

N7024300



TRW NOTE NO. 70-FMT-819

PROJECT APOLLO
TASK MSC/TRW A-50

APOLLO MISSION 11, TRAJECTORY
RECONSTRUCTION AND POSTFLIGHT ANALYSIS
VOLUME 1

16 MARCH 1970

FACILITY FORM 808	N70 - 24300	
	(ACCESSION NUMBER)	(THRU)
	<u>137</u>	<u>1</u>
	(PAGES)	(CODE)
<u>NASAC # 108 349</u>	<u>31</u>	
(NASA CR OR TMX OR AD NUMBER)	(CATEGORY)	

Prepared for
MISSION PLANNING AND ANALYSIS DIVISION
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
MANNED SPACECRAFT CENTER
HOUSTON, TEXAS
NAS 9-8166

TRW NOTE NO. 70-FMT-819

PROJECT APOLLO
TASK MSC/TRW A-50

APOLLO MISSION 11, TRAJECTORY
RECONSTRUCTION AND POSTFLIGHT ANALYSIS
VOLUME I

16 MARCH 1970

Prepared for
MISSION PLANNING AND ANALYSIS DIVISION
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
MANNED SPACECRAFT CENTER
NAS 9-8166

Approved by W. P. Girod
W. P. Girod, Manager
MSC/TRW Task A-50

Approved by J. E. Alexander
J. E. Alexander, Manager
Guidance and Control
Systems Department

Approved by H. L. Moore
H. L. Moore, Manager
MSC/TRW Task A-50

Approved by D. G. Saile
D. G. Saile, Assistant
Project Manager
Guidance and Performance
Mission Trajectory Control Program

Approved by R. K. Petersburg
R. K. Petersburg, Manager
Systems Evaluation
Department

Approved by R. P. Parten
R. P. Parten, Chief
Mission Planning Support
Office
NASA Manned Spacecraft
Center

FOREWORD

This report is submitted to the NASA Manned Spacecraft Center in accordance with MSC/TRW Task A-50 Contract NAS 9-8166. This report contains the postflight analysis performed in conjunction with the Apollo II mission and is issued as supplement one to the Apollo II Mission Report (NASA/MSC Report MSC-00171, Nov. 1969).

The report is issued in two volumes. Volume I contains details of the analysis and results obtained, including appendixes. Volume II contains a listing of the 45-day best estimated trajectory (BET) for the Apollo II mission in the NASA Apollo Trajectory (NAT) format. The listing is not generally distributed but is available from NASA/MSC upon request. Requests should be made to:

NASA/MSC Computations and Analysis Division
Central Metric Data File
Code ED-5, Building 12, Room 133
Houston, Texas 77058

Page Intentionally Left Blank

TABLE OF CONTENTS

	<u>Page</u>
7.1 INTRODUCTION AND SUMMARY	7-1
7.1.1 Apollo 11 Mission	7-1
7.1.2 Postflight Analysis	7-1
7.2 ORBIT ANALYSIS	7-7
7.2.1 Methods of Reconstruction	7-7
7.2.2 CSM Best Estimate of Trajectory	7-8
7.2.3 LM Best Estimate of Trajectory	7-9
7.2.3.1 Descent Phase Trajectories	7-9
7.2.3.2 Rendezvous Trajectories	7-10
7.3 ONBOARD TRACKING DATA ANALYSIS	7-15
7.3.1 Introduction	7-15
7.3.2 Onboard Measurements	7-16
7.3.3 Evaluation of Onboard Tracking Data	7-16
7.4 LANDING RADAR DATA ANALYSIS	7-63
7.4.1 Descent Trajectories	7-63
7.4.2 Landing Radar Velocity Residuals	7-65
7.4.3 Lunar Surface Altitude Along Groundtrack	7-75
REFERENCES	R-1
APPENDIX A	A-1
APPENDIX B	B-1
APPENDIX C	C-1
APPENDIX D	D-1

Page Intentionally Left Blank

LIST OF TABLES

	<u>Page</u>
7.1 Apollo 11 Sequence of Events	7-3
7.2 Descent and Rendezvous Maneuver Summary for Apollo 11	7-4
7.3 Matchpoint Comparisons of Trajectories Produced with the R2 and L1 Lunar Potential Models	7-12
7.4 Apollo Mission 11 BET Summary	7-14
7.5 Summary of Rendezvous Radar Residual Statistics	7-18
7.6 Rendezvous Radar Only Solution Residual Statistics	7-28
7.7 Summary of VHF Ranging Residual Statistics	7-33
7.8 Summary of Sextant Residual Statistics	7-39
7.9 Comparison of Rendezvous Radar Noise Estimates with Specification Requirements	7-61
7.10 Comparison of VHF Ranging and Sextant Noise Estimates with Specification Requirements	7-61
7.11 LM Landing Site Coordinates	7-67
7.12 Landing Radar Velocity Residual Statistics	7-68

Page Intentionally Left Blank

LIST OF ILLUSTRATIONS

		<u>Page</u>
7-1a	Relative Motion of the LM for Apollo 11 Descent - DOI to Landing (CSM Centered)	7-5
7-1b	Relative Motion of the LM for Apollo 11 Rendezvous - Ascent to Docking (CSM Centered)	7-6
7-2	Tracking Data (Onboard and Ground Based) Timeline for Apollo 11 Descent and Rendezvous	7-13
7-3a	Rendezvous Radar Angle Residual Statistics	7-19
7-3b	Rendezvous Radar Range and Range Rate Residual Statistics	7-20
7-4	Rendezvous Radar Residuals (Insertion to CSI)	7-21
7-5	Rendezvous Radar Residuals (CSI to CDH)	7-23
7-6	Rendezvous Radar Residuals (CDH to TPI)	7-25
7-7	Rendezvous Radar Shaft Noise as a Function of Average Range	7-30
7-8	Rendezvous Radar Trunnion Noise as a Function of Average Range	7-31
7-9	Rendezvous Radar Range Noise as a Function of Average Range	7-32
7-10	VHF Ranging Residuals (DOI to PDI)	7-34
7-11	VHF Ranging Residuals (CSI to CDH)	7-35
7-12	VHF Ranging Residuals (CDH to TPI)	7-36
7-13	VHF Ranging Noise as a Function of Average Range	7-37
7-14	VHF Ranging Residual Statistics	7-38
7-15	Sextant Residuals (DOI to PDI)	7-41
7-16	Sextant Residuals (Insertion to CSI)	7-42
7-17	Sextant Residuals (CSI to CDH)	7-43
7-18	Sextant Residuals (CDH to TPI)	7-44
7-19	Sextant Angular Random Noise as a Function of Average Range	7-45
7-20	Sextant Residual Statistics	7-46
7-21	Out-of-Plane Component of LM Position Relative to CSM (DOI to PDI)	7-47
7-22	Differences Between Position Components of Relative Trajectories (DOI to PDI)	7-48

LIST OF TABLES

	<u>Page</u>
7.1 Apollo 11 Sequence of Events	7-3
7.2 Descent and Rendezvous Maneuver Summary for Apollo 11	7-4
7.3 Matchpoint Comparisons of Trajectories Produced with the R2 and L1 Lunar Potential Models	7-12
7.4 Apollo Mission 11 BET Summary	7-14
7.5 Summary of Rendezvous Radar Residual Statistics	7-18
7.6 Rendezvous Radar Only Solution Residual Statistics	7-28
7.7 Summary of VHF Ranging Residual Statistics	7-33
7.8 Summary of Sextant Residual Statistics	7-39
7.9 Comparison of Rendezvous Radar Noise Estimates with Specification Requirements	7-61
7.10 Comparison of VHF Ranging and Sextant Noise Estimates with Specification Requirements	7-61
7.11 LM Landing Site Coordinates	7-67
7.12 Landing Radar Velocity Residual Statistics	7-68

LIST OF ILLUSTRATIONS
(Cont)

		<u>Page</u>
7-23	Differences Between Velocity Components of Relative Trajectories (DOI to PDI)	7-49
7-24	Differences Between Position Components of Relative Trajectories (Insertion to CSI)	7-51
7-25	Differences Between Velocity Components of Relative Trajectories (Insertion to CSI)	7-52
7-26	Out-of-Plane Component of LM Position Relative to CSM (Insertion to CSI)	7-53
7-27	Out-of-Plane Component of LM Position Relative to CSM (CSI to TPF)	7-54
7-28	Differences Between Position Components of Relative Trajectories (CSI to TPF)	7-55
7-29	Differences Between Velocity Components of Relative Trajectories (CSI to TPF)	7-58
7-30	LM Landing Site Coordinates	7-66
7-31	Landing Radar X-Antenna Velocity Residuals (BET #3)	7-69
7-32	Landing Radar Y-Antenna Velocity Residuals (BET #3)	7-70
7-33	Landing Radar Z-Antenna Velocity Residuals (BET #3)	7-71
7-34	Landing Radar X-Antenna Velocity Residuals (Onboard/MSFN H-S)	7-72
7-35	Landing Radar Y-Antenna Velocity Residuals (Onboard/MSFN H-S)	7-73
7-36	Landing Radar Z-Antenna Velocity Residuals (Onboard/MSFN H-S)	7-74
7-37	Surface Altitude Along Ground Track	7-77
7-38.1	} Groundtrack of LR Range Beam Piercepont	7-79
7-38.2		7-81
7-38.3		7-83
7-39	Altitude of LM During LR Range Sampling	7-85

7.0 APOLLO MISSION 11 TRAJECTORY RECONSTRUCTION

AND POSTFLIGHT ANALYSIS

7.1 INTRODUCTION AND SUMMARY

7.1.1 Apollo 11 Mission

The Apollo 11 mission was launched from the Kennedy Space Center at 13:32:00 (hrs:min:sec) Greenwich Mean Time on 16 July 1969. Apollo 11 was the third manned lunar mission and the first to attempt and accomplish a landing on the lunar surface. A summary of the major events is presented in Table 7.1.

The descent phase of the mission was initiated during the thirteenth revolution of the moon at approximately 100^h.07^m Ground Elapsed Time (GET). The lunar module (LM) successfully landed on the lunar surface at approximately 102^h45^m GET.

The rendezvous phase began with ascent ignition during the 25th CSM revolution and ended with docking at 128^h03^m GET. A summary of the CSM and LM maneuvers performed during descent and rendezvous is presented in Table 7.2 and a graphical representation of these phases of the mission which depicts the motion of the LM relative to the CSM is shown in Figures 7-1a and 7-1b.

7.1.2 Postflight Analysis

The objective of the postflight analysis task was, in general, to generate trajectory parameters and data for the command and service modules (CSM) and LM from S-IVB/CSM separation to the end of mission. As in the Apollo missions 9 and 10, a preliminary trajectory was generated from the best available RTCC vectors. The bulk of the postflight analysis effort was then concentrated on reconstruction of the two periods of flight from LM/CSM undocking to LM touchdown (descent phase) and from LM ascent to LM/CSM docking (rendezvous).

The RTCC vectors used to generate the preliminary NAT (NASA Apollo Trajectory) are summarized in Appendix A. Most of the lunar trajectories were generated using RTCC SS2 (inclination constrained) solution vectors rather than SS1 (no a priori) solution vectors. Unlike the Apollo 10

SS2 vectors which were constrained to the pre-LOI1, rev 18, and rev 29 planes, the Apollo 11 SS2 vectors were constrained on a rev to rev basis. Each SS2 vector contained two revs of data and was constrained to the SS1 solution plane of one of these two revs (exceptions existed at maneuvers). This technique prevented the accumulation of a large error in the out-of-plane component of position. The lunar potential model used in the generation of the preliminary NAT and for propagation of RTCC vectors was the Boeing R2 model defined in Appendix B.

The final NAT was produced by updating the preliminary NAT to include reconstructions of critical maneuvers for which telemetered acceleration data was available and to reflect the results of the trajectory reconstruction efforts performed on the descent and rendezvous periods of the mission. These reconstructions will be discussed in detail in the following sections.

In general, the postflight analysis was accomplished without difficulty. Coincident with the trajectory reconstruction activities, analyses were performed to determine the quality of the onboard tracking data (LM rendezvous radar, CSM sextant, CSM VHF ranging, and LM landing radar). The results of these analyses are also included in this report.

Table 7.1 Apollo 11 Sequence of Events

	GET h:m:s	GMT d:h:m:s
Range Zero	00:00:00	16:13:32:00
Insertion	00:11:49.3	16:13:43:49.3
Translunar Injection Ignition	02:44:16.2	16:16:16:16.2
S-IVB/CSM Separation	03:17:04.6	16:16:49:04.6
First Docking	03:24:03.1	16:16:56:03.1
Spacecraft Ejection	04:16:59.1	16:17:48:59.1
Midcourse Correction #1	26:44:58.7	17:16:16:58.7
Lunar Orbit Insertion #1	75:49:50.4	19:17:21:50.4
Lunar Orbit Insertion #2	80:11:36.8	19:21:43:36.8
Undocking	100:12:00	20:17:44:00
CSM Separation	100:39:52.9	20:18:11:52.9
Descent Orbit Insertion	101:36:14	20:19:08:14
Powered Descent Initiation	102:33:05.2	20:20:05:05.2
Touchdown	102:45:39.9	20:20:17:39.9
Liftoff	124:22:00.8	21:17:54:00.8
Coelliptic Sequence Initiation	125:19:36	21:18:51:36
Constant Differential Height	126:17:49.6	21:19:49:49.6
Terminal Phase Initiation	127:03:51.8	21:20:35:51.8
Terminal Phase Finalization	127:46:09.8	21:21:18:09.8
Second Docking	128:03:00	21:21:35:00
Final Separation	130:30:01	22:00:02:01
Transearth Injection	135:23:42.3	22:04:55:42.3
Midcourse Correction #2	150:29:57.4	22:20:01:57.4
CM/SM Separation	194:49:12.7	24:16:21:12.7
Entry Interface	195:03:05.7	25:02:35:05.7

Table 7.2 Descent and Rendezvous Maneuver Summary for Apollo 11

Maneuver	Type of Maneuver	Ignition Time (h:m:s) GET	Cutoff Time (h:m:s) GET	T/M Coverage	ΔV (FPS)
SEPARATION	CSM/RCS	100:39:52.9	100:40:01.9	Yes	2.6
DOI	LM/DPS	101:36:14.0	101:36:44.0	No	76.4
PDI	LM/DPS	102:33:05.0	102:45:42.2	Yes	6775.8
ASCENT	LM/APS	124:22:00.8	124:29:15.7	Yes	6070.1
CSI	LM/RCS	125:19:36	125:20:23	No	51.5
CDH	LM/RCS	126:17:49.6	126:18:07.4	Yes	19.9
TPI	LM/RCS	127:03:51.8	127:04:14.5	Yes	25.3
TPF	LM/RCS	127:46:09.8	127:46:38.1	No	31.4

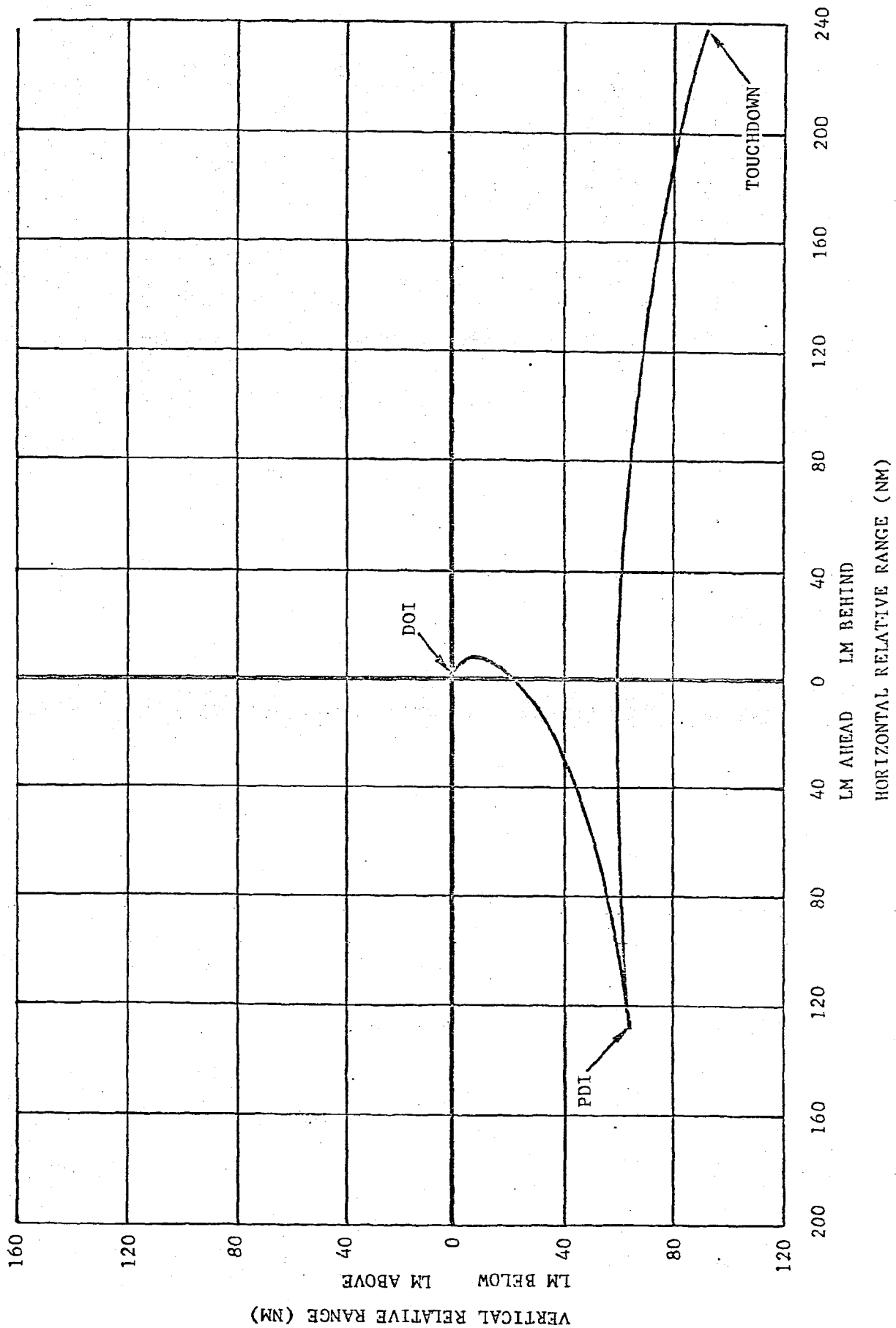


Figure 7-1a Relative Motion of the LM for Apollo 11 Descent - DOI to Landing (GSM Centered)

