

Development of EBCs with Enhanced Durability and Temperature Capability

**Kang Lee
Cleveland State University
NASA Glenn Research Center
Cleveland, OH**

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Co-workers

Dennis Fox - water vapor thermogravimetry

Craig Robinson - high pressure burner rig test

Narottam Bansal - fabrication of hot pressed EBC

Jeff Eldridge - phase stability/stress measurement

Dongming Zhu/Robert Miller - thermal conductivity

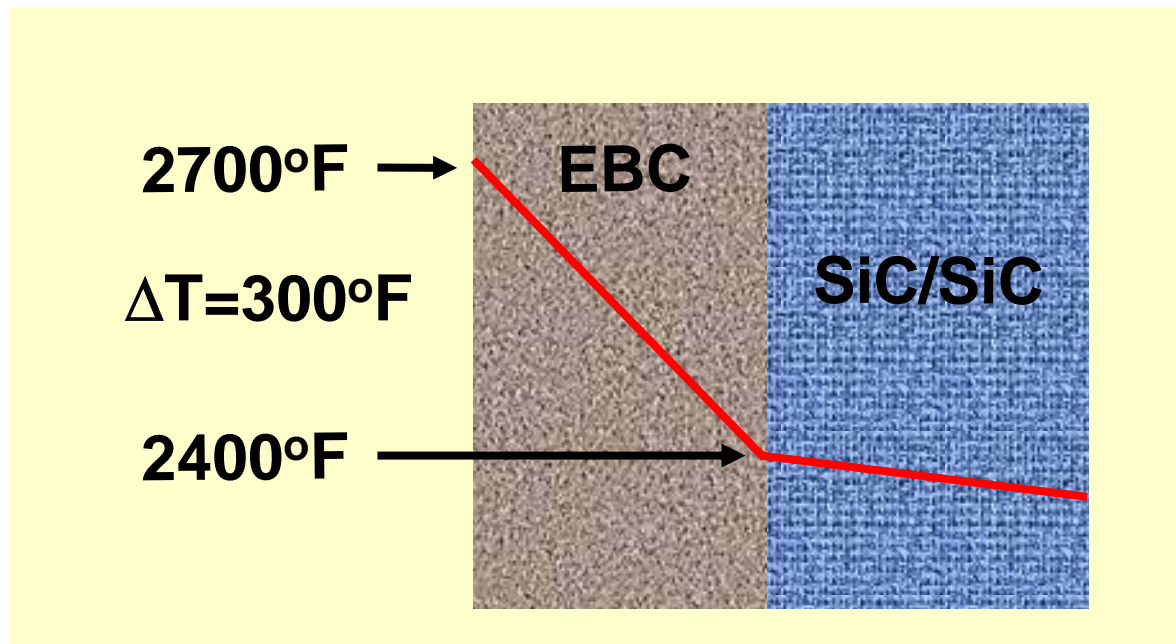
Outline

- **Objective**
- **Approach**
- **Results (SiC/SiC, Si₃N₄)**
- **Conclusions**

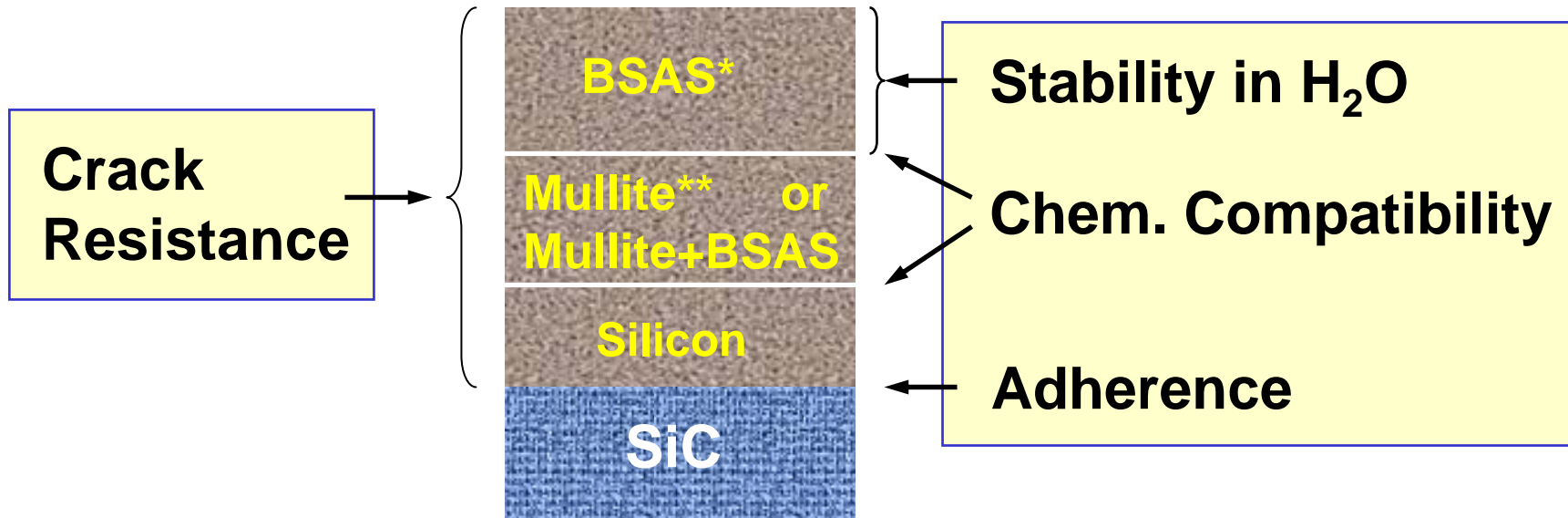
Objective

Develop advanced EBCs:

>1000 hr life at 2700°F (1482°C) EBC temp and
2400°F (1316°C) CMC temp



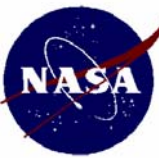
Current EBCs



* $x\text{BaO} \cdot (1-x)\text{SrO} \cdot \text{Al}_2\text{O}_3 \cdot 2\text{SiO}_2$

** $3\text{Al}_2\text{O}_3 \cdot 2\text{SiO}_2$

- Developed in the NASA HSR-EPM Program in joint research by NASA-GE-PW (2001 R&D 100 Award)
- Successful 15,000h engine test in Solar Centaur 50s SiC/SiC combustor liners under the DOE-CSGT Program (scaled up by Pratt & Whitney)



