UTRC Environmental Barrier Coating Development and Demonstration

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Team members

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Outline

- EBC development and demonstration on SiC/SiC CMC
- Evaluation of EBC_{SiC} on silicon nitride
- Key issues associated with EBC_{SiN} development







EBC Background

- EBC development was initiated under NASA HSCT EPM program, to prevent recession of SiC/SiC CMC in gas turbine combustion environment due to accelerated oxidation and subsequent volatilization of silica
- Three-layer EBC systems were developed

Moisture Barrier	Barium Strontium Aluminum Silicate (BSAS)
Intermediate Layer	BSAS+mullite mixture, or mullite
Bond Layer	Silicon
SiC/SiC CMC	





The Current "Standard" EBC (EBC_{SiC})

- EBC scaled-up and improved under DOE/Solar Turbines CSGT program
- Earlier work encountered difficulty in producing dense and crack free layers. Cracks affect EBC protectiveness.
- The current "standard" EBC_{SiC} is a three layer system with a Si layer, a BSAS-mullite intermediate layer, and a BSAS top layer.











EBC_{SiC} Demonstrated for Long-Term 1200°C Applications

>40,000 hours field tests in Solar Turbines, Inc. Centaur 50S engine Bakersfield, CA: 13,937 hours, test terminated Malden Mill II, MA: >15,000 hours, test terminated



CMC well protected in areas where EBC remained intact

COMPO 15.0kV



6

UTRC



WD 25.2mm

100µm

X90

EBC_{Sic} Optimization for 30,000 Hr Life Goal

• Improvement based on post-test analysis results and borescope observations

Observations	Resulting Improvement
Oxidation / recession from uncoated edges	UTRC applied EBC on liner edges
EBC spallation at combustor liner edges due to mechanical interference	Solar modified combustor liner design and mounting scheme
Surface asperity causing EBC spallation*	CMC manufacturers minimized tooling bumps on surface
BSAS recession*	UTRC optimized EBC composition for better stability in steam

*Karren More et al 2002 IGTI paper

Coating refurbishment process developed to meet the long term ceramic component life goal**
 **Harry Eaton et al 2001 IGTI paper

7



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SAS as "New" Steam Barrier Layer in EBC_{SiC}

SAS exhibits better steam stability compared to BSAS
SAS-based EBC has been applied to Solar CMC liners



*Hot-pressed sample at 2732°F (1500°C) in 90% Steam



SAS EBC applied to Solar Centaur 50S combustor liners for Malden Mill II test





EBC_{SiC} Demonstration Status Overview

Site	Substrate	EBC		Operation Period
Bakersfield 1 st Set	HACI MI	Si/mullite/BSAS		4/1999 – 11/2000
	HACI CVI	Si/mullite-BSAS/BSAS		13,937 hrs
Malden Mills	BFG MI	Si / mullite-BSAS / BSAS	Coated liner edges	8/1999 – 10/2000
No. I	HACI CVI	Si / mullite-BSAS / BSAS		7,238 hrs, refurbished
Malden Mills No. 2	BFG MI	Si / mullite-BSAS / BSAS	8/20 15,1	8/2000 – 7/2002 15 144 brs
	HACI CVI	Si / mullite-BSAS / BSAS		10,177 1113
Bakersfield	BFG MI	Si / BSAS	Coating refurbishment9/2On	9/2001 –
2 ^m Set	HACI CVI	Si / mullite-BSAS / BSAS		
Malden Mills No. 2 2 nd Set	HACI	Si / SAS		7/2002 –
	HACI	Si / mullite-SAS / SAS		Un going



 EBC_{SiC} for Si_3N_4 Applications

Known Factors:

- EBC_{SiC} (RT-1200°C: ~5.0 ppm/°C)
- Si₃N₄ (RT-1200°C: 3.0-3.2 ppm/C)
- Vertical through thickness cracks observed in as-processed coating

Original Concerns:

- Coating spallation due to thermal cycling
- Substrate oxidation due to the presence of cracks







Major Achievement / Findings on EBC_{SiC} for Si_3N_4 Applications

- EBC coating process developed for vane geometry.
- EBC_{sic} remained intact after thermal cycling tests and trip shutdowns during FT8 sector rig tests despite mismatch cracks.
- EBC_{SiC} provided effective environmental protection to silicon nitride substrate, indicating effective barrier function
- At room temperature, EBC_{SiC} significantly reduced the strength (>50%) of silicon nitride. However, at high temperature, EBC_{SiC} caused <15% strength reduction in silicon nitride (AS800).