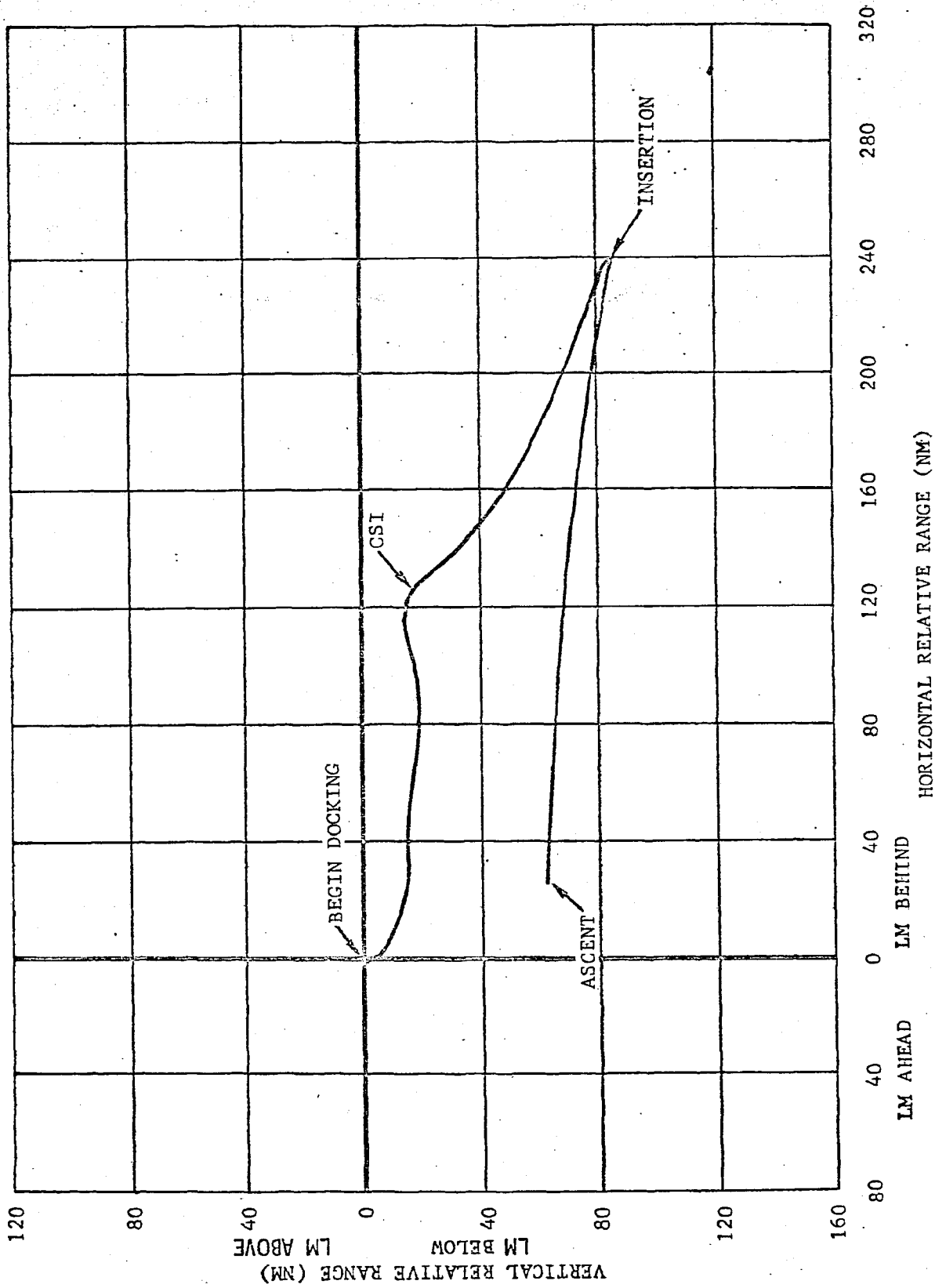


Figure 7-1b Relative Motion of the LM for Apollo 11 Rendezvous - Ascent to Docking (CSM Centered)

7.2 ORBIT ANALYSIS

7.2.1 Methods of Reconstruction

The HOPE Program was used as the basic orbit determination tool. The program utilizes a weighted least squares differential correction technique to regress on a large set of parameters. It is capable of handling two vehicles, and can use both ground based and onboard tracking data. An additional capability is the IGS (Inertial Guidance System) burn option which models the Apollo inertial measurement unit and uses telemetered acceleration data to reconstruct maneuvers.

The orbit determination was accomplished using four basic fit techniques. These techniques are described as follows:

- a) MSFN free flight - regression on the state vector over free flight intervals as defined by spacecraft maneuvers using MSFN data.
- b) MSFN IGS - regression on the state vector using, at maximum, one revolution of MSFN data and incorporating the spacecraft maneuvers which had telemetry coverage by means of the HOPE IGS burn model.
- c) Onboard free flight - regression on the state vector over free flight intervals using available onboard tracking data to correct the LM trajectory with respect to a fixed CSM trajectory (MSFN fits).
- d) Onboard IGS - regression on the state vector using available onboard data to correct the LM trajectory with respect to a fixed CSM trajectory and incorporation of the LM maneuver (which had telemetry coverage) by means of the HOPE IGS burn model.

More accurate trajectories are usually produced with techniques (b) and (d) since they take advantage of longer tracking data arcs. This factor is important in descent and rendezvous trajectory reconstruction since the tracking intervals between some maneuvers are too short to produce a representative trajectory over the whole segment.

As a result of the analysis of various lunar potential models contained in Reference 7, and on the basis of improved observation residuals and propagation characteristics, the L1 model (Langley Model 1) was used

in the orbit analysis. This model is basically the Boeing R2 model augmented by a C33 term. Table 7.3 shows improvements in propagation characteristics of the L1 over the R2 model. Both models are defined in Appendix B.

The trajectories for both Apollo 11 vehicles during descent and rendezvous were reconstructed using the methods summarized above. The data used in these reconstruction activities primarily included low speed MSFN, high speed MSFN, rendezvous radar, VHF ranging, and sextant data. Telemetered acceleration data were used to reconstruct maneuvers where available and applicable. Table 7.2 lists the maneuvers performed during the descent and rendezvous periods. Figure 7-2 shows the tracking data arcs (which were available over the periods of interest) as a function of ground elapsed time. In Figure 7-2, the solid bars represent the transmitting (two-way) MSFN station and the numbers represent the number of observations upon which final fits were based. Note that some stations operated in the dual mode (simultaneous tracking of both the CSM and the LM).

The following paragraphs describe the trajectories which were used as the final BET for both vehicles.

7.2.2 CSM Best Estimate of Trajectory

The trajectories for the CSM lunar revolutions 13, 14, 25 and 26 were reconstructed from low speed MSFN tracking data compacted to a rate of two samples per minute or, in the case of stations operating in the dual mode, one sample every 36 seconds. The data used are summarized in Figure 7-2. The quantity of data obtained for revolutions 13 and 26 was good. Because of the partial data arcs from some stations on revolutions 14 and 25, the data quantity in these revs could only be rated as fair. Inclusion of data from a southern hemisphere station (Ascension) enhanced the geometry of the active tracking network configuration and contributed to the quality of all the fits.

Two reconstruction techniques were used to obtain the CSM BET's. The MSFN IGS fit technique was used on revolution 13 because of the presence of telemetered acceleration data from the CSM separation burn performed

in the MSFN data arc. BET's for the remaining orbits of interest (14, 25 and 26) were obtained from MSFN free flight fits.

In general, the CSM BET's were of good quality. This is illustrated to some extent by the good position and velocity comparisons between revolutions (Table 7.4) and by the residual statistics listed for each fit in Appendix B. These statistics (standard deviation of .1 to .25 cycles per second) compare very well with Apollo 8 (standard deviations between .3 and .6 cycles per second) and Apollo 10 (standard deviations between .2 and .4 cycles per second). A portion of this improvement may be attributed to the better fit produced by the L1 lunar potential model.

Table 7.4 contains a summary of the final BET's giving fit type (technique), data interval, NAT trajectory interval, and position and velocity differences at matchpoints between segments.

7.2.3 LM Best Estimate of Trajectory

A major portion of the postflight analysis effort was directed towards reconstruction of the LM trajectories from undocking to landing and from liftoff through rendezvous. A discussion of the origin and quality of the final trajectories is included in the following paragraphs.

7.2.3.1 Descent Phase Trajectories

The descent phase was reconstructed in three segments; undocking to DOI, DOI to PDI, and PDI to Touchdown. The BET for undocking to DOI was obtained from a MSFN free flight fit based upon the entire data arc from revolution 13. The quantity of data was considerably better than for the CSM since five stations were tracking the LM. Residual statistics (summarized in Appendix B) compare well with the MSFN residual statistics obtained from the CSM fits. Note from Figure 7-2, which shows the tracking history, that the tracking station geometry was good.

The BET for the period from DOI to PDI was obtained from an onboard free flight fit based on CSM sextant and VHF ranging data taken prior to PDI. As can be seen in Figure 7-2 the data quantity was good, with 18 VHF ranging observations and 13 sextant sightings. Data quality is discussed more thoroughly in Section 7.3. The CSM trajectory which was used as the reference trajectory was the revolution 14 BET discussed in paragraph 7.2.2.

The BET for the powered descent segment of the flight was originally based on a fit obtained from low speed MSFN data taken from revolution 14 acquisition of signal to touchdown. The trajectory obtained from this fit was modified to force the landing point to coincide with the current best estimate of landing site location. Landing site parameters obtained from this descent trajectory (BET #3) were $.6358^{\circ}$ latitude, 23.4938° longitude, and -8557 feet altitude (referenced to the mean lunar radius). These figures compare well with the value published in Reference 10 as the best estimate (latitude $.647^{\circ}$ and longitude 23.505° , determined from postflight photo reduction).

Since the BET #3 was constrained to impact a desired landing site, the quality of the trajectory at PDI is not the best available. A subsequent reconstruction using a combination of onboard plus high speed MSFN data is discussed in Section 7.4 of this report. This combination of high speed data from acquisition of signal to landing and relative tracking data obtained prior to PDI produces a consistent and continuous representation of the LM trajectory from DOI to touchdown.

7.2.3.2 Rendezvous Trajectories

The BET for LM ascent was initialized with landing site coordinates of $.6357^{\circ}$ latitude, 23.4701° longitude, and a height of -8607 feet above the mean lunar radius. These initial conditions were then propagated to insertion using accelerometer data to model the ascent burn.

The LM BET for the period from insertion to TPF was reconstructed in two segments; insertion to CSI and CSI to TPF. The trajectory for the insertion to CSI segment was obtained from a MSFN free flight fit. The data arc and trajectory interval are described in Table 7.4. The MSFN data was good both quantitatively and qualitatively as can be seen in Figure 7-2. The residual statistics, summarized in Appendix B, show that the standard deviations of the doppler residuals are larger in this segment than in segments which have a less severe orbital geometry. This characteristic also existed in the Apollo 10 postflight results.

The second rendezvous segment covered the period from CSI to TPF. The BET chosen was obtained from an onboard data, IGS fit. The data used in the fit included LM rendezvous radar, CSM sextant, and CSM VHF ranging observations. In addition, telemetered acceleration data was used in the IGS burn option of HOPE to reconstruct the CDH and TPI burns. The data arcs are shown in Figure 7-2, and the residual statistics are summarized in Appendix B. Data quality was good, and the resulting BET produced an accurate relative trajectory. The CSM trajectory chosen as the reference for the relative observations was the revolution 26 trajectory described in paragraph 7.2.2. (The quality of the data and the reconstruction are discussed in more detail in Section 7.3 of this report.)

Table 7.3 Matchpoint Comparisons of Trajectories Produced with R2 and L1 Lunar Potential Models

Revolutions Compared	R2		L1	
	POS RSS (ft)	VEL RSS (ft/sec)	POS RSS (ft)	VEL RSS (ft/sec)
11-12	10,637	7.187	7544	5.756
12-13	9,936	8.178	4817	3.046
13-14	8,643	8.723	1555	2.53
25-26	9.595	9.139	2147	3.173

RSS = Square root of the sum of the squares of the differences between position (POS) or velocity (VEL) components.

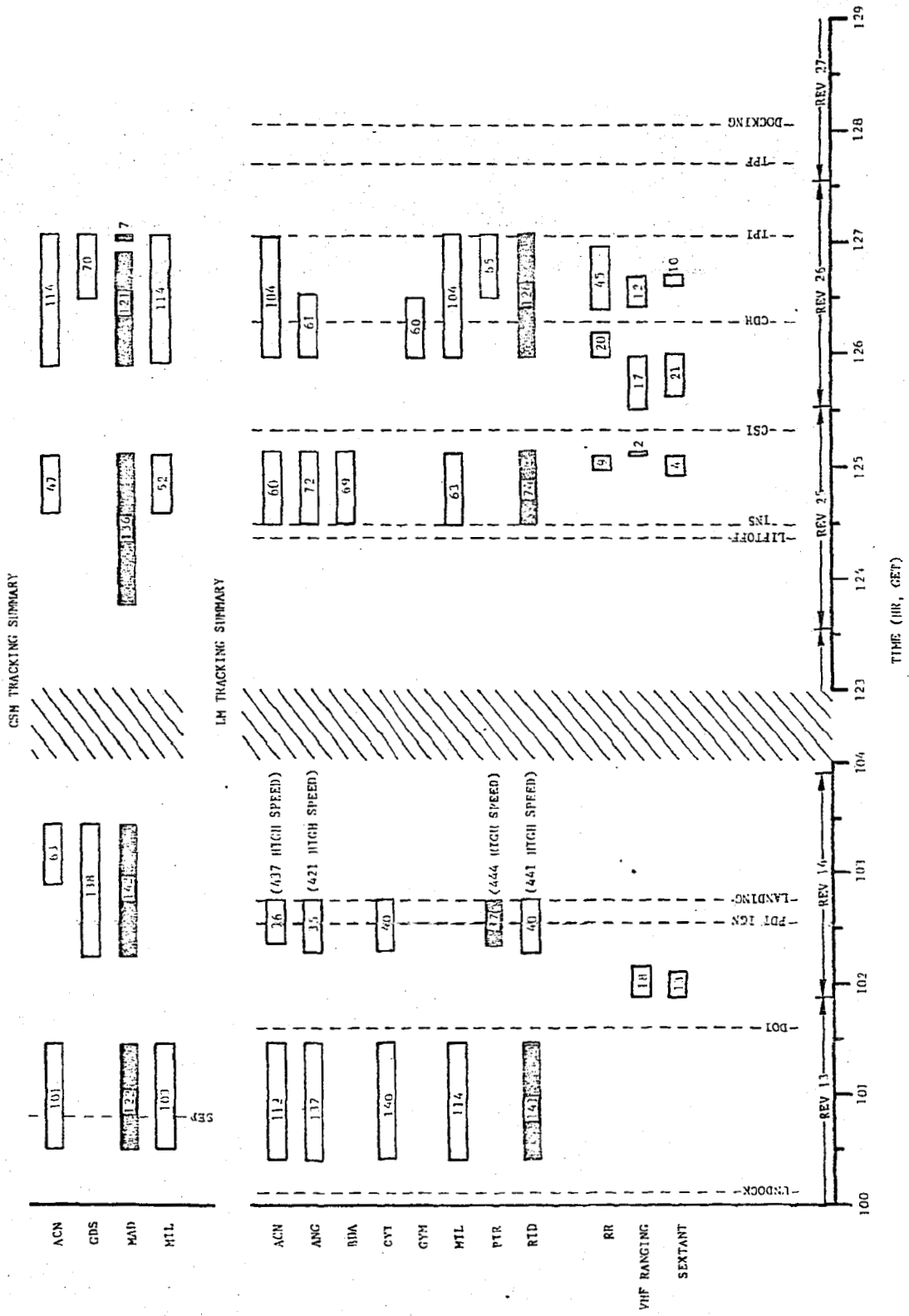


Figure 7-2 Tracking Data (Onboard and Ground Based) Timeline for Apollo 11 Descent and Rendezvous

Table 7.4 -- Apollo Mission 11 BET Summary

Vehicle	Flight Segment	Fit Type	Data Interval		Trajectory Interval		ΔR ft	ΔV ft/sec
			(d:h:m:s) GMT	(d:h:m:s)	(d:h:m:s) GMT	(d:h:m:s)		
LM	Undock to DOI	MSFN F.F.	20:17:48:45	20:18:59:57	20:17:39:00	20:19:08:14.7	--	--
LM	DOI to PDI	O/B, F.F.	20:19:25:03	20:19:43:56	20:19:08:44.5	20:20:04:00	2007.	75.9*
LM	Powered Descent	Modified MSFN ICS	20:19:48:51	20:20:17:45	20:20:04:37.2	20:20:17:45.2	N/A	N/A
LM	Ascent	IMU only	---	---	21:17:53:58.9	21:18:03:08.9	--	--
LM	Insertion to CSI	MSFN F.F.	21:18:03:03	21:18:40:15	21:18:01:21	21:18:51:36	N/A	N/A
LM	CSI to POST TPI	O/B ICS	21:19:02:17	21:20:29:38	21:18:52:00	21:21:15:08	4253.	50.8**
CSM	Revolution 13	MSFN ICS	20:17:54:24	20:18:59:36	20:17:22:00	20:19:22:00	--	--
CSM	Revolution 14	MSFN F.F.	20:19:47:27	20:20:58:21	20:19:22:00	20:21:17:00	1555.	2.5
CSM	Revolution 25	MSFN F.F.	21:17:27:51	21:18:38:33	21:16:57:00	21:18:57:00	--	--
CSM	Revolution 25	MSFN F.F.	21:19:26:15	21:20:36:51	21:18:57:00	21:20:57:00	2147.	3.2

*DOI Burn ΔV = 76.4 ft/sec

**CSI Burn ΔV = 51.5 ft/sec

7.3 ONBOARD TRACKING DATA ANALYSIS

7.3.1 Introduction

Analysis of the LM rendezvous radar data from Apollo missions 9 and 10 and CSM VHF ranging data from Apollo 10 resulted in the conclusions that both data types were of high quality and, in general, produced trajectories consistent with those obtained from ground based tracking data (References 1 and 5).