Current EBC Issues - Recession

Projected BSAS Recession

(1000 hr, 6 atm, $v_{\text{gas}} = 24$ m/sec)

1300°C : 28 μm

1400°C : 67 μm

1500°C : 268 μm

Silica Volatility Model (Smialek et al)

$$\text{Volatility} \propto \frac{v^{1/2} \times P(\text{H}_2\text{O})^2}{(P_{\text{TOTAL}})^{1/2}}$$

v : gas velocity

$P(\text{H}_2\text{O})$: water vapor pressure

P_{TOTAL} : total pressure

Assumptions

- Observed weight loss is due to silica loss only
- $\text{Si}(\text{OH})_4$ is the major volatile species

Approach to Develop New EBC Systems



Temp goals

1482°C
(2700°F)

1400°C
(2552°F)

1316°C
(2400°F)

**New
Top coat**

Mullite or
Mullite+BSAS**

Silicon

MI SiC/SiC

- **Identify new top coats**

- water vapor stability @ $T > 1482^\circ\text{C}$
- chemical compatibility @ $T > 1400^\circ\text{C}$
- mechanical compatibility

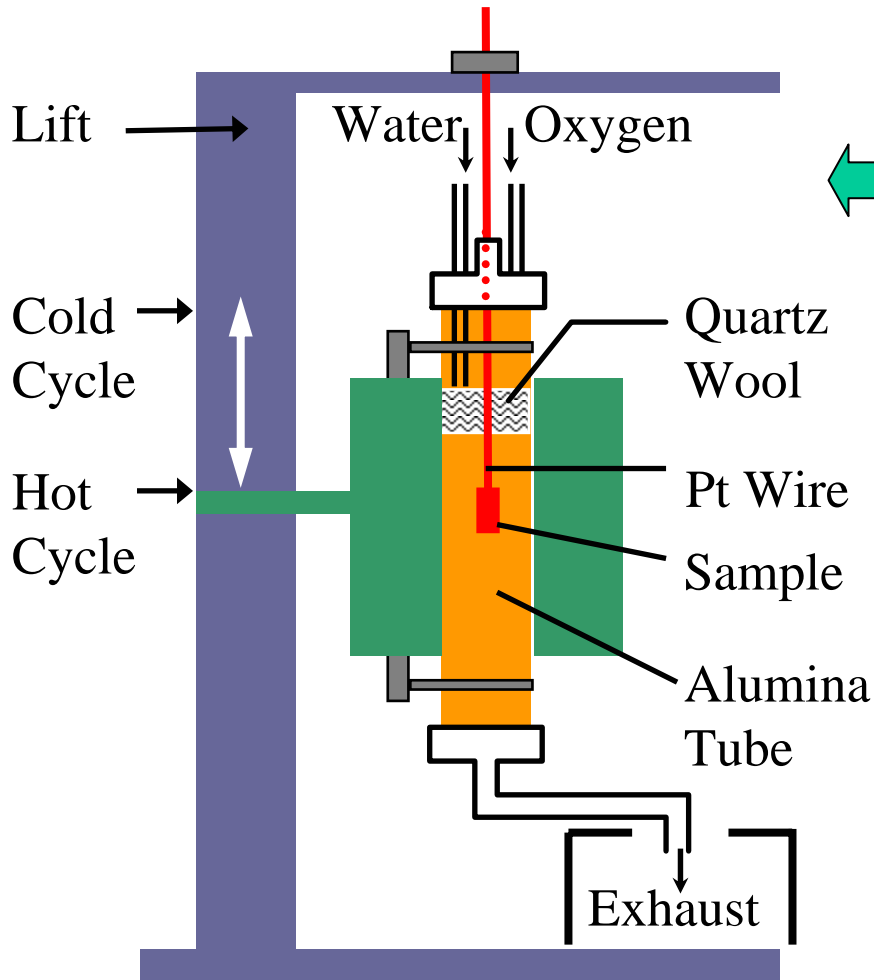
- **Water vapor TGA test**

- Water vapor stability of top coat @ 1500°C

- **High steam thermal cycling test**

- Chemical & Environmental durability @ $1300 - 1400^\circ\text{C}$

High Steam Thermal Cycling Rig (HSTC)

