### Assessments of ATMS and SSMIS Data for NWP Applications

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JCSDA Seminar, June 20, 2012



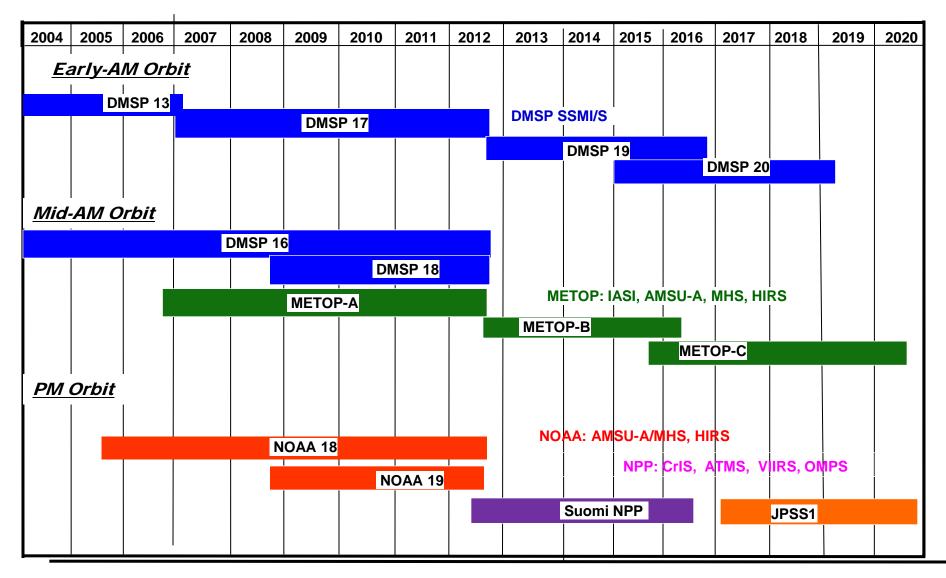
- Current Observing System for NWP
  - Data used in GFS
  - Impacts from uses of satellite data
  - Optimal global observing system
- Microwave Sounding Systems Conical vs Cross-Track
  - Pros and Cons
  - Convertibility issue of TDR to SDR
  - Changing practices in forward model

#### Assessments of Instrument Performance wrt NWP

- Cloud algorithm for NWP control controls
- ATMS sounding channels
- SSMIS sounding Channels

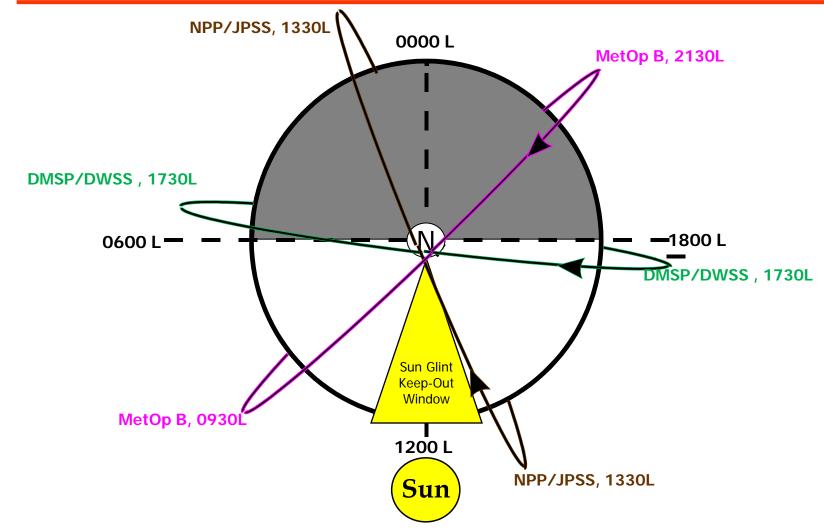
#### Summary and Conclusions

#### **US Current Polar Satellite System Program**



# NORTH TO FORMER

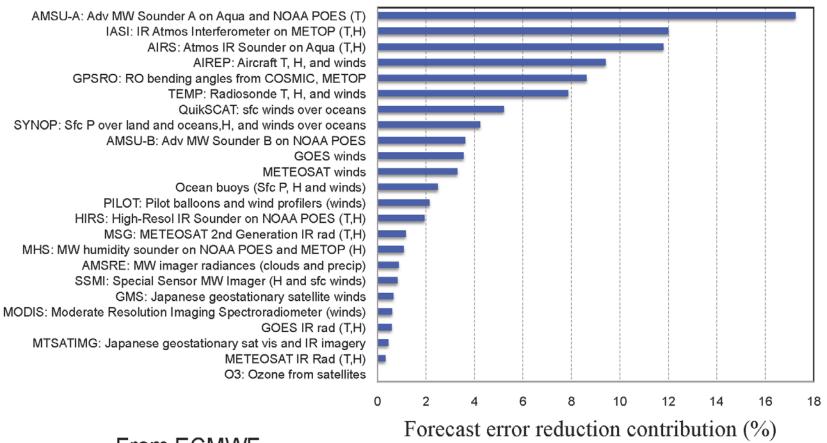
#### **JPSS Era System**



JPSS will Extend and Replace the Heritage NOAA Satellites that are Crucial to National Operational Capabilities of both NOAA and DoD



#### Impacts of Satellite/Conventional Data on Global Forecast Model

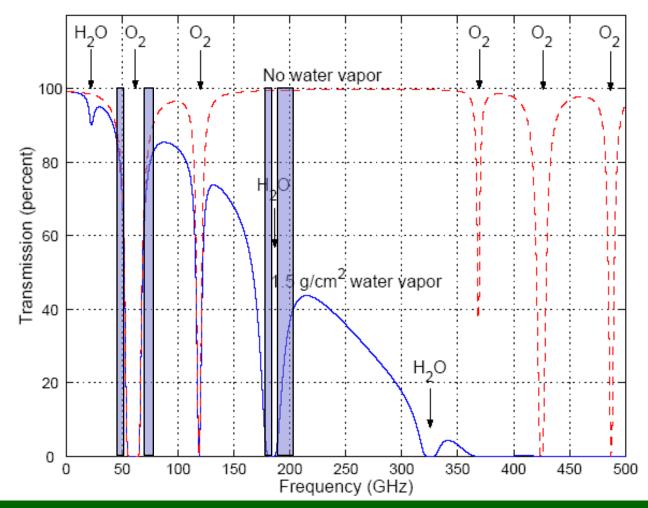


From ECMWF

Impacts from assimilation of satellite data on NWP forecasts are larger than those from conventional data. Much of the impacts is attributed to use of microwave/IR/GPSRO sounding data



#### Atmospheric Transmission at Microwave Wavelengths



The frequency dependence of atmospheric absorption allows different altitudes to be sensed by spacing channels along different absorption lines.

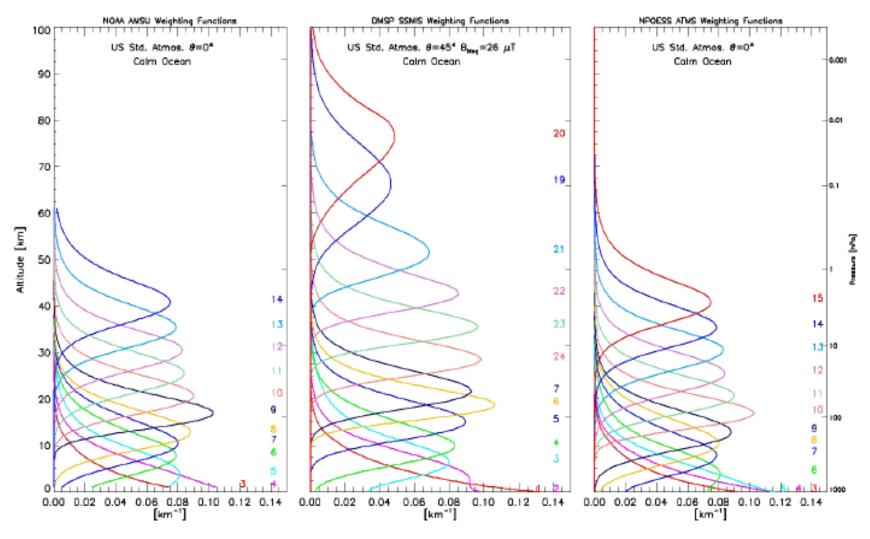


#### Microwave Temperature Sounding Vertical Resolution



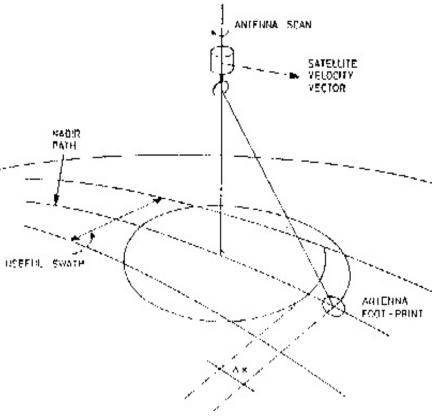
SSMIS

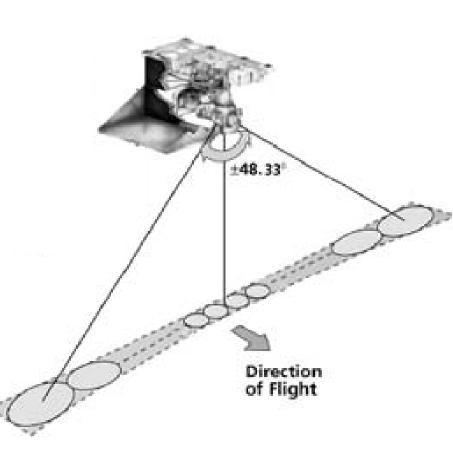






#### **Conical vs Cross Track Sounding**



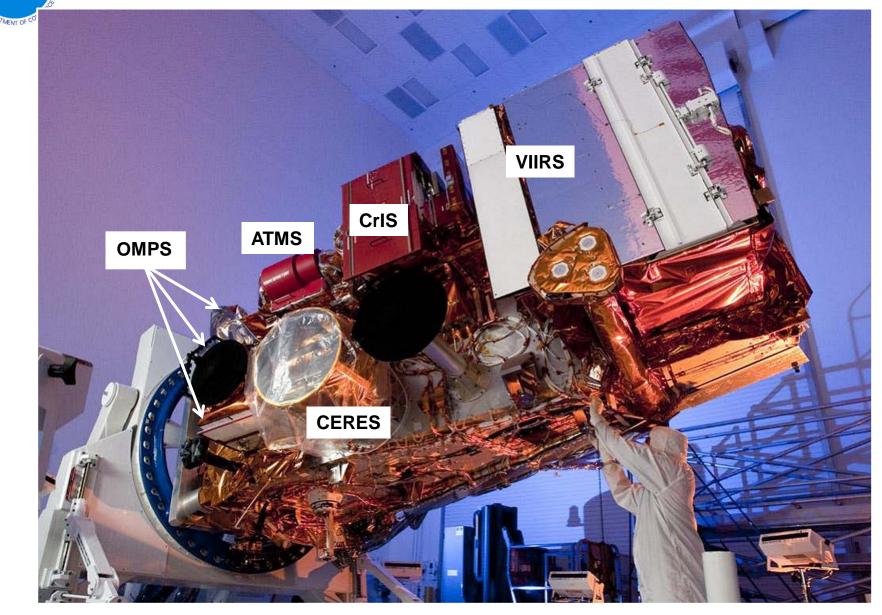


Large scan swath width (no orbit gap)
Same resolution for all frequencies
Mixing pol as scan from nadir to limb
Res varies with scan angle