A similar analysis of the onboard tracking data obtained during the Apollo 11 mission was performed with the following objectives:

- a) Determine the consistency of the LM rendezvous radar data and the CSM VHF ranging data with similar data from Apollo missions 9 and 10.
- b) Using these data as a standard of comparison, evaluate the LM sightings made with the CSM sextant.
- c) Determine the consistency of all onboard data with the ground based data.
- d) Use the onboard data to construct a more accurate LM rendezvous trajectory.

The onboard tracking data were obtained from the downlink telemetry tapes by a special purpose computer program designed to read the tape, and output the desired observations and associated information on punched cards. The format of the punched cards was the specified input to the HOPE Program. Editing of bad data was performed manually.

Onboard tracking data yields a measure of the position and velocity of one vehicle relative to another. It is necessary, therefore, to obtain a good, independent estimate of the trajectory of one vehicle and fix this as a reference trajectory. Since the LM trajectory is perturbed by several maneuvers during the descent and rendezvous mission periods, it is logical to fix the trajectory of the relatively quiescent CSM as the reference.

As discussed in Section 7.2, the CSM trajectory was reconstructed in four single revolution fits from MSFN tracking data. The three segments of interest here were MSFN free flight fits on revolutions 14, 25, and 26.

Trajectories for the LM free flight segments were also reconstructed from MSFN tracking data. The ground based MSFN tracking available for use during the periods of interest are summarized in timeline form in Figure 7-2.

The GSM trajectory was fixed as the reference, and the LM MSFN free flight trajectories were then used to initialize fits based on onboard data in the four segments where relative data were available. A priori confidence values of 10,000 feet were placed on each component of position and 10 feet per second on each component of velocity in the initial conditions.

The reconstruction activities will be discussed in more detail in the following sections. In addition, various tables and figures are included which serve to describe the operations performed and show the accuracy and validity of the data.

7.3.2 Onboard Measurements

Rendezvous radar data were obtained during three periods of the Apollo 11 mission; these were Insertion to CSI (9 observations), CSI to CDH (20 observations), and CDH to TPI (45 observations). As in previous missions, the amount of rendezvous radar data obtained was limited to those periods when telemetry coverage was available.

VHF ranging data were obtained from the CSM during four segments of the flight; these were DOI to PDI (18 observations), insertion to CSI (2 observations), CSI to CDH (17 observations), and CDH to TPI (12 observations). Since only two observations were obtained from Insertion to CSI, no meaningful statistics could be obtained.

Sextant data were obtained during the same periods of flight as were VHF ranging data; 13 observations between DOI and PDI, 4 observations from insertion to CSI, 21 observations from CSI to CDH, and 10 observations between CDH and TPI. Listings of all the data are included in Appendix D.

7.3.3 Evaluation of Onboard Tracking Data

Rendezvous Radar Data

In order to determine the quality of the rendezvous radar data, the residuals (differences between the actual measured value and a measurement

value computed from given CSM and LM trajectories) were examined. The CSM trajectories used in obtaining these residuals were the BET's discussed in Section 7.2 of this report (one rev MSFN free flight fits). The LM trajectories were obtained by using technique (c) described in paragraph 7.2.1 (onboard free flight fits). All available onboard data were used in these fits.

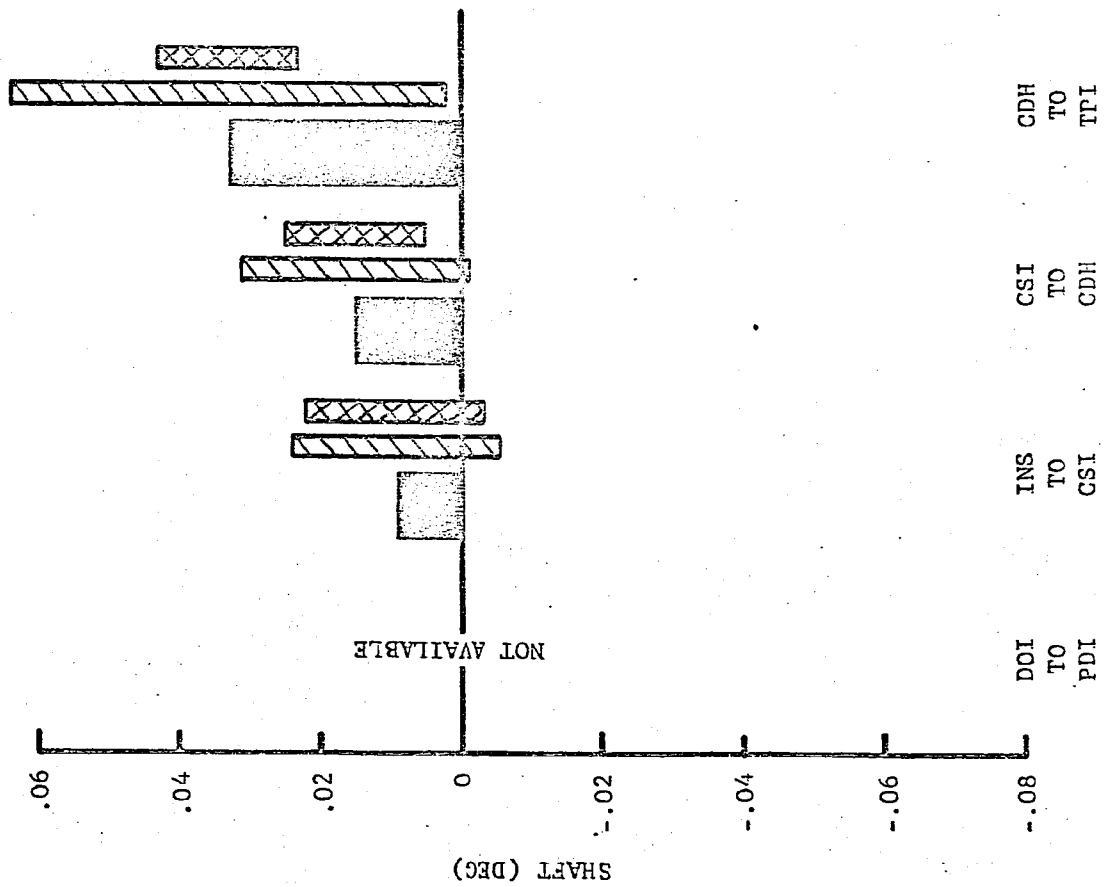
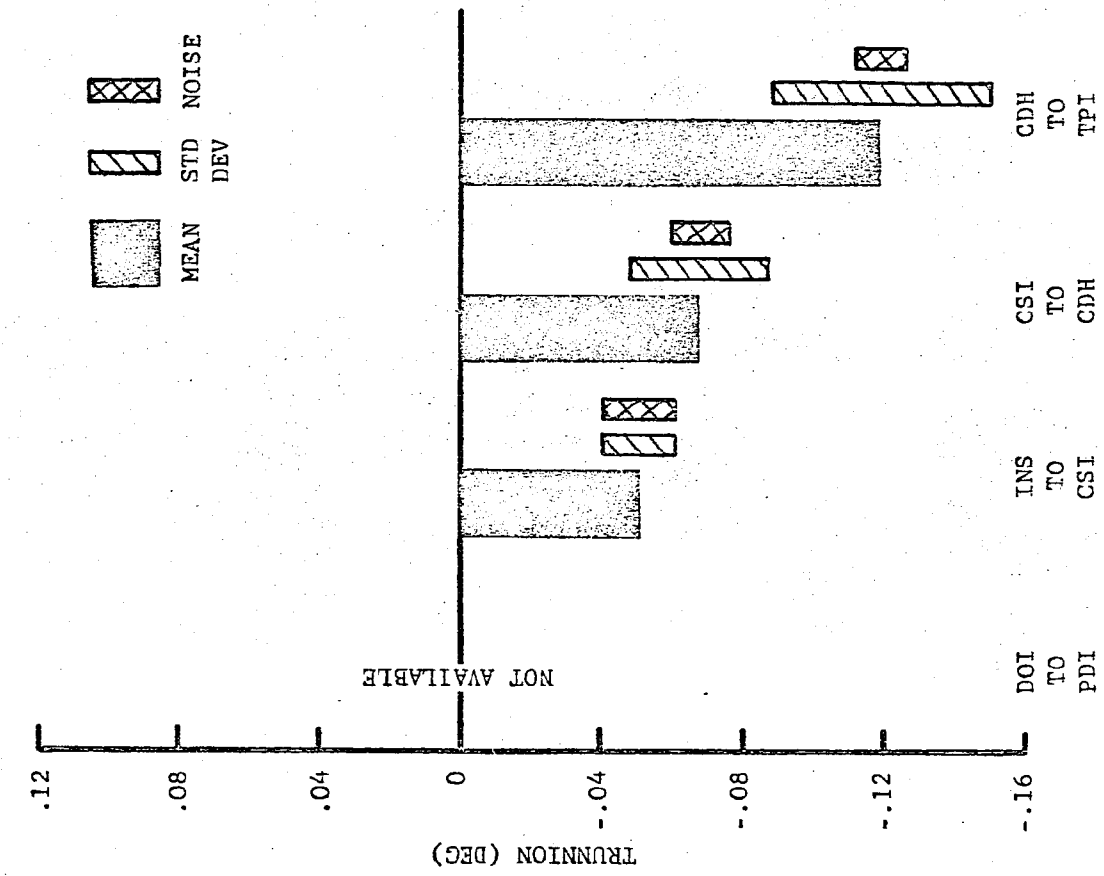
Table 7.5 lists residual statistics (mean, standard deviation, one-sigma noise estimate) computed from the onboard free flight fits of each segment and Figures 7-3a and 7-3b illustrate the results graphically. The data were generally well behaved as can be seen in the residual plots (Figures 7-4 through 7-6). The relatively large differences between the standard deviation and noise computed for shaft and trunnion in the CSI to CDH and CDH to TPI periods can partly be attributed to the fact that both rendezvous radar and sextant data were used in the fit. As the two data sets become more equal in size (weighted effect) or the sampling arcs more coincident, residual statistics deteriorate. This effect is demonstrated by the statistics listed in Table 7.6 which were obtained from fits made with only rendezvous radar data included. Note that when sextant data is eliminated, the RR shaft and trunnion means and deviations decrease in all segments. The shaft statistics are still relatively high (especially in the CDH and TPI period), indicating that a systematic error is still present in the shaft measurement. It should also be noted that the rendezvous radar residual statistics from Apollo 10 exhibited a similar characteristic (Reference 1). In Apollo 10, the standard deviations for both shaft and trunnion measurement are relatively large in the CDH to TPI period (no sextant data were included in Apollo 10 solutions). The large mean values seen in Table 7.5 are also a result of the inclusion of sextant data in the solution data sets. When only rendezvous radar data was included, the mean values decreased to near zero values.

The range residual statistics exhibited characteristics similar to the Apollo 10 data. When VHF ranging data is removed from the solution data set, standard deviations decrease and become, in two segments, almost equal to noise estimates. The mean values also approach zero, indicating that no bias is present.

Table 7.5 Summary of Rendezvous Radar Residual Statistics

	Insertion To CSI	CSI To CDH	CDH To TPI	
Shaft (deg)	.009	.015	.033	Mean
	.015	.016	.031	S. Dev.
	.013	.010	.010	Noise
Trunnion (deg)	-.051	-.068	-.119	Mean
	.010	.019	.031	S. Dev.
	.010	.008	.007	Noise
Range (feet)	79.	75.	55.	Mean
	144.	63.	92.	S. Dev.
	39.	37.	27.	Noise
Range Rate (fps)	.604	-.243	-.305	Mean
	.173	.339	.277	S. Dev.
	.6278	.6278	.6278	Q. E.*

* Quantization Error.



