiency |
| PE | photo-electron |
| PLC | programmable logic controller |
| PMT | photomultiplier tube |
| PPE | personnel protective equipment |
| PRV | pressure-relief valve |
| psi | pounds per square inch |
| QC | quality control |
| QFP | quad flat pack |
| R | roentgen |
| rad | radian (also mrad) |
| RC | resistive capacitive |
| ROOT | An object oriented framework for large-scale data analysis developed at CERN |
| s | second (also ns, μ s, ms) |
| S/N | signal-to-noise |
| scfm | standard cubic foot per minute |
| SCR | silicon controlled rectifier |
| SDSTA | South Dakota Science and Technology Authority |
| SHV | safe high voltage, a type of HV cable connector |
| SIMOPS | simultaneous operations study |
| SiPM | Silicon photomultiplier |
| SM | stress-migration |
| SS | stainless steel |
| T | Tesla |
| t | ton |

| | |
|------|---|
| TC | thermal cycling |
| TDDB | time-dependent dielectric breakdown |
| TDU | timing distribution unit |
| TPB | tetraphenyl butadiene, a wavelength shifting chemical |
| TPM | TPC module |
| USB | universal serial bus |
| UV | ultraviolet |
| V | volt (also mV, kV, MV) |
| VA | volt-ampere (also mVA, kVA, MVA) |
| VAC | Volts alternating current (also mVAC, kVAC) |
| VME | a computer bus standard |
| VUV | vacuum ultraviolet light |
| VUV | vacuum ultraviolet |
| W | watt (also mW, kW, MW) |
| WARP | Wimp Argon Program |
| WBS | Work Breakdown Structure |
| WLS | wavelength shifter |
| yd | yard |

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1 Introduction

1.1 Introduction to the LBNE Project

The Long-Baseline Neutrino Experiment (LBNE) Project team has prepared this Conceptual Design Report (CDR) which describes a world-class facility to enable a compelling research program in neutrino physics. The ultimate goal in the operation of the facility and experimental program is to measure fundamental physical parameters, explore physics beyond the Standard Model and better elucidate the nature of matter and antimatter.

Although the Standard Model of particle physics presents a remarkably accurate description of the elementary particles and their interactions, it is known that the current model is incomplete and that a more fundamental underlying theory must exist. Results from the last decade, revealing that the three known types of neutrinos have nonzero mass, mix with one another and oscillate between generations, point to physics beyond the Standard Model. Measuring the mass and other properties of neutrinos is fundamental to understanding the deeper, underlying theory and will profoundly shape our understanding of the evolution of the universe.

1.1.1 About this Conceptual Design Report

The LBNE Conceptual Design Report is intended to describe, at a conceptual level, the scope and design of the experimental and conventional facilities that the LBNE Project plans to build to address a well-defined set of neutrino-physics measurement objectives. At this Conceptual Design stage the LBNE Project presents a *Reference Design* for all of the planned components and facilities, and alternative designs that are still under consideration for particular elements. The scope includes

- an intense neutrino beam aimed at a far site
- detectors located at the near site just downstream of the neutrino source

- a massive neutrino detector located at the far site
- construction of conventional facilities at both the near and far sites

The selected near and far sites are Fermi National Accelerator Laboratory (Fermilab), in Batavia, IL and Sanford Underground Laboratory at Homestake (Sanford Laboratory), respectively. The latter is the site of the formerly proposed Deep Underground Science and Engineering Laboratory (DUSEL) in Lead, South Dakota.

This CDR is organized into six stand-alone volumes, one to describe the overall LBNE Project and one for each of its component subprojects:

- Volume 1: The LBNE Project
- Volume 2: The Beamline at the Near Site
- Volume 3: Detectors at the Near Site
- Volume 4: The Liquid Argon Detector at the Far Site
- Volume 5: Conventional Facilities at the Near Site
- Volume 6: Conventional Facilities at the Far Site

Volume 1 is intended to provide readers of varying backgrounds an introduction to LBNE and to the following volumes of this CDR. It contains high-level information and refers the reader to topic-specific volumes and supporting documents, also listed in Section 1.1.5. Each of the other volumes contains a common, brief introduction to the overall LBNE Project, an introduction to the individual subproject and a detailed description of its conceptual design.

1.1.2 LBNE and the U.S. Neutrino-Physics Program

In its 2008 report, the Particle Physics Project Prioritization Panel (P5) recommended a world-class neutrino-physics program as a core component of the U.S. particle physics program [3]. Included in the report is the long-term vision of a large detector at the Sanford Laboratory and a high-intensity neutrino source at Fermilab.

On January 8, 2010, the Department of Energy (DOE) approved the Mission Need for a new long-baseline neutrino experiment that would enable this world-class program and firmly establish the U.S. as the leader in neutrino science. The LBNE Project is designed to meet this Mission Need.

With the facilities provided by the LBNE Project, the LBNE Science Collaboration proposes to mount a broad attack on the science of neutrinos with sensitivity to all known parameters in a single experiment. The focus of the program will be the explicit demonstration of leptonic CP violation, if it exists, by precisely measuring the asymmetric oscillations of muon-type neutrinos and antineutrinos into electron-type neutrinos and antineutrinos.

The experiment will result in the most precise measurements of the three-flavor neutrino-oscillation parameters over a very long baseline and a wide range of neutrino energies, in particular, the CP-violating phase in the three-flavor framework. The unique features of the experiment – the long baseline, the broad-band beam, and the high resolution of the detector – will enable the search for new physics that manifests itself as deviations from the expected three-flavor neutrino-oscillation model.

The configuration of the LBNE facility, in which a large neutrino detector is located deep underground, could also provide opportunities for research in other areas of physics, such as nucleon decay and neutrino astrophysics, including studies of neutrino bursts from supernovae occurring in our galaxy. The scientific goals and capabilities of LBNE are outlined in Volume 1 of this CDR and described fully in the LBNE Case Study Report (Liquid Argon TPC Far Detector) [4] and the 2010 Interim Report of the Long-Baseline Neutrino Experiment Collaboration Physics Working Groups [5].

1.1.3 LBNE Project Organization

The LBNE Project Office at Fermilab is headed by the Project Manager and assisted by the Project Engineer, Project Systems Engineer and Project Scientist. Project Office support staff include a Project Controls Manager and supporting staff, a Financial Manager, an Environment, Safety and Health (ES&H) Manager, a Computing Coordinator, Quality Assurance and Risk Managers, a documentation team and administrative support. The Beamline, Liquid Argon Far Detector and Conventional Facilities subprojects are managed by the Project Office at Fermilab, while the Near Detector Complex subproject is managed by a Project Office at Los Alamos National Laboratory (LANL).

More information on Project Organization can be found in Volume 1 of this CDR. A full description of LBNE Project management is contained in the LBNE Project Management Plan [6].

1.1.4 Principal Parameters of the LBNE Project

The principal parameters of the major Project elements are given in Table 1-1.

Table 1-1: LBNE Principal Parameters

| Project Element Parameter | Value |
|---|---|
| Near- to Far-Site Baseline | 1,300 km |
| Primary Proton Beam Power | 708 kW, upgradable to 2.3 MW |
| Protons on Target per Year | 6.5×10^{20} |
| Primary Beam Energy | 60 – 120 GeV (tunable) |
| Neutrino Beam Type | Horn-focused with decay volume |
| Neutrino Beam Energy Range | 0.5 – 5 GeV |
| Neutrino Beam Decay Pipe Diameter \times Length | 4 m \times 200 m |
| Near Site Neutrino Detector Type | Liquid Argon Time Projection Chamber (LArTPC) Tracker |
| Near Site Neutrino Detector Active Mass | 18 ton |
| Far Detector Type | LArTPC |
| Far Detector Active (Fiducial) Mass | 40 (33) kton |
| Far Detector Depth | 1480 m |

1.1.5 Supporting Documents

A host of information related to the CDR is available in a set of supporting documents. Detailed information on risk analysis and mitigation, value engineering, ES&H, costing, project management and other topics not directly in the design scope can be found in these documents, listed in Table 1-2. Each document is numbered and stored in LBNE's document database, accessible via a username/password combination provided by the Project. Project documents stored in this database are made available to internal and external review committees through Web sites developed to support individual reviews.

Table 1-2: LBNE CD-1 Documents

| Title | LBNE Doc Number(s) |
|---|--------------------|
| Acquisition Plan | 5329 |
| Alternatives Analysis | 4382 |
| Case Study Report; Liquid Argon TPC Detector | 3600 |
| Configuration Management Plan | 5452 |
| DOE Acquisition Strategy for LBNE | 5442 |
| Integrated Environment, Safety and Health Management Plan | 4514 |
| LAr-FD Preliminary ODH Analysis | 2478 |
| Global Science Objectives, Science Requirements and Traceback Reports | 4772 |
| Preliminary Hazard Analysis Report | 4513 |

| | |
|--|-------------|
| Preliminary Project Execution Plan | 5443 |
| Preliminary Security Vulnerability Assessment Report | 4826 |
| Project Management Plan | 2453 |
| Project Organization Chart | 5449 |
| Quality Assurance Plan | 2449 |
| Report on the Depth Requirements for a Massive Detector at Homestake | 0034 |
| Requirements, Beamline | 4835 |
| Requirements (Parameter Tables), Far Detector | 3747 (2843) |
| Requirements, Far Site Conventional Facilities | 4408 |
| Requirements, Near Detectors | 5579 |
| Requirements, Near Site Conventional Facilities | 5437 |
| Risk Management Plan | 5749 |
| Value Engineering Report | 3082 |
| Work Breakdown Structure (WBS) | 4219 |

1.2 Introduction to the Liquid Argon Far Detector

1.2.1 Overview

The reference design Far Detector for LBNE is a liquid argon time projection chamber (LArTPC). The basic components of this detector include a cryostat to contain the liquid argon (LAr), a TPC detection mechanism immersed in the LAr, readout electronics and a cryogenic system to keep the LAr temperature at 89 K and maintain the required purity.

The LBNE LArTPC, referred to as the LAr-FD, consists of two massive cryostats in a single cavern, oriented end-to-end along the beam direction (roughly east-to-west), and located at the 4850 level (4850L) of the Sanford Laboratory. The fiducial mass of each, as defined for neutrino-oscillation studies, is 16.3 kton and the active (instrumented) mass is 20 kton, resulting in a total active mass of 40 kton. Figure 1-1 shows the proposed layout of the Far Site, and Figure 1-2 shows the detector configuration. The cavern and related conventional facilities are described in Volume 6 of this CDR.

In an LArTPC a uniform electric field is created within the TPC volume between cathode planes and anode wire planes. Charged particles passing through the TPC release ionization electrons that drift to the anode wire planes. The bias voltage is set on the anode plane

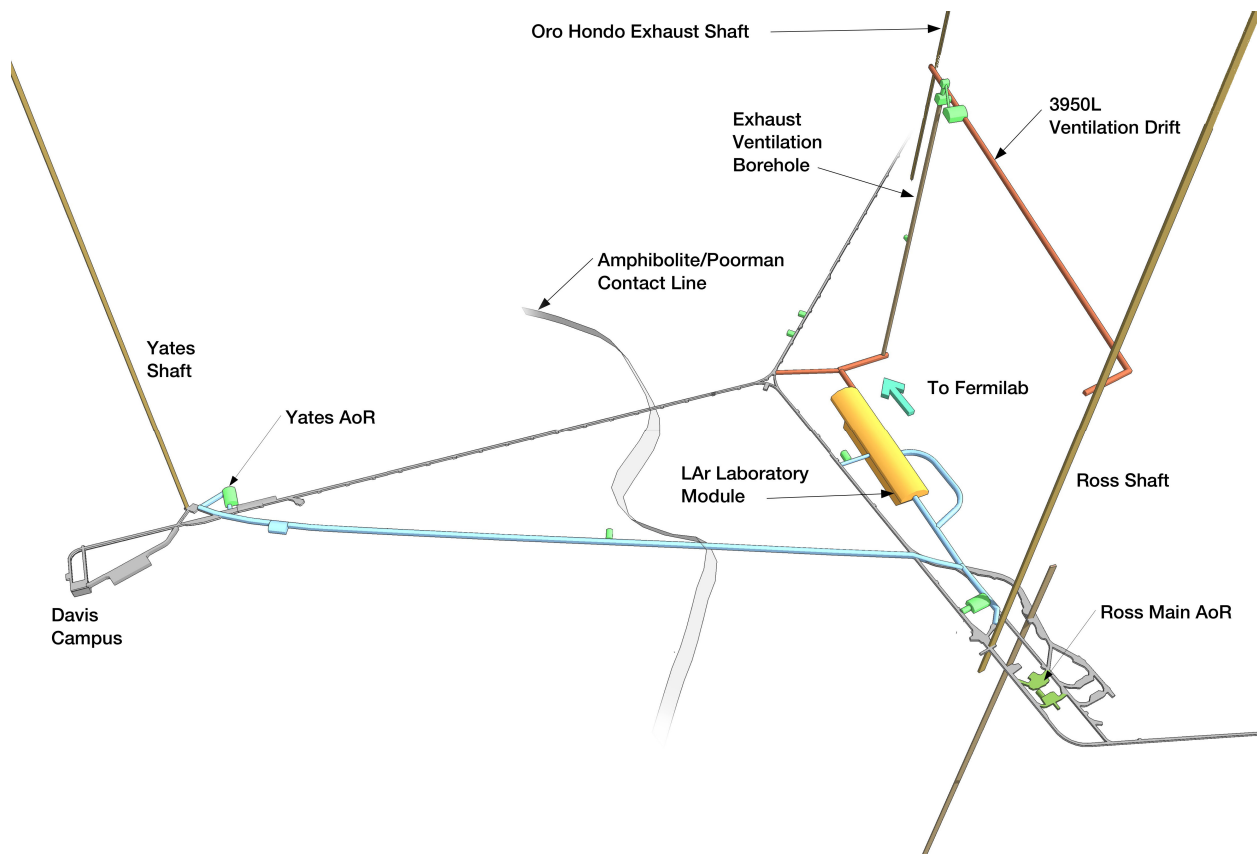


Figure 1-1: Location of LAr-FD at the 4850L (Dangermond Keane Architecture, courtesy Sanford Laboratory)

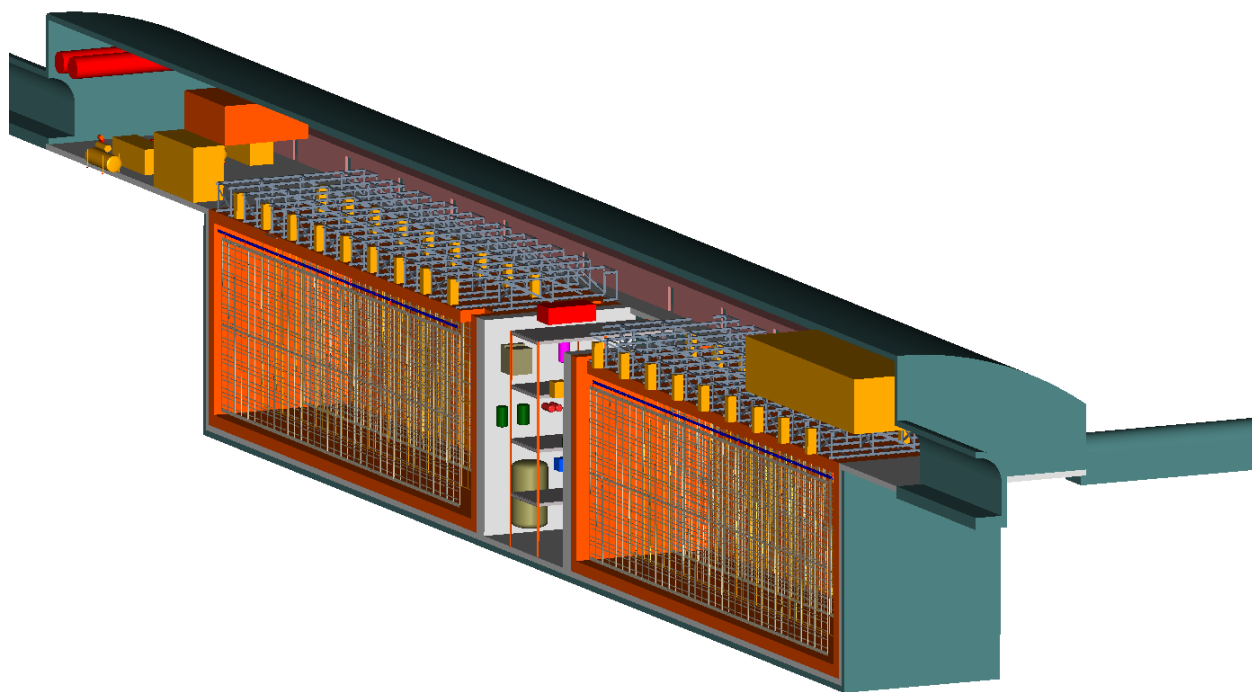


Figure 1–2: Detector configuration within the cavern. The TPC is located within a membrane cryostat, shown in orange. The interior dimensions of each cryostat are 24 m wide \times 18 m high \times 51 m long. The highbay is 150 m long and has a 32 m span. Cryogenic equipment is located in the septum area between the two cryostats.

wires so that ionization electrons drift between the first several (induction) planes and are collected on the last (collection) plane. Readout electronics amplify and continuously digitize the induced waveforms on the sensing wires at several MHz, and transmit these data to the data acquisition (DAQ) system for processing. The wire planes are oriented at different angles allowing a 3D reconstruction of the particle trajectories. In addition to these basic components, a photon-detection system provides a trigger for proton decay and galactic supernova neutrino interactions.

The design of the LAr-FD has been developed and refined over the past three years. The starting point was the ICARUS T600 system [7], and the process was informed and guided by the experience with small LArTPCs in the U.S., particularly ArgoNeuT [8], the development of designs for MicroBooNE [9]. The LAr-FD concept is designed for assembly from small, independent elements which can be repeated almost indefinitely in any dimension to form the entire assembly within a large cryostat. Each of the elements provides an independent mechanical structure to support the elements it contains. To a large extent, scaling from detector volumes containing anywhere from a few to several hundred such elements is straightforward with small and predictable risk. Scaling in other areas of LArTPC detector technology, namely cryostat construction, LAr purification and electronics readout has been incorporated into the design.

1.2.2 Cavern Layout

The LAr-FD cavern will be excavated into a “mailbox” cross section, as illustrated in Figure 1-2. The long axis of the cavern will point towards Fermilab, parallel to and in line with the neutrino beamline. The pit, the below-grade rectangular cavity inside of which the cryostats fit, will have dimensions of 26.1-m wide by 18-m high by 119.2-m long. Dimensions are detailed in Figure 1-3.

The pit will be segmented into three volumes: a space for each cryostat plus a 15-m-wide clear space in the middle, called the septum, separated from the cryostat areas by 1.5-m-thick walls. Equipment and vessels used for filtering LAr to achieve high purity levels will be placed in the septum area.

The cavern roof, located above the pit, will be 33 m wide by 163 m long, and its rounded “mailbox top” upper surface will arch to ~ 9.5 m in the center and ~ 5 m on the sides. The curved upper surface will extend beyond each end of the dual-cryostat detector along the long axis to provide additional space for ancillary equipment as well as for installation and maintenance operations.

Space will be provided for the nitrogen refrigerators, nitrogen compressors and necessary cryogen-storage vessels. Walkways will run the entire length of the cavern on each side of the detector to provide personnel access. A separately ventilated emergency-egress passageway

will be located on one side of the cavern. The cryostat cover will be designed to allow easy access to the emergency passageway from the other side of the cavern. One walkway will be 3 m wide to provide sufficient clearance to transport equipment from one end of the cavern to the other.

1.2.3 Cryostat Construction

The cryostat construction uses commercial stainless-steel membrane technology engineered and produced by industry. These vessels are widely deployed in liquefied natural gas (LNG) tanker ships and tanks, and are typically manufactured in sizes much larger than that of the LAr-FD. This is an inherently clean technology with passive insulation.

The LAr-FD cryostat reference design was selected on the recommendation of the experienced engineering consultants from ARUP USA, Inc. [10] after consideration of an alternative design that uses segmented, internally self-supporting, evacuable, modular cryostats. Evacuation, in particular, appears not to be necessary; in September 2011, the Fermilab Liquid Argon Purity Demonstrator (LAPD) achieved purity levels of less than 100 ppt oxygen-equivalent, using the method that is planned for use in the LAr-FD. This confirms that the method works, obviating the risk to LBNE that an evacuable vessel will be required. Operation of a 35-ton prototype using membrane-cryostat technology will provide a further demonstration.

The LAr-FD membrane cryostats are sealed containers supported by the surrounding rock. This “in-ground” configuration offers access only from the top and protects against possible cryogen leaks out of the tank. The side walls consist of a series of membranes, foam insulation and reinforced concrete poured against the shotcrete covered rock. The inner (primary) membrane liner, made of stainless steel, is corrugated to provide strain relief from temperature-related expansion and contraction.

1.2.4 Cryogenic Systems

The LAr must be initially transferred to the cryostats and must be kept cold, pure and circulating smoothly during operations in order to maintain a sufficiently long drift lifetime for the ionization electrons. The major cryogenic systems used to perform these functions include the cryogen supply for cool-down and fill, gas filtration, argon condensing, liquid filtration and circulation and argon purity analysis.

The overall cryogenic system’s layout and location is intended to optimize safety and efficiency. It is designed to minimize:

- the risk of personnel injury to any Oxygen Deficiency Hazard (ODH)

- heat ingress to the cryogenic system (by minimizing piping length and pump power)
- the volume of the argon system external to the cryostat and hence the potential for argon escape or contamination

It is also designed to provide safe access to refrigeration equipment that requires periodic maintenance.

The re-condensers and purifiers will be located underground, adjacent to the cryostat. A surface facility will be located at the Oro Hondo Fan site. The surface facility will include a 50-m³ LAr receiving dewar and a 50-m³ LN dewar. Three 75-kW liquid nitrogen (LN) refrigerators (two operating and one spare) and a 50-m³ LN dewar will be located in the cavern. Risers and other pipework linking the cavern to the surface are routed through a new 4.0 m diameter vertical exhaust ventilation borehole (shown in Figure 1-1) that connects to the existing Oro Hondo exhaust shaft. The shaft will contain LN supply and return pipes, an LAr pipe, vent piping and control-system wiring.

The required flow rate of liquid argon to be sent for purification is expected to decrease over time. The initial maximum flow rate will be 136 m³/hr (600 gpm) from each cryostat, resulting in a complete volume turnover every five days. Longer term, the rate will decrease to 34 m³/hr with a turnover time of 20 days. As point of comparison, ICARUS T600 has a maximum turnover rate of eight to ten days.

1.2.5 LAr Purification

The purification of LAr is accomplished with standard industrial equipment, using molecular sieves and chemically reducing materials, which are scalable within the contemplated range to accommodate the estimated irreducible material-outgassing from warm materials in the vapor space above the liquid argon, called the ullage.

1.2.6 Time Projection Chamber

The Time Projection Chamber (TPC) is the active detection element of the LAr-FD. The construction concept is shown schematically in Figure 1-4. The TPC is located inside the cryostat vessel and is completely submerged in LAr at 89 K. Its active volume is 14 m high, 22.4 m wide and 45.6 m long in the beam direction. It has four rows of Cathode Plane Assemblies (CPA) planes interleaved with three rows of Anode Plane Assemblies (APA) planes that are oriented vertically, parallel to the beamline, with the electric field applied perpendicular to the planes. The maximum electron-drift distance between a cathode and an adjacent anode is 3.7 m. Both the cathode and anode plane assemblies are 2.5 m wide and 7 m high. Two 7-m modules (either APA or CPA) stack vertically to instrument the

14-m active depth. In each row, 18 such stacks are placed edge-to-edge along the beam direction, forming the 45.6-m active length of the detector. Each cryostat houses a total of 108 APAs and 144 CPAs. A “field cage” surrounds the top and ends of the detector to ensure uniformity of the electric field. The field cage is assembled from panels of FR-4 sheets with parallel copper strips connected to resistive divider networks.

Each APA has three wire planes that are connected to readout electronics; two induction planes (labeled U and V in Figure 1-4) and one collection plane (X). A fourth wire plane, grid plane (G), is held at a bias voltage but is not instrumented with readout electronics. The grid plane improves the signal-to-noise ratio on the U plane and provides electrostatic discharge protection for the readout electronics.

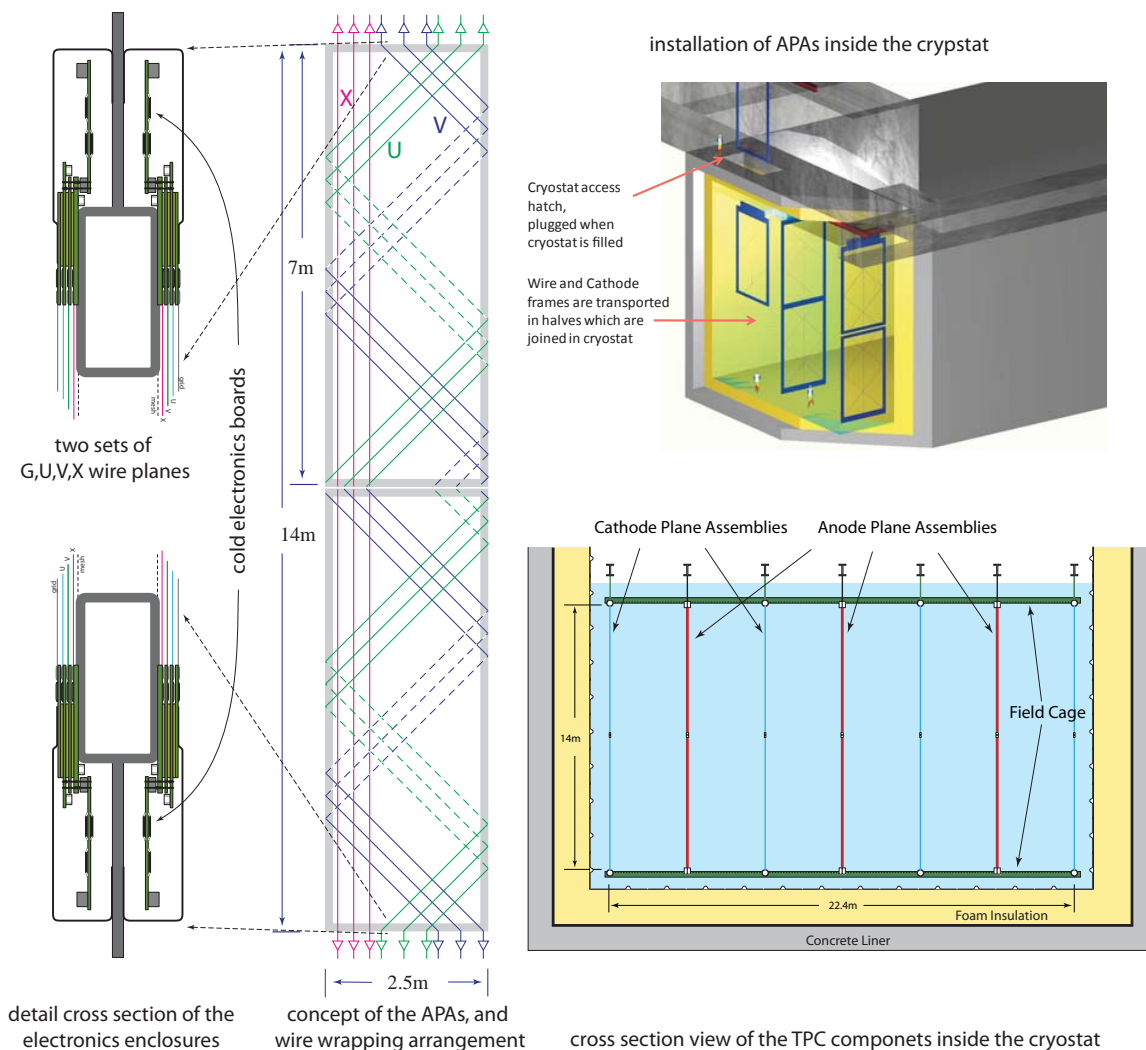


Figure 1-4: TPC modular construction concept

1.2.7 Electronics, Readout and Data Acquisition

Requirements for low noise and for extreme purity of the LAr motivate locating the front-end electronics in the LAr (hence “cold electronics”) close to the anode wires, which reduces the signal capacitance (thereby minimizing noise). The use of CMOS electronics in this application is particularly attractive since the series noise of this process has a noise minimum at 89 K. The large number of readout channels required to instrument the LAr-FD TPCs motivates the use of CMOS ASICs. Signal zero-suppression and multiplexing will be implemented in the ASIC, minimizing the number of cables and feedthroughs in the ullage gas, and therefore reducing contamination from cable outgassing. Figure 1-5 shows the conceptual architecture of a front-end electronics design that meets the requirements for LAr-FD. The entire electronics chain is immersed in the LAr.

All signal feedthroughs will be placed at the top of the cryostat, where they are easily installed, are always accessible, are at low hydrostatic pressure and pose no risk of LAr leakage. The cold electronic system will include digitization, buffering, and a high level of digital output multiplexing (ranging from 1/128 to 1/1024). Output data links will include redundancy to eliminate the effect of any single-point failure.

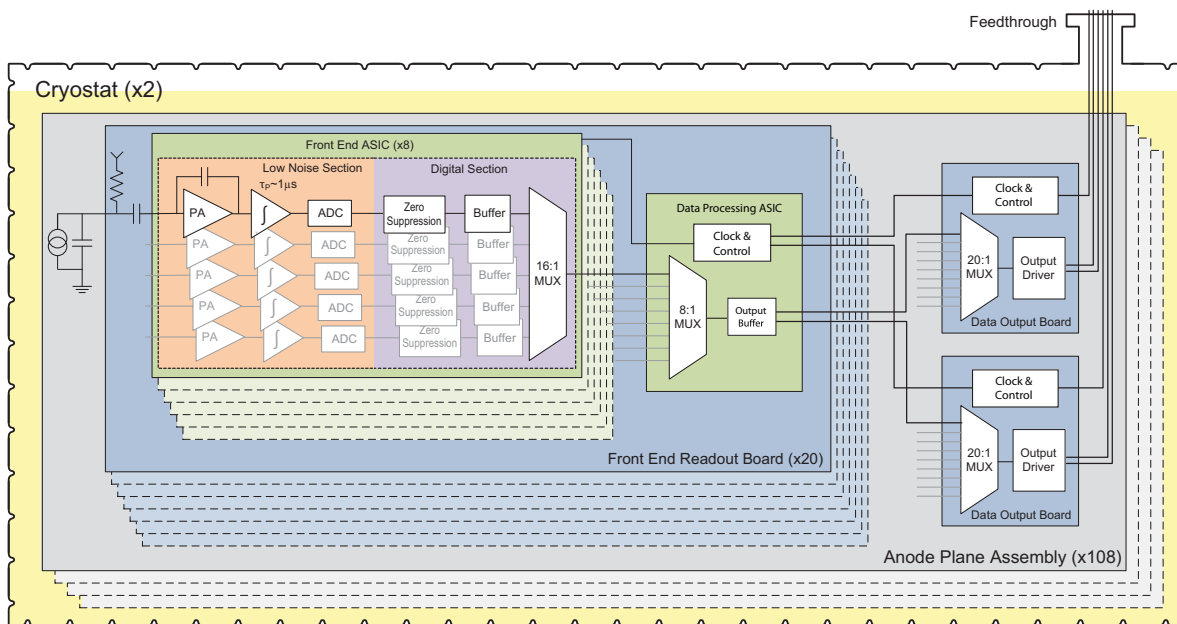


Figure 1-5: Conceptual front-end electronics architecture

1.2.8 Photon-Detection System

Identification of the different possible charged-particle types depends on accurate measurements of ionization along tracks. This requires accurate determination of the time of interaction, or event time, t_e , which leads to the absolute location of the event along the drift axis, and allows the determination of Q_0 , the true ionization charge.

For non-accelerator physics events, t_e is not known a priori. However, LAr is an excellent scintillator, generating of order 10^4 128-nm photons per MeV of deposited energy. Detection of scintillation photons provides a prompt signal that allows unambiguous location of particle positions along the drift axis.

The photon-detection system consists of acrylic light-guides that lead to small photomultiplier tubes (PMTs), as shown in Figure 5-2. Approximately 5% of the converted photons incident on a light guide are captured within it and travel to the PMT. A wavelength-shifting coating on the light guides efficiently converts the scintillation photon wavelength from 128 nm to 428 nm where the PMT is most sensitive. The fast PMT signals will be routed out of the cryostat to standard readout electronics.

Ten light-guide and PMT assemblies, or “paddles”, will be installed within each APA frame prior to wire winding. The PMT signals will be used as a software “trigger” in the DAQ to define the event time, t_e , for non-accelerator events. This system provides a t_e signal throughout the entire detector in contrast to a system similar to that used in MicroBooNE and ICARUS, where light is detected outside the detector volume.

1.2.9 Detector Installation and Operation

Detector components will be shipped in sealed containers to the Far Site by truck and lowered to the cavern. Shipping containers will be moved to a clean area over the septum area between the cryostats where components will be lowered through an access hatch into each cryostat.

The construction of the two cryostats and the installation and commissioning activities will be staged such that both TPCs can be tested cold while one cryostat still remains available as a potential LAr storage vessel. The LAr in one cryostat can be transferred to the other, and back again, if necessary, until all the tests complete successfully. Once both TPCs are known to work properly at LAr temperature, the second fill will take place.

To protect the membrane on the floor of the cryostat during TPC installation, a temporary floor will be installed. After each pair of APAs is installed, they will be connected to the DAQ system and the wire integrity tested. All wires on previously installed APA pairs will also be tested. The wire integrity test will be performed during cryostat cool-down as well. A

relatively slow cool-down rate will ensure that the temperature-induced stresses in the APA frames and wires are kept well below the level experienced during testing.

An installation and integration detector mock-up will be constructed at Fermilab to confirm that interfaces between detector systems are well defined and to refine the installation procedures.

1.2.10 The LAr1 One-kiloton Detector Prototype

The LBNE LAr-FD subproject will include the construction and operation of a 1-kton LArTPC prototype cryostat in order to establish a credible basis for scaling to a 40-kton size and to further develop the LAr-FD design. The LAr1 engineering prototype will be built during the CD-1 to CD-2 (preliminary design) phase of the LBNE project. LAr1 will include a representative sample of Anode Plane Assemblies (APAs) and Cathode Plane Assemblies (CPAs) in a membrane-style cryostat. The APA read-out chain with cold electronics, cable routing, and data acquisition (DAQ) system will be a scaled-down version of the full LBNE LAr-FD, however many features of LAr1 will be at a 1:1 scale. Analysis software will be used to reconstruct cosmic-ray events. This prototype is discussed in Chapter 7.

1.3 Principal Parameters

The principal parameters of the LAr-FD are given in Table 1-3.

Table 1-3: LAr-FD Principal Parameters

| Parameter | Value |
|---|---|
| Active (Fiducial) Mass | 40 (33) kton |
| Number of Detector Modules (Cryostats) | 2 |
| Drift Cell Configuration within Module | 3 wide \times 2 high \times 18 long drift cells |
| Drift Cell Dimensions | 2 \times 3.7 m wide (drift) \times 7 m high \times 2.5 m long |
| Detector Module Dimensions | 22.4 m wide \times 14 m high \times 45.6 m long |
| Anode Wire Spacing | \sim 5 mm |
| Wire Planes (Orientation from vertical) | Grid (0°), Induction 1 (45°), Induction 2 (-45°), Collection (0°) |
| Drift Electric Field | 500 V/cm |
| Maximum Drift Time | 2.3 ms |

1.3.1 Design Considerations

TPCs operated to date have been constructed with an anode wire spacing in the range of 3 mm (ICARUS) to 4.8 mm (Fermilab cosmic-ray stand). The amount of ionization charge collected on the wires increases with larger wire spacing, resulting in a better signal-to-noise ratio without serious consequences (the radiation length of LAr is ~ 30 times larger than the typical wire spacing). The electron- π^0 separation efficiency of a TPC with 5-mm wire spacing is only a few percent lower than one with 3-mm wire spacing. It is also clear that a TPC with larger wire spacing requires fewer wires and readout channels, resulting in lower cost.

Only two wire planes are required to reconstruct events in three dimensions, however three wire planes will be implemented to provide N+1 redundancy. The third will improve the pattern-recognition efficiency for a subset of multi-track events in which trajectories can overlap in two views. The collection-plane wires are most commonly used for calorimetric reconstruction and are oriented vertically (0°) to minimize both the wire length and the electronics noise.

A study of wire orientation has shown that for a TPC with three instrumented wire planes, the optimum orientation of the induction plane wires should be between $\pm 40^\circ$ and $\pm 60^\circ$ when the collection plane wires are at 0° . The ideal orientation for the more isotropic low-energy events, e.g., supernova-neutrino interactions, is $\pm 60^\circ$. The selected induction-plane wire orientation of $\pm 45^\circ$ has better position resolution in the vertical direction than $\pm 30^\circ$ and has shorter wires compared to a wire orientation of $\pm 60^\circ$. The induction plane wires are wrapped around the APA frames so that the readout electronics can be located on the top or bottom of the TPC. As a result, it is natural to arrange the APAs vertically in a two-high configuration.

Access to the top of the cryostat is required to install and connect cabling. Therefore, risk of personnel injury and detector damage, both of which increase with height, form the primary considerations for the the detector height, 14 m. The The height of the APA has been chosen, accordingly, to be 7 m, resulting in 7-m-long collection-plane wires and 10-m-long induction plane wires.

The 2.5 m width of the APA was chosen to facilitate construction and to allow standard, over-the-road transport.

The choice of cryostat width is based on the desired cryostat shape and cavern span. From a cryogenics standpoint, the ideal cryostat for a modular TPC would be a cube since membrane-cryostat capital and operating costs scale linearly with the surface area. This shape is not ideal for cavern excavation, however. In the absence of a geotechnical investigation for the cavern location, the cavern span has been limited to ~ 30 m on the advice of rock engineers. A detector width of 22.4 m results after making allowance for cryostat insulation,

and personnel access both above and within the cryostat.

A drift field of 500 V/cm was chosen based on experience from similar detectors such as ICARUS, ArgoNeuT and the Fermilab cosmic-ray test stand. At this electric field, $\sim 30\%$ of the ionization electrons produced by the passage of a minimum ionizing particle (MIP) recombine and create scintillation light that provides a fast trigger. The remaining ionization electrons drift to the APA and produce wire-plane signals. The TPC could function at higher or lower drift fields but the relative yields of scintillation light and ionization electrons would change. The use of a higher drift field would require more care in the design of the high-voltage systems. The electron drift velocity is 1.6 mm/ μ s at 500 V/cm.

The 22.4 m width of the the detector can be divided into $2 \times N$ drift cells, resulting in a range of maximum drift lengths of 11.2 m (N=1), 5.6 m (N=2), 3.7 m (N=3), 2.8 (N=4), and so on. The maximum drift cell length of 3.7 m was chosen based on experience from other detectors, the required minimum signal-to-noise ratio and high-voltage considerations. The required minimum signal-to-noise ratio of 9:1 ensures that the tracking efficiency will be 100% throughout the entire drift cell. The TPC must be capable of detecting the smallest signal (1 MeV) produced in interactions that LBNE will study. This situation occurs when a MIP travels parallel to a wire plane and perpendicular to the orientation of the wires in the plane. A MIP loses 2.1 MeV of energy in each cm of travel, producing $\sim 40,000$ ionization electrons along every 5 mm section of the track. About 28,000 electrons escape recombination and, ignoring the effects of LAr purity and diffusion, would all drift to one collection plane wire. The capacitance due to the maximum-length 10 m wire is 226 pF resulting in an equivalent noise charge (ENC) of 530 electrons in the CMOS amplifiers. The signal-to-noise ratio would therefore be 53:1 if all of the ionization electrons arrived at the wire. For a maximum drift distance of 3.7 m and a drift field of 500 V/cm, the required voltage on the cathode plane is 185 kV. This is within the range of commercially available high-voltage cables and within the range of current designs for cryogenic feedthroughs. Additional R&D would be needed for longer maximum drift lengths.

Ionization electrons will be lost due to impurities in the LAr. The fraction that survive passage to the anode planes is $e^{t/\tau}$, where t is the drift time and τ is the drift-electron lifetime. The maximum drift time is the maximum drift length divided by 1.6 mm/ μ s which equals 2.3 ms for LBNE. The ICARUS detector has achieved a drift electron lifetime of 6 – 7 ms. The Materials Test Stand (described in Section 8.3.1) regularly achieves a drift-electron lifetime of 8 – 10 ms. The Fermilab Liquid Argon Purity Demonstrator achieved a lifetime of > 3 ms during initial tests. Based on this experience, and by careful selection of materials in the ullage, a drift-electron lifetime at least as good as ICARUS is expected. The signal-to-noise ratio would be 36:1 for a drift electron lifetime of 6 ms. A minimum lifetime of 1.4 ms is required to meet the 9:1 signal-to-noise ratio requirement.

The cloud of drifting ionization electrons will spread out in space due to the effects of diffusion. The maximum transverse *RMS* width of the electron cloud is 2.4 mm for the chosen drift distance and drift field. This is well matched to the 5 mm wire spacing.

1.4 Detector Development Program

As mentioned above, the design of the LAr-FD has benefited greatly from other experiments and related development activities. Development activities in the U.S. are described in the *Integrated Plan for LArTPC Neutrino Detectors in the US* [11]. This program includes non-LBNE activities such as the Fermilab Materials Test Stand, Fermilab electronics test stand, LAPD, photon detection, ArgoNeUT and MicroBooNE as well as LBNE activities such as the 35-ton prototype and LAr1. The development plan is described in detail in Chapter 8.

1.5 Participants

The design for the LBNE Far Detector is being carried out by an LBNE subproject team, headed at Fermilab but with participants also from Brookhaven National Laboratory, Argonne National Laboratory and Indiana University, in conjunction with an engineering design firm, Arup USA, Inc. This firm has assisted with cryostat and cryogenic-plant design. The detector is planned for construction at the Sanford Laboratory site, which is managed by the South Dakota Science and Technology Authority (SDSTA).

The LBNE Far Detector, called the LAr-FD, is managed by the Work Breakdown Structure (WBS) Level 2 Manager for the Far Detector subproject. The supporting team includes a WBS Level 3 Manager for each of its component systems: Cryogenics & Cryostat, Time Projection Chamber (TPC), Trigger & Data Acquisition (DAQ), Installation & Commissioning and Photon Detector. There is also an L3 manager for the LAr1 Prototype. Figure 1-6 shows an organization chart down to Level 3 (L3).

The Conventional Facilities Level 3 Far Site Manager is the LBNE Project liaison with the LAr-FD subproject to ensure the detector requirements are met; this person is responsible for all LBNE scope at the Far Site. [Management of the Sanford Lab and the organizational relationship between it and the LBNE Project and Fermilab are in the process of being determined; this section will be updated when that is known.]

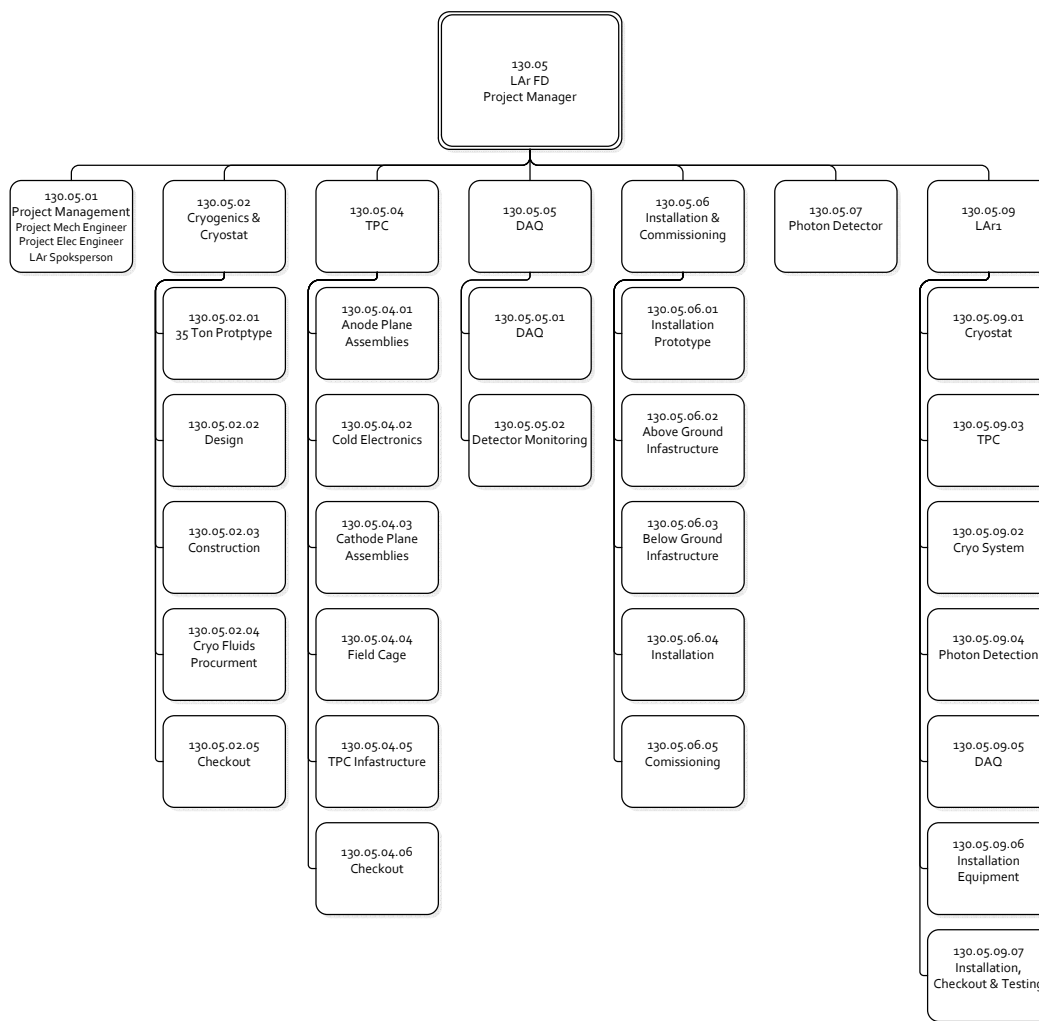


Figure 1-6: Organization chart for the Far Detector subproject

2 Cryogenics System and Cryostat (WBS 130.05.02.01)

The scope of the Cryostat and Cryogenics subsystem includes the design, procurement, fabrication, testing, delivery and installation oversight of (a) a cryostat to contain the liquid argon (LAr) and the TPC, and (b) a comprehensive cryogenics system that together meet the required performance for acquiring, maintaining and purifying the LAr in the detector. This chapter describes a reference design for these interdependent detector elements.

The scope of the reference-design membrane cryostat encompasses the following components:

- Two 20-kton cryostats for a LAr-FD
- LAr tanker truck receiving facilities
- Transfer system to deliver LAr to the underground detector cryostats
- Boil-off gas reliquefaction equipment
- LAr-purification facility
- Cryostat-purge facilities

2.1 Introduction

The conceptual reference-design for the LAr-FD specifies two rectangular vessels each measuring 24.14 m in width, 15.98 m in height and 48.96 m in length, and containing a total active mass of 20 kton of LAr. This membrane design is commonly used for liquified natural gas (LNG) storage and transport tanks (Figure 2-1). A membrane tank uses a stainless-steel liner to contain the liquid cryogen. The pressure loading of the liquid cryogen is transmitted through rigid foam insulation to the surrounding rock, which provides external support for the liner. The membrane liner is corrugated to provide strain relief resulting from temperature-related expansion and contraction (Figure 2-2).

The advantages offered by the membrane design relative to a self-supporting cryostat are:

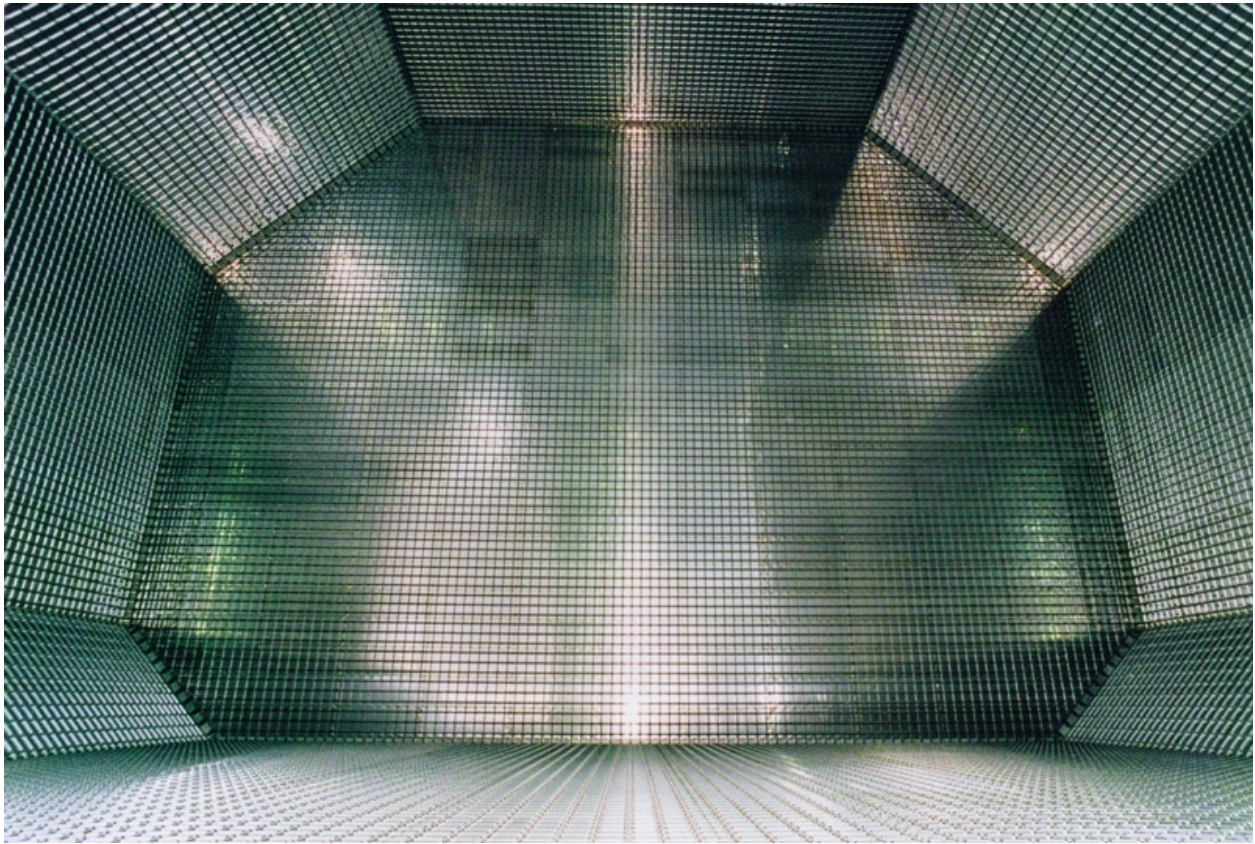


Figure 2-1: Interior of a LNG ship tanker. The tank shown is 24 m high by 35 m wide with interior grid-like corrugations on a 0.34-m pitch. By comparison, a single LAr-FD cryostat is 16 m high by 24 m wide.

The corrugated stainless steel primary barrier:

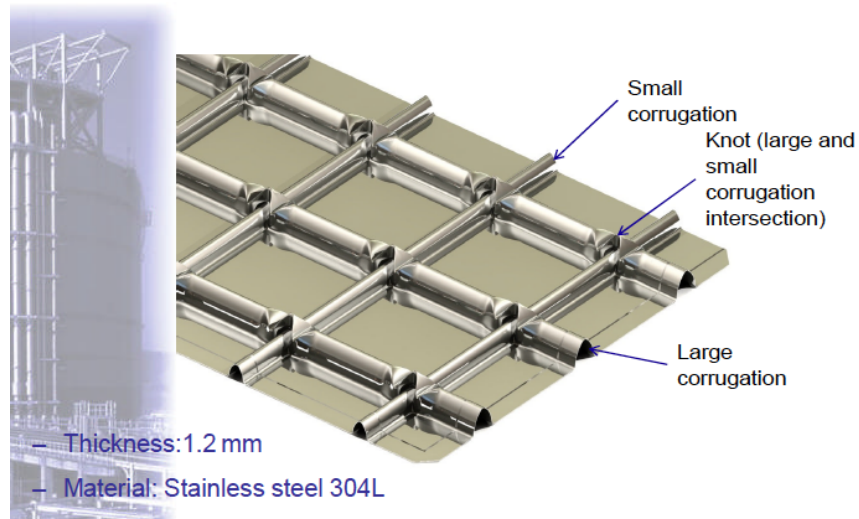


Figure 2-2: Primary membrane section (courtesy GTT)

- Efficient use of the underground cavern volume due to its direct attachment to the rock on floor and sides, which reduces the civil construction costs for the project
- Higher ratio of usable (fiducial) mass to total mass

Conceptual design studies and studies done by ARUP, USA [10] indicate that the implementation strategy for the cryogenics system is independent of the cryostat design. During the conceptual design studies two membrane cryostat vendors have been identified. Those vendors are GTT (Gaztransport & Technigaz) and IHI (Ishikawajima-Harima Heavy Industries). Each is technically capable of delivering a membrane cryostat that meets the design requirements for the LAr-FD. To provide clarity, only one vendor is represented in this CDR (GST system from GTT); this is for informational purposes only and should not be construed as preferring GTT over IHI. Nothing inherent in the IHI design changes the design approach.

2.2 Design Parameters

The requirements and parameters for the cryostat and cryogenic system design are within the LAr-FD requirements documentation [12] [13] and the parameter tables [14], respectively. The overarching system requirements are to provide a high-purity, stable liquid argon environment for the TPC and to provide mechanical support for the TPC. For components that pass through the ullage (the vapor space above the LAr), no sources of reliquefaction may be present. Tables 2-1 and 2-2 offer a brief overview of parameters for a single cryostat of LAr-FD.

2.3 Cryostat Configuration

2.3.1 Sides and Bottom of Tank

The membrane tank is a sealed container that relies on external support from the surrounding rock to resist the hydrostatic load of the contents. In order from innermost to outermost layers, the side walls of the membrane tank consist of the stainless-steel primary membrane; insulation; a secondary, thin aluminum membrane that contains the LAr in case of any leaks in the primary membrane; more insulation; a barrier to prevent water-vapor ingress to the cryostat; concrete; shotcrete and rock. This “in-ground” tank arrangement (i.e., offering access only from the top) protects against possible cryogen leaks out of the tank — there is no place for the cryogen to go because it is surrounded on all sides by rock. The membrane cryostat is considered a “full containment” system in the LNG industry lexicon. The basic components of the membrane tank are illustrated in Figure 2-3.

Table 2-1: Design parameters for one LAr-FD Cryostat

| Parameter | Value |
|---|--|
| Total Volume: | 18,887 m ³ |
| LAr Total Mass: | 25 kton |
| Inner Height of the Tank: | 15.98 m |
| Inner Width of the Tank: | 24.14 m |
| Inner Length of the Tank: | 48.62 m |
| Insulation: | Reinforced Polyurethane; inner layer is 30 cm thick, outer layer is 70 cm thick |
| Primary Membrane(GTT): | 1.2-mm thick type 304L stainless steel with corrugations on 340 mm × 503 mm rectangular pitch |
| Secondary Containment(GTT): | ≈ 0.07-mm thick aluminum between fiberglass cloth; overall thickness is 0.8 mm located between insulation layers |
| Outer Concrete Layer: | 0.5 m thick, inner surface treated with a vapor barrier |
| LAr Temperature: | 89 ± 1 K |
| Depth of the Liquid (Liquid Head): | 15.0 m |
| Design Operating Pressure (Above Liquid): | 0.113 MPa |
| Design Operating Pressure (Bottom of Liquid): | 0.2217 MPa |
| Rated Pressure Capacity of Tank: | 0.52 MPa (calculated according to BS EN 14620) (British-Adopted European Standard / 29-Dec-2006 / 60 pages ISBN: 0580497763) |

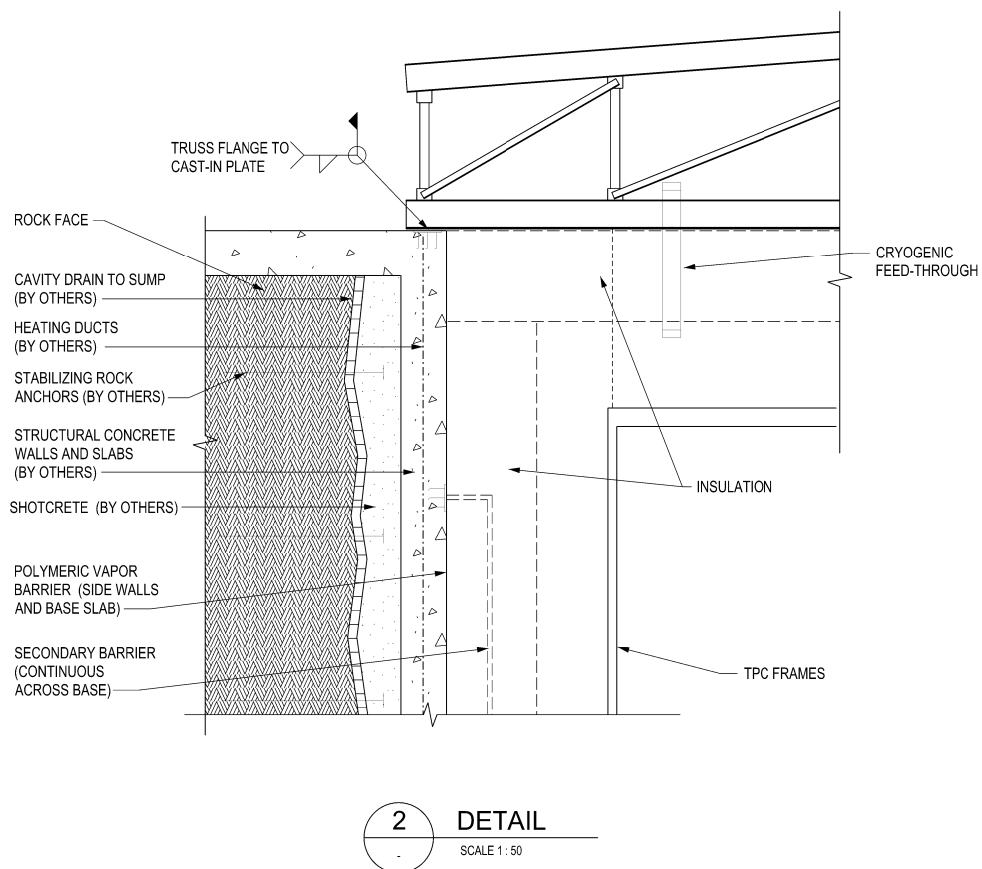


Figure 2-3: Composite system as installed for the LAr-FD reference design

Table 2–2: Summary of parameters for the 4850L membrane cryostat

| Property | Reference-Design Cryostat |
|---|---|
| Personnel access to cavern | Ross Shaft |
| Equipment transport to cavern | Yates Shaft |
| Construction access to pit | Through temporary construction opening in side walls at base of pit |
| Type of crane in cavern | 20 ton bridge crane |
| Base slab | Reinforced concrete |
| Side walls | Reinforced concrete |
| Heating system | Redundant / Replaceable Electric system |
| Roof | Pre-fabricated steel truss modules with lower steel plate |
| Vapor barrier | Polymeric on concrete surfaces / steel plate on roof |
| Insulation / Secondary barrier / Membrane | GST system by GTT |
| TPC | Individual 2.5 x 7m frames lowered through 2m x 4m roof hatch. Assembled within cryostat and suspended by hangers passing through the roof. |
| LAr containment system | Full containment: Membrane / Secondary Barrier / Concrete Liner |

2.3.2 Concrete Liner and Vapor Barrier

The formed concrete liner will be poured against the sides and bottom of the excavated rock pit. Conduits and heating elements will be embedded in the concrete liner to maintain rock temperatures above freezing to preclude any problems associated with freezing water and heaving as the rock temperature drops below the freezing point of water. The embedded conduits are encased approximately midway in the concrete side walls, end walls and bottom floor slab as depicted in Figure 2–4. The concrete liner and conduits are provided under the conventional facilities scope. The heating elements are provided by LAr-FD scope.