Narrow scan swath width with orbit gap
FOV size is the same for all positions but varies with frequencies
Same pol for all scan positions

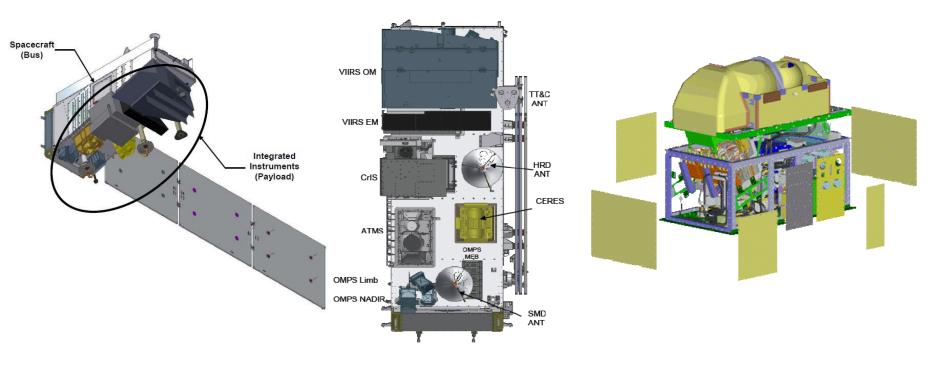
#### **Suomi NPP Spacecraft and Payloads**

NOAA





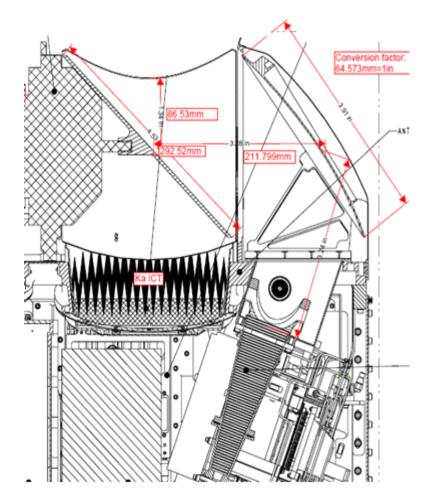
#### **ATMS after Assembly**

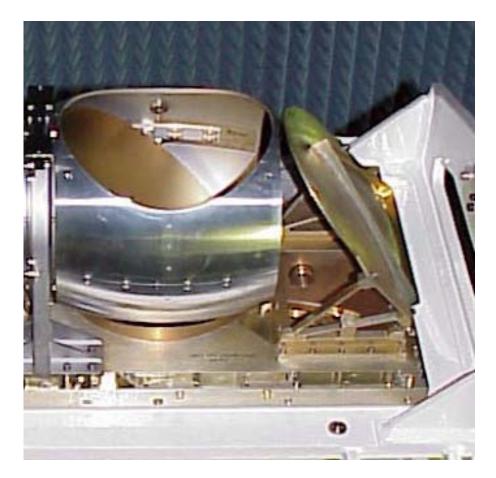


- 70x40x60 cm
  - 110 W
  - 85 kg
  - 8 year life



#### **ATMS Scanning Assembly**





#### **Courtesy of NGES**

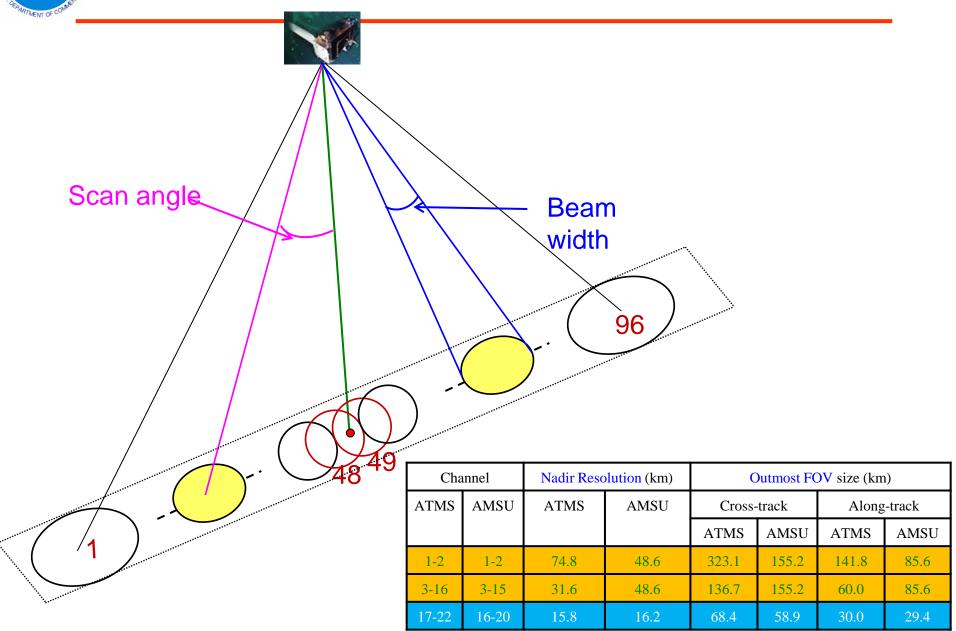
### Channel Characteristics of ATMS and AMSL

NOAA,

Channel		Frequency (GHz)		ΝΕΔΤ (Κ)		Beam width (°)		Peak WF (hPa)	
ATMS	AMSU	ATMS	AMSU	ATMS	AMSU	ATMS	AMSU	ATMS	AMSU
1		23.8		0.50	0.30	5.2	3.3	Surface	
2		31.4	31.399	0.60	0.30	5.2	3.3	Surface	
3		50.3	50.299	0.70	0.40	2.2	3.3	Surface	
4		51.76		0.50		2.2		Surface	
5	4	52.8		0.50	0.25	2.2	3.3	1000	
6	5	53.596±0.115		0.50	0.25	2.2	3.3	700	
7	6	54.4		0.50	0.25	2.2	3.3	400	
8	7	54.94		0.50	0.25	2.2	3.3	270	
9	8	55.5		0.50	0.25	2.2	3.3	180	
10	9	57.29		0.75	0.25	2.2	3.3	90	
11	10	$57.29 \pm 0.217$		1.00	0.40	2.2	3.3	50	
12	11	$57.29 \pm 0.322 \pm 0.048$		1.00	0.40	2.2	3.3	25	
13	12	$57.29 \pm 0.322 \pm 0.022$		1.25	0.60	2.2	3.3	12	
14	13	$57.29 \pm 0.322 \pm 0.010$		2.20	0.80	2.2	3.3	5	
15	14	$57.29 \pm 0.322 \pm 0.0045$		3.60	1.20	2.2	3.3	2	
16	15	88.2	89.0	0.30	0.50	2.2	3.3	Surface	
17	16	165.5	89.0	0.60	0.84	1.1	1.1	1000	Surface
18	17	183.31±7.0	157.0	0.80	0.84	1.1	1.1	800	Surface
19	18	183.31±4.5	$183.31 \pm 1.0$	0.80	0.60	1.1	1.1	700	400
20	19	183.31±3.0		0.80	0.70	1.1	1.1	600	
21	20	183.31±1.8	183.31±7.0	0.80	1.06	1.1	1.1	500	800

### **ATMS Scan Angle and Beam Width**

**ND ATMOSPH** 





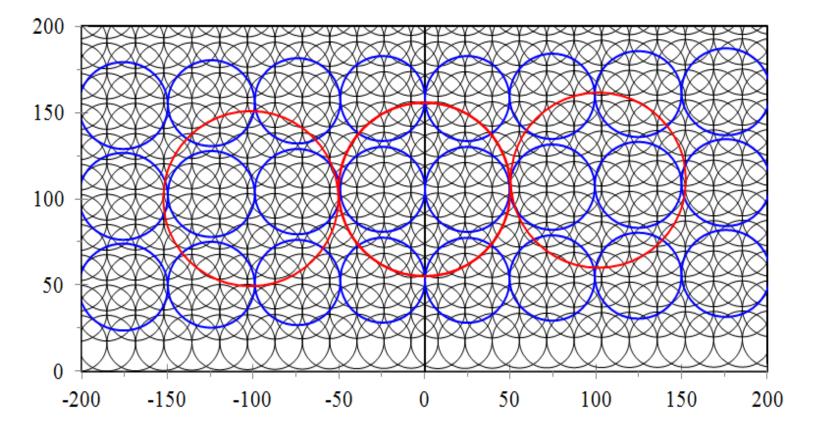
Beamwidth (degrees)ATMSAMSU/MHS23/31 GHz5.23.3			Spatial sampling				
	ATMS	AMSU/MHS		ATMS	AMSU/MHS		
23/31 GHz	5.2	3.3	23/31 GHz	1.11	3.33		
50-60 GHz	2.2	3.3	50-60 GHz	1.11	3.33		
89-GHz	2.2	1.1	89-GHz	1.11	1.11		
160-183	1.1	1.1	160-183 GHz	1.11	1.11		
GHz							
			Swath (km)	~2600	~2200		

ATMS scan period: 8/3 sec; AMSU-A scan period: 8 sec

#### Impacts of ATMS Spatial Re-sampling on NWP O-B

ATMS Field of View Size for the beam width of 2.2° – black line

ATMS Resample to the Field of View Size for the beam width of 3.3°- blue line



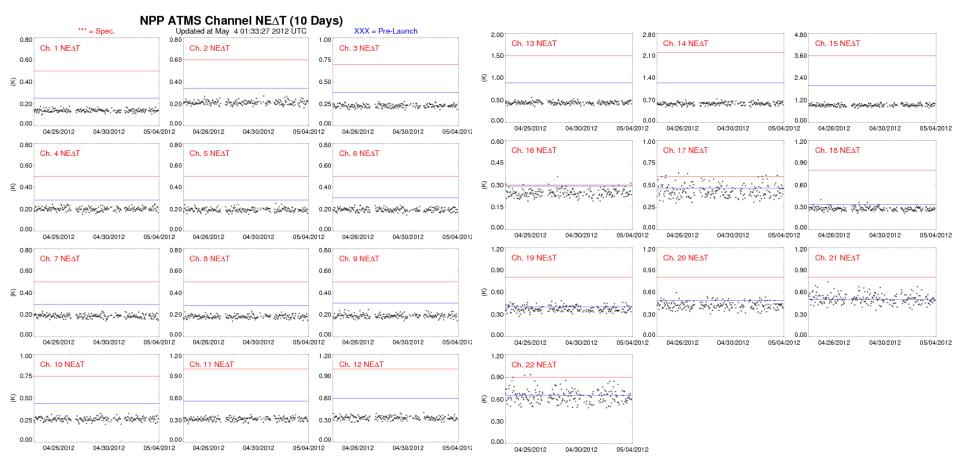


### **ATMS CalVal Status**

- Beta version of ATMS TDR data was declared on February 22, 2012
  - ATMS instrument noises (NEDT) are characterized and all channels meet the requirements
  - Onboard calibrators are functioning normally and stable, providing good quality data for TDR calibration
  - Cold space calibration profile one is selected for ATMS calibration
- Provisional version of ATMS TDR has been achieved in May, 2012
  - Accuracy of ATMS sounding channels is characterized with GPSRO data and within 1.0-2.0 K
  - Effects of ATMS spectral response function are investigated and the improvements for forward modeling are significant (up to 0.2 to 0.3K)