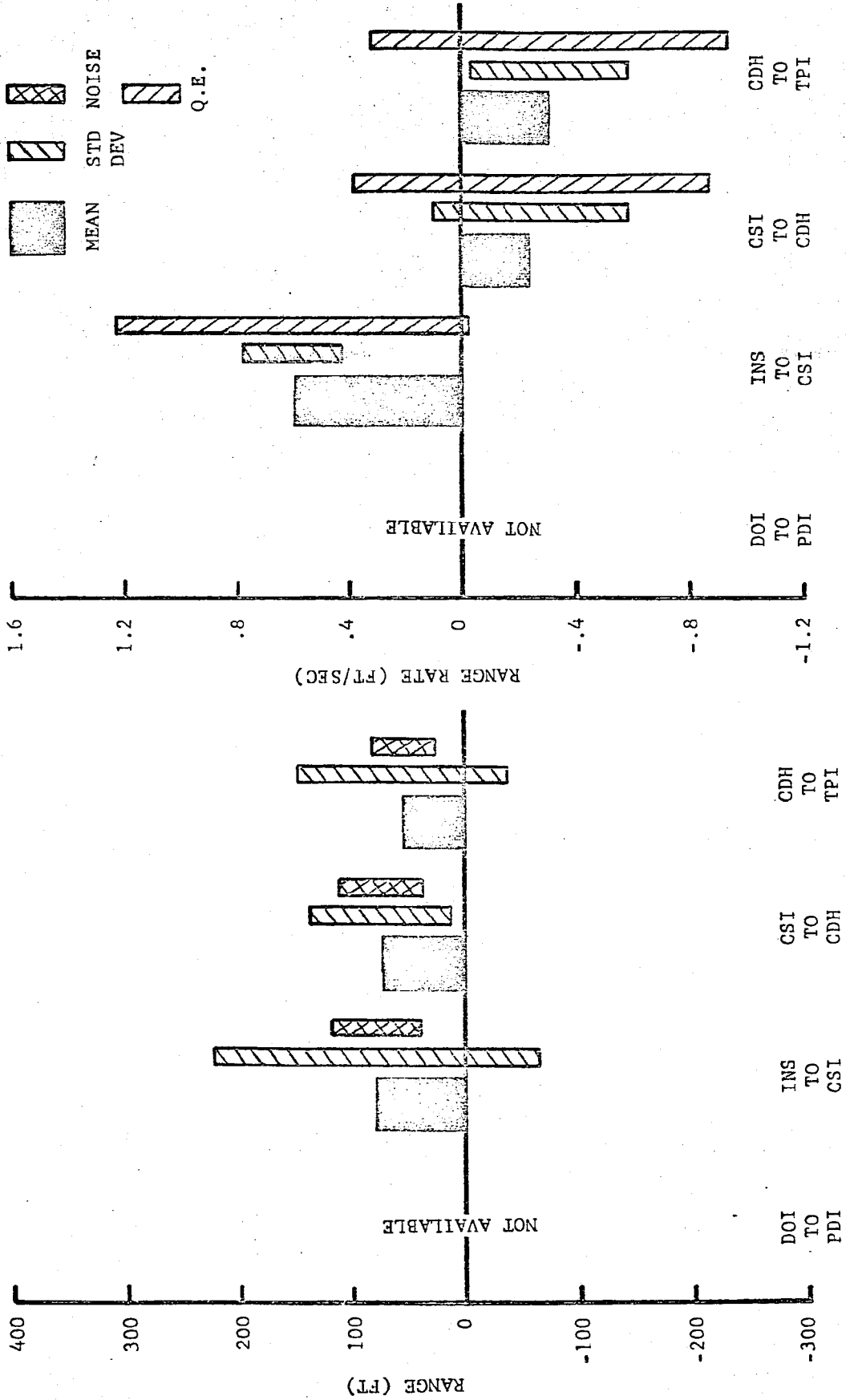


Figure 7-3b Rendezvous Radar Range and Range Rate Residual Statistics

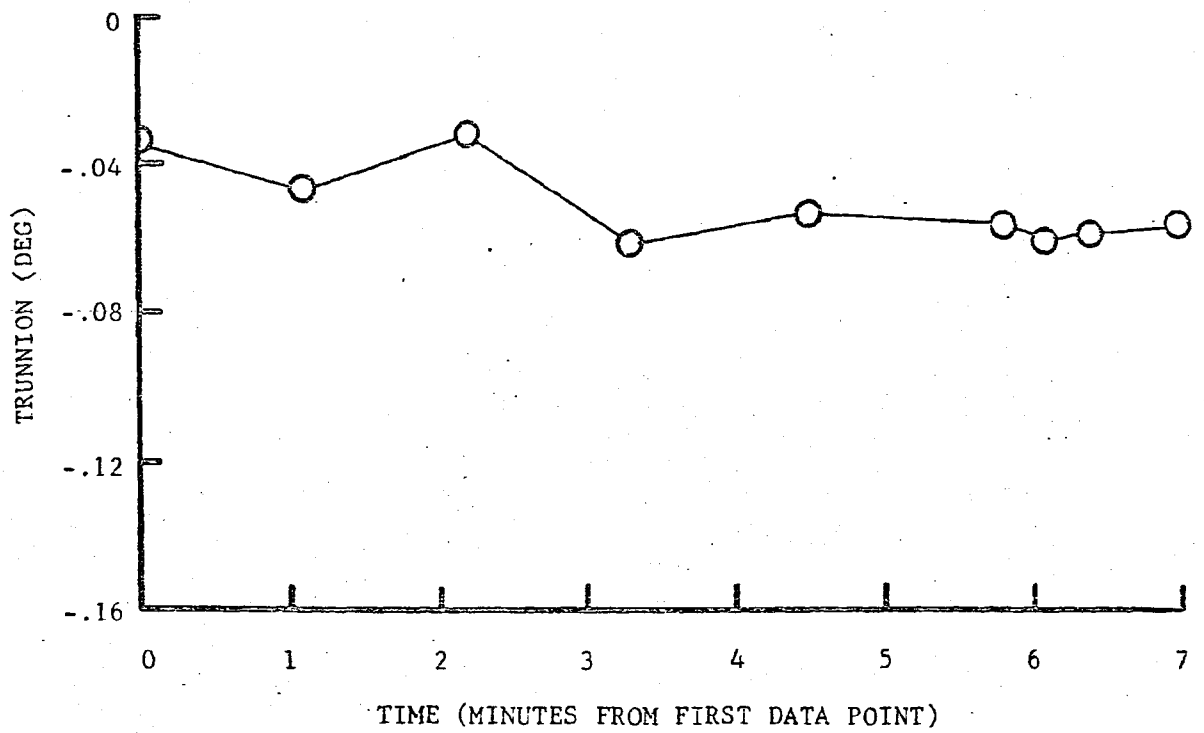
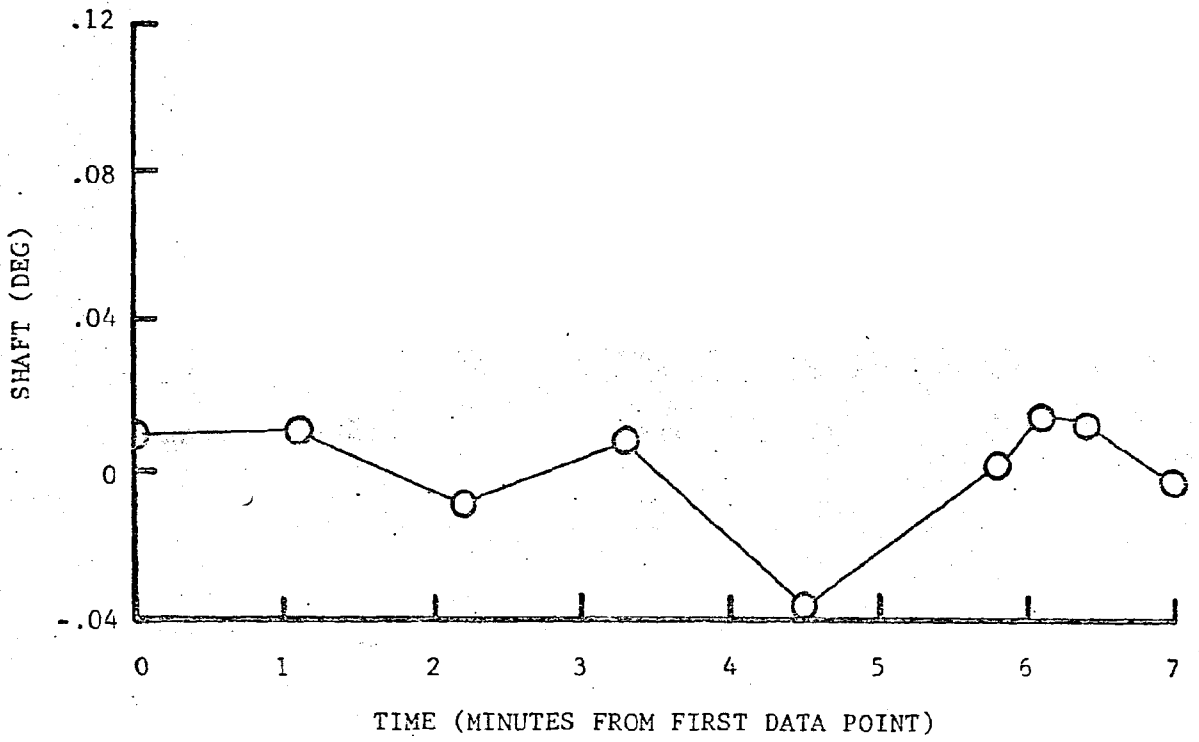


Figure 7-4 Rendezvous Radar Residuals (Insertion to CSI)

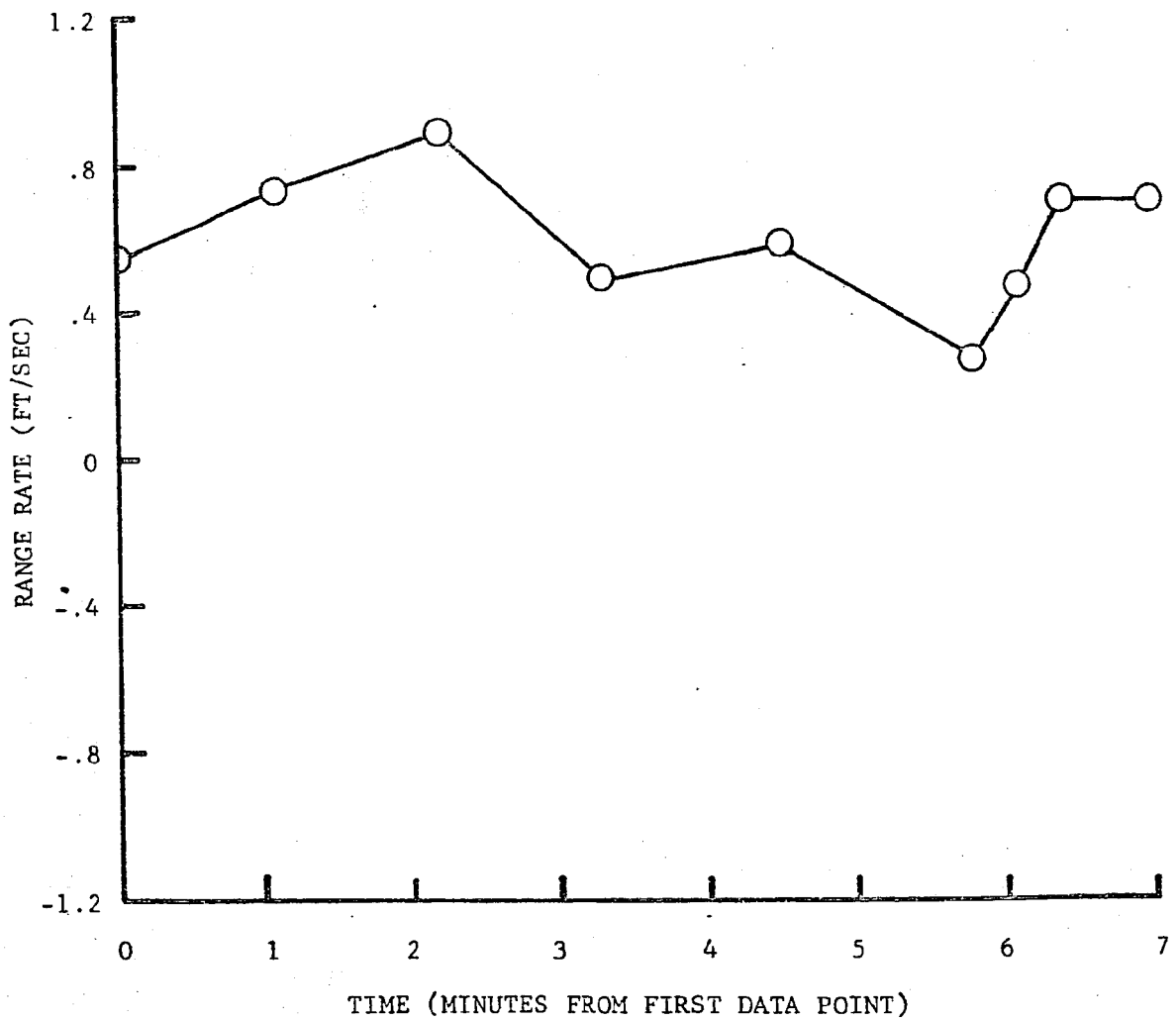
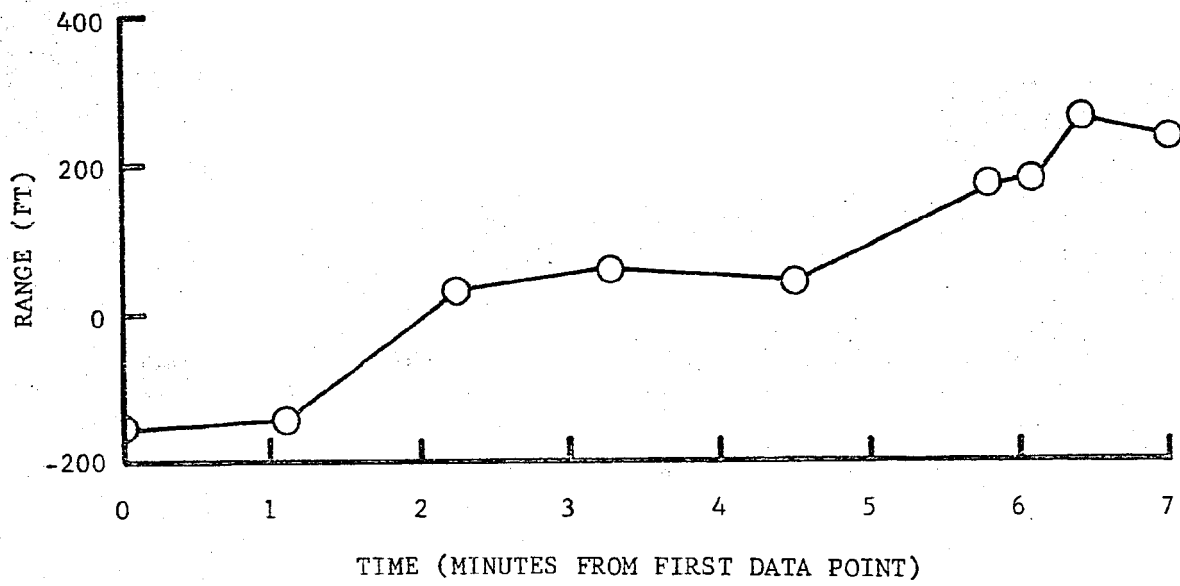


Figure 7-4 Concluded

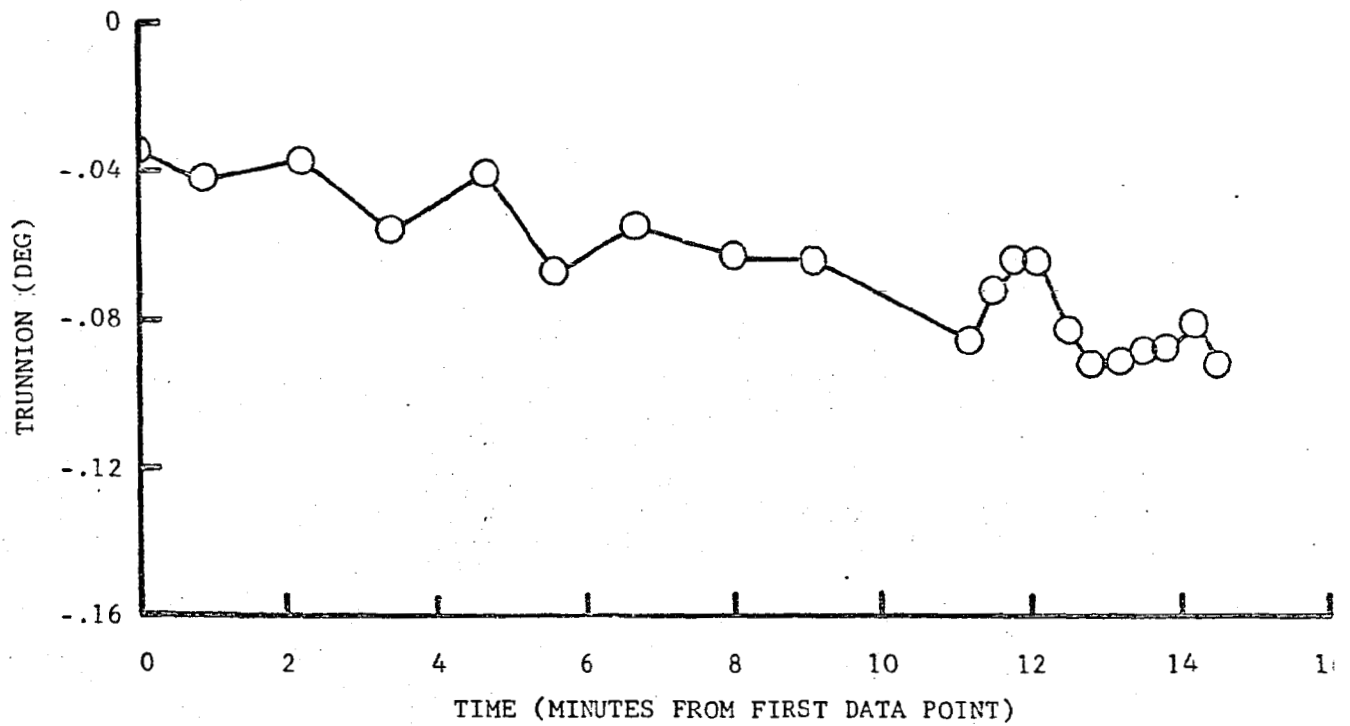
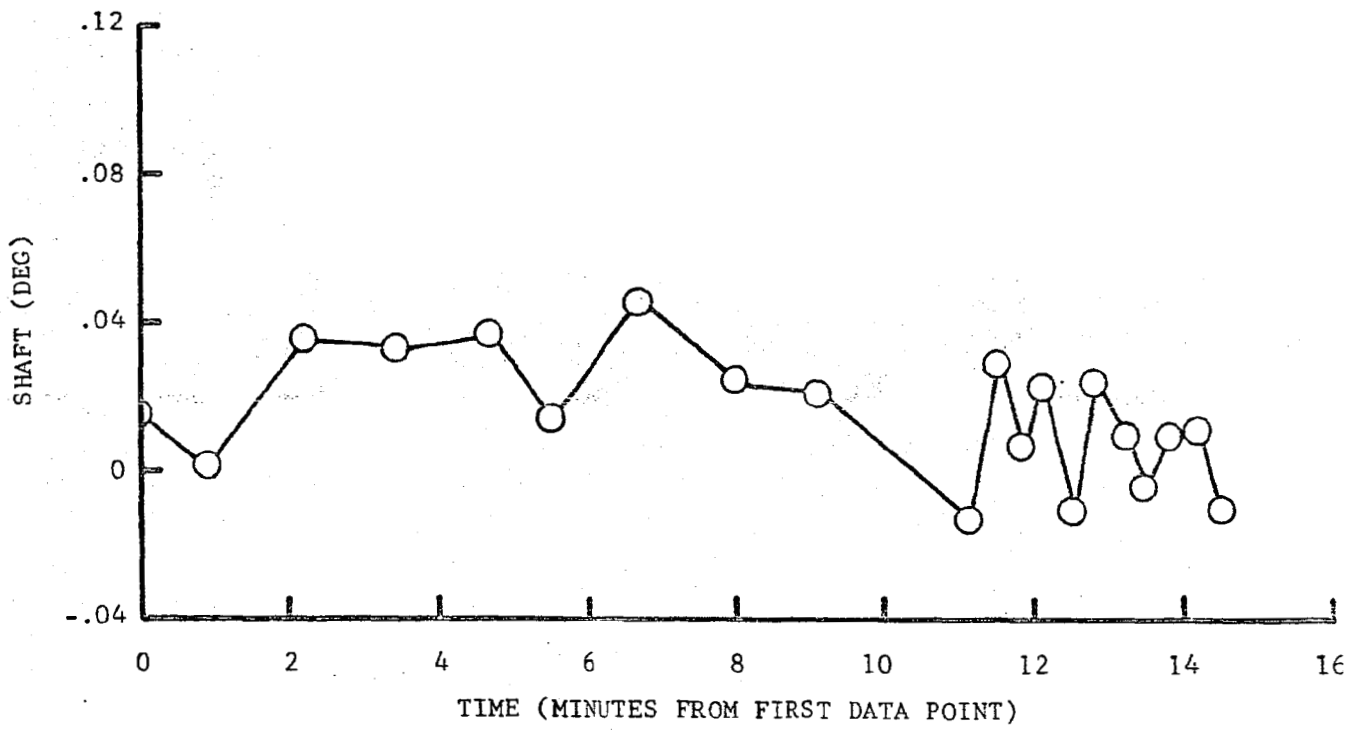


Figure 7-5 Rendezvous Radar Residuals (CSI to CDH)