A vapor barrier is required on all internal surfaces of the concrete liner (base slab, side walls, end walls) and the roof to prevent the ingress of any water vapor into the insulation space. If water vapor were permitted to migrate into the insulation space it could freeze and degrade the thermal performance of the insulation. The barrier must also reliably absorb the stresses and strains from all normal loading conditions. The selected vapor barrier material is a polymeric liner for the side and bottom surfaces. This has been used extensively in onshore LNG tank applications. The vapor barrier for the top will be solid steel plate welded together and to the underside of the roof truss.

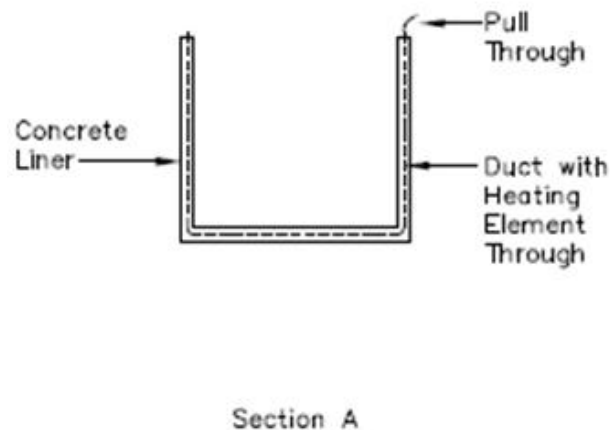


Figure 2-4: End view of concrete liner showing embedded conduits for heating elements (Courtesy Arup)

Table 2-3: Heat load calculation (Thickness = 1 m for all)

| Element | Area (m ²) | C (W/m-K) | ΔT (K) | Heat input (kW) |
|------------|------------------------|-----------|----------------|-----------------|
| Base | 1324 | 0.0283 | 191 | 7.2 |
| End walls | 888 | 0.0283 | 191 | 4.8 |
| Side walls | 1781 | 0.0283 | 191 | 9.6 |
| Roof | 1324 | 0.0283 | 207 | 7.2 |
| Total | | | | 28.7 |

2.3.3 Insulation System and Secondary Membrane

The membrane cryostat requires insulation applied to all internal surfaces of the concrete liner (base slab, side walls, end walls) and roof in order to control the heat ingress and hence the required refrigeration load. Choosing a reasonable, maximum insulation thickness of 1 m, and given an average conductivity coefficient for the insulation material of $C \approx 0.0283$ W/m-K, the heat input from the surrounding rock is expected to be 28.7 kW total. This is shown in table 2-3.

The insulation material, a solid fiberglass foam, is manufactured in 1-m \times 3-m composite panels. The panels will be laid out in a grid with 3-cm gaps between them (that will be filled with loose fiberglass) and fixed onto anchor bolts embedded into the concrete at \sim 3-meter intervals. The composite panels contain the 70-cm thick outermost insulation layer, the secondary membrane and the 30-cm innermost insulation layer. After positioning adjacent composite panels and filling the 3-cm gap, the secondary membrane is spliced together by epoxying an additional layer of secondary membrane over the joint. All seams are covered so that the secondary membrane is a continuous liner.

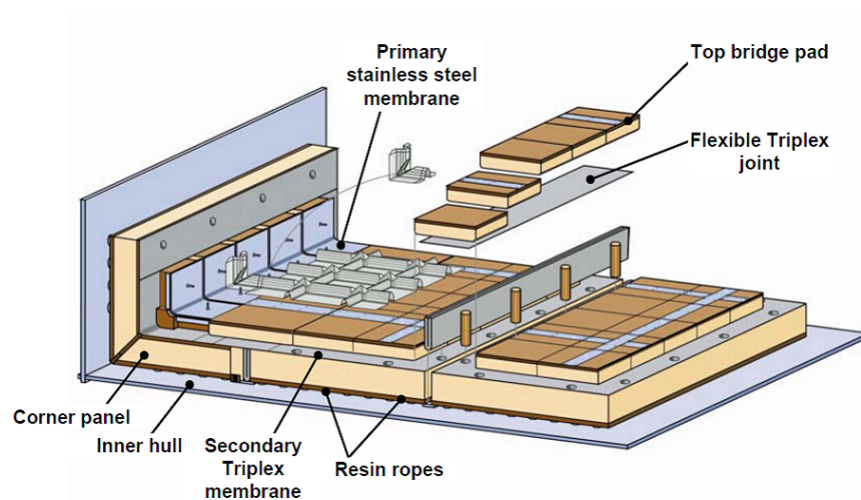


Figure 2-5: Membrane corner detail

The secondary membrane is comprised of a thin aluminum sheet and fiberglass cloth. The fiberglass-aluminum-fiberglass composite is very durable and flexible with an overall thickness of ~ 1 mm. The secondary membrane is placed 70 cm in from the concrete (30 cm from the primary membrane), within the insulation space. It surrounds the internal tank on the bottom and sides, and it separates the insulation space into two distinct, leak-tight, inner and outer volumes. The outer, 70-cm thick volume of insulation separates this membrane from the concrete. This sheet is connected to embedded metal plates in the vertical concrete wall at the upper edge of the tank. In the unlikely event of an internal leak from the cryostat's primary membrane into the inner insulation space, it will prevent the liquid cryogen from migrating all the way through to the concrete liner where it would degrade the insulation thermal performance and could possibly cause thermal stress cracks in the surrounding concrete. The liquid cryogen in the case of leakage through the inner (primary) membrane will be contained in the secondary membrane volume.

2.3.4 Tank Layers as Packaged Units

Membrane tank vendors have a “cryostat in a kit” design that incorporates insulation and secondary barriers into packaged units. See Figure 2-6. Figure 2-3 illustrates how these layers would be used in the LAr-FD reference design.

2.3.5 Top of Tank

The stainless-steel primary membrane and the intermediate layers of insulation and water-vapor barrier continue across the top of the detector, providing a vapor-tight seal. Note that no secondary membrane is used or required for the cryostat top. The cryostat roof is a steel

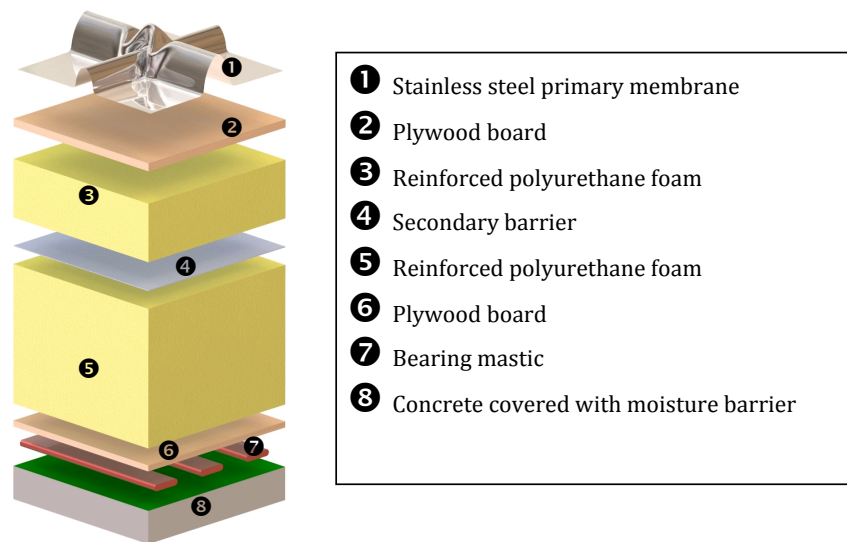


Figure 2-6: GST composite system from GTT

truss structure that bridges the detector. Stiffened steel plates hermetically welded to the underside of the truss form a flat vapor barrier surface onto which the roof insulation attaches directly; this is described more fully below. Fully fabricated roof trusses are placed across the roof with a 2-m spacing. Field-fabricated truss members and steel plates are welded between the prefabricated trusses to connect the two prefabricated sections. The roof is built up of alternating prefabricated and field-fabricated “in-fill” roof sections. This configuration was selected during the screening process because it provides an efficient, gas-tight solution that can be readily constructed within the cavern space.

The truss structure rests on the top of the concrete wall as shown in Figure 2-3 where a positive structural connection between the concrete and the truss is made to resist the internal upward force caused by the slightly pressurized LAr in the cryostat. The hydrostatic load of the LAr in the cryostat is carried by the floor and the side walls. Everything else within the cryostat (TPC planes, electronics, sensors, cryogenic- and gas-plumbing connections) is supported by the steel plates under the truss structure. All piping and electrical penetrations into the interior of the cryostat are made through this top plate to minimize the potential for leaks.

Studs are welded to the underside of the steel plates to bolt the insulation panels to the steel plates. Insulation plugs are inserted into the bolt-access holes. The primary membrane panels (also manufactured in smaller sheets) are first tack-welded then fully welded to complete the inner cryostat volume. Feed-through ports as shown in Figure 2-8 are located at regular intervals within the corrugation pattern of the primary membrane to accommodate TPC hangers, electrical and fiber-optic cables, and piping. The roof truss will be anchored to the top of the poured-concrete liner walls to resist the uplift caused by internal tank overpressure (Figure 2-7). The roof truss will be pre-fabricated off-site in ~2 m wide, fully-welded modules

and transported to the cavern as required by the installation schedule.

The prefabricated steel roof-truss modules are relatively lightweight ($\sim 8,000$ kg each) and require only moderate crane capacity. If the steel trusses need to be separated into smaller pre-fabricated units for transport to the underground site, assembly within the cavern space prior to installation is relatively straightforward. The truss construction is illustrated in Figure 2-7.

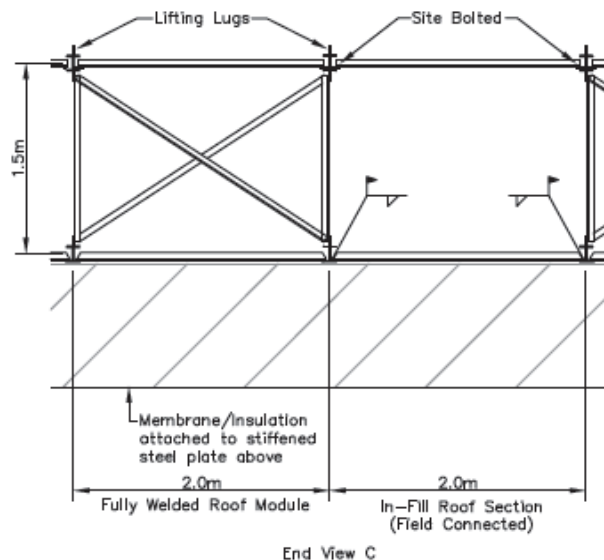


Figure 2-7: Roof truss structure (Courtesy Arup)

Some equipment, such as monitoring instrumentation and pumps, will be installed within wells extending through the roof structure. All connections into the cryostat will be made via nozzles or penetrations above the maximum liquid level and mostly located on the roof of the cryostat. See figure 2-8 for a typical roof-port penetration.

2.4 LAr Circulation and Temperature-Profile Modeling

The liquid circulation in the LAr-FD cryostats has been modeled using computational fluid dynamics modeler software (ANSYS CFX). A field cage, described in Section 3.5, was modeled with half-inch slots cut every five inches, yielding a 10% porosity. Standard insulation thermal conductivity of 0.0283 W/m-K was used to model the heat flux into the LAr, and used an exterior temperature of 278 K and an internal temperature of 87.15 K. The temperature stability requirement on LAr-FD is ± 1 K. The model in Figures 2-9 and 2-10 clearly identifies convective currents flowing through the entire liquid bath with the highest velocities < 0.034 m/s, and a temperature gradient much less than 0.1 K across the entire fluid body. This indicates conformity with LAr-FD requirements and parameters.



Figure 2-8: Nozzle in roof membrane cryostat (Figure courtesy GTT)

2.5 Leak Prevention

The primary membrane will be subjected to several leak tests and weld remediation, as necessary. All (100%) of the welds will be tested by an Ammonia Colorimetric Leak Test (ASTM E1066-95) in which welds are painted with a reactive yellow paint before injecting gas with 25% ammonia into the bottom insulation space of the tank. Wherever the paint turns purple or blue, a leak is present. Both membrane cryostat manufacturers use this technique for certifying that a cryostat is leak-tight. Any and all leaks will be repaired. The test lasts a minimum of 20 hours and is sensitive enough to detect defects down to 0.003 mm in size and to a 10^7 std-cm³/s leak rate (equivalent leak at standard pressure and temperature, 1 atm and 273 K).

Both membrane cryostat manufacturers use this technique for certifying that a cryostat is leak-tight.

To prevent infiltration of water-vapor or oxygen through microscopic membrane leaks (below detection level) the insulation spaces will be continuously purged to provide one volume exchange per day. The gaseous space in the insulation under the primary membrane is estimated to be 116 m³ which requires an argon gas flowrate of 16.3 m³/hr at standard temperature and pressure.

The insulation space between the primary and secondary barriers will be maintained at 0.103 MPa, slightly above atmospheric pressure. This space will be monitored for changes that might indicate a leak from the primary membrane. The outer insulation space will also be purged with argon at a slightly different pressure. The pressure gradient across the

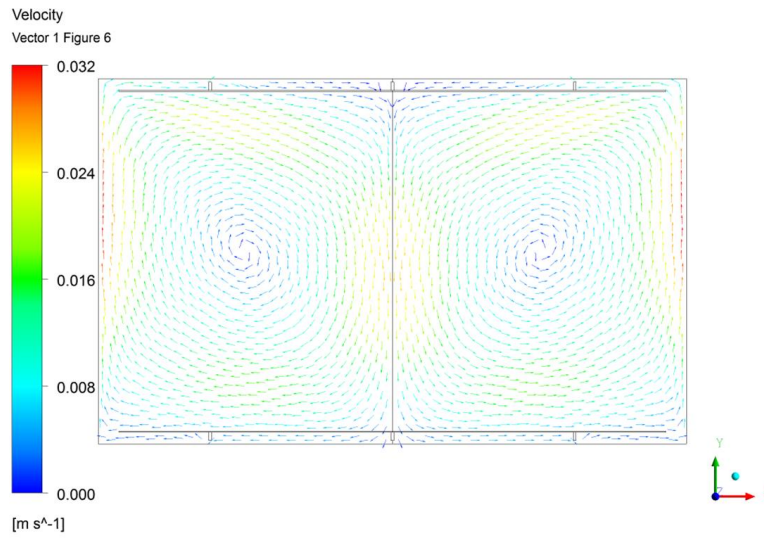


Figure 2-9: LAr velocity profile

ANSYS

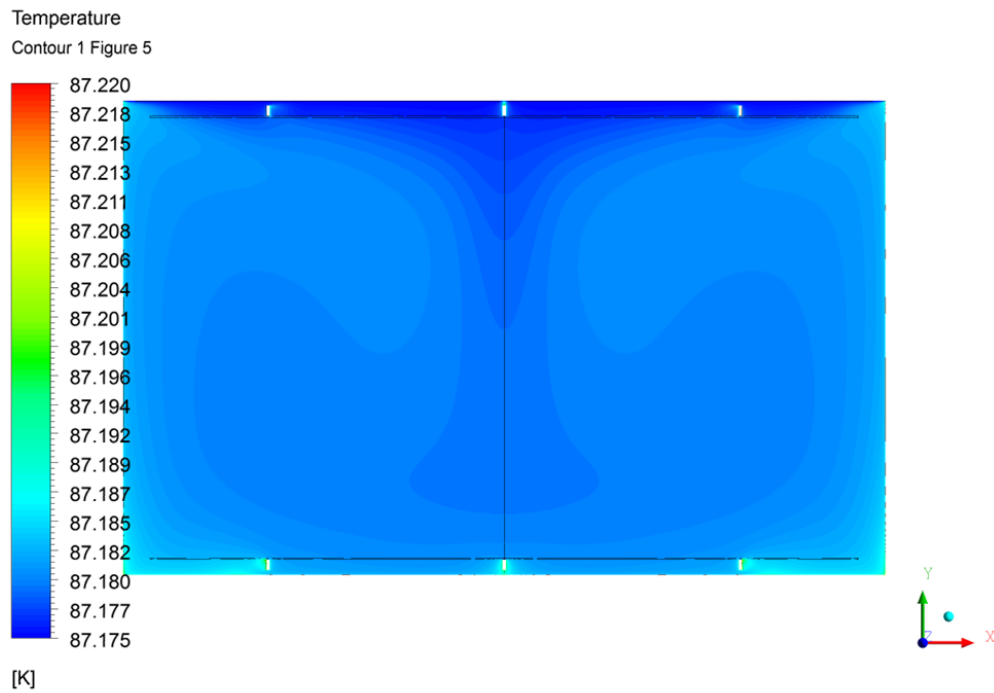


Figure 2-10: LAr temperature profile

Table 2-4: Cryogenic system equipment location for 4850L siting

| Equipment | Location |
|---------------------------------------|----------|
| Nitrogen liquefier plants | cavern |
| 50 m ³ LN backup dewar | cavern |
| 50 m ³ LN receiving dewar | surface |
| 50 m ³ LAr receiving dewar | surface |
| Argon recondensers | cavern |
| Argon Purification Plants | cavern |

membrane walls will be maintained in the outward direction. Pressure-control devices and relief valves will be installed on both insulation spaces to ensure that the pressures in those spaces do not exceed the operating pressure inside the tank.

The purge gas will be recirculated by a blower to a small purge gas dryer and reused as purge gas. The purge system is not safety-critical, and an outage of the blower would have only a minimal, short-term impact on operations [15].

2.6 Cryogenic Systems Layout

Cryogenic systems are located on the surface and within the cavern. Within the cavern, the cryogenic systems external to the cryostat are located in the 15 m septum space between the two cryostats. The cavern cryogenic systems layout and design take into account the depth of the cryostat and the impact on ODH. The layout design:

- minimizes the exposure of personnel to any ODH
- minimizes passive heat ingress to the cryogenic system
- minimizes the potential for argon escape or contamination by minimizing the volume of the argon system external to the cryostat

The schematic in Figure 2-16 delineates the services located above and below ground. The equipment is listed in Table 2-4. The recondensers and purifiers will be located underground, adjacent to the cryostats. The purification vessels will be located within the septum space. Required cryogenic piping which must travel from the surface to the cavern will be routed through the Oro Hondo shaft to a utility access shaft and into the cavern. The surface facility will include a 50 m³ LAr receiving dewar and a 50 m³ liquid-nitrogen dewar.

Each cryostat will have its own argon recondenser, argon-purifying equipment and overpressure-protection system, each set of systems placed on the septum side of the wall closest to the

cryostat it serves. Four 34-m³/hr circulation pumps will be placed within each membrane cryostat to circulate liquid from the bottom of the tank through the purifier. With the purifier in the cavern, these pumps each add about 6.5 kW of pump energy to the liquid. The required refrigeration needs are dependent on the mode of operation within both detector modules. The combined plant capacity of 225 kW (sum of three refrigerators) surpasses estimated loads during all modes of operation.

The following equipment will be placed directly on or within the cryostat:

- Circulation pumps to deliver LAr to purifiers
- “Commissioning” circulation pumps to deliver LAr to purifiers at an elevated rate during the initial purification period
- LAr return/fill pipe
- Boil-off gas line
- Cryostat pressure-control valves
- Cryostat pressure-relief valves
- Purge pipework
- Cryostat-monitoring equipment (for pressure, temperature, density, depth, and so on)

Equipment located within the cryostat, such as monitoring instrumentation and pumps, will be installed within wells extending through the roof structure. Figure 2-11 shows a pump tower located in the hull of a membrane cryostat tanker.

2.7 Pipework between Surface and Cavern

2.7.1 Service Duty, Size and Material

Risers and all other pipework carrying cryogenic fluids that link the cavern to the surface will be routed from the 4850L through a new 4.0-m-diameter vertical exhaust-ventilation shaft that connects to the existing Oro Hondo Shaft at the 3650L. The shaft will contain the LN supply, an LAr pipe, vent piping and control-system wiring. Because of the long distance required for transporting LN₂ and LAr from the surface for the initial fill, each of these circuits will have pressure-reducing stations within the shaft. Table 2-5 lists the lines, service duty and size of piping that will run between the surface and cavern.

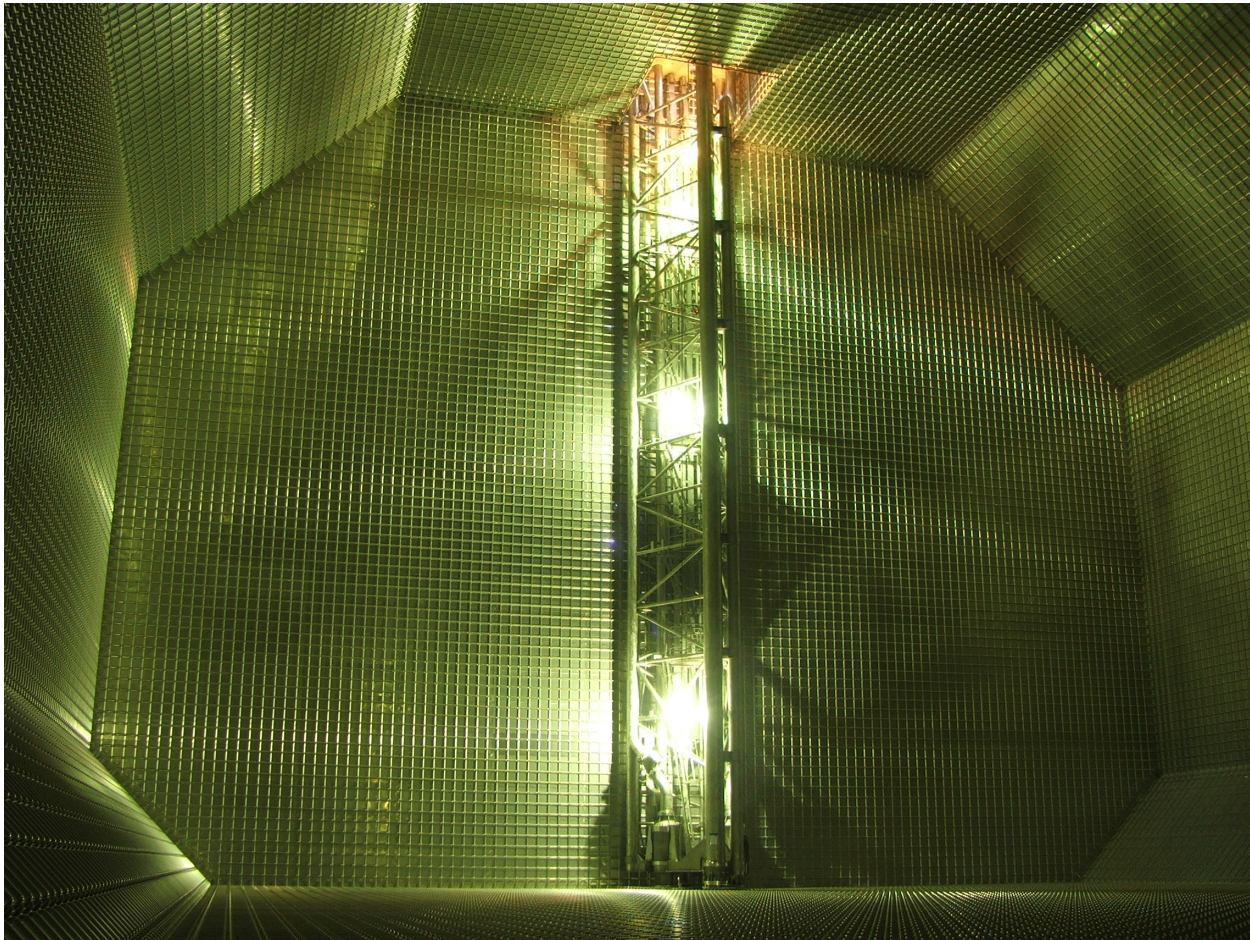


Figure 2-11: Pump tower in LNG membrane cryostat

Table 2-5: Piping between surface and cavern; location, duty and required size

| Start and End Points | Duty | Inner Pipe Size and OD (in) | Outer Pipe Size and OD (in) |
|---|--------------------------------------|-----------------------------|-----------------------------|
| LAr supply from surface dewar to cryostat | filling only | 3, 3 1/2 | 5, 5 5/16 |
| LAr to purifiers | filling only | 6, 6 5/8 | 8, 8 5/8 |
| Purifier return to cryostat | continuous | 6, 6 5/8 | 8, 8 5/8 |
| Argon vent to surface | emergency relief | 6, 6 5/8 | 8, 8 5/8 |
| LN vent from recondenser to surface | backup operation or emergency relief | 6, 6 5/8 | 8, 8 5/8 |
| LN supply from refig plant to recondenser | continuous | 6, 6 5/8 | 8, 8 5/8 |
| LN return from recondenser to refig plant | continuous | 6, 6 5/8 | 8, 8 5/8 |

Metallic piping will be designed in accordance with ASME B31.3 “Process Piping” and ASME B31.5 “Refrigeration Piping and Heat Transfer Components”. The piping material will be seamless Austenitic Stainless Steel in accordance with ASTM A312 Grade TP304 or TP304L. No corrosion allowance is required with this material in this service as stainless steel is resistant to corrosion.

2.7.2 Venting and Fluid Flow

Cryogenic piping will be slightly inclined such that any boil-off vapor either vents through the dedicated vent system or returns to the cryostat, interim storage dewar, or another vessel designed for vapor recovery.

Wherever the pipe-system design leaves open the potential for isolated pipe sections to fill with cryogenic fluid (e.g., fluid trapped between valves), overpressure protection will be provided.

Smaller-bore vents and drains will be used as necessary to facilitate testing, commissioning and cool-down of the piping system. This will also allow minimal use of of flange-type or bayonet-type connections and thus reduce the risk of leakage. Flanges may be required for the connection of the piping to proprietary equipment and vessels. When flanges are required, the minimum flange rating will be in accordance with ASME B16.5 class 150. Since high flow velocities can cause abrasion on interior piping surfaces that in turn can cause erosion of these surfaces, all piping systems will be sized to limit liquid-flow velocities to a maximum of 2.5 m/s and gaseous flows to less than 20 m/s.

2.7.3 Contamination Prevention

In addition to the minimization of flanges, all piping components and equipment will be selected and evaluated with consideration given to the potential for introducing contamination into the system. Instead of the standard valves (that can cause initial contamination due to trapped gases in the packing), the use of valves with bellows-sealed valve stems that provide a hermetic seal may be warranted on continuous-service lines. Where leak paths cannot be eliminated, for instance in rotating shaft seals, a guard atmosphere of gaseous argon will be implemented so that any ingress into the system will be essentially pure argon rather than cavern air.

All pipe sections and equipment will be designed to be isolated from service for maintenance or replacement. All cryostat piping connections will have isolation valves for this purpose. Also critical relief and safety devices will have online spares to facilitate isolation and maintenance of these devices.

2.7.4 Insulation

All cryogenic piping will be insulated to minimize heat transfer to the fluid and to provide personnel protection. Vacuum-insulated pipe will minimize heat load and simplify pipe-support arrangements on straight runs. Internal-bellows arrangements are preferred to compensate for thermal length changes.

For vacuum-insulated pipe, the outer pipe will be pressure-rated for at least the design pressure of the internal pipe in order to provide secondary containment in the event of inner pipe failure. The vacuum-insulated pipe will be designed to contain the vapor or liquid cryogens in the event of an inner pipe failure, and safety-relieve these cryogens to the vent (to surface). Proper relief sizing will be determined through an engineering design process and reliefs will be installed at every vacuum-insulated section. The use of foam insulation will be minimized. When foam insulation is used, a waterproof cladding will be applied to prevent water from entering the insulation.

2.7.5 Mechanical and Thermal Stress

In accordance with the requirements of ASME B31, all piping systems will be adequately supported to limit stress and vibration from thermal, mechanical and environmental sources, and from installation, long-term fatigue and seismic loads. Piping within the cryostat or within other vessels will require adequate support or guidance to minimize stress and to accommodate differential thermal contraction as the various components cool.

2.8 Equipment Redundancy

Safety-critical equipment will be implemented, generally, on the basis of N+1 units. This will ensure continuity of operation and adherence to safety standards whenever a unit is unavailable, e.g., for maintenance. Items for short-term use and those already present in multiple units are exempted; a low probability of failure is assumed for the former and minimal impact to the performance of the overall system for the latter.

Where common-function equipment could be used for alternate systems, interconnections will be provided to increase flexibility during operation. An example of this would be an interconnection between argon and nitrogen delivery lines at the surface facility and in the cryostat cavern. The argon piping will only be used during LAr delivery to the cryostat and the LN₂ will only be used in a power outage emergency and initial N₂ charging of compressors. Later during the lifetime of the experiment, if the LN delivery system had a problem such as excessive heat leak due to a degradation of vacuum, the LAr piping could be used in its place.

2.9 Cryogenic System Processes

The functionality of the cryogenic system is shown in Figure 2-12. This diagram shows the interconnectivity between major functions of the cryogenic system. The details showing only major piping connections can be seen in Figure 2-13. The major functions serving the cryostat are cryogen supply for cool down and fill, gas filtration, condensing, liquid filtration and circulation, argon-purity analysis, and argon condensing. The methods presented in this section are motivated by experience from other LAr TPC cryogenic systems such as ICARUS and LAPD.

2.9.1 Cryostat Initial Purge and Cool-down

After cryostat construction and following installation of all scientific equipment, the cryostat will be cleaned, purged and cooled. Construction procedures leading up to this point will ensure that the completed cryostat does not contain debris and is free of all loose material that may contaminate the LAr.

2.9.1.1 Initial Purge

Argon piping will be isolated, evacuated to less than 0.1 mbar absolute pressure and backfilled with high-purity argon gas. This cycle will be repeated several times to reduce contamination levels to the ppm level in the piping. The reference-design choice for removing air from the membrane cryostat will be to flow/piston-purge argon, introducing the heavy argon gas at the bottom of the tank and removing the exhaust at the top. The bottom field cage (part of the TPC, described in Section 3.5), serves an additional role as a flow diffuser during the initial purge. A matrix of small holes in the field cage will provide a uniform flow, approximately 10-mm diameter at a 50-mm pitch.

The flow velocity of the advancing argon-gas volume will be set to 1.2 meters/hour. This is twice the diffusion rate of the air downward into the advancing argon so that the advancing pure argon-gas wave front will displace the air rather than just dilute it. A 2D ANSYS model of the purge process shows that after 20 hours of purge time and 1.5 volume changes, the air concentration will be reduced to less than 1%. At 40 hours of elapsed time and three volume changes, the purge process is complete with residual air reduced to a few ppm. This simulation includes a representation of the perforated field cage at the top and bottom of the detector and heat sources due to the readout electronics. The cathode planes (Section 3.4) are modeled as non-porous plates although they will actually be constructed of stainless-steel mesh.

The computational fluid dynamics (CFD) model of the purge process has been verified with

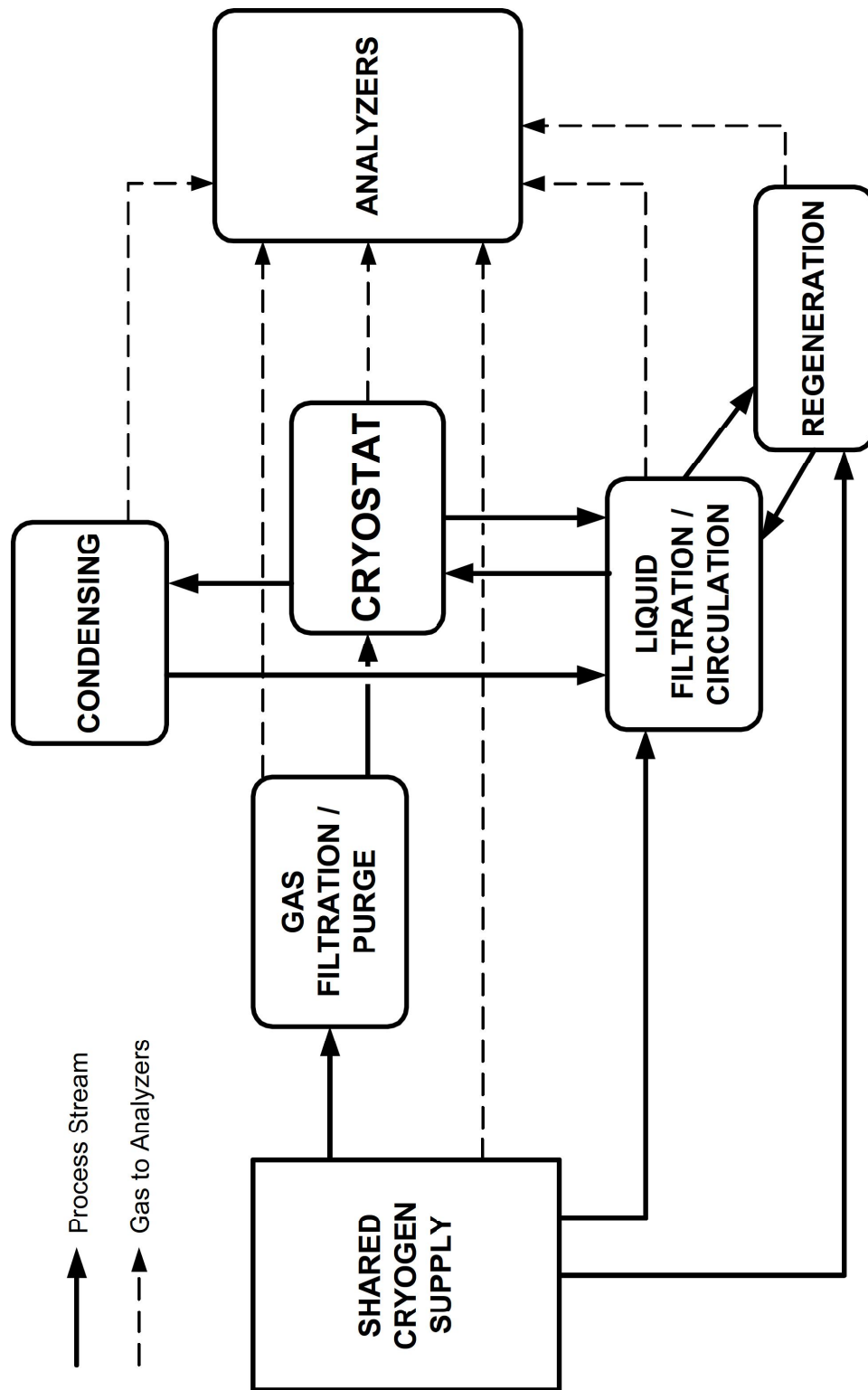


Figure 2-12: Cryogenic system functions

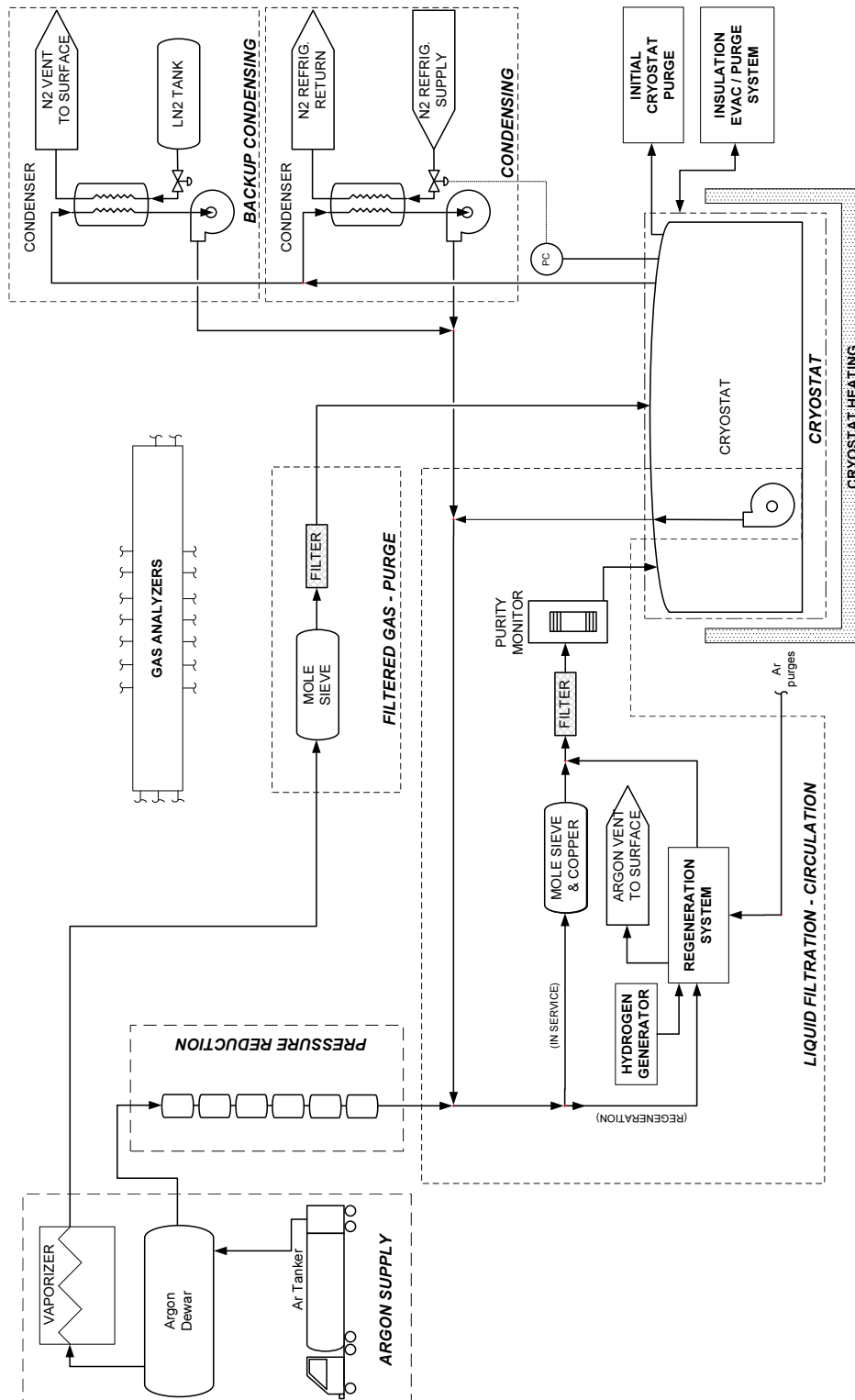


Figure 2-13: Cryogenic system block flow diagram

purge tests in an instrumented 1-m-diameter by 2-m-tall cylinder. Recognizing that obtaining the required purity levels by the flow-purging method needs to be clearly demonstrated, a Liquid Argon Purity Demonstrator (LAPD) project is underway at Fermilab. The LAPD is a right-cylindrical vessel, 3 m in diameter and 3 m tall. LAPD took gas-sampling measurements at varying heights and times during the purge process. Experimental measurements taken by the Liquid Argon Purity Demonstrator (LAPD) have verified the previous modeling of this purge process. LAPD used nine volume changes to reach single-digit contamination levels (ppm by volume) for oxygen, water and nitrogen. Following the purge, the LAPD vessel was cooled down and filled with LAr.

On September 30, 2011, LAPD leaders presented initial results of this purge process in which purity levels of less than 100 ppt oxygen-equivalent were achieved. This confirms that the purge method works and it minimizes the risk to LBNE that an evacuable vessel will be required. This test will be repeated with a 35-ton membrane cryostat prototype.

2.9.1.2 Water Removal via Gas Flow

Water and oxygen will continue to be removed from the system for several days following the initial purge. Flowing gas will be used at the same rate, however at this stage the gas will be filtered and recirculated. Each cryostat contains ~ 3 tons of FR4 circuit-board material and a smaller inventory of plastic-jacketed power and signal cables. These somewhat porous materials may contain as much as 0.5% water by weight. Water-vapor outgassing from these materials will be entrained in the gas flow exiting the top of the cryostat and will be removed from the gas stream by filters. Adsorbed water will also be removed from the metallic inner surfaces of the cryostat and piping system. Water deep within porous materials will remain; this is not a problem since the water diffusion rate in FR4 at room temperature is already quite low ($0.3 \mu\text{m}^2/\text{s}$) and the FR4 assemblies are relatively thick (1 cm).

2.9.1.3 Alternative Water-Removal Method: Evacuation

The traditional method for removing water and oxygen from LArTPCs is to evacuate the tank to $\sim 10^{-4}$ mbar and then backfill it with pure argon gas. Although the primary membrane of the reference-design cryostat is not a vacuum vessel, it is possible to evacuate it. A membrane-cryostat vendor reports that evacuation of the insulation spaces (external to the primary membrane) is normally done during the construction and leak-checking phases. A vacuum pressure of less than 200 mbar absolute in the insulation spaces has so far been achieved. As long as the pressure-differential direction across the walls is kept outward, it is possible to reduce the internal membrane-tank volume to these pressures as well.

2.9.1.4 Initial Cool-Down

Purified LAr will be distributed across the bottom of the cryostat to cool down the cryostat. The boil-off gas will flow through the volume of the cryostat, then routed to the recondenser and liquid-filtration system. Simulation has shown that the liquid cool-down method can be controlled to stay within the available recondenser capacity. The required cooling rate is determined by the maximum stress that detector components can tolerate. For example, the 150- μm APA wires will cool much more rapidly than the APA frames. A mass flow-control system with temperature-monitoring system will be used to control the temperature difference across the cryostat. The exact temperature difference required is yet to be determined; it will be based on input from the cryostat designer and the requirements of the TPC components and structure.

2.9.1.5 Initial Purge and Cool-Down Design Features

Internal piping is positioned within the cryostat to support the purge and cool-down procedure. Heavy argon vapor, which is a result of cooling down the membrane bottom with liquid, will promote purging after it rises from the base of the cryostat and is vented from the roof level. The LAr-supply pipework will have nozzles spaced along its length to distribute equal liquid-delivery flow rates across the bottom of the cryostat. The flow nozzles will be directed downward or to the side so that the injection velocity will not cause local vertical gas plumes or turbulent mixing but rather will spread across the bottom of the tank and produce a stable, upwardly advancing argon wave front. The vertical velocity of 1.2 m/hr for the gas purge includes a contingency for some level of turbulent mixing.

Main gas returns, used for pressure control, will be distributed along the cryostat roof. All nozzles and dead-end (stagnant) volumes located at the top of the cryostat will have gas-exhaust lines for the initial purge and for continuous sweep-purge of those volumes during normal operations. The sweep-purge during the initial stage of purging will be vented outside of the cavern. After all but trace amounts of air have been expelled, the gas returns will be routed to the recondensers before being returned to the cryostat. When cool-down to 120 K is complete (and during steady state operations), the gas returns will be sent to the recondenser to be liquefied by heat exchange with a liquid nitrogen stream. The recondensed liquid will be filtered and sent back to the cryostat to complete the cool-down operation. All purge gas will be contained and either vented outside of the cavern at a remote location, or recondensed and reused.

2.9.2 Liquid Argon Receipt

Each LAr detector module will hold an inventory of 25 kton of liquid argon. Initial purge operations are expected to consume and exhaust about 0.25 kton per module. Considering

that some product will also be lost in transit, approximately 52 kiloton of LAr will need to be procured. Planning the logistics and supply of LAr to the facility requires consideration of the following issues:

- total capacity of commercial air-separation plants within freight distance of the facility (the peak delivery potential)
- extent of boil-off that will occur in transit
- number of vehicle movements required and their impact on the local community
- costs and benefits associated with stockpiling LAr at the facility ahead of commencing the purge, cool-down and fill procedure
- provision of a temporary air-separation plant at the facility to generate liquid argon
- availability and cost associated with the delivery of high-purity LAr as opposed to lower-quality, commercial-grade argon combined with on-site, coarse purification

Total argon production in the United States is currently approximately 3.6 kton per day. Argon is normally co-produced along with large volumes of oxygen, so any project that requires large oxygen quantities may also spur additional argon production, enhancing the supply capacity. A 2013-2018 market-forecast report by the Freedonia group [16] indicates that the demand for argon will increase at a rate of 3.4% per year whereas the demand for oxygen will increase 4.8%.

The standard grade specification for argon is a minimum purity of 99.995%, allowing a maximum concentration of 5.0 ppm for O₂ and 10.5 ppm for H₂O. This is designated as Grade 4.5 in the gas-supply industry. Requiring higher-purity product would significantly reduce the volume of product available to the experiment, increasing cost and pushing out the schedule. Therefore, standard product will be procured from multiple vendors. The most efficient mode of argon delivery is over-the-road tank truck with a maximum capacity of 18.7 metric ton (MT). The expected number of such deliveries needed is 1,400 over six months to fill one cryostat. Rail delivery is not cost-effective as there are no rail spurs leading to the site. This mode would require transfer of product from rail tanker to a tank truck, introducing cost that exceeds the benefit.

Surface facilities are required for the offloading of LN and LAr road tankers. It will be necessary to procure approximately four trailer loads of liquid nitrogen (about 40 tons) for the initial filling of the LN refrigeration dewar and charging of a single refrigeration plant. Vehicle access and hard-surfaced driving areas are required adjacent to the LN dewar and LAr-supply piping. An interim LAr storage dewar will hold the contents of a road tanker in order to minimize off-loading time. Road tankers will connect to a manifold located close to the 300L access drift and will use their on-board pumps to transfer the LAr to the storage

Table 2-6: Estimated heat loads within the cryostat

| Item | Heat Load (kW) |
|--------------------------|----------------|
| Insulation heat loss | 28.7 |
| Electronics power | 5.2 |
| Recirculation-pump power | 6.5 |
| Total | 40.4 |

dewar. Each tanker will be tested to ensure that the LAr meets the purity specification. The LAr will be gravity-fed to the cavern where it will be routed through the purification system prior to discharging into one of the cryostats.

2.9.3 Cryostat Filling

Liquid argon will be delivered to the cryostat through the cryostat-filling pipework. Because of the depth of the cryostat location there is a need to have pressure-reducing stations for both the LN2 and LAr cryogen circuits. These stations will be located at specified intervals throughout the vertical drop in the shaft and are designed with recirculation to operate away from geysering conditions. This minimizes pressure swings and improves the stability of operations.

The filling process will take place over many weeks due to the delivery schedule of liquid argon described in the previous section. Liquid-argon purification can begin once the liquid depth reaches about 2 m in the cryostat. At this depth, the recirculation pumps can safely turn on and direct up to 136 m³/hr (600 gpm) of liquid argon through the purification system.

2.9.4 Argon Reliquefaction and Pressure Control

High-purity liquid argon stored in the cryostat will continuously evaporate due to the unavoidable heat ingress discussed in Section 2.3.3. The argon vapor (boil-off gas) will be recovered, chilled, recondensed and returned to the cryostat. A closed system is required in order to prevent the loss of the high-purity argon.

During normal operation the expected heat ingress of approximately 40.4 kW to the argon system will result in an evaporation rate of 1100 kg/hr and expanding in volume by a factor of 200 when it changes from the liquid to vapor phase. This increase in volume within a closed system will, in the absence of a pressure-control system, raise the internal pressure. In LAr-FD, argon vapor will be removed from the top of the cryostat through the chimneys that contain the cryogenic feedthroughs. As the vapor rises, it cools the cables and

feedthrough, thereby minimizing the outgassing, and sweeping any contaminants up and out of the cryostat.

The exiting gaseous argon will be directed to a heat exchanger (recondenser) in which it is chilled against a stream of liquid nitrogen and condensed back to a liquid. As the argon vapor cools, its volume reduces and in the absence of pressure control further gas would be drawn into the heat exchanger, developing a thermal siphon. Therefore, a pressure-control valve on the boil-off gas lines will control the flow to the recondenser to maintain the pressure within the cryostat at $0.113 \text{ MPa} \pm 0.003 \text{ MPa}$. The liquid nitrogen stream (that provides the coolant for the recondenser) will be supplied from a closed-loop LN2 refrigeration plant. The commercial refrigeration plant uses compression/expansion and heat rejection to continuously liquefy and reuse the returning nitrogen vapor.

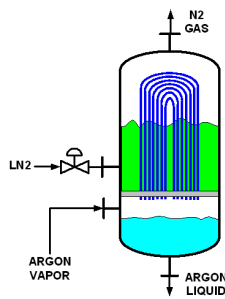


Figure 2-14: Liquid argon recondenser

Each module of the detector has a dedicated nitrogen-refrigeration plant and a third plant will be on standby during normal operations. Further, each module will also have one operating and one standby argon recondenser. This will ensure high availability of the recondensing system, minimize the need to vent high-purity argon and allow down-time for maintenance of the recondensers and the refrigeration plants. The condensed stream of LAr will be combined with the circulating stream of LAr and purified prior to being returned to the cryostat.

The heat load introduced to the cryostat for the initial circulation rate will increase the total heat ingress significantly, in fact beyond the nominal 60-65 kW per refrigerator capacity. Therefore, additional refrigeration is required during the commissioning phase. However the refrigeration plant and recondenser have been provided on a fully redundant basis to ensure high availability. During the initial period the plants will be operated in parallel to provide the increased cooling capacity that will be required.

2.9.5 Argon Purification

Since the tank is designed without penetrations below the liquid level, pumps must be used to transfer LAr from the cryostat. Vertical submersible pumps will be inserted into pump

wells extending down from the cryostat roof. The pump suction must be located a minimum distance (normally ~ 1.5 to 2 m) below the lowest liquid level at which they are to pump in order to prevent cavitation and vapour-entrapment. The pumps and pump wells will extend to the bottom of the cryostat. They could also be staggered at different elevations to allow flexibility in drawing liquid from different elevations. Vertical, submersible, cryogenic pumps are supplied by manufacturers such as Ebara and Carter Cryogenic Products.

The required flow rate of liquid argon to be sent for purification is expected to decrease over time. The initial maximum flow rate will be $136 \text{ m}^3/\text{hr}$ (600 gpm). The liquid-argon volume in one cryostat will turn over every five days at this rate. Longer term, the rate will decrease to $34 \text{ m}^3/\text{hr}$ with a turn-over rate of 20 days. As a point of comparison, ICARUS T600 has a maximum turn-over rate of eight to ten days. See Table 2-7 for a comparison of purification rates among other experiments and LBNE. To achieve the turn-down required between the short-term commissioning flow rate of $136 \text{ m}^3/\text{hr}$ (600 gpm) and the long-term operational flow rate of $34 \text{ m}^3/\text{hr}$ (150 gpm) through the purifiers, four removable $34 \text{ m}^3/\text{hr}$ (150 gpm) pumps will be located at the end of the cryostat. Placing the pumps at this end of the cryostat will keep space clear for TPC installation. The purification skids are located at the 20 m wide septum which separates the two cryostats. The multiple-pump arrangement will provide a very high level of redundancy, which will extend the maintenance-free operating period of the cryostat.

Since only one detector module at a time will require the $136 \text{ m}^3/\text{hr}$ (600 gpm) flows, it is proposed to purchase only six pumps. This is a significant cost savings to the project and satisfies all cool-down scenarios. Two pumps can be removed from the first module after purification and placed in the second. After fulfilling the purity requirements in the second module, two pumps may be removed from the second module and placed on the shelf as spares.

The purification system consists of three identical filter vessels, each containing molecular-sieve and copper media filters. Each filter is 2.4 m in diameter by 3.8 m tall. The filters are sized to provide effective media usage at low pressure drop (2 kPa or 0.3 psi) over the expected range of flow rates. Only one filter will be in service at any time. For the initial fill and purification this redundant equipment will allow:

- Staggered filter regeneration and cool-down to lower the utility load (electric heating and LN2 cooling)
- Short turn-around time for regenerated filter (2 days)
- Maintenance of the purification beds

and for long-term operations it will allow:

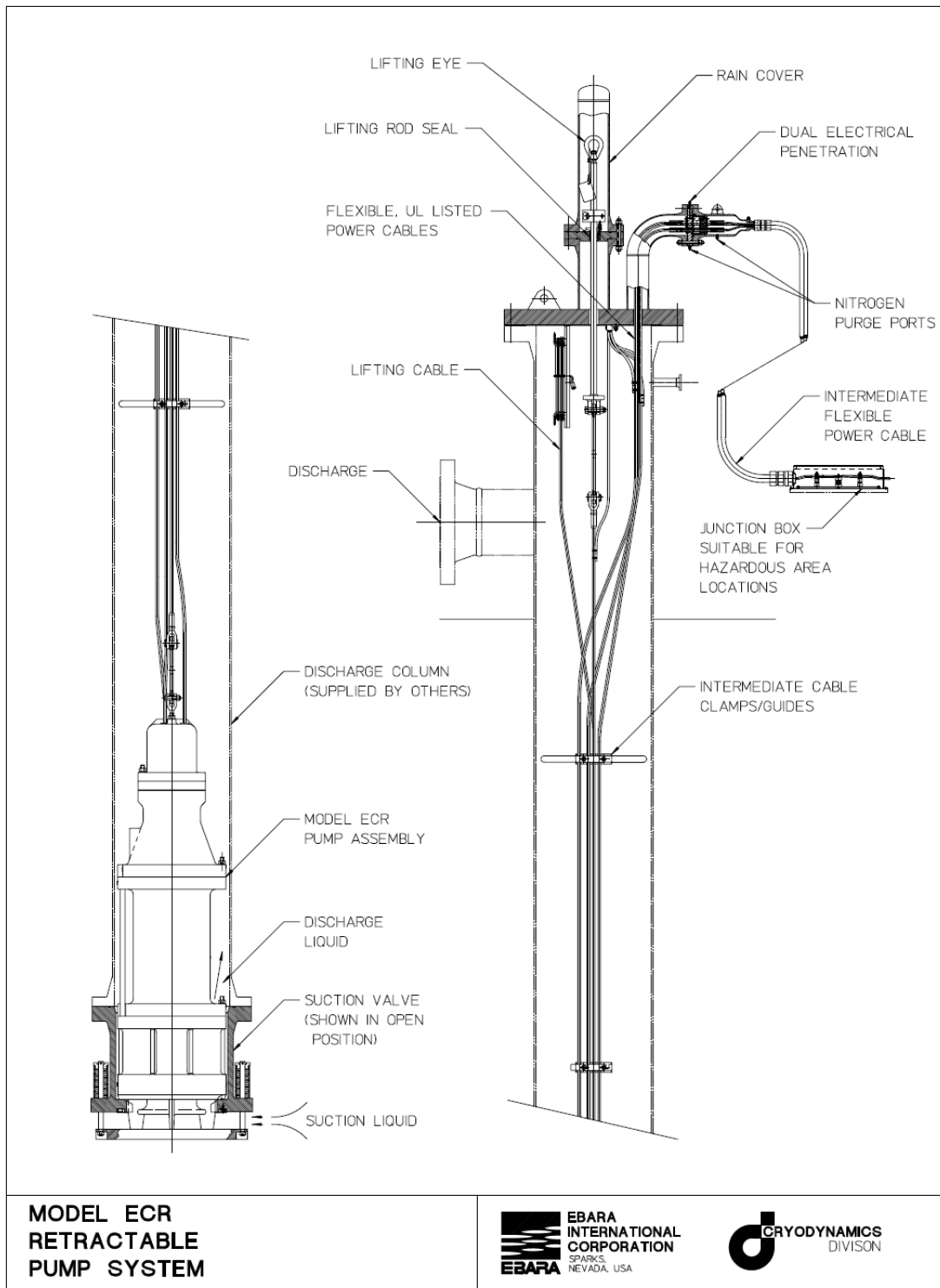


Figure 2-15: Vertical submersible pump

- High available capacity at the reduced flow rate
- Relaxation of regeneration time to minimize utility swings
- Possibility of filter maintenance or filter media replacement without disruption

The purifiers will be located close to the cryostat to minimize both the volume of LAr in the circulation pipework and the pump power required to achieve the desired flow rate.

The cryostat liquid argon inventory is circulated through a purification filter to achieve and maintain the required purity. The purification filter, containing molecular sieve media to remove water and copper media to remove oxygen, will become saturated. The nearly saturated purification filter is regenerated to vent the contaminants. The liquid argon flow is switched to another purification filter for uninterrupted filtration. A purity monitor after the purification filter will monitor the filter effectiveness. (Purity monitors measuring electron lifetime will also be in the LAr bath and resident in the cryostat. It is a requirement that purity levels reach < 200 ppt oxygen equivalent to match the required electron lifetime of the detector).

The regeneration of a filter is done in several steps. A saturated purification filter is first warmed with heated argon gas to an elevated temperature driving the captured water into the gas. Hydrogen gas is generated and mixed with the circulating Argon gas up to 1.5% hydrogen by volume. The hydrogen reacts with the oxygen and makes water that is also released into the circulating argon gas. Argon gas is vented to purge water from the hot circulating gas. The hot filter full of regenerated media is cooled by circulating chilled argon gas. The circulating argon gas is chilled first with a heat exchanger using a commercial R-404A refrigeration unit until the filter is cold. The commercial refrigeration unit accumulates the R-404A liquid and shuts down. The filter is next cooled down to cryogenic temperatures by circulating argon gas chilled by a second heat exchanger with a liquid nitrogen coolant. This completes the regeneration steps for a purification filter. The filter is now ready to be switched into service or held cold until needed. Two spare purification filters are used with separate heating and cooling loops to reduce the usage rate of electricity and liquid nitrogen. This also reduces the stresses on heat exchangers by decreasing their temperature swings.

2.9.6 Pressure Control

2.9.6.1 Normal Operations

The pressure-control valves are sized and set to control the internal cryostat pressure under normal operating conditions to the nominal design pressure of 0.113 MPa. Fluctuations within the range 0.105 MPa (50 mbarg) to 0.120 MPa (200 mbarg) will be allowed. Ten percent excursions above or below these levels will set off alarms to alert the operator to

Table 2-7: Purification comparison data for LArTPCs

| Experiment | LAr Vol (m ³) | LAr Mass (tons) | Liquid Purif Rate (kg/hr) | Gas Purif Rate (kg/hr) | Vol Change Rate (days) | Electron Life(ms) |
|------------------|---------------------------|-----------------|---------------------------|------------------------|------------------------|-------------------|
| ICARUS T600 | 550 | 761 | 2766 | 168 | 10.8 | 1.8 |
| ICARUS prototype | 10 | 13.9 | 692 | 0.69 | 0.8 | 1.1 |
| MTS at FNAL | 0.25 | 0.35 | 167 | 5.56 | 0.1 | >5 |
| ArgoNeuT | 0.55 | 0.76 | 0 | 4.3 | 7.3 | 0.75 |
| MicroBooNE | 123 | 170 | 6875 | 83 | 1.0 | TBD |
| LAPD | 22.2 | 30.7 | 3791 | 46.7 | 0.3 | TBD |
| LBNE LAr-FD | 18887 | 25000 | 188000 | 1100 | 5.5 | 1.4 req'd |

Table 2-8: Important Pressure Values

| | |
|--|--------------------------------|
| Vessel ullage maximum operating pressure | 0.121 MPa, 200 mbarg, 2.9 psig |
| Relief valve set pressure | 0.125 MPa, 250 mbarg, 3.5 psig |
| Roof truss design working pressure | 0.135 MPa, 350 mbarg, 5.1 psig |

intervene. Further excursion may result in automatic (executive) actions. These actions may include stopping the LAr circulation pumps (to reduce the heat ingress to the cryostat), increasing the argon flow rate through the recondenser, increasing the LN flow through the recondenser vessel, powering down heat sources within the cryostat (e.g., detector electronics). Eventually, if the pressure continues to rise, it will trigger the pressure-relief valves to operate. Table 2-8 gives important pressure values.