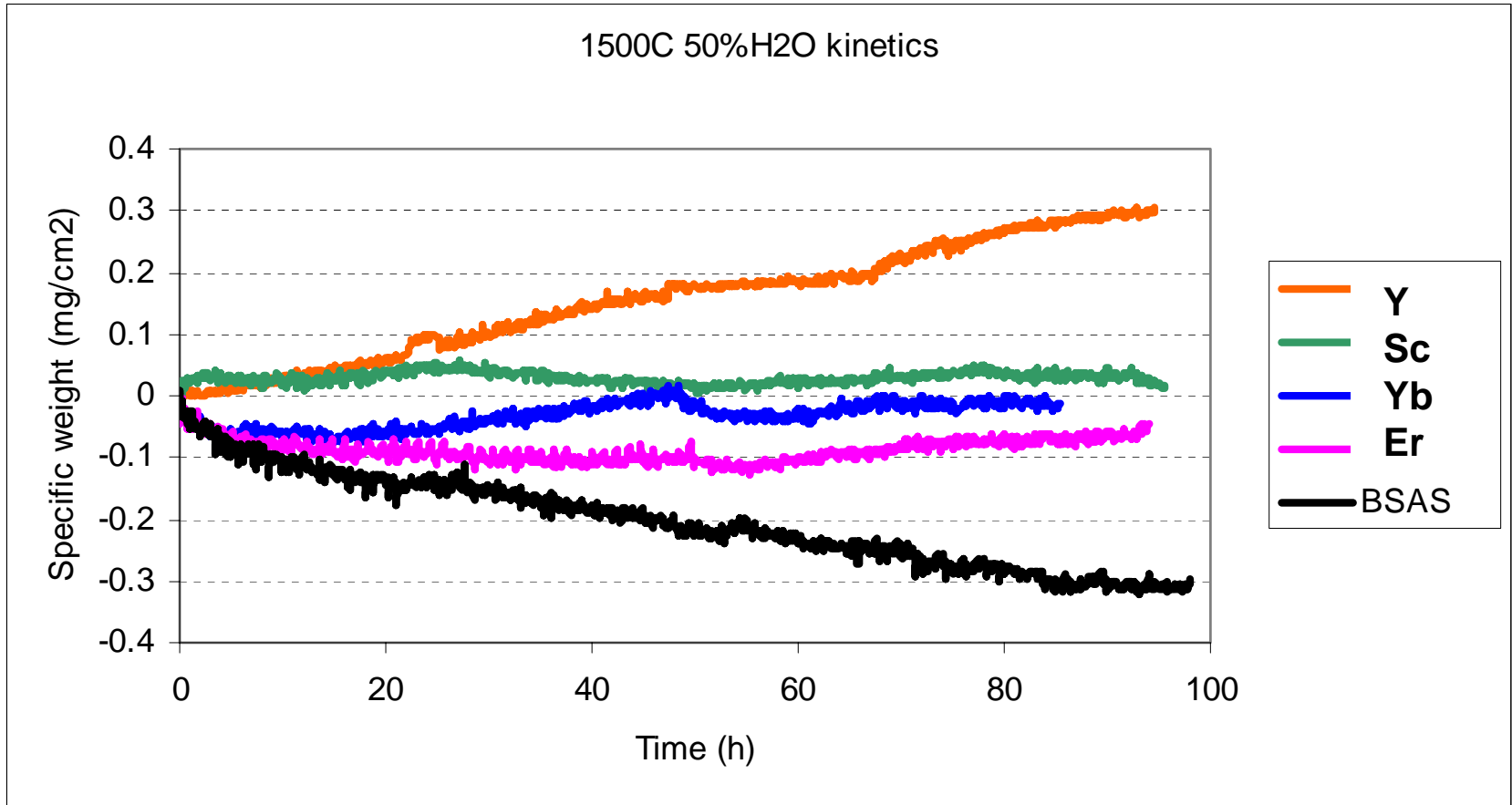


- HSTC Rig
 - simulate fuel lean environment ($p_{\text{H}_2\text{O}} = 0.9$, $V_{\text{gas}} = 2.2 \text{ cm/sec}$)
 - environmental durability
 - chemical stability

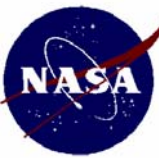
- Water vapor TGA
 - volatility



Rare-Earth Silicate EBC Identified



US PATENT PENDING



Rare Earth Silicate CTE

Material	CTE ($10^{-6}/^{\circ}\text{C}$)
Y_2SiO_5	5 ~ 6*
Er_2SiO_5	5 ~ 7
Yb_2SiO_5	3.5 ~ 4.5
Sc_2SiO_5	5 ~ 6

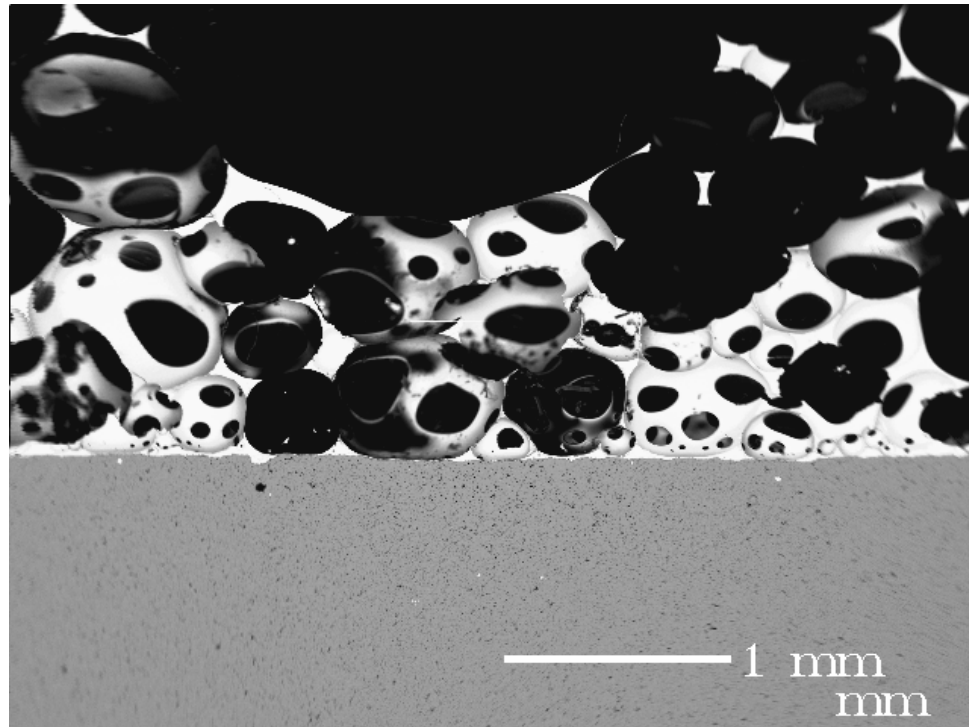
Material	CTE ($10^{-6}/^{\circ}\text{C}$)
SiC^{**}	4.5 ~ 5.5
$\text{Si}_3\text{N}_4^{**}$	3 ~ 4
Si^{**}	3.5 ~ 4.5
BSAS (celsian)	4 ~ 5
Mullite	5 ~ 6

* Ogura et al.