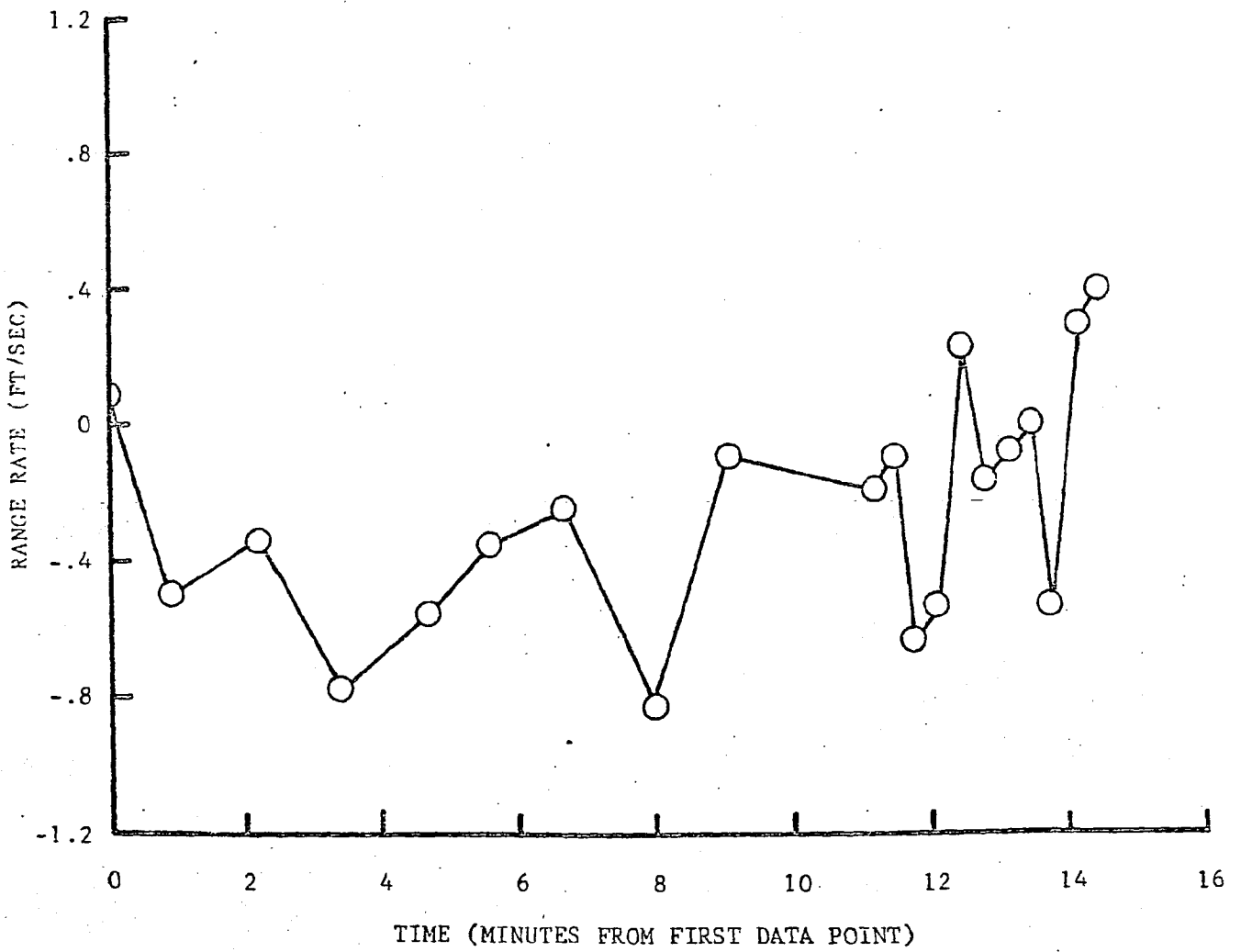
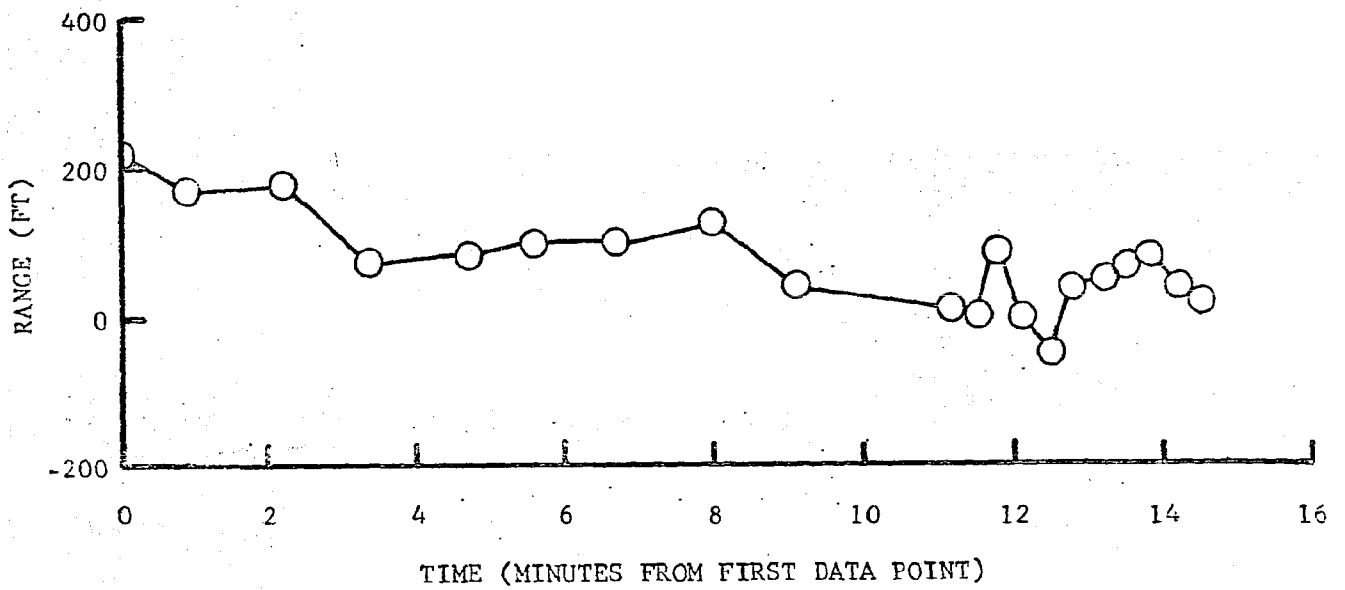


Figure 7-5. Concluded

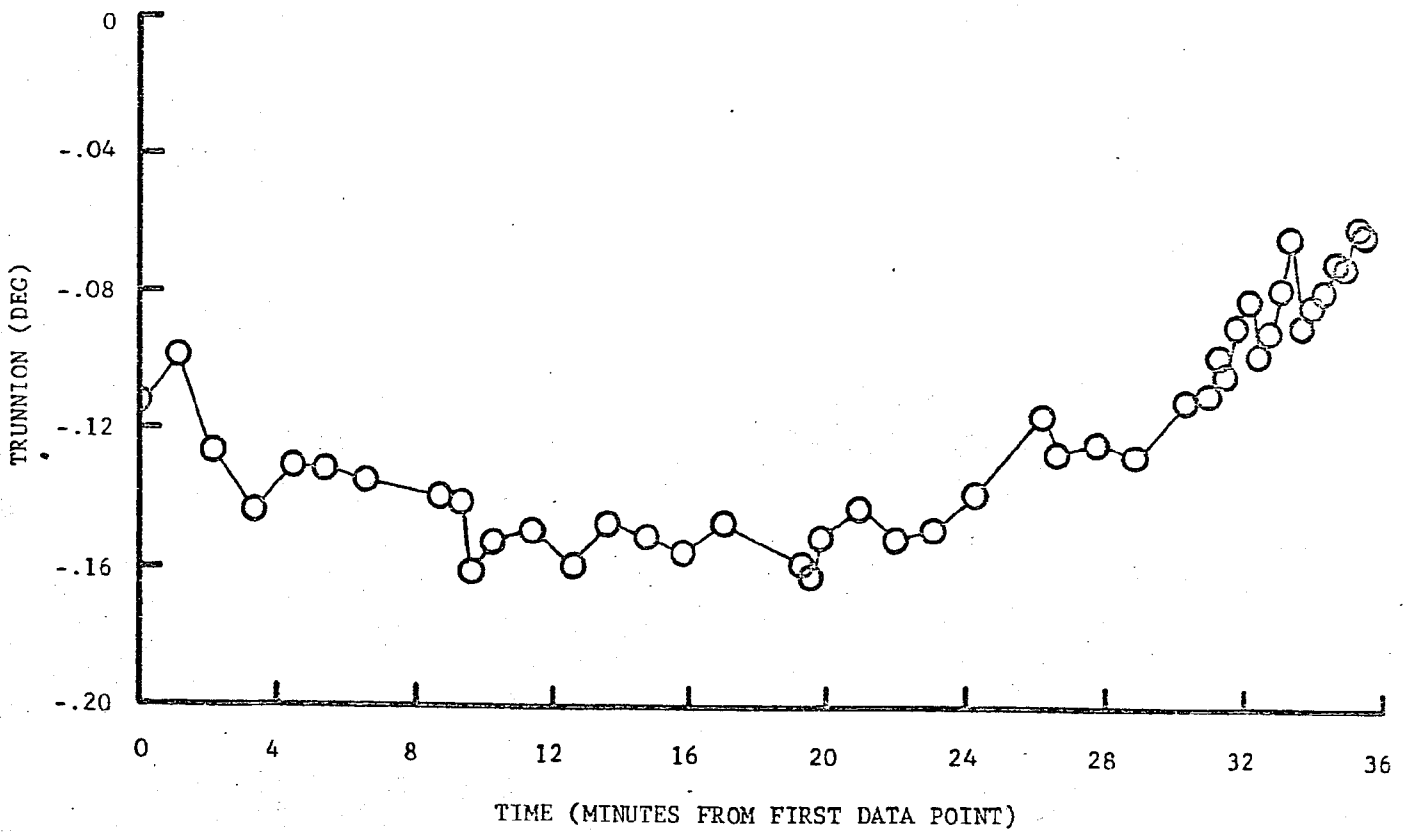
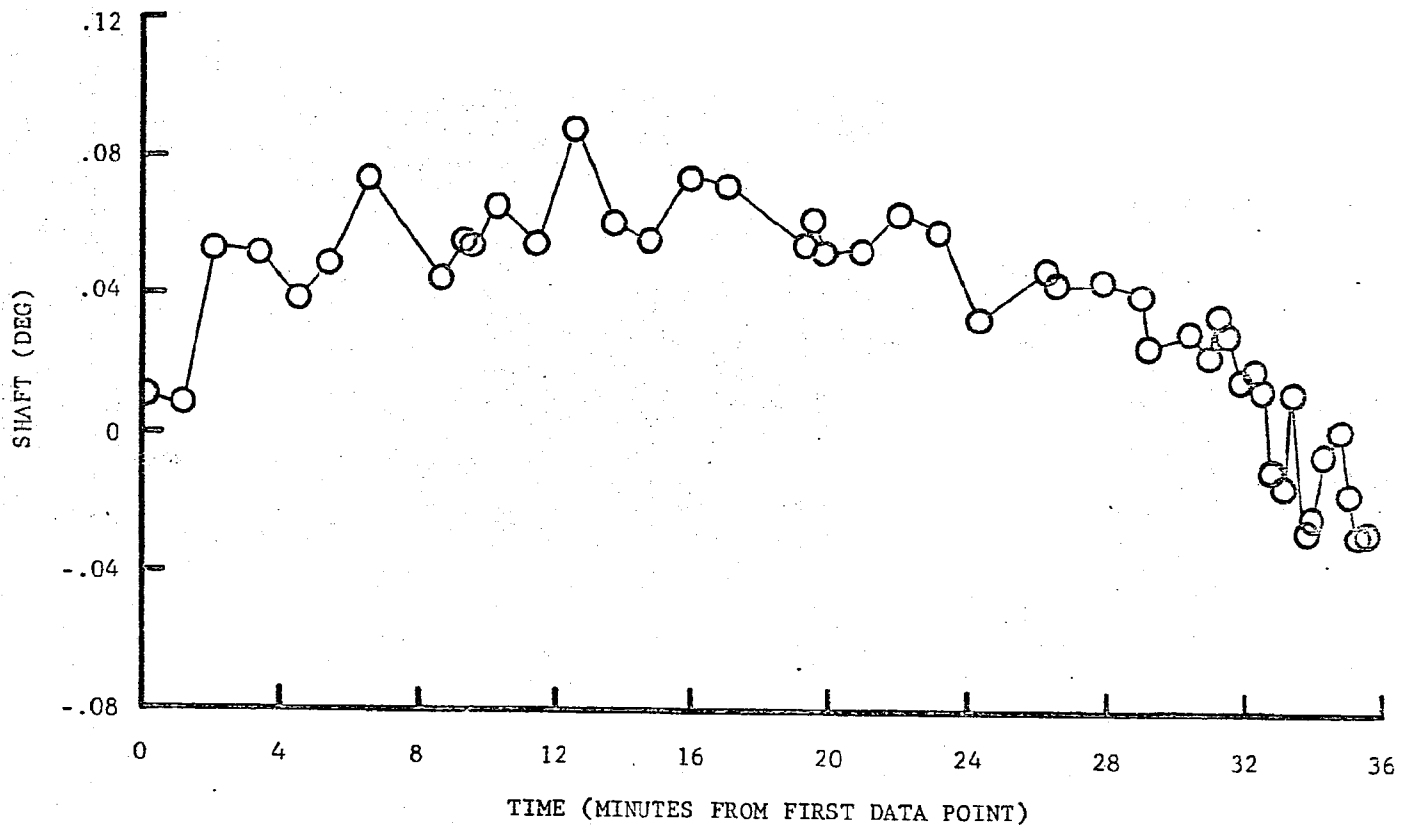


Figure 7-6 Rendezvous Radar Residuals (CDH to TPI)

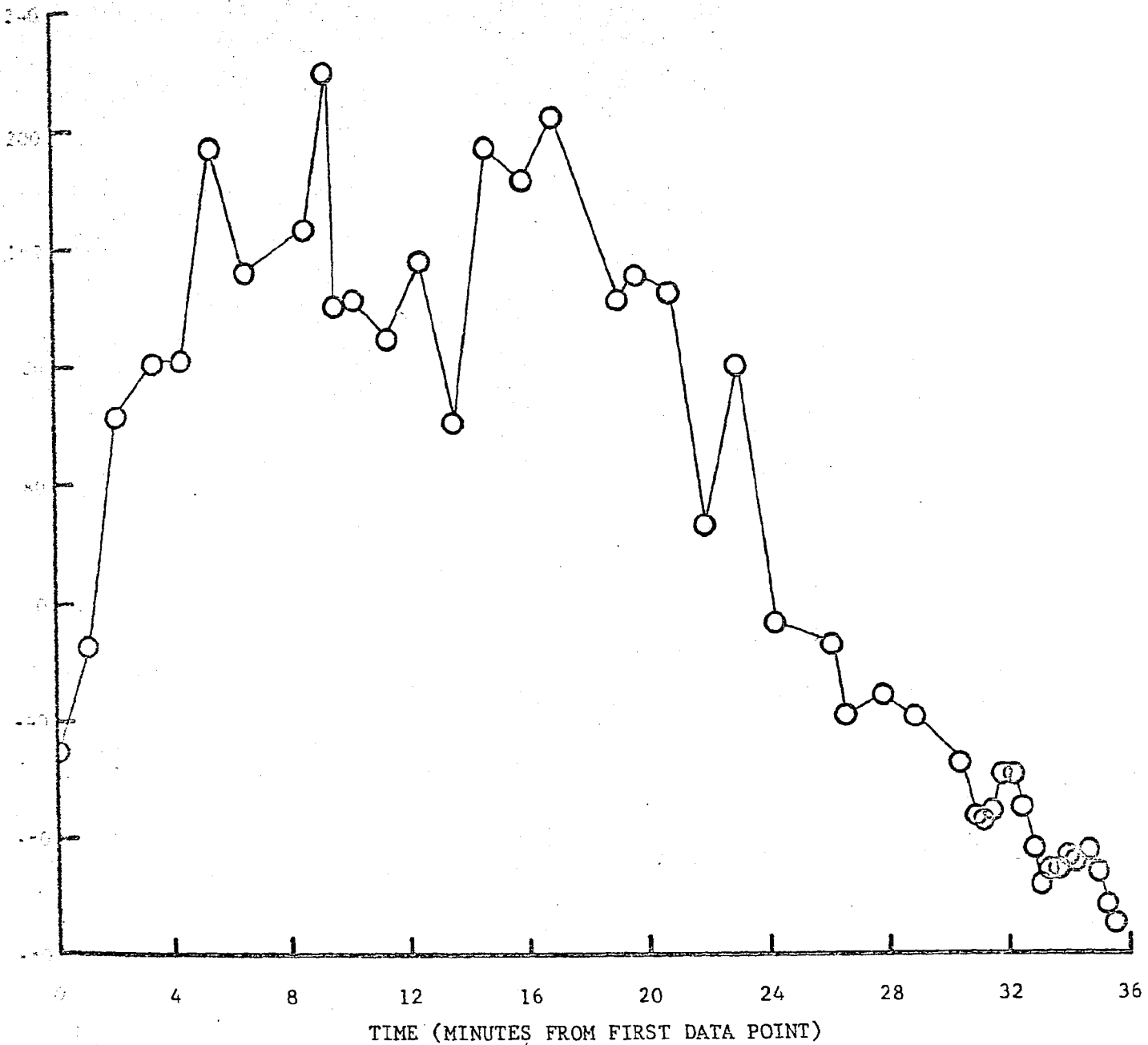


Figure 7-6 Continued

Reproduced from
best available copy

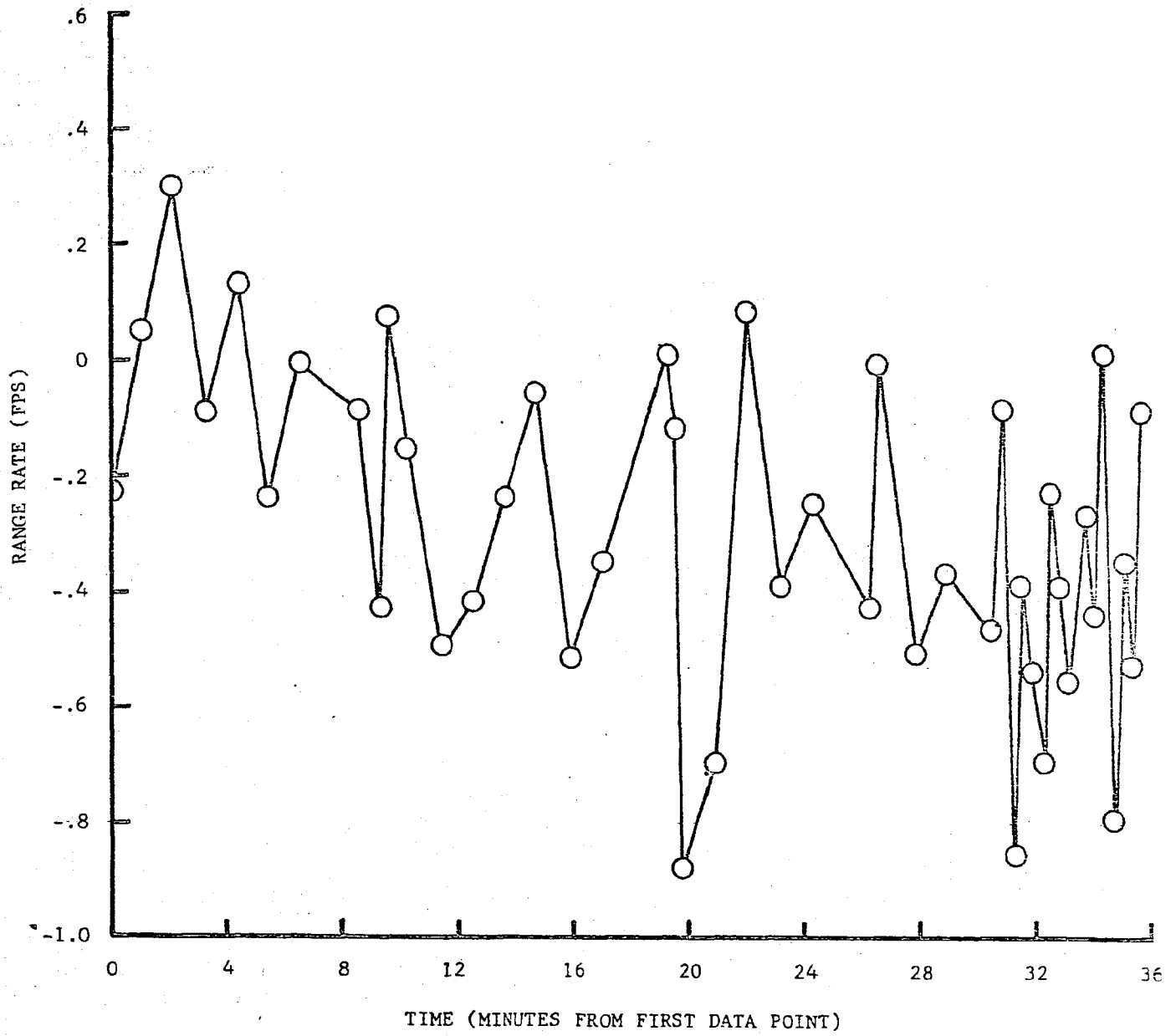


Figure 7-6 Concluded

Table 7.6 Rendezvous Radar Only Solution Residual Statistics

	INS-CSI	CSI-CDH	CDH-TPI	
Shaft	-.00007	.00028	.018	Mean
	.015	.014	.031	S. Dev.
Trunnion	-.00013	.00002	-.0005	Mean
	.0066	.0074	.0085	S. Dev.
Range	.55	.026	2.89	Mean
	39.	37.2	72.45	S. Dev.
Range Rate	-.343	-.105	-.196	Mean
	.187	.314	.304	S. Dev.