The ability of the control system to maintain a set pressure is dependent on the size of pressure upsets (due to changes in flow, heat load, temperature, atmospheric pressure, etc.) and the volume of gas in the system. The reference design has 0.8 meters of gas at the top of the cryostat. This is 5% of the total argon volume and is the typical vapor fraction used for cryogenic storage vessels. Reaction time to changes in the heat load are slow, on the order of an hour. At the expected heat-load rate of 62.5 kW, and for an isolated or un-cooled cryostat, the rate of pressure rise would be 200 mbar (3.0 psi) per hour. We plan to provide two redundant pressure control valves to maintain the required pressure range, each sized to handle at least 1500 kg/hr of argon flow to the recondenser.

2.9.6.2 Overpressure Control

In addition to the normal-operation pressure-control system, it is planned to provide a cryostat overpressure-protection system. This will need to be a high-integrity, automatic, failsafe system capable of preventing catastrophic structural failure of the cryostat in the case of

excessive internal pressure.

The key active components of the planned system are pressure-relief valves (PRVs) located on the roof of the cryostat that will monitor the differential pressure between the inside and the outside of the cryostat and open rapidly when the differential pressure exceeds a preset value. A pressure sensing line is used to trigger a pilot valve which in turn opens the PRV. A pressurized reservoir of power fluid is provided to each valve to ensure that the valves will operate under all upset and/or shutdown scenarios. The PRVs are self-contained devices provided specially for tank protection; they are not normally part of the control system.

The installation of the PRVs will ensure that each valve can periodically be isolated and tested for correct operation. The valves must be removable from service for maintenance or replacement without impacting the overall containment envelope of the cryostat or the integrity of the over-pressure protection system. This normally requires the inclusion of isolation valves upstream and downstream of the pressure-relief valves and at least one spare installed relief valve ($n + 1$ provision).

When the valves open, argon is released, the pressure within the cryostat falls and argon gas discharges into the argon vent riser. The valves are designed to close when the pressure returns below the preset level.

2.9.6.3 Vacuum-Relief System

The cryostat vacuum-relief system is a high-integrity, automatic, failsafe system designed to prevent catastrophic structural failure of the cryostat due to low internal pressure. The vacuum-relief system protects the primary membrane tank. Activation of this system is a non-routine operation and is not anticipated to occur during the life of the cryostat.

Potential causes of reduced pressure in the cryostat include operation of discharge pumps while the liquid-return inlet valves are shut, gaseous argon condensing in the recondenser (a thermo-siphon effect) or a failure of the vent system when draining the cryostat. Vacuum-relief valves are provided on LNG/LPG storage tanks to protect the structure from these types of events.

The key active components of this additional protection system are vacuum-relief valves located on the roof of the cryostat that will monitor the differential pressure between the inside and the outside of the cryostat and open when the differential pressure exceeds a preset value, allowing cavern air to enter the cryostat to restore a safe pressure.

2.9.7 LN Refrigeration System

Three commercial LN-refrigeration plants will be procured for LAr-FD. Each cryostat will have a dedicated LN plant for steady-state operations. A third plant will be used as a stand-by unit. The plants will be located in the cavern in the 4850L configuration. Each will be a closed-loop system supplying LN to the argon recondenser. The nominal rating of the quoted refrigerators is in the range of 60-65 kW.

Two-phase nitrogen is delivered from the cold end of the refrigerator into a 50 m³ buffer tank in the cavern. Pure liquid is withdrawn from the buffer tank and is supplied via transfer line to a pressure-reducing valve and phase-separator tank also located within the cavern. LN is then withdrawn from the bottom of the phase-separator tank, at a pressure of 2.0 bar and temperature of 84K, and directed to the recondenser. This results in a 5K temperature difference relative to the 89K argon recondenser temperature. The buffer tank size was chosen because it is one of the largest available vertical dewar volumes available. The chosen 50 m³ size will allow for up to forty hours of refrigeration time. This time window is adequate to cover most power outages, refrigerator performance problems and refrigerator switch-overs.

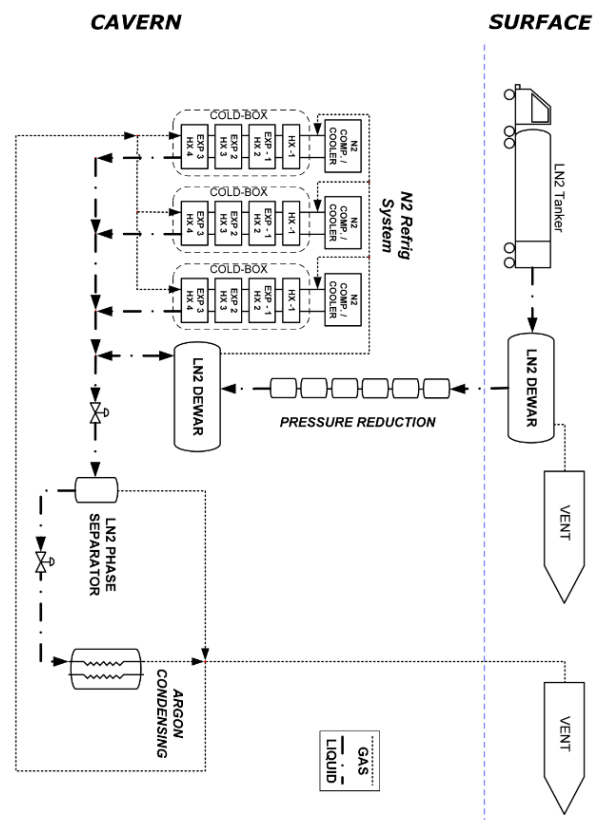


Figure 2-16: Nitrogen refrigeration-plant flow diagram

The refrigeration system operation, illustrated in Figure 2-16, is based on a screw compressor package and three turbo expanders. This system is expected to be capable of running continuously for at least a year, and then require only minor servicing. The system will be equipped with automatic controls and a remote monitoring so that no operator will be required during normal operation. Estimated maximum power requirement is 750 hp (560 kW), not taking into account the power generated by the expanders. The LAr-FD reference design places the nitrogen compressor in a surface-level equipment building. A closed-loop water system with evaporative-cooling tower removes heat from the compressor. Compression is carried out at close-to-ambient temperature. A compressor aftercooler is provided to reject heat.

The fluid is next routed to a 'cold box' consisting of four heat exchangers. This series of exchangers provides staged heat transfer from a cooling nitrogen stream to a warming one. The expanders are connected between the heat exchangers to progressively reduce the pressure of the cooling nitrogen stream to isentropically reduce the pressure and temperature of the nitrogen stream, eventually leading to a large liquid-nitrogen fraction at the coldest end of the cold box.

The main cold box shell is 1.22 m (4 ft) in diameter and 8.2 m (27 ft) tall. The expanders are adjacent to the cold box at three elevations and extend about 1 m to the side of the cold-box shell. The cold box will weigh 5670 kg. The compressors are located at the surface inside an equipment shed. The compressor skid (frame) is 4.3 m long, 1.8 m wide and 2.7 m tall and will weigh approximately 3630 kg.

2.9.8 Refrigeration Load Scenarios

In order to determine the optimal plant capacity and number of plants required, a plan has been developed which forecasts the LN refrigeration loads and plant capacity needed over seven scenarios. Those scenarios are described below and summary tables taken from Arup Concept Report LAr-FD at 800L are shown in Figures 2-18 and 2-19. The conclusion points to the requirement of three 60 to 65-kW plants. Each of these plants can achieve a 20% turn up or turn down thereby allowing a minimum single-plant capacity of 72 kW. Scenario 4 imposes the most severe requirement. In this scenario, all three plants will be required to operate at a duty cycle of approximately 59%. This scenario includes one completely filled and purified cryostat while the second is being purified and requiring frequent filter filtration.

Scenario 1 The initial operation will be the purging, cooling and filling of the first cryostat. The surface LAr and LN dewars will be operational and the cooling load for the dewars will come directly from the refrigeration plant. The pipework and vessels will be cold, the LAr in the cryostat will be circulating at high flow rate through the purification plant, and the cryostat will be cold. Both recondensers on the cryostat being filled will be in service to remove this heat. The LAr-FD subproject has assumed that the cryostat cool-down rate will be governed by the residual capacity of the two

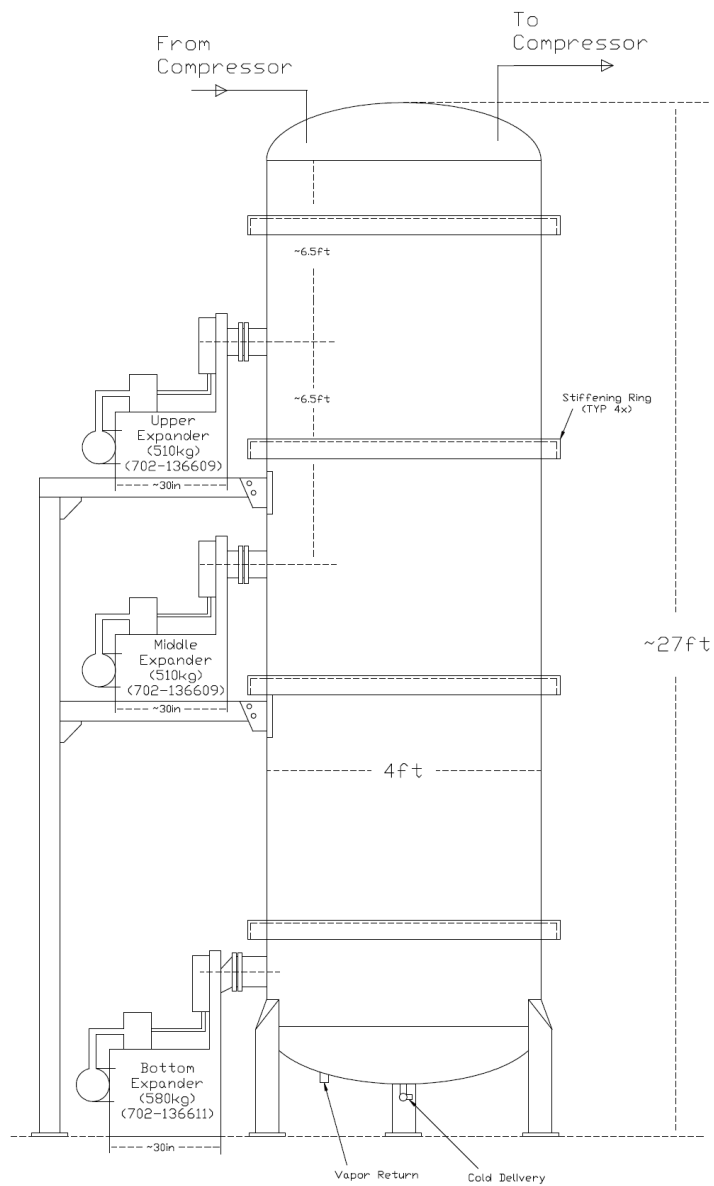


Figure 2-17: Nitrogen refrigeration plant

recondensers operating at 110% of their rated load. Each recondenser has been sized to accommodate the long-term refrigeration load associated with the cryostat. As the LAr is circulated to achieve the operational purity the filtration plant will need to be regularly regenerated. This will mean that the associated refrigeration load will normally be present.

Scenario 2 Once the first cryostat is filled with LAr the cool-down load will reduce to zero and the cryogenic plant will run for several months purifying the LAr inventory.

Scenario 3 When the LAr in the cryostat reaches the required purity level, the circulation flow rate will be reduced and the detector electronics will be turned on. At this stage the recondenser refrigeration load falls such that only one recondenser is required and the second unit can operate as a spare unit.

Scenario 4 The first cryostat continues to operate in normal experimental mode while the second cryostat is being purged, cooled down and filled with LAr.

Scenario 5 The second cryostat is full and LAr is circulated at high flowrate through the purification plant. The first cryostat continues to operate as normally.

Scenario 6 Both cryostats are operating in normal experimental mode. A spare recondenser is available on each cryostat to facilitate maintenance.

Scenario 7 It is assumed that a total failure of the refrigeration plant has occurred. All non-critical heat sources are isolated and liquid nitrogen from the surface dewar is utilized to recondense the inventory of high purity LAr. When the liquid nitrogen reservoir is exhausted an additional supply will be required or the high purity argon will need to be vented. In the locked-down state the circulation pumps and the purification plants are shut down.

2.10 Prototyping Plans

The development of the LAr-FD from conceptual to preliminary design will include a prototyping program. The most significant issue to resolve is whether a membrane cryostat the size of LAr-FD can achieve the required electron-drift lifetime. The Liquid Argon Purity Demonstrator (LAPD), already in progress and introduced in Section 8.3.3, is an off-project prototype being built as part of a development program at Fermilab to study the scaling of LArTPCs to kiloton sizes. In addition, two membrane-cryostat prototypes are being developed as an LBNE effort to confirm, among other things, that at larger scales the required LAr purity can still be achieved. The prototypes will be used to repeat the purge process accomplished in the LAPD to confirm that initial evacuation of the cryostat is unnecessary. They will also seek to confirm that an LAr purity level sufficient to enable the required drift times in a membrane cryostat can be achieved

| Heat Demand | Unit Loads (kW) | Scenarios | | | | | | |
|---|-----------------|-----------|------|------|------|------|------|------|
| | | 1 | 2 | 3 | 4 | 5 | 6 | 7 |
| Recondenser Load (1st Cryostat) | | | | | | | | |
| Cryostat (W) Heat Ingress | 28.7 | 28.7 | 28.7 | 28.7 | 28.7 | 28.7 | 28.7 | 28.7 |
| Recirculation Pump in cryostat W (1 off) | 8.3 | | | 8.3 | 8.3 | 8.3 | 8.3 | |
| Recirculation Pump in cryostat W (4 off) | 33.3 | 33.3 | | | | | | |
| Piping and Purification vessel Heat ingress (W) | 2 | 2.0 | 2.0 | 2.0 | 2.0 | 2.0 | 2.0 | 2.0 |
| Detector Electronics in cryostat W | 4.7 | | | 4.7 | 4.7 | 4.7 | 4.7 | 4.7 |
| Cryostat Cooldown load | | 27.7 (1) | | | | | | |
| Recondenser Load (kW) | | 92 | 64.0 | 43.7 | 43.7 | 43.7 | 43.7 | 28.7 |
| Recondensers in Operation | | 2 | 2 | 1 | 1 | 1 | 1 | 1 |
| Recondenser Duty (min) | | 46 | 32 | 43.7 | 43.7 | 43.7 | 43.7 | 43.7 |
| Recondenser Load (2nd Cryostat) | | | | | | | | |
| Cryostat (E) Heat Ingress | 28.7 | | | | 28.7 | 28.7 | 28.7 | 28.7 |
| Recirculation Pump in cryostat E (1 off) | 8.3 | | | | | | 8.3 | |
| Recirculation Pump in cryostat E (4 off) | 33.3 | | | | 33.3 | 33.3 | | |
| Piping and Purification vessel Heat ingress (E) | 2 | | | | 2.0 | 2.0 | 2.0 | 2.0 |
| Detector Electronics in cryostat E | 4.7 | | | | | | | 4.7 |

Figure 2-18: Refrigeration loads

| Heat Demand | Unit Loads (kW) | Scenarios | | | | | | |
|-------------------------------------|-----------------|-----------|------|------|----------|-------|-------|------|
| | | 1 | 2 | 3 | 4 | 5 | 6 | 7 |
| Cryostat Cooldown load | | | | | 27.7 (1) | | | |
| Recondenser (E) Load | | | | | 92 | 64.0 | 43.7 | 28.7 |
| Condensers (E) in Operation | | | | | 2 | 2 | 1 | 1 |
| Condenser Duty (minimum) | | | | | 46 | 32 | 43.7 | 43.7 |
| Dewar Refrigeration Load | | | | | | | | |
| LAr Storage dewar recondenser | 2 | 2 | 2 | 2 | 2 | 2 | 2 | 2 |
| LN Storage dewar recondenser | 2 | 2 | 2 | 2 | 2 | 2 | 2 | 2 |
| Total Continuous Refrigeration Load | | 94 | 68.0 | 47.7 | 139.7 | 111.7 | 91.4 | 61.4 |
| Refrigeration Plants in Operation | | 2 | 2 | 2 | 3 | 2 | 2 | 0 |
| Intermittent Refrigeration Loads | | | | | | | | |
| Purifier Regeneration Load | 30 | 30 | 30 | 30 | 30 | 30 | 30 | |
| LAr Sub cooling during Filling | 8 | 8 | | | 8 | | | |
| Peak Refrigeration Load | | 132 | 98.0 | 77.7 | 177.7 | 141.7 | 121.4 | 61.4 |
| Refrigeration Plants in Operation | | 2 | 2 | 2 | 3 | 2 | 2 | 0 |
| Duty per plant | | 66 | 49 | 39 | 59 | 71 | 61 | |

Figure 2-19: Refrigeration loads, continued

2.11 ES&H

During all phases of LAr-FD and the proposed prototypes, Fermilab ES&H standards and Sanford Laboratory ES&H codes and standards will guide the design, procurement and installation phases of the project. Particular attention will be paid to critical sections of Chapter 5000 [17] relating to ODH, standards for piping construction and vessel design. The planned work process will provide for reviews throughout all phases of the project to guarantee stringent adherence to the safety requirements. Requirements on the membrane-cryostat materials and their fabrication will be strictly outlined in the specification documents. Close communication between the vendors, Fermilab's cryogenic and process engineers, and Fermilab and Sanford Laboratory ES&H personnel will be maintained at all times.

3 Time Projection Chamber and Electronics (WBS 130.05.04.01)

The scope of the Time Projection Chamber (TPC) subsystem includes the design, procurement, fabrication, testing, delivery and installation of the mechanical components of the TPC:

- anode plane assemblies
- cathode plane assemblies
- field cage

and of all the in-vessel electronics, signal and power cables, their feedthroughs, as well as the low- and high-voltage power supplies feeding the electronics. This chapter describes the reference design for the TPC that meets the required performance for charge collection in the LBNE liquid argon detector, LAr-FD.

3.1 Introduction

The Time Projection Chamber (TPC) is the active detector element of LAr-FD. It is located inside the cryostat vessel and is completely submerged in liquid argon at 89 K. The TPC consists of alternating anode plane assemblies (APAs) and cathode plane assemblies (CPAs), with field-cage panels enclosing the four open sides between the anode and cathode planes. A uniform electric field is created in volume between the anode and cathode planes. A charged particle traversing this volume leaves a trail of ionization. The electrons drift toward the anode wire planes, inducing electric current signals in the front-end electronic circuits connected to the sensing wires.

TPC subsystem interfaces to the cryostat and cryogenic subsystem through the TPC mounting fixtures, and it interfaces with the DAQ subsystem through the signal feedthroughs.

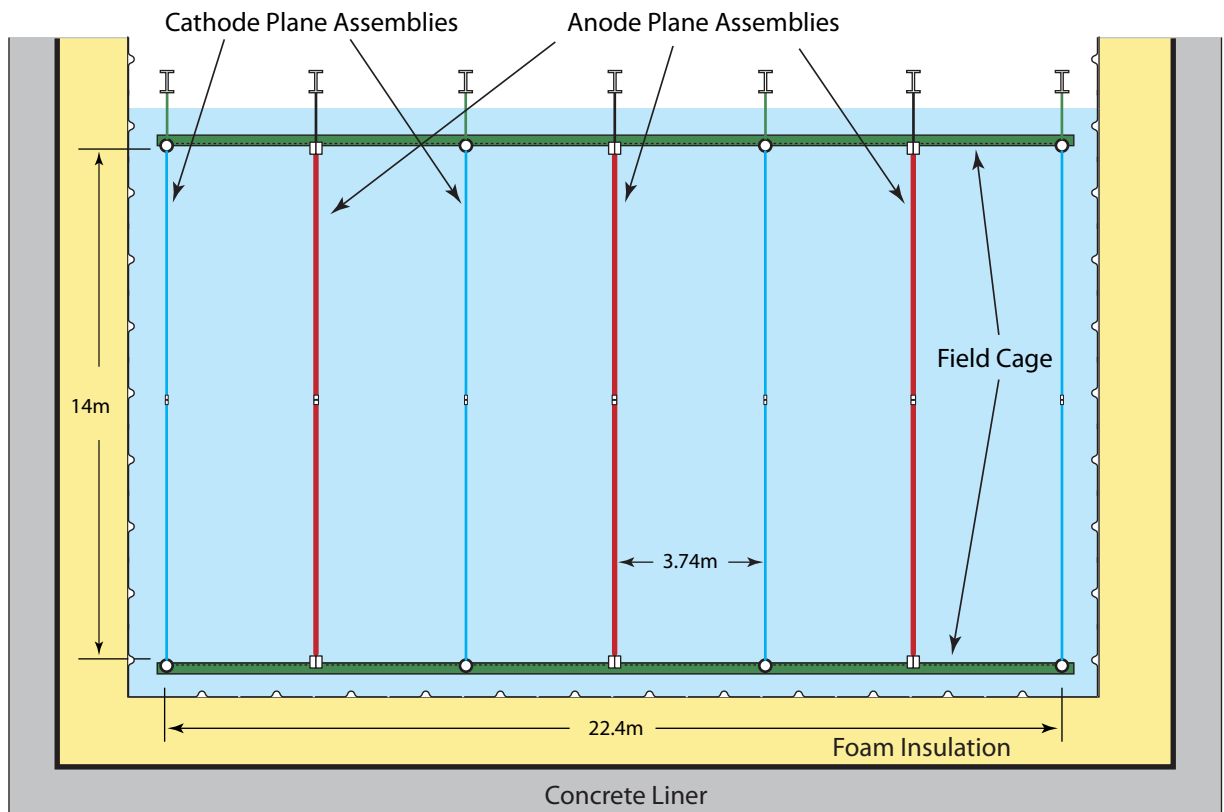


Figure 3-1: Cross section of the TPC inside the cryostat. The length of the TPC is 45.6 m, along the direction of the neutrino beam (into the paper).

The TPC's active volume (Figure 3-1) is 14 m high, 22.4 m wide and 45.6 m long in the beam direction. Its four rows of CPA planes interleaved with three rows of APA planes are oriented vertically, parallel to the beamline with the electric field applied perpendicular to the planes. The maximum electron-drift distance between a cathode and an adjacent anode is 3.7 m. Both the cathode and anode plane assemblies are 2.5 m wide and 7 m high. Two 7 m modules (either APA or CPA) stack vertically to instrument the 14 m active depth. In each row, 18 such stacks are placed edge-to-edge along the beam direction, forming the 45.6 m active length of the detector. Each cryostat houses a total of 108 APAs and 144 CPAs. Each facing pair of cathode and anode rows are surrounded by a "field cage," assembled from panels of FR-4 sheets with parallel copper strips connected to resistive divider networks.

On each APA, four planes of wires cover each side of a frame (the "wire frame"). See Figure 3-2. The inner three planes of wires are oriented, going from the inside out: vertically, and at $\sim \pm 45^\circ$ to the vertical, respectively. Each wire is connected to a front-end readout channel. The wires on the outermost plane are oriented vertically, and are not connected to the readout electronics. At a nominal wire pitch (center-to-center separation) of 4.5 mm, the total number of readout channels in an APA is 2560, for a total of 276,480 in each cryostat.

The readout electronics are optimized for operation in the cryogenic environment. The front-end ASIC chip utilizes a mixed-signal design. It has 16 channels of preamp, a shaper and an analog-to-digital converter (ADC) in its analog section, followed by a large, shared buffer and the digital IO interface. Eight such chips are mounted on a single readout board, instrumenting 128 adjacent wires in one plane. A digital-processing ASIC with an 8:1 multiplexer on this board further increases the multiplexing factor to 128:1, resulting in a single output channel. Data from this output channel will be transmitted through two redundant LVDS (low-voltage differential signaling) output cables to two APA-level multiplexing boards. Each of these boards connects to the 20 front-end readout boards on the APA, and provides a further 20:1 multiplexing. The output from each of these APA-level multiplexing boards are again duplicated for redundancy.

One cable bundle is planned for each APA to connect to the outside of the cryostat. The bundle will consist of wires for low-voltage power, wire-bias voltages, data out, clock in, digital control IO and an analog monitoring output. Redundant cables will be provided for many of these functions. The 108 cable bundles will be connected through 27 feedthroughs distributed on the roof of the cryostat. Additional feedthroughs will be required for redundant high-voltage connections to the CPAs.

3.2 Design Considerations

The requirements for the TPC can be found in the requirements documentation [12]. The most significant ones are the following:

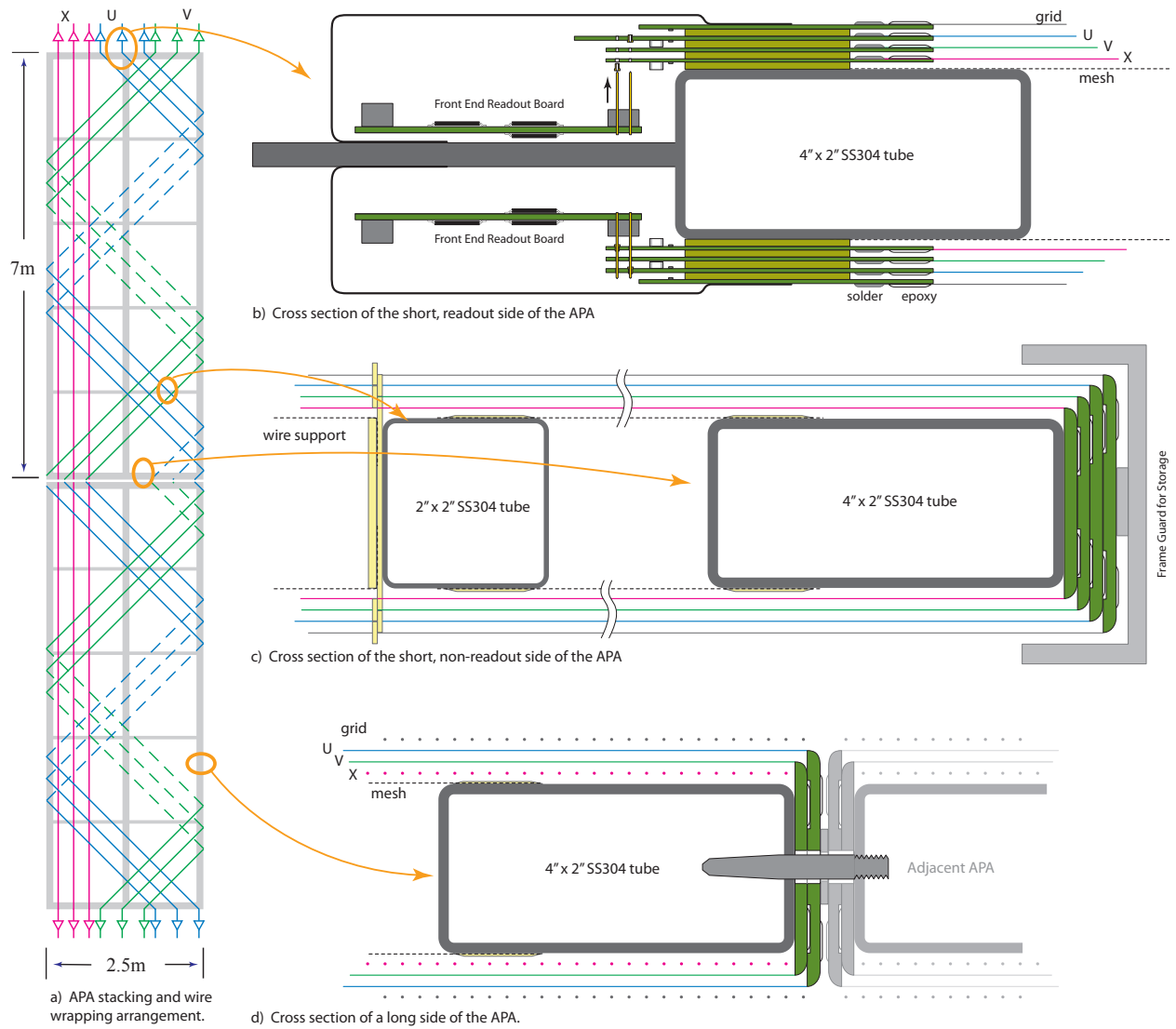


Figure 3-2: Illustration of the APA wire wrapping scheme, and three cross sectional views.

- Provide the means to detect charged particles in the detector and transmit the detector signals to the Data Acquisition System (DAQ)
- Meet the physics requirement for electron/photon discrimination; the TPC wire spacing will be < 5 mm
- Limit variation in the wire sag to < 0.5 mm such that it does not significantly impact the position and energy resolution of the detector
- Provide redundancy in the discrimination of electrons from photon conversions and ensure long-term reliability over the life of the experiment; configuration will use three instrumented wire planes
- Optimize the measurement of high-energy and low-energy tracks from accelerator-neutrino interactions; the wire-plane orientation is optimized for neutrinos in the LBNE energy range
- Enable the detector to distinguish a Minimum Ionizing Particle (MIP) from noise with a signal-to-noise ratio $> 9:1$
- Enable the detector to measure the ionization up to 15 times that of a MIP particle; this is necessary to perform particle identification of stopping kaons from proton decay
- Enable the in-vessel electronics to operate for the life of the facility
- Record the wire-signal waveforms continuously without dead time
- Use only materials that are compatible with high-purity liquid argon

3.3 Anode Plane Assemblies (WBS 130.05.04.02)

The APAs are 2.5 m wide, 7 m long, and ~ 9 cm thick. The 7-m length is chosen to limit the maximum wire length to 10 m (which caps the input capacitance to the preamps to ~ 200 pF) at a 45° wire angle, and the 2.5-m width is set to fit in a standard HiCube container for storage and transport. Each APA is constructed from a framework of light-weight, stainless-steel rectangular tubing, with four layers of wires wrapped over both sides of the frame. The front-end electronics boards are mounted on one end of the wire frame and protected by a metal enclosure.

3.3.1 Wires

The wires used in the TPC must provide:

- High break load to withstand the applied tension
- Good conductivity to minimize noise contribution to the front-end electronics
- Comparable thermal-expansion coefficient to that of the stainless-steel frame to avoid tension change after cool-down

Both stainless-steel and copper-beryllium (CuBe) wires are potential candidates. Stainless steel was the choice of ICARUS, while a copper-plated stainless-steel wire was chosen by MicroBooNE (to reduce resistance). Both experiments use a wire-termination technique that is labor-intensive and impractical for LAr-FD. Previous experience from FNAL [18] has shown that a CuBe wire under tension can be reliably bonded to a copper-clad G10/FR4 (glass epoxy material) surface by a combination of epoxy (mechanical bonding) and solder (electrical connection). This bonding technique greatly simplifies the electrical connection to the readout electronics and it can be easily automated with commercial equipment. Therefore CuBe wire is selected as the reference design wire of choice.

At 150 μm diameter, the breaking tension of a hardened CuBe wire is ~ 30 N. To ensure no wire breakage in the TPC, e.g. during cryostat cool-down, the nominal operating tension of the wire will be set at 5 N. Periodic support structures on the wire frame will limit the unsupported wire length to less than 2 m, resulting in less than 0.2 mm deflection due to gravitational or electrostatic forces. Wire ends will be glued and soldered (if electrical connection is needed) onto printed circuit boards attached to the wire frame.

3.3.2 Wire Planes

Four planes of wires are installed on each side of an APA as shown in Figure 3-2. A nominal wire pitch of 4.5 mm is selected to meet the position resolution and signal-to-noise ratio requirement. The distance between wire planes is set to 4.8 mm (3/16 in) to use standard printed circuit board thickness, while maintaining optimal signal formation. These four planes (along the direction of electron drift) are labeled as: the *grid plane*, the *first induction plane* (U), the *second induction plane* (V), and the *collection plane* (X). The wires on the grid and the collection planes are vertically oriented, while the two induction planes are oriented at $\sim \pm 45^\circ$ to the vertical. This wire layout is shown to be the best for reconstructing beam-neutrino events [19]. The wires on the grid plane are not connected to the readout electronics; they shield the first induction wire plane from being influenced by distant ionizations. The four wire planes will be electrically biased so that electrons from an ionizing-particle track completely drift past the first three planes and are collected by the fourth plane. Calculations show that the minimum bias voltages needed to achieve this goal are $V_G = -665\text{V}$, $V_U = -370\text{V}$, $V_V = 0\text{V}$ and $V_X = 820\text{V}$ respectively. A grounded mesh plane, located 4.8 mm behind the collection plane, prevents the electric field around this set of wires from being distorted by the metal frame structure and the wires on the

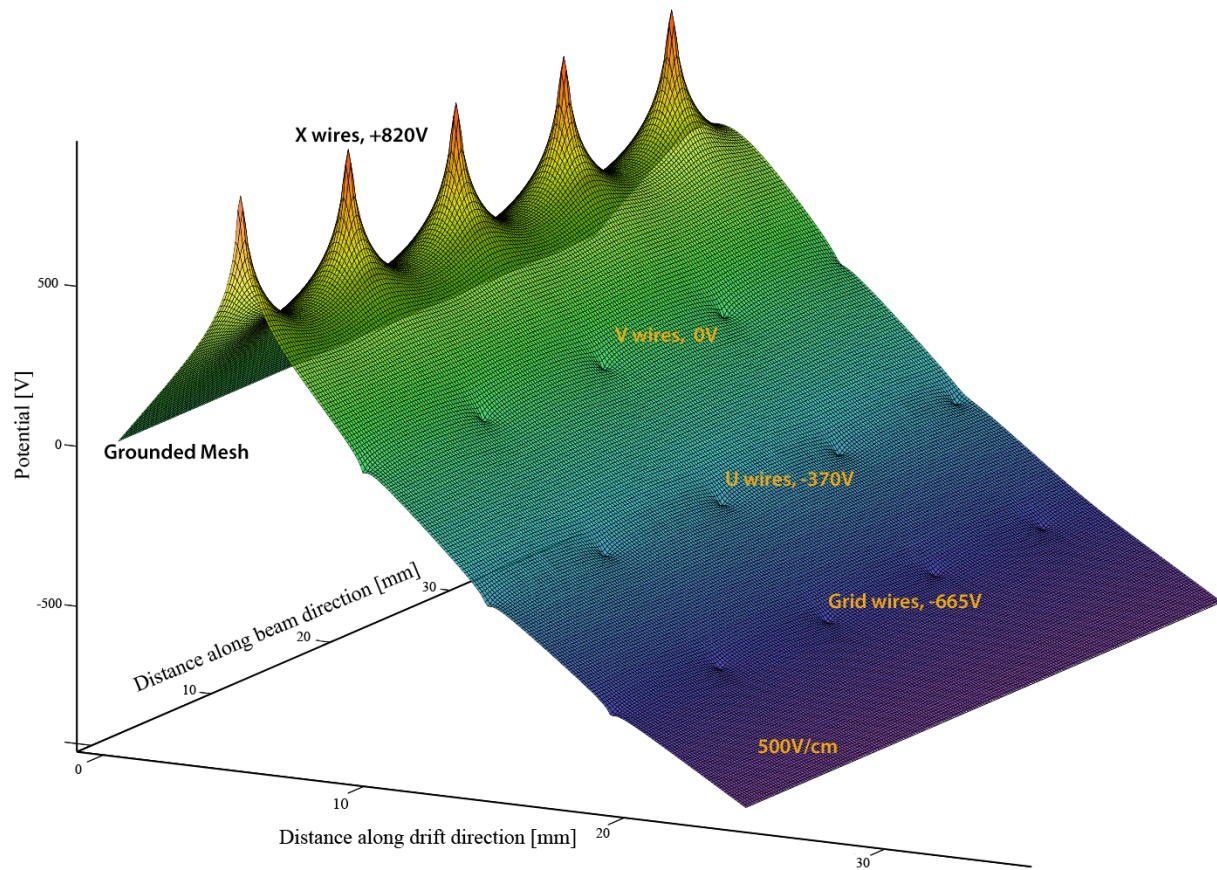


Figure 3-3: A surface plot of the electric potential distribution near the wire planes. The voltages on the wire planes are biased to provide complete electron transparency through the first three planes, and complete collection on the fourth plane.

opposite side of the frame. It also shields the sensing wires from potential EM interferences from the photomultiplier tubes (PMTs), discussed in Chapter 5, mounted within the frame. The mesh should have a wire pitch less than 2 mm to ensure a uniform electric field and a high optical transparency. Figure 3-3 shows the electric potential distribution near the APA frame with the wire planes biased with the appropriate voltages.

The V wire plane is directly connected to the front-end electronics, i.e. $V_V = 0V$, to simplify the coupling and reduce the maximum bias voltages on the other planes. The wires on the two induction planes (U & V) are wrapped in a helical pattern around the long edges of the wire frame (Fig.3-2a). This technique makes it possible to place readout electronics only at one short edge of a wire frame, enabling joining the APAs on the other three sides with minimal dead space. It slightly complicates the track reconstruction because the U & V wires are sensitive to tracks on both sides of the APA. The upper APAs in the cryostat will have their readouts at the top edge of the frame (as shown in Figure 3-2), while the lower APAs will mount their electronics at the bottom edge.

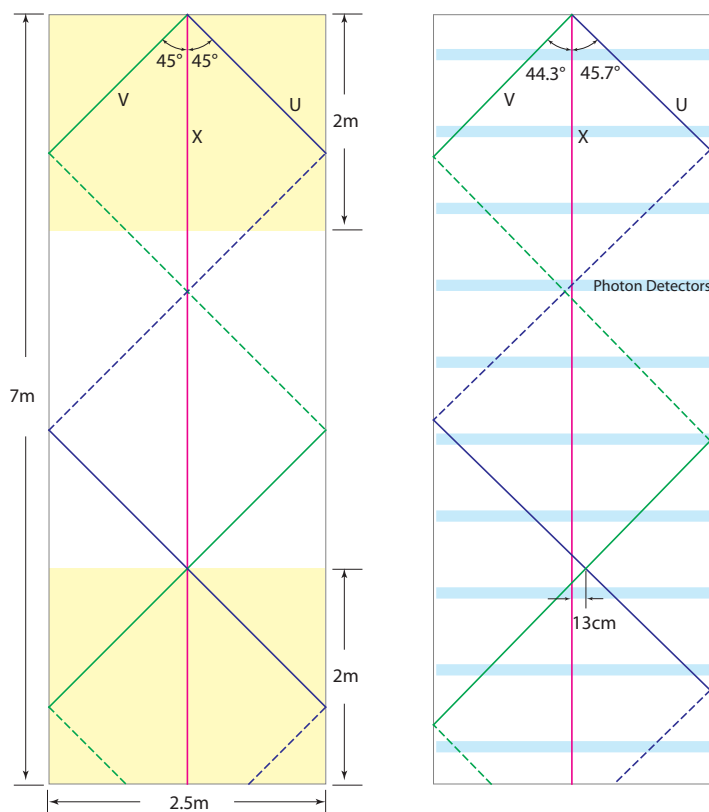


Figure 3-4: Left: illustration of the ambiguity problem if the U & V wire angles are equal; Right: slightly changing the wire angles and photo detectors help to resolve the problem.

If the wire angles of the U & V planes are equal to 45° , both wires wrap around the frame one complete cycle, covering a 5-m height, then repeat the same pattern for another 2 m. An ambiguity problem arises when wire pattern of all three sensing planes in the top 2-m

section is identical to that in the bottom 2-m section. The detector will not be able to tell if a track is in the top or the bottom section of an APA (see Figure 3-4). To resolve this ambiguity, the wire angles of U & V planes are set slightly differently: $45^\circ \pm \delta$, such that the same three wires do not cross again. In addition, the multiple photon detectors embedded in the APA frame will help to identify the vertical location of an ionizing track.

The angles and pitches of the U & V wires are chosen such that (1) a modularity of 128 channels form at the readout end of the APA, with the X wire pitch at 4.5mm; (2) a modularity on the side wrapping boards form with a module length similar to that of the readout board. The current configuration has the U wires at 45.7° from vertical, at a 4.9-mm pitch, while the V wires at 44.3° from vertical, and 5.0-mm pitch.

The APA readout electronics are divided into 20 identical modules. Each module covers 128 readout wires, consisting of 56 X, 36 U and 36 V wires. These 128 readout wires span a width of 252 mm. Using this modularity, the APA's active width is set to 2520 mm. There are 1120 X wires, 720 U wires, 720 V wires, and 1120 grid wires for each APA. The total number of readout channels is 2560 per APA. The total number of wires per APA is 3680.

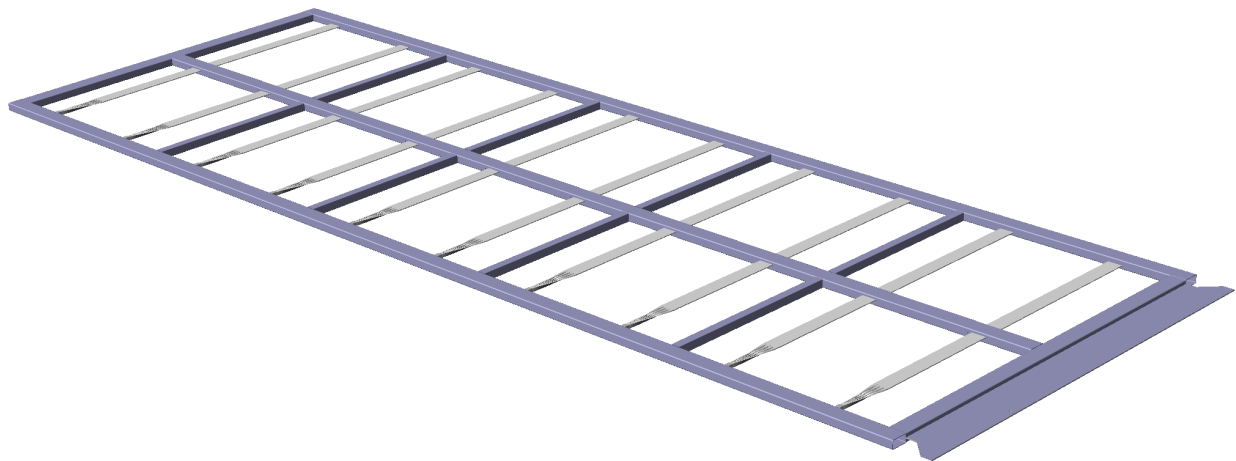


Figure 3-5: Conceptual design of a wire frame (shown without wires). The photon detectors are shown installed on the APA frame.

3.3.3 APA Frame

At a nominal wire tension of 5 N, the 3680 wires exert a force of ~ 6.4 kN/m on the short edges of the APA, and a ~ 2 kN/m force on the long edges. The wire frame must be able to withstand the wire tension with a minimal distortion, while minimizing the thickness of the frame to reduce the resulting dead space. A conceptual design of the wire frame is shown in Figure 3-5. It is constructed from all stainless-steel tubes welded in a jig. Finite element analysis has shown that the maximum distortion of the frame due to wire tension is under

0.5 mm. The total mass of a frame is ~ 250 kg. All hollow members of the frame are vented to prevent the creation of trapped volumes. The two long outer members of the frame are open-ended, so that signal and power cables can be threaded through them to reach the readout end of the lower APA.

3.3.4 Wire Wrapping Around an APA

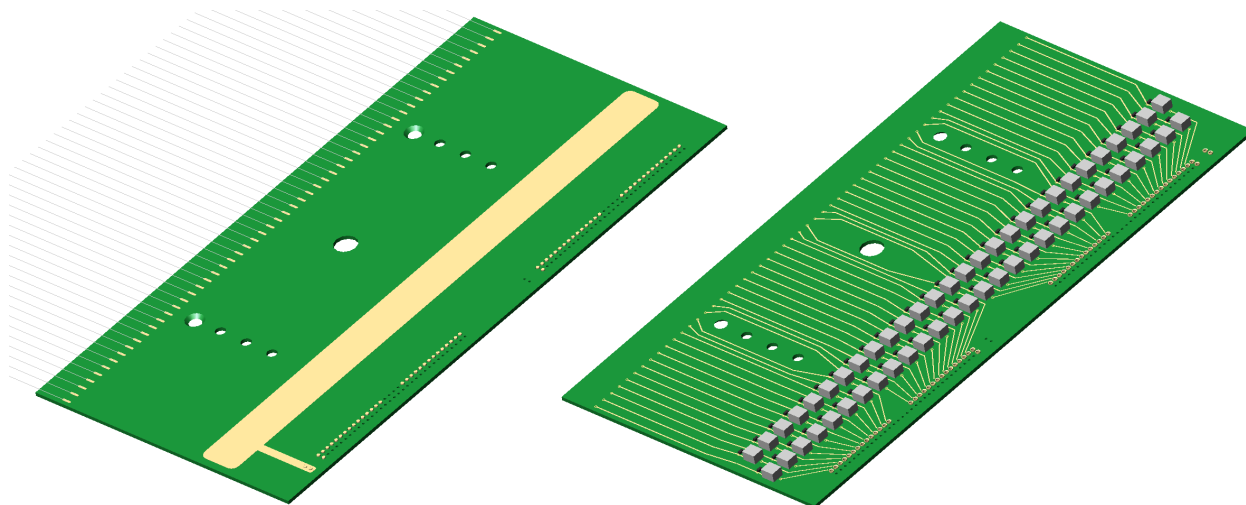


Figure 3-6: Conceptual design of a wire bonding board for the x wires. Left: An array of wires are glued on the leading edge of the top surface, and then soldered onto the soldering pads; Right: the bottom side of the board has the RC network for the bias voltage.

Figure 3-2b-d shows three major cross sections of an APA. The top figure is the cross section of the readout end of an APA. The four planes of wires are attached to their respective wire-bonding boards through a combination of epoxy and solder. Figure 3-6 shows both sides of the board for the x wires. The wires are wound on the top surface of the board (left) using a winding machine. The wires are then glued down with a bead of epoxy at the leading edge of the board. After the epoxy has cured, the wires are soldered onto the copper pads under each wire, and the wires are cut beyond the pads. On the bottom of this board, the copper traces connect the wires to the bias voltage supply through an RC network. The resistors in this network have values about $20\text{ M}\Omega$, such that in the event that a wire from a different plane breaks and is shorted to these wires, the bias voltages on the rest of the wires will not be affected. The AC-coupled signals from the wires are connected to sockets that will mate with the front-end readout boards.

Similar boards for the U & V planes are aligned and stacked above the X boards. An array of pins on the front-end readout boards is pushed through the stack of wire bonding boards, making electrical connection between the readout electronics and the matching wires. These readout boards, as described in Section 3.7, process the analog signals from the wires and transmit the digital information through feedthroughs to the DAQ system outside the

cryostat. The electronics on the readout boards dissipate an estimated ~ 60 W of heat per APA and may generate a small quantity of argon bubbles. Two stainless-steel covers are placed over the readout boards to contain the bubbles and direct them to the gas volume of the cryostat. In the case of the lower APAs, the bubbles, if not already recondensed, will be funneled through the vertical hollow frame members to the top of the cryostat.

Figure 3-2c shows the cross section of the short, non-readout end of an APA. All wires are mechanically terminated on this end on the four layers of wire-wrapping boards. No electrical connections are needed.

Figure 3-2d shows the cross section of a long edge of an APA. Only two layers (U & V) of wire wrapping boards are needed here. The wire-wrapping boards are made from printed circuit boards, shown in Figure 3-7. The boards are attached to the APA frame, and then a winding machine wraps a wire around the APA in a helical fashion, placing the wire into the grooves on the edges of each board. After winding, the wires are glued down to the wire-wrapping boards near the grooved edges. To clear the mounting holes on the board, a few wires are soldered onto the copper pads, and cut, leaving the copper traces to bridge the connection.

Figure 3-8 is a closeup view of a corner of an APA frame with some wires and various wire bonding boards to demonstrate the assembly.

After the grid plane wires are placed on the APA, metal guards are placed along the three wrapped edges of an APA. These guards protect the fragile wires during APA handling, storage and transport.

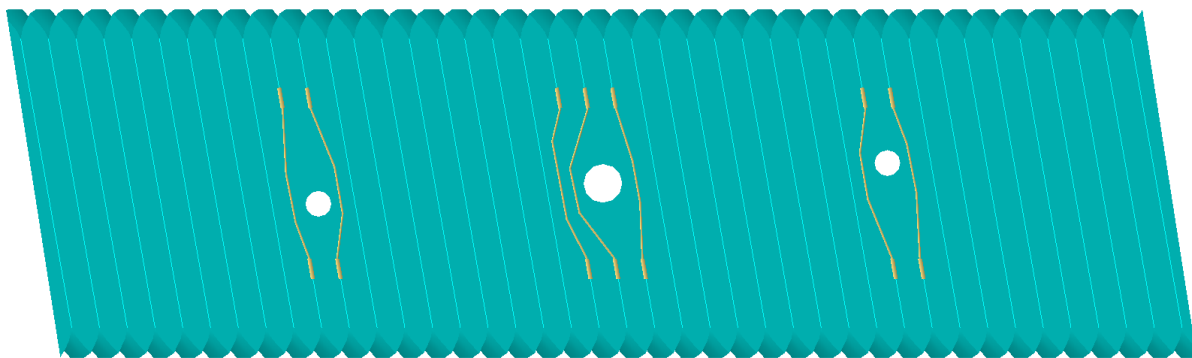


Figure 3-7: Conceptual design of a wire wrapping board for the U wires on a long edge of an APA. The light cyan colored lines represent the wires wrapped over the board surface. Some wires near the mounting holes must be soldered to the copper traces and then cut.

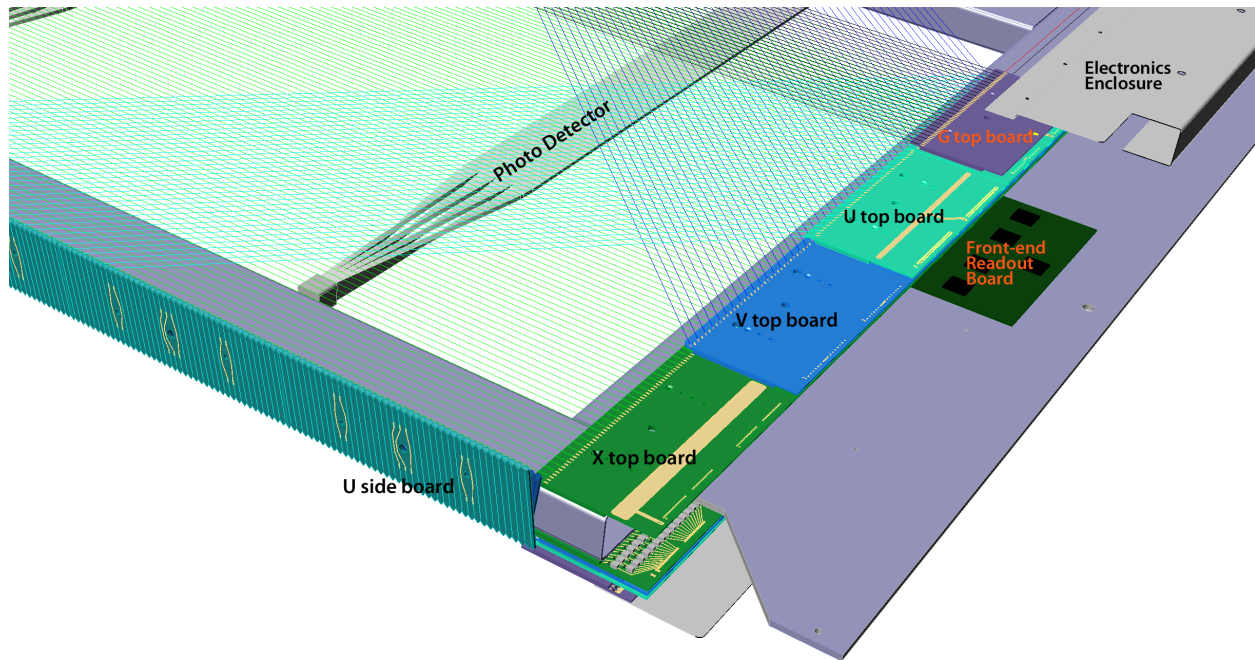


Figure 3–8: A closeup view of a partially assembled corner of an APA. For clarity, only every other wire is shown in this illustration.

3.3.5 Wire Supports on Inner Frame Members

The left of Figure 3–2b also shows the wire-support structure mounted on one of the inner horizontal frame members. A detailed rendering of this concept is illustrated in Figure 3–9. The support structure is composed of strips of thin fiberglass boards, with notches machined at specific intervals. The support strips for X plane is directly mounted on the inner frame members. After all X wires have been placed into the slits, the V support strip (shown in green) is glued onto the tips of the X strips, trapping the X wires in position. After the V wires are placed into the slits, the U support strip (identical to the V strip) is glued to the V strip, trapping the V wires. These structures are repeated four times along the 7-m length of an APA, limiting the unsupported wire length on any wire plane to < 2 m, while introducing only millimeter-scale dead regions. These wire supports play a key role in minimizing wire deflection due to gravity and electrostatic force, enabling the use of a moderate wire tension and reducing the risk of wire breakage.

3.3.6 Wire-Winding Machines

Two winding machines will be constructed to lay the 3680 wires on each APA. Their working concepts are illustrated in Figure 3–10. For the X and the grid wire planes, (left figure), the wire frame will be standing on one of its long edges, while a wire is wrapped around the frame

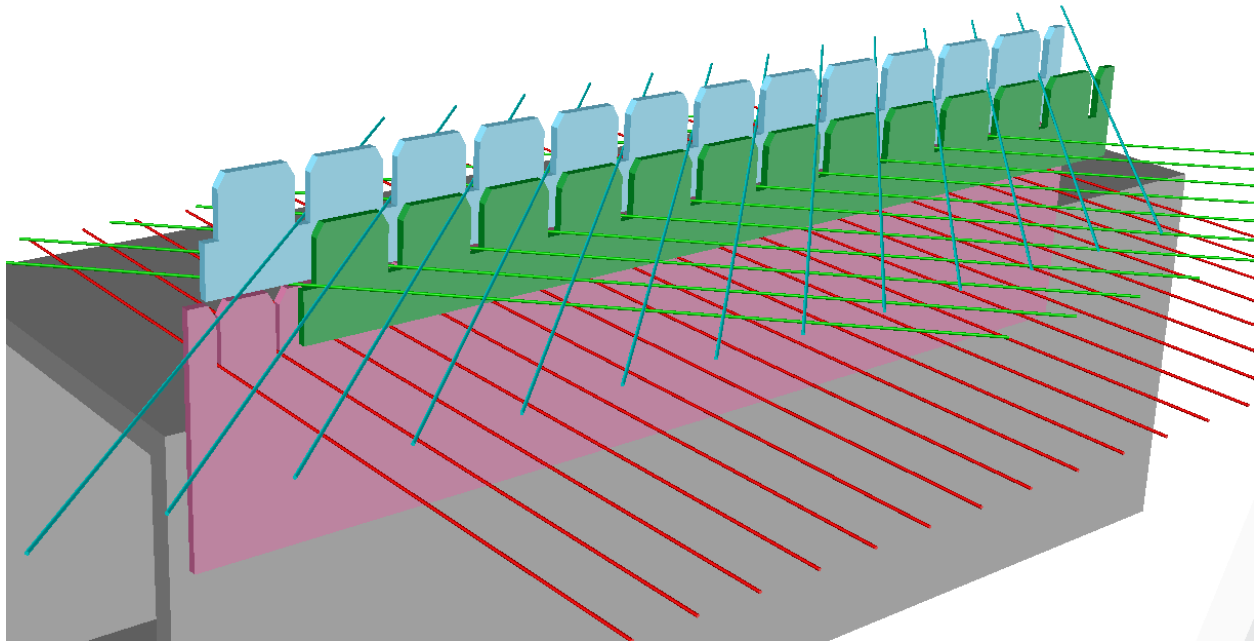


Figure 3-9: Conceptual design of the wire support for the U, V & X wires. Similar structures will be used to support the grid wire plane.

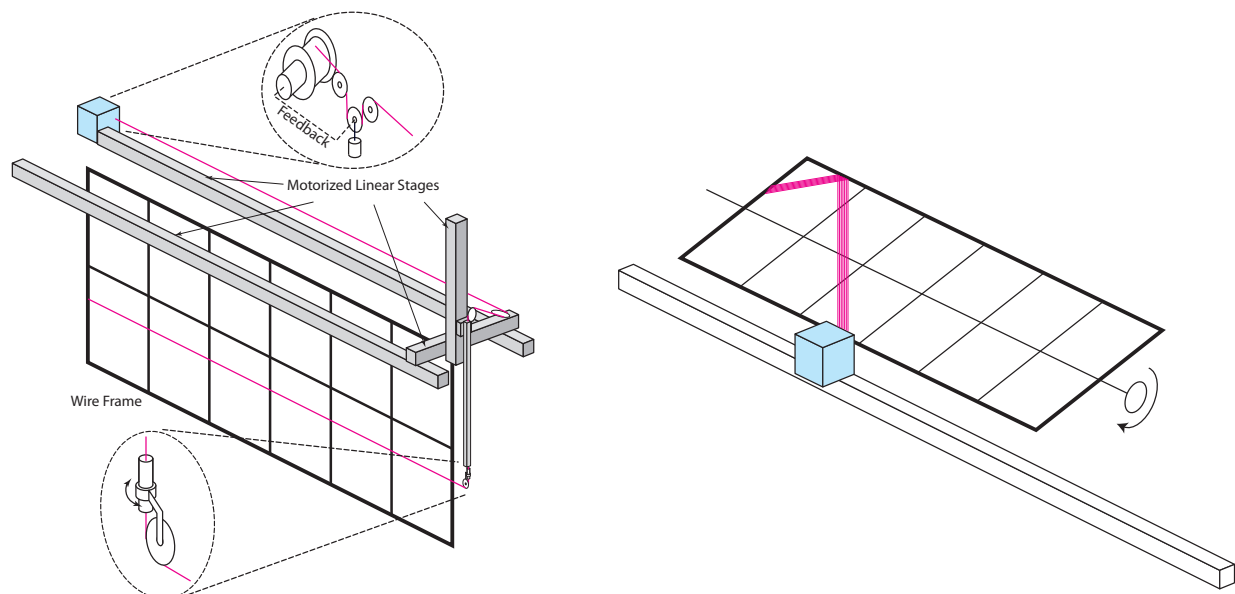


Figure 3-10: Two winding machine concepts. The left figure is for the X and grid wires. The right figure is for the U & V wires. The 'blue boxes' are wire spools with automatic tension control.

using a commercial 3-axis gantry robot. A wire-tension controller maintains the wire tension and feed rate of the wire off the spool. Although the entire plane of wires can be wound in one pass, a more fault-tolerant procedure is to pause the winding machine periodically and solder the last wire onto the frame. This intermediate soldering step will prevent the unraveling of a large section due to an accidental broken wire. The winding machine for the U & V planes will wrap a group of wires (~ 10) as a band in each pass. An automatic soldering robot will solder the wire ends after the group of wires has been laid down on one side of the frame. A wire-tension measuring device will scan the newly placed wires and record the wire tension of each wire. Any wires with abnormal tension will be replaced manually.

3.3.7 Alternative APA Construction

The APA design described above requires two customized machines capable of placing wires directly on the $2.5 \text{ m} \times 7 \text{ m}$ APA frame with a precision of the order of 1 mm. Cleaning the solder flux off the APA after wire bonding also poses a challenge on an object this size. An alternate APA design is under development to alleviate these difficulties. In this design, the APA frame dimensions and the wire geometry remain approximately the same, but the wires are attached to the APA frame in a two-step process.