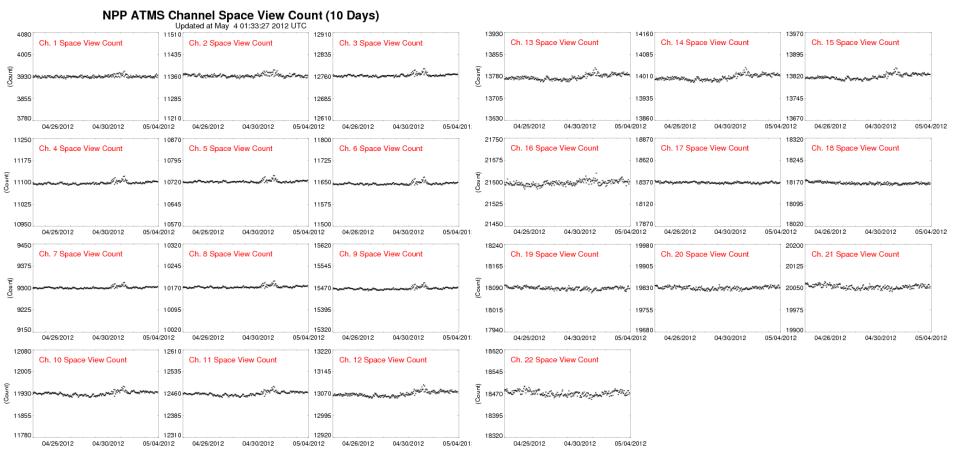
## ATMS Channel NEAT



NPP ATMS on-orbit NE $\Delta$ T (black) are within specifications (red)



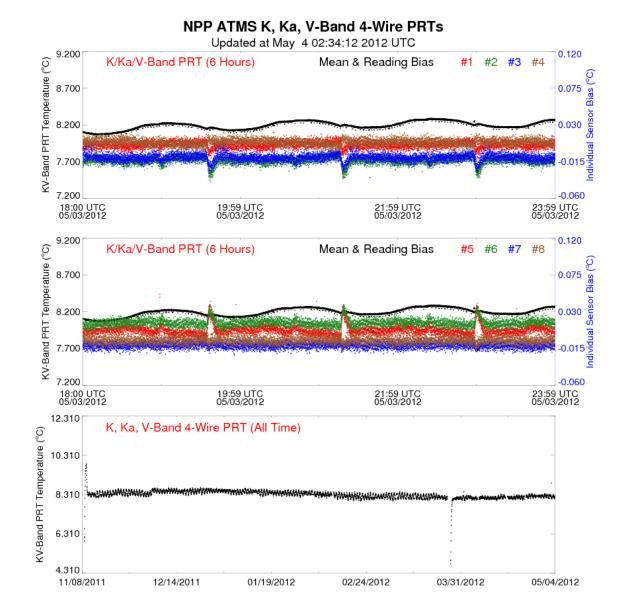
# **ATMS Space View Count**



- NPP ATMS space view calibration counts are stable
- NPP ATMS warm load calibration counts and gain are also stable (not shown)

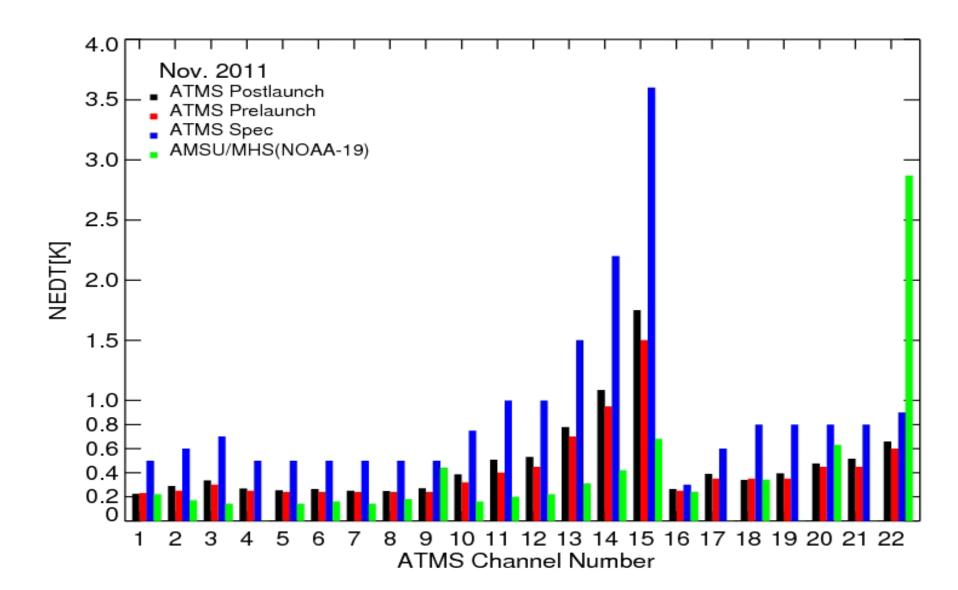
# K/Ka Band 4-Wire PRT Temperature

- K/Ka band 4-wire warm load PRT temperature (8 readings) is stable
- W/G band 4-wire warm load PRT temperature (7 readings) is stable too (not shown)





#### ATMS Noise Equivalent Differential Temperature (NEDT)





#### Assessments of ATMS Performance Using NWP O-B

#### Approach 1: no antenna models applied for observations or simulations

Observation (O): Antenna brightness temperature (TDR) from satellite observations

Background (B): Simulated brightness temperature (SDR) from NWP forecast fields or other profiles

#### Approach 2: Antenna models for converting TDR to SDR

Observation (O): Sensor brightness temperature (SDR) from satellite observations.

Background (B): Simulated brightness temperature (SDR) from NWP forecast fields or other profiles

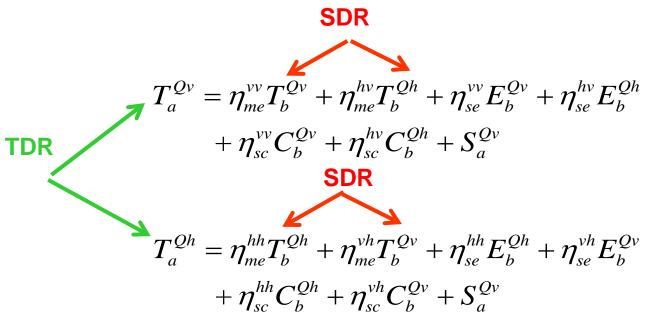
Approach 3 : Antenna model applied for converting simulated SDR to TDR

Observation (O): Antenna brightness temperature (TDR) from satellite observations

Background (B): Simulated antenna brightness temperature (TDR) from NWP forecast fields or other profiles



#### Antenna/Sensor Brightness Temperature for Cross-Track Scanning Instrument



The first two terms are Quasi-V and Quasi-H brightness temperature from earth in the main beam (main lobe earth), the 3rd/4th terms are those from the side-lobe earth, the 5/6th terms are side-lobe cold space,, the last two are near-field satellite radiation. Specifically

$$T_b^{Qv} = T_b^v \cos^2 \theta + T_b^h \sin^2 \theta \qquad T_b^{Qh} = T_b^v \sin^2 \theta + T_b^h \cos^2 \theta$$

Under a polarized earth scene, the cross-polarization term can result in large errors in computing SDR from TDR data if the antenna has a significant spill-over effect and the cross-polarization term is neglected.



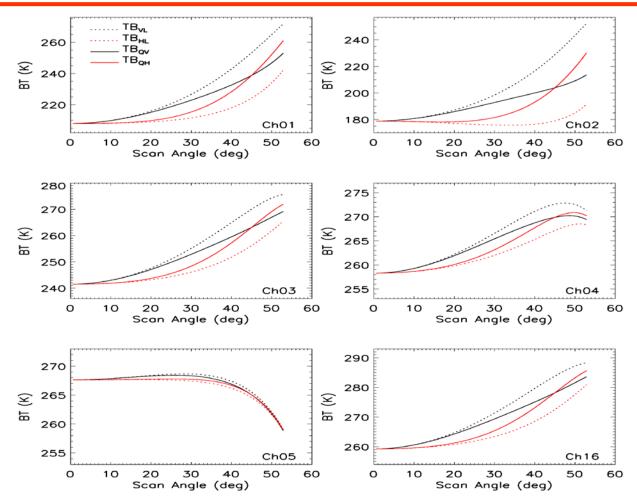
- Need to correct side-lobe radiation from far-field earth and near-field satellites
- For a sensor with a significant cross-polarization spillover, an inversion from TDR to SDR is problematic if a single polarization measurement is provided
- For un-polarized surface and atmospheric conditions, the inversion from TDR to SDR is possible with a single polarization measurement.



#### **ATMS Antenna Beam Efficiency**

Frequency	$\eta^{\scriptscriptstyle pp}_{\scriptscriptstyle me}$ (%)			$\eta^{\scriptscriptstyle pq}_{\scriptscriptstyle me}$ (%)			$\eta_{se}^{pp} + \eta_{sc}^{pp} + \eta_{ss}^{pp}$ (%)		
(GHz)	B1	B48	B96	B01	B48	B96	B01	B48	B96
23.8	99.48	99.61	99.53	0.52	0.39	0.46	0.003	0.0002	0.0025
31.4	99.59	99.60	99.60	0.40	0.40	0.39	0.003	0.0003	0.0024
50.3	99.43	99.39	99.56	0.57	0.61	0.44	0.001	0.0006	0.0008
51.8	99.45	99.47	99.73	0.55	0.53	0.27	0.001	0.0004	0.0007
52.8	99.48	99.46	99.36	0.51	0.54	0.64	0.001	0.0004	0.0010
53.6	99.49	99.43	99.31	0.51	0.57	0.68	0.001	0.0004	0.0008
54.4	99.51	99.51	99.55	0.49	0.49	0.44	0.001	0.0006	0.0006
54.9	99.48	99.49	99.21	0.51	0.51	0.78	0.001	0.0004	0.0007
55.5	99.50	99.52	99.54	0.50	0.48	0.46	0.001	0.0004	0.0007
57.3	99.48	99.49	99.48	0.52	0.51	0.52	0.001	0.0006	0.0007
88.2	97.73	97.70	97.92	2.27	2.30	2.07	0.002	0.0012	0.0035
166	98.00	97.77	96.92	1.98	2.21	3.06	0.013	0.0147	0.0085
176	97.92	97.77	96.17	2.07	2.21	3.81	0.009	0.0115	0.0075
183	97.69	98.48	98.86	2.29	1.50	1.12	0.009	0.0108	0.0083
190	98.23	97.94	97.80	1.75	2.03	2.18	0.011	0.0138	0.0111

#### Simulated ATMS Brightness Temperatures over Ocean Conditions



For a scan angle ranging from 15 to 45 degrees, ATMS brightness temperatures at ch1, 2, 3,4 and 16 are highly polarized over ocean conditions.