The range rate residuals were also of good quality. Mean values were all less than the downlink readout error (.6278 fps).

One sigma noise calculations for shaft, trunnion, and range rate from three missions are plotted as a function of average range in Figures 7-7 through 7-9. These figures show that the Apollo 11 noise estimates compare well with similar estimates from missions 9 and 10. Note that no definite trend is apparent in the angular noise as relative range varies. Figure 7-9 does seem to indicate, however, that the noise estimate for the range measurement does increase as average range increases. The Apollo 11 noise estimates for all three observables appear to be generally smaller than those obtained from previous missions.

VHF Ranging Data

Table 7.7 contains a summary of VHF ranging data residual statistics obtained from onboard free flight fits made over the three segments where adequate amounts of data were available. Figures 7-10, 7-11, and 7-12 contain plots of these residuals. Since only two observations were obtained from the insertion to CSI segment, only the DOI to PDI (18 observations), CSI to CDH (17 observations), and CDH to TPI (12 observations) segments are considered.

The VHF ranging data were generally of good quality. As expected, the smallest mean value was obtained during the DOI to PDI period when VHF ranging was the only range data type measuring the distance between vehicles. The mean values become increasingly large as more rendezvous radar data are included in the data set or as the data arcs become coincident in time. This can be seen in the large mean value for the CDH to TPI period. This large mean, however, is still within the bias specification limit of ± 270 feet.

Figure 7-13 shows that the calculated noise values compare favorably with Apollo 10 results and are relatively constant when compared to those obtained from Apollo 10. The residual statistics listed in Table 7.7 are illustrated graphically in Figure 7-14.

Sextant

The residual statistics shown in Table 7.8 indicate that the CSM sextant is a very accurate instrument. Sextant observations were obtained

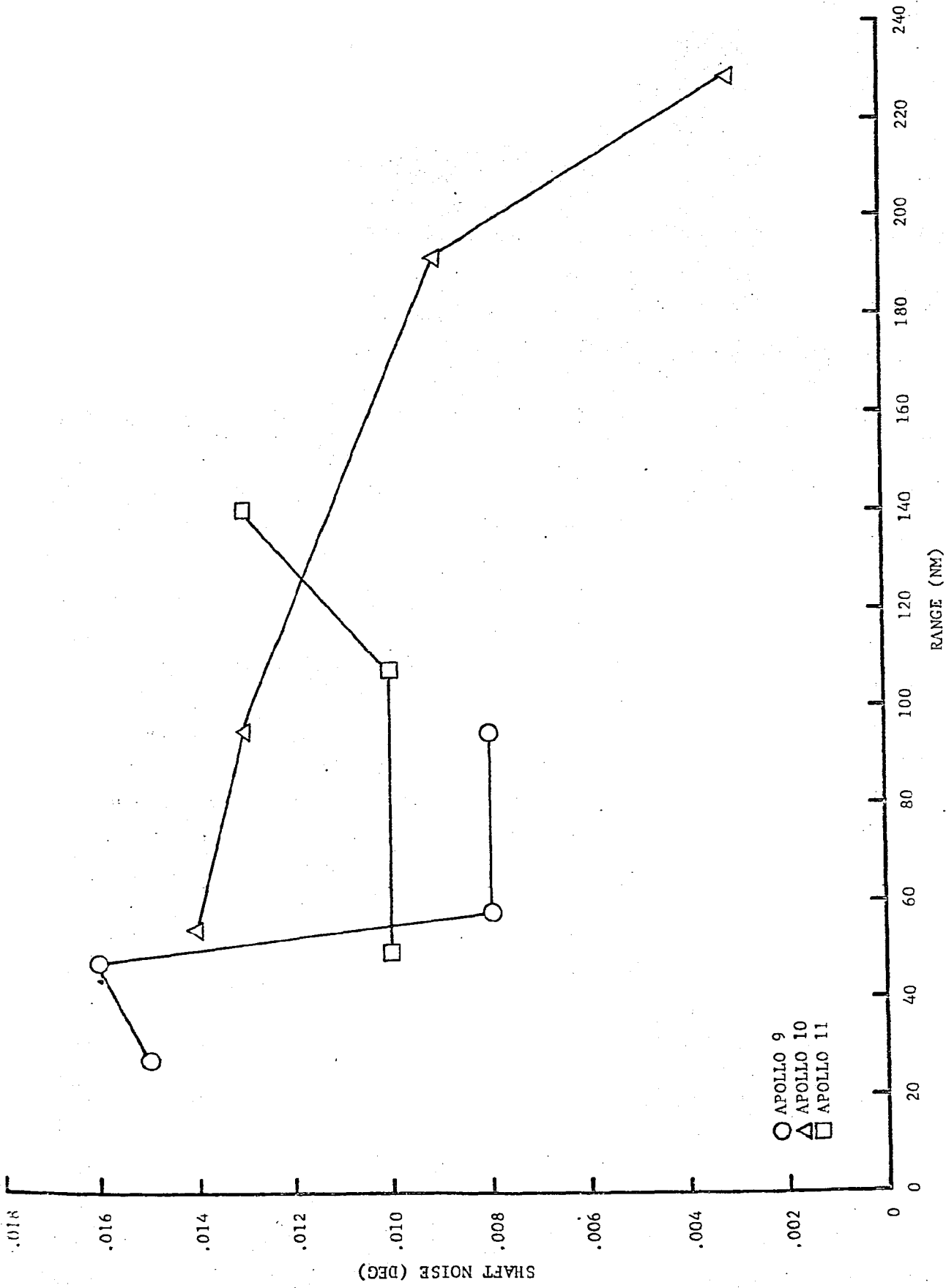


Figure 7-7 Rendezvous Radar Shaft Noise as a Function of Average Range

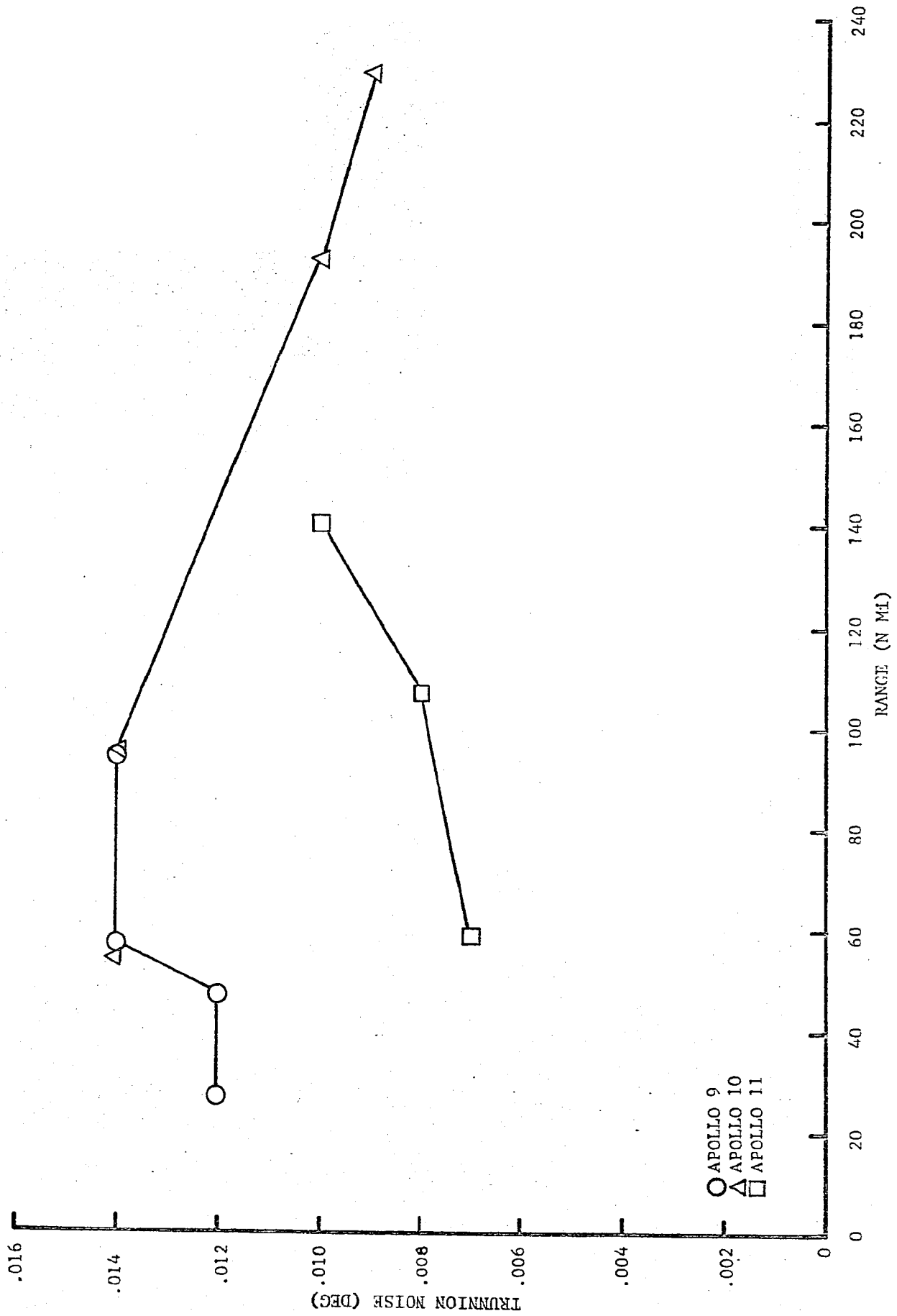


Figure 7-8 Rendezvous Radar Trunnion Noise as a Function of Average Range

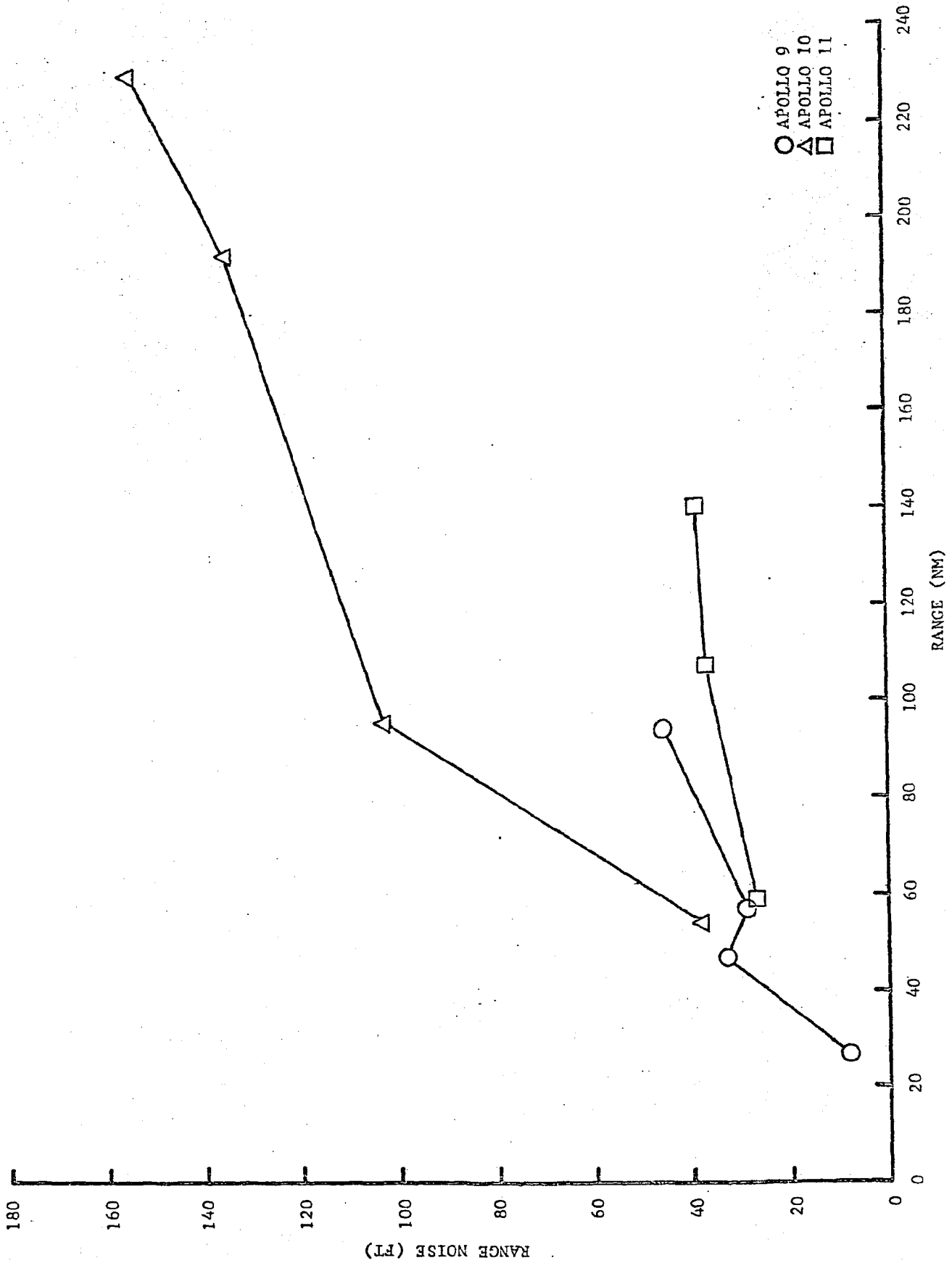


Figure 7-9 Rendezvous Radar Range Noise as a Function of Average Range

Table 7.7 Summary of VHF Ranging Residual Statistics

	DOI-PDI	CSI-CDH	CDH-TPI	
Range	-26.	-86.	-216.	Mean
(feet)	74.	104.	48.	S. Dev.
	23.	23.	19.	Noise

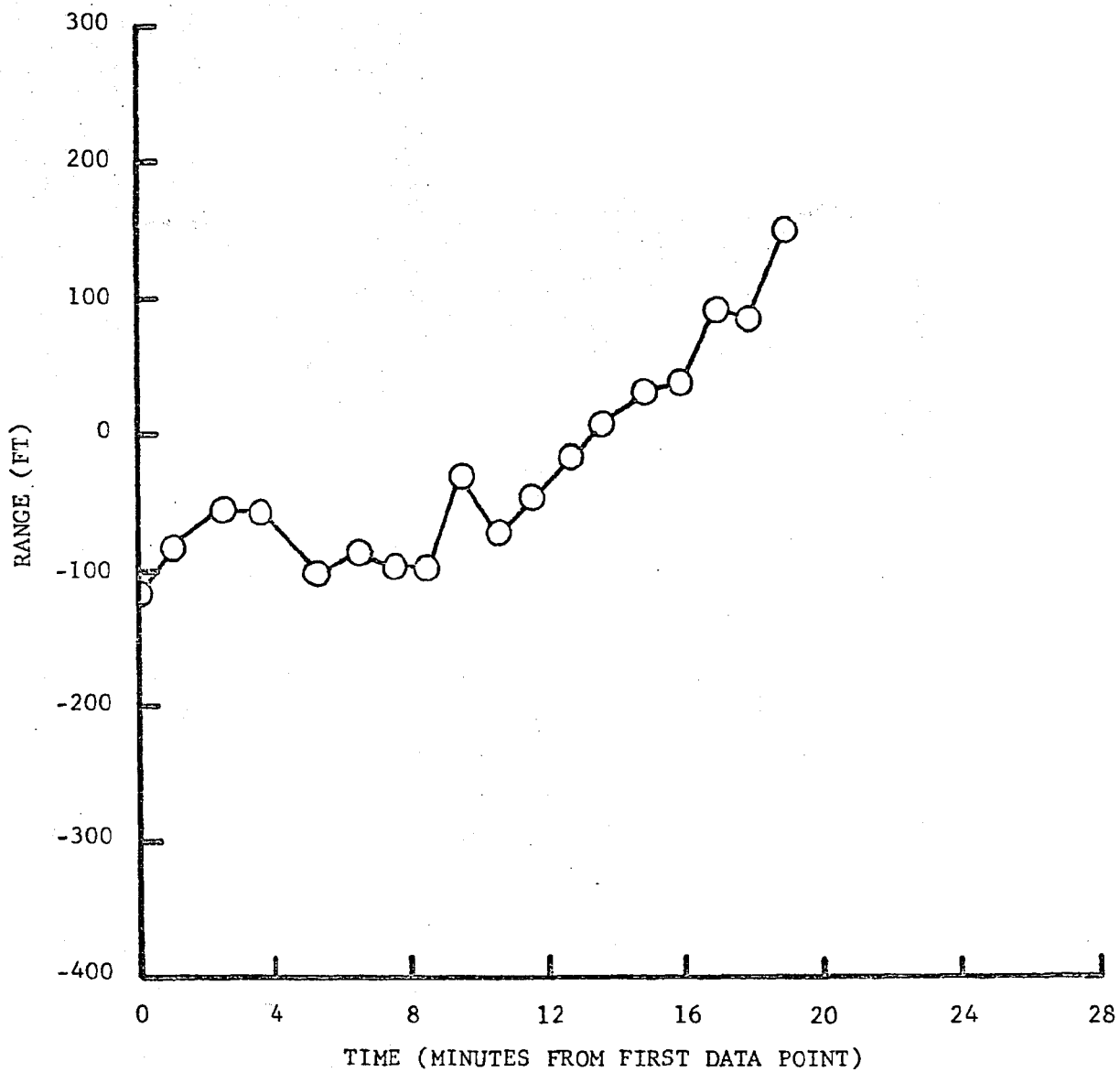


Figure 7-10 VHF Ranging Residuals (DOI to PDI)