The first step is to bond a group of wires (~ 64) with epoxy and solder on two FR-4 boards to form a wire module. The second step is to mount all the assembled wire modules onto an APA frame using a stretching table. The wire lengths in each wire module are determined in such a way that when a module is mounted on the frame, all wires will reach a uniform tension of 5 N. Jumper cables interconnect the corresponding U or V wire modules along the two long edges of an APA to complete the helical wire wrapping pattern electrically. The assembly of an APA frame using prefabricated modules will be easier and faster than direct winding on the big frame. If a wire is broken while being attached to the frame, it is relatively straightforward to replace the affected wire module. The details of APA assembly are shown in Figure 3-11.

One major drawback of this alternative design is a relatively large dead space between any APA joints ($\sim 2 \times 35 \text{ mm}$) due to the FR-4 board size. The electric field in the drift region directly above the FR4 boards will have some irregularities, causing minor distortion to the track trajectories. To eliminate these problems, additional special wire modules could be placed over the dead region at the APA edges. For example, a module with two layers of 18 wires ($\sim 90 \text{ mm}$ wide) would completely cover the FR-4 boards between two APA frames. The upper layer wires maintain the uniform drift field, collect the electrons, and provide the X position information, just like the regular X readout wires. The obtained X coordinates and charge information partially restore the detector sensitivity at the APA joints. The scheme with additional wire modules is shown in Figure 3-12.

Using the wire geometry shown in Figure 3-12 the bias voltages for wire planes have been

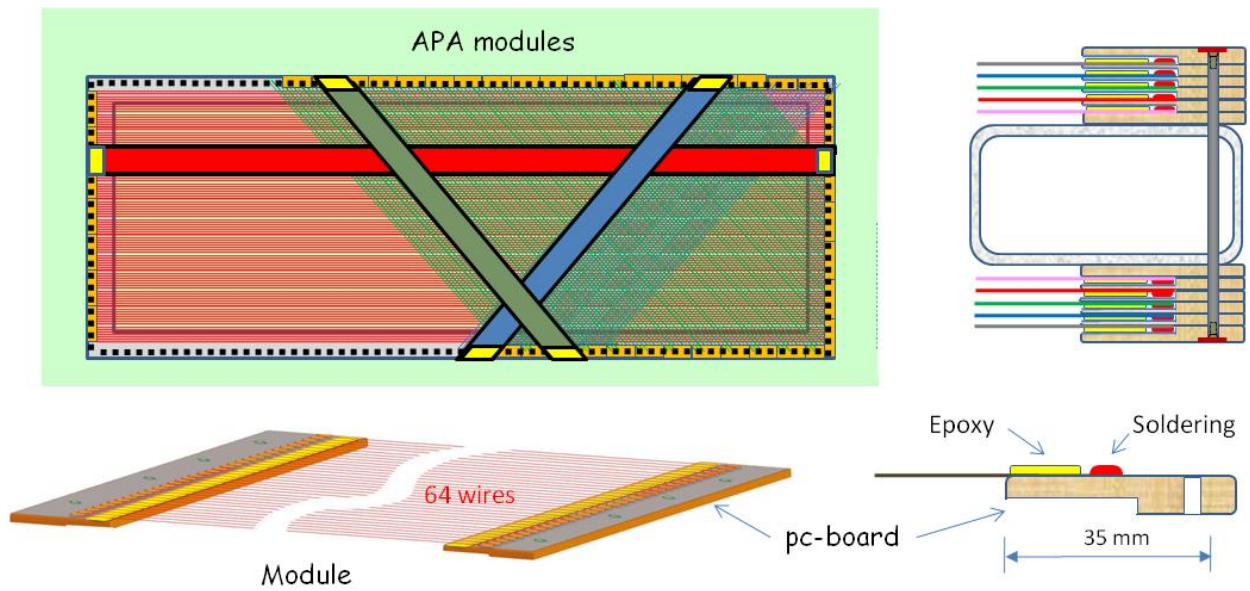


Figure 3-11: Concept of the alternative APA wiring scheme using pre-assembled wire modules

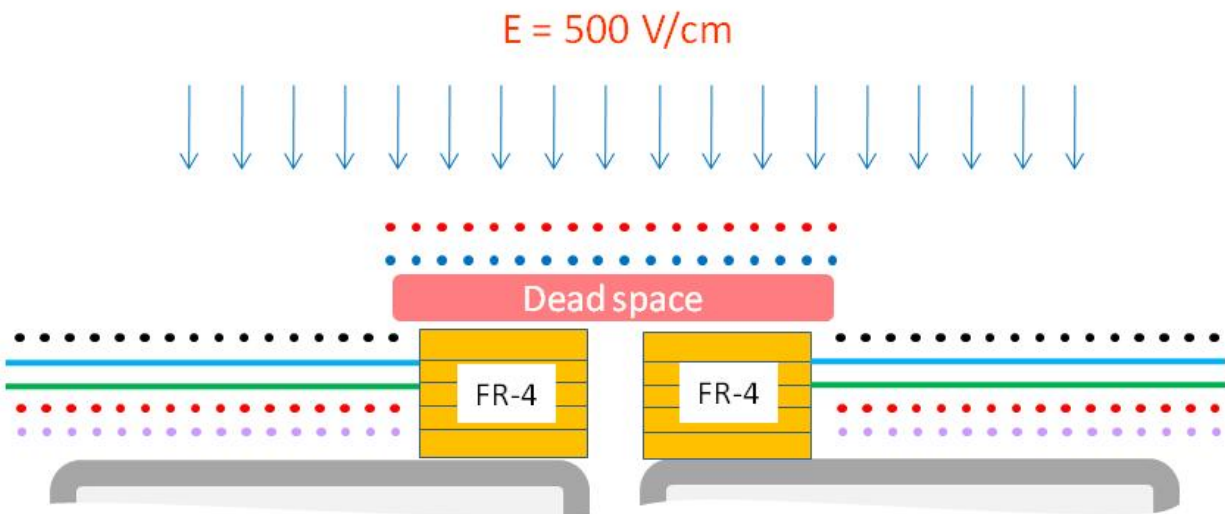


Figure 3-12: Illustration of dead space above FR-4 boards and possible solution with installation of two additional wire planes above.

defined and electric field strength calculated in 2D. The calculation confirmed that the drift field above the FR-4 boards is uniform, and full collection of electrons can be achieved on the additional wire modules. From past experience, the electric field is not expected to change very much in a 3D simulation. Nevertheless, it is necessary to develop a full 3D model for more accurate field calculations and simulation of electron drift in real APA geometry.

One of the advantages of this alternative design is the modularity of wire planes that will simplify the wire placement and APA assembly. The winding procedure becomes simple and reliable and a simple winding machine with replaceable wheels is required. The module production could be done at several sites to accelerate the APA assembly schedule. The completed modules will be cleaned in an ultrasonic bath, dried in a vacuum oven and stored in dry air. A set of tests will be performed to ensure that modules meet the technical specifications. The winding scheme for module fabrication is presented in Figure 3-13.

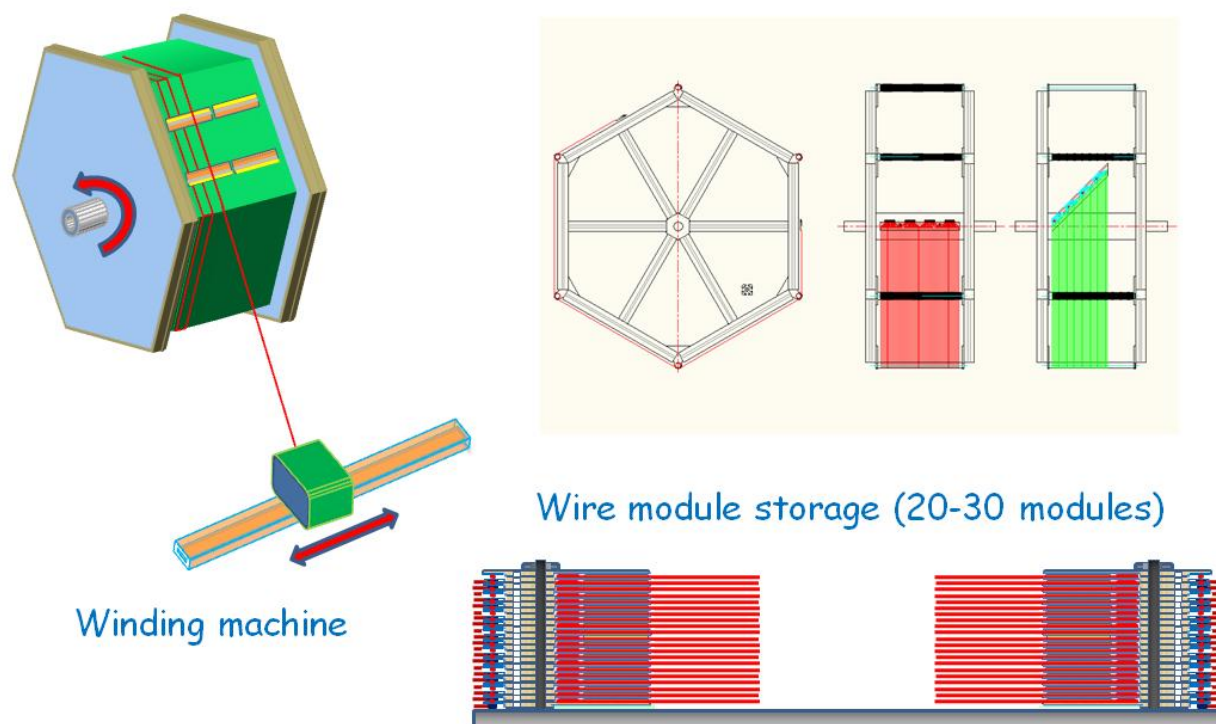


Figure 3-13: Conceptual designs of the wire winding machine and wire module storage fixture. Upper left: the fabrication of the wire modules are accomplished by attaching the wire module boards on to a large drum with specific circumference, and laying the wires over the boards at a constant pitch. Upper right: placement of the wire module boards for the X and grid wires (red), and U, V wires (green). Bottom: completed wire modules are stacked on a storage fixture.

For the assembly of the wire modules onto APA frames, a special assembly table will be used. The table will allow frame rotation in order to allow access to both sides of an APA frame. A special movable, stretching mechanism will be mounted on the table edge for installation of wire modules. After assembly of each plane the final wire tension and wire spacing will

be measured. If some wires' tension or spacing are out of specifications, their corresponding wire modules will be replaced. A schematic design of a wire-stretching table is shown in Figure 3-14.

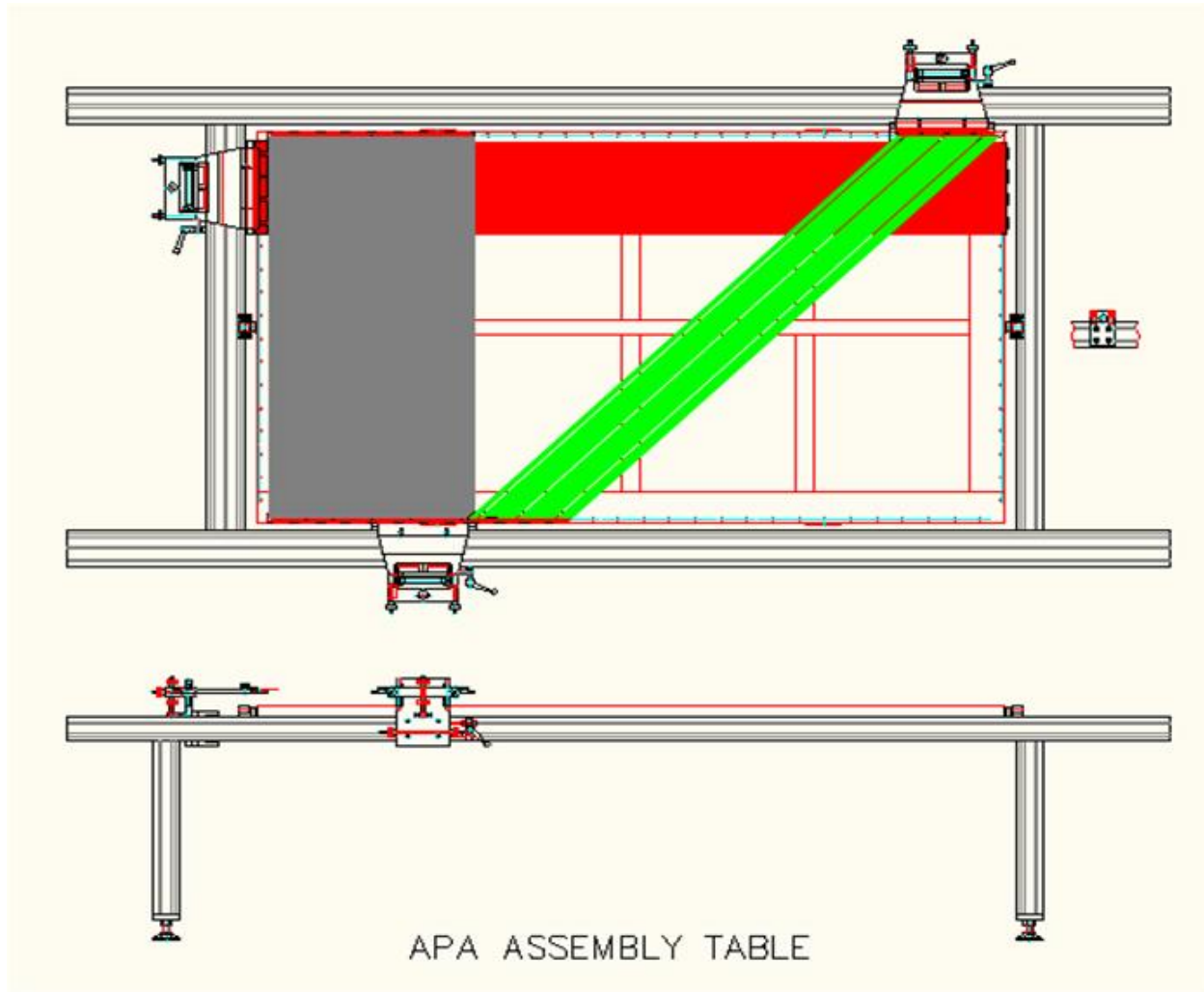


Figure 3-14: Assembly table with equipment for stretching wires.

Demonstration modules of both APA designs are being constructed. They will be evaluated to determine which design will be used in the final APA construction.

3.4 Cathode Plane Assemblies (WBS 130.05.04.04)

The cathode plane assemblies (CPAs) have similar dimensions to the APAs, 2.5 m wide and 7 m high. Each CPA is made of a stainless-steel framework, with a layer of stainless-steel wire mesh stretched over one side of the frame. To reduce drift-field distortion, all surfaces that

rise significantly above the mesh, including the stainless-steel frame structure on the other side of the mesh, are covered with field-shaping electrodes biased at appropriate voltages. Figure 3–15 illustrates the concept of the CPA.

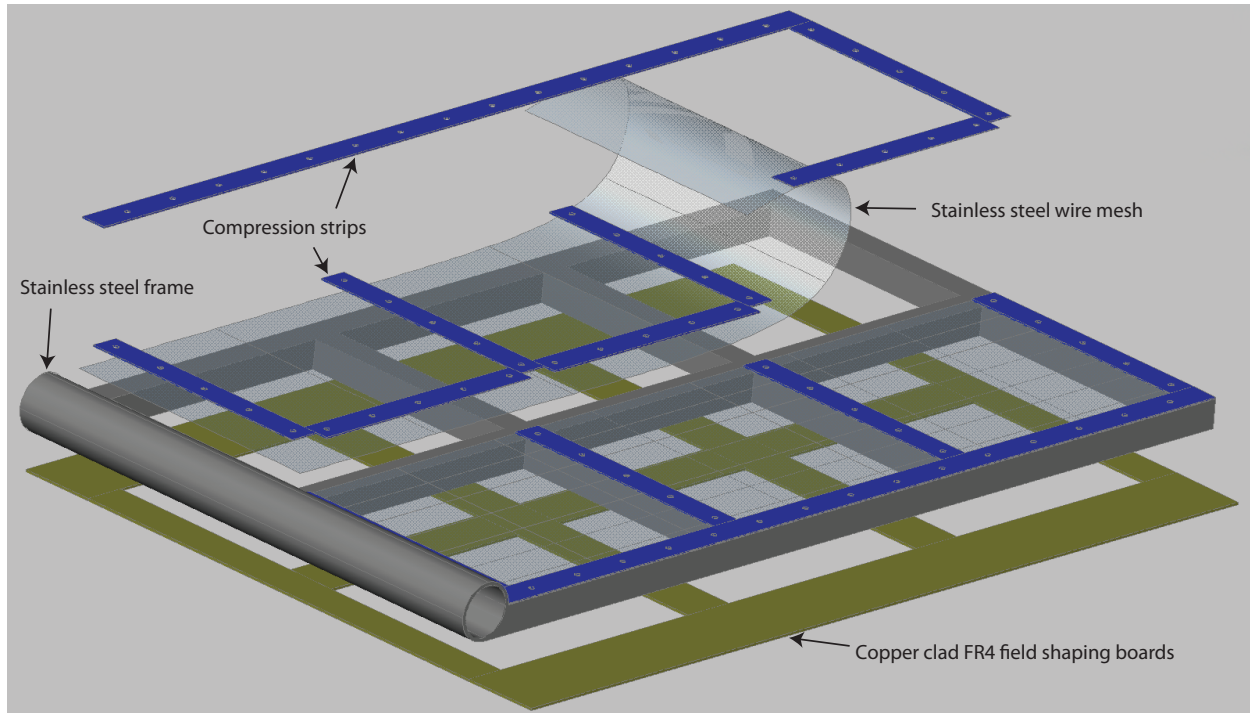


Figure 3–15: Conceptual design of a cathode plane assembly (not to scale). The assembly is 2.5 m wide and 7 m tall, similar to the APAs.

To achieve a 500 V/cm drift field over a 3.7 m distance, the bias voltage on the cathode plane must reach -185 kV. The minimal high-voltage bias system requires four high-voltage (HV) power supplies, one for each CPA row in the cryostat. Further partitioning of the CPA row and the field cages is being considered to reduce the loss of active volume in the event of HV failure inside the cryostat. For example, an eight-zone configuration will require that a gap of several centimeters be maintained in the middle of each CPA row and field cage. Failure to maintain HV in one zone will result in the loss of no more than 20% of the active volume.

3.5 Field Cage (WBS 130.05.04.05)

Each pair of facing cathode and anode rows forms an electron-drift region. A field cage completely surrounds the four open sides of this region to provide the necessary boundary conditions to ensure a uniform electric field within, unaffected by the presence of the cryostat walls.

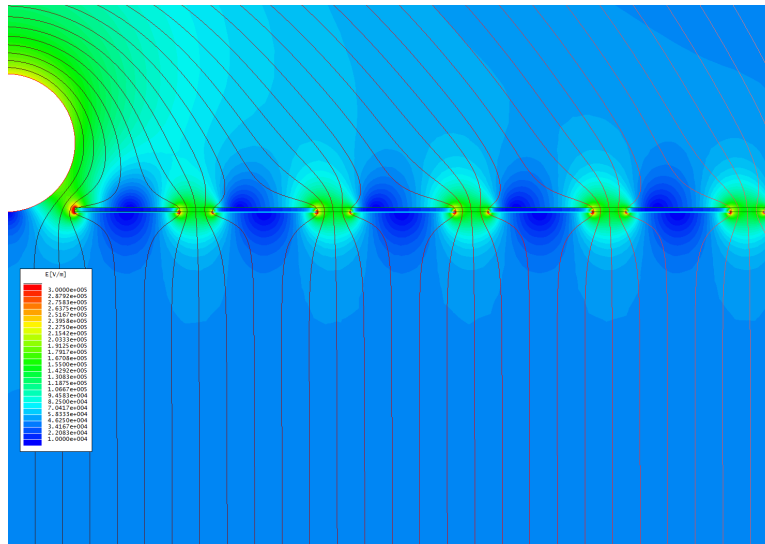


Figure 3–16: Electrostatic simulation of the electric field near a section of the field cage. The filled color contours represent the electric field strength. The line contours represent the electric potential at 500 V intervals. The pitch of the electrodes is 5 cm in this model.

The entire TPC requires $\sim 2700\text{m}^2$ of field cage material. The field cages are constructed using copper-clad FR4 sheets reinforced with fiber glass I-beams to form panels of $2.5\text{ m} \times 3.7\text{ m}$ in size. Parallel copper strips are etched/machined on the FR4 sheets. Strips are biased at appropriate voltages provided by a resistive-divider network. These strips will create a linear electric-potential gradient in the LAr, ensuring a uniform drift field in the TPC’s active volume. Figure 3–16 shows the results from an electrostatic simulation of a particular strip pattern. The drift-field non-uniformity quickly drops below 1%, roughly a strip pitch away from the field-cage surface. Since the field cage completely encloses the TPC drift region on four (of six) sides, the FR4 sheets must be frequently perforated to allow natural convection of the liquid argon. The “transparency” of the perforation will be determined by a detailed LAr computerized fluid dynamic (CFD) study.

The resistor-divider network will be soldered directly onto the field-cage panels. Multiple resistors will be connected in parallel between any two taps of the divider, in order to provide fault tolerance. One end of the divider chain is connected directly to the cathode, while the other end is connected to ground at the APA through resistors of the appropriate value.

3.6 TPC Assembly in the Cryostat

Figure 3–17 shows a partial assembly of a section of the TPC. The finished cryostat has seven rows of anchor points distributed along the ceiling (not shown in the figure). A mounting rail is suspended through stainless-steel rods to each row of the anchor points. Under these

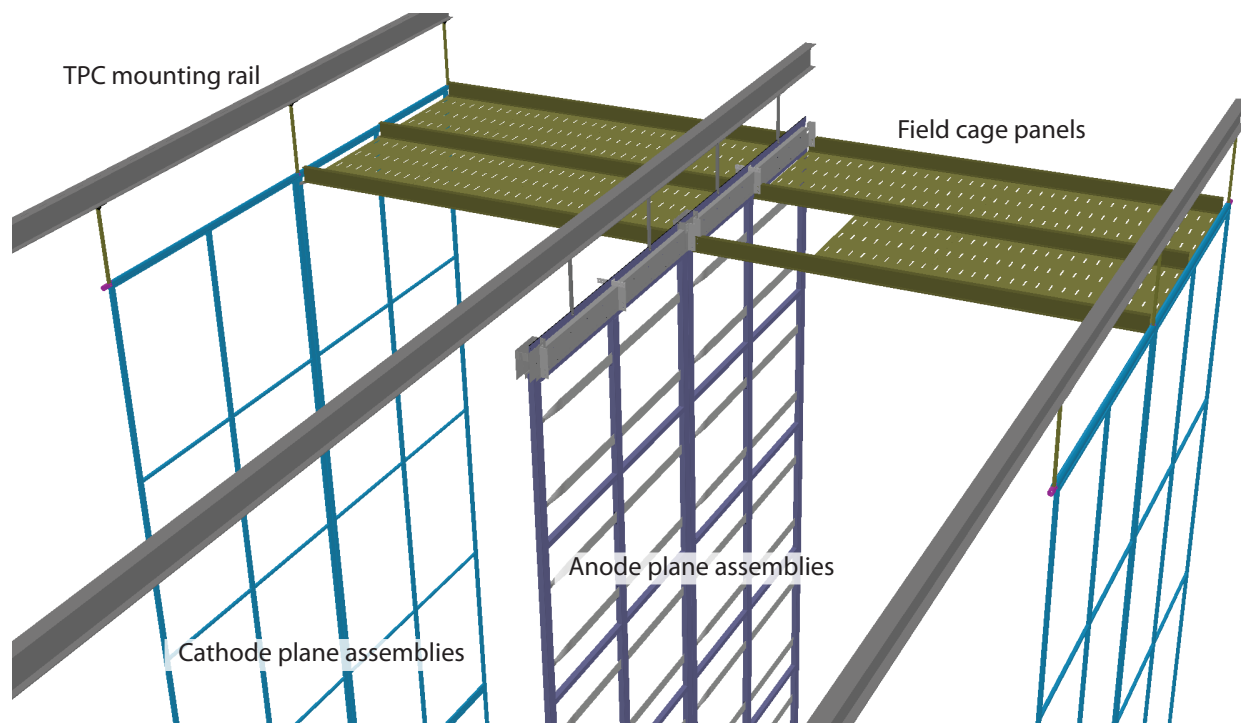


Figure 3-17: A partial assembly of the TPC showing all major components

seven mounting rails, rows of CPAs and APAs are suspended in an interleaved fashion. Because the cathodes are at a high voltage, the CPAs are attached to their mounting rails through G10 rods. The distance between the facing anode and cathode is maintained by the pultruded fiberglass I-beams holding the FR4 sheets forming the field cage. The TPC installation procedure is discussed in Chapter 6.

3.7 In-Vessel Front-End Electronics (WBS 130.05.04.03)

The front-end electronics will operate at cryogenic temperatures. The system must provide amplification, shaping, zero suppression, digitization, buffering and multiplexing of the signals. All functions of the readout chain will be programmable via a register on the ASICs. The APAs will need to be self triggering since, unlike neutrino-oscillation measurements for which the pulsed beam can provide a trigger for readout, most other measurements, such as proton decay and supernova bursts, will have no trigger.

3.7.1 Architecture

The large number of readout channels (276,480) required to instrument each cryostat of the LAr-FD TPC dictates the use of CMOS ASICs for the electronics. Requirements of low noise

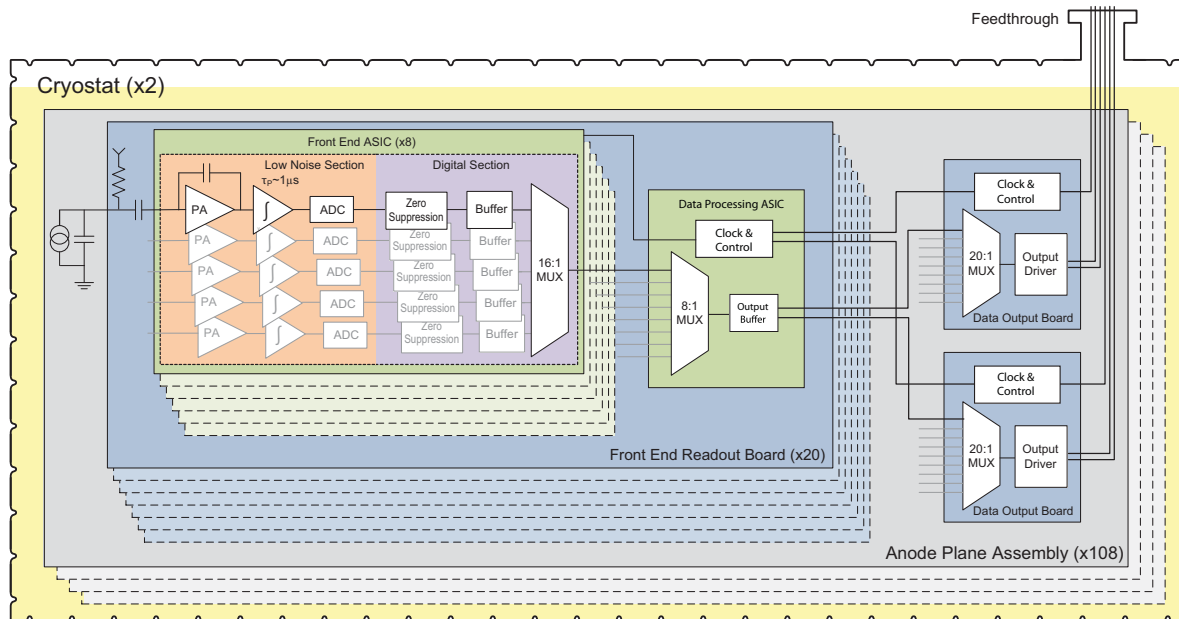


Figure 3-18: Conceptual architecture of the front-end electronics operating in LAr

(less than 1000 rms electrons for a wire capacitance of 220 pF) and for achieving extreme purity of the LAr dictate that the front-end electronics be located at the signal wires in the LAr. This reduces the signal capacitance and hence the noise. Significant levels of signal multiplexing can then also be used, minimizing the number of cables in the ullage gas and the attendant outgassing of electronegative impurities. This electronics architecture, combined with the modular TPC elements, also leads to a TPC implementation that can be readily scaled to any detector size or geometry.

Figure 3-18 shows the architecture of a front-end electronics design that meets the requirements for the LAr-FD TPC. The entire electronics chain is immersed in the LAr and operates at 89 K. It is composed of a 16-channel front end implemented as a mixed-signal ASIC providing amplification, shaping, digitization, buffering, a 16:1 multiplexing stage, a driver and voltage regulators. Eight front-end ASICs plus a single digital ASIC implementing an 8:1 multiplexer, clock and control circuitry comprise a single 128-channel front-end readout board. A third digital ASIC containing a multiplexing stage (20:1) and driver is used for each APA module to serialize all of the data from the 20 readout boards on a single APA (for a total of 2,560 channels per APA) and transmit it out of the cryostat on either a twisted copper pair or an optical fiber. The choice between twisted-pair copper and optical fiber will be postponed until further work is completed to evaluate the long-term reliability of optical drivers at LAr temperature. The data rates per APA are not expected to be high enough to require the use of optical fibers. For either choice, a two-fold redundancy per APA will be implemented to minimize the probability of the loss of an entire APA.

Figure 3-19 shows a block diagram of the proposed 16-channel front-end ASIC. Each channel includes a charge amplifier with a selectable of gain 4.7, 7.8, 14 and 25 mV/fC (full scale charge of 55, 100, 180 and 300 fC), a high-order anti-aliasing filter with adjustable time constant (peaking time 0.5, 1, 2, and 3 μ s), an option to enable an AC coupling, baseline adjustment for operation with either the collecting or the non-collecting wires, a 12-bit 2 MS/s ADC and a zero-suppression/data compression stage. Shared among the 16 channels are the bias circuits, programming registers, a temperature monitor, the digital multiplexer (16:1), an analog buffer for signal monitoring, and the digital interface. A 600-kb buffer will be included in this design, capable of storing a 1.5-ms event sampled at 2 MS/s in each channel assuming no compression. Two or more events can be stored with compression. The layout size is on the order of 10×5 mm², which would provide a yield in excess of 90%. The estimated power dissipation is below 15 mW per channel at 1.8 V supply.

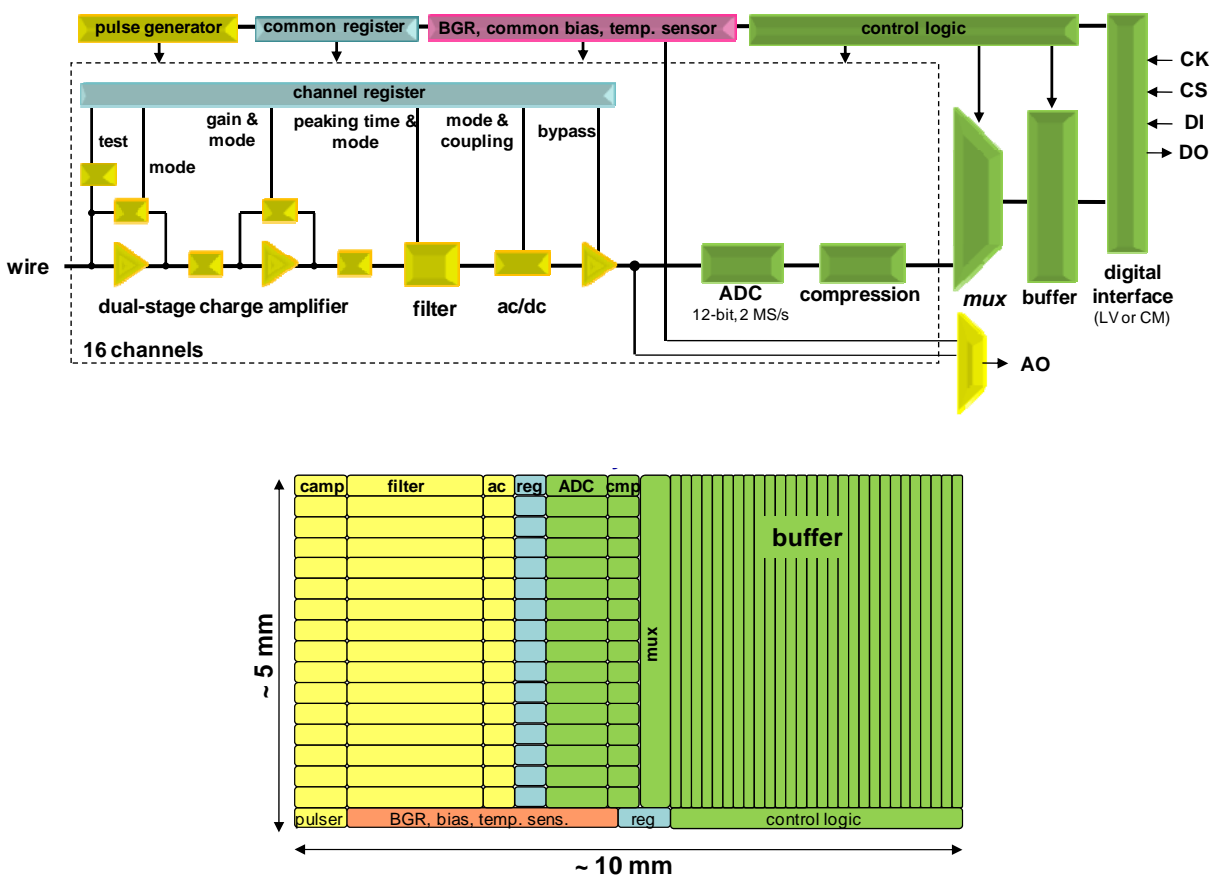


Figure 3-19: Architecture and layout of the 16-channel front-end mixed-signal ASIC

3.7.2 Data Rates

For neutrino-oscillation measurements the pulsed beam can provide a trigger for readout, but for most other measurements, such as proton decay and supernova bursts, there will be

no trigger. Therefore the APAs will be self triggering.

Data rates will then be dependent on the event rates for background processes. Dominant backgrounds are decays of radionuclides in the LAr, predominantly the naturally occurring Ar^{39} . The cosmic-muon rate at the 4850 Level is approximately 2.3 mHz per APA. Reading an APA (2,560 wires) for one drift time (4,625 samples) gives 142 Mb of data. At the cosmic-event rate, the net data rate is 0.3 Mb/s per APA. For radioactive decay of Ar^{39} the rates are much higher: the specific activity is 1 Bq/kg which results in 183 kHz/APA. At this rate, the APA will be continuously read out, at a prohibitive data rate of 61 Gb/s; of course, since the ionization produced by these events is highly localized, most of the drift “frame” is empty.

The mean range of the beta from Ar^{39} decay is only 0.3 mm, so all of the signal is contained in a single over threshold sample of a single wire, for a true “information” rate of 0.6 Mb/s. In order to reduce the rate (and volume) of recorded data to tolerable values, zero suppression and data buffering must be provided. The simplest scheme is to do this at the APA level, deriving a write-enable from the logic OR of discriminators on all the charge collecting (Y) wires in an APA, and then writing all data to a buffer while the write-enable is true. Baseline samples could be recorded with reduced range (4 bits) and compacted into full words. This mode would still record large volumes of data without useful information, particularly for simple event topologies and from low-energy events (radioactive decays) and noise.

A more effective zero-suppression can be implemented at the chip level with a dedicated buffer for each channel having a write-enable consisting of the OR of that channel and its two nearest neighbors. A few samples before and after the write-enable will also be recorded to capture the below-threshold leading and trailing edges of the signal waveform. This mode provides maximal zero-suppression that is insensitive to localized low-energy events and noise. This scheme would reduce the data rate per APA for Ar^{39} decays to 18 Mb/s. The rate for Kr^{85} decays would be about four times lower.

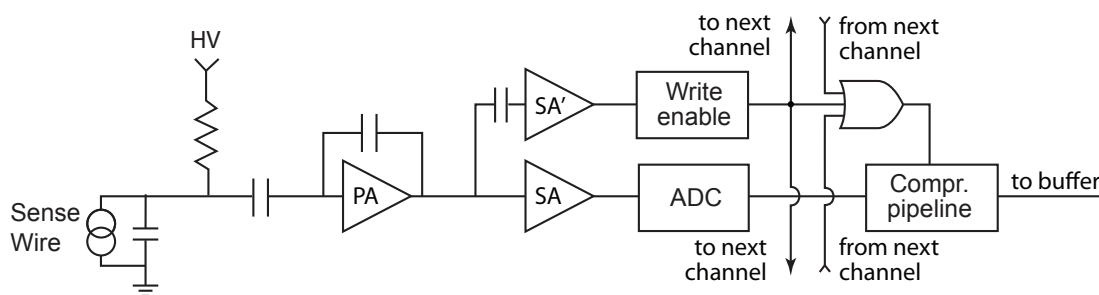


Figure 3–20: Conceptual architecture of zero-suppression using a write enable using separate analog shaping

A conceptual diagram of this zero-suppression implementation is shown in Figure 3–20. The write-enable for each channel is derived from a separate shaping amplifier with AC coupling

to remove low-frequency, baseline fluctuations. Multiple programmable discriminators in the “write-enable” logic are needed to sense both unipolar (collection wire) and bipolar (induction wire) waveforms. The write-enable gate passes to, and is OR-ed with, channels of adjacent wires to record the below-threshold portion of charge shared with these wires. As individual channel buffers fill, they are transferred to a large output buffer shared with all 16 channels for transmission off the chip.

3.7.3 CMOS Circuit Design

To successfully design CMOS circuits that will operate at cryogenic temperatures, two critical issues must be addressed and resolved. The first is the requirement for realistic models at the operating temperature of active devices, interconnects and passive components (resistors and capacitors) in order to reliably predict operating points, signal response and noise during the design process. The second critical requirement is that the design must ensure a long operational lifetime, since once the TPC is filled with LAr the detector must operate for about 15 years without any access to the electronics for repair or replacement. Concerning the availability of realistic models, our preliminary results from the cryogenic characterization (down to 40 K) of a complete mixed-signal ASIC [20] in a commercial CMOS 0.25 μm technology, originally developed for room-temperature applications, indicates that the models are useful to first order. To refine these models, several single-transistor test structures were fabricated on the first prototype of the 0.18 μm device. Measurements of the properties of these structures at cryogenic temperatures have been used to refine the device models at 89 K.

The lifetime of CMOS circuits is limited by several mechanisms which degrade the performance over time, eventually causing the circuit to fail to perform as specified. The rates of most degradation mechanisms in CMOS, such as electro-migration (EM), stress migration (SM), time-dependent dielectric breakdown (TDDB), thermal cycling (TC), and negative bias-temperature instability (NBTI), all scale with temperature such that cryogenic operation is favored [21][22]. The only mechanism that could affect the lifetime at cryogenic temperature is the degradation due to impact ionization, which causes charge trapping in the MOSFET gate oxide at large drain-current densities (the “Hot Carrier” effect). Results from a CMOS reliability study [23] provide general design guidelines (for device geometry, bias and current density) that should guarantee a lifetime well in excess of 15 years for continuous cryogenic operation. These design guidelines also provide information for designing test conditions to observe the deterioration mechanism and to extrapolate from accelerated deterioration rates, measured under stressed conditions within practical times, to the ultimate lifetime under normal operation.

A monitor of the impact ionization is the bulk current, which reaches a maximum at $V_D S = V_D D$ and at $V_G S = 0.5 V_D D$. When operating constantly in this condition at room temperature, a properly designed device will typically have a lifetime (defined as a 10%

degradation in g_m) of about 10 years. The bulk current (i.e., the impact ionization) increases by roughly a factor of four from 300 K to 77 K [23] and a circuit designed for operation at room temperature would have a proportionately shorter useful life at cryogenic temperature. As stated above, in order to guarantee the required lifetime at cryogenic temperatures, design guidelines must be modified for both analog and digital circuits. For analog circuits, this is done by operating the devices at moderate-to-low drain current densities, where impact ionization becomes negligible. For digital circuits, operating the devices with reduced $V_D D$ (about 20%) and using non-minimum channel length L , which is easily accommodated since at cryogenic temperature the speed of the digital circuit increases, compensating for the increased L . These guidelines will be verified with accelerated aging tests, at increasing values of $V_D D$, on dedicated structures. Such tests also will be conducted on prototype samples throughout the development process to verify the long-term reliability of the final ASICs.

The development of the readout ASIC has begun by designing and fabricating in a commercial CMOS process (0.18 μm and 1.8V) a 16-channel ASIC implementing the complete analog front-end section of the scheme shown in Figure 3-19. This process is expected to be available for at least another 10 years. The charge amplifier input MOSFET is a p-channel biased at 2 mA with a L/W (channel length/width) ratio of 0.27 $\mu\text{m} / 10 \text{ mm}$, followed by dual cascade stages. The charge amplification and shaping filter have digitally programmable gain and peaking time as described above. Each channel also implements a high-performance output driver which, in the final version, will be replaced with a sample-and-hold stage preceding the ADC. The ASIC integrates a band-gap reference (BGR) to generate all the internal bias voltages and currents. This guarantees a high stability of the operating point over a wide range of temperatures, including cryogenic. The ASIC is packaged in a commercial, fully encapsulated plastic QFP 80 package.

This ASIC has now been through three design/fabrication/testing revision cycles. Prototypes from each cycle have been evaluated and characterized at room (300 K) and liquid nitrogen (77 K) temperatures. During these tests the circuits have been cycled multiple times between the two temperatures and operated without any change in performance. Figure 3-21 shows the measured pulse response, along with details on the adjustability of the gain, peaking time and baseline. These results are in close agreement with the simulations and indicate that both the analog and the digital circuits and interface operate as expected in a cryogenic environment. Simulations and experimental results show that the pole-zero cancellation needs to be optimized, which will be done in the next revision of the design. Also reported in Figure 3-21 are the outputs of the BGR and temperature sensor, which are in close agreement with the simulations as well.

Figure 3-22 shows the measured ENC versus filter-time constant (peaking time). At 1 μs about 650 e^- was measured, to be compared to the simulated value of 500 e^- . The difference is mainly due to the thermal noise from a $\sim 11\text{-ohm}$ parasitic resistance of the input line (shown in the detail of Figure 3-22), which contributes about 350 electrons at 77 K. The width of the line will be increased in a revision in order to make this contribution negligible. A second contribution, on the order of 100 e^- , was due to the dielectric loss from the capacitor

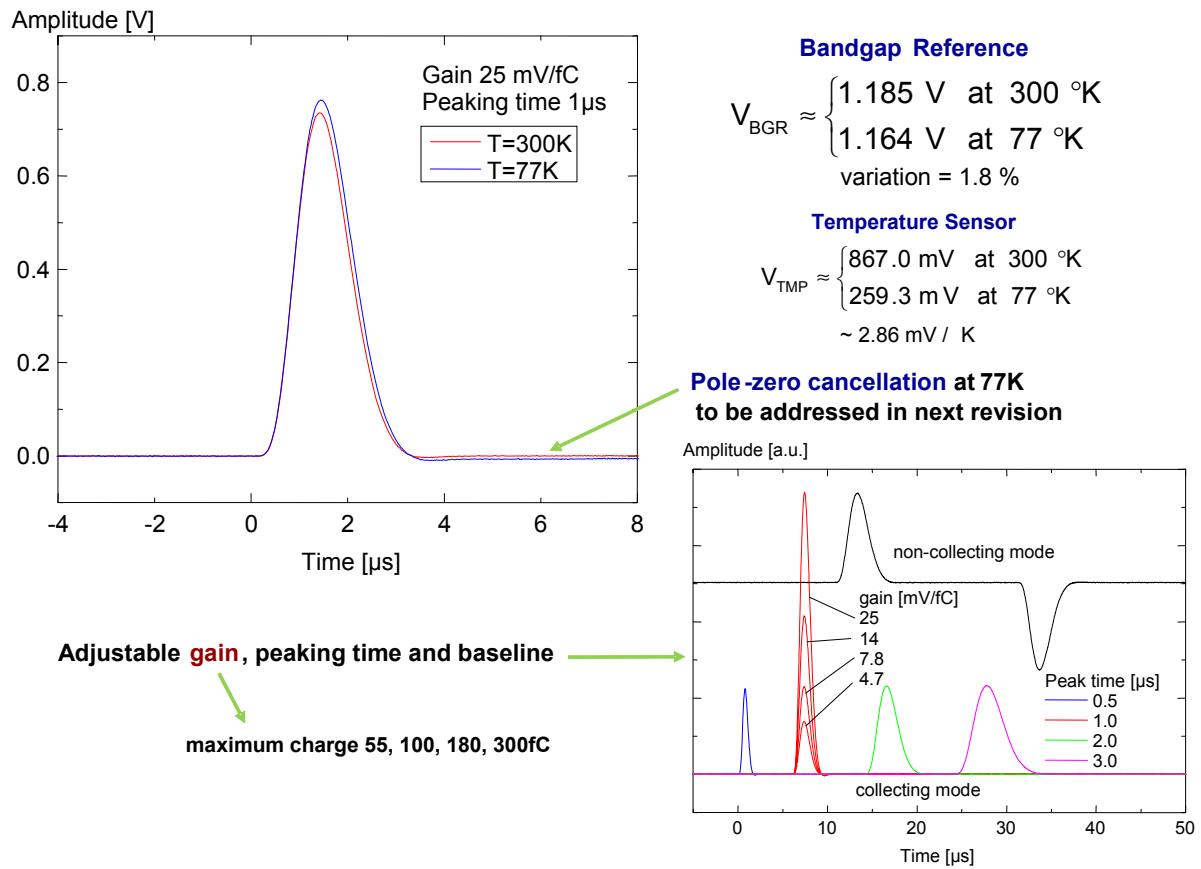


Figure 3-21: Measured pulse response with details on gain, peaking time and baseline adjustments

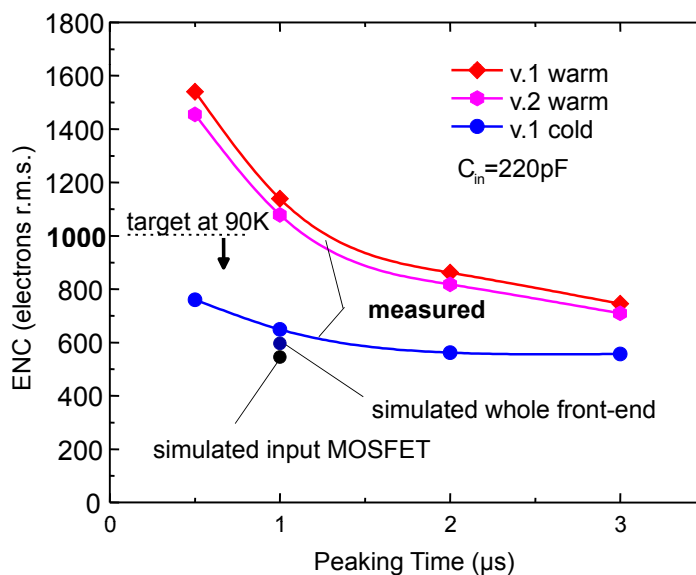


Figure 3-22: Measured ENC vs filter time constant from the first two versions of the analog front end ASICs

(220 pF) used to simulate the wire (the cases of MICA and NPO ceramic were compared). This contribution would not be present with the input connected to a sense wire in the TPC.

Each channel is equipped with an injection capacitor which can be used for test and calibration and can be enabled or disabled through a dedicated register. The injection capacitance has been measured using a calibrated external capacitor. The measurements show that the calibration capacitance is extremely stable, changing from 184 fF at room temperature to 183 fF at 77 K. This result and the measured stability of the peaking time demonstrate the high stability of the passive components with the temperature. Channel-to-channel and chip-to-chip variation in the calibration capacitor are typically less than 1%. Measurements are being carried out on the individual test structures fabricated on this ASIC to confirm device models and design guidelines.

All data, control, bias and power supply lines will be duplicated to provide redundancy to avoid the loss on an entire APA. Four APAs will be cabled to warm feedthroughs in “chimneys” in the roof of the cryostat that contain the support rods for the TPC planes.

3.8 TPC Infrastructure (WBS 130.05.04.06)

The TPC infrastructure includes low-voltage and high-voltage supplies, all power cables and cable routing to the supplies, signal cables and cable routing to the DAQ, and all cryogenic feedthroughs.

3.8.1 Design Considerations

- All power supplies must be able to be monitored and controlled both locally and remotely through the DAQ system. They must have over-current and over-voltage protection circuits.
- The power supplies for the TPC cathode planes must be able to provide -200 kV at 1 mA current. The output voltage ripple must not introduce more than 10% of the equivalent thermal noise from the front-end electronics. The power supplies must be programmable to trip (shutdown) their output at a certain current limit. During power on and off, including output loss (for any reason), the voltage ramp rate at the feedthrough must be controllable to prevent damage to the in-vessel electronics through excess charge injection.
- The power supplies for the wire-plane bias voltages must provide sufficient current. The output-voltage ripple must not introduce more than 10% of the equivalent thermal noise from the front-end electronics.
- High-voltage feedthroughs must be able to withstand -250 kV at their center conductors in 1 atm air or argon gas environment when terminated in liquid argon.
- Medium-voltage feedthroughs must be able to withstand twice their nominal operating voltages with a maximum specified leakage current in 1-atm argon gas.
- Low-voltage power feedthroughs must be able to deliver sufficient DC current.

3.8.2 Reference Design

There are three types of power supplies and matching feedthroughs in the cryostat: TPC high voltage, wire-bias voltages and low-voltage DC power to the readout electronics. There are two additional types of feedthroughs carrying digital signal: LVDS and optical fiber.

With the exception of the TPC high-voltage connections, all other cables inside the cryostat will be attached to their corresponding feedthroughs distributed throughout the cryostat roof. The other ends of the cables will be connected to the matching connectors on the APAs in the cryostat. The cables for the lower APAs must be carefully threaded through the hollow frames of the APA stacks. The cables will be strain-relieved on the mounting rails above the APAs.

Measurements in the Materials Test Stand at Fermilab (described in Section 8.3.1) have shown that impurities (principally O_2 and H_2O) embedded in objects submerged in the liquid argon do not result in a decrease in electron-drift lifetime, whereas impurities in objects located in the warmer gas phase do. This indicates the importance of minimizing the amount of material in the gas ullage at the top of the cryostat. Therefore it may be desirable

to connect all cables to feedthroughs below the liquid surface, and then pass the cables out of the cryostat, through an evacuated volume that traverses the gas and cryostat insulation, to a matching set of feedthroughs to the outside.

3.8.3 TPC High Voltage

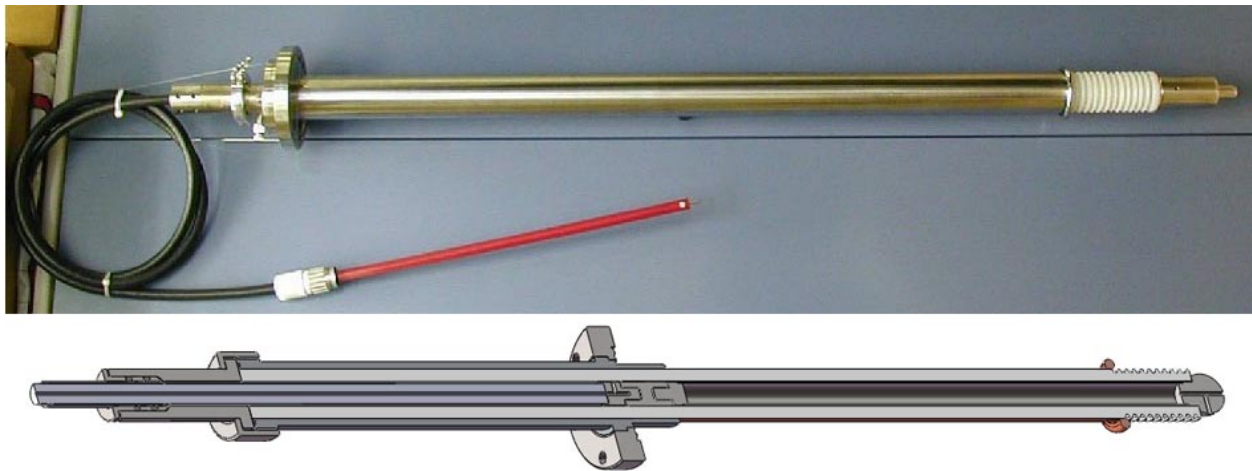


Figure 3-23: Top: A high voltage feedthrough developed by the UCLA group for the Icarus experiment. It was tested up to 150 kV. Bottom: a conceptual design of a new feedthrough for the LAr-FD.

The cathode planes are biased at -185 kV to provide the required 500 V/cm drift field. At a minimum, four high-voltage power supplies, each connecting through their own feedthroughs, will be used. Each supply will provide high voltage to one of the four rows of the cathode plane assemblies.

The current candidate for the high-voltage power supplies is the Heinzinger PNC hp series, which is used by the ICARUS experiment. Additional filtering of the voltage ripples is done through the intrinsic HV cable capacitance and series resistors. Established techniques and practices will be implemented to eliminate micro-discharges and minimize unwanted energy transfer in case of an HV breakdown.

To ensure safe and reliable operation, the feedthroughs will be tested at a much higher voltage than expected in routine operation (~ 250 kV) in liquid argon. The feedthroughs will be mounted on the ceiling of the cryostat, their cold ends reaching through the gas ullage space and submerging into the liquid argon. The center conductor on the cold side of a feedthrough will be insulated and shielded by a grounded shroud at least 50 cm below the surface of the liquid. Connections between the feedthroughs and the CPA rows are made through stainless-steel pipes in the liquid argon. Figure 3-23 shows an example of a feedthrough made by the UCLA group for the ICARUS experiment, as well as the conceptual design of a feedthrough suitable for the LAr-FD TPCs.

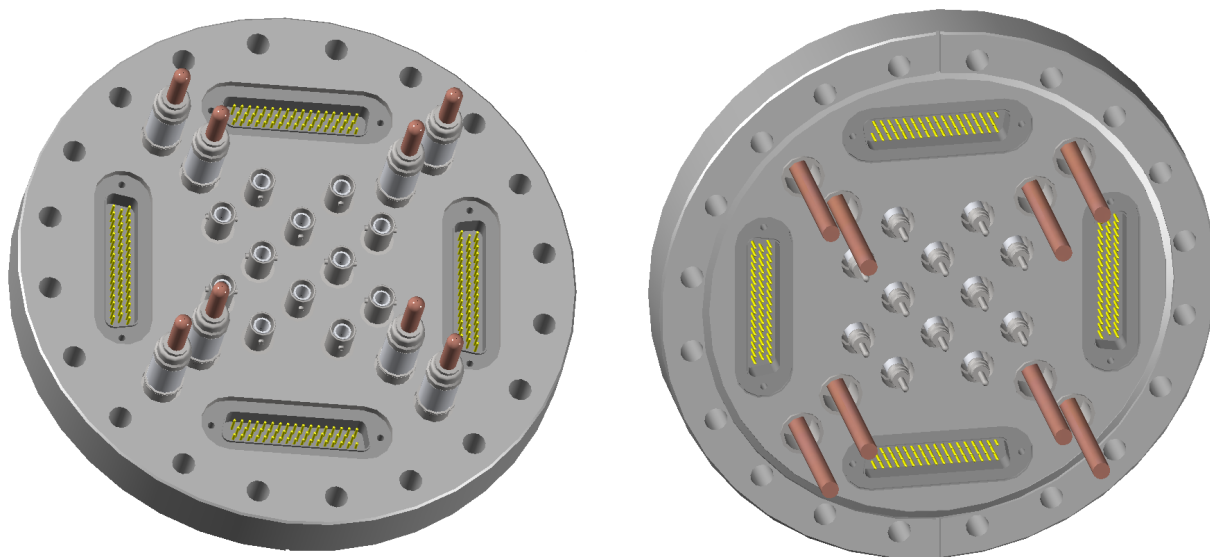


Figure 3-24: A conceptual design of a signal/power feedthrough using all off-the-shelf commercial components

3.8.4 Wire-Bias Voltages

Each anode plane assembly requires three bias voltage connections at $+820\text{V}$, -370V , and -665V . The current on each of these supplies is expected to be zero at normal operation. However the ripple voltage on the supply must be carefully controlled to avoid noise injection into the front-end electronics.

The power supplies for the wire bias will be similar to those used for conventional multi-wire proportional chambers. Additional filtering networks will be needed to further reduce voltage ripples. The default feedthroughs are the commercial SHV type. However, other, higher-density multi-channel feedthroughs capable of withstanding the maximum voltage are under investigation.

3.8.5 Power for the Cold Electronics

The power-per-channel for the front-end ASIC is designed to be about 15 mW and the total power requirement for each APA is expected to be about 65 W . Power will be supplied to the electronics on each APA separately by low-noise power supplies outside the cryostat, either directly by low-voltage (1.8 V), high-current (36 A) conductors or by high-voltage (48 V) low-current (2 A) conductors to DC-DC converters placed locally in the LAr. The use of DC-DC converters requires conductors with smaller cross section, minimizing heat input to the cryostat (and ice formation of the feedthroughs). However, the power dissipated by the (somewhat inefficient) converters in the LAr will create boiling which may introduce

contamination directly into the high-purity LAr, and if enough LAr is vaporized, may also produce strong mixing of the ullage gas, driving more impurities into the liquid. These effects of boiling LAr, unless they can be demonstrated to be harmless, will drive a preference for eliminating DC-DC converters, and directly powering the front-end readout boards.

Heat conduction through the high-current feedthroughs and the self-heating ($I \cdot R$) of the wires are the factors contributing to additional heat load on the cryogenic system. The sum of these two factors as a function of the wire gauge, however, has a minimum due to the two opposing dependencies on the copper-wire cross section. An optimum wire gauge can be chosen to minimize heat input to the cryostat.

3.8.6 Digital Data IO Feedthroughs

The TPC data rate per APA, using the zero-suppression and full event-buffer scheme described earlier, appears sufficiently low that it is within the capability of a single LVDS channel on copper. Optical fiber will be used if data must be transmitted at a much higher rate. In this case, the number of optical fibers will be two per APA for redundancy, or 336 in the entire module. Commercial optical-fiber feedthroughs are available to meet this demand.

In addition to the high-speed data-output channels, LVDS connections will be made to each APA to distribute a clock signal and control information. These data can be transmitted at a lower bit rate, clearly within the capabilities of LVDS. The number of channels for these signals are on the order of thousands in the entire detector, easily covered by commercial multichannel feedthroughs.

A conceptual design of an APA signal/power feedthrough flange is shown in Figure 3-24. Based on a standard 8-in conflat flange with all commercial off-the-shelf components, each of these feedthroughs will serve the bias/power/digital IO needs of four APAs.

3.9 TPC Prototyping, Test and Checkout

3.9.1 TPC Prototyping

Several prototype TPC modules will be constructed during the design phase. The initial prototypes will be fraction scale or partial models of the APA and CPA. The CPA prototype will be used to evaluate the wire-mesh tension and field-shaping electrode attachment techniques. The APA prototype will be used to study the placement of the wire-wrapping boards and wire-support structures. It will also be used to develop the prototype winding machines. The prototypes will undergo numerous thermal cycles down to liquid-nitrogen temperature to test the integrity of the wire-to-board and board-to-frame bonds.

The second set of prototypes will be full/near-full scale models of the APA and CPA. They will be used to validate the designs and to evaluate production procedures. Prototype front-end electronics boards are expected to be available at this stage. A partially or completely instrumented APA can be assembled and tested in a prototype cryostat.

The 1-kton engineering prototype (LAr1), described in Chapter 7, requires three full-size APAs with fully instrumented readout electronics, six full-size CPAs, and complete field-cage coverage. The TPC will be constructed using identical APAs, CPAs and field-cage panels as designed for the LAr-FD. Additional features will be installed to ensure proper TPC operation given the half-height cryostat configuration. The construction and assembly of all TPC mechanical components will use the same materials and techniques as designed for LAr-FD, with the exception of a reduced degree of automation that will be used to wire APAs for the LAr-FD.