#### ATMS O-B Using Approach One: (Observation TDR- Simulation SDR)

#### For Quasi-V:

$$T_{a}^{Qv} - T_{b}^{Qv} = (\eta_{me}^{vv} - 1)T_{b}^{Qv} + \eta_{me}^{hv}T_{b}^{Qh} + \eta_{se}^{vv}E_{b}^{Qv} + \eta_{se}^{hv}E_{b}^{Qh} + \eta_{sc}^{vv}C_{b}^{Qv} + \eta_{sc}^{hv}C_{b}^{Qh} + S_{a}^{Qv}$$

#### For Quasi-H:

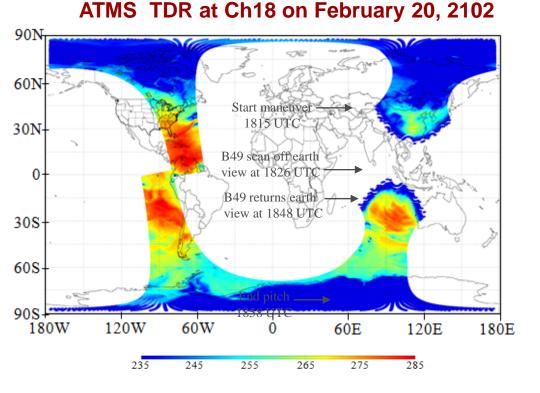
$$\begin{split} T_{a}^{Qh} - T_{b}^{Qh} &= (\eta_{me}^{hh} - 1)T_{b}^{Qh} + \eta_{me}^{vh}T_{b}^{Qv} + \eta_{se}^{hh}E_{b}^{Qh} + \eta_{se}^{vh}E_{b}^{Qv} \\ &+ \eta_{sc}^{hh}C_{b}^{Qh} + \eta_{sc}^{vh}C_{b}^{Qv} + S_{a}^{Qv} \end{split}$$

For ATMS temperature sounding channels, the earth scene is un-polarized and the bias is mainly driven by the near-field side lobe effect after the farfield side-lobe effects are neglected. Fortunately, the near field side-lobe contribution is being characterized by means of its pitch-over measurements:

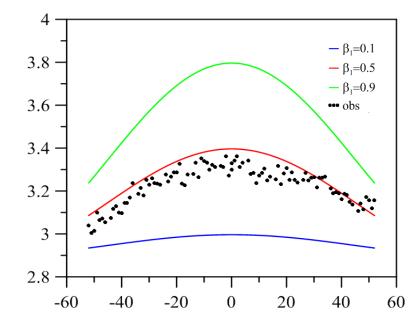
$$T_a^{Qh} - T_b^{Qh} \approx S_a^{Qh}$$



#### **ATMS Pitch-Over Maneuver Data**



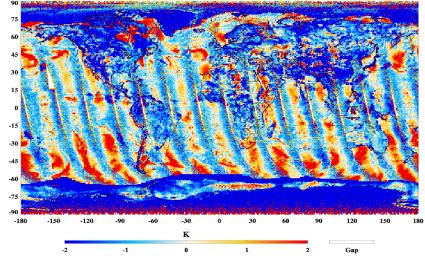
#### **ATMS Pitch-Over TDR at Ch4**



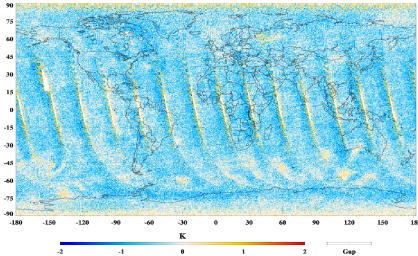


### ATMS O (TDR) - B (SDR) wrt ECMWF

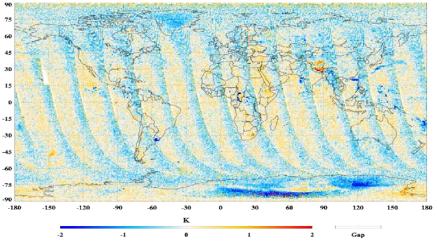
ATMS Clear Sky Over Ocean Ch.5 52.8 GHz Scan Date: 2012-06-13



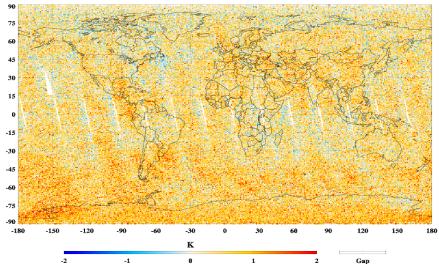
ATMS Clear Sky Over Ocean Ch.9 55.50 GHz Scan Date: 2012-06-13



ATMS Clear Sky Over Ocean Ch.7 54.40 GHz Scan Date: 2012-06-13

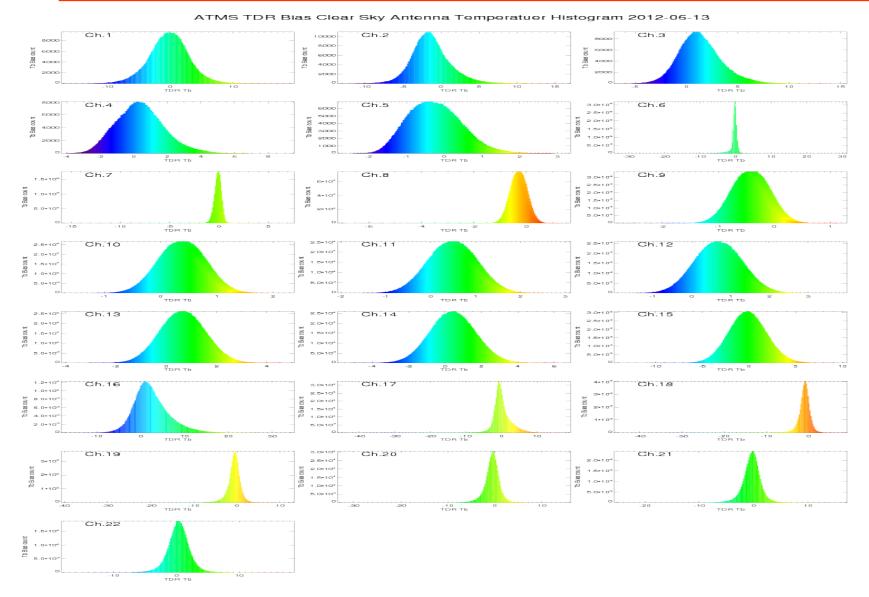


ATMS Clear Sky Over Ocean Ch.11 57.29034 +/- 0.217 GHz Scan Date: 2012-06-13





#### ATMS O-B Distribution for Clear Sky Conditions





### **Cloud Liquid Water Algorithm**

$$L = a_0 \mu [\ln(T_s - T_{b31}) - a_1 \ln(T_s - T_{b23}) - a_2]$$

 $\mu$  — cosine of satellite zenith angle

 $T_{\rm s}$  — surface temperature

 $T_{b23}$ ,  $T_{b31}$  — brightness temperature at 23.8 and 31.4 GHz

$$a_{0} = -0.5\kappa_{v23}/(\kappa_{v23}\kappa_{l31} - \kappa_{v31}\kappa_{l23}),$$
  

$$a_{1} = \kappa_{v31}/\kappa_{v23},$$
  

$$a_{2} = -2.0(\tau_{o31} - a_{1}\tau_{o23})/\mu + (1.0 - a_{1})\ln(T_{s}) + \ln(1.0 - \varepsilon_{31}) - a_{1}\ln(1.0 - \varepsilon_{23})$$

 $K_{\nu}$  — water vapor absorption coefficients

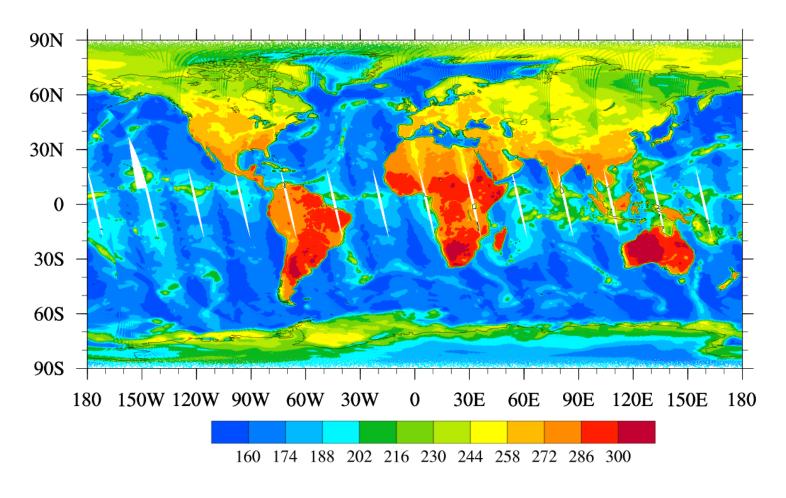
 $\kappa_l$  — cloud liquid water mass absorption coefficients

$$au_o$$
 — optical thickness

Weng, F., L. Zhao, R. R. Ferraro, G. Poe, X. Li, and N. C. Grody, Advanced microwave sounding unit cloud and precipitation algorithms, Radio Sci., 38(4), 8086, 2003.



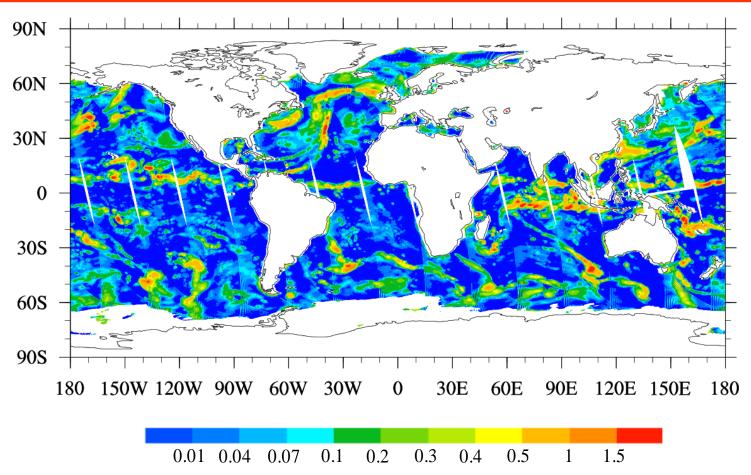
#### **ATMS Brightness Temperature at CH 2**