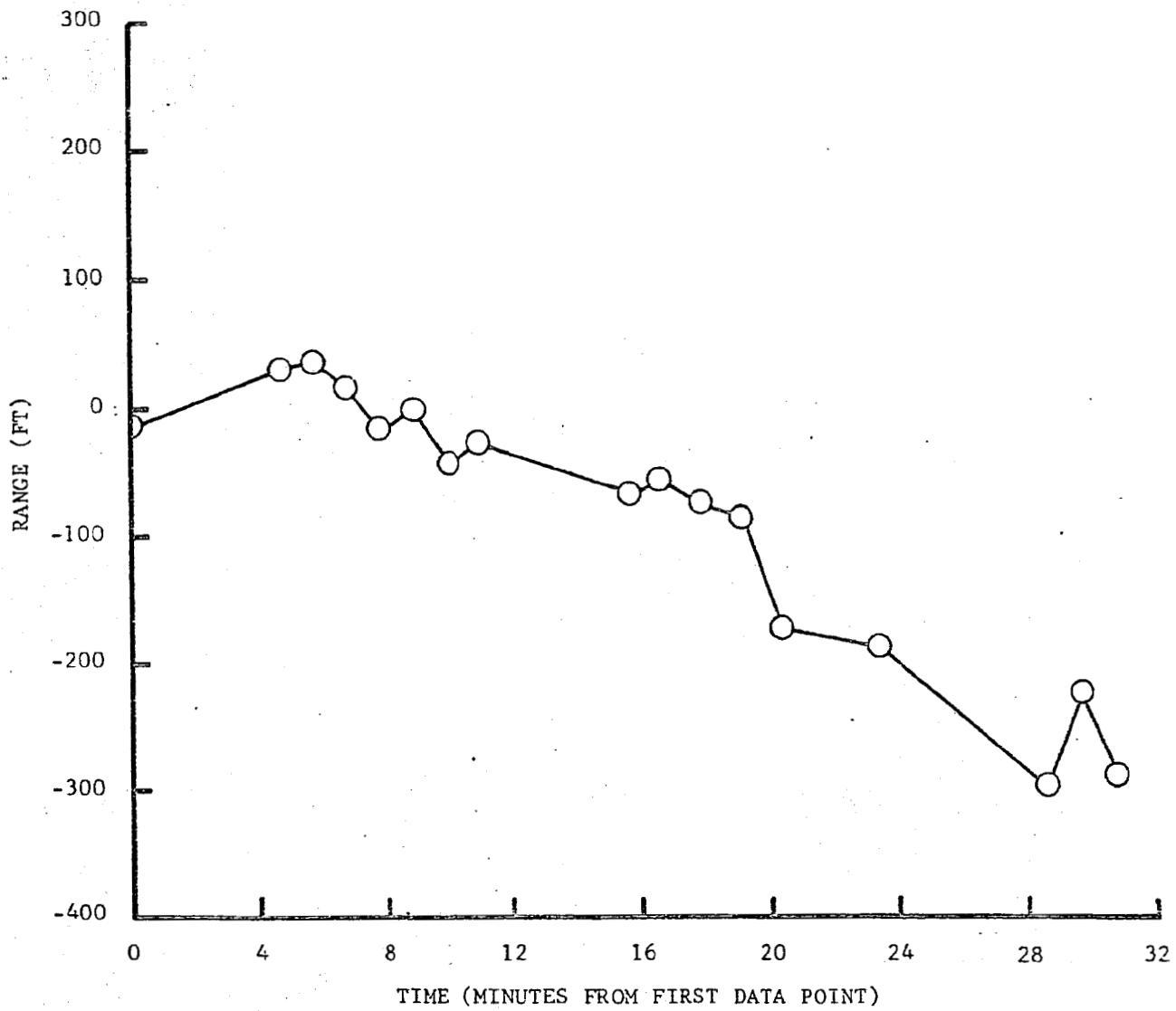


Figure 7-11 VHF Ranging Residuals (CSI to CDH)

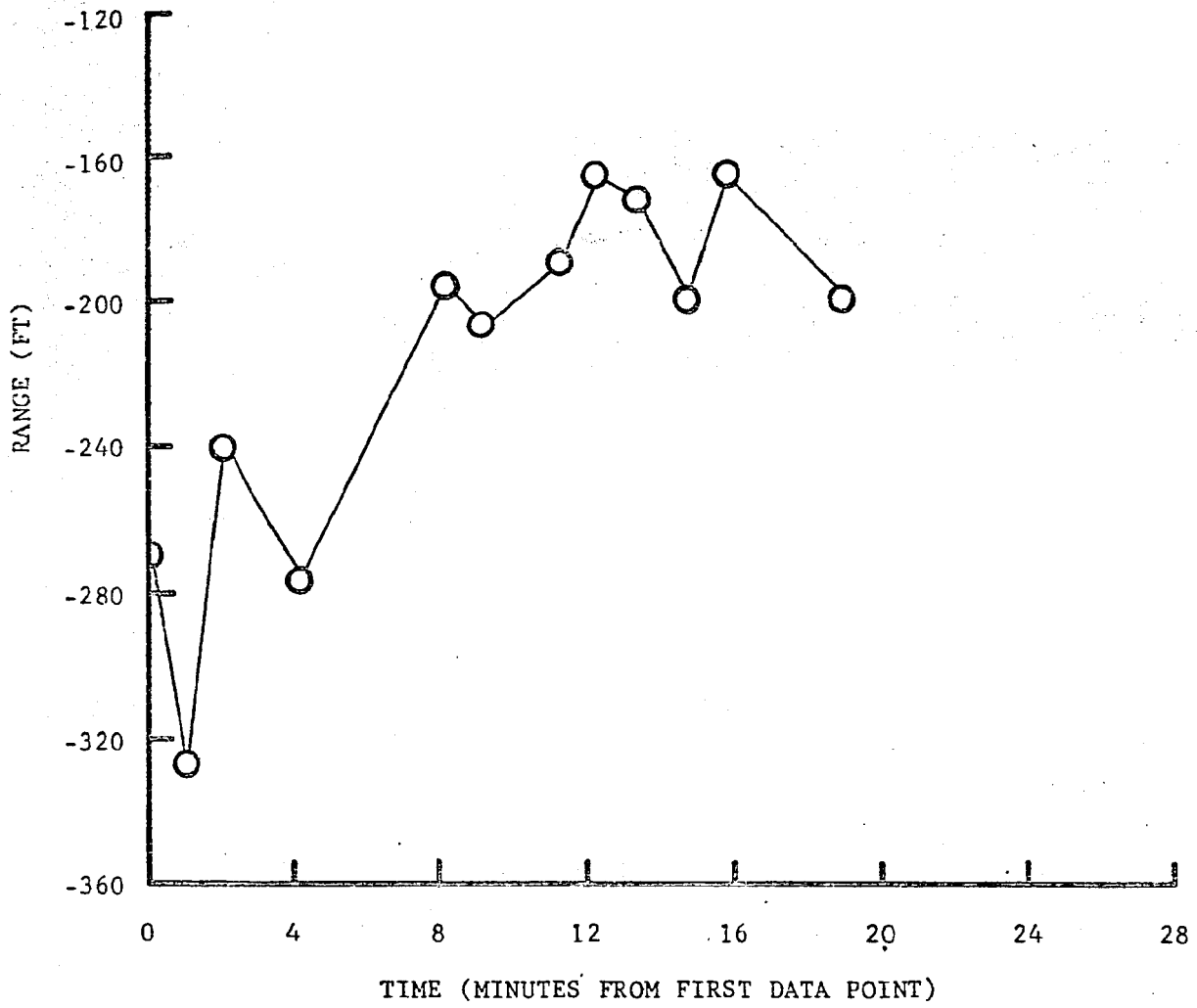


Figure 7-12 VHF Ranging Residuals (CDH to TPI)

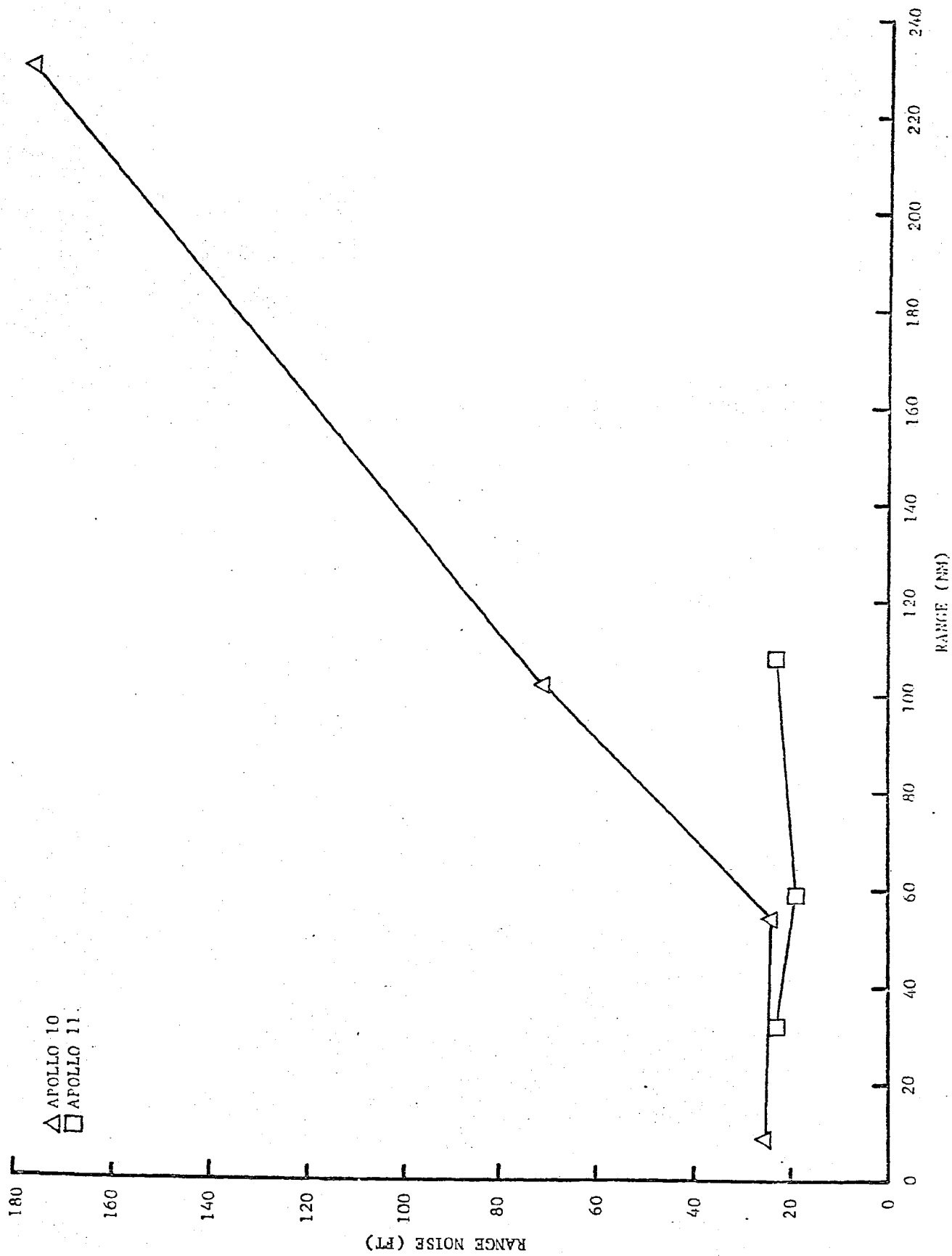


Figure 7-13 VIF Ranging Noise as a Function of Average Range

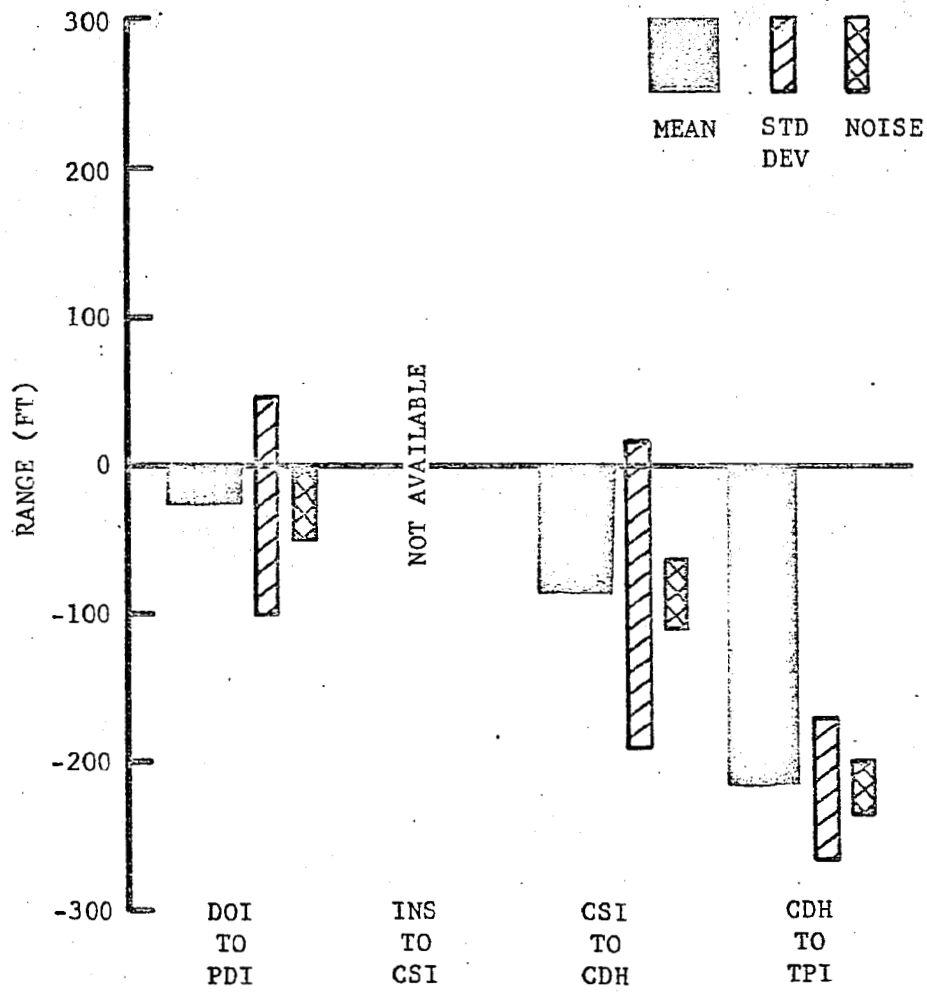


Figure 7-14 VHF Ranging Residual Statistics

Table 7.8 Summary of Sextant Residual Statistics

DOI To PDI	Insertion To CSI	CSI To CDH	CDH To TPI	Mean	S. Dev.	Noise
-.001	.0005	.0002	.0017			
.015	.005	.011	.013			
.014	Insufficient Data	.010	.011			
-.0004	-.000009	.0004	-.001			
.004	.003	.006	.008			
.003	Insufficient Data	.006	.011			

Shaft
(deg)

Trunnion
(deg)

in four of the free flight segments; DOI to PDI (13 sightings), insertion to CSI (4 sightings), CSI to CDH (21 sightings), CDH to TPI (10 sightings).

The close agreement of the standard deviations with the noise estimates and the very small means listed indicate that there are essentially no biases in either angle.

The residual patterns (Figures 7-15 through 7-18) are very well behaved. The random noise estimates (Figure 7-19) compare well with rendezvous radar angular noise estimates and no trend can be identified with respect to average range. Note the good agreement with the Apollo 9 noise estimates plotted in Figure 7-19.

Onboard Tracking Data Consistency

In order to determine the consistency of trajectories reconstructed from onboard tracking data with those obtained from MSFN tracking data, state vector comparisons were made over the propagation intervals. These comparisons were made in a UVW-type coordinate system and the results are presented in graphic form. In the figures presented, RZ is the negative of the U or radial component, RX is the V or downrange component, and RY is the negative of the W or crossrange component of a system centered at the CSM. RXD, RYD and RZD are the respective velocities.

Three LM trajectories were obtained for the period from DOI to PDI. Figure 7-21 plots (as a function of time) the out-of-plane component of LM position relative to the CSM for a MSFN free flight trajectory, an onboard data free flight trajectory, and the final BET (combined high speed MSFN and onboard tracking data). It can be seen that the addition of onboard tracking data drastically improves this component of position. Figures 7-22 and 7-23 show the differences between relative trajectories obtained from the MSFN and from the onboard tracking free flight fits. There are large differences in the trajectories which are primarily due to the poor quality of the MSFN free flight fit, but the comparisons do show that the downrange and radial components compare fairly well inside the MSFN data arc.