A complete set of cold electronics will be installed on the APAs. The electronics components will closely resemble those designed for the LAr-FD. All key features of the LAr-FD electronics chain, including preamp, shaper, ADC, digital buffer, zero suppression and multiplexing will be implemented. Some LAr1 electronics may be in prototype or functional-equivalent form.

3.9.2 Assembly Testing

The front-end readout boards will be thoroughly tested.

- A small number of the ASICs will undergo a complete suite of tests, including thermal cycling to determine the batch yield.
- If the yield is high ($> 95\%$), all ASICs will be mounted on the front-end boards. Tests will be performed on each board and bad chips replaced as needed.
- If the yield is not high, an automated test fixture will be fabricated to validate every ASIC chip before mounting on the readout boards. Board-level tests after mounting the ASICs will be conducted.
- The fully assembled front-end boards will be thermally cycled multiple times while connected to a simple DAQ system to ensure reliable operation.
- The wire-carrier boards will be thermally cycled and HV stressed.

The APAs will also undergo testing.

- The tension and electrical continuity of each wire will be measured after the plane of wires is bonded to the frame.

- After the front-end electronics boards have been installed on the APA, an initial calibration of all electronic channels will be performed. The electronic gains and noise levels of all channels will be recorded in a database.
- A cool-down stress test will be performed on each completed APA in a liquid-nitrogen environment. Electronic calibration on all channels will be performed while the APA is cold and again after it is warmed up. Significant differences in the cold and warm calibration results will be investigated and remediated.

For the CPAs, a cool-down stress test will be performed on each completed CPA in a LN2 environment. After warming up, the tension of the wire mesh will be checked.

For the field cages, the resistance will be measured along each copper strip, and between strip pairs. The resistance between two strips should exceed $1\text{ G}\Omega$, without the resistive divider.

3.9.3 Checkout (WBS 130.05.04.07)

After passing the tests at the assembly level, the APAs will be put into storage, and later transported to the LBNE Far Site. Prior to installation, another round of electronic calibration will be performed on the APAs to validate their acceptable status.

During installation, the DAQ system will be running continuously. As soon as each stack of APAs is connected to the pre-routed cables, a suite of calibration runs will be performed to validate that all connections have been made properly. Repair or replacement at this stage will still be straightforward.

After the entire TPC is assembled, a system-wide calibration will be performed at room temperature and again at cryogenic temperature in argon gas. Repair or replacement would require partial disassembly of the TPC and should be avoided unless absolutely necessary.

The responsibility and authority for the design, installation and use of the detector quiet-power distribution and detector-grounding system is held by the subproject electrical engineer. This engineer has oversight responsibility for all electrical and electronics design and installation tasks, including all attachments to the detector that create an electrical connection.

4 Data Acquisition (WBS 130.05.05)

The scope of the data acquisition (DAQ) subsystem includes the design, procurement, fabrication, testing, delivery and installation of a combination of custom and commercial electronics modules, (including commodity computing and networking hardware), as well as both commercial and internally developed software.

4.1 Introduction

The DAQ subsystem will perform the primary functions of:

- Configuration, online calibration/checkout, and control of operations of detector subsystems, including the generation and distribution of timing and control signals,
- Readout of raw data from the TPC and other detector subsystems,
- Filtering the data and constructing event records to be logged to persistent storage media,
- Control of, and readout of data from, devices providing real-time information on detector, subsystem and environmental conditions,
- Providing user/operator interfaces for these functions via a run control system, and
- Receiving and handling the LBNE beam-spill signal.

In this chapter, a reference design for the DAQ subsystem is presented. The development of this design is guided by recent experience gained in the development of relevant systems for the $\text{NO}\nu\text{A}$ [24] and MicroBooNE [25] experiments, as well as from running experiments with comparable channel counts and/or experimental conditions, such as D-Zero, CDF, MINOS and ICARUS.

The DAQ subsystem is to be located external to the cryostat vessel, with components in the detector hall and in an on-site control room. The primary interface is with the TPC front-end electronics. Additional interfaces are with the front-end electronics systems for the photon-detector subsystem, with the Fermilab Accelerator complex (the beam-spill signal), and with the cryogenics subsystem (for logging of conditions).

The DAQ subsystem reference design described in this chapter consists of the following components:

- custom ‘Data Concentrator’ modules located in the detector hall to receive data from the TPC (transmitted via redundant LVDS lines) and to carry out low-level data processing operations (these connect to the network, listed next)
- a network consisting of commercial ethernet switches located in the detector hall and a commercial router located in the counting house/control room, (for the transmission of data to the farm, listed next)
- a local farm of commodity computers that provide trigger event-building and real-time processing/event reconstruction functions
- a custom timing system consisting of a master unit that locks onto a GPS clock and distributes timing signals to the data concentrator modules via slave units
- dedicated computer nodes that host run control, routing control, node supervisor and slow controls processes

The DAQ subsystem does not include power-supply hardware for the TPC or front-end electronics, nor does it include the cryogenics subsystem process-control and monitoring functions.

4.2 Design Considerations

4.2.1 Physics Considerations

Physics considerations determine the scale of the primary tasks of digitized TPC data read-out, event building and online processing. In addition to rates for processes of interest, the DAQ subsystem design depends critically on the specifications for the TPC and front-end electronics systems, chosen to satisfy the LBNE physics requirements. As described in Chapter 3, obtaining sensitivity to signals that occur independently of the LBNE beam spill, such as those from nucleon decay or supernova-neutrino bursts, requires a free-running transmission of data from the TPC front-end electronics. The sampling rate of 2 MHz has been

chosen so as to achieve the required position resolution along the ionization drift direction. The task of data transfer is facilitated by multiplexing and data compression/zero-suppression in front-end ASICs in the LAr, and by redundant data lines that provide connection to data-acquisition hardware located outside the cryostat. With these specifications, event building and triggering/filtering can be accomplished outside the cryostat. The LBNE beam-spill signal and data from the photon-detection system are considered part of the data stream, and can be used at this stage to select events for processing and storage through physics-specific data streams, as desired.

4.2.2 Technical Considerations

In addition to physics considerations, DAQ design goals include minimizing the impact of single-point failures and maximizing the use of commercial components. For the reference design described here, sited at the 4850L of the Sanford Laboratory, the atmospheric-muon rate is small enough – 0.1 Hz within the full LAr-FD active volume – to contribute only negligibly to the DAQ bandwidth requirement. For reference, the rate at the alternate 800L site is estimated to be 500 Hz within the active volume. This and other assumptions are discussed below in Section 4.2.3. The requirements on the DAQ system are listed in the requirements documentation [12].

4.2.3 Event Rates and Timing

Signals associated with beam events will be localized within the TPC and synchronous with discrete ($\mathcal{O}(1\text{ s})$ rep rate) beam-spill intervals spanning approximately $10\ \mu\text{s}$. However other physics events of interest will occur at random times, and can be dispersed throughout the TPC volume as in the case of neutrino bursts from supernovae. Other specific signatures, such as very slow-moving magnetic monopoles ($\beta < 10^{-3}$) may involve signals spanning sample times exceeding the 2.3-ms maximum ionization-drift time.

Cosmic-ray muons dominate the physics rate, even at the proposed 4850L site. However, this rate is negligible with respect to noise sources. This is not the case at other possible detector depths: at the alternate site identified at 800L, the total rate in the detector is estimated to be approximately 500 Hz, accounting for variations in overburden due to topography at the prospective cavern site, while the rate at the 300L is expected to be higher than this by a factor of 10. For shallow depths, the frequency of muon incidence is comparable to the maximum duration (2.3 ms) of the signal from an event, and hence cosmic-ray muons would have an impact on the required bandwidth of the DAQ system: while the system described here has sufficient bandwidth to operate at depths as shallow as the identified 800L site, going shallower still would impact the design.

As described earlier in this report (see Figure 3–18), the cold electronics for a single Anode

Plane Assembly will consist of twenty 128-channel Front-End Readout Boards, each providing a single digital input to a 20-channel Data Output Board, which includes a $20 \times$ MUX stage into a driver for a redundant pair of LVDS outputs.

The Front-End Boards will generate zero-suppressed data: worst-case scenarios (i.e., > 10 GeV EM showers contained within a single APA) indicate roughly a factor of ten reduction in the number of samples read out with respect to the maximum (2304 wires \times 4625 $0.5\text{-}\mu\text{s}$ samples per wire). For cosmic-ray muons, the rejection factor is estimated to be ~ 200 . The rejection factor is of course much higher in APAs not containing any portion of a physics event. Radioactive decay from ^{39}Ar and ^{85}Kr in the LAr, and to a lesser extent from detector materials (U/Th/Co/K), is estimated to provide a 65-kHz/ APA rate of activity of energy above about 300 keV (0.3 MIPs) but less than ~ 5 MeV, while electronics noise (assuming $10:1$ S/N for 1 MIP, and a threshold of 0.3 MIPs) will contribute a relatively low rate per APA of singles. Table 4-1 provides a summary of these rate estimates. Work is ongoing to further refine them.

Table 4-1: Per-APA estimates of rates and data sizes/rates for various processes. Unless otherwise stated, estimated numbers of samples and data rates assume suppression of signals below 0.3 MIP. ‘Inst. Data Rate’ refers to the number of bits in a 2.3-ms long data block divided by this time interval, while ‘Avg. Data Rate’ factors in the process rate. A 12-bit ADC is assumed, and no allowance is made for data items such as time-stamp, channel identifier, etc.

| Process | Rate (kHz/APA) | Samples (per APA) | Inst. Data Rate (Mbps) | Avg. Data Rate (Mbps) |
|---|--------------------|--------------------|------------------------|-----------------------|
| Generic 2.3 ms interval (not zero-suppressed) | 0.43 | 1.06×10^7 | 55,000 | 55,000 |
| Cosmic ray muons (4850L) | 6×10^{-7} | 5×10^4 | 260 | 1×10^{-4} |
| Cosmic ray muons (800L) | 0.0034 | 5×10^4 | 260 | 2.0 |
| 10 GeV EM shower | — | 1×10^6 | 5,200 | — |
| Radioactivity: U/Th (γ 's) | ~ 1 | 40 | 0.48 | 0.48 |
| $^{39}\text{Ar}/^{85}\text{Kr}$ (β 's) | 63 | 24 | 18 | 18 |
| Electronics noise (not common mode) | ~ 1 | 15 | 0.2 | 0.2 |

It can be concluded from the table that the average data rates out of the front-end electronics system are manageable: about 20 Mbps of ‘salt and pepper’ per APA due to radionuclides in the Ar and TPC materials. Large beam- or atmospheric-neutrino interactions or showering ultra-high-energy cosmic-ray muons will result in high (Gbps-level) instantaneous rates on

the scale of the maximum ionization drift period, but contribute negligibly to the average rate. With sufficient buffering in front-end ASICs, as described in Chapter 3, the plan of having a single LVDS output line per APA (plus a second one for redundancy) is easily realizable. However, to be conservative, and to provide opportunities for collecting data with relaxed zero-suppression, the DAQ reference design described below allows for as many as 20 output lines per APA (one per front-end board), each operating below 24 Mbps. This leads to a capacity for APA output rates up to 480 Mbps, well above the ~ 20 Mbps expected.

4.3 Architecture Summary (WBS 130.05.05.02)

The reference design of the DAQ system is summarized in block diagram form in Figure 4-1. Component counts are given in Table 4-2. The main elements of the design are described in the following sections.

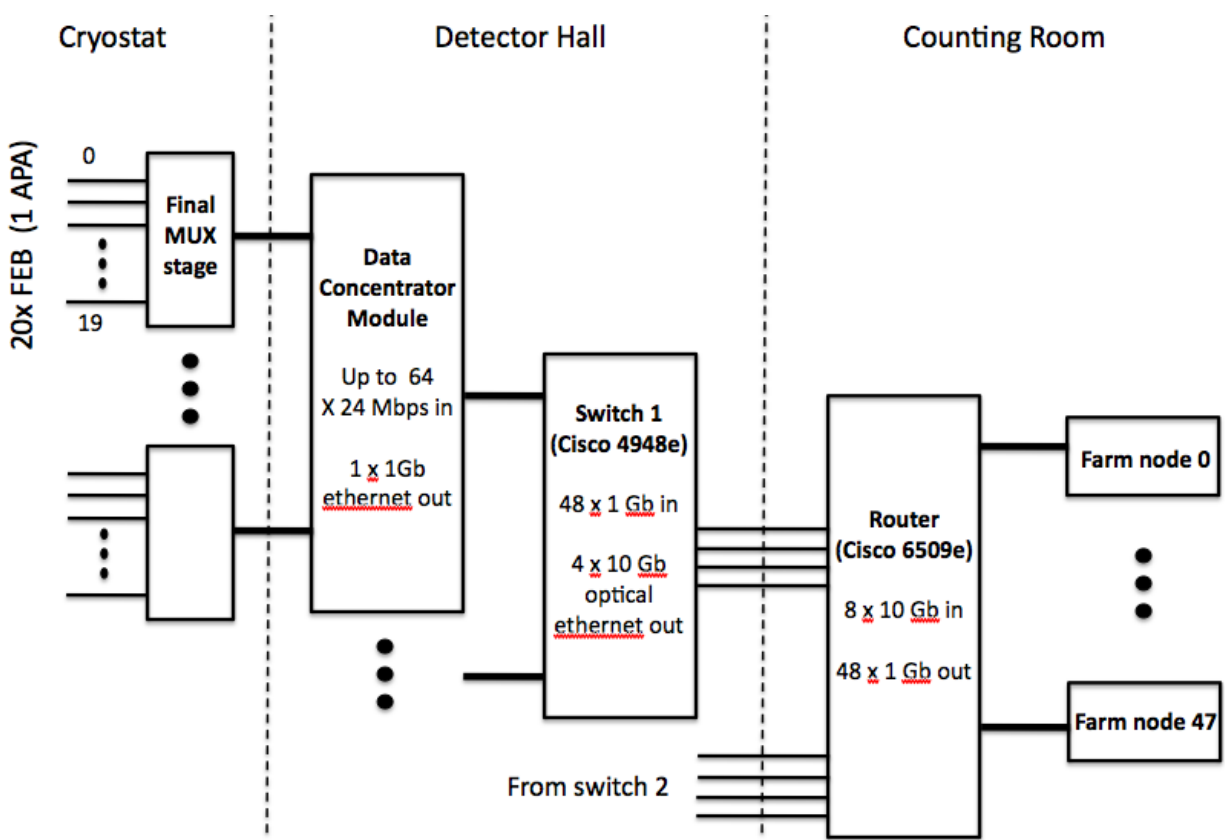


Figure 4-1: Block diagram depicting the DAQ reference-design architecture

Table 4-2: DAQ subsystem component counts for one 20-kton module/cryostat. The total component count for the two-cryostat LAr-FD will be twice what is shown.

| Quantity | Description |
|----------|--|
| 54 | Data Concentrator Modules |
| 2 | Ethernet Switches (Cisco 4948E or similar) |
| 1 | Ethernet Switch Chassis (Cisco 6509E or similar), with 8-port input optical 10-GB ethernet interface module, and 48-port output 1-GB ethernet interface module |
| 1 | |
| 1 | |
| 48 | Data Farm compute nodes |
| 1 | Readout Supervisor compute node (not shown in Figure) |
| 1 | Routing Master compute node (not shown in Figure) |
| 1 | Master timing unit + GPS receiver (not shown in Figure) |
| 9 | Slave timing units (not shown in Figure) |
| 1 | Run Control compute node (not shown in Figure) |
| 1 | Slow Controls compute node (not shown in Figure) |

4.4 Data Concentrator Module (WBS 130.05.05.03)

The LBNE/LArTPC Data Concentrator Module (DCM) serves as the primary interface between the cold TPC electronics and the DAQ subsystem, with the main task of receiving serial data pushed out of the front end. It also packetizes and transmits the data to ‘data farm’ computers for event building via an ethernet switch array, described in Section 4.5. Finally, it will provide the interface for transmitting timing and control signals to the cold electronics. As such, it is envisioned to provide the functionality of NO ν A’s custom electronics module of the same name, and similarly, the digital portion of MicroBooNE’s “Front End Module” (FEM) cards.

For the purposes of this conceptual design, the NO ν A DCM is considered as is. Several NO ν A prototype modules are shown in Figure 4-2. The NO ν A DCM consists of 64 input ports (RJ45 sockets), and a single 1 GB output line. A large FPGA provides preliminary data processing capability. A processor (running Linux) provides local control for configuration, buffering and routing functions.

Assuming the Data Output Board is implemented in the cold volume with 20 \times muxing to provide a single output line per APA, only a handful of NO ν A-style DCMs would be required to read out the entire detector. Considering typical data rates of order 20 Mbps per APA (see Table 4-1), the 1 Gb (80 MB/s) DCM output bandwidth could comfortably accommodate more than a few APA’s. More likely, the LBNE DCMs would be designed for fewer input lines in this case, in order to distribute them rationally.

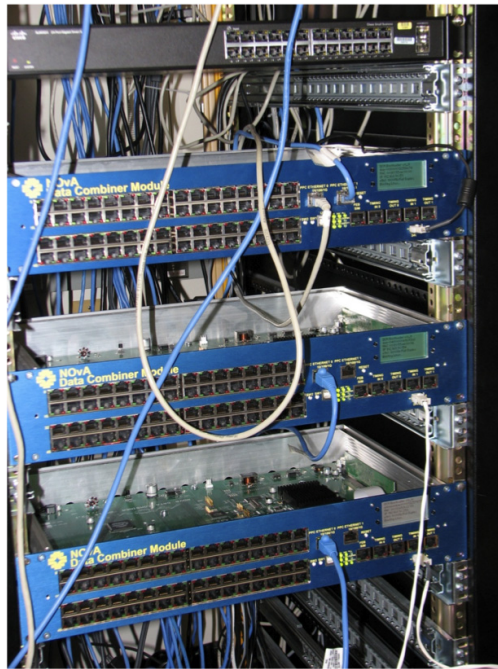


Figure 4–2: Photograph of several prototype NO ν A Data Concentrator Modules.

On the other hand, allowing for the possibility of 20 output lines per APA, the NO ν A DCM footprint is well matched to the LBNE APA granularity. In this case, a single DCM could serve two or three APAs. Outputs from auxiliary detector elements could make use of the otherwise unused DCM input channels. To obtain a conservative estimate of costs, this configuration is considered with two APA's per DCM.

4.4.1 Data Processing/Handling

As currently imagined, buffers in the front-end electronics will constitute a source of variable-length (i.e., zero-suppressed) time-ordered sequence of samples for each channel (wire) registering activity, along with a channel address and time stamp. Although this format is likely to be suitable for transmission to the data farm, the DCM's task of generating ethernet packets also provides an opportunity for additional data processing. If re-formatting or additional zero suppression is desired, it could be done here.

4.4.2 Timing, Control and Configuration Signals

The DCM will provide the interface for the transmission of timing, control and configuration signals to the TPC/front-end electronics. More detail on these signals is given in following sections.

4.5 Ethernet Switch Network (WBS 130.05.05.04)

The network accomplishing transmission of data from DCMs to the data farm will consist of two layers of ethernet switch arrays. The first level will reside in the detector hall, and is imagined to be able to operate with little external control. Commercial switch modules such as the Cisco 4948E are well suited for this application. The 4948E has 48 1-GB input ports and four 10-GB optical output ports, as well as 175 MB of buffer memory. Two modules will support the required data throughput for the entire detector.

The second level will be deployed in the counting room and will serve as a router to the data farm nodes located there. For this application the Cisco 6509E switch chassis provides a possible implementation. This would be loaded with a single 8-port 10 GB blade for input data from the 4948E's, and a single 48-port 10/100/1000 MB blade for 1-GB output to farm nodes.

Routing information will be provided to DCMs by a Routing Master. This task will run on a computer located in the counting room. It will monitor the state of data farm nodes, and provide routing information to DCMs.

4.6 Event Building and Triggering (WBS 130.05.05.05)

The event building and triggering function of the LAr-FD DAQ system will be performed by the data-farm computers. Event data will be staged locally before being transmitted in quasi real-time (nominally to Fermilab) for archival to persistent storage.

4.6.1 Event Building

At present it is imagined that an event will consist of raw data from the entire detector, spanning a time interval yet to be determined. To construct such an event, DCM packets corresponding to data with a common (range of) timestamp value(s) will be routed to a particular data-farm node.

An alternate scenario considers events as being localized to individual APAs, or possibly small APA clusters. Individual farm nodes would work only on the corresponding data to generate event records. This concept is attractive in that (1) the routing of data to farm nodes is simplified, and (2) event record sizes are kept as small as possible. The main drawbacks are that (1) offline processing/analysis of these event records for physics events with activity spanning geographical boundaries would become more cumbersome; (2) certain physics studies, such as proton-decay searches, might benefit from simple access to data-registering

activity in remote sections of the detector ; and (3) auxiliary data would either be unnecessarily duplicated or would have to be stored in its own event record, again adding complexity to the data-analysis process. Evaluation of this alternative is ongoing.

4.6.2 Event Data Model

We will need to develop an Event Data Model (EDM). It may be advantageous to implement the raw data EDM in a custom format, as opposed to one based on ROOT. Experience with MicroBooNE will be helpful in optimizing the design for this.

4.6.3 Triggering and Selection for Output Streams

Significant work remains to understand how data-farm nodes will carry out event filtering and creation of separated physics/task-specific data streams. Use of the LBNE beam-spill signal to identify events recording beam-induced activity is expected to be straightforward. However, identifying events of interest that lack such a signal requires study. Is it sufficient to find a suitable way to generically veto events based on lack of coherent detector activity, or must ‘positive’ signatures for each of the physics processes of interest be identified for triggering? To indicate the range of signatures, these processes of interest include, for example, (1) beam-induced events for which the corresponding beam-spill signal is missing due to network failure or other malfunction, (2) atmospheric-neutrino interactions, (3) supernova-neutrino bursts, (4) proton decay, and (5) magnetic-monopole incidence.

4.6.3.1 Rejection of Event Records

Since the dominant rate is due to dispersed low-energy activity associated with radionuclide decays, simple trigger primitives can be generated and combined so as to reject event records failing well-defined (and easy-to-model) criteria. For example, event records in which no APAs register energy deposition in excess of some threshold (say 5 MeV) will be sufficient to reject most background events.

4.6.3.2 Event selection for Physics-Specific Streams

Even given the above statements about event rejection, it is desired to perform some type of high-level event reconstruction to identify candidates compatible with specific physics signatures. This level of analyses is essential both for online detector performance diagnostics as well as for the case where candidate event records of particular types are to be written to parallel output streams. For the former application, an unanticipated shortfall in the

DAQ data farm computing capacity could be easily addressed through establishment of a separate computer farm for online analysis. For the latter application, it would be necessary to adjust the size of the data farm depending on the processing requirements. These are not known at this time. However, with anticipated costs for commodity computing systems, it is not expected that the overall cost of the DAQ/online computing systems would increase significantly relative to the currently budgeted system.

4.7 Timing System (WBS 130.05.05.06)

Comparable requirements and conditions suggest a timing system similar to that being implemented for $\text{NO}\nu\text{A}$. That system meets the requirements of deterministic timing and coherence of signals distributed across the entire detector. It consists of a Master Timing Unit (MTU) whose main task is generation of GPS-based timing packets, and an array of Timing Distribution Units (TDUs). The TDUs are geographically distributed: they compensate for propagation delays before transmitting timestamp packets to the front ends via DCMs. Such a system could work well for LBNE and may be able to be adapted with only minor design modifications. For $\text{NO}\nu\text{A}$, each TDU is associated with 12 DCMs; for LBNE a reasonable distribution could be achieved with 9 TDUs, one for every 6 DCMs (i.e., spaced at intervals of 5 m along the length of the detector).

4.8 Run Control (WBS 130.05.05.07)

The scope of functionality of the Run Control system includes operation of DAQ subsystem components, configuration of front-end electronics, control of power supplies and other auxiliary equipment, and control of data collection. Development of a user interface for experimenters during data-taking and for technical personnel to assist with commissioning and debugging activities is key. To date, limited effort has been put forth in the design of the Run Control system. To the extent that the challenges faced are similar to those that have been addressed at MINOS and ICARUS, no technical obstacles are foreseen. As the designs of the DAQ and other subsystems continue to develop, specifications of the Run Control system will become more concrete.

4.9 Slow Control Systems (WBS 130.05.05.08)

The Slow Control system is a critical element of the DAQ, providing the main interface to power supplies for the detector and electronics as well as to equipment used to monitor the operational status of the detector and supporting systems. As in the case of the Run Control system, the development of the conceptual design for the Slow Control system is in its early

stages. Again, based on experience from other experiments, no obstacles are foreseen with regard to the development of a robust system.

4.10 DAQ Infrastructure (WBS 130.05.05.09)

4.10.1 Wide Area Network

As in the case of MINOS and NO ν A, it is expected that event data can be transmitted over the network to Fermilab. Although rates for events of interest are comparable, data throughput for the LBNE LArTPC is expected to be at least an order of magnitude higher. A detailed analysis of the requirements of Sanford Laboratory for the appropriate level of connectivity within and off the lab site will need to be carried out.

4.10.2 Online Data Storage

To protect against significant periods of absent network connectivity, it is desired to store a significant amount of the data emerging from the DAQ to local storage. A local data storage facility of ~ 100 TB is expected to be more than adequate for five days worth of detector data, even without prescaling cosmic-ray muon events.

4.10.3 Power and Cooling

Power and cooling requirements for the DAQ system described here are modest. DCMs operate at below 50 Watts each, while the maximum power consumed by each of the two Cisco 4948Es is 275 Watts. Assuming power supplies that operate at 75% efficiency, and accounting for other components, the total DAQ subsystem budget for power in each 20-kton detector/cryostat hall is likely to be below 15 kW.

5 Photon Detector (WBS 130.05.07)

5.1 Introduction

The scope of the photon detector system includes the design, procurement, fabrication, testing, delivery and commissioning of a subsystem that meets the performance requirements for light collection in the LAr-FD. This subsystem will provide absolute event timing, thereby enhancing the detector's event localization and particle-identification capability. Absolute event timing also makes possible time-of-flight measurements.

5.2 Physics Motivation for Photon Detection

5.2.1 Photon Production in Liquid Argon

Photons are produced in liquid argon by scintillation and by Cherenkov radiation induced by charged particles. The former dominates by a factor of five. The Cherenkov light is directional, while the scintillation light is isotropic. Twenty-three percent of the scintillation signal is in prompt 6 ns light and 77% of the signal is in late 1.6 μ s light. The decay photons are in the VUV and have a wavelength $\lambda = 128$ nm. The photon-detector system is designed to detect these 128-nm scintillation photons.

Identification of the different charged-particle types depends on accurate measurements of ionization along tracks. Some drifting ionization electrons are lost due to impurities in the LAr. The measured ionization charge, Q , is therefore lower than the true ionization charge, Q_0 , and is expressed as $Q = Q_0 e^{-t_{drift}/\tau} = Q_0 e^{-(t_{arr}-t_0)/\tau}$, where t_{drift} is the measured drift time and τ is the lifetime of the drifting ionization electrons, measured by purity monitors in the cryostat. The drift time is the time difference between the time of occurrence of the event, t_0 , and the arrival time of electrons at the anode plane, t_{arr} . The photon-detector system detects the arrival time of the event, t_0 , which allows one to both correct for recombination and locate the event in the detector.

5.2.2 Impact on Physics Analyses

The photon-detector system contributes to three physics analyses: proton decay, supernova physics and time-of-flight studies. The system also helps to identify spallation backgrounds and δ -rays from cosmic-ray muon interactions in the LAr. For proton decay, the photon-detector system identifies the K^+ decay background entering the sides of the detector from outside the fiducial volume. It also recovers information from the K^+ decay channel into pions. For supernova physics, the photon detector is important in determining both the time structure of the arriving “collapse” neutrinos as well as providing a robust estimate of their energy. For the time-of-flight analysis, the reference design avoids introducing systematic errors that would comprise the measurements; however the timing system would require the addition of atomic clocks to reduce the errors to a level that makes the analysis feasible. The photon-detection system can locate a cosmic-ray muon in the detector. With that information, a gate can be opened and a search initiated that identifies decays by short-lived radioactive isotopes induced by spallation reactions and δ -rays generated along the track.

The “golden” proton-decay channel in LAr ($p \rightarrow \nu K^+$) is guaranteed to be rare, so it is crucially important to eliminate all backgrounds that could masquerade as the signal. Photon detection offers high efficiency for the elimination of the important backgrounds that occur when a cosmic-ray interaction produces a K^+ or a K_L^0 that undergoes a charge-exchange reaction to a K^+ outside the exterior cathode planes and enters the fiducial volume of the detector. The decay of the resulting K^+ in the TPC could mimic a proton decay if not positively identified. The photon-detection system allows better use of the outside drift cells thereby increasing the fiducial mass of the detector.

The key to discovering supernovae with LBNE is the efficient detection of the neutrinos liberated by core collapse. The photon-detection system provides an excellent additional handle for detecting supernova bursts that can be used in combination with counting wire-plane clusters. By including both the prompt and late scintillation light in the trigger, a supernova-burst search can be extended reliably out to distances well beyond 10 kpc. These detected neutrinos see deep into the collapsing core and reveal both the time history of the collapse and the energy of the neutrinos produced. In addition, the oscillation of these neutrinos as they propagate to earth reveal important fundamental properties of the neutrinos themselves.

One potential source of background in LAr is spallation reactions with Ar induced by the muon as it traverses the detector. Short-lived isotopes of C, N, O, Be and Li that β decay with $\tau_{1/2} < \text{few seconds}$ could introduce backgrounds that the photon-detection system would identify by their proximity to the track. Another source of background is the delta-ray generation along the track. These are not backgrounds that are easily calculable because the cross sections for muons on Ar are unknown. But by opening a time window along the known track position, these background events can be identified and corrected for.

The OPERA experiment reported on the detection of superluminal speeds for muon neutrinos from CERN to the Gran Sasso [26] however this effect is now thought to be due to instrumental error. The magnitude of the OPERA effect at LBNE would be that ν_μ neutrinos arrive $\sim (1300/730) \times 61 \text{ ns} = 110 \text{ ns}$ early. In the reference design described below, the electronics introduce an error in $t_0 \sim 10 \text{ ns}$. The electronics proposed for the light detection system is not a barrier to investigation of this disputed measurement.

5.3 Reference Design

The reference design is built around adiabatic light guides, as first demonstrated by Bugel *et al.* [27]. These are extruded polystyrene bars with a co-extruded layer of polystyrene embedded with the waveshifter tetraphenyl butadiene or TPB [2] that converts the 128-nm scintillation photons to visible light. The light-guide core channels the waveshifted 420-nm photons to a photomultiplier tube (PMT) modified to work efficiently at LAr temperatures. Four light guides are ganged together into a “paddle” held by a frame that mounts onto an APA (described in Chapter 3). This paddle frame includes a mount for a PMT. There are ten paddle frame assemblies on each APA. Each PMT has a custom cryogenic base. The HV is brought into the base and the readout signals taken out on the same cable. There are therefore ten cables for the photon-detection system per APA. There are 216 APA frames (108 per cryostat) in the reference design. There are 4 light guides/paddle \times 10 paddles/APA frame \times 216 APA frames = 8640 adiabatic light guides required in the reference design.

The readout electronics are warm (outside the LAr) and set to trigger at 0.5 PE. The readout is based on the MicroBooNE 60 ns shaper and commercial electronics where possible. The signal is digitized at 65 MHz and the integration time is $3.2 \mu\text{s}$, or twice the lifetime of the late light. The digitization of the 1.6-ns early light pulse gives four samples for the 60-ns shaped pulse; there are 208 samples per trigger. Once a trigger is received, a signal is sent to adjacent PMTs to begin recording.

5.3.1 Photon Production and Detection

For a MIP particle the number of photoelectrons detected in the the LAr-FD detector, N_{PE} , per MeV of energy of the ionizing particle can be written

$$\begin{aligned} N_{PE}/\text{MeV (MIP)} &= (N_{128}/f) \times \epsilon_E \times (N_{420}/N_{128}) \times DQE \times \Omega \\ &= 0.2 \text{ PE/MeV}, \end{aligned}$$

where (N_{128}/f) is the number of prompt 128-nm photons/MeV, corrected by $f = 0.23$ to the total (prompt + late) light signal = 3.3×10^4 for a MIP particle [28]; ϵ_E ($= 0.6$) is the reduction in light due to the 0.5-kV/cm electric field [29]; (N_{420}/N_{128}) (≈ 1) is the number of 420-nm photons emitted by the waveshifter for every absorbed 128-nm VUV photon [2]; DQE

= detector efficiency = 2.7×10^{-3} [29] and includes first estimates of the light-system design parameters for the baseline detector described here; and $\Omega = 0.0036$ geometric acceptance, including wire transmission [29].

The number of photons produced by a muon in a K^+ decay and by a typical ν in a supernova burst can be estimated with this equation:

- For a $K^+ \rightarrow \mu^+$ that decays to a 250-MeV muon: $N_{prompt} = 12$ PE; $N_{tot} = 50$ PE
- For a supernova neutrino that produces a 20-MeV electron: $N_{prompt} = 1$ PE; $N_{tot} = 4$ PE

Backgrounds in LAr were studied in the 2.3-liter WARP detector [1]. The energy spectra for these backgrounds are shown in Figure 5-1.

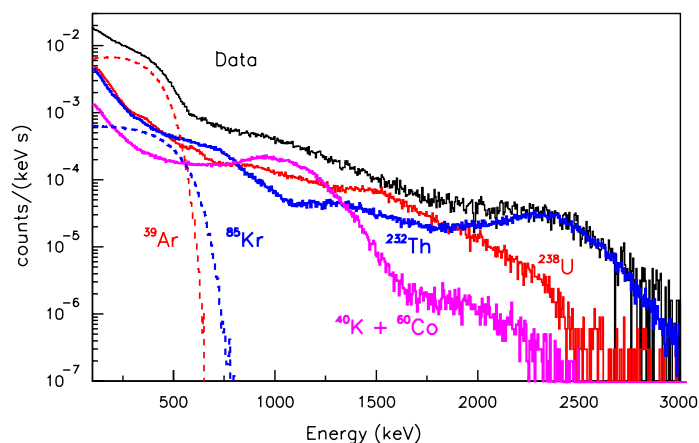


Figure 5-1: Backgrounds measured in the WARP detector [1]

The dominant internal background is from Ar^{39} beta decays. The dominant external background comes from Co^{60} contamination in the stainless steel. For a typical 0.5-MeV β from Ar^{39} , $N_{prompt}(\text{Ar}^{39}) = 0.05$ PE; $N_{tot}(\text{Ar}^{39}) = 0.2$ PE and does not present a problem. For the Co^{60} in the stainless steel, the 1.2/1.3 MeV γ 's give $N_{prompt}(\text{Co}^{60}) = 0.06$ PE and $N_{tot}(\text{Co}^{60}) = 0.3$ PE. But this background has a rate that is an order of magnitude lower at 0.5 MeV and two orders of magnitude lower at 1.25 MeV than the Ar^{39} background; it is therefore also not a problem.

5.3.2 Adiabatic Light Guides

The basic elements of the photon-detection system are adiabatic light guides. The light guides are made from extruded polystyrene bars with a co-extruded layer embedded with

TPB. Scintillation photons from the LAr are converted to 420 nm in this co-extruded layer. The optical photons propagate into the polystyrene core, where they are transported to the end by total internal reflection to the PMT.

5.3.2.1 Design

Figure 5-2 shows the details of the light-guide design. The dimensions have been chosen so that four light guides completely cover the photocathode area of the PMT.

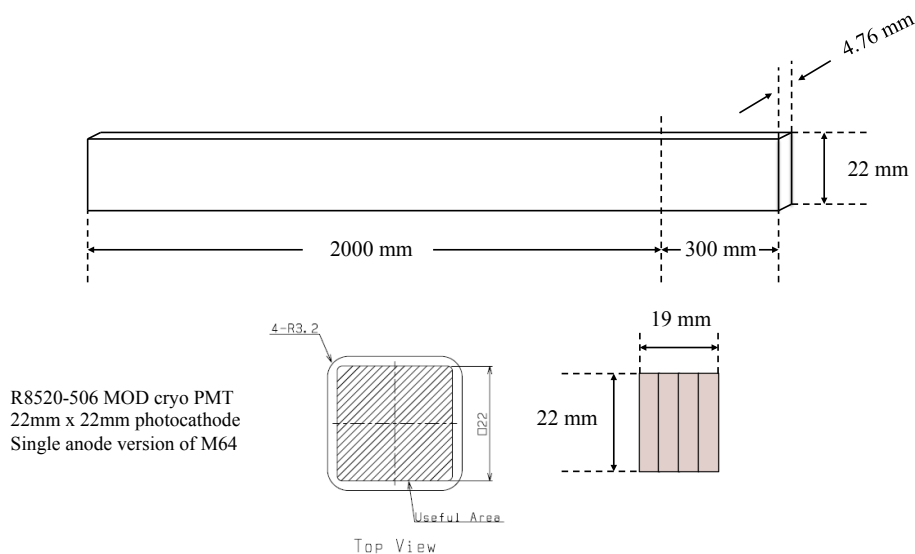


Figure 5-2: *top:* Dimensions of adiabatic light guide. *bottom:* Four light guides are ganged together to fill the photocathode area of a Hamamatsu R8520-MOD PMT.

5.3.2.2 Extrusion Technology

Particle detection using extruded scintillator is a mature technology. MINOS has demonstrated that co-extruded solid scintillator with TiO_2 embedded in the co-extrusion can be manufactured with excellent quality control and uniformity in an industrial setting. The co-extruded scintillator elements can be produced at Fermilab using the extrusion line jointly operated by Fermilab and the Northern Illinois Center for Accelerator and Detector Development (NICADD) at Northern Illinois University (NIU). NIU physicists and mechanical engineers have formed a collaboration to support development of the next generation of detectors at Fermilab's Scintillator Detector Development Technical Center. The extrusion line was purchased by NICADD in 2003. The co-extruder line was purchased by Fermilab in 2005. Fermilab and NICADD support and operate the extruder to ensure that the High Energy Physics community has access to high-quality extruded scintillator. Fermilab and NICADD personnel have been responsible for commissioning the extruder, for simulations,

production and prototyping of dies associated with specific detectors, and for productions of extrusions for prototypes and detector construction.

Scintillator strips are composed of a polystyrene core and a co-extruded, 0.25-mm thick polystyrene layer embedded with TPB. This layer is introduced in a single step as part of a co-extrusion process. The ruggedness of this co-extruded layer enables the direct gluing of the strips to each other to form paddles.

The scintillator-bar production process is characterized by an ‘in-line’, continuous-extrusion process. The polystyrene pellets are dried in a nitrogen atmosphere and automatically conveyed to a gravimetric feeder. The dopant mixture is added periodically to a different gravimetric feeder that works surrogated to the pellet feeder. These feeders have the necessary precision and reliability to ensure delivery of a constant ratio. The pellet feeder is controlled by a computer that drives the output of the twin-screw extruder to ensure the correct composition and processing. The extruder is responsible for melting and mixing the polystyrene pellets and the dopants. A twin-screw extruder will provide the highest degree of mixing to achieve a very homogeneous concentration. The outer waveshifting coating is added through material injected from a second extrusion machine (co-extruder) which mixes the polystyrene and tetra-phenyl butadiene (TPB) pellets. Currently the co-extruder is manually operated to start-up and to vary the thickness of the waveshifter coating. As the plastic emerges from the die, it goes directly into the cooling tank. There it is formed into the final shape using the sizing tooling and vacuum. It continues to be cooled with water and air until it can be handled.

5.3.2.3 Waveshifter (TPB)

The detection of 128-nm UV photons emitted by Ar is a challenging problem. Short wavelengths are ordinarily “strongly absorbed by nearly all materials used for visible optics, such as quartz or glass windows of PMTs, but they are not energetic enough to be treated calorimetrically like x-rays or γ -rays’ [2]. This obstacle can be overcome by down-shifting the VUV photons into visible photons with a waveshifter. Typically the waveshifter used is tetra-phenyl butadiene (TPB). TPB has been well-studied in this application [2,30].

The re-emission spectrum of TPB when illuminated by 128-nm photons is shown in Figure 5-3 [2]. The efficiency for TPB to convert input VUV photons is shown in Fig. 5-4 [2]. The photon conversion efficiency ($N_{420\text{nm}}/N_{128\text{nm}}$) ~ 1 used in the calculation of photon production has been taken from this figure.

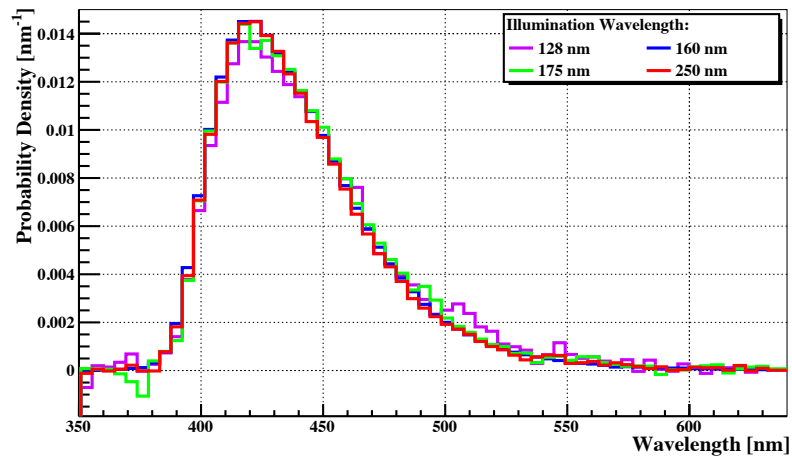


Figure 5-3: TPB re-emission spectrum from Gehman *et al.* [2].

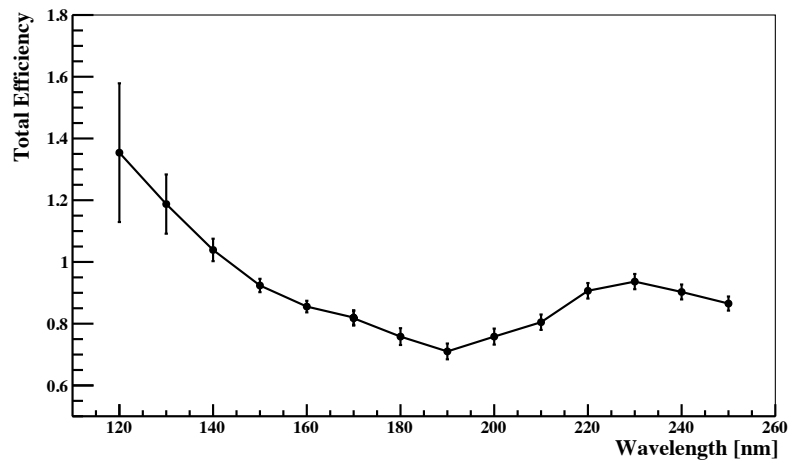


Figure 5-4: TPB efficiency as a function of VUV photon wavelength from Gehman *et al.* [2].

5.3.3 Light Guide Paddles

5.3.3.1 Design

To maximize photocathode coverage, four adiabatic light guides are ganged together into a “light guide paddle”, as shown in Figure 5–5.

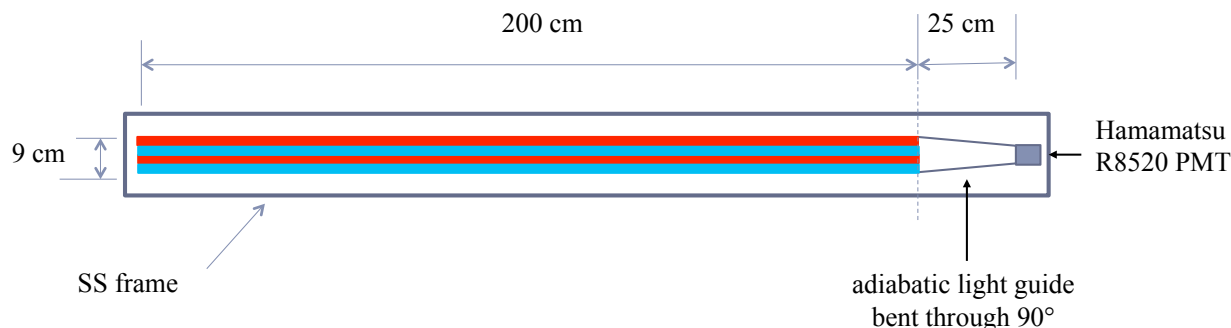


Figure 5–5: Light Guide Paddle: four adiabatic light guides bent onto a single R8520-MOD PMT.

The reference design for the assembly of four light guides into a paddle is given in Figure 5–6. The two center drawings show how the light guides are bent through 90° to fit onto the PMT.

The figure also shows how the light guides are held and clamped onto the PMT. To minimize feedthrough holes in the APA frame, the light guides are all bent in the same sense. The details of the design for the PMT holder and the PMT clamp are shown in Figure 5–7.

Figure 5–8 shows the reference design for fitting ten light-guide paddles onto an APA frame. Details of the bracket that holds the PMT to the frame are shown in Detail A. The feedthrough for the paddles through the central strut is shown in Detail B. The design for the fixturing for clamping the PMT to the frame is shown in Detail C.

5.3.3.2 Assembly

The procedures for assembling the paddles will be developed during LAr1 construction (Chapter 7). Initially, once the light guides are extruded, they will be heated in an oven and then bent through 90° into the proper shape. As seen in Figure 5–6, each of the four light guides in a paddle requires a somewhat a different shape, depending on its position in the paddle. The bent light guides will then be set into a custom-designed fixture and glued. It is important that the four ends of the light guides be clamped in the fixture during this gluing assembly step to assure that they are properly aligned for flycutting. Once the glue

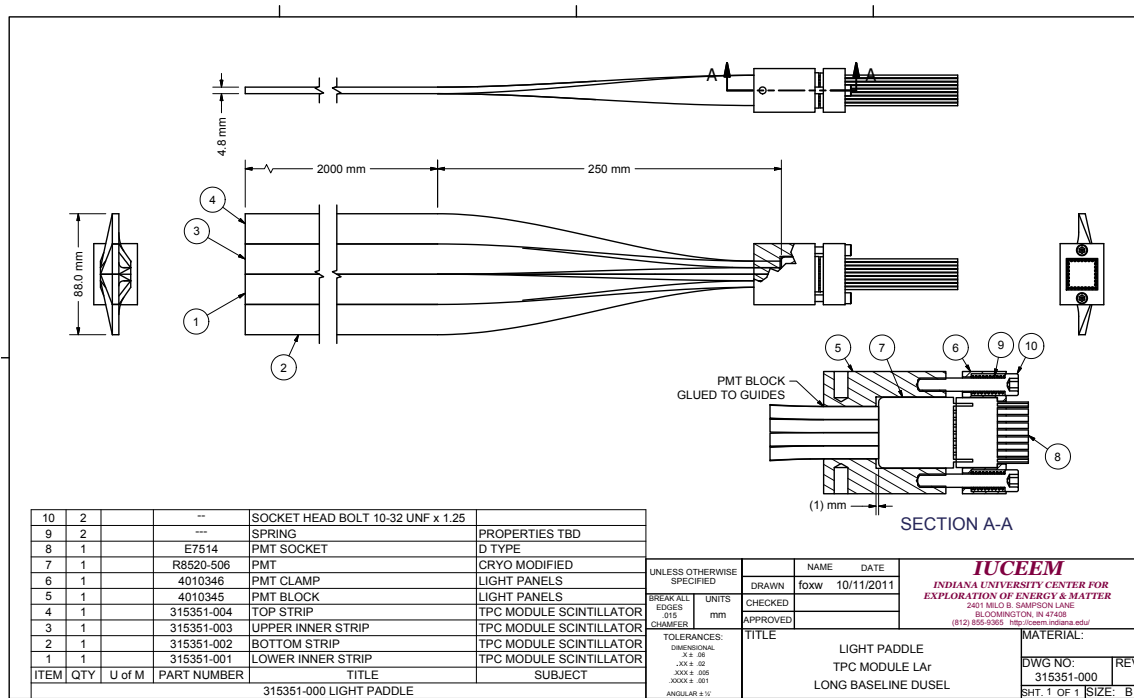


Figure 5-6: Four adiabatic light guides bent through 90° and clamped onto an R8520-MOD PMT.

sets, the ends will be flycut and set into a PMT block. Finally, the paddle assembly will undergo quality control testing, most likely with an LED in a dark box.

5.3.4 Photomultiplier Tubes

The PMTs in the reference design photon-detection system are Hamamatsu R8520-MOD cryogenic tubes, shown in Figure 5-9.

These 1-in tubes are single-channel, cryogenic versions of the Hamamatsu M16 tubes used in the MINOS far detector. General information on these tubes is given in Figure 5-10. The PMT characteristics are given in Figure 5-11.

5.3.5 Electronics

Scintillation light from LAr comes from two different excited states with lifetimes of 6 ns and 1.6 s. Only a limited amount of light is collected by this system, so the electronics is designed to collect the light from both excited states. This system needs to determine the event time to a precision of 1 to 2 μs, so including the late light has no impact on the desired resolution. The basic architecture is shown in Figure 5-12.

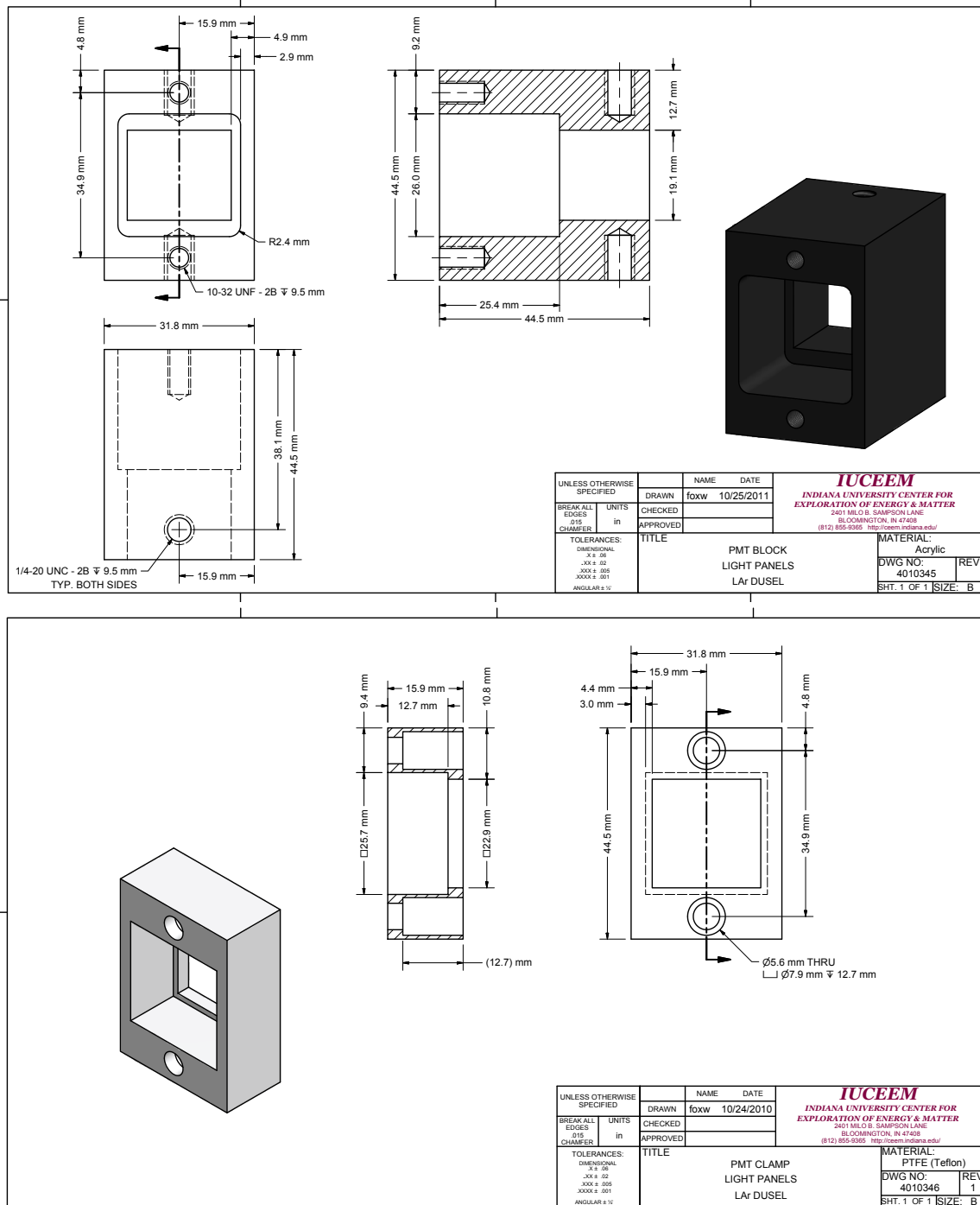


Figure 5-7: top: PMT holder. bottom: PMT clamp.

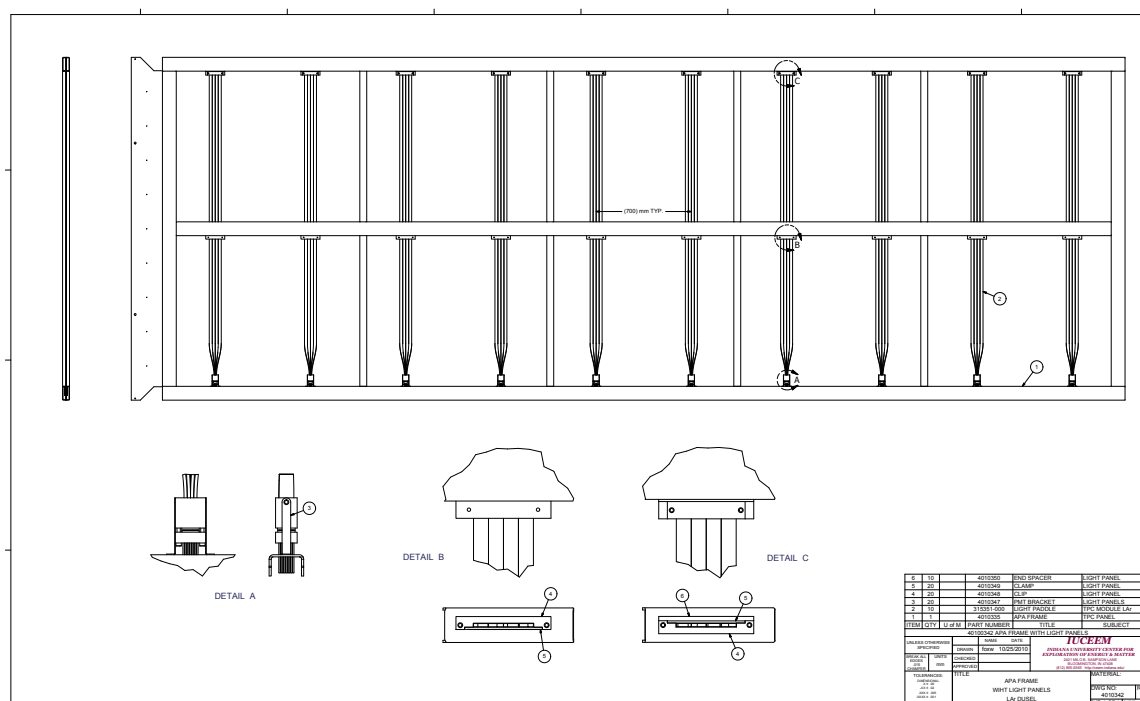


Figure 5-8: An APA frame fit with ten light-guide paddles.

Signals from the PMT are transmitted from the LAr to electronics racks located near the ports on top of each cryostat. The signals on the HV cable are picked off using a blocking capacitor and then sent to a pulse shaper with a 60-ns shaping time. From there, the shaped signals are sent to a commercial 32-channel wave-form digitizer module and digitized to 12 bits at a frequency of 62.5 MHz. One channel of the digitizer is used to digitize the experimental clock which has both the 32-MHz clock and special time markers encoded on it that are identical to those generated in the DCMs used to read out the TPC data. This allows the onboard FPGA to determine the time that an event occurred and also to determine when a time slice has ended so that data from the last time slice can be sent to the host PC. The FPGA is also used to zero-suppress the PMT data, time-stamp it and send the time-stamped data to the host PC. A simplified diagram of the FPGA logic is shown in Figure 5-13.

The host PC then groups the data into time blocks (which may contain multiple time slices) and sends the data to the farm node that is processing the APA data for a given time.

The photocathode is located close to the APA wire planes, which are at or near ground potential. In order to eliminate any possibility of HV breakdown, the photocathodes are run at ground potential. The PMT output signal is then offset from ground by the full DC bias voltage. This has the advantage that only one cable per PMT is required but it has the disadvantage that the signals must be AC-coupled into the readout electronics.

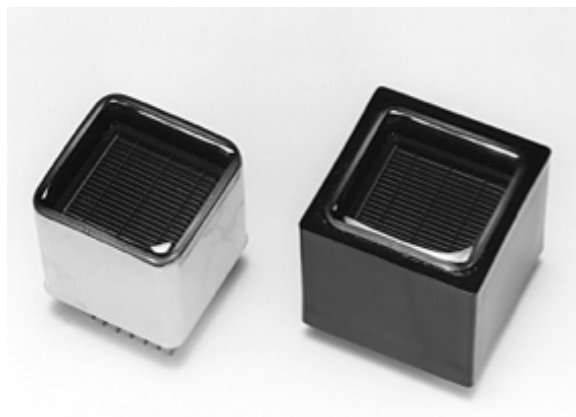


Figure 5-9: Hamamatsu R8520-MOD PMT.

General

| Parameter | | Description / Value | Unit |
|-------------------------------|------------------------|----------------------|--------|
| Spectral Response | | 300 to 650 | nm |
| Window Material | | Borosilicate Glass | - |
| Photocathode | Material | Bialkali | - |
| | Minimum Effective Area | 22 x 22 | mm |
| Dynode | Structure | Metal channel Dynode | - |
| | Number of Stages | 10 | - |
| Weight | | Approx. 28 | g |
| Operating Ambient Temperature | | -186 to +50 | deg. C |
| Storage Temperature | | -186 to +50 | deg. C |

Figure 5-10: Hamamatsu M16 tubes, general information

Since the electronics are outside the cryostat, there must be a HV-feedthrough connector. There are a total of 1,080 connections per cryostat, assuming there are 216 APAs with ten PMTs per APA and two PMTs per HV power supply. Consequently, this must be a very reliable connector that occupies little space. The current plan uses the Reynolds 8-channel HV connector. This connector has eight channels in a round connector that is 1.187 inches in diameter and can withstand 10 kV. This connector has been used in D-Zero for the LAr calorimeter. It is mounted in the ullage and has been operated at 2 kV for nearly 25 years.

The D-Zero system has eight tefzel-insulated cables inside an overall braided shield. This cable is ~0.25 cm in diameter. Since the signals are also on this cable, the lack of individual shields may lead to unwanted crosstalk and poor signal propagation. The best solution is to add a braided or foil shield to the individual wires in the assembly. This will significantly increase the diameter of the cable, but its diameter should still be much less than that of the connector. If the individual coax cables have too small a diameter, the impedance will be less than 50 ohms, which effectively sets the lower limit on the cable size.