** Touloukian et al, "Thermophysical Properties of Matter"

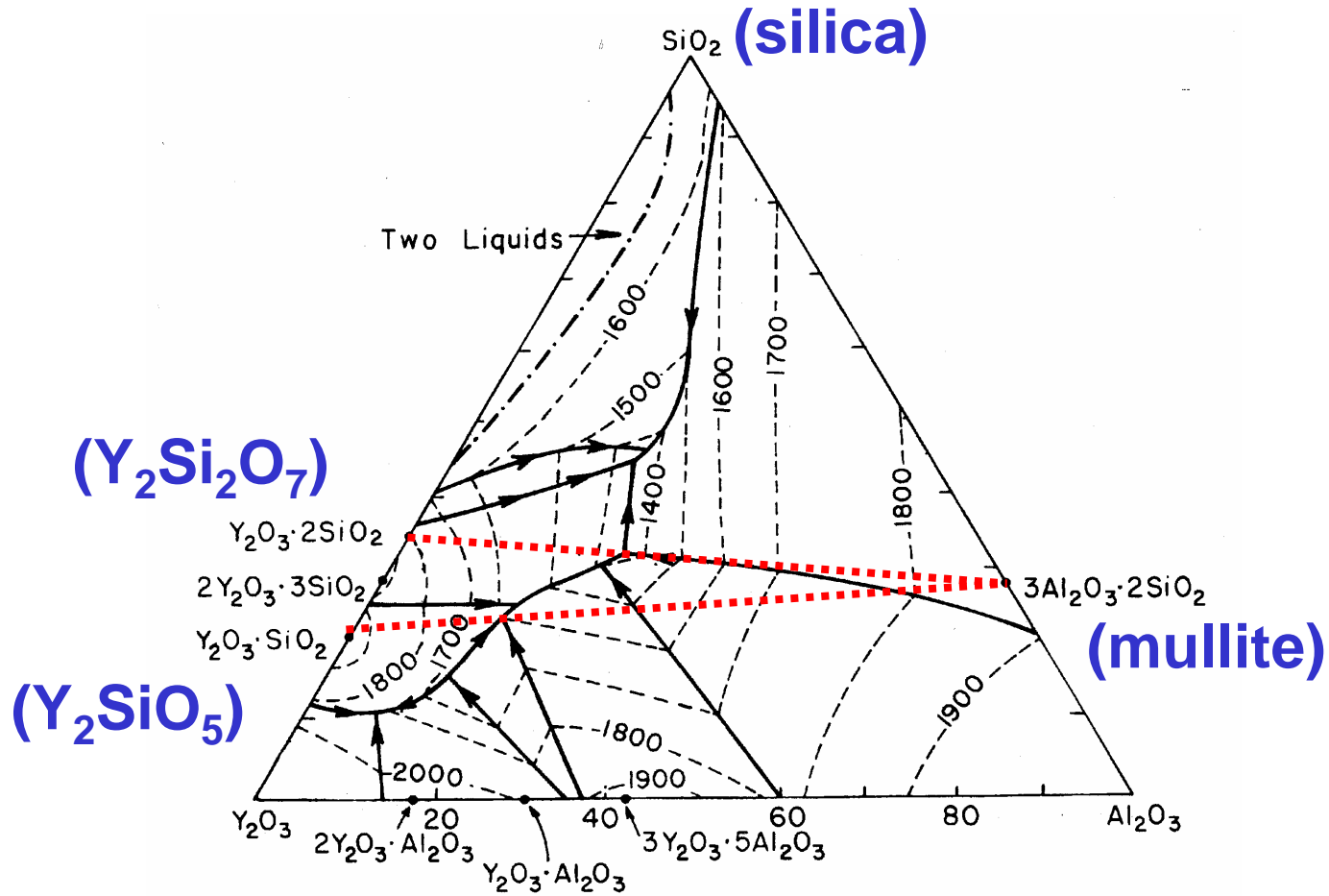
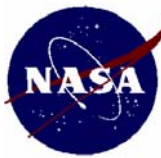
Mullite/ Y_2SiO_5

1400°C- 46 hr, 1h cycles, HSTC



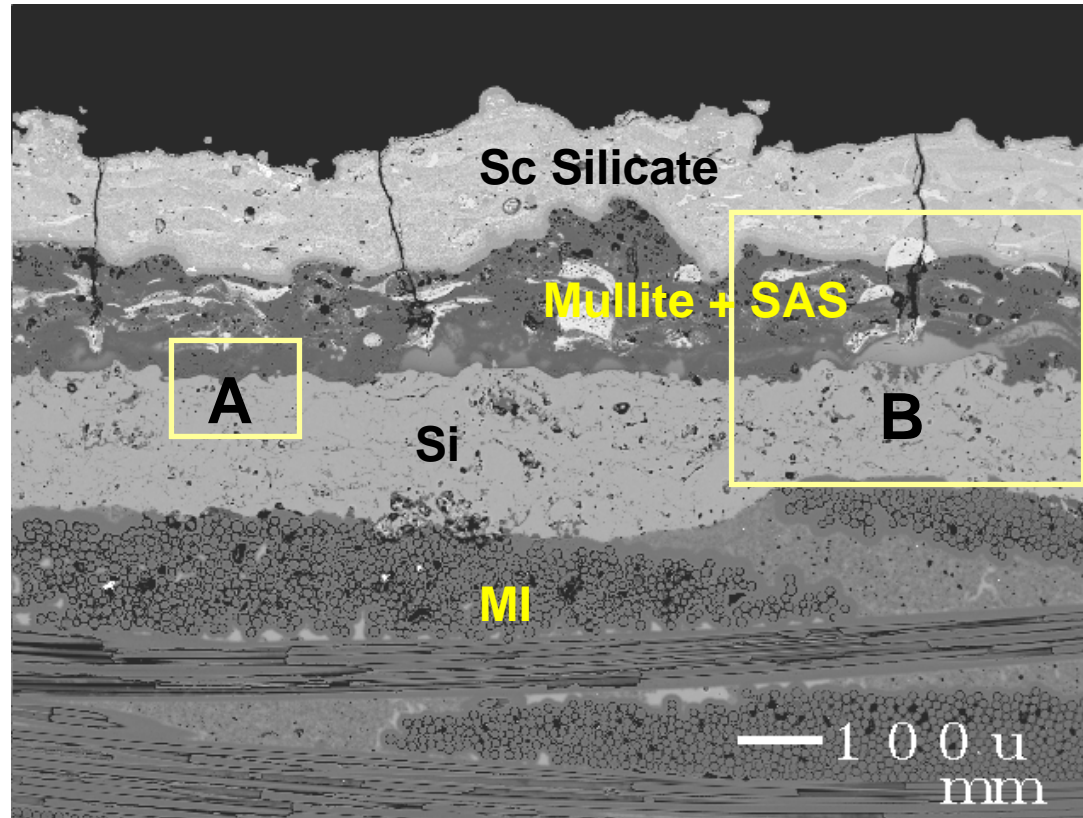
- Coating turned into Y-Al-Silicate bubbles