Reproduced from
best available copy

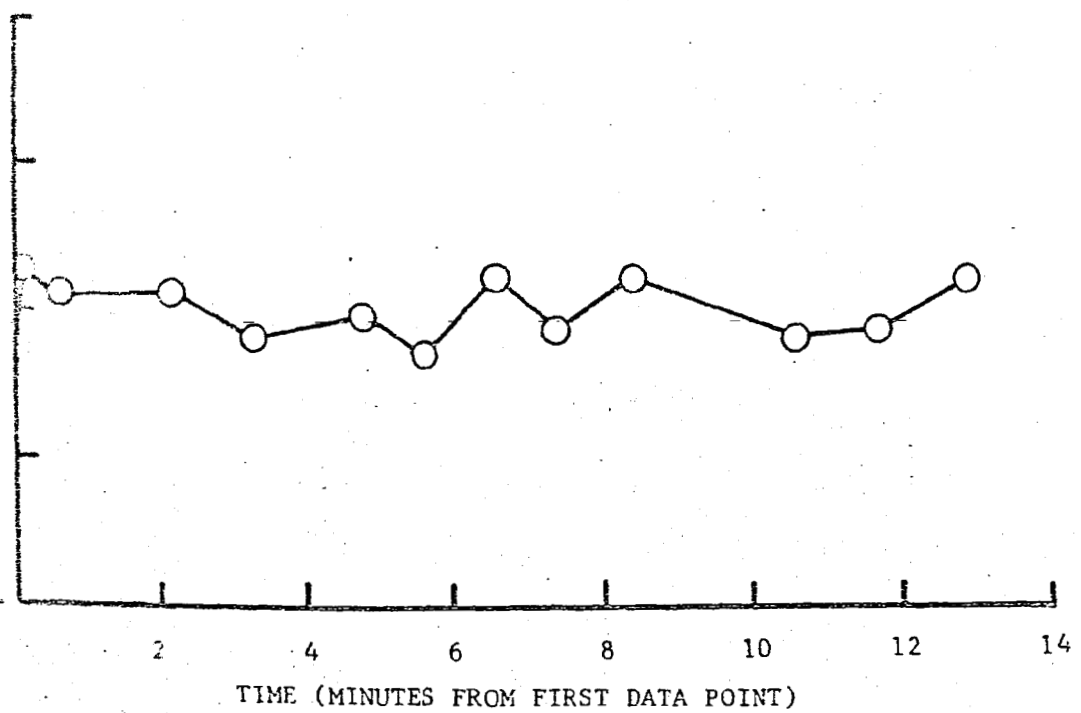
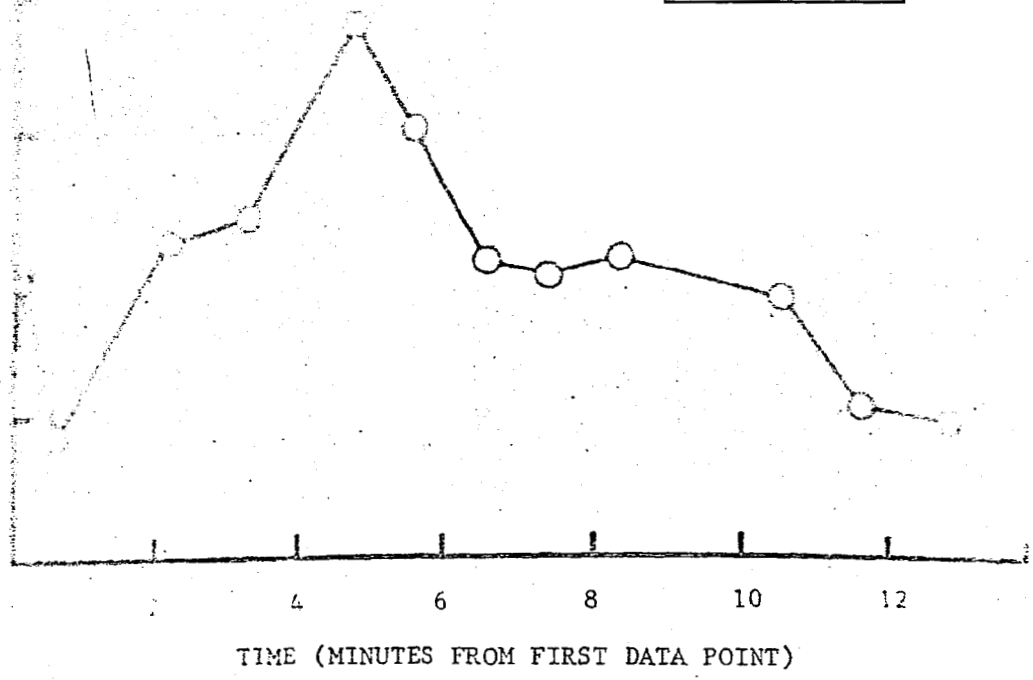


Figure 7-15 Sextant Residuals (DOI to PDI)

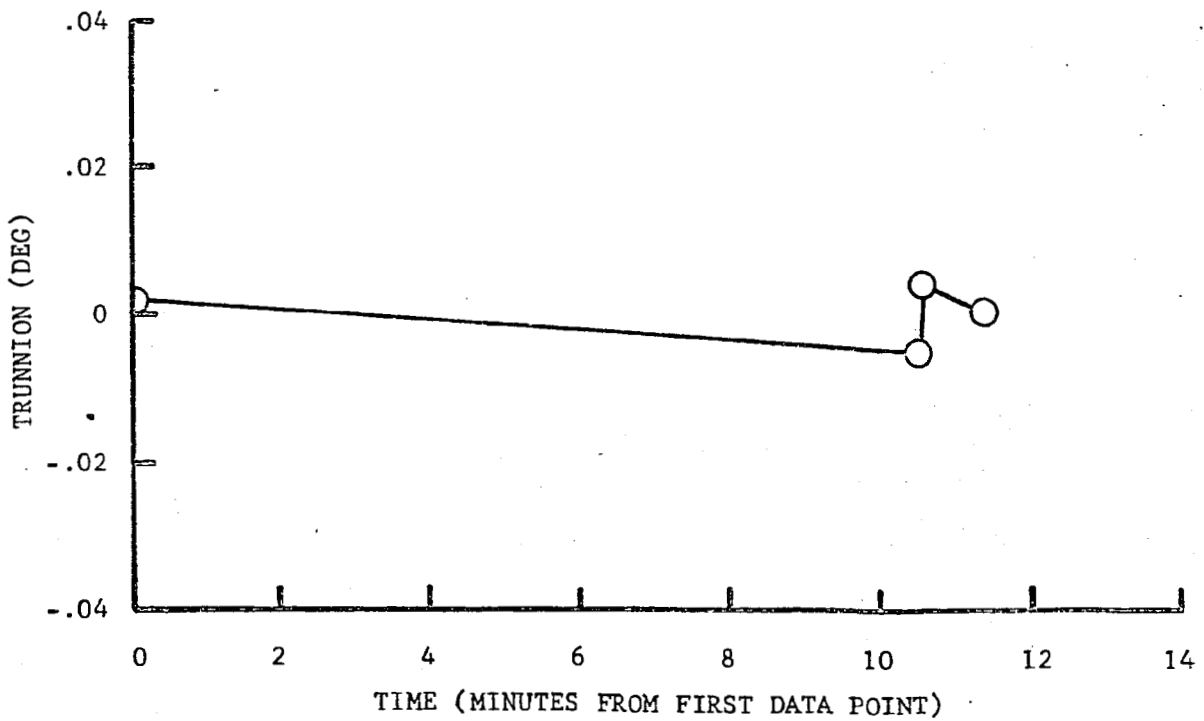
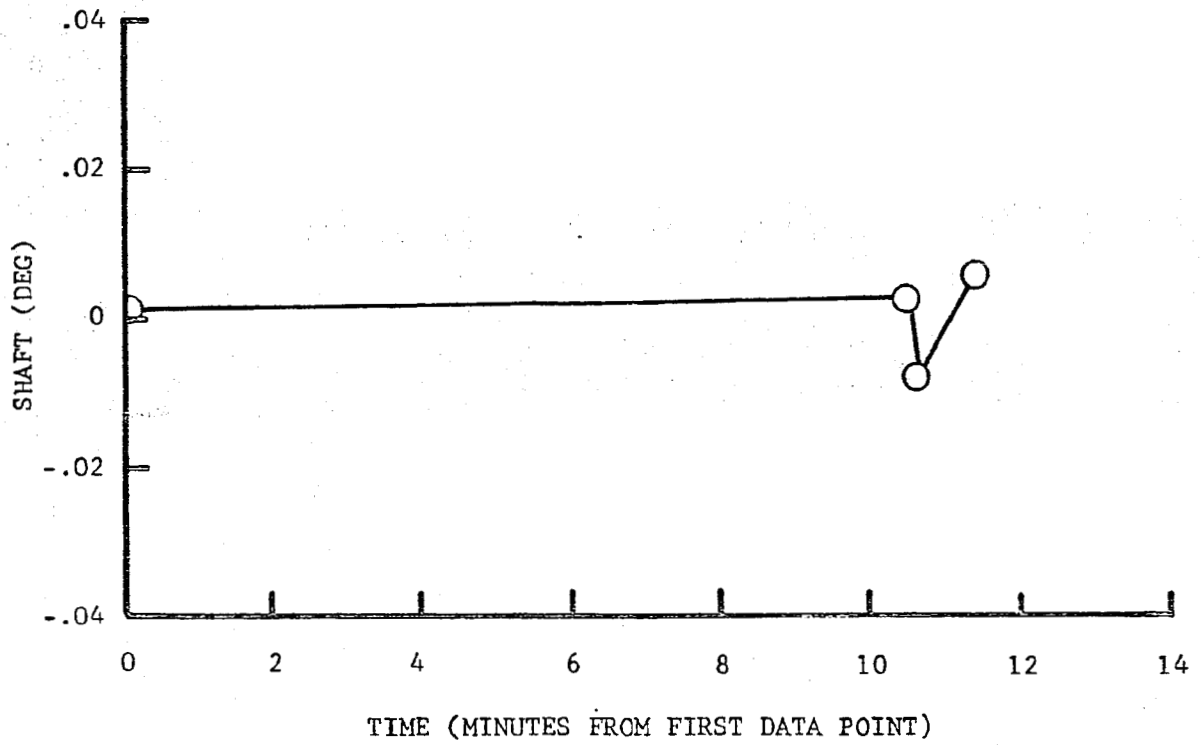


Figure 7-16 Sextant Residuals (Insertion to CSI)

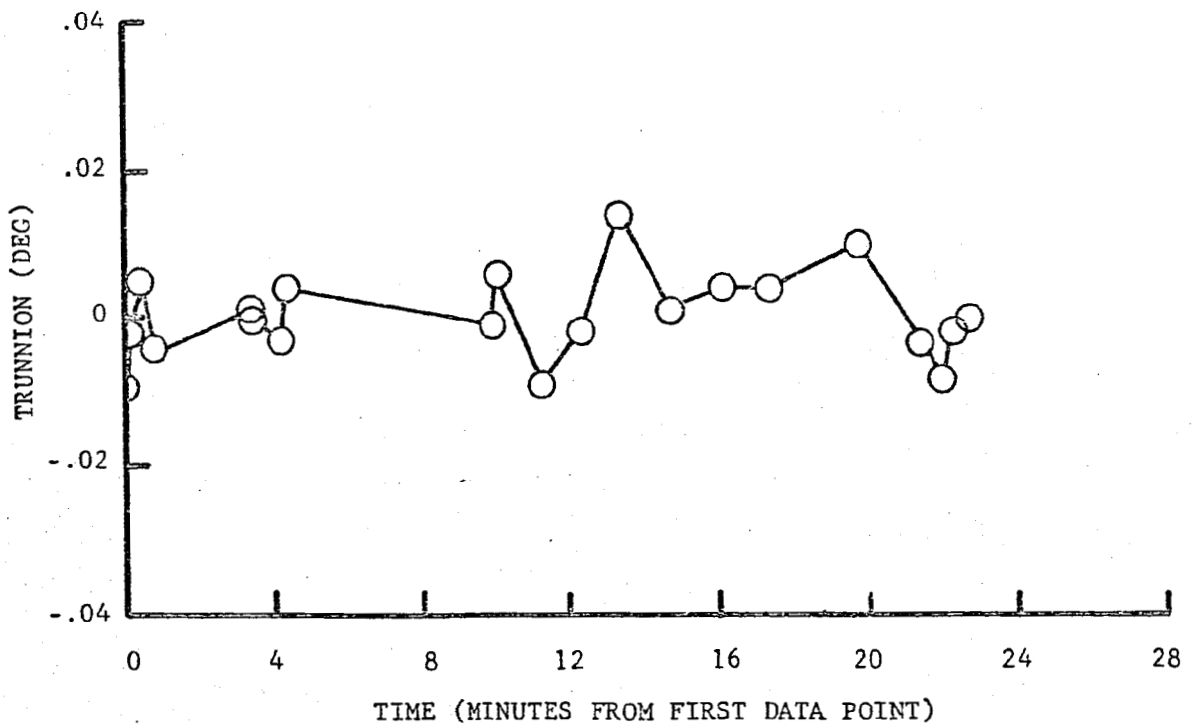
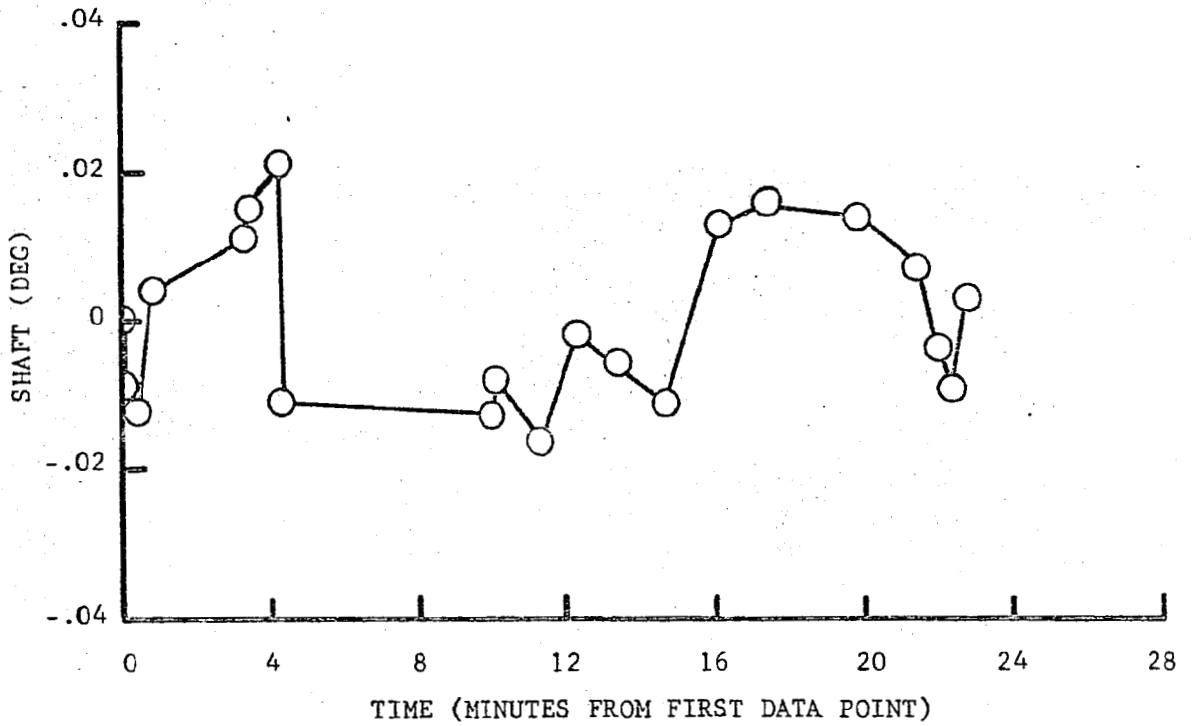


Figure 7-17 Sextant Residuals (CSI to CDH)

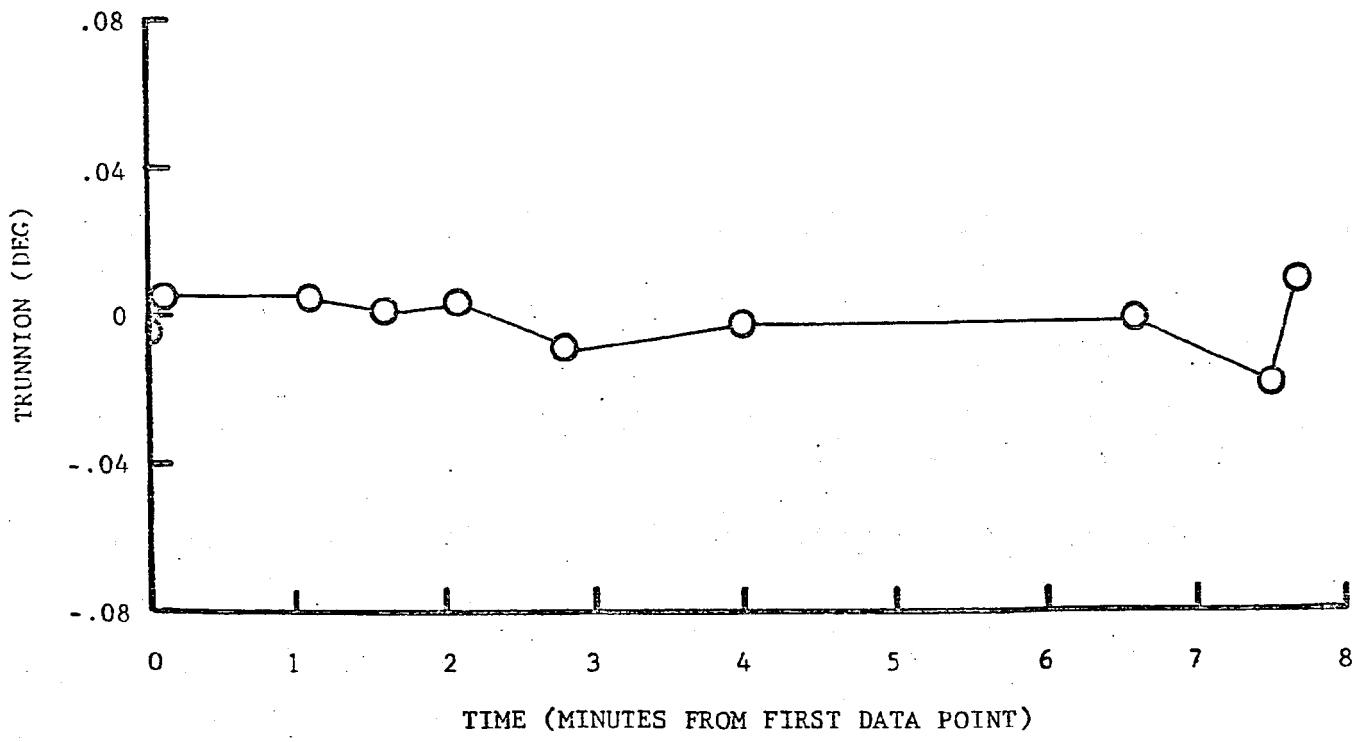