The Reynolds connector has a common ground so all the braids will be connected together

Characteristics at 25 deg. C

| Parameter | | Min. | Typ. | Max. | Unit |
|--|------------------------------|------|-------------------|------|-------|
| Cathode Sensitivity | Luminous (2856K) | - | 100 | - | uA/lm |
| | Quantum Efficiency at 340 nm | - | 25 | - | % |
| Anode Sensitivity | Luminous (2856K) | - | 100 | - | A/lm |
| Gain | | - | 1.0×10^6 | - | - |
| Anode Dark Current (after 30 min. storage in darkness) | | - | 2 | 20 | nA |
| Time Response | Anode Pulse Rise Time | - | 1.8 | - | ns |
| | Transit Time Spread (FWHM) | - | 0.8 | - | ns |
| Pulse Linearity at +/-2% deviation | | - | 30 | - | mA |
| P/V Ratio | | 1.1 | - | - | - |

Figure 5-11: Hamamatsu M16 tubes characteristics

at the feedthrough. The connector is less than 1/40 of a wavelength for frequencies up to 100 MHz so this interruption in the coaxial cable should not be a significant problem.

The interior cryostat cables will be terminated at the PMT end with a coaxial HV connector such as SHV or LEMO and be plugged directly to the PMT base. This will minimize the number of connections in the cryostat. It also allows the cable to be threaded through the 1-by-3 inch openings in the cryostat frames.

There are two candidates for the HV system. One is a VME-based system designed by Fermilab and manufactured by BiRa corporation. The other is a custom system manufactured by Weiner corporation. The advantage of the BiRa system is that most of the supplies can be recovered from the D-Zero and CDF experiments. Since the purchase price of the Weiner system is less than the BiRa system (assuming no recovery from completed experiments), the Weiner one has been used in the cost estimate. Each port will require 20 channels (40 tubes). Both systems have more than 20 channels per crate so the crates will be located at every other port (if VME) or at the the end of each row of racks (if Weiner).

The signal splitter will house a HV-blocking capacitor and a 60-ns shaper that is identical to the one designed for the MicroBooNE experiment. The shaper is shown in Figure 5-14.

Output from the splitter will be fed into a commercial waveform digitizer.

Unlike a digital oscilloscope or many commercial digitizers, LBNE needs continuous digitization followed by zero suppression and time stamping. The zero suppression is controlled by a user-programmable FPGA.

Time-stamping can be done either by internal timing or by using one channel of the digitizer to digitize the experiment clock. At least one vendor has a product that uses a GPS signal to derive the time stamp. A second vendor who makes a similar product has been identified. No difficulty is anticipated in finding a commercial product that will meet the requirements.

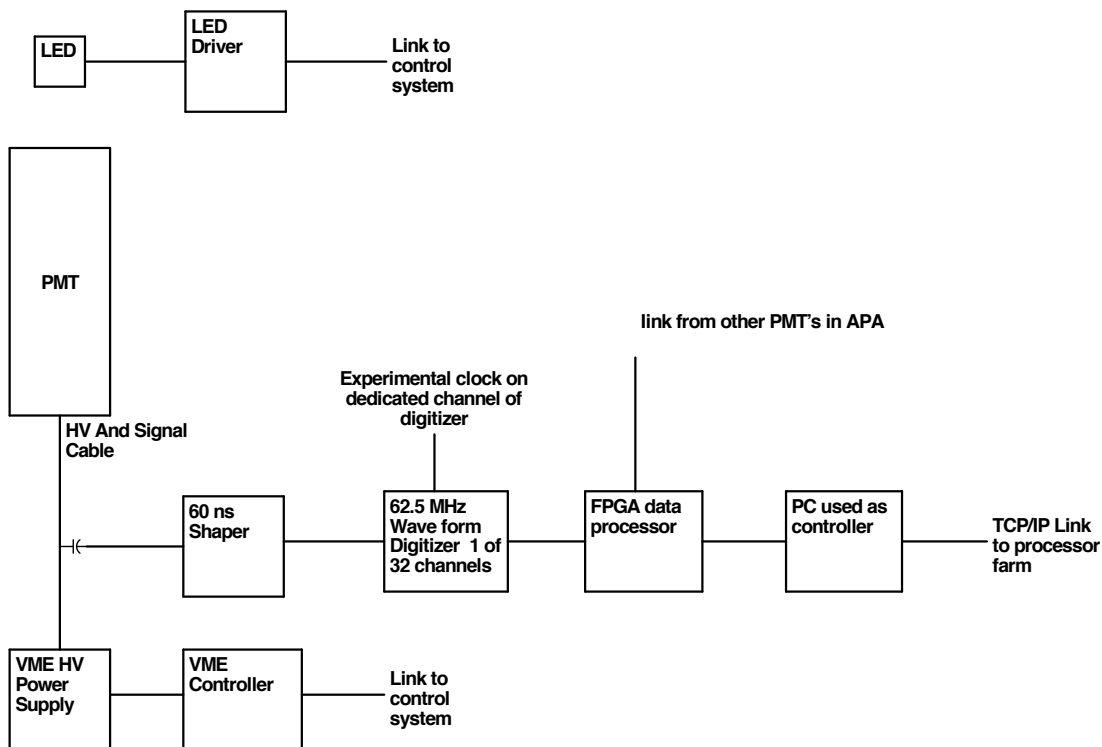


Figure 5-12: Block diagram of the electronics chain for the PMT system.

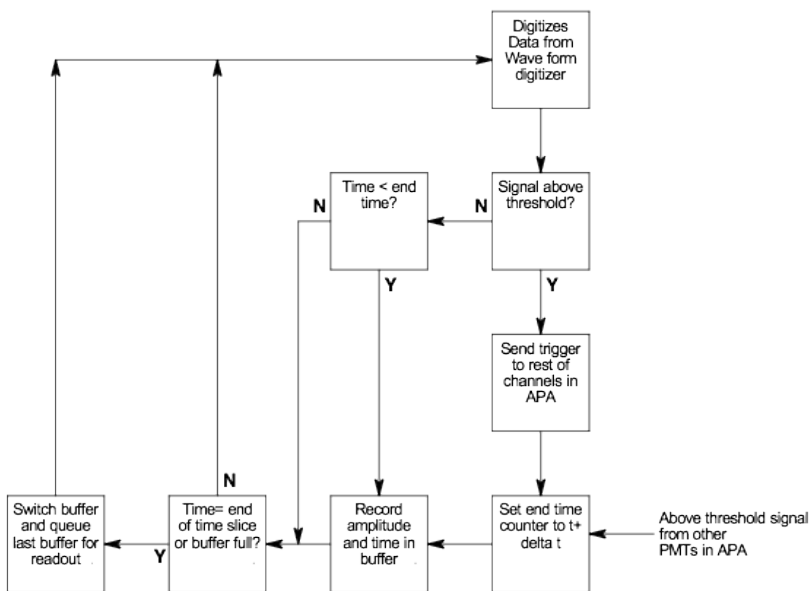
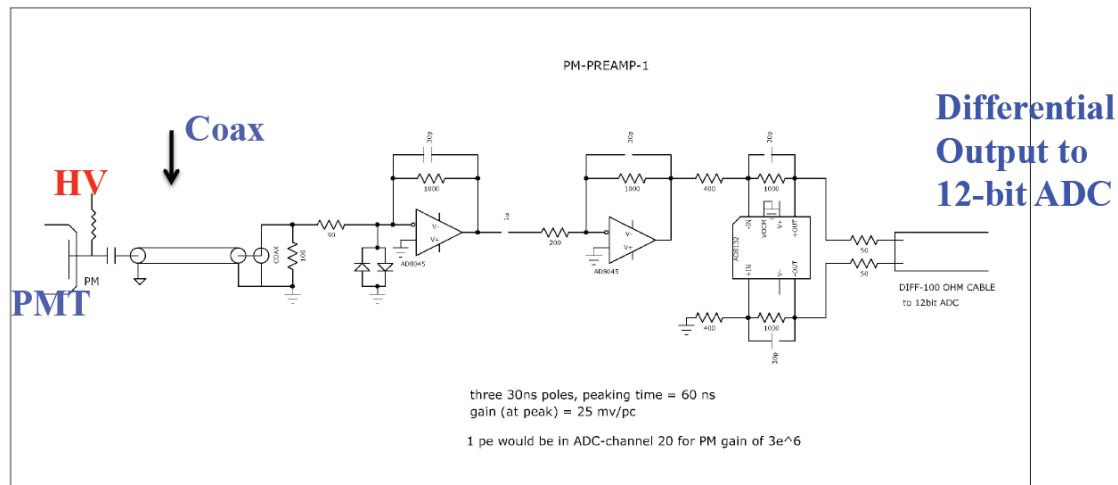


Figure 5-13: FPGA logic diagram. The time information will come from a dedicated channel in the digitizer. This logic will actually be implemented as a finite-state machine.



- Three 30 ns poles, peaking time = 60 ns
- Gain (at peak) = 25mV/pC
- 1pe would be in ADC=channel 20 for PM gain of 3×10^6

Figure 5-14: Shaper circuit designed for the MicroBooNE experiment. This figure has been taken from the MicroBooNE CDR.

All of the commercial digitizers under consideration send data to a PC over a PCI or USB bus. The digitizer FPGA will be programmed to provide appropriate timing information to the PC so that it can assemble the PMT data into packets spanning time periods that are identical to those generated in the DCMs first for the drift-chamber data. These packets will be sent to the appropriate farm node using an algorithm similar to the one in the DCMs.

5.4 Alternatives

5.4.1 Waveshifters

Two different waveshifters are being explored as more cost-effective alternatives to TPB, bis-MSB and p-Terphenyl. Studies are underway to extrude light guides at Fermilab with co-extruded layers doped with the same concentrations of TPB, bis-MSB and p-Terphenyl. The purpose of these studies is to quantify the relative light output of these three waveshifters. To make these comparisons, a NIST-calibrated photodiode will be used to calibrate the VUV vacuum monochromator at Fermilab.

5.4.1.1 bis-MSB

The waveshifter bis-MSB is widely used to convert the UV emission from primary scintillants like pseudocumene or doped polystyrene to wavelengths that can be absorbed by WLS fibers (MINOS, NO ν A, MINER ν A). Bis-MSB is inexpensive relative to TPB. Figure 5-15 shows the typical emission spectrum of bis-MSB in the range 360-500 nm which is well-matched to the R8520-MOD PMT. In this figure, the absorption spectrum is also shown for the typical range of interest down to 280 nm.

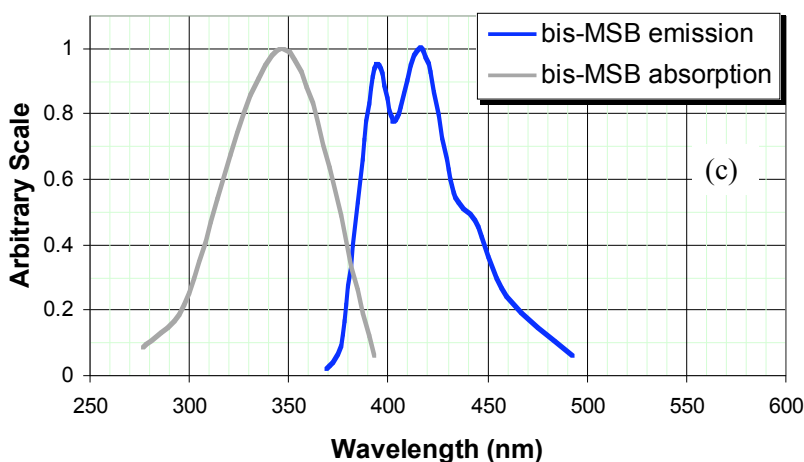


Figure 5-15: Absorption and emission spectra of bis-MSB. The absorption below 280 nm has not been found in the literature.

The group at Indiana University did an exploratory study on light guides extruded at Fermilab having a co-extruded layer doped with different concentrations of bis-MSB. Figure 5-16 shows the results of this preliminary study. The spectra in this figure have not been corrected for the spectrum of the deuterium lamp. The figure shows that there is clear absorption of 128-nm photons with re-emission at 420 nm when compared with an undoped bar. The current effort is aimed at comparing the 420-nm light output of bars doped with bis-MSB to bars doped with TPB when illuminated at 128 nm.

5.4.1.2 p-Terphenyl

Another cost-effective waveshifter that may prove to absorb at 128 nm and re-emit at 420 nm is p-Terphenyl. The absorption and emission spectra are shown in Figure 5-17. Bars doped with p-Terphenyl will be compared to bars doped with bis-MSB or TPB.

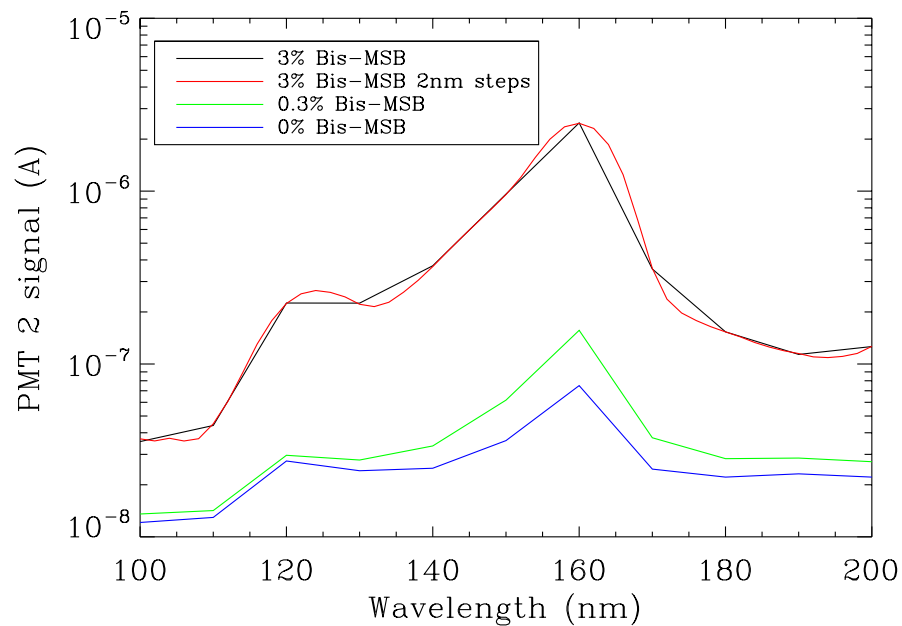


Figure 5-16: Relative emission at 420 nm for co-extruded bars doped with bis-MSB as a function of illuminating wavelength in the VUV monochromator at Fermilab. The co-extruded bars, doped with different concentrations of bis-MSB, are compared with an undoped bar. The spectra are not corrected for the deuterium-lamp spectrum.

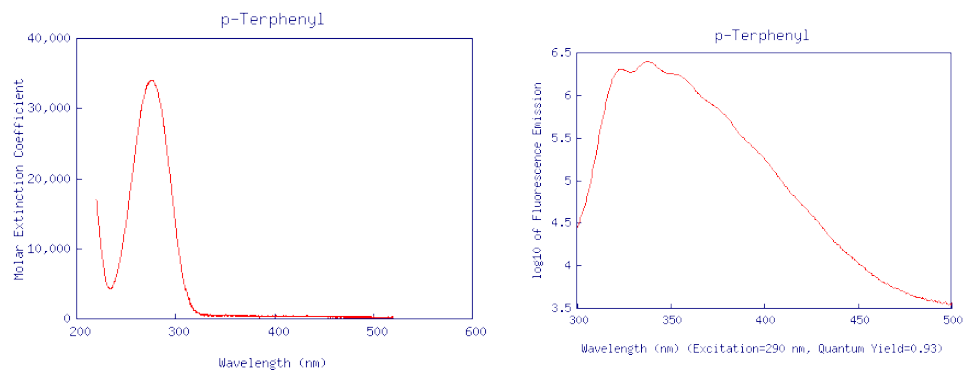


Figure 5-17: p-Terphenyl absorption (*left*) and emission (*right*) spectra. The absorption below 200 nm appears to be rising.

5.4.2 Cast-Acrylic Light Guides

As a test of the technology, several cast-acrylic bars were obtained from a commercial vendor to compare with the extruded polystyrene bars. Work at MIT suggests that cast acrylic may prove to be a superior method producing light guides. The first tests of the commercial cast-acrylic bars compared their attenuation lengths to those of extruded polystyrene bars. In addition, comparisons were done between the cast-acrylic bars and extruded-acrylic bars produced at the Fermilab extrusion line.

As seen in Figures 5-18 and 5-19, cast acrylic offers great promise because its attenuation length at 420 nm is significantly longer than that of extruded acrylic and extruded polystyrene. In addition, it might be possible to cast the light-guide paddle in one piece, which would significantly improve the efficiency of paddle construction. Clearly, cast-acrylic adiabatic light paddles show great potential.

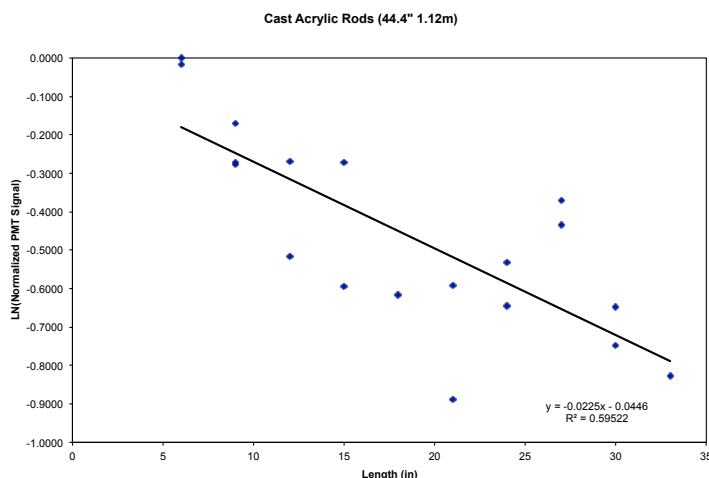


Figure 5-18: Attenuation lengths at 420 nm for cast acrylic, $l = 44.4''$

However, as also seen in Figure 5-18, fluctuations in the cast acrylic are much larger than in the extruded parts shown in Figure 5-19. The cast acrylic clearly shows potential but the production of acceptable light guides with this technology requires improvement in uniformity. In addition, techniques for coating the cast acrylic with waveshifter for the large number of required paddles would need to be developed.

5.4.3 Silicon Photomultipliers

Over the last decade a new type of photon detector has been developed, generically called Silicon Photomultipliers (SiPM), which may provide an attractive alternative to PMTs as detectors for the photon-detector system in the LArTPC.

An SiPM is a photon-counting multi-pixel device, consisting of silicon-based avalanche photodiodes (APD) operating in Geiger mode. The characteristics that make them attractive for this task are their high photon-detection efficiency (PDE), high intrinsic gain (10^5 to 10^7), lack of need for an HV system, increased performance in cryogenic environments, low cost per sensitive area and low peripheral costs. There are now many manufacturers of SiPMs (CPTA, SensL, MEPhi/Pulsar, Hamamatsu and AdvansID, among others).

Figure 5–20 shows the PDE for the Hamamatsu’s S11064, a 4×4 channel array, in which each channel is a 3×3 mm² SiPM. The PDE has been corrected for crosstalk and afterpulses, as measured by the manufacturer. This device is being studied at Fermilab.

In the particular case of the above device, the effective active area-per-channel is 9 mm² and for 16 (4×4) channels, the photosensitive area is 144 mm². The fill factor is 0.308, reducing the effective photon-collection area to 44.35 mm².

The Indiana University group is studying the Hamamatsu S10985-100C, a 2×2 channel array in which each channel is 3×3 mm². There are 900 pixels per channel with a pixel size of 100 μ m². There is an effective active area-per-channel of 9 mm², giving a total area of 33 mm² for the four channels. The fill factor is .785, reducing the effective collection area to 28.26 mm². At peak sensitivity of 440 nm, the PDE is 50%.

5.5 Preliminary Engineering Design

The preliminary engineering design studies will concentrate on production of prototype light-guide paddles with the goal of producing the 30 paddles needed for LAr1. These prototypes will be based on the reference design. The light guides will be tested in an apparatus at Indiana University that is based on the MIT design [27]. The paddle assemblies and their associated photodetectors will likely be tested at a more appropriate facility like Fermilab’s planned 35-ton membrane-cryostat prototype, discussed in Section 8.3.7, to be built well before LAr1.

As the prototypes are developed, three areas will be investigated as potential paths to improve the light guide/paddle performance and cost. As described above, these include (1) alternative production technologies for light guides; (2) alternative, more cost-effective waveshifters; and (3) SiPM photodetectors.

5.5.1 Light-Guide Testing at Indiana University

The test apparatus shown in Figure 5-21 was built at and funded by Indiana University. It consists of a jacketed LHe dewar that has been fitted with a frame to support light guides. The hold time for LAr in this dewar is significantly increased with LN₂ in the outer jacket. Since the purity of the LAr is vital in these test measurements (as described by the MIT group [27]) and high-purity LAr is quite expensive, the increased hold time is an important advantage to this setup. The 1- μ Ci Am²⁴¹ α source generates the scintillation photons in LAr and can be moved up and down a shaft to illuminate the light guide along its length. The electronics are fast enough (1.4 GB/s) to read out the waveform of the prompt 1.6 ns signal and the fast waveform digitizer has enough memory to see the late signal as well.

5.5.2 Alternative Waveshifters

A program is being started to compare the waveshifting properties of TPB, bis-MSB, and p-Terphenyl under analogous conditions. For the initial tests, polystyrene bars have been extruded with a co-extrusion layer that contains 3% waveshifter and 97% polystyrene. These bars will be tested in two ways. First, it is necessary to test their absolute light output with the McPherson vacuum VUV monochromator at Fermilab. These measurements will be at room temperature. A NIST-calibrated photodiode has already been purchased and the monochromator will be calibrated in the near future. This investigation will directly compare the absolute optical-light output from these bars when illuminated by UV light from 100 nm to 200 nm to determine the efficiency for UV-to-optical light conversion for the different waveshifters.

These bars will be tested in LAr at both IU and MIT in a test apparatus similar to that shown in Figure 5-21. These tests will measure the relative light output for the different waveshifters with the same concentration in the same type of co-extruded light guide. The monochromator measurements and the measurements in LAr should give enough information to make an optimum choice for the waveshifter.

The doping concentration will be optimized after the waveshifter has been selected. For these tests, different concentrations of waveshifter will be loaded into the co-extrusion then the light output will be measured with the VUV monochromator and in the LAr test apparatus. Since the costs of the waveshifters are known, these are studies that trade light for cost to produce light guides with adequate light for the funding available.

5.5.3 Cast-Acrylic Light Guides

As seen in Figure 5-19, cast acrylic has a significantly longer attenuation length than extruded polystyrene. In fact, the relatively short attenuation length for extruded polystyrene

may prove a barrier to this technology for a 2.3-m light guide. A second advantage of cast acrylic is that it offers the potential to cast a single, monolithic adiabatic paddle, instead of four light guides that must be assembled into a paddle. In this case, a thin, flat paddle would be drawn into the square shape of the PMT. However, several issues associated with cast-acrylic light guides require investigation.

The first issue is whether light paddles can be cast into the required shape that adiabatically tapers a flat paddle into the square PMT profile. A second issue is whether this method would be cost-effective. If silicon photomultipliers (SiPMs) prove to be a realistic alternative to PMTs in their light-gathering power, however, a complicated mold may not even be necessary since the SiPMs could be manufactured to populate the edge of a flat panel and no bending would be necessary. It must be demonstrated that cast-acrylic paddles or light guides be made with equal uniformity to those from commercial off-the-shelf cast-acrylic stock.

A final issue is the uniform coating of light paddles with waveshifter. Hand-coating the 2,160 light guides by dissolving the waveshifter in polystyrene + toluene and air-brushing the mixture onto the cast acrylic paddles (as is currently done at MIT for MicroBooNE) would be very labor-intensive and would likely introduce significant variations in the coatings. Two paths for coating the bars will be pursued. (1) Casting the light guides with the waveshifter in the mold. It is unknown whether this can be achieved. (2) The development of a machine to coat the bars. There is experience from MINOS and NO ν A in developing machines to deposit the uniform coatings of glue for assembly of these experiments, which seems to be a very similar problem. It is clearly important to learn how these coatings would perform in LAr.

5.5.4 SiPMs

At Fermilab, the development plan includes the continuing work on the absolute calibration of SiPMs. In addition, the plan includes tests directed at detecting VUV light in LAr using SiPMs coated with TPB [31]. This work has already begun.

The Indiana University group has been awarded a research grant to purchase a small number of Hamamatsu SiPMs. These will be substituted for the R8520-MOD PMTs on the extruded light guides and comparison tests of them in LAr will be made.

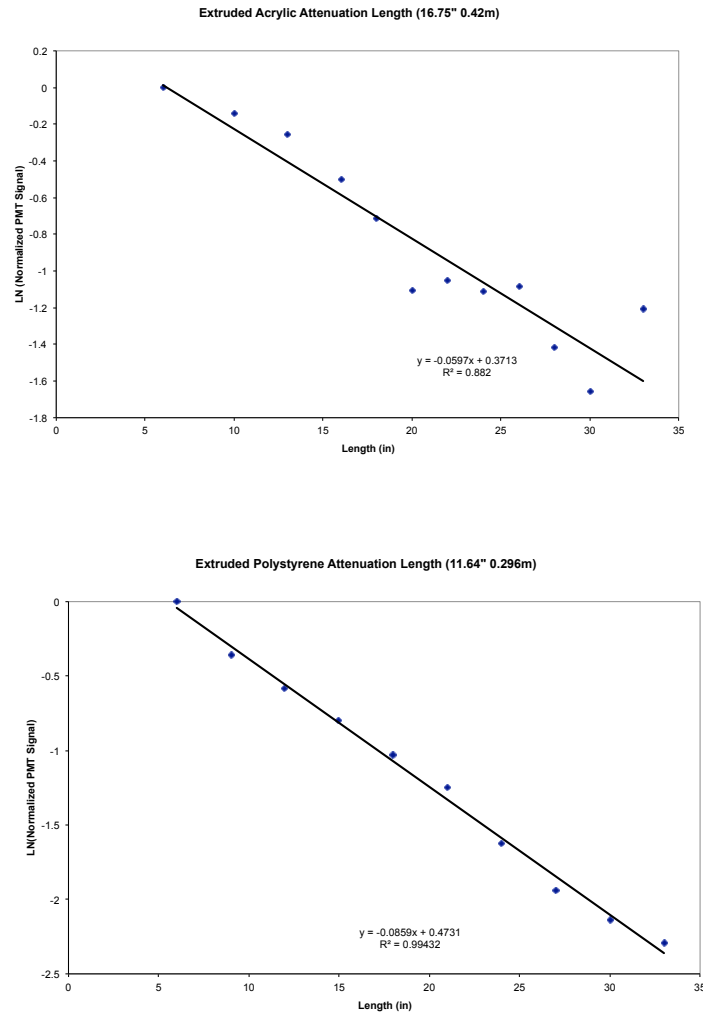


Figure 5-19: Attenuation lengths at 420 nm. (*top*): extruded acrylic, $l = 16.8''$; and (*bottom*): extruded polystyrene, $l = 11.6''$.

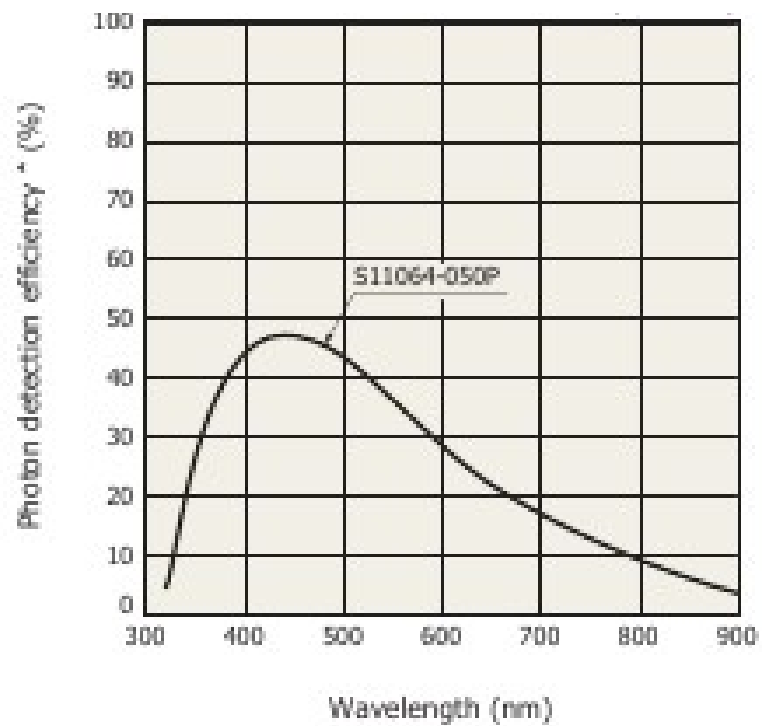


Figure 5-20: PDE for the Hamamatsu S11064 SiPM.

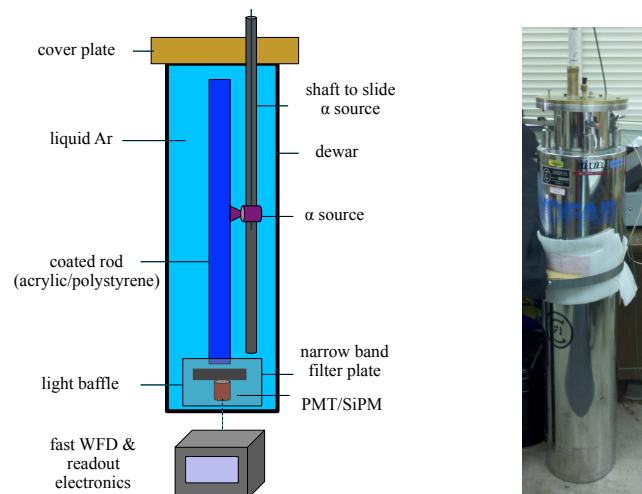


Figure 5-21: Apparatus for testing light guides in LAr. *left:* schematic of light guide testing apparatus; *right:* light guide testing apparatus.

6 Installation and Commissioning (WBS 130.05.06)

6.1 Introduction

This chapter discusses the LAr-FD Installation and Commissioning system's activities and responsibilities. LAr-FD construction and installation will occur in a series of distinct phases:

- installation planning and prototyping
- surface storage identification and operation
- excavation and outfitting of the cavern; this activity is the responsibility of the Conventional Facilities subproject (CF), WBS 130.06
- construction and installation of the cryogenics system and cryostats by a construction management firm; this activity is the responsibility of the cryogenics system (WBS 130.05.02)
- construction of LAr-FD components at collaborating institutions and shipment to the Far Site
- installation of detector components and installation management
- commissioning activities leading to CD-4

The Installation and Commissioning system will accept responsibility for the LAr-FD cavern, associated tunnels, infrastructure and above-ground facilities from Conventional Facilities upon completion of the Conventional Facilities contract. The cavern will be outfitted with the following utilities upon receipt of beneficial occupancy:

- ventilation in accordance with OSHA standards
- electrical power sufficient for the HVAC, cryogenics plant cooling and general 110-V service

- quiet power for the electronics with a double Faraday-shielded transformer located some distance from the cavern to take advantage of the inductance of the power lines. The primary shield will be connected to the main substation via a grounded feed wire and the secondary shield will be connected to the Ufer ground.
- communications consisting of telephone lines and computer network
- cavern lighting in accordance with OSHA regulations for industrial use
- tunnel lighting with battery-powered backup or emergency circuit backup
- environmental monitoring of oxygen, carbon monoxide, smoke and temperature
- dual isolation bulkheads separating the cavern from the existing Far-Site facility
- sump pumps for groundwater removal
- a 20-ton bridge crane in the cavern

After beneficial occupancy of the completed cavern from the CF subproject, the cavern ventilation system will be tested to assure adequate performance with regard to ODH requirements. The system will be tested with oxygen monitors distributed around the cavern and a controlled argon spill. Remedial action will be taken if required during cryostat and cryogenics construction.

The cryostat and cryogenics contractor will retain responsibility for the site during construction of the cryostat and cryogenics system. The Cryostat and Cryogenics System group will provide oversight during this phase. Upon completion of this contract, the facility will be in the following state:

- the LN₂ refrigeration system will be constructed and commissioned with liquid nitrogen
- the LAr systems and cryostats will be constructed and tested without the introduction of cryogens
- the access hatches on the cryostat and the cryostat feedthroughs will be temporarily sealed
- the APA- and CPA-installation support beams will be in place
- the cryostat will be connected to the steel roof structure thus connecting the detector ground to the Ufer ground

The detector will utilize the cryostat pit rock bolts as part of the grounding scheme. The rock bolts in the pit will extend through the shotcrete that lines the cavern and be attached to the reinforcing steel network within the cryostat concrete liner, forming an Ufer ground. The reinforcing steel will be connected to the steel truss cover. The Ufer ground will be connected to the detector ground (the cryostat SS liner) through a low-impedance connection.

The Installation and Commissioning group will be responsible for all LAr-FD-related activities at the Far Site from this point in time until the end of the LAr-FD project. Close coordination is clearly required between this group, system groups that provide components and other Far Site construction activities.

On-project commissioning activities include the coordination of system-checkout activities, culminating in the approval to introduce LAr into the detector modules, and managing the steps required to meet the CD-4 goals.

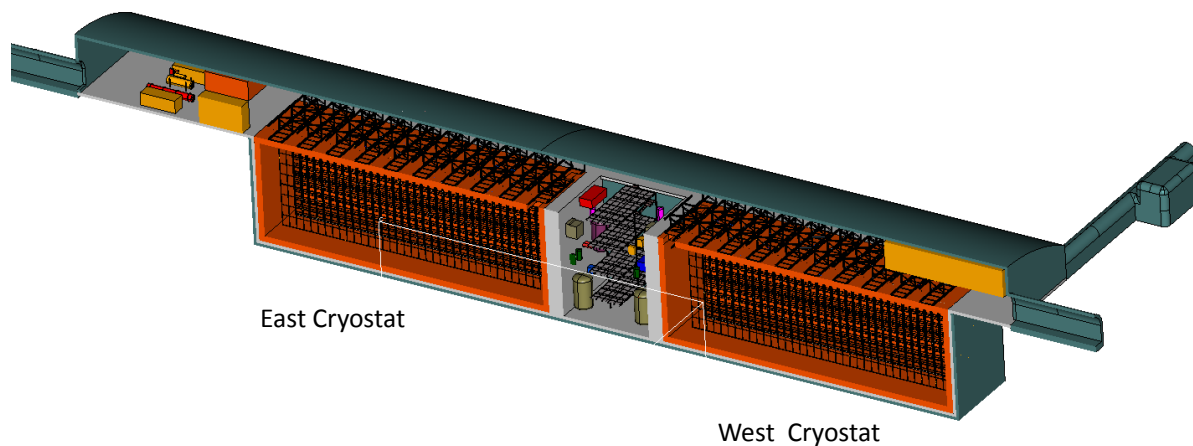


Figure 6-1: Cutaway view of two cryostats with TPC detectors installed

6.2 Above-ground Pre-Installation

Detector components will be delivered to the Far Site over a period of many months and will need to be stored in a surface-storage facility. This will allow the supply of material to be maintained ready for installation. A facility for this purpose will be identified and an agreement will be made for its use during LAr-FD construction.

The initial surface-storage plan actually includes three facilities: a 4,000 ft² warehouse structure for material receiving and unpacking, an on-site hardstand for storage of a small number of cargo containers and an off-site hardstand area for the storage of a larger number of cargo containers. Material will be transferred from these areas to the cavern for installation, as required. Each system group is responsible for delivery of its components to the local storage at the Far Site or to the off-site location. The Installation and Commissioning group will provide the management and labor resources for inventory control, material handling and transport from the off-site and on-site storage facilities to the cavern.

6.2.1 Cryostat Materials

The on-site warehouse storage space will initially be made available to the cryogenics system contractor who will construct the cryostats. The cryostat insulation will comprise the largest bulk of material; approximately 5,500 m³. Figure 6-2 shows membrane-cryostat components staged in the hull of an LNG tanker under construction. Cryostat materials will come from overseas in approximately 100 shipping containers that will be stored off-site. The containers will be brought to the shaft headframe building for unloading and transportation down the shaft.

Control of the storage area will revert to the Installation and Commissioning group when the cryostat-construction contract is completed.

6.2.2 TPC Materials

APAs and CPAs, referred to as “TPC panels” in this chapter, will be constructed before arrival at the Far Site. The electronics will already be installed and the cold-testing performed. They will be shipped in sealed shipping containers to either the on- or off-site hardstand area, depending on the TPC production rate. No significant preparation or extensive testing of these components is required after arrival and prior to installation. The entire set of TPC panels for the two-cryostat detector will require approximately fifty shipping containers. Figure 6-3 shows a group of TPC panels in a shipping container.

Whereas general material will be lowered in a lift cage, the TPC panels are too long to



Figure 6-2: Membrane cryostat components staged in a LNG transport ship under construction.

fit in the lift cage and require special containers and special handling. The TPC panels, grouped in their enclosed, special containers, will be lowered down the Yates shaft since its lift has provisions to attach long objects to the bottom of the cage. They will descend the shaft in a vertical orientation (the same orientation as they will be installed), be rotated to a horizontal orientation, then moved along the access drift on a cart or rail to the cavern. Figure 6-4 shows a TPC container exiting at the bottom of the shaft. The TPC panels may be transported to the Far Site in these special containers or transferred to them after arrival.

TPC-panel containers will be transported to the cavern and unloaded every few days to supply the TPC components for installation. These containers may need to be moved during off-work hours to avoid conflicts with other users of the shaft lifts. TPC-panel containers may be stored in the cavern on the deck on top of the cryostat trusses. Enough parts will be stored in the cavern to ensure that a sufficient supply of TPC parts is always on hand for installation.

6.2.3 Liquid Argon Receipt

Delivery of LAr will occur over a four-to-six month period for each cryostat, with approximately eight tank trucks arriving on site per day. Each tank truck will have been loaded with 18.8 tons of LAr which will be purity-tested by the vendor before the truck departs. Minor losses of LAr will occur during shipment. The purity will be checked again during unloading. If the impurities are found to exceed the required specification, the partially emptied tank truck will be returned to the vendor.

LAr will be transferred to a buffer-storage vessel that allows the tank truck to unload as quickly as possible. Under normal circumstances transfer of the LAr out of each tank truck will require 1 hour, including the time for making hose connections and purging the hoses. The storage vessel will deliver argon to the filtering loop that processes argon from the cryostat recirculation and condensing loops.

6.3 Below-ground Pre-Installation (WBS 130.05.06.06)

The below-ground pre-installation activities must be completed prior to the start of TPC installation into the cryostat. These activities include design, procurement and installation of detector-specific infrastructure such as man-lifts, lifting fixtures, catwalks, ladders, tools, and so on. The major items include the support rails for the TPC panels, the lower staging platform for joining two panels, installation monorails for moving the panels inside and outside of the cryostat, and a mobile scaffold for providing access to the top and middle of stacked TPC panels.

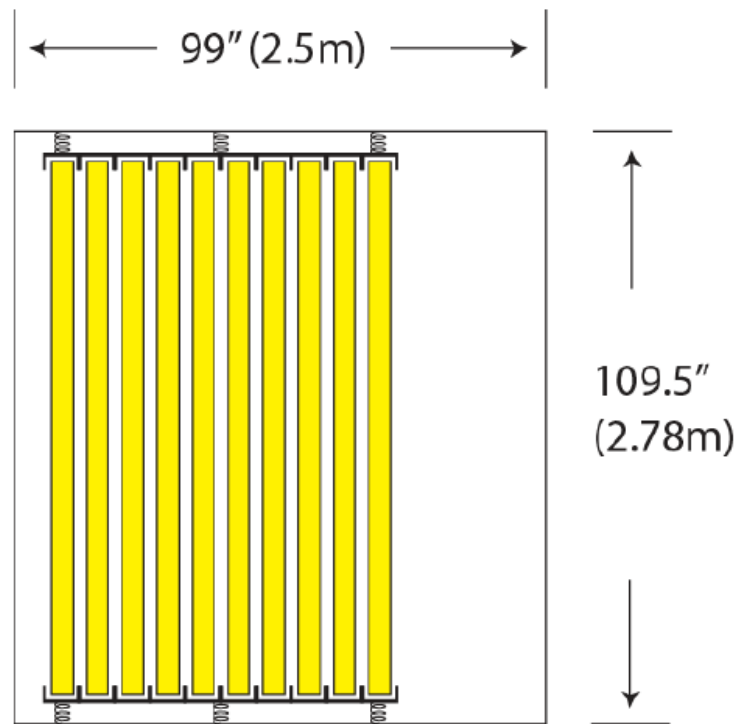


Figure 6-3: Concept for APA shipping containers - cross section view

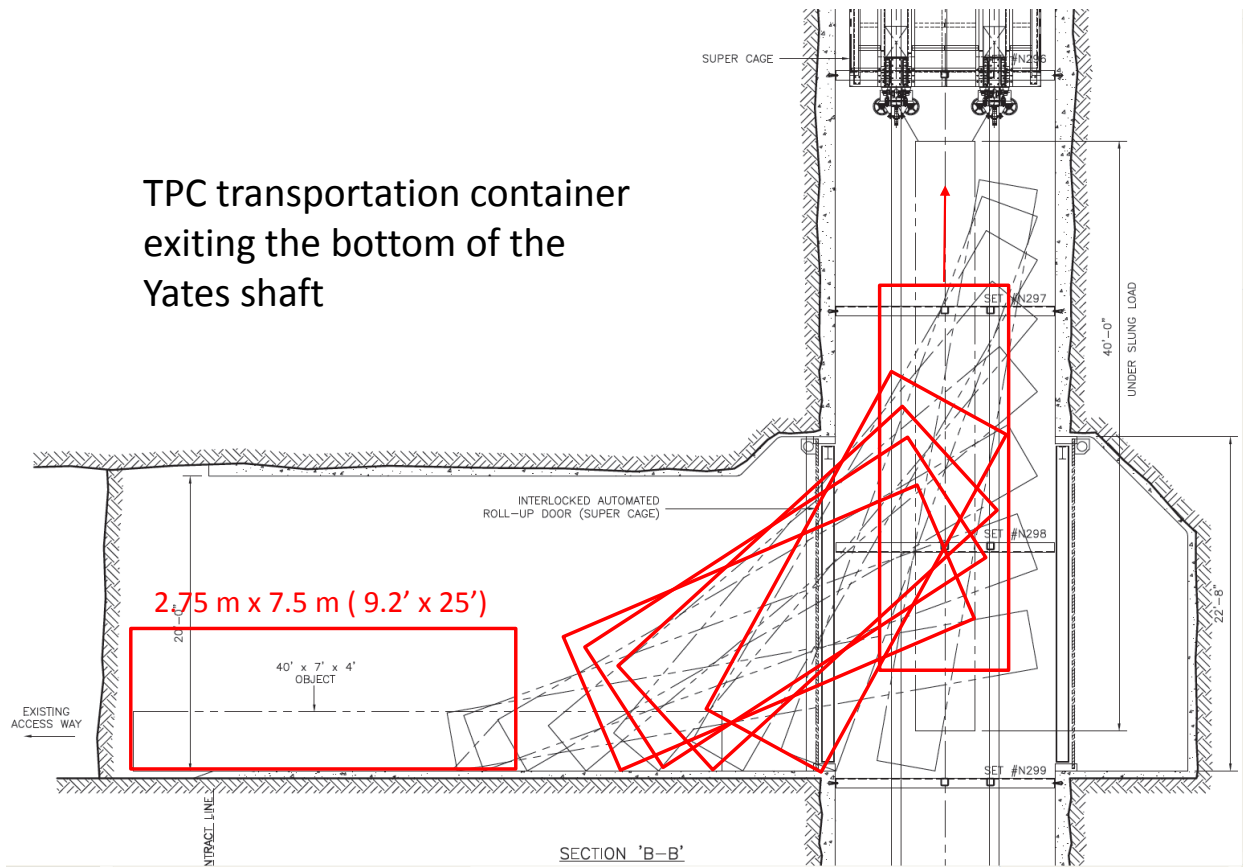


Figure 6-4: TPC-panel containers exiting the Yates shaft

6.3.1 Equipment Required for Detector Installation

Several items will be installed for use during TPC installation, and removed afterwards.

- A temporary lighting system inside the cryostat with emergency backup lighting for TPC installation
- A ventilation system and air-monitoring sensors with alarms connected to the cryostat to assure adequate air quality for the personnel working inside the cryostat. The system will also include a high-sensitivity smoke-detection system that is interlocked to the power for all devices inside the cryostat.
- A raised-panel floor to protect the cryostat floor; see Figure 6-5. The raised-panel floor will have support spacers located between the convolutions in the stainless-steel primary membrane to provide a flat surface for moving equipment around within the cryostat. The pressure limit for the insulation in the floor is 0.5 MPa and the load of the vacuum that will be used to monitor leakage during installation reduces the effective limit to 0.4 MPa. The stock round spacers in the raised-panel floor are 10-cm diameter and would support a load of 310 kg. The load can be increased by adding larger-diameter plates under the standard spacers.
- A transfer rail with motorized trolleys to move TPC panels within the cryostat

6.3.2 Clean Area

A clean-area vestibule in the range of class 10,000 (ISO 7 equivalent) will be constructed in the septum area around the entrance to the cryostat to keep the area around the open hatch isolated from the drift access. The vestibule will have an area for personnel to gown with the appropriate clean-room clothing and safety shoes. A large, closable door, next to which the TPC-storage containers can be parked, will allow unloading of the TPC components directly from the container into the vestibule.

A crane that operates inside the clean-room vestibule enclosure will be used to transfer TPC panels from the vestibule into the cryostat. The crane will have a clean-room-style hoist and trolley to prevent contamination of the cryostat or vestibule area.

The Double Chooz detector developed a cleanliness plan to ensure that dust contamination did not contribute more than a specified amount to the detector signal. Measurements were made of the activity of rock in the underground laboratory which was assumed to be the source of airborne dust. Maximum allowable dust concentrations and the clean-room class and cleanliness practices were determined such as to meet the requirements for contamination. The cavern for LAr-FD and many of the access tunnels will involve new excavation

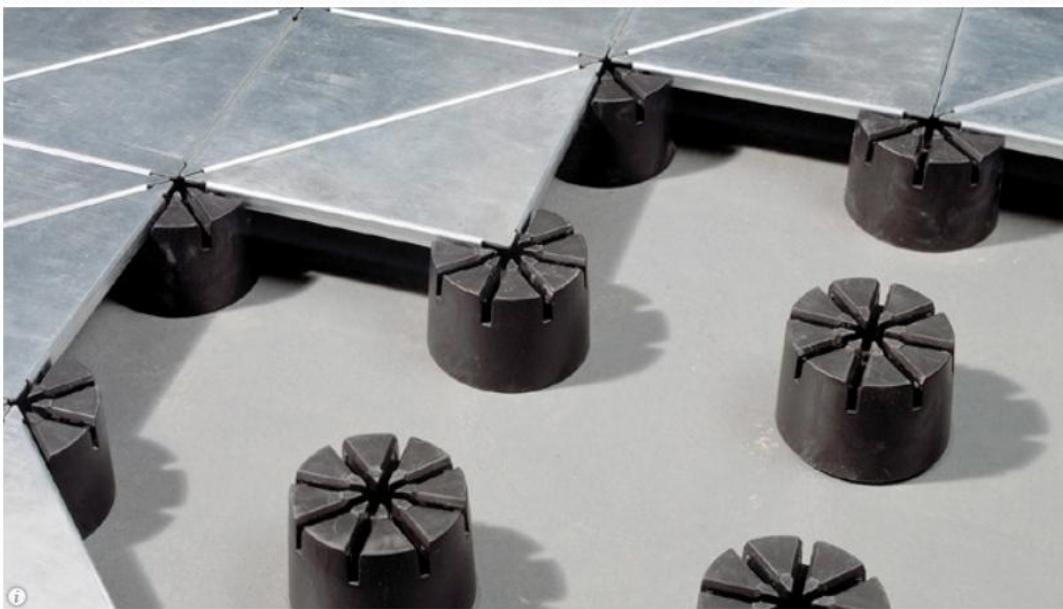


Figure 6-5: Raised-panel floor to protect the cryostat's primary membrane

with shotcrete covering the native rock. The Installation and Commissioning group will need to evaluate the dust sources in the LAr-FD cavern and determine if a similar approach to setting the clean room requirements is appropriate.

6.3.3 Rails for TPC-Panel Support and Transfer

A set of seven support rails, shown in Figures 6-11 and 6-6, permanently mounted in each cryostat, will provide the support for the APA and CPA panels and a track for moving the panels into position. Rods spaced at 5-m intervals will support the rails. The rods will be hung from anchor points mounted in the top of the cryostat. The rods will be adjustable to enable level installation of the rails, which will be done in rail segments with the aid of a laser level. The rails shrink approximately 13 cm along their entire length during cooldown. The rods will be installed with an angle bias that allows the rails to return to level after the cryostat and TPC is cooled.

The mass of each stacked set of APA panels is 600 kg and the mass of each stacked set of CPA panels is 250 kg. The load of the TPC on the support rails comes to 200 kg/m for the APA rails and 100 kg/m for the CPA rails. The rail segments will be constructed from 20-cm-deep laser-welded, W-shaped, stainless-steel beams. The rail segments will be joined end-to-end with large pin connections. The upper support rods will be made from 15-mm-diameter stainless steel.

The rail installation can be completed most efficiently while the large scaffolding system used for cryostat construction is still in place, therefore it will be part of the cryostat-construction contract.

Signal and power cables will be installed from the cryostat feedthrough ports next to the APA support rods, shown in Figure 6-7, along the rails to the point where the connection to the APAs will be made. The cables will be preplaced and tested before APA installation begins.

The top and bottom TPC panels of a stacked pair will be moved into the cryostat separately, then connected together below the cryostat equipment hatch. The lower panel is held temporarily with a staging platform while the upper panel is connected. After connection, a motorized trolley will move the stacked panels from the hatch area to the final position. Since the duty cycle of the trolley is rather low, the trolley could be battery-powered to avoid the need for cable festooning. The trolley initially moves along a transfer rail until it reaches the end where the stacked panel will be permanently mounted. An arrangement of transfer switches is used to connect the transfer rail to the final support rail. See Figure 6-8 for an example of a transfer rail.

A rolling scaffold with an integral stair tower will allow personnel to access the top of the

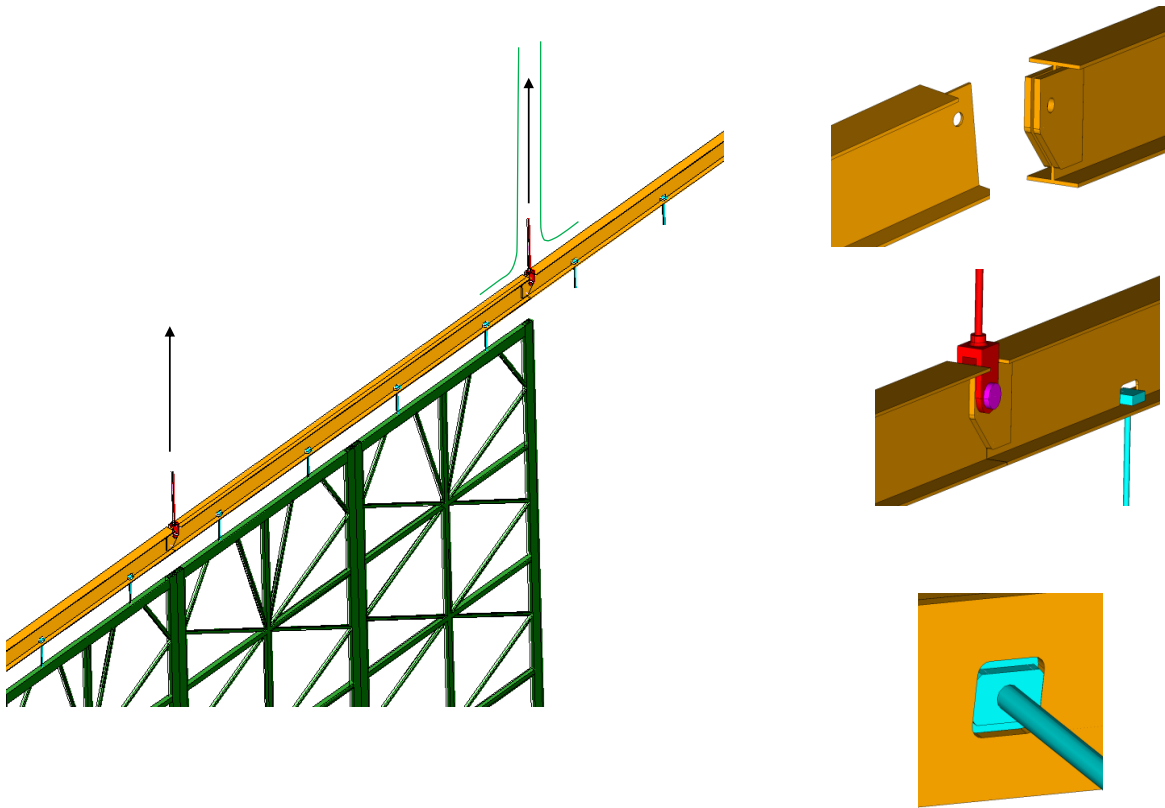


Figure 6-6: Support rails inside cryostat

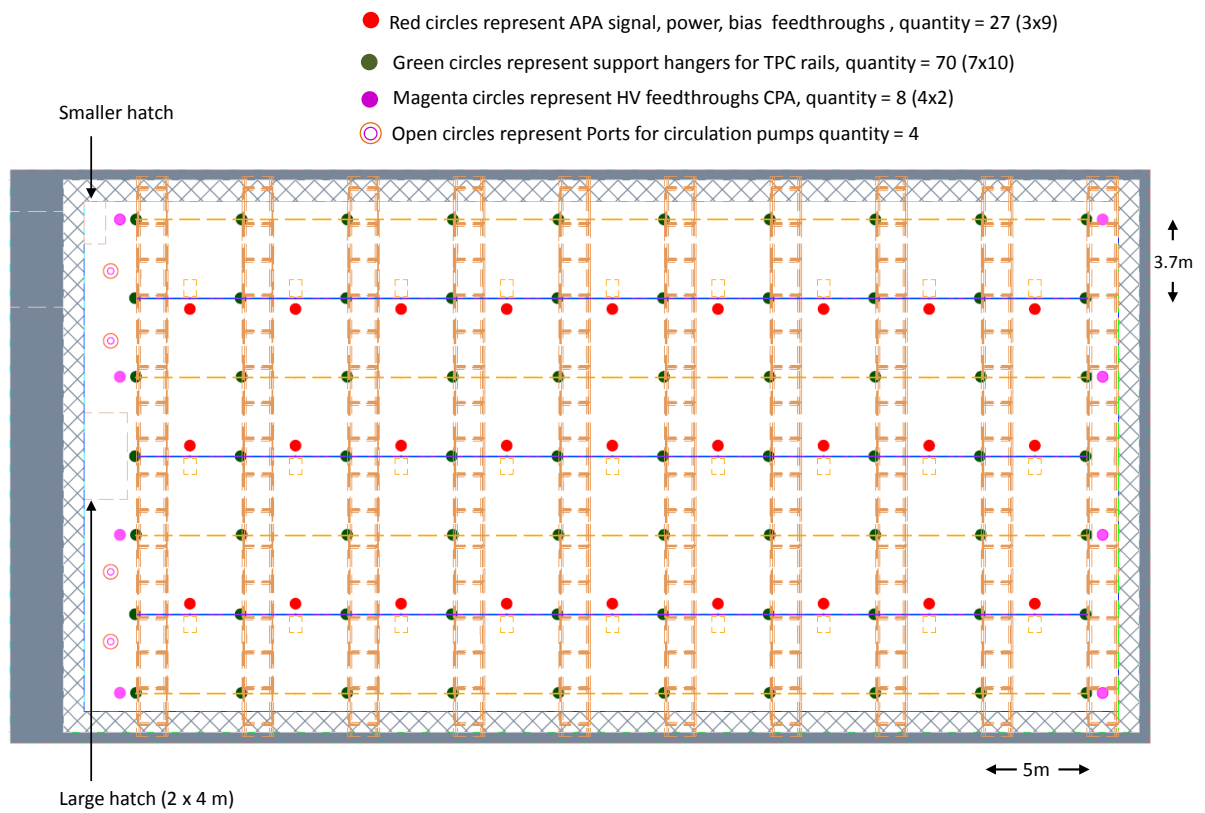


Figure 6-7: Feedthroughs

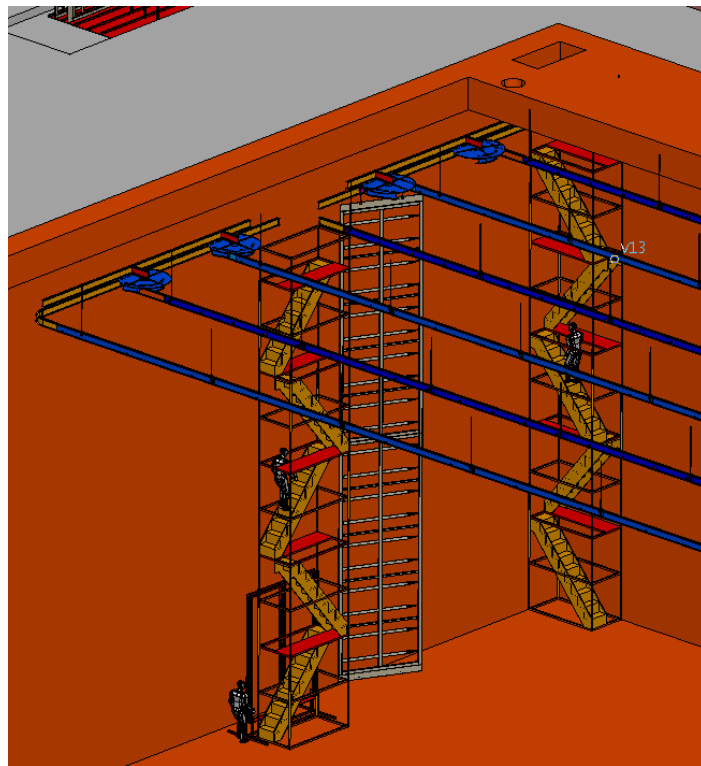


Figure 6–8: TPC installation monorail with APA moving to support rail

stacked panel. Fixed scaffolds with stair towers will provide access into the cryostat.

6.3.4 Detector Electrical Ground

The LAr-FD will have nearly a quarter million channels of electronics with an intrinsic noise level of less than 1,000 electrons. These channels will be connected to wires that are seven meters long. Thus, grounding, shielding and power distribution are critical to the success of the experiment. In the reference design the entire cavern will be treated as the detector ground for the following reasons.

The cryostat has a large number of penetrations for supporting the APAs that extend to the support structure above the cryostat. There are also 27 signal feedthrough ports which connect to racks located on the top of the cryostat. Achieving and maintaining adequate AC isolation on such a large structure during construction will be difficult. In addition, the cavern has very few connections to the outside world so it is much easier to isolate the cavern as a whole than the individual pieces.

Secondly, it is necessary to provide a conductive body with a large enough self-capacitance that its voltage changes only negligibly when electric charge flows onto it. In this way the cavern can serve as a sink of unwanted current without generating noise in the detector.

6.3.4.1 Reference Design

Earth ground will be provided by the Ufer grounding system in the concrete walls and other concrete support structures. This ground will be attached to the rock bolts in the cavern walls and augmented by the large amount of steel in the roof-support trusses and the upper metal floor. To be an effective ground, all of the steel support structure will be welded together. The welds need not be structural; their use is only to assure reliable long-term electrical connections.

In order for this system to work, the cavern must be kept noise-free, i.e., all connections, except AC power, between the cavern and the outside world must be electrically isolated either by dielectric breaks or optical isolators. Electric motors will be restricted to three-phase induction motors except for special, well-controlled cases. Electric heaters will be controlled by switches rather than SCRs. Digital equipment such as network switches will be run at frequencies of 30 MHz or higher whenever possible so that the noise remains outside the bandwidth of the TPC preamplifier. Some equipment, such as switching power supplies, will still need special attention to ensure that they do not generate noise in the detector.