EBC Coating Process Developed for Airfoil Geometry and Cooling Holes

Uniform coating thickness around airfoil





EBC for cooling hole demonstrated



Airfoil-to-platform transition capability demonstrated









 EBC_{SiC} Applied to Cooled FT8 Si₃N₄ Vane and Survived Two Trip Shutdowns



- SN282 vane uncoated
- AS800 vane coated with $\mathsf{EBC}_{\mathrm{SiC}}$
- 30 hours at 70% power (1230°C at coated vane), 2 trip shutdowns from ~60% power, 1 hour at 80% power (1260°C at coated vane)



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Pressure Side



EBC_{SiC} Provided Environmental Protection to the Si₃N₄ Vane



No reaction/oxidation observed in the substrate of coated vane







oxidation occurred with silica formation and RE-silicate white powder on the surface of uncoated vane





 EBC_{SiC} Provided Effective Protection to Si_3N_4 in ORNL Keiser Rig





Uncoated surface after 1500-hr steam exposure at 1200°C

Results provided by Karren More, ORNL



- No oxidation observed in the AS800 substrate underneath after 2000-hr steam exposure at 1200°C
- Coating remained intact except through thickness cracks due to CTE mismatch





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EBC_{SiC} Significantly Reduces Room Temperature Strength of Si₃N₄



- RT strength knockdown first discovered in Solar SN282 nozzle, later confirmed in AS800 and SN88
- <15% knockdown at high temperature





Mechanisms Causing RT Strength Knockdown Identified

- No effect of surface preparation
- No evidence of chemical interaction between coating and substrate causing strength debit.
- Thermal expansion mismatch between coating and substrate induces cracks in the coating. The cracks in the coating act as strength controlling flaws and reduce substrate strength.







EBC for Silicon Nitride (EBC_{SiN}) Currently Being Developed

Issues Identified and Need to Be Addressed for EBC_{SiN} Development

- 1. Steam stability and barrier function
- 2. Chemical compatibility between the individual coating layers and with substrate material
- 3. Composition and phase stability over temperature range
- 4. Coating adherence satisfying transient conditions and sustaining centrifugal force on rotating components
- 5. Creep resistance at high temperatures, especially under centrifugal force on rotating components
- 6. Erosion resistance
- 7. Effect on substrate performance
- 8. Non-line-of-sight coating process for complex shaped components







Other Areas Required to Enable EBC_{SiN} Development

- 1. Cost effective long term durability test facility with steam environment and gas velocity
 - Most of the existing steam rigs for durability test operate under minimum gas velocity
 - High pressure burner rig type of test is of very high cost
- 2. Mechanical testing method to assess coating performance at HT
- 3. Recession mechanism of potential EBC materials





Test Method Developed for Coating Adhesion at RT and High Temperature





Recession Mechanism of BSAS, SAS and Other Potential Silicate-Based Coating Materials Need to be Understood

•What are the recession mechanism and recession rate?

•Can equations based on SiC be applied?

$$\mathsf{R}(\mathsf{T}) \propto \frac{|\mathsf{V}|^{1/2} \bullet \mathsf{P}_{\mathsf{H2O}}^2}{\mathsf{P}_{\mathsf{total}}^{1/2}} \bullet \mathsf{e}^{-1/\mathsf{KT}}$$

Temperature	Recession Rate / 1000hrs	Test	
	(normalized for P & v)*		
2200°F (1200°C)	BSAS ~2.3 μm	Solar Bakersfield Engine Test, ~ 14,000 hrs	
2370°F (1300°C)	BSAS/mullite ~ 11 μ m	NASA EPM HPBR Test, 90 hrs	
2600°F (1425°C)	BSAS-mullite ~17 μ m	UTRC Steam Rig, 1024 hrs	
2730°F (1500°C)	BSAS ~61 μm	UTRC Steam Rig, 365 hrs	

Model to predict EBC life and recession behavior is critical to technology insertion





Summary

- EBC_{sic} has been demonstrated for SiC/SiC CMC for >10,000 hour applications.
- Applicability of EBC_{sic} to silicon nitride has been examined. The coating provide excellent environmental protection but affect substrate material performance.
- EBC_{SiN} is currently under development. Key requirements and enabling technologies have been identified.