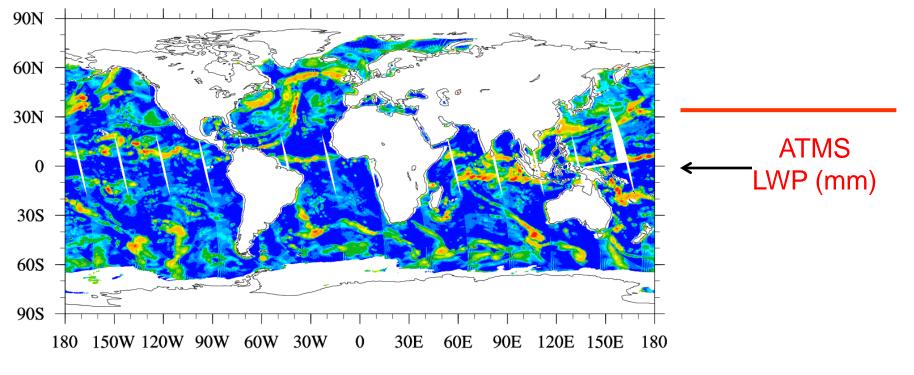


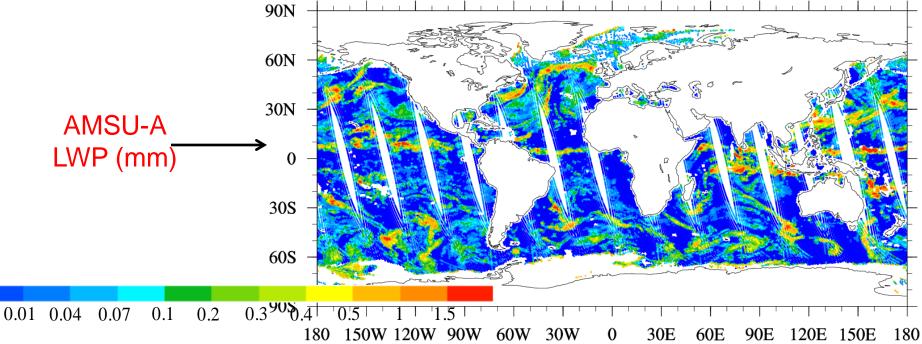
December 20, 2011

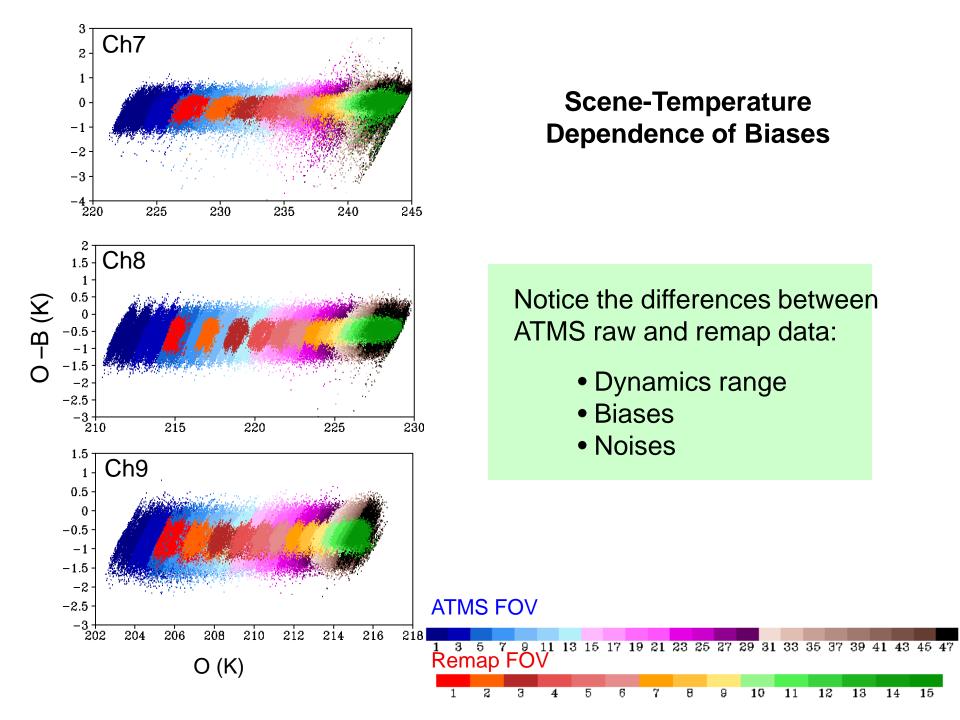


#### **ATMS Cloud Liquid Water**

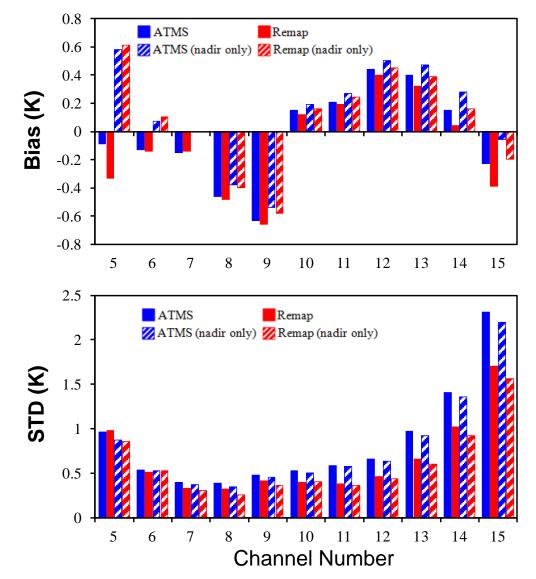




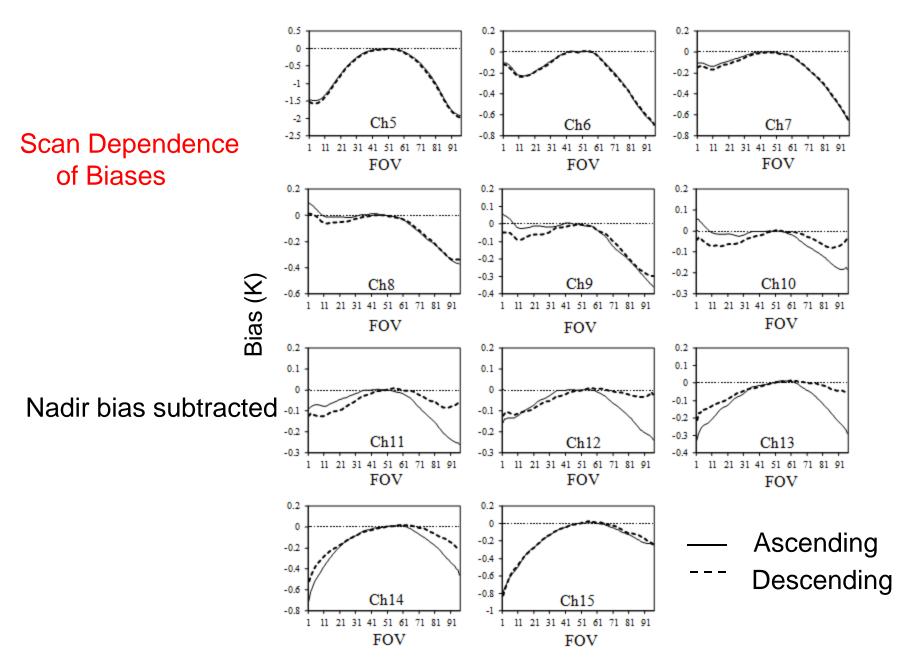




#### **Global Biases and Standard Deviations**

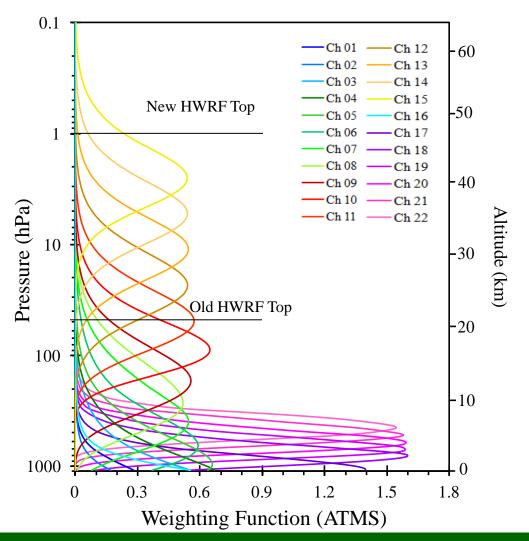


Within 60S-60N, clear-sky, ocean only, 20-27 December 2011



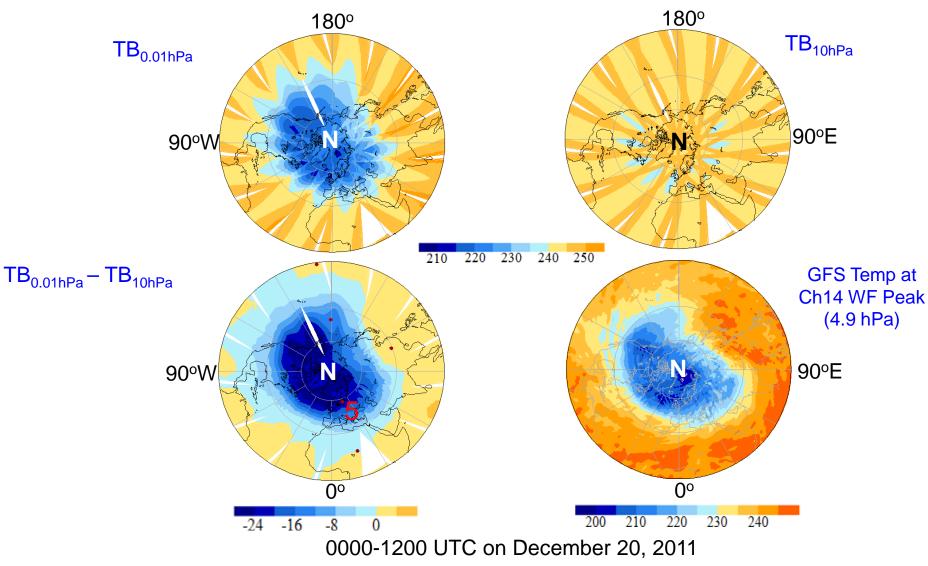
# Effects of NWP Model Top Heights on O-B

MENT C

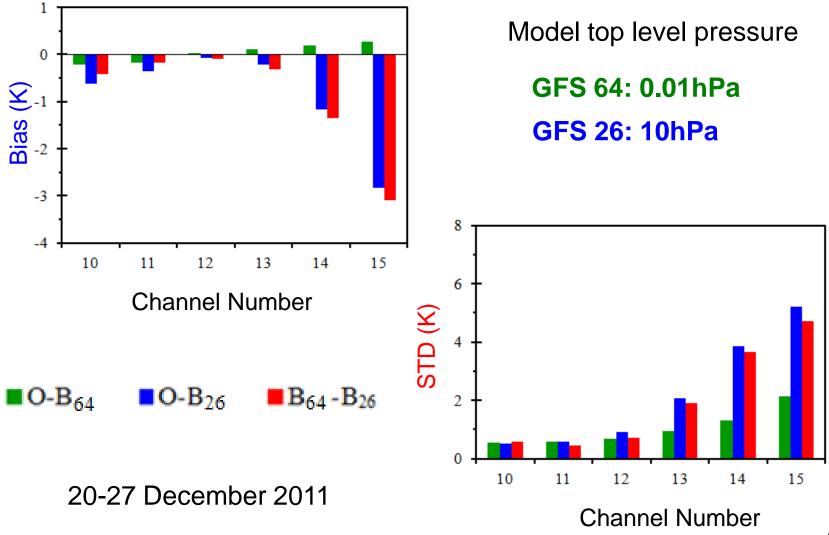


If an NWP model top (such as HWRF) is run at 50 hPa, many of satellite upper level sounding channels can not be assimilated due to large O-B

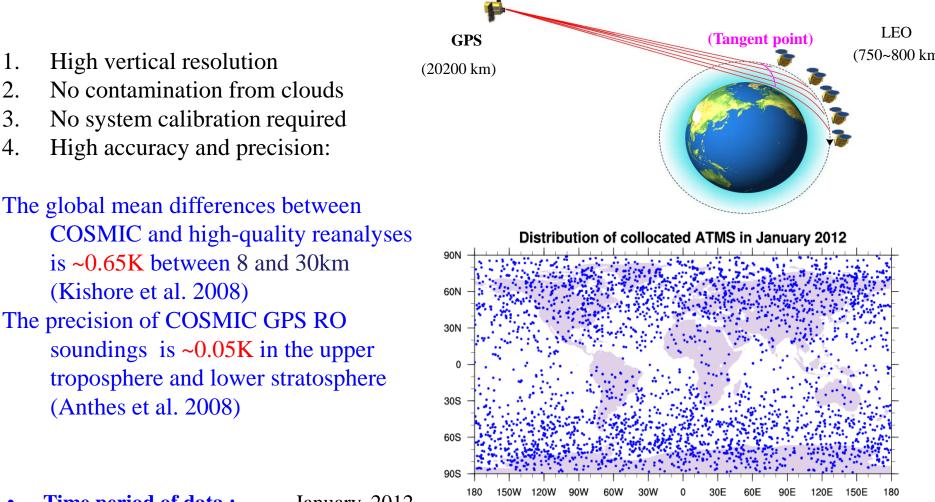
## ATMS Simulated Tb at Ch14 and Temperature at 4.9 hPa



# Comparison of GFS 64-level and 26-level Data



# Uses of GPSRO Data for ATMS Bias Characterization



**Time period of data :** January, 2012

1.

2.

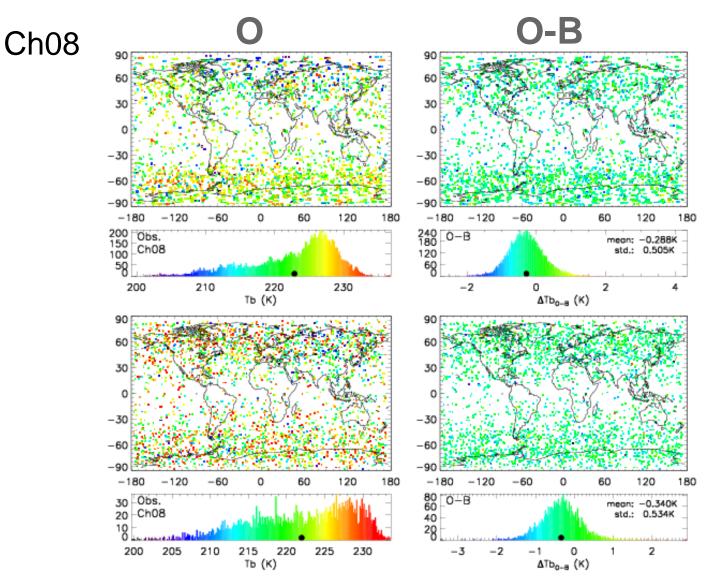
3.

4.

**Collocation of ATMS and COSMIC data:** Time difference < 0.5 hour, Spatial distance < 30 km, GPS geolocation at 10km altitude is used for spatial collocation)



# **ATMS Bias Distribution Using GPSRO**

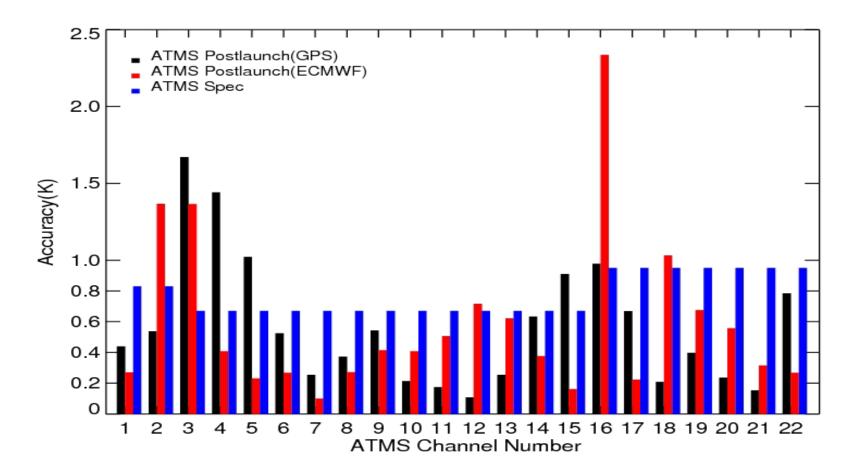


December, 2011

January, 2012



# Post-Launch ATMS Calibration Accuracy Using COSMIC Data

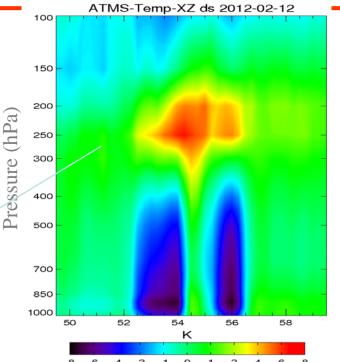


On-orbit ATMS calibration accuracy is quantified using GPSRO data as input to LBLRT model and the results are better than specification for most of sounding channels



# Suomi NPP ATMS Resolves Warm Core Structure of Tropical Cyclones





A warm core of 8K ore more at 250 hPa from ATMS indicated a category 4 to 5 hurricane intensity

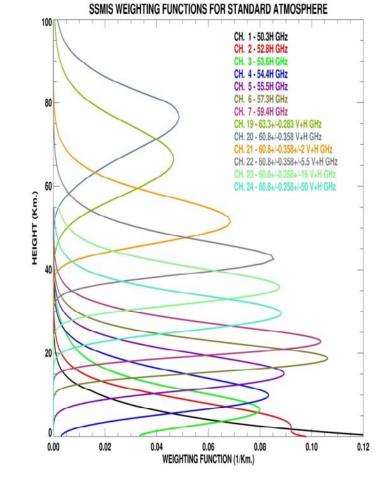




# **SSMIS Channel Characteristics**

### **SSMIS Sensor Characteristics**

Char	nnel Center	Passband	Freq. Pol	NEDT	S
	Freq.(GHz)	(MHz)	Stab.(MHz)	(Max)(K)	In
1	50.3	400	10 H	0.4	
2	52.8	400	10 H	0.4	
3	53.596	400	10 H	0.4	
4	54.4	400	10 H	0.4	
5	55.5	400	10 H	0.4	
6	57.29	350	10 *	0.5	
7	59.4	250	10 *	0.6	
8	150	1500	200 H	0.88	
9	183.31+/-6.6	1500	200 H	1.2	
10	183.31+/-3	1000	200 H	1.0	
11	183.31+/-1	500	200 H	1.25	
12	19.35	400	75 H	0.7	
13	19.35	400	75 V	0.7	
14	22.235	400	75 V	0.7	
15	37	1500	75 H	0.5	
16	37	1500	75 V	0.5	
17	91.655	3000	100 V	0.9	
18	91.655	3000	100 H	0.9	
19	63.283248	3	0.08 V + H	2.4	
	+/-0.285271				
20	60.792668	3	0.08 V + H	2.4	
	+/-0.357892				
21	60.792668	6	0.08 V + H	1.8	
	+/-0.357892				
	+/-0.002				
22	60.792668	12	0.12 V + H	1.0	
	+/-0.357892				
	+/-0.006				
23	60.792668	32	0.34 V + H	0.6	
	+/-0.357892				
	+/-0.016				
24	60.792668	120	0.84 V + H	0.7	
	+/-0.357892				
	+/-0.050				



Notes:

1. The sampling interval refers to the along scan direction and is based on nominal spaceci

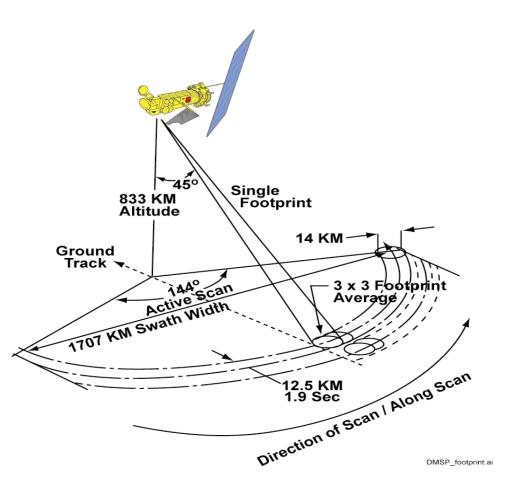
3. \* = These channels are not polarization dependent.

<sup>2.</sup> The radiometer integration time is 4.20msec for a single 12.5km sample interval.

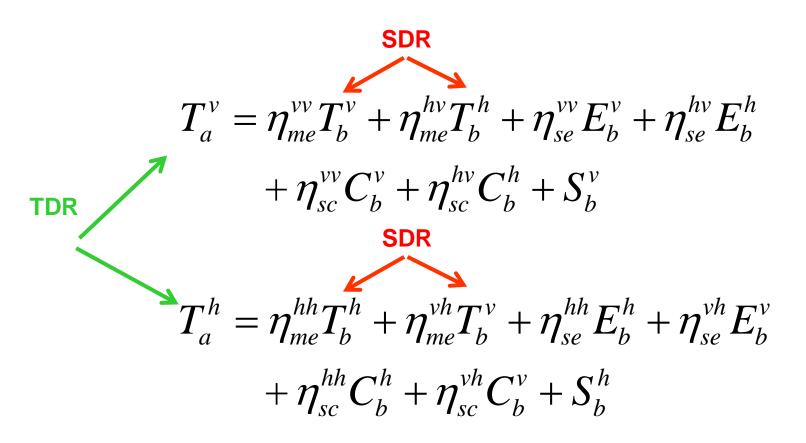


# **SSMIS Scanning Characteristics**

- 24 Channels (19-183 GHz)
- Conical Scan Geometry
- Mesospheric Sounding
- Improved Sounding HCS
- Swath Width 1700 km
- Scan Rate 31.6 rpm
- Calibration Accuracy
  - Better than 1K
  - Warm and Cold Targets each Scan





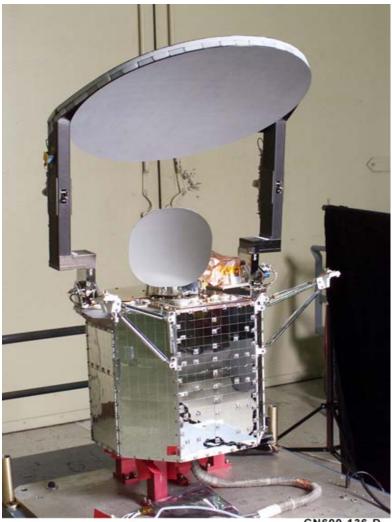


For a microwave conical scanning radiometer, the cross-polarization term is measured at imaging channels. Thus, sensor brightness temperatures can be derived from antenna brightness temperature measurements



# **Anomalies in SSMIS Antenna System**

- Main-reflector conically scans the earth scene
- Sub-reflector views cold space to provide one of two-point calibration measurements
- Warm loads are directly viewed by feedhorn to provide other measurements in two-point calibration system
- The SSMIS main reflector emits radiation from its coating material
  - SiOx VDA (coated vapor-deposited aluminum)
  - SiOx and AI VDA Mixture
  - Graphite Epoxy
- Warm load calibration is contaminated by solar and stray lights
  - Reflection Off of the Canister Top into Warm Load
  - Direct Illumination of the Warm Load Tines
- Space view is also occasionally contaminated



CN600-136-D



# Antenna/Sensor Brightness Temperature with an Emitting Antenna

$$\begin{cases} T_a^{\nu} = (1 - \varepsilon_r^{\nu})[\eta_{me}^{\nu\nu}T_b^{\nu} + \eta_{me}^{h\nu}T_b^{h}] + \varepsilon_r^{\nu}T_r \\ T_a^{h} = (1 - \varepsilon_r^{h})[\eta_{me}^{hh}T_b^{h} + \eta_{me}^{\nu h}T_b^{\nu}] + \varepsilon_r^{h}T_r \end{cases}$$

Or

$$\begin{cases} T_a^{\nu} = \gamma_{me}^{\nu\nu} T_b^{\nu} + \gamma_{me}^{h\nu} T_b^{h} + \mathcal{E}_r^{\nu} T_r \\ \\ T_a^{h} = \gamma_{me}^{hh} T_b^{h} + \gamma_{me}^{\nu h} T_b^{\nu} + \mathcal{E}_r^{h} T_r \end{cases}$$

Where

$$\gamma_{me}^{pq} = (1 - \mathcal{E}_r^p) \eta_{me}^{pq}$$
  
 $T_r$  is the antenna physical temperature



# **SSMIS Sounding Channel O-B** (Observation TDR - Simulation SDR)

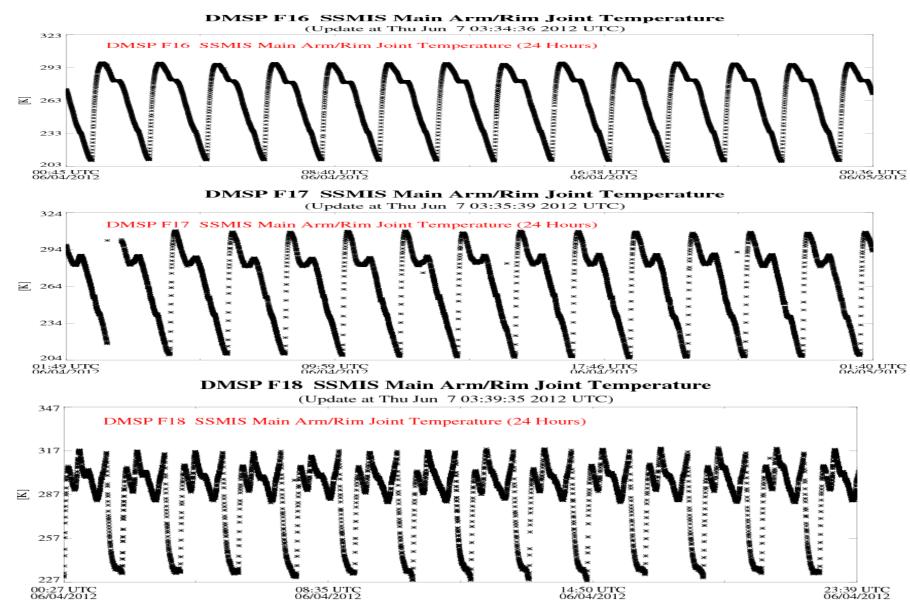
$$T_a^h - T_b^h = (\eta_{me}^{hh} + \eta_{me}^{vh} - 1 - \varepsilon_r^h \eta_{me}^{hh} - \varepsilon_r^v \eta_{me}^{vh})T_b^h + \varepsilon_r^h T_r$$

$$T_a^h - T_b^h \approx \varepsilon_r^h (T_r - \eta_{me}^{hh} T_b^h)$$

For the SSMIS temperature sounding channels, the bias is mainly driven by the difference between the antenna reflector temperature and the earth scene brightness temperature when the side-lobe effects from near- and far- fields are negligible. If  $T_r - \eta_{me}^{hh} T_b^h \approx 100$ K and an emissivity of 0.02, the bias can be on an order of 2.0K

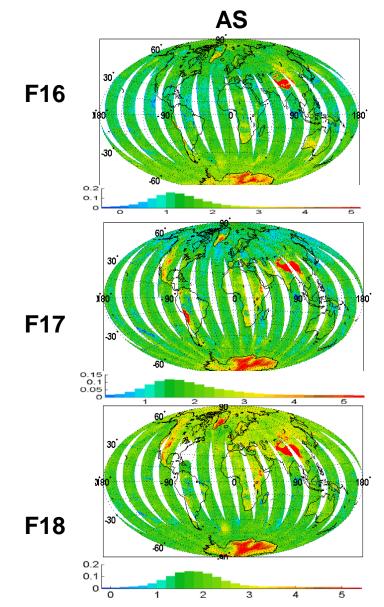


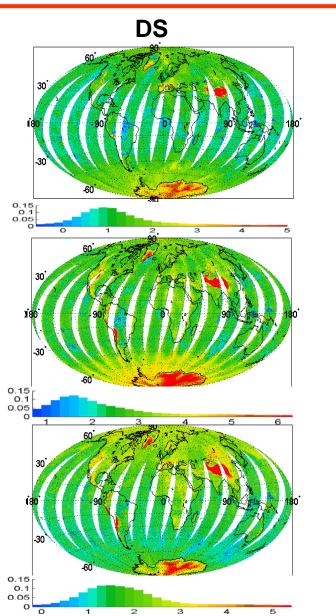
# **SSMIS Arm/Rim Temperature**





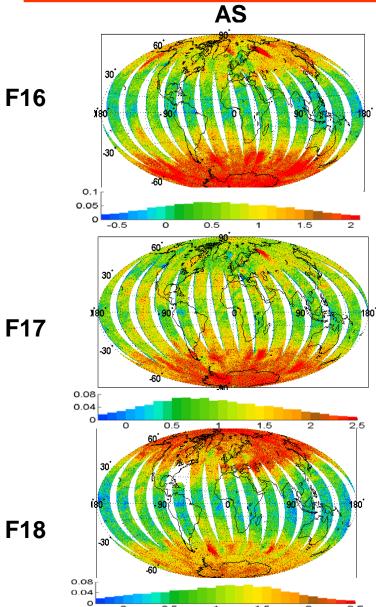
# SSMIS O-B at ch3 (53.59 GHz) February 15, 2012

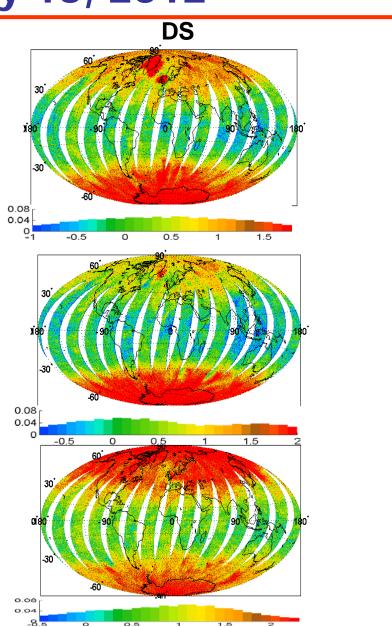






# SSMIS O-B at ch5 (55.5 GHz) February 15, 2012

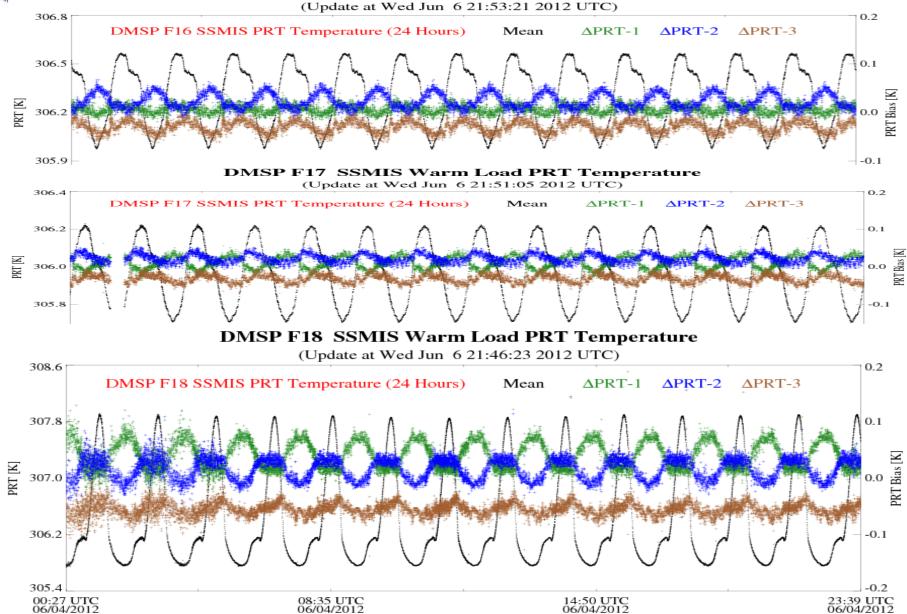






# **SSMIS PRT/Count**

#### DMSP F16 SSMIS Warm Load PRT Temperature





# **SSMIS Preprocessing and** Calibration Algorithms

- NRL/UK MetOffice SSMIS Unified Pre-processor (UPP data) (Bell et al. 2008)
  - Correction of antenna emission for LAS
  - Correction of warm load anomaly
  - Linear mapping of SSMIS imager to its predecessor (SSM/I)
  - Doppler shift correction for UAS
  - Spatial averaging to reduce to the sub-Kelvin levels
- NOAA/NESDIS SSMIS Pre-processor (NESDIS Data)

(Yan and Weng 2009)

- Correction of antenna emission for LAS
- Correction of warm load anomaly
- UAS bias removal using SABER (Sounding of the Atmosphere using Broadband Emission Radiometry) measurements simulated as truth
- Spatial filter for noise reduction
- Linear mapping of SSMIS imager to its predecessor (SSM/I) using the F15 and F16 Simultaneous Conical Overpass observations
- Inter-sensor calibration for SSMIS imager non-linearity (for climate reprocessing)



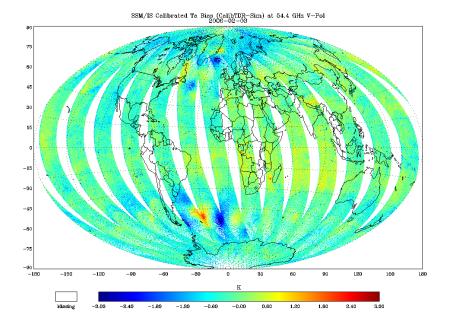
# F16 SSMIS O-B (Ch5)

## Before anomaly correction

# SIM/IE Te Blus (bes-Sig) at 64.4 GEV-Pol

Uncorrected Ch 4 OB-BK (GDAS CRTM)

## After anomaly correction



## Corrected Ch 4 OB-BK (GDAS CRTM)



# Fast SSMIS Zeeman Splitting Absorption Model

## **Energy level splitting:**

In the presence of an external magnetic field, each O2 energy level associated with the total angular momentum quantum number J is split into 2J+1 levels corresponding to the azimuthal quantum number M = -J, ..., 0, ..., J

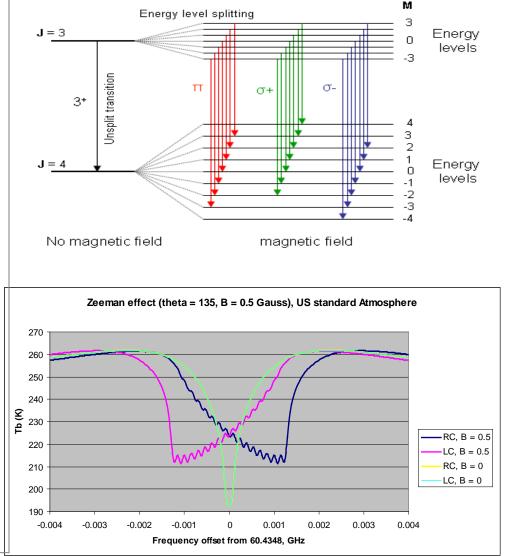
## Transition lines (Zeeman components) :

The selection rules permit transitions with  $\Delta J = \pm 1$  and  $\Delta M = 0, \pm 1$ . For a change in J (i.g. J=3 to J=4, represented by 3<sup>+</sup>), transitions with

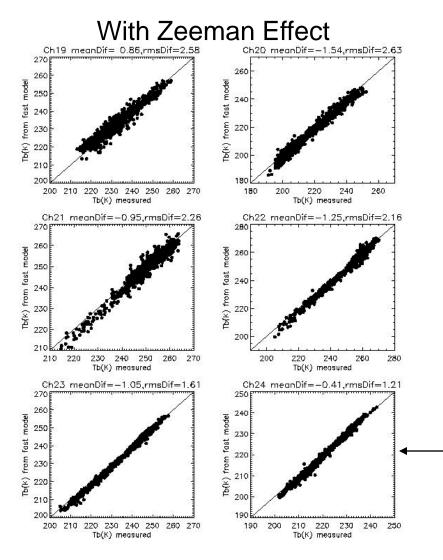
- $\Delta M$  = 0 are called  $\pi$  components,
- $\Delta M$  = 1 are called  $\sigma$ + components and
- $\Delta M$  = -1 are called  $\sigma$  components.

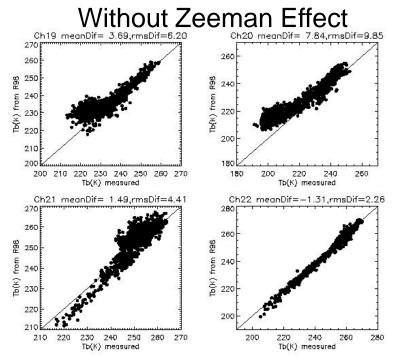
## **Polarization:**

The three groups of Zeeman components also exhibit polarization effects with different characteristics. Radiation from these components received by a circularly polarized radiometer such as the SSMIS upper-air channels is a function of the magnetic field strength  $|\mathbf{B}|$ , the angle  $\theta_B$  between **B** and the wave propagation direction **k** as well as the state of atmosphere, not dependent on the azimuthal angle of **k** relative to **B**.



# Comparison between SSMIS Observations and Simulations w/o Zeeman-effect





Without including Zeeman-effect in the model.

-Channels 23 & 24 are not affected by Zeeman-splitting

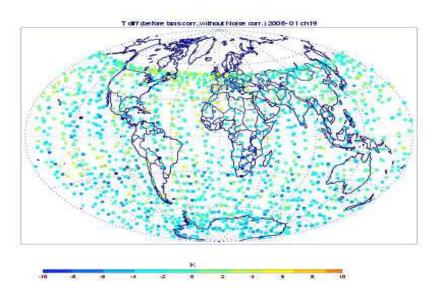
Collocated temperature profiles for model input are retrievals form the SABER experiment.

Sample size: 1097 samples



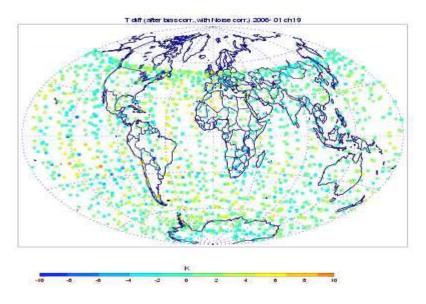
# SSMIS Upper Atmospheric Sounding Channels

## Before anomaly correction



No correction TB(OBS) – TB(BK) at Ch. 19

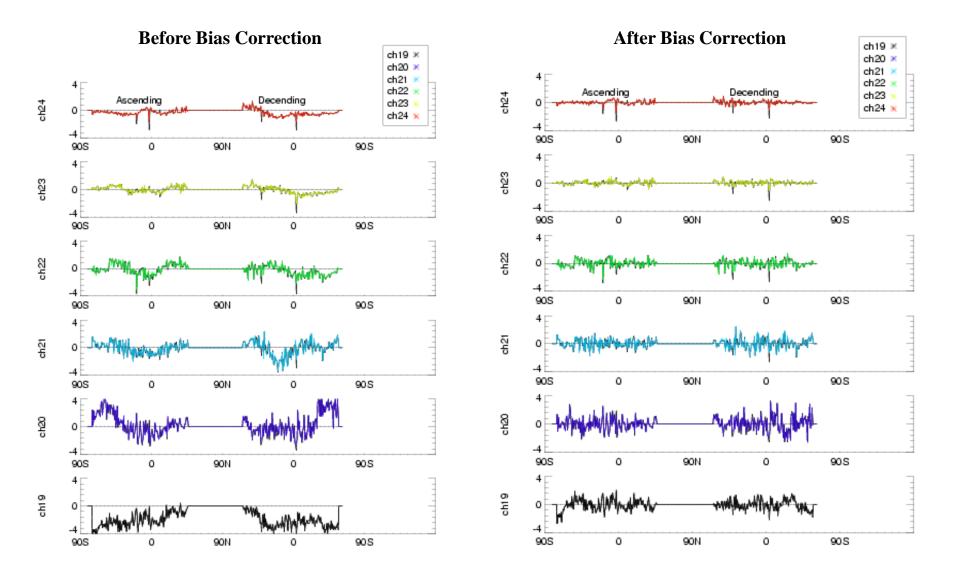
## After anomaly correction



Correction TB(OBS) – TB(BK) at ch. 19



# **SSMIS UAS Correction**



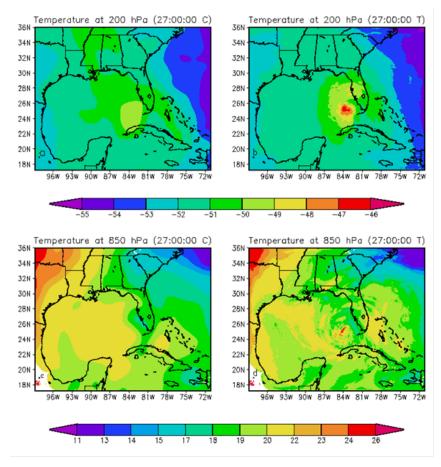


# SSMIS LAS Radiance Assimilation in WRF

DMSP F-16 SSMIS radiances is at the first time assimilated using NCEP 3Dvar data analysis. The new data assimilation improves the analysis of surface minimum pressure and temperature fields for Hurricane Katrina. Also, Hurricane 48-hour forecast of hurricane minimum pressure and maximum wind speed was significantly improved from WRF model

Significance: Direct assimilation of satellite radiances under all weather conditions is a central task for Joint Center for Satellite Data Assimilation (JCSDA) and other NWP centers. With the newly released JCSDA Community Radiative Transfer Model (CRTM), the JCSDA and their partners will be benefited for assimilating more satellite radiances in global and mesoscale forecasting systems and can improve the severe storm forecasts in the next decade





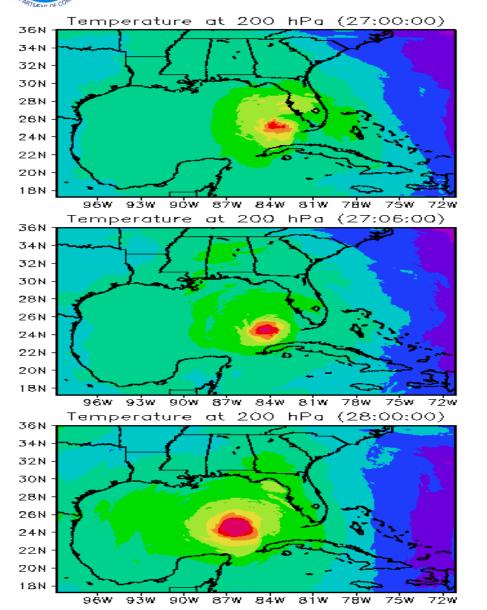
The initial temperature field from control run (left panels) w/o uses of SSMIS rain-affected radiances and test run (right panels) using SSMIS rain-affected radiances

# Katrina Warm Core Evolution through NCEP GSI Analysis

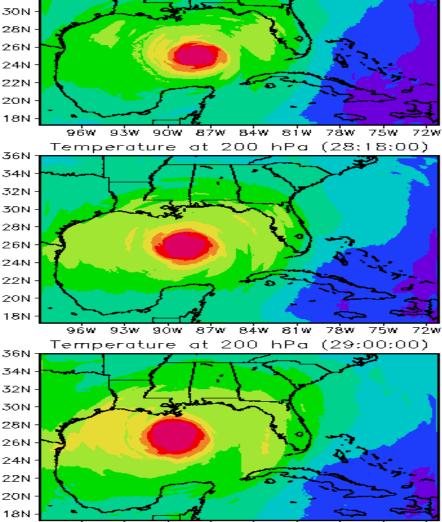
36N

34N

32N



NOAA



Temperature at 200 hPa (28:12:00)

96W 93W 90W 87W 84W 81W 78W 75W 72W



# **Summary and Conclusions**

- The noise of Suomi NPP ATMS antenna temperature data is well characterized and meets the specification. The instrument performance is stable and produce the data for user applications
- ATMS TDR biases at the temperature sounding channels with respect to NWP simulations are uniform over the global conditions. The biases also display some angular dependence but smaller than those from AMSU-A. This dependence could be related to near-field side-lobe effects from satellites
- The effects of the NWP model top on the ATMS biases are well understood. To fully utilize the ATMS sounding capabilities, NWP model top must be set to 0.1hPa or higher.
- To derive ATMS SDR products, the conversion from a single polarization measurement may become non-unique for some sounding channels (e.g. channel 16) due to a larger cross-polarization spill-over effect
- DMSP SSMIS TDR data display strong latitudinal-dependent O-B. This dependence is related to the emitting antenna. Three SSMIS instruments all have similar anomalous features. Uses of TDR data in NWP systems need a careful bias corrections scheme.
- There are some other calibration anomalies in SSMIS sounding systems related to the warm load temperature and counts.