As the only conducting link to the outside world, the AC power must be filtered to eliminate any electrical noise. Since the currents are large, it is most economical to implement a filter

using the inductance of the power cable itself along with some capacitors to form a capacitor-inductor filter.

The reference design will keep the power for the conventional facilities and for each cryostat independent of one another so that they can be operated independently. As shown in Figure 6-9 (top), one 500-KVA transformer for each cryostat along with a one-MVA transformer for the conventional facilities will be placed in a separate utility room located a short distance from the detector cavern. These transformers will have double faraday shielding with the primary shield returned on the ground wire to the substation. The second shield will be connected to the local Ufer ground and to the cavern by the required grounding wire.

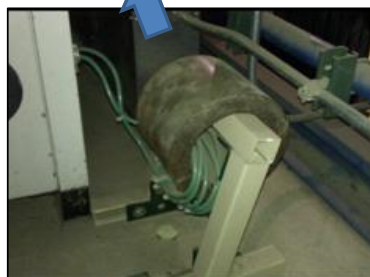
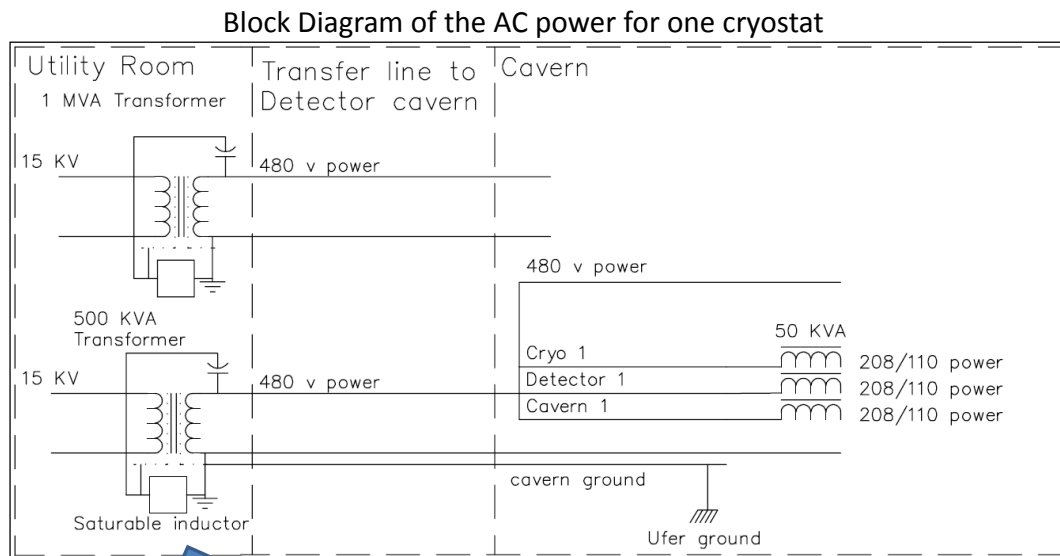
A saturable inductor between the two shields serves to isolate the primary and secondary shields and separate their grounds. The secondary shield and transformer frame are connected to Earth ground at the transformer. The ground wire running back to the substation is only locally grounded through the saturable inductor but it is fully grounded at the substation. A fault between the secondary and primary shields would trip the substation breaker due to current flowing in the return ground wire. If this wire were disconnected, the current would flow through the saturable inductor to Earth ground and also trip the substation breaker. Figure 6-9 (bottom) shows an inductor used at D-Zero and the voltage developed across the inductor as a function of the fault current. The inductance is completely saturated by 100 A and the developed voltage is still safe.

The 480-V power will be transmitted from the utility room to a separate switch gear for each detector module. The switch gear, located adjacent to its module, will directly distribute power to all the 480-V loads. The switch gear will also feed up to three 50-KVA single-faraday-shielded transformers. Two of these will provide 208-V power to the module and its associated cryogenic equipment, and the third will provide additional cavern power. The number and size of the transformers may change depending on the final load.

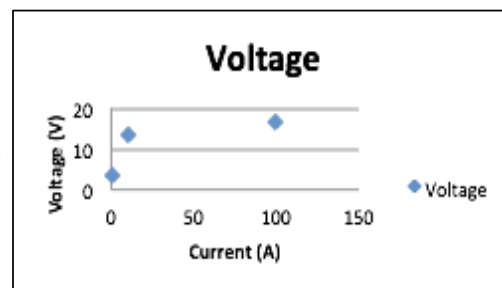
The filter on the conventional-facilities transformer will be similar to the one for the detector modules. In addition to the conventional facilities, this transformer will power all welding outlets. This arrangement would allow some minor welding on one cryostat, if needed, while the other is taking data.

6.3.4.2 Features

It is important that all the components inside each detector module be connected to a common ground. The best candidate for this common ground, and the one chosen for the reference design, is the top of the cryostat. The APAs will have ground braid connections to the roof at two points on each panel. The lower APAs will be connected to the upper ones by their mechanical mounting connections. The front-end boards, phototube ground and the reference ground for the bias voltages will all be connected to the APA frame.



Saturable inductor used for ground isolation



Voltage developed across the inductor as a function of the fault current

Figure 6-9: Block diagram of the AC power for one cryostat with a saturable inductor to separate the grounds of the transformer shields

The top of each cryostat is made of 1.2-mm or 2.0-mm-thick stainless steel, a poor conductor, and therefore does not serve as an adequate ground plane. The best solution would be to add a copper sheet directly on top of the cryostat, however this is difficult to do. Instead the copper sheet will be installed over the insulation that is on top of the cryostat membrane, about one meter from the cryostat, and connected to it via a grid of connections spaced 2.5 m apart. This spacing will give good electrical performance up to 12 MHz – well above the input bandwidth of the amplifier. The connections will be made with copper strips that fit in gaps in the insulation. Power supplies and power-supply filters (including the cathode supply) will all be grounded to this plane.

A port is located above every other APA junction and each port will serve the four APAs located directly underneath it. Cables for the bottom two APAs will be routed through the hollow cable frames to provide both mechanical support and electrical shielding. The digital cables will be located in the left frame member, and the power and bias lines will be routed through the right frame member. The horizontal portion of the cable run will be outside of the wire planes, so the use of doubly shielded cable such as Amphenol skew clear may be adequate. If not, then custom metal shields will be mounted on the front-end circuit boards.

The racks will be mounted either directly over the ports or adjacent to them with an extension to the rack that covers the port. All cables can be brought out of the cryostat into a grounded and shielded enclosure. All cables to the APAs will be the same electrical length to ensure uniformity of signal travel time. This will require storing an extra 14 meters of cable for each top APA. The plan is to use 36-in-deep racks and construct a shielded area on the back side of the rack to hold the excess cable. Twenty-seven racks will be required for each cryostat, and rack space will be shared between the TPC and photon-detection system readout and power supplies. A modest number of racks will be required for the DAQ in the surface control room. All relay racks will be equipped with rack protection and monitoring. The racks will be supplied by the detector installation effort.

The digital electronics does not present a grounding issue in the strict sense, but it does affect noise. The plan is to operate the digital system with at least a 32-MHz clock that is well above the upper bandwidth of the input amplifier (the 3-db point is less than 1 MHz).

6.4 Below-ground Installation Activities

The following list of detector components and systems will be installed below-ground:

- Relay Racks with rack protection (27 on each cryostat plus a few in the control room)
- Cable trays and power distribution to racks
- Cable inside and outside of the cryostat including cable feedthroughs

- DAQ crates and power supplies in the relay racks
- The 108 APAs per cryostat with integrated photon detectors
- The 144 CPAs per cryostat
- Any cryogenic instrumentation not installed by the cryostat construction vendor (e.g. purity monitors)
- Note: Support rails and hangers are installed during cryostat construction

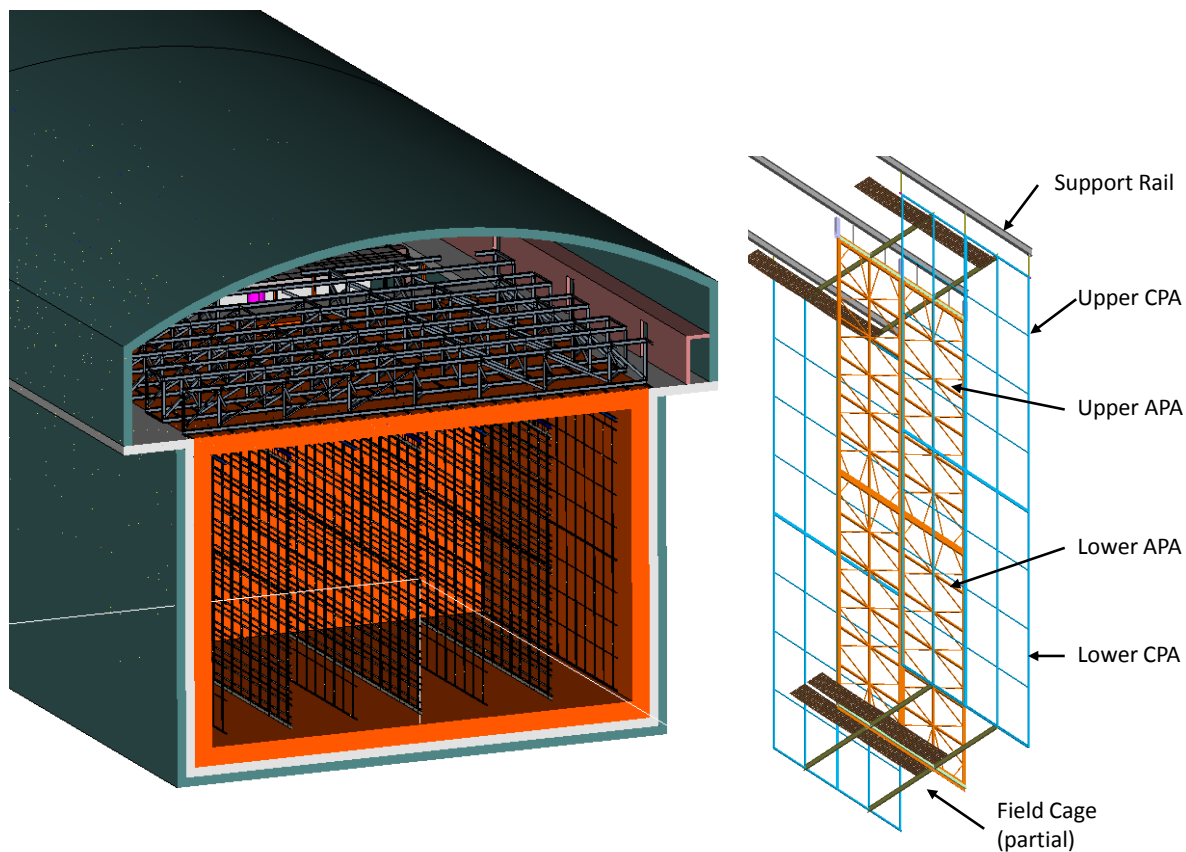


Figure 6-10: TPC panels installed in cryostat

6.5 TPC Installation

Each APA and CPA panel will be carefully tested after transport into the clean area at the septum and before installation into one of the cryostats. Immediately after a panel is installed

it will be rechecked. Throughout the installation period it will be checked periodically. The serial stacking of the APA and CPA panels along the rails means that removing and replacing one of the early panels in the row after several others are installed would be very costly in effort and time. Therefore, to minimize the risk of damage, as much work around already-installed panels as possible will be completed before proceeding with further panels.

The installation sequence is planned to proceed as follows:

1. Install the monorail or crane in the staging area outside the cryostat, near the equipment hatch.
2. Install the relay racks on the top of the cryostat and load with the DAQ and power-supply crates.
3. Dress cables from the DAQ on the top of the cryostat to remote racks.
4. Construct the clean-room vestibule outside the cryostat hatch.
5. Install the raised-panel floor inside the cryostat.
6. Insert and assemble the stair tower and mobile scaffold.
7. Install the transfer rail with switches and the staging platform inside the cryostat
8. Install protection on (or remove) existing cryogenics instrumentation in the cryostat.
9. Install the cryostat feedthroughs and dress cables inside the cryostat along the support beams.
10. Begin regular transport of TPC panels in shipping boxes into the cavern.
11. Install TPC panels:
 - (a) Install a connected APA-CPA panel pair.
 - (b) Connect power and signal cables.
 - (c) Test each APA wire for expected electronics noise. Spot-check electronics noise while cryogenics equipment is operating.
 - (d) Connect field cage in sections as the APA and CPA installation progresses.
 - (e) Perform electrical test on CPAs and field cage.
 - (f) Remove temporary floor sections as the TPC installation progresses.
 - (g) Install sections of argon-distribution piping as the TPC installation progresses.
12. Complete the field cage.
13. Remove the transfer rail, staging platform, moving platform and stair tower.

14. Temporarily seal the cryostat and test all channels for expected electronics noise.
15. Seal the access hatch.
16. Perform final test of all channels for expected electronics noise.

In general, APA and CPA panels will be installed in order starting with the panel furthest from the hatch side of the cryostat and progressing back towards the hatch. The field cage will be installed in stages as the installation of APA and CPA panels progresses. The only requirement for survey or alignment is to maintain the edges of a row of APA panels to 3-mm alignment along each beam. A laser guide or optical transit in combination with the adjustment features of the tie rods will be used to establish the alignment. After the stacked panel is attached to the support rods the electrical connections will be made to cables that were already dressed to the support beams and electrical testing will begin. Periodic electrical testing will continue to assure that nothing gets damaged during the additional work around the installed panel.

The TPC installation will be performed in three stages, each in a separate location; the locations, or zones, are shown in Figure 6-11. First, in the clean room vestibule, a crew will move the APA and CPA panels from storage racks, rotate to the vertical position and move them into the cryostat. Secondly, in the panel-staging area immediately below the equipment hatch of the cryostat, a second crew will transfer the lower panels from the crane to the staging platform, connect the upper and lower panels together, route cables to the top of stacked panels and finally transfer the stacked panels on to the monorail trolley that moves within the cryostat. A third crew will reposition the movable scaffolding and use the scaffold to make the mechanical and electrical connections at the top for each APA and CPA as they are moved into position. The monorails inside and outside the cryostat will each have two motorized trolleys so that work can be conducted by all three crews in parallel. The steady-state rate for installation, given this work plan and a single-shift schedule, is estimated to be two stacked panels per day.

Wire integrity will be confirmed by measuring the Equivalent Noise Charge (ENC) of each electronics channel and comparing it with the expected noise for a properly connected wire. A properly connected wire provides an input capacitance of 240 pF to the front-end amplifier resulting in an ENC at room temperature of 1,100 electrons. The wire-integrity test also ensures that coherent noise sources, e.g., a mechanical connection between the detector ground and the Ufer ground, are discovered promptly. Error budgeting, regular noise monitoring and mitigation will ensure that the TPC reaches and maintains the required noise performance before the cryostat is cooled down.

The detector installation system is also responsible for developing and implementing the procedure for monitoring the integrity of the membrane-cryostat primary liner during installation. The space between the primary liner and the secondary liner will be held under

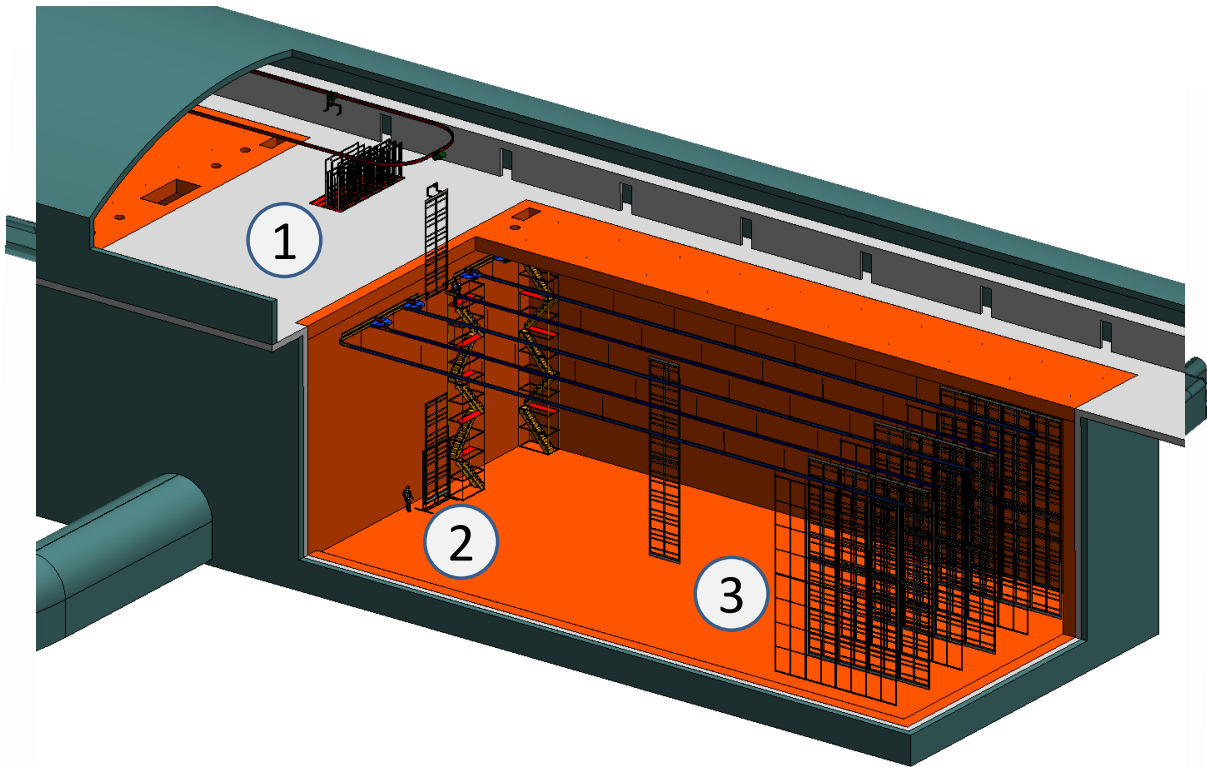


Figure 6–11: The three main work zones for TPC Installation. TPC components are lowered into the cryostat in zone 1. TPC components are connected together in zone 2, transferred to the support rails and then rolled into final position (zone 3).

vacuum during installation. The vacuum level will be automatically monitored and will alarm if any leaks develop in the primary membrane during TPC installation.

6.6 Installation Prototype

The goals of the installation prototype are to test and verify the key elements of the equipment and process for TPC installation and to serve as a training tool for personnel who will perform the TPC installation. The initial testing of the equipment at Fermilab will be used to verify or refine the installation concepts. Complete testing of the final equipment and operations will occur at Fermilab before the installation equipment is moved to the Far Site.

For the installation prototype, a similar, 70%-height detector will be constructed in a suitable location at Fermilab, e.g. the Wideband lab or the CDF or DZero assembly hall. The cryostat mock-up will include a representation of a roof hatch, APA and CPA rail supports and feedthroughs. Multiple mock-ups of APA and CPA stacked panels will be installed. The APAs will include all mechanical mounting points and electrical connections such as optical-fiber readout cables and power cables. Prototype versions of the special equipment required for TPC installation will be used, including the lower panel staging platform, a transfer rail with trolley, and rolling-cart scaffolding. Scaffolding elements will be rented and the scaffolding will be erected by a contract or as part of a training program.

Initial tests, where appropriate, will be performed at a low elevation. For example, the installation trolley and single-switch rail segment will be tested at a low elevation with a dummy weight. After successful demonstration of the features at low elevation, the components will be moved to a high elevation for testing with full size mock-up panels.

6.7 Training and Access Control

Installation and Commissioning will be responsible for the personnel, equipment and procedures for providing cavern-access controls. These controls will include a mechanism for checking the training status of personnel, badge-in/badge-out procedures and closed-circuit-television monitoring of the entrance portal. Only trained personnel will be allowed below-ground. Members of the installation crew will be trained on specific installation tasks and must pass a qualification test. The training will likely use mock-up APAs constructed for the installation prototype. The training program will be modeled after the Fermilab “NuMI underground training” and will be developed in collaboration with Sanford Laboratory ES&H personnel.

Installation and Commissioning will provide all of the general tools and equipment needed

to support the personnel in their installation work. It will include hand tools, power tools, material-handling equipment, ladders, lifts, electrical meters and personal protective equipment (PPE). The detector system groups will provide any special equipment to check out or debug the power and read-out chain of the detectors.

6.8 Detector Commissioning

The construction of the two cryostats and the installation and commissioning activities will be staged such that both TPCs can be tested cold while one cryostat still remains available as a potential storage vessel. The LAr in one cryostat can be transferred to the other, and back again, if necessary. Once both TPCs are known to work properly at LAr temperature, the second fill will take place.

The commissioning sequence, illustrated in Figure 6-12, will start with the installation of the TPC in the east cryostat. During this time, construction of the west cryostat will be completed. The east cryostat will be filled initially with room-temperature GAr, injected from the bottom in order to displace the air upward through the top ports. The procedure used for this gas purge will be developed during prototype work. Following that, the LAr fill of the east cryostat will take four to six months, assuming continuous LAr deliveries. LAr purification will begin when the liquid level is high enough to start the recirculation pumps. The commissioning of the cryogenics system can also begin at this point but its full commissioning will require a fully loaded cryostat. When the cryostat is full, the TPC will undergo testing for about one month. In-vessel purity monitors will operate during this time.

Construction of the west cryostat is expected to finish during the LAr fill of the east cryostat. After the west cryostat is leak-checked and cleaned, its TPC installation will begin. After TPC installation and testing, the purging and cooling phases will proceed. Then the LAr from the east cryostat will be transferred to the west one, at which point the west TPC will begin a month of testing. During this time the east cryostat will be maintained cold via continuous GAr circulation. Once tests on the west cryostat are completed successfully, the east cryostat will be refilled with LAr, and its roughly five-month commissioning period begins.

6.9 ES&H

Careful consideration for ES&H will be demonstrated in the planning and execution of the installation and commissioning. Safety professionals will be involved in all phases. Hazards of note for the TPC installation are work at elevated heights and working in the confined space of the cryostat.

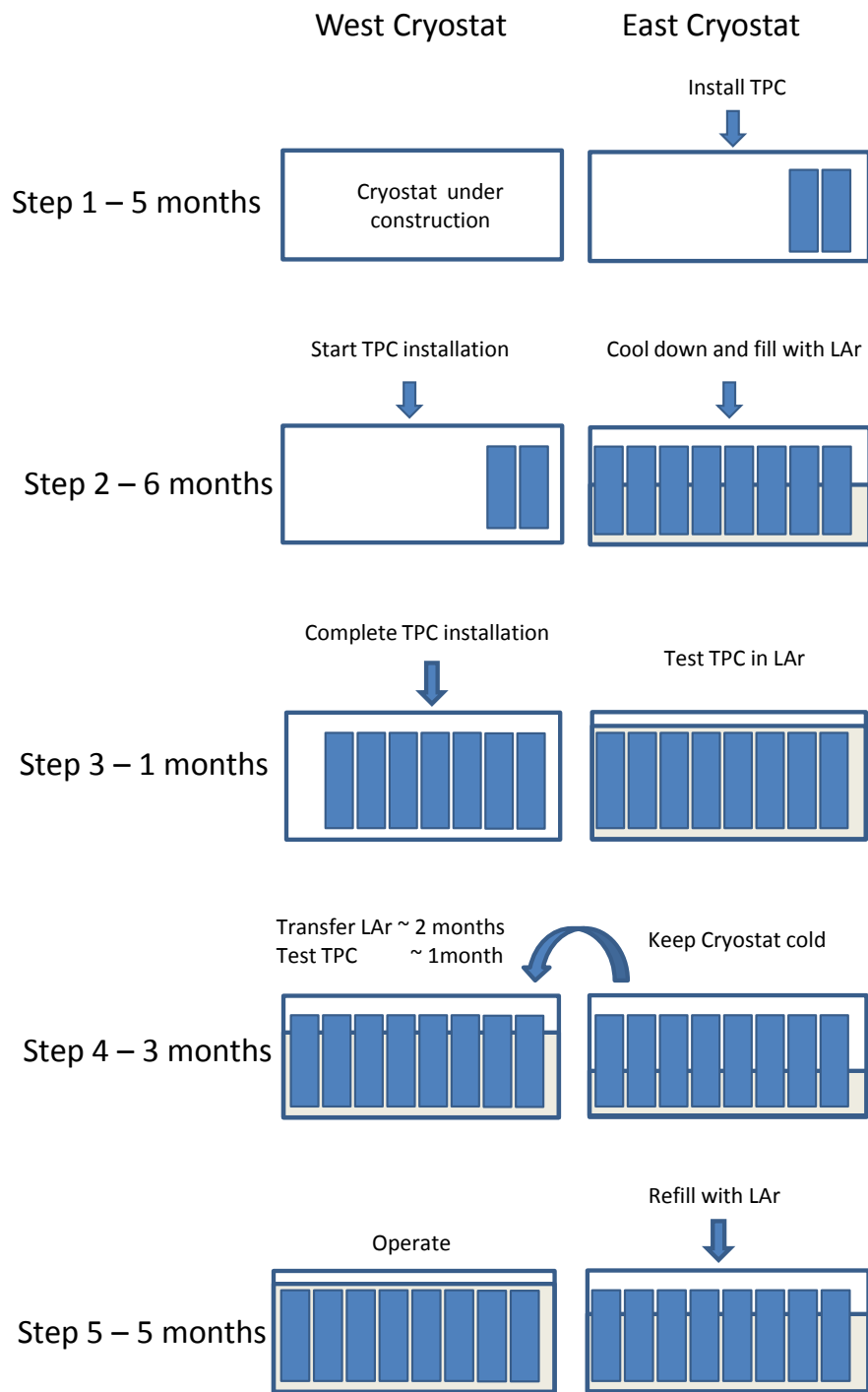


Figure 6-12: Startup and Commissioning Sequence

7 Engineering Prototype - LAr1 (WBS 130.05.09)

7.1 Introduction

The LBNE Liquid Argon Time Projection Chamber (LArTPC) project will include the construction and operation of a 1-kton LArTPC prototype cryostat in order to establish a credible basis for scaling to a 40-kton size and to further develop the LAr-FD design. The LAr1 engineering prototype will be built during the CD-1 to CD-2 (preliminary design) phase of the LBNE project. Many features of LAr1 will be at a 1:1 scale.

LAr1 will include a representative sample of Anode Plane Assemblies (APAs) and Cathode Plane Assemblies (CPAs) in a membrane-style cryostat. The APA read-out chain with cold electronics, cable routing, and data acquisition (DAQ) system will be a scaled-down version of the full LBNE LAr-FD. Analysis software will be used to reconstruct cosmic-ray events. The LBNE LAr subproject has determined that it is not necessary to install this engineering prototype in a beam line. To realize the LAr1 engineering prototype at the minimum cost and shortest schedule it will be constructed in the Fermilab D-Zero assembly building and reuse some of the D-Zero experiment infrastructure.

LAr1 will provide a high-level integration test of the operation of a large, modular-construction TPC suitable for deployment in detector of LAr-FD's scale. LAr1 will provide an opportunity to discover any problems of the TPC design that might not be exposed in more limited test stations that simulate fewer features of the full detector. A prototype TPC will be tested to help guide the final design and a production TPC will be tested to provide validation of the LAr-FD TPC design.

In addition to the integrated operations, other tests will be performed:

- Argon purity and drift length
- Argon purification techniques for large volumetric flow rates
- Mechanical structure of the APA both dry and after immersion in liquid argon

- Full readout chain including cold electronics
- High-voltage system, e.g., cabling, filtering and feedthroughs
- Cabling and cable feedthroughs
- The grounding scheme and common-mode noise control

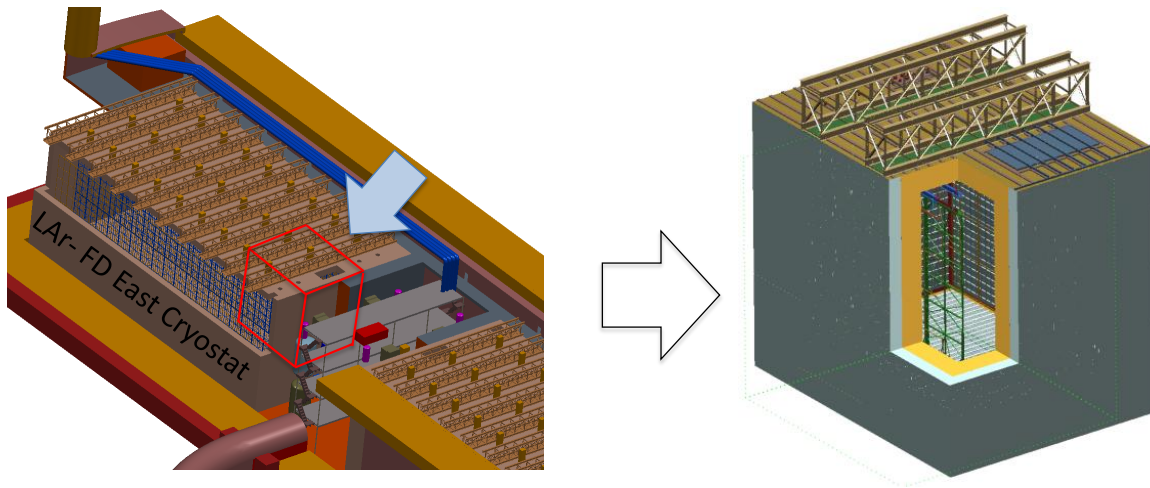


Figure 7-1: Engineering Prototype is like a section of a LAr-FD cryostat

7.2 Cryostat

The LAr1 cryostat will be constructed using the same stainless-steel, membrane-cryostat technology as planned for LAr-FD, with 0.8-m insulation thickness. The dimensions of the interior will be approximately 7.1 x 10.2 x 9.2 m high. The volume and surface area of the cryostat are scaled to, respectively, 1/30 volume and 1/10 surface area of a single LAr-FD cryostat. The ullage (gas height) will be the same as a LAr-FD cryostat (0.98 m). The depth

| Parameter | units | LAr1 | Single LAr-FD Cryostat | LAr-FD/LAr1 ratio |
|---------------------------------------|----------------|-------|---------------------------|----------------------|
| Total LAr Mass | kton | 0.826 | 25 | 30.3 |
| Cryostat Width | m | 7.14 | 24.14 | 3.4 |
| Cryostat Length | m | 10.2 | 48.96 | 4.8 |
| Cryostat Depth | m | 9.18 | 15.98 | 1.7 |
| Cryostat Surface Area | m ² | 464.0 | 4700 | 10.1 |
| LAr Depth | m | 8.2 | 15 | 1.8 |
| Ullage depth | m | 0.98 | 0.98 | 1.0 |
| Ullage % | % | 10.7 | 6.5 | 0.6 |
| APA quantity | | 3 | 108 | 36 |
| CPA quantity | | 6 | 144 | 24 |
| Drift Distance | m | 3.65 | 3.65 | 1 |
| Number of Signal/Bias feedthroughs | | 2 | 27 | 14 |
| Number of HV feedthroughs | | 2 | 8 | 4 |
| Total Heat Gain | W | 4419 | 50100 | 11.3 |
| Liquid Argon free surface area | m ² | 72.8 | 1181.9 | 16.2 |
| Cryostat Metal Surface Area in Ullage | m ² | 106.8 | 1325.2 | 12.4 |

Figure 7-2: Comparison of LAr1 and LAr-FD

of liquid argon within the cryostat for LAr1 will be 8.2 m which is about half the depth of a LAr-FD cryostat.

The cryostat will incorporate the following similar features as LAr-FD but scaled down in quantity/size consistent with a three-APA TPC.

- Access hatch
- Signal/power feedthrough port
- HV feedthrough port
- TPC support hangers
- Port for removable pump

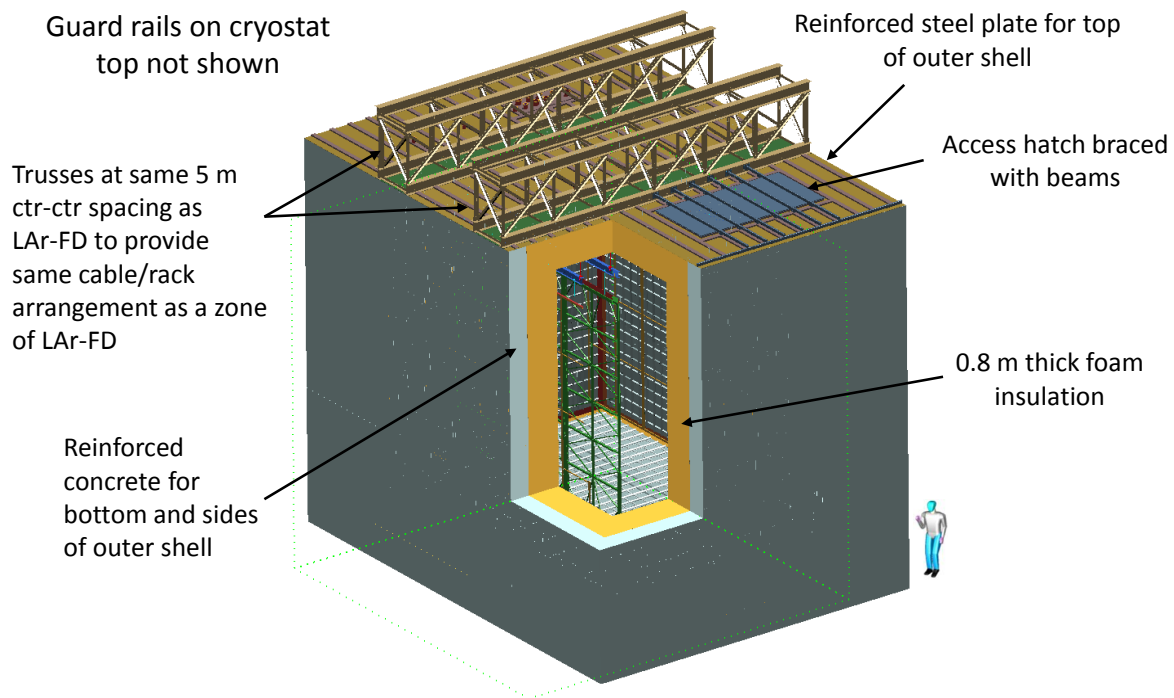


Figure 7-3: LAr1 cryostat enclosure

The bottom and sides of the enclosure for the cryostat will be constructed from reinforced concrete. Tubes will be embedded in the concrete base to allow the installation of heating elements. The top will be constructed from structural steel and the top will be configured to represent the geometry of a portion of the top of a LAr-FD module. The enclosure will be designed for pressures due to the liquid argon static head plus a gas ullage design pressure of 350 mbarg.

Fermilab will manage the design of the cryostat and will have design contracts with a membrane-cryostat company and a conventional-construction architectural and engineering company. Fermilab will develop the general arrangement of equipment inside and outside of the cryostat with a 3D CAD model. In addition, Fermilab will develop a set of design specifications with construction requirements for the membrane cryostat and a set of design specifications with construction requirements for the cryostat enclosure. Fermilab will establish a contract with an established membrane-cryostat company to develop the final design of the cryostat with oversight by Fermilab. Two membrane cryostat companies have been identified. Fermilab will also establish a contract with an A&E company to create the final design of the concrete enclosure and steel roof.

The concrete enclosure and steel roof will be of a conventional construction, built commercially. While the enclosure is being constructed, Fermilab will order the membrane-cryostat materials if they are not supplied directly by the membrane-cryostat company. The membrane-cryostat company will provide technical assistance with sourcing the membrane-cryostat materials. The standard set of quality assurance tests will be performed by the vendor or Fermilab. These tests include verifying the insulation and structural properties of the insulation panels. They also include verifying the material certification of the stainless steel membrane skin and verifying weldability.

The installation of the membrane cryostat will begin after the cryostat enclosure is complete and a moisture barrier is applied to the concrete interior. A temporary ventilation system with air monitoring will be setup inside the cryostat along with temporary lighting. The large hatch in the top of the cryostat will be used for moving material, equipment and personnel into the cryostat. Scaffolding with a stair tower will extend from the bottom of the cryostat to the hatch(s). A temporary raised floor will be installed at the bottom of the cryostat to protect the stainless-steel membrane once it is complete.

Installation of the cryostat insulation and stainless steel membrane will be performed by Fermilab personnel or contractors under the supervision of Fermilab personnel. The cryostat membrane company will provide onsite technical assistance throughout the membrane-cryostat installation process. The technical assistance will include training of technicians and welders and oversight of the installation. Specialized equipment including an automated welding machine will be rented from the membrane-cryostat company. After the top of the cryostat is installed the rails used to support the TPC will be mounted from the top of the cryostat. When the entire stainless steel membrane is complete the cryostat will be leak checked by the standard procedure used for LNG tanks. The process involves using a col-

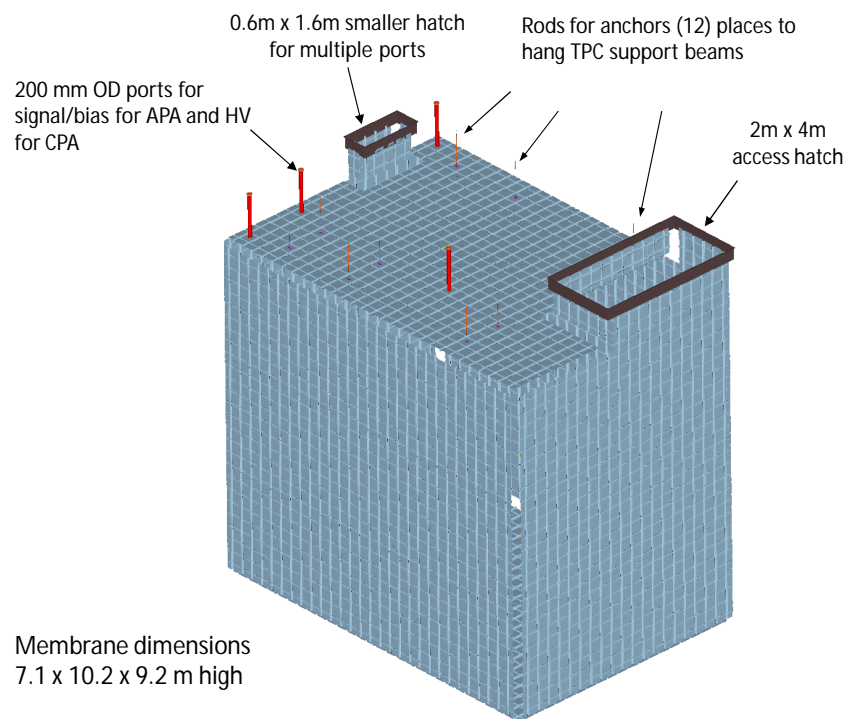


Figure 7-4: LAr1 cryostat membrane

chrometric paint applied to the weld joints and a gas containing ammonia injected into the insulation space. Leaks will be identified by a change in color at the location of the leak. The cryostat will be cleaned after the leak check and any repairs.

7.3 Cryogenic System

The LAr1 cryogenic system, which ensures cryogenic stability and LAr purity, will be built to satisfy the LAr-FD detector requirements (temperature stability to 1 K and purity to < 200 ppt oxygen equivalent contamination). The existing D-Zero Detector cryogenic infrastructure will supply LAr and LN2 to the prototype. The LAr1 cryogenic system will become an extension of this infrastructure. The existing infrastructure at D-Zero also provides a process-control system for the cryogenics and a fully developed ODH-monitoring system.

The cryogenic system will incorporate the following features that are also planned for LAr-FD:

- An LAr-pumping system with two permanent pumps and provisions for one removable pump
- A condenser vessel for boil-off gas reliquefaction
- Liquid filtration, mole sieve and copper to remove contaminants
- Gas filtration for the initial cryostat purge gas (GAr)
- Purity monitoring
- Cryostat monitoring instrumentation

LAr1 will not incorporate the use of a LN2 refrigerator because of the adequate on-site availability of LN2 storage dewars. The cryogenic system will be designed, assembled, and installed by Fermilab personnel. The existing D-Zero ODH exhaust system is sufficient so that the D-Zero assembly building can be ODH class 0 risk level after the addition of LAr1.

A LAr storage vessel will be constructed outside of the D-Zero assembly building which will allow LAr1 to be emptied and refilled with a minimum cycle time. The storage vessel will be constructed as a low pressure flat bottom cylinder and will be insulated by spray-on foam insulation with a protective mastic coating.

Table 7-1: LAr1 cryogenic system comparison to LAr-FD

| Parameter | LAr-FD (single cryostat) | LAr1 |
|---|--------------------------|--------------------------------|
| LAr total mass | 25 kton | 826 ton |
| Cryostat volume | 18887 m ³ | 668 m ³ |
| Ullage Depth | 0.96 (6.0 percent) | 0.98 (11.9 percent) |
| Heat leak to bath | 50.1 kw | 4.45 kw |
| Recirculation rate for initial purification | 0.2 vol changes per day | 0.2 to 0.5 vol changes per day |
| Ullage operating pressure | 130 mbarg | 130 mbarg |
| Relief valve set point | 250 mbarg | 250 mbarg |
| Pressure at bottom | 2217 mbarg | 1061 mbarg |

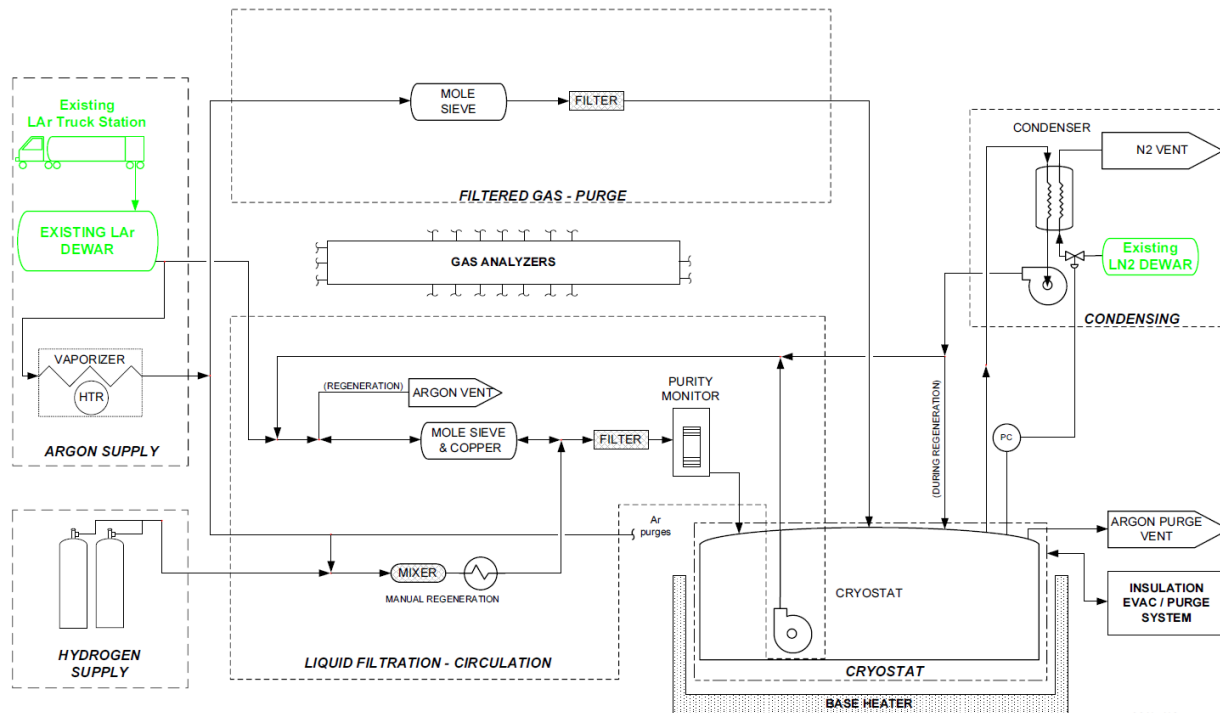


Figure 7-5: LAr1 cryogenic system process-flow diagram

7.4 Time Projection Chamber

The TPC will include three Anode Plane Assemblies (APA), six Cathode Plane Assemblies (CPA) and field cage elements. The TPC will be constructed using identical APAs, CPAs and FCAs as designed for LAr-FD. LAr1 will have no vertical stacking of the APAs and CPAs, so the active depth of the TPC is only seven meters. Additional features will be installed to ensure proper TPC operation given the half-height cryostat configuration. Some of these features will allow the field cage elements to attach to the end of the APAs and CPAs that is normally in the middle of the detector for a full height cryostat. The construction and assembly of all TPC mechanical components will use the same materials and techniques as designed for LAr-FD, with the exception of a reduced degree of automation that will be used to wire APAs for the full detector.

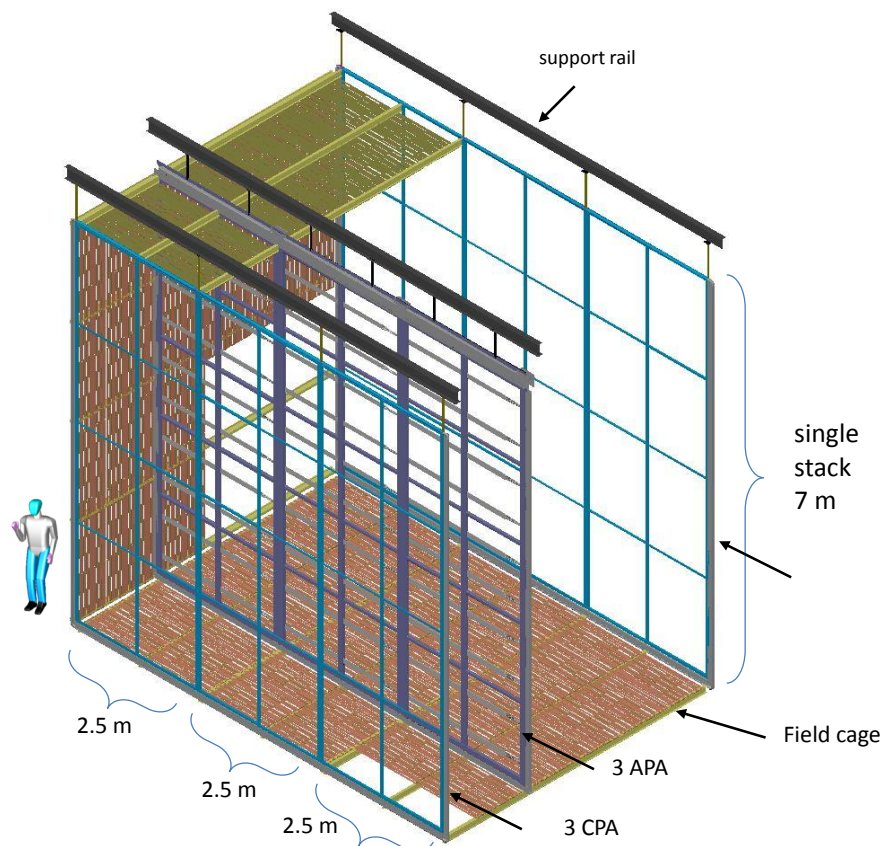


Figure 7-6: LAr1 TPC

Cold electronics will be installed on the APAs. The electronics components will closely resemble those designed for the LAr-FD. All key features of the LAr-FD electronic chain, including preamp, shaper, ADC, digital buffer, zero suppression and multiplexing will be implemented.

Due to the higher cosmic-ray flux expected in LAr1, the electronic readout system may not guarantee recording and transmitting 100% of the detected signal. Some LAr1 components may still be in prototype form. The TPC will include APA power supplies, signal/power cables with signal/power feedthroughs.

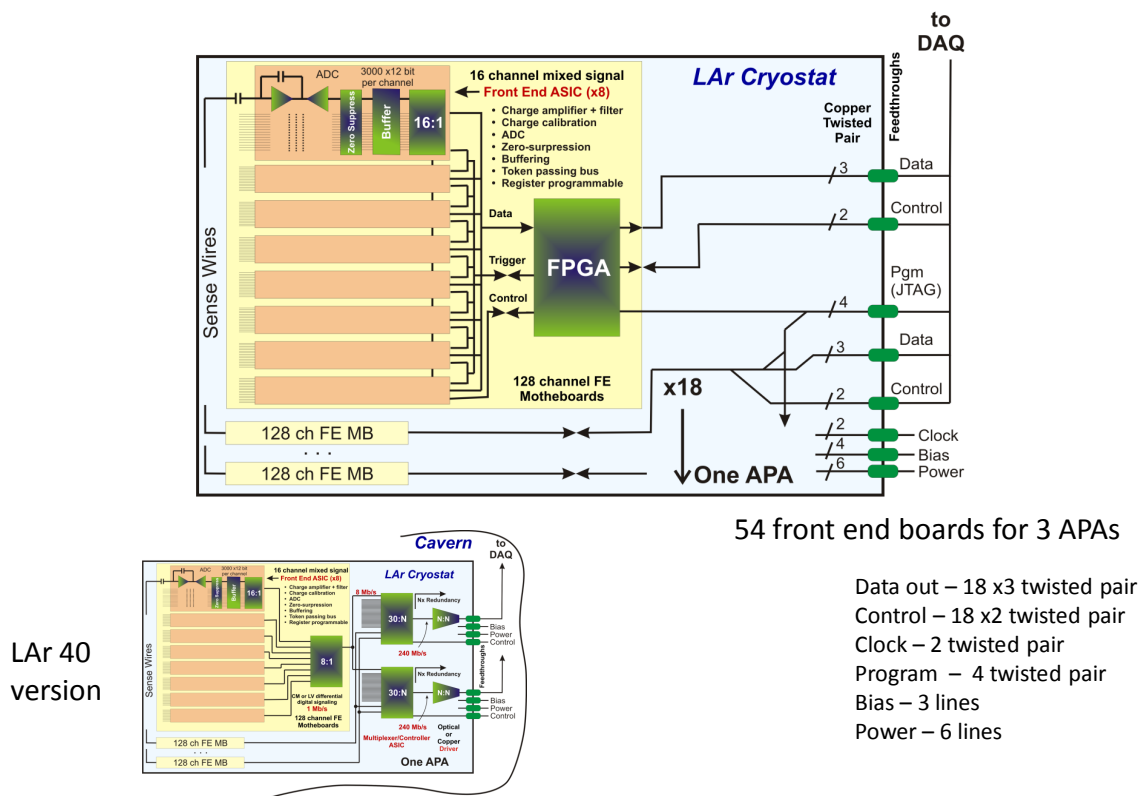


Figure 7-7: LAr1 cold electronics

Two different drift configurations are possible. One option has symmetric drift lengths of 2.72m. The second more likely option is an asymmetric 3.74m + 1.70m configuration. The 3.74m drift would match the drift planned for LAr-FD. The TPC also includes HV power supply, feedthrough and connections to the Cathode Plane Assemblies to achieve an electric field of 500 V/cm in the drift region. If an asymmetric drift configuration is implemented two HV power supplies will be required.

7.5 Data Acquisition

The LAr1 DAQ system will communicate with the front-end TPC electronics, including downloads of configuration, timing and control signals, as well as read out of TPC data and system status information. The DAQ system will transmit data to computers for buffering and online processing/triggering tasks, as well as subsequent storage to persistent media. To carry out these tasks, the DAQ system will include custom Data Concentrator Modules (DCM) and Timing Distribution electronics similar to components of the NOvA DAQ system. The DAQ will also include commodity hardware in particular network switches, routers, as well as computers and peripherals to be used in the online processing/trigger farm, DAQ supervisor and run control systems. These will be similar to the components that will be used for the LAr-FD DAQ. The DAQ footprint will be larger than a simple scaling from LAr-FD, due to the higher cosmic-ray flux expected in LAr1.

The DAQ system will include hardware and software systems that will continuously monitor various environmental and detector conditions. The DAQ system task will include development of software that will provide operator interfaces to the detector operations, TPC data readout, online processing and detector/environmental conditions monitoring tasks. The DAQ system may be required to provide read out and integration of signals from auxiliary detector elements inside or outside the TPC. Auxiliary elements are expected to have low channel counts which will be a minor load addition to the DAQ. These elements are not specified at this time.

7.6 Photon Detection

The planned LAr-FD photon-detection system technology will be tested in LAr1. The LAr-FD photo-detection system has the light guides integrated into the APA. Testing the photon-detection technology in LAr1 will identify and quantify unforeseen integration issues, e.g., signal/noise, mechanical fit-up and argon purity, for the benefit of LAr-FD.

7.7 Infrastructure

The infrastructure for LAr1 will be a combination of items re-used from D-Zero and items unique to the LAr-FD detector or the engineering prototype.

Infrastructure reused from D-Zero includes:

- 9000 gallon and 20,000 gallon liquid nitrogen storage tanks.

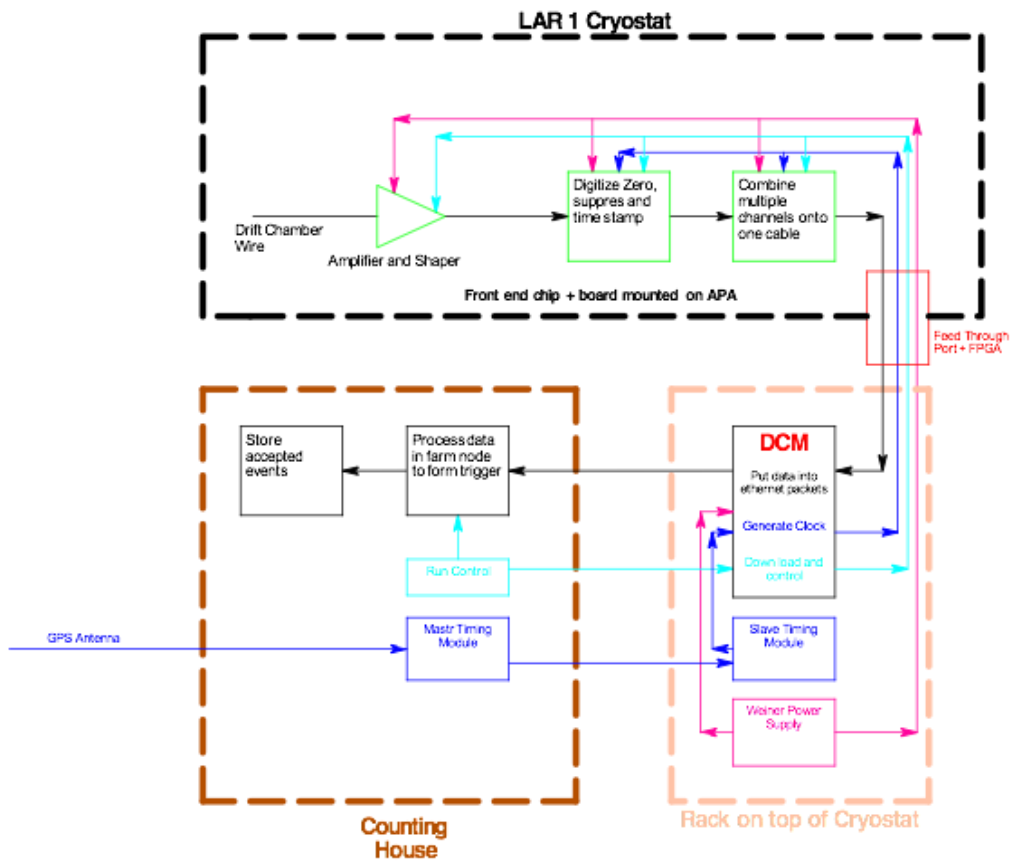


Figure 7-8: LAr1 DAQ block diagram

- Liquid nitrogen receiving station
- Liquid nitrogen sub-cooler on supply line
- Liquid nitrogen recapture and recirculation system on nitrogen vent line
- Liquid argon receiving station and liquid argon purity sampling station
- Liquid argon and liquid nitrogen piping to southeast corner of pit
- Argon floor drain and 100 ton underground insulated sump for argon spill scenario
- Dedicated 4500 and 13,000 cfm ventilation system for argon spill scenario
- Oxygen Deficiency Hazard (ODH) monitor system with 24 oxygen sensors distributed through the building and pit
- Gas shed and gas recirculation systems (piped to southeast corner of pit)
- Gas sampling and gas analyzer racks
- Two independent 100 scfm, 90-psig dry compressed instrument air systems
- Utility and insulating vacuum-pump systems (piped to southeast corner of pit)
- Vacuum-jacketed 4-in \times 6-in exhaust vent (piped from pit to outdoor vent stack)
- Operations control room with process control system (e.g., PLC, I/O bases, IFIX user interface)
- Chilled cooling water system and two De-Ionized chilled water systems
- 50 ton building crane and 10 ton building crane
- Welding shop and machine shop
- Emergency electrical power from a large diesel generator
- Fire safety systems with High Sensitivity Smoke Detection (HSSD) and FIRUS (provides 24/7 monitoring)

New infrastructure that will be added for the engineering prototype includes:

- Relay Racks with rack protection
- Power distribution to the relay racks and cable trays
- Detector reference ground

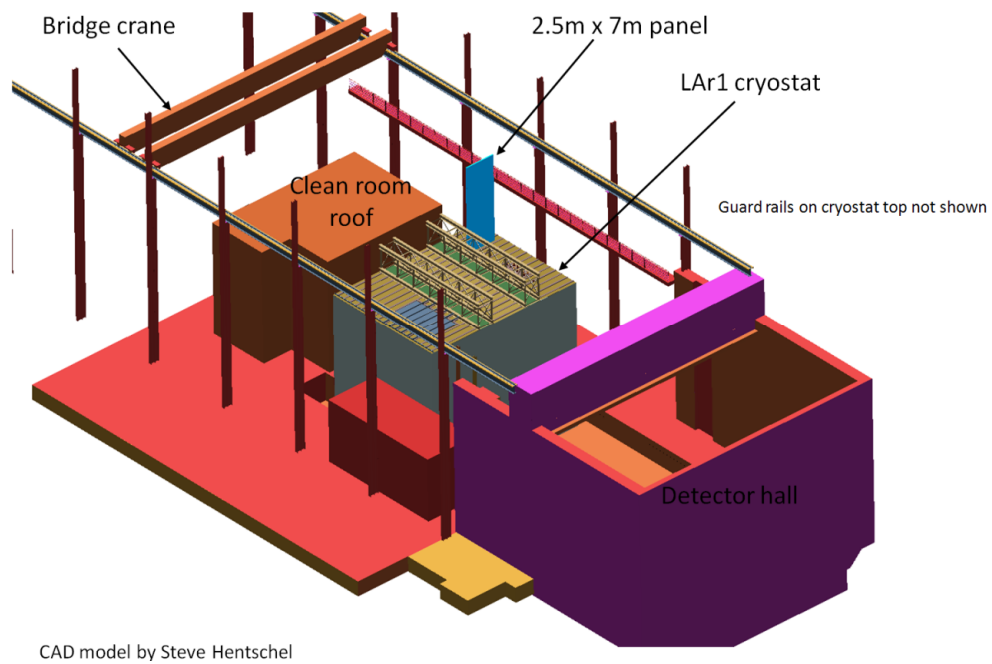


Figure 7-9: LAr1 Sited at D-Zero

- Support beams and hangers for the TPC
- Temporary ventilation and lighting in the cryostat
- Scaffolds
- Temporary raised floor for cryostat
- Clean room vestibule around cryostat hatch
- TPC handling equipment

The TPC power supplies and warm electronics will be mounted in the same style relay rack as planned for LAr-FD. The relay racks and cable routing on LAr1 will be the same as a segment of the LAr-FD layout near the feedthrough ports. The rack protection and monitoring on the LAr1 racks will be the same as LAr-FD.