Y_2O_3 - Al_2O_3 - SiO_2 Phase Diagram



Si/Mullite+SAS/Sc Silicate

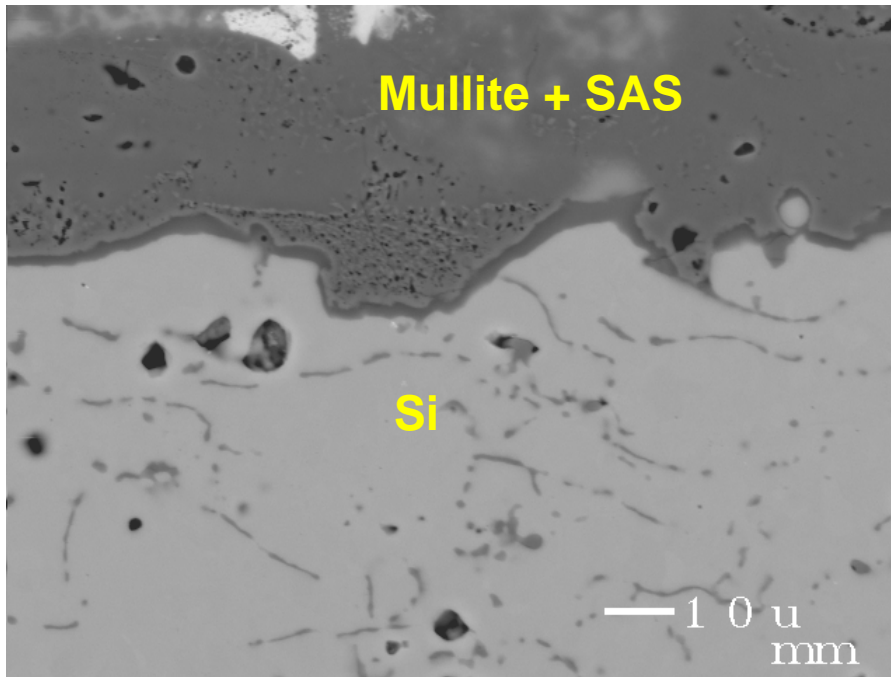
1380°C- 300 hr, 1h cycles, HSTC



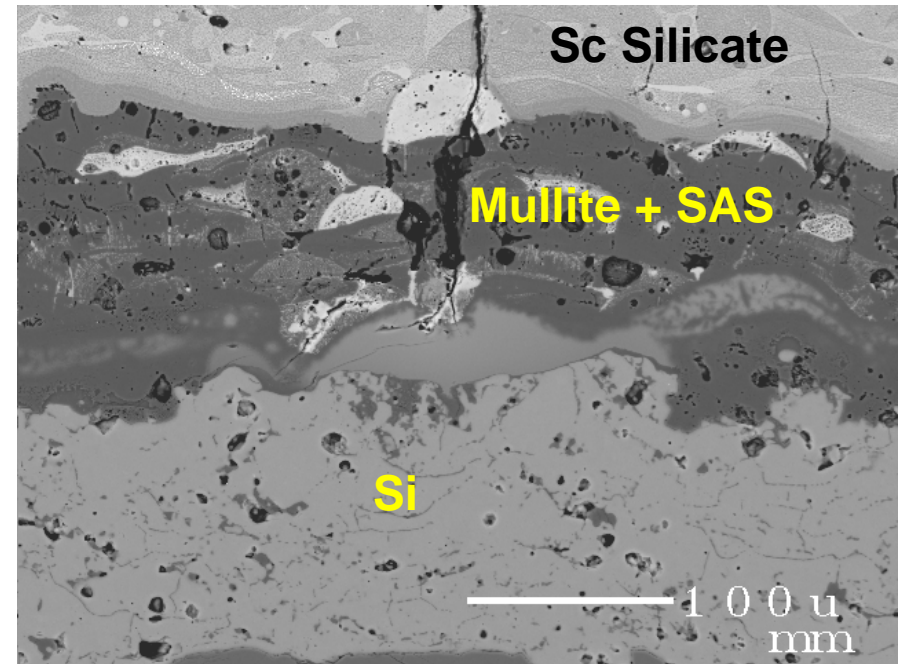
- Through-thickness cracks in ceramic layers
 - CTE mismatch, phase instability?, etc.
 - relax stress

Si/Mullite+SAS/Sc Silicate

1380°C- 300 hr, 1h cycles, HSTC



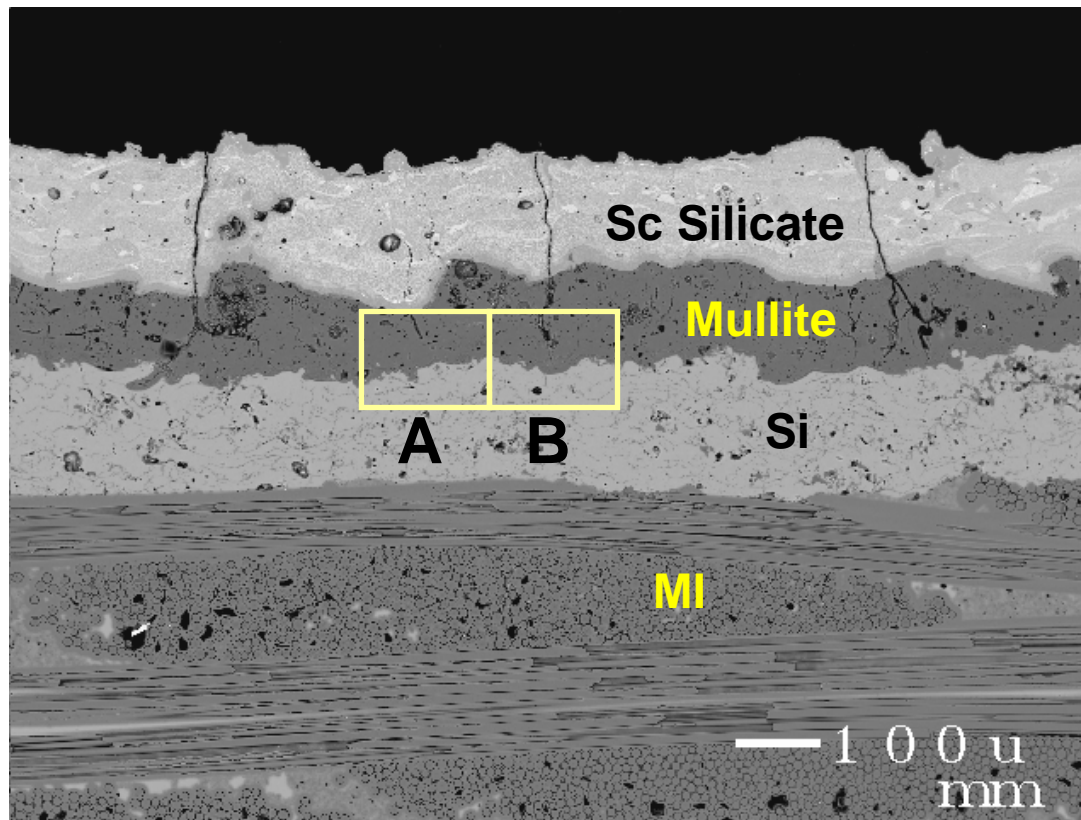
A: Minimal oxidation



B: Enhanced oxidation at the crack tip
- glass formation

Si/Mullite/Sc Silicate

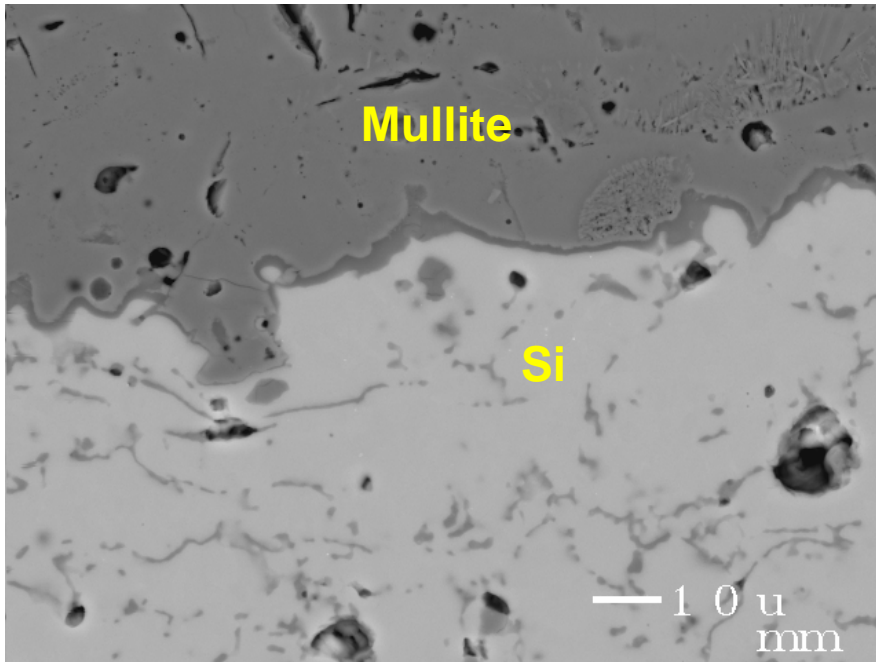
1380°C- 300 hr, 1h cycles, HSTC



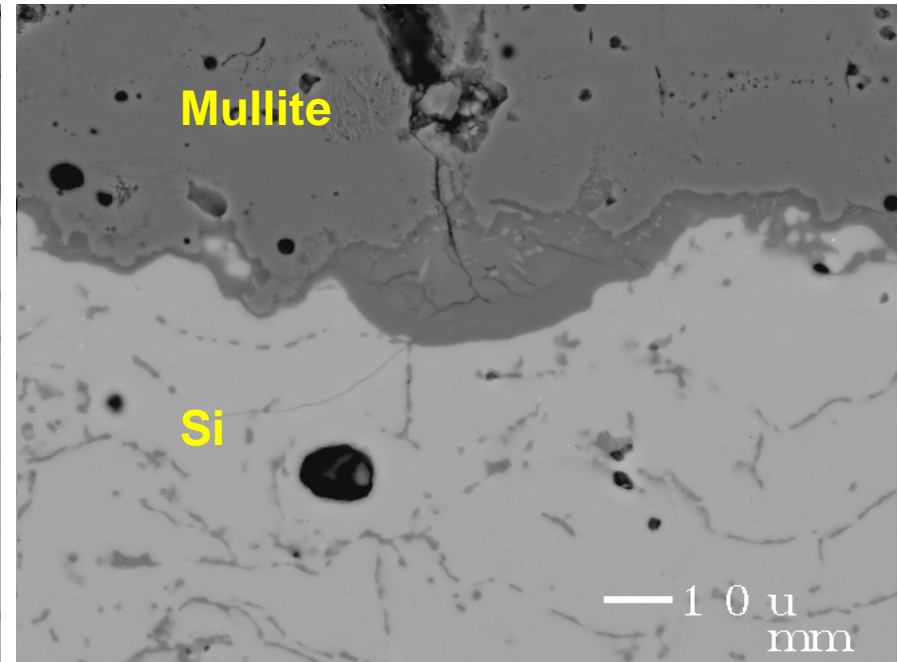
- Through-thickness cracks in ceramic layers
 - more prone to cracking with mullite intermediate layer

Si/Mullite/Sc Silicate

1380°C- 300 hr, 1h cycles

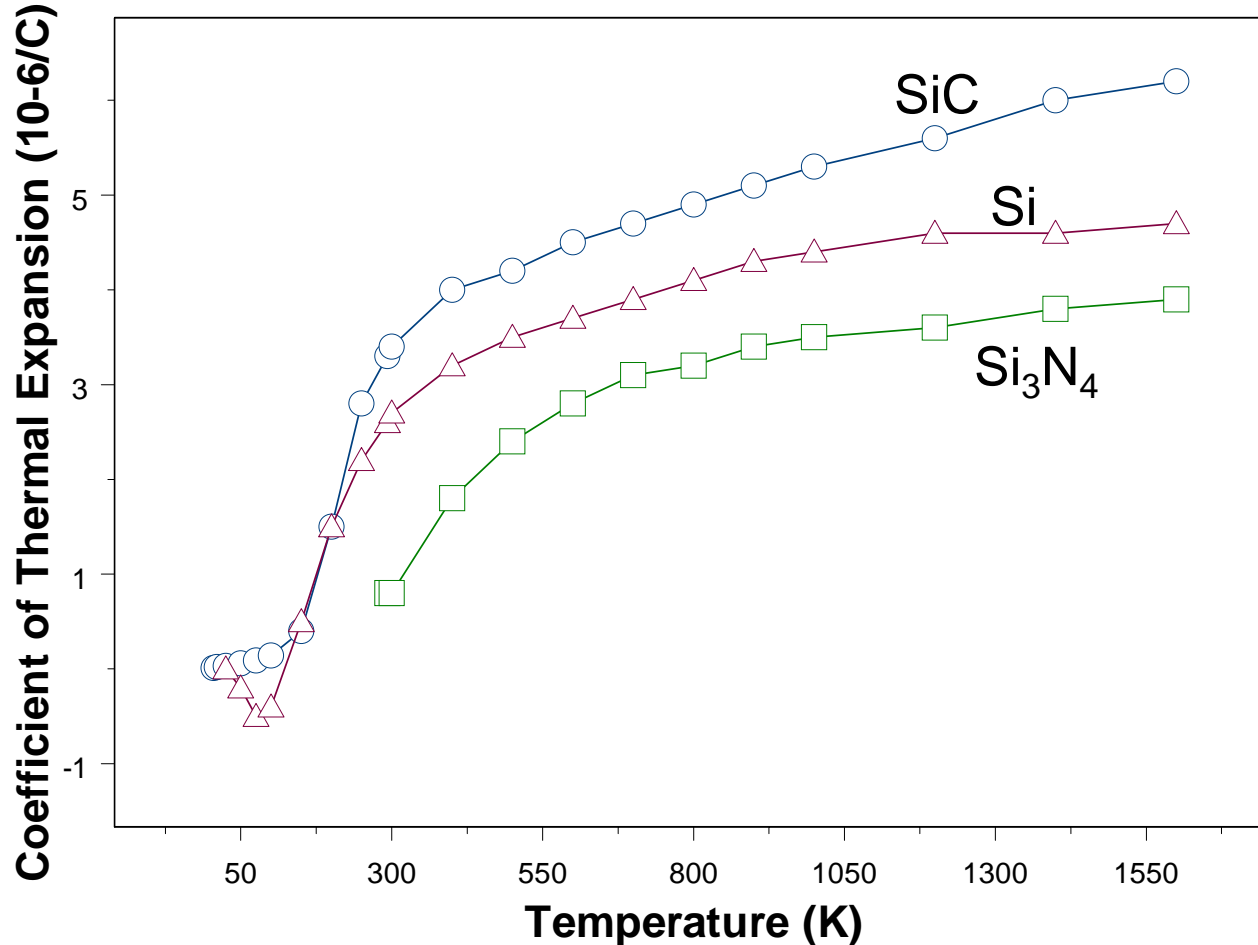


A: Minimal oxidation



B: Enhanced oxidation at the crack tip
- no glass formation

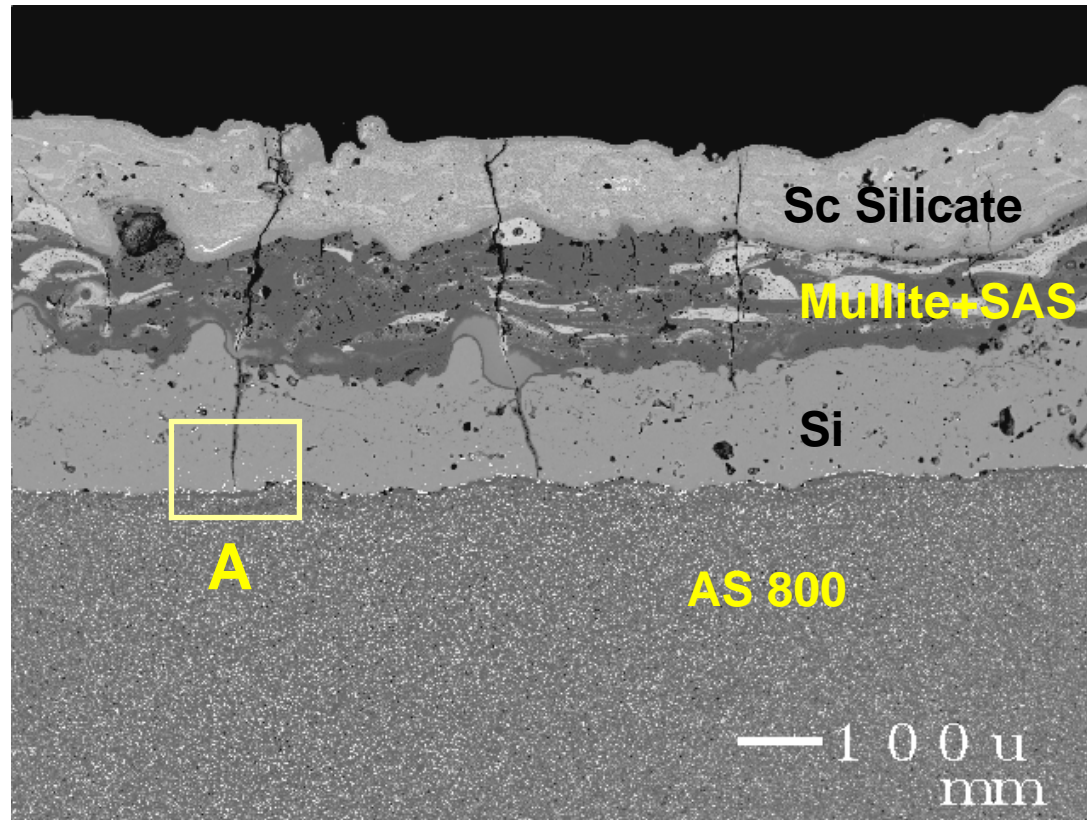
CTE of Si, SiC and Si₃N₄



- Lower CTE of Si₃N₄ causes bigger CTE mismatch

Si/Mullite+SAS/Sc Silicate

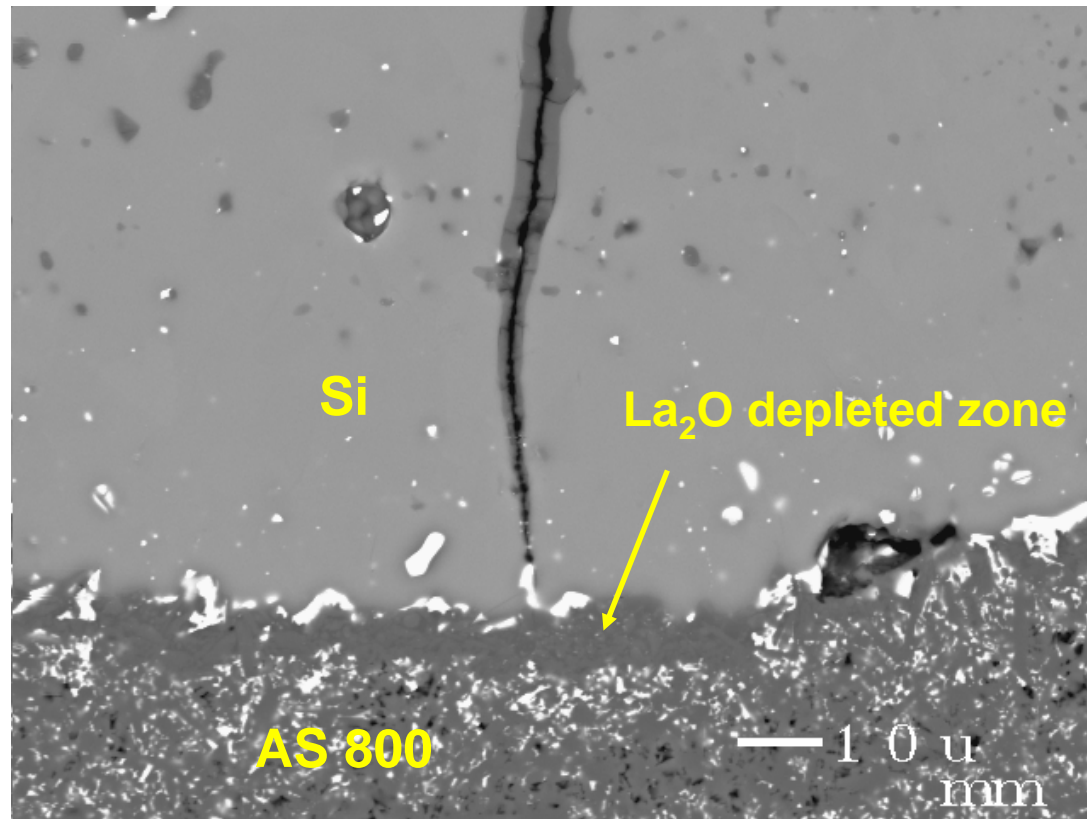
1380°C- 200 hr, 1h cycles, HSTC



- Through-thickness cracks in all three layers (Si bond coat in tension)

Si/Mullite+SAS/Sc Silicate

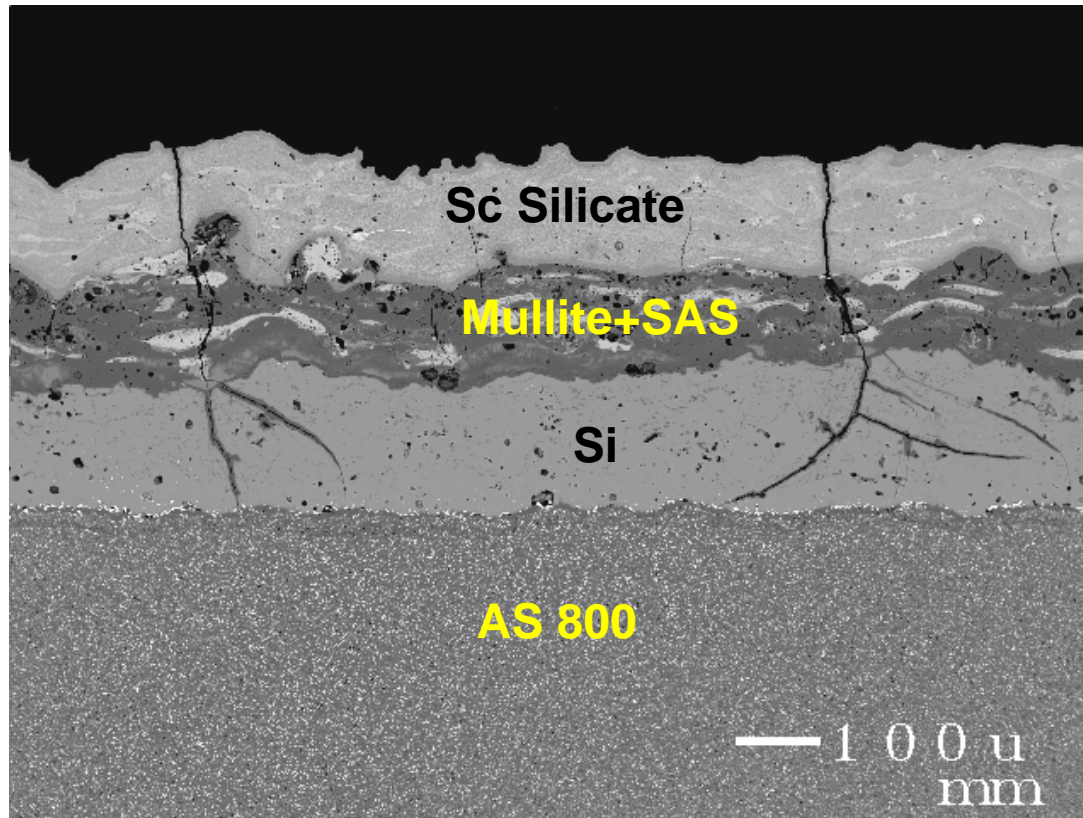
1380°C- 200 hr, 1h cycles, HSTC



A: Cracks provide path for water vapor which can rapidly oxidize Si bond coat

Si/Mullite+SAS/Sc Silicate

1380°C- 200 hr, 1h cycles, HSTC

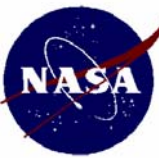


- some cracks branch laterally
- can cause coating delamination



Conclusions

- **Rare earth silicates (Sc, Yb, Lu) are promising as 2700°F EBC top coat**
 - **Low CTE ($4 \sim 7 \times 10^{-6}/^{\circ}\text{C}$)**
 - **Superior water vapor stability compared to BSAS**
 - **Superior chemical compatibility compared to BSAS**
 - **Low thermal conductivity (equal to or better than YSZ)**
- **Issues with rare earth silicates**
 - **Through-thickness cracks**
 - **What is the effect on long-term durability?**
 - **Economy (high cost of Sc, Lu)**



Conclusions (cont)

- **Si₃N₄ EBC issues**
 - Cracks into Si bond coat can rapidly oxidize Si bond coat
 - should not be as severe in infrequent cycling (industrial turbines)
 - Cracks branching laterally can cause delamination
 - Adherence on as processed surface?
- **Selection of EBC system depends on operating conditions**
 - EBC surface temperature (*BSAS vs. RE silicates*)
 - Substrate temperature (*Mullite+BSAS vs. Mullite*)
 - Life goal
 - Thermal cycling frequency
 - Aero or industrial gas turbines?