A temporary floor will be installed in the cryostat for the installation of the TPC. It will be a LAr-FD-style modular floor. Fixed and movable scaffolds or personnel lifts will be used to enter the cryostat. Lifting equipment and procedures required to move APAs and CPAs from the horizontal position to the vertical position will be similar to those for LAr-FD. Equipment will be required to move single height APAs and CPAs from the hatch area to the final location on the supports rails. These may be fixtures that operate on the floor or motorized trolleys as planned for LAr-FD. Equipment and frame lifting will be handled by the D-Zero crane and a temporary hoist.

7.8 TPC and DAQ Installation

The steps for the installation of the TPC and DAQ for the engineering prototype will be similar to the steps for LAr-FD. The major differences in the steps are due to the prototype cryostat being approximately half the height of LAr-FD and the TPC being single stack. The DAQ will be installed and operational before the TPC installation. Cables for the TPC will be routed along the support rails. TPC installation begins once the DAQ and cables are ready. The APAs will be tested before and after installation. The DAQ will be used to continuously monitor installed APAs during the subsequent installation work. The prototype TPC installation schedule includes time allowances for a learning curve and for correcting unforeseen problems.

7.9 Filling and Testing

Cryostat filling, with argon, begins after Operation Readiness Clearance is granted by Fermilab management, and completion of TPC and DAQ testing. The three-stage LAr-FD filling

process will be used:

1. Argon piston purge - Introduce gaseous argon to the bottom of the cryostat at a rate sufficient to produce a virtual piston of argon that moves from the bottom of the cryostat towards the top at a rate greater than the diffusion of other gas down into the argon wave front
2. Cool down with the slow introduction of liquid argon to the bottom of the cryostat
3. Recirculate liquid argon through the liquid filters until sufficient electron drift lifetime is achieved

The length of time required for each of these steps will be better understood with the information obtained from the testing of similar operations with LAPD and the 35 ton LAr membrane-cryostat prototype. Filling of the cryostat will take approximately one month with the delivery of two tanker trucks per day. The argon will be tested upon delivery to verify the specified purity level has been met. The argon will pass through the purification filter/sieve before being transferred to the cryostat. For the initial purification the LAr will be circulated at a rate between one volume change each 2 to 5 days. The initial purification will last for up to one month based on the experience of the ICARUS detector. The purification time and the filter regeneration time will be better understood after the operation of the 35 ton prototype and MicroBooNE. Measurement data and log book records will be collected for all aspects of the argon fill and purification process.

7.10 ES&H

The Fermilab ES&H Manual shall be used as the guideline for the TPC installation. Hazards of note for the TPC installation are work at elevated heights and working in the confined space of the cryostat. Safety professionals will be involved in the planning and execution of the TPC and DAQ installation.

With the existing ODH monitoring and ventilation systems the D-Zero assembly hall is categorized as ODH class 0. A preliminary analysis of the additional hazards presented by the engineering prototype has been prepared and the hall is expected to remain ODH class 0.

8 Detector Development Program

8.1 Introduction

This chapter describes the development program designed to ensure a successful and cost-effective construction and operation of the massive, dual-cryostat LArTPC detector for LBNE and to investigate possibilities for enhancing the performance of the detector. The feasibility of the LArTPC as a detector has been demonstrated most impressively by the current state of the ICARUS experiment currently taking data at Gran Sasso.

It is understood that for successful operation an LArTPC has stringent requirements on

- argon purity which must be of order 200 ppt O₂ equivalent or better
- long-term reliability of components located within the liquid argon; in particular, the TPC and field cage must be robust against wire-breakage and must support a cool-down of over 200 K
- the front-end electronics which must achieve a noise level ENC of 1000e or better

The design of the LBNE LArTPC has evolved significantly from earlier concepts based on standard, above-ground, upright cylindrical LNG storage tanks which envisioned single TPC sense and high-voltage planes spanning the full width of the tank – essentially a direct scaling of previous detectors. Problems with the actual construction of such massive planes and with the logistics of being able to construct the TPC only after the cryostat was complete are avoided in the present design. In our design, TPC ‘panels’ are fully assembled and tested – including the electronics – independently of the cryostat construction. This modular approach is a key feature of the design. It has the benefit not only of improving the logistics of detector construction, but also the individual components can be of manageable size. It should also be noted that the cryostat itself is formed of modular panels designed for quick and convenient assembly.

8.2 Components of the Development Program

Programs of ongoing and planned development to allow the construction of massive LArTPCs in the U.S. have been developed and described in the *Integrated Plan for LArTPC Neutrino Detectors in the US* [11]. To advance the technology to the detectors proposed for LBNE, the U.S. program has three aspects:

- a demonstration that the U.S program can reproduce the essential elements of the existing technology of the ICARUS program
- a program of development on individual elements to improve the technology and/or make it more cost-effective
- a program of development on how to apply the technology to a detector module

A summary of the items in the program is given in the following tables. Table 8-1 lists the activities that are part of the LBNE Project (“on-project”) described in this chapter, a short description of the information needed and the LBNE milestone corresponding to when the information is required. Table 8-2 lists off-project activities, the aspect of these activities that is applicable to LAr-FD and the LBNE milestone at which the information is required. These aspects will be described in more detail in the following sections. As will be explained below, these are not R&D activities, but rather elements of the preliminary engineering design process.

8.3 Scope and Status of Individual Components

8.3.1 Materials Test System

An area for LAr detector development, shown in Figure 8-1, has been established in the Proton Assembly Building at Fermilab. The Materials Test System (MTS) has been developed to determine the effect on electron-drift lifetime of materials and components that are candidates for inclusion in LAr-FD. The system essentially consists of a source of clean argon (< 30 ppt O₂ equivalent), a cryostat, a sample chamber that can be purged or evacuated, a mechanism for transferring a sample from the sample chamber into the cryostat, a mechanism for setting the sample height in the cryostat so that it can be placed either in the liquid or in the gas ullage above the liquid, a temperature probe to measure the temperature of the sample, and an electron-lifetime monitor. The system is fully automated and the lifetime data are stored in a single database along with the state of the cryogenic system.

Table 8-1: LBNE on-project development activities

| Activity | LAr-FD Information | Need by |
|-----------------------|----------------------------------|---------------------------|
| In-liquid Electronics | Low noise readout, long lifetime | LAr1 construction |
| TPC Construction | Mechanical design | LAr1 construction |
| 35-ton Prototype | Cryostat construction | LAr1 cryostat procurement |
| LAr1 | detector integration | LBNE CD-2 |

Table 8-2: LBNE off-project development activities

| Activity | LAr-FD Applicability | Status | Need by |
|--------------------------|------------------------|----------------|-------------------|
| Yale TPC | None | Completed | NA |
| Materials Test System | Define requirements | Completed | NA |
| | Materials testing | Operating | As Req'd |
| Electronics Test Stand | Electronics testing | Operating | As Req'd |
| LAPD | Purity w/o evac. | Operating | LBNE CD-2 |
| | Convective flow | Operating | LBNE CD-2 |
| Scintillator Development | Photon Det. Definition | Completed | LAr1 Construction |
| | Industrialization | Not started | LBNE CD-3 |
| ArgoNeuT | Analysis tools | On-going | LBNE CD-2 |
| MicroBooNE | Electronics tests | Construction | LBNE CD-3 |
| | DAQ algorithms | In development | LBNE CD-3 |
| | Analysis tools | In development | LBNE CD-2 |
| | Lessons learned | Not started | LBNE CD-3 |



Figure 8-1: Liquid argon area at the Proton Assembly Building at Fermilab

A noteworthy feature is the novel bubble-pump filter inside the cryostat. In case of argon contamination, this can filter the cryostat volume in a few hours, allowing us to continue studies without having to refill. A schematic of the MTS is shown in Figure 8-2.

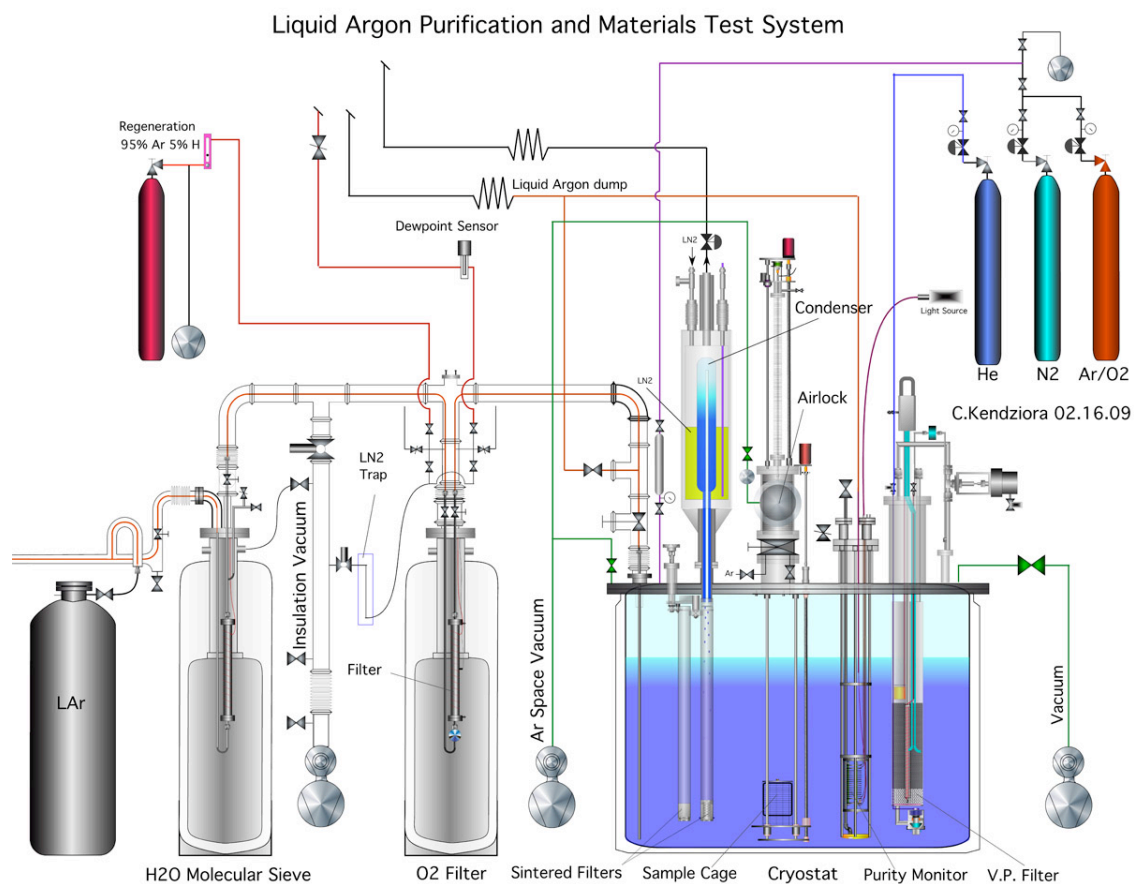


Figure 8-2: Schematic of the Materials Test System (MTS) cryostat at Fermilab

The major conclusions of the studies to-date are summarized here. No material has been found that affects the electron-drift lifetime when the material is immersed in liquid argon – this includes, for example, the common G-10 substitute, FR-4. On the other hand, materials in the ullage can contaminate the liquid; this contamination is dominated by the water outgassed by the materials and as a result is strongly temperature-dependent. Any convection currents that transport water-laden argon into the LAr and any cold surfaces on which water-laden argon can condense will fall into the LAr and reduce the electron lifetime. Conversely, a steady flow of gaseous argon of a few ft/hr away from the LAr prevents any material in the gas volume from contaminating the LAr.

These results are taken into account in the design of both MicroBooNE and LAr-FD. For LBNE they have been cast as detector requirements. The MTS will continue to be used by MicroBooNE and LBNE to test detector materials such as cables that will reside in the ullage.



Figure 8–3: Electronics test TPC insertion into cryostat

8.3.2 Electronics Test Stand

The Electronics Test Stand is also installed in the Proton Assembly Building at Fermilab. It consists of a cryostat, served by the same argon source as the Materials Test System, and a TPC with a 50-cm vertical drift terminating in three planes of 50 wires each, arranged at 120 degrees. Figure 8–3 shows the TPC being inserted into the cryostat. Two sets of small scintillation counters outside the cryostat are used to trigger the system on cosmic rays. The data from this system provide a crucial check of simulations of the electron drift and the signals induced on the wires.

The front-end electronics and the crucial shaping filters for the ArgoNeuT experiment were developed on this system from the sort of data shown in Figure 8–4. These data also led to the development of the FFT technique for reconstructing track hits. A hybrid pre-amplifier has been successfully tested in LAr and has demonstrated excellent noise performance.

The ASIC front-end amplifiers for LBNE will also be tested, providing assurance that bench measurements are reproducible in a functional TPC.

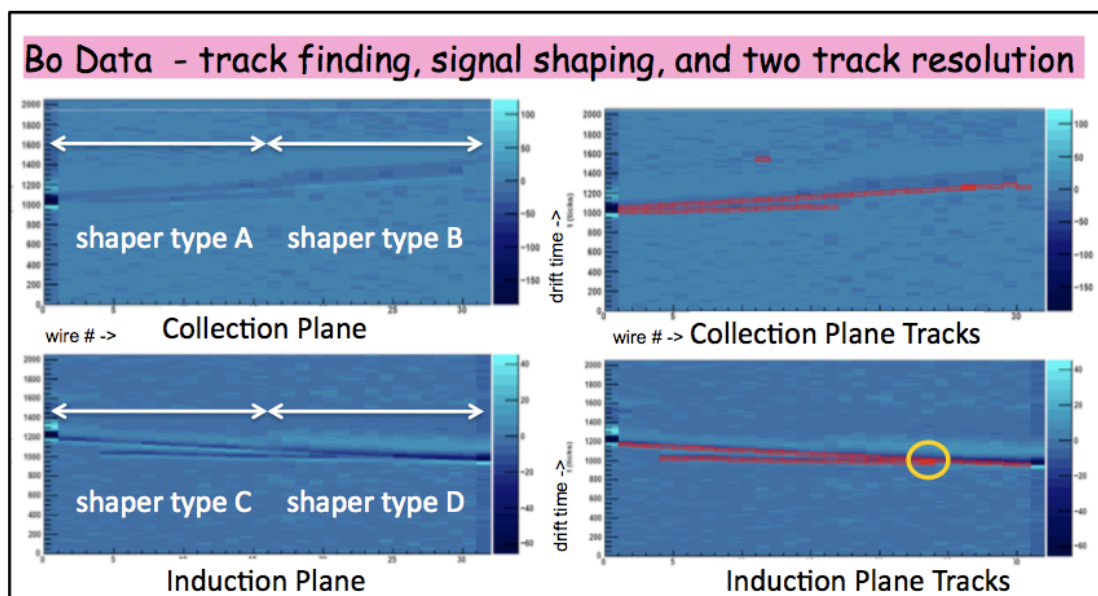


Figure 8-4: A cosmic ray with a delta electron as seen in the Electronics Test Stand TPC. Note the different width of the shaded portions along the tracks in the left panels. These are due to the two shaping circuits being tested. The readout scheme for MicroBooNE and LBNE uses digital signal processing and has minimal filtering before the ADC.

8.3.3 Liquid Argon Purity Demonstrator

The Liquid Argon Purity Demonstrator (LAPD) is a significant-scale system, 30 tons, consisting of a complete cryogenic recirculation system and a vessel capable of achieving large-drift LArTPC purity specifications without initial evacuation. While large cryogenic systems have been built by U.S. groups, this is the first with a purification system intended specifically for an LArTPC. As such it is providing valuable experience and data for the construction of future systems. Figure 8-5 shows the filtration system and the LAPD tank in place at Fermilab's PC-4 facility.

The tank is instrumented with several systems. An array of sniffers measures the oxygen concentration at six heights during the gas-purge; a system of temperature monitors can be lowered and raised inside the tank to measure temperature gradients; an ICARUS-style purity monitor is deployed on the input line to the tank; and four ICARUS-style purity monitors measure electron-drift lifetime within the tank.

Analytic instrumentation includes a water meter and an oxygen meter, both with ppb sensitivity, and a nitrogen meter with 10-ppb sensitivity, installed for the purpose of taking measurements throughout the system. The scheme for displacing the air (at atmospheric pressure) in the cryostat interior volume by introducing and purifying LAr consists of several steps: a gas purge with argon, gas recirculation with purification, an LAr fill while



Figure 8-5: Liquid Argon Purity Demonstration filtration and tank at the PC-4 facility

venting (to drive out any contaminants remaining in the gas) and recirculation of the LAr through the purification system.

The primary motivation of LAPD is to demonstrate an electron-drift lifetime of several milliseconds, which has now been accomplished. LAPD began purging with argon gas in September 2011. The tank was filled with LAr to the 40% level on November 1. A drift electron lifetime of 3 ms was achieved in late November after some re-work was done on the filter material containment system. The filters became saturated (with water) in mid December, were regenerated and were returned to service in late January 2012. The drift-electron lifetime reached 3 ms after four days of operation and has continued to improve as of February 2.

Now that high purity has been achieved, the effect of varying the conditions (gas and/or liquid recirculation, different recirculation rates, no recirculation) can be investigated. The size of the tank allows us to use the measurements from the temperature-monitor system to check the accuracy of ANSYS simulations of the temperature profile within the tank. This will check the simulations and should confirm the small temperature gradients and resulting low velocities of the convection flows predicted for large LArTPCs.

A problem was found with the building electrical system in early March 2012 that necessitated ending the planned suite of LAPD tests prematurely. The current plan is to perform additional re-work on the filter material containment system over the next few months, followed by a second phase of testing. At this time, LAPD management and operations will be transferred to LBNE to support the 35-ton membrane-cryostat prototype described in Section 8.3.7.

8.3.4 Photon Detection R&D

The R&D program for Photon Detection is based on a promising, new, cost-effective scheme for light collection in LArTPCs as described in a NIM article by Bugel et al [27]. The design is based upon lightguides fabricated from extruded or cast acrylic and coated with a wavelength-shifter doped skin. Multiple acrylic bars are bent to guide light adiabatically to a single cryogenic PMT. Prototypes of the basic detector elements have been shown to perform well. These lightguides have a thinner profile than the usual TPB-coated PMT-based system, occupying less space in an LAr vessel and resulting in more fiducial volume. Another advantage of this system is that the bars are inexpensive to produce. The most convenient place for the paddles is between the wire planes that wrap around the APAs.

Lightguide R&D has advanced rapidly since the initial publication resulting in $\sim 3 \times$ higher light yields. The R&D is now sufficiently advanced to provide a technical basis for the LAr-FD reference design. On-going design efforts at MIT, Indiana University, and Fermilab are directed toward industrial-scale production and the evaluation of lower-cost fluors

that are effective in converting VUV photons. These efforts also include investigating PMTs with increased quantum efficiency as well as other efficient light-collection technologies, such as Geiger-mode avalanche photodiodes (commonly known as Silicon Photomultipliers, or SiPMs).

8.3.5 TPC Design

The design for LBNE has adopted the basic ICARUS multi-plane, single-phase TPC concept and has incorporated new features suitable for a very large detector. The main emphasis of the development program is to develop a TPC design that is highly modular, low-cost, robust and easily installed inside a finished cryostat.

A significant effort has been focused on minimizing the dead space between detector modules to improve the fiducial versus total LAr-volume ratio. The APA reference design accomplishes this goal but requires making ~ 2 million high-quality wire terminations. The wire-termination scheme used by ICARUS has proven to be very reliable but it is too labor-intensive to fabricate for a million-channel detector system. We have adopted the wire-solder + wire-epoxy termination scheme that has been used for decades on drift chambers and proportional wire chambers to mount Cu-Be wires. The termination scheme was used to terminate 2.5 million anode wires in the CMS end-cap muon system. Cu-Be wires have excellent mechanical properties and the advantage of low resistance compared to stainless steel. A study is currently underway within the LAr-FD subproject to identify the optimum wire-bonding parameters. The focus is currently on finding a commercial epoxy that optimizes the qualities of bond strength, cure time and low-temperature operation.

8.3.6 Electronics Development

The work to-date on cold electronics has established that no show-stoppers exist. The remaining activities outlined here concern performance optimization of the CMOS ASICs, the evaluation of several widely available CMOS technologies, and the development of readout architectures appropriate and timely for various scenarios of very large detectors.

8.3.6.1 CMOS Transistors: Lifetime Verification and Technology Evaluation

The results of the design of the CMOS electronics for operation at LAr temperature (89 K), performed so far by the MicroBooNE and LBNE collaborations, as well as by a large collaboration led by Georgia Inst. of Technology, are summarized in Section 3.7.

Briefly, the fundamentals are: charge-carrier mobility in silicon increases at 89 K, thermal fluctuations decrease with kT/e , resulting in a higher gain (transconductance/current ratio

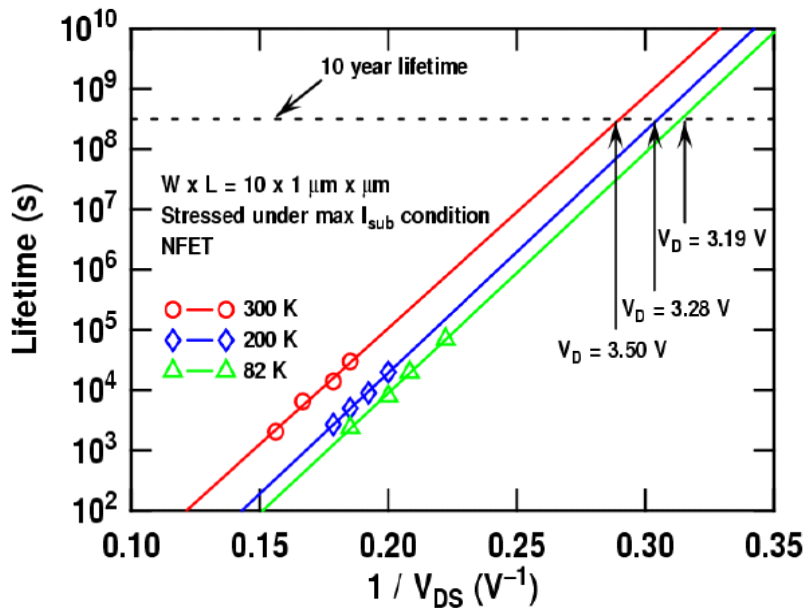


Figure 8-6: Lifetime at different temperatures vs V_{DS}

$= g_m/i$), higher speed and lower noise. For a given drain-current density the same degree of impact ionization (measured by the transistor substrate current) occurs at a somewhat lower drain-source voltage at 89 K than at 300 K. The charge trapped in the gate oxide and its interface with the channel causes degradation in the transconductance (gain) of the transistor and a threshold shift. The former is of major consequence as it limits the effective lifetime of the device (defined in industry and the literature as 10% degradation in transconductance). Thus an MOS transistor has equal lifetime due to impact ionization at 89 K and at 300 K at different drain-source voltages. This is illustrated in Figure 8-6.

This feature offers a tool for accelerated-lifetime testing by stressing the transistor with both increased current and increased voltage, and monitoring the substrate current and the change in g_m due to impact ionization. In these conditions, the lifetime can be reduced arbitrarily by many orders of magnitude, and the limiting operating conditions for a lifetime in excess of ~ 20 years can be determined. With this foundation, more conservative design rules (lower current densities and voltages) can be derived and applied in the ASIC design, as has been done for the ASIC described in Section 3.7.1. The goal of this part of the program is to verify by accelerated testing the expected lifetimes for the several widely available CMOS technologies under consideration (TSMC, IBM, AMS). It should be noted that this is a standard test method used by the semiconductor industry. These methods are used to qualify electronics for deep space NASA missions as well as commercial PCs.

8.3.6.2 Readout Architectures, Multiplexing and Redundancy

A high degree of multiplexing after digitization of signals is essential for a TPC with 0.5 million wires in order to reduce the cable plant and the attendant outgassing. Just how high a multiplexing factor should be chosen is a matter of study, considering the risk of losing one output data link. A part of the program will include system designs with redundant links and redundant final multiplexing stages to minimize the risk of losing the data from a significant fraction of the TPC (note that even with a multiplexing factor of 1/1024 and no redundancy, one failed link would result in a loss of 0.2%).

8.3.7 Cryostat Development: 35-ton Prototype

The next step in the cryostat-prototype program is intended to address project-related issues: (1) to gain detailed construction experience, (2) to develop the procurement and contracting model for LAr-FD and (3) to incorporate the design and approval mechanism in the Fermilab ES&H manual. (Membrane cryostats are designed in accordance with European and Japanese standards.) At present, we are in the process of procuring the cryostat components for a 35-ton membrane cryostat from IHI.

The LBNE project has contracted with the Japanese company IHI to build a small prototype membrane cryostat at Fermilab. This approximately 35-ton unit is to be built and made operational in 2012 at Fermilab's PC-4 facility where LAPD is located. It is intended to demonstrate high-purity operation in this type of cryostat and the suitability of the planned LAr-FD construction techniques and materials. The testing programs for LAPD and the small prototype will be similar. LBNE's 35-ton membrane cryostat will use a large portion of the cryogenic-process equipment installed for LAPD.

The prototype membrane cryostat's total size, including insulation and concrete support, is approximately 4.1 m × 4.1 m × 5.4 m, and will hold approximately 826 tons of LAr. The insulation thickness will be 0.4 m rather than the 1.0 m chosen for our reference design. The techniques of membrane-cryostat construction will be demonstrated to be a fit for high-purity TPC service. Welding of corrugated panels, removal of leak-checking dye penetrant or ammonia-activated leak-detecting paints, and post-construction-cleaning methods will be tested for suitability of service. Residual contamination measurements at different elevations during the initial GAr purge process will be compared to computational predictions and will validate the purge-process modeling of a large rectangular vessel. The prototype membrane cryostat will be filled with LAr. Purity levels of the liquid with time and electron-drift times will be measured using purity monitors installed in the liquid bath. Heat-load measurements will be made and compared to calculations. Eventually, connectors and feedthroughs, ports and other features that are planned for the reference design will be incorporated into the prototype. Materials and cold-electronics testing can be done along with electron-drift-time measurements.

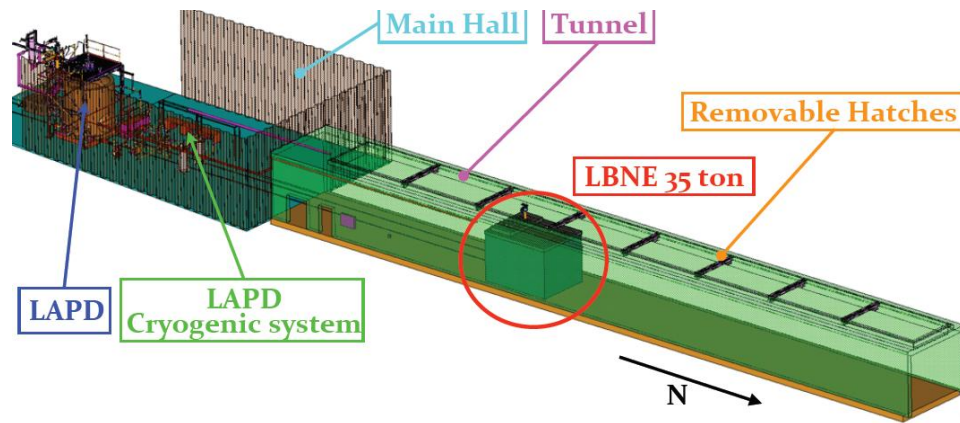


Figure 8-7: Layout of 35-ton prototype at Fermilab's PC-4 facility

In principle, a thin-walled membrane cryostat is as suitable as a thick-walled cryostat for use with high purity LAr. Both would be constructed with 304 stainless steel with a polished surface finish. Both would use passive insulation. The total length of interior welds required for construction would be similar in both cases. The leak checking procedure would be the same in both cases.

The significant difference between membrane cryostats and thick-walled cryostats is the depth of the welds used to construct the vessel. The majority of membrane-cryostat welds are completed in one or two passes with automatic welding machines. A second difference, and a major advantage, is that the membrane cryostat is a standard industrial design that has been in use for over 40 years. A thick-walled cryostat vessel would be custom designed and would require significant engineering and testing. A third difference, and another major advantage, is the ability to purge the membrane cryostat insulation space with argon gas so that a leak cannot affect the purity if it escapes detection and repair.

A 3-m × 3-m wall panel shown in Figure 8-8 was constructed at Fermilab using materials and technical guidance from GTT. The labor hours used in construction are consistent with the vendor estimates. The wall panel was leak tested (none were found) and vacuum tests were performed on the insulation system. We found that the insulation system is designed to allow vacuum pumping of the main cryostat volume to a hard vacuum. This result demonstrates that vacuum pumping of a membrane cryostat is feasible, if it is found to be required. No modifications to the vendor-supplied components are required to accomplish this.



Figure 8-8: Membrane panel assembly and components

8.3.8 One-kiloton Engineering Prototype: LAr1

The last major project of the development program is the construction of a 1-kton cryostat, called LAr1, to be equipped with the same TPC and electronics as LAr-FD. The proposed cryostat will reside at Fermilab's D-Zero Assembly Hall using the LAr and LN infrastructure already existing for the D-Zero detector, which is no longer operating. The end of the Tevatron era allows LBNE to make immediate use of this existing building and the existing cryogenic processes it houses. This prototype will serve to test all the major components of LAr-FD, in particular:

- Gain further experience in cryostat construction and procurement
- Exercise the assembly and testing procedures for all the individual components
- Test the integration features of the detector design including:
 - the TPC supports
 - the cold electronics, signal feedthroughs and power feedthroughs
 - the grounding scheme and control of electrical noise
 - the purification system
 - the cryogenic system and the cryogenic control system

- Exercise the installation procedure

The means of demonstrating that these goals have been met is via the acquisition and reconstruction of a sample of through-going cosmic rays. LAr1 is being managed as a semi-independent subproject. See Chapter 7 for further details. The subproject was subjected to a CD-1-style technical, cost, schedule and management review in August 2011. LAr1 construction could be completed in early 2014 if both funding and resources are available.

An attractive extension of the LAr1 goals would be to measure the cosmic-ray spallation background rates expected in LAr-FD. A sample of cosmic-ray spallation interactions in the LAr1 volume could be acquired in a relatively short period of time. Charged kaons associated with the interaction could be identified and their momentum measured. Charged kaons produced by K^o charge-exchange could also be identified and their rate and angular distribution measured.

8.3.9 Physics Experiments with Associated Detector-Development Goals

Two projects, ArgoNeuT and MicroBooNE, which are physics experiments in their own right, are also contributing to the development of the LBNE experiment. Their most important role is in providing data and motivation for the development of event reconstruction and identification software.

8.3.9.1 ArgoNeuT - T962

The Argon Neutrino Test (ArgoNeuT) is a 175-liter LArTPC which completed a run in the NuMI neutrino beam. The $0.5\text{ m} \times 0.5\text{ m} \times 1\text{ m}$ LArTPC was positioned directly upstream of the MINOS near detector, which served as a muon catcher for neutrino interactions occurring in ArgoNeuT.

ArgoNeuT began collecting data using the NuMI anti-muon neutrino beam in October 2009 and ran until March 1, 2010. ArgoNeuT's $\sim 10\text{k}$ events motivate the development of analysis tools, and are the basis for the first measurements of neutrino cross sections on argon. An event with two π^0 decays is shown in Figure 8-9. ArgoNeuT was also the first LArTPC to be exposed to a low-energy neutrino beam and only the second worldwide to observe beam-neutrino interactions. The ArgoNeuT collaboration is currently preparing (1) a NIM paper that documents the detector performance using NuMI beam muons and (2) the first physics paper on muon-neutrino charged-current differential cross sections on argon. See Figures 8-11 and 8-12.

A deconvolution scheme using an FFT has been applied to the ArgoNeuT data. This procedure eliminates a problem with the ArgoNeuT electronics (which were D-Zero spares and

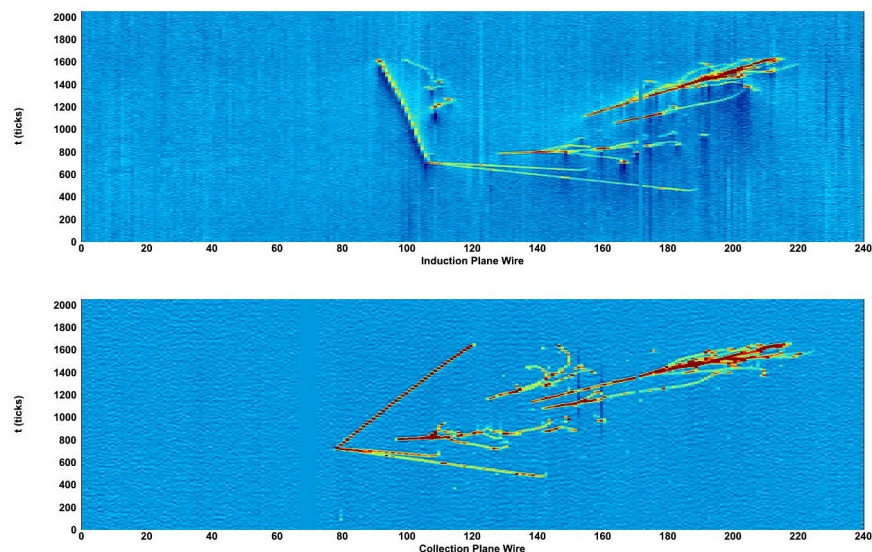


Figure 8-9: A neutrino event with four photon conversions in the ArgoNeuT detector. The top (bottom) panel shows data from the induction (collection) plane after deconvolution.

could not be modified for ArgoNeuT). Another more significant benefit of deconvolution is that bi-polar induction-plane signals can be transformed into uni-polar collection-plane signals. An example of this is shown in Figure 8-10. A selection of figures from the draft NIM paper are reproduced below.

The applicability of ArgoNeuT is that it provides a set of data in the same range of energy as the LBNE neutrino beam, enabling the development of analysis algorithms that can be utilized for LAr-FD physics analysis with little or no modification.

8.3.9.2 MicroBooNE E-974

The MicroBooNE experiment is an 86-ton active mass LArTPC, (170-ton argon mass) in the construction phase. It has both a physics program and LArTPC development goals.

MicroBooNE received stage 1 approval from the Fermilab director in 2008, partial funding through an NSF MRI in 2008 and an NSF proposal in 2009. MicroBooNE received DOE CD-0 Mission Need in 2009 and CD-1 in 2010, and CD-2/3a review in September 2011. It plans to start running in early 2014.

As well as pursuing its own physics program, MicroBooNE will collect a large sample ($\sim 100k$) of low-energy neutrino events that will serve as a library for the understanding of neutrino interactions in LAr. Because MicroBooNE is at the surface, it will also have a large sample of cosmic rays with which it can study potential backgrounds to rare physics. The process of designing MicroBooNE has naturally stimulated several developments helpful to the

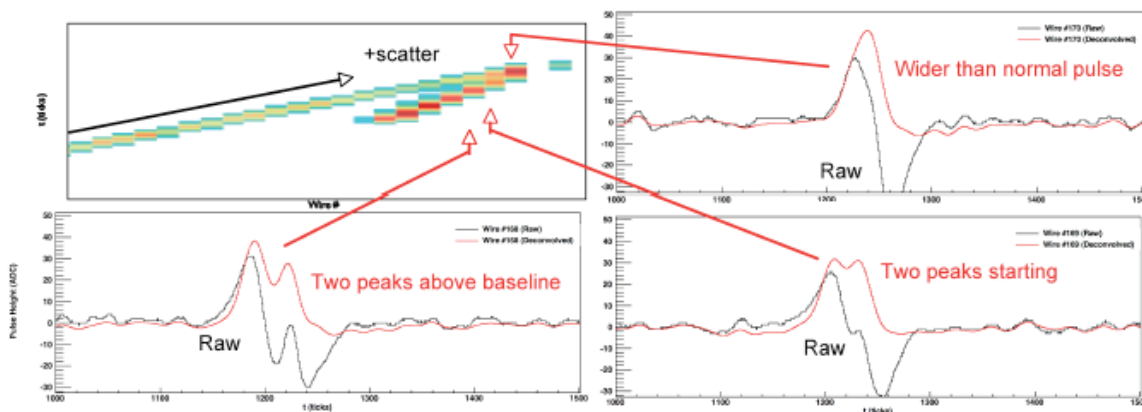


Figure 3: (Upper left) A set of tracks as seen on the (deconvoluted) induction plane. The wire views on three adjacent wires are also shown in order to demonstrate the effects of deconvolution on the raw wire pulses. The raw data can be seen in black and the deconvoluted data can be seen in red.

Figure 8-10: Figure from the ArgoNeuT draft NIM paper.

LBNE program. Studies of wire material, comparing Be-Cu with gold-plated stainless steel in terms of their electrical and mechanical properties at room and LAr temperatures, and techniques for wire-tension measurement are immediately relevant. Expertise has been developed generating simulations of electrostatic-drift fields as well as simulations of temperature and flow distributions in LAr cryostats which is being applied to the LAr-FD TPC and cryostat. MicroBooNE will use the front end of the proposed in-liquid electronics as the wire-signal amplifiers and the DAQ developed for MicroBooNE will exploit compression and data-reduction techniques to record data with 100% livetime.

In summary, MicroBooNE's LArTPC development goals that are pertinent to LAr-FD are

- large-scale testing of LBNE front-end electronics, similar in scale to LAr1
- testing of continuous data-acquisition algorithms
- refinement of the analysis tools developed in ArgoNeuT
- provide costing and construction lessons-learned

8.4 Summary

Impressive progress has been made in the development of LArTPC technology over the last few years. All elements of the development program have completed the R&D phase. Credible

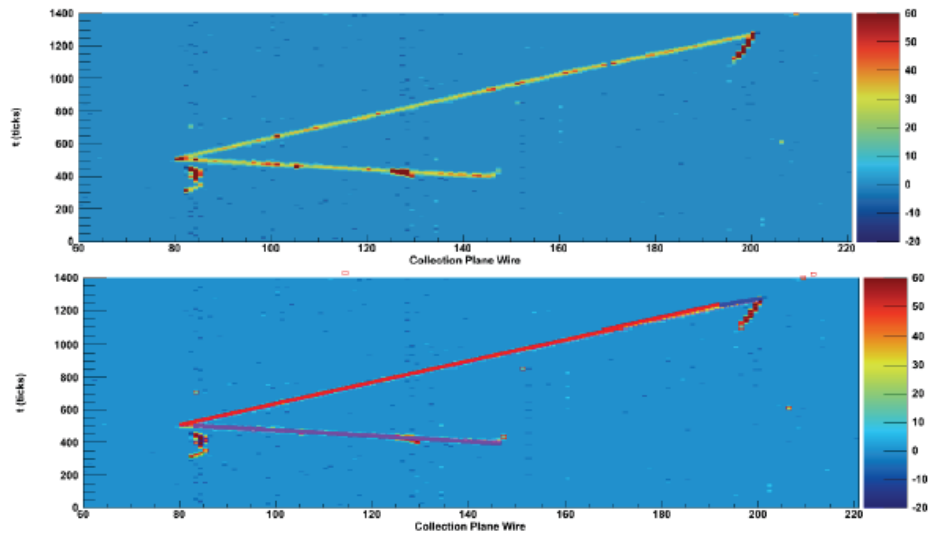


Figure 6: (Top) A neutrino candidate in ArgoNeuT as seen on the collection plane. (Bottom) The Hough lines found with the line-finding algorithm overlaid on the particle tracks.

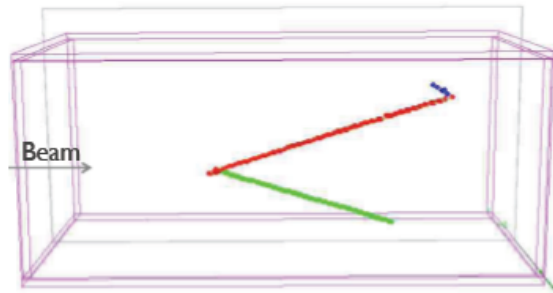


Figure 7: The neutrino event shown in Figure 6 reconstructed in three dimensions.

Figure 8–11: Figure from the ArgoNeuT draft NIM paper showing the status of 3D reconstruction

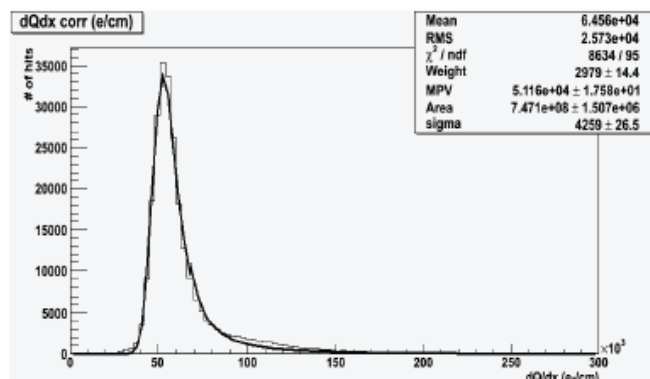


Figure 13: dQ_0/dx distribution (in ADC/cm) obtained for the through-going muon data sample having corrected for the electron lifetime and quenching effect on the ionization charge and properly taken into account the contribution due to δ -rays, as reported in the previous Section. A Landau-Gaussian fit is also reported.

Figure 8-12: Figure from the ArgoNeuT draft NIM paper showing the status of calorimetric reconstruction.

conceptual designs exist for all systems in LAr-FD. The technical activities described in this chapter are properly characterized as preliminary engineering design.

The most significant deficiency is the lack of fully-automated event reconstruction. Algorithms have been developed within the LAr community and are being successfully applied to ArgoNeuT data as well as to simulated MicroBooNE data. The algorithms have individually shown that the high efficiency and excellent background rejection capabilities of an LArTPC are achievable. The task remains to combine them into a single package.

9 Alternatives Not Selected

Alternative detector configurations and design parameters that were considered but ultimately rejected for other designs are discussed in this chapter.

9.1 Detector Configuration

9.1.1 Double Phase Readout

The European GLACIER collaboration is pursuing a novel double-phase readout detector technology that has potential advantages. In this scheme, ionization electrons are drifted upwards under the influence of an electric field towards the liquid-vapor interface. The electrons are extracted from the liquid into the vapor by an electric field of 2.5 kV/cm. The electrons then drift to two stages of Large Electron Multipliers (LEM). Electrical signals are induced on segmented electrodes on the LEM.

This method requires far fewer readout channels than our reference design, however significant R&D is required to demonstrate the viability of this technique for a large detector. This design requires very long electron-drift lengths ($\sim 20\text{m}$) in order to be cost-effective.

9.1.2 Cryostat Shape

Storage tanks can be classified by shape (upright cylinder, horizontal cylinder, rectangular parallelepiped) and means of support (self supporting, externally supported).

A horizontal cylindrical tank would require significant structural support to withstand the gravitational load. On the other hand, upright self-supporting cylindrical tanks are commonly used for surface storage of cryogenic liquids. The proposed above-ground LArTPC experiment FLARE utilized a tank of this configuration. An upright cylindrical tank is also proposed for the 100-kton GLACIER underground detector. In contrast, the 600-ton ICARUS detector is a rectangular parallelepiped.

A study was performed [32] to compare the cost for three configurations of equivalent active mass: upright cylindrical cryostat (soup can), rectangular parallelepiped externally-supported cryostat (membrane) and the rectangular parallelepiped self-supporting cryostat (modular). The study considered the cost of rock excavation, the cost of the total inventory of LAr required for a detector with a fixed active mass and a rough estimate of the cryostat cost.

The active/total mass fraction for the three configurations ranges between 70% and 74% and is therefore not a significant factor in the cost difference. The major cost factor is the volume of rock that must be excavated for these configurations. There is a significant amount of unused cavern space if an upright cylindrical tank is located within a rectangular parallelepiped cavern. The amount of unused space can be reduced by excavating a cylindrical cavern but the excavation cost would increase significantly ($\sim 30\%$).

The study results show a cost range of 10% - 20% for the different configurations. This is within the uncertainty of the estimates ($\sim 50\%$) so none of the options can be rejected purely on economic grounds given our current state of knowledge. Given sufficient resources, all configurations could be more fully developed to make a more informed decision, however any potential value would be offset by the cost of pursuing multiple design paths.

The study results also indicate that a membrane cryostat is the most cost-effective solution. It clearly maximizes the use of the excavated rock volume. A membrane cryostat is also inherently more cost-effective than a self-supporting cryostat since the hydrostatic pressure of the liquid is constrained by the cavern walls and not the cryostat walls, thereby reducing the amount of structural steel required.

9.1.3 Modular Cryostat

The modular LANNDD detector concept was considered. The main benefit of the LANNDD concept is that the cryostat is evacuable. However, reinforcing members within the cryostat would be required to withstand the vacuum load. Physics cuts around these members would reduce the fiducial mass of the detector significantly relative to the membrane-cryostat reference design. Results from the Liquid Argon Purity Demonstrator have shown that cryostat evacuation is not required to achieve excellent LAr purity.

9.2 Depth Options

9.2.1 Surface and 300L

Cosmic-ray backgrounds for accelerator-based neutrino analyses are negligible in an LArTPC located on the surface of the earth. The cosmic-ray rate on the top of an APA drift cell of size $2.5 \text{ m} \times 5 \text{ m}$ would be 2 Hz if it were located at the surface. A neutrino event is fully contained within the drift cell and the drift time is 1.5 msec, so the rate of accidental cosmic rays in a drift cell containing a neutrino event is 0.2%. The physics scope of either the surface or 300L options would be restricted to accelerator neutrinos since a competitive proton-decay search could not be performed with such a high background rate. The cost of underground construction is significantly higher than conventional surface construction so there is no clear benefit from the 300L option.

Significant cost savings (several \$100M) would result for the surface option. However, space charge build up due to the relatively high cosmic ray flux on the surface and the slow ion drift velocity may result in unacceptable distortions in the electric field using the long 3.7 m drift that is planned for the LAr-FD. This can be mitigated by reducing the drift length, which would result in a substantial increase in the number of channels and a modest reduction in the ratio of fiducial volume to total volume. These would, in turn, increase the cost of the detector, partially offsetting the cost reduction from moving to the surface.

9.2.2 800L

In addition to the reference design described in this CDR for a LAr-FD at the 4850L, a candidate design for the 800L was also developed [10]. At this depth, there is an unacceptable level of background to $p \rightarrow \nu K^+$ due to the creation of K0L by cosmic ray muons in the surrounding rock, which subsequently charge exchange within the LAr detector to create an isolated K^+ . To mitigate this effect, a large-area veto system was included in the 800L design, in order to tag muons which could generate such background and thereby eliminate candidate proton decay events in coincidence with a detected muon. Given the goal of essentially zero background for the proton decay search ([33] quotes a background rate on the order of 1 event in 30 years), the veto system would be required to be extremely efficient, and therefore essentially completely hermetic. However, due to the practical engineering considerations of how to implement a veto system extending 7 m into the rock at both top and bottom of the main detector, it proved difficult to achieve the required hermetic coverage. In addition, at the position of the proposed cavern at the 800L, the cosmic ray flux angle and energy depend on the details of the surface topography, which would require extensive simulations to be able to be assured of the efficacy of the veto system for reducing the background to the required level. Moving the detector to the 4850L reduces the cosmic ray background by three orders of magnitude, essentially eliminating these concerns.

Cosmic ray muons passing through the LAr detector itself can produce spallation products with lifetimes too long to allow signals resulting from their subsequent decay to be vetoed by observing the muon that created them. These would create backgrounds that could compromise low-energy physics searches, of which the observation of neutrinos from a distant core-collapse supernova is the prime, but not only example. The spectrum of spallation products of argon is not well known, nor is the spectrum of low-energy searches that the LBNE LAr-FD may be called upon to do, so it is difficult to quantify the effect of such background. Since the spallation backgrounds cannot be vetoed, the only way to reduce them is to move the detector to greater depth.

Based on these considerations, as well as the fact that siting LBNE at the 4850L would help enable other deep underground science in the U.S., the LBNE Collaboration Executive Committee, in its meeting in December 2011, issued a statement that “There was a very strong preference for siting the experiment at the 4850L depth.”

9.3 Cryogenics Plant

9.3.1 LAr Supply using a Temporary Air-Separation Plant

We have considered whether the provision of a temporary, dedicated air-separation plant could be justified based on the elimination of LAr losses due to boil-off during transportation, elimination of vehicle movements and the potential increase in the supply reliability. These advantages must be offset against the net capital cost of the temporary plant, the operating cost of the plant and the relative inefficiency of the small temporary plant as compared to a large commercial plant. We have been advised by LAr suppliers that this would not be cost-effective.

9.3.2 LAr Storage

It would be desirable to provide temporary storage of LAr to decouple the delivery schedule from the detector construction schedule. Ideally, temporary storage could reduce the schedule by six months since argon deliveries can occur in parallel with detector construction. This is only possible if construction of the temporary storage facility occurs concurrently with other activities.

Temporary storage would also mitigate the risk of a detector failure necessitating access to the cryostat. If a sudden detector-wide failure occurred and temporary storage were not available, the argon would be vented to the atmosphere and the \$25M investment of LAr would be lost. The cryostat filling sequence described in Chapter 6 effectively eliminates the likelihood of failure during detector commissioning. A sudden detector-wide failure occurring during

operations is highly unlikely. It is more likely, but still unlikely, that a gradual degradation of detector performance would occur over an extended period of time, allowing time for a decision to be made to construct a storage tank.

9.3.3 Common Riser for LAr and LN

The potential for the replacement of the two liquid cryogen risers (LAr and LN) with a common cryogenic riser may be considered during later design phases. The fill lines are not normally used during operation. After initial filling of the cryostat the argon line is not planned to be used unless the cryostat is to be drained. The common line would be configured in standby mode for LN service in case of unplanned outage of the refrigeration plant.

9.3.4 Common Vent Lines

A common vent system could be adopted that would be suitable for both nitrogen and argon. It would need to be sized for the peak combined flow rate. The probability of either argon or nitrogen being vented under normal operations is low, but a Simultaneous Operations Study (SIMOPS) would be required to demonstrate that cross-contamination could not occur during commissioning, or normal or emergency operations. A greater level of design detail is required to support this study than has been possible to-date and therefore separate vents have been included at this stage.

9.4 Cryostat Insulation

The original LANNDD modular cryostat included the use of a vacuum-insulated space between the inner and outer tanks, resulting in a significant decrease in the required refrigeration load. This benefit would need to be weighed against the added capital cost of a vessel that can withstand the vacuum load. Arup eliminated the vacuum-insulated option during the screening process, citing the benefit of not having an outer tank designed for a full vacuum load.

The use of a vacuum-jacketed cryostat was rejected. The most serious concern is an accident scenario that would result in a significant leak in the inner vessel. The outer tank and the argon venting system would need to be designed to cope with the large volume of released gas.

9.5 TPC

9.5.1 TPC Configuration

9.5.1.1 Reference Design 1a

The first proposed Reference Design, 1a, relied on a minimal extrapolation of the MicroBooNE TPC module design. The detector would be installed in a cavern with drive-in access and would consist of a rectangular stack of TPC modules that had been constructed above-ground.

A modest extrapolation of the MicroBooNE module design parameters does not introduce large cost uncertainty if done properly. The extrapolations are in dimensions that do not challenge the limits of the technology. Rather, this detector configuration suffers from a poor fiducial/total mass fraction. Only 56% of the LAr in the detector would be useful for physics and the cost is ~ 3 times the cost of the selected reference design.

9.5.1.2 Reference Design 2a

Reference Design 2a is similar to the selected reference design. The major difference is the use of room-temperature, accessible electronics. The concept is shown in Figure 9-1.

Short low-impedance cables route wire signals from the Anode Plane Assemblies (APAs) to a cold feedthrough which is kept at ~ 120 K. The readout electronics are located within the feedthrough. The benefit of this design is that the electronics boards can be replaced without removing LAr from the detector.

There are a number of disadvantages to this concept, however. The cable lengths must be kept to ~ 1 m, the wire spacing must be increased to 5 mm and the wire angle reduced from 45^{deg} to 30^{deg} in order to achieve a marginally acceptable signal-to-noise ratio. As a result, the track resolution in the vertical direction is ~ 1 cm. A large number of feedthroughs are required in order to keep the cable lengths short. One feedthrough containing 576 electronics channels is required for every 53 cm along the top of each APA. Each feedthrough would be $34 \text{ cm} \times 22 \text{ cm}$ in cross section and would be ~ 2 m long.

9.5.2 Wire Spacing

The distinguishing feature of the LArTPC is the ability to distinguish one MIP electrons from 2 MIP electron-positron pairs close to the interaction vertex. A wire spacing smaller

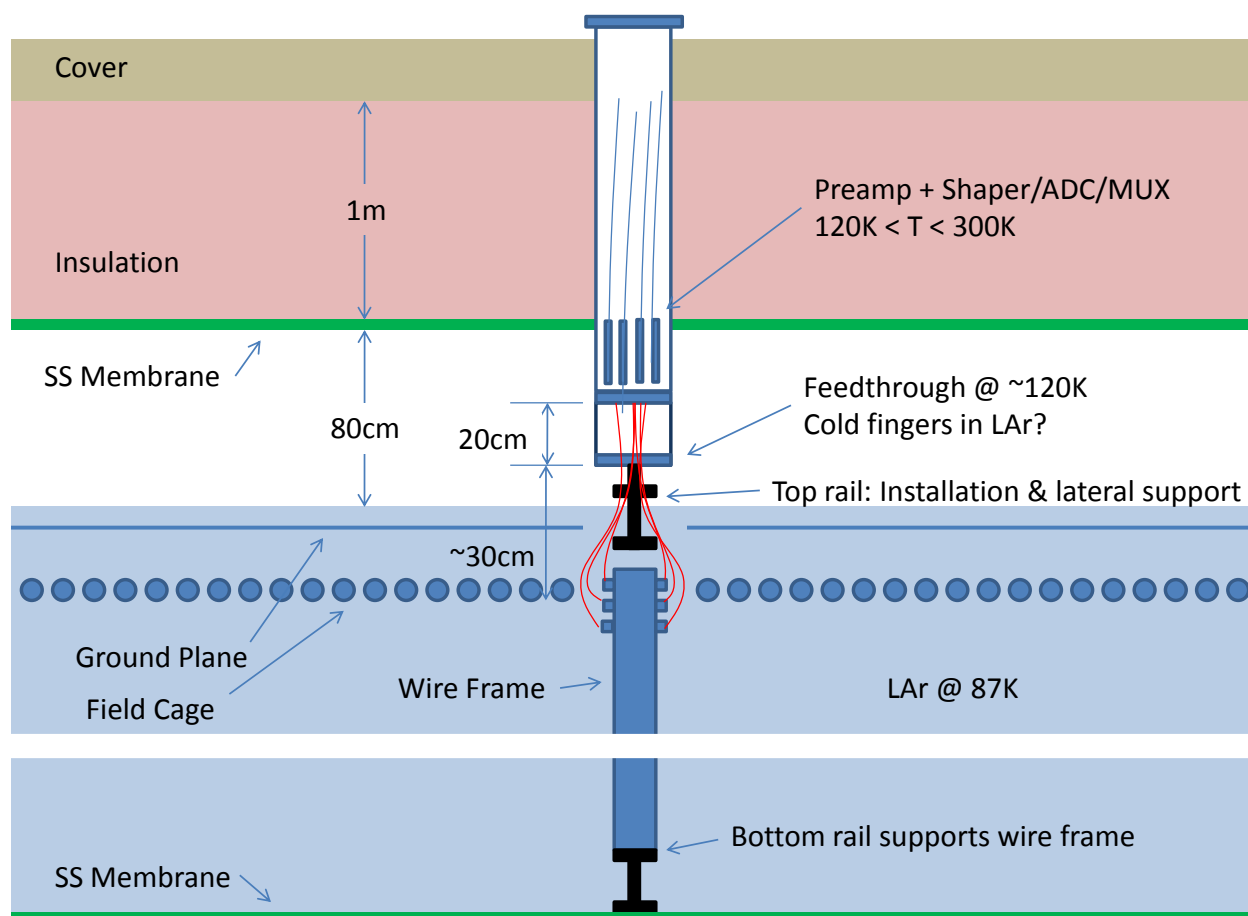


Figure 9-1: Reference Design 2a, showing cable routing from an Anode Plane Assembly to a cold feedthrough. The warm readout electronics are located within the feedthrough.

than the design (5 mm) would have smaller signal-to-noise ratio (S/N) commensurate with the wire spacing. The design S/N ratio could be restored by decreasing the noise. The most direct means of accomplishing this would be to reduce the wire length, thereby increasing the number of electronics channels. Reducing the wire spacing below 5 mm will only offer minimal benefit since the background from $NC\pi^0$ events is already quite small.

9.5.3 Number of Wire Planes

The reference design includes three instrumented wire planes and one un-instrumented grid plane. Reducing the number of instrumented planes to two would reduce the detector cost by \sim \$3M with a negligible loss of ν_e identification efficiency. This would also reduce the readout redundancy, however, and potentially affect the long-term reliability of detector operations. We do not consider this a credible value-engineering option.

The un-instrumented grid plane could be eliminated, saving \sim \$1M. The grid plane is used to create equal signal levels in both induction planes. The signal level in the first induction plane would be reduced by \sim 2 times if the grid plane were eliminated. The grid plane also provides electrostatic discharge protection for the instrumented wires. We do not consider this a credible value-engineering option.

9.5.4 Drift Length

Selection of the design drift length is highly coupled with the expected LAr purity, the wire spacing and the required ν_e identification efficiency and background rejection. The ICARUS detector is currently operating with a drift electron lifetime of 6 to 7 ms. We expect the LAr purity in LAr-FD to be equal or superior to ICARUS. Increasing the drift distance beyond the reference design of 3.7 m would result in significant cost savings. The 5 mm wire spacing is well matched to this drift distance; it is roughly twice the transverse diffusion RMS. As an example, if the drift distance were increased to 5 m, the wire spacing could be increased to \sim 6 mm. The minimum signal-to-noise ratio would suffer a minor decrease from 36:1 to 30:1 if the electron lifetime was 6 ms. The effect of such a change on physics analyses has not yet been explored.

9.6 DAQ Cable Routing

The concept of routing raw signals from all wires out of the cryostat, as ICARUS does, was considered. The large data rate would require a huge number of cables and feedthroughs with little benefit since the vast majority of the sampled wire signals have no information. The large cable plant would be a major contributor to LAr impurities.

9.7 Installation & Commissioning

The initial Installation and Commissioning concept was to support the TPC on the floor of the cryostat. Cross-bracing in both directions would be required to ensure mechanical stability and would compromise the TPC design. Hard points in the cryogenic insulation would be needed which would likely require design modifications to the standard vendor-supplied membrane-cryostat insulation system.

Several alternatives for accessing the top of the detector during installation were considered. A scissors lift was considered and rejected due to concerns that the lift could sway and damage detector components. Consideration was given to a moveable platform that would

traverse the top of the detector on rails, or alternatively, using a temporary catwalk. These systems would provide access to the top of the detector but not at intermediate heights.

Several installation sequences were considered; row-wise versus column-wise installation of APAs and CPAs. The current installation sequence was deemed superior in that it minimizes work activities in previously installed sections of the detector. This reduces the risk of damage during installation.

9.8 Photon Detection

Most LArTPCs use TPB-coated PMTs to detect scintillation light. Light emitted more than a few meters from a PMT is diffused by Rayleigh scattering ($\lambda \approx 90$ ccm), so PMTs would need to be placed between drift cells. This configuration is not compatible with the APA concept. Conceptually, each interior cathode plane can be replaced with two cathode planes separated by sufficient space for an array of PMTs. This would increase the width of the detector and the cavern by ~ 1 m and result in a lower fiducial mass.

A variety of options were considered for the wavelength-shifting scheme and the light guides. These options suffer from lower light-collection efficiency and result in higher cost for the same performance.

9.9 LAr1 Prototype

A variety of alternatives to the LAr1 prototype have been considered, e.g. constructing a series of smaller prototypes to validate the integration of the detector systems. However, smaller prototypes may not catch some unknown issues that this technology, in its current state of maturity, may present.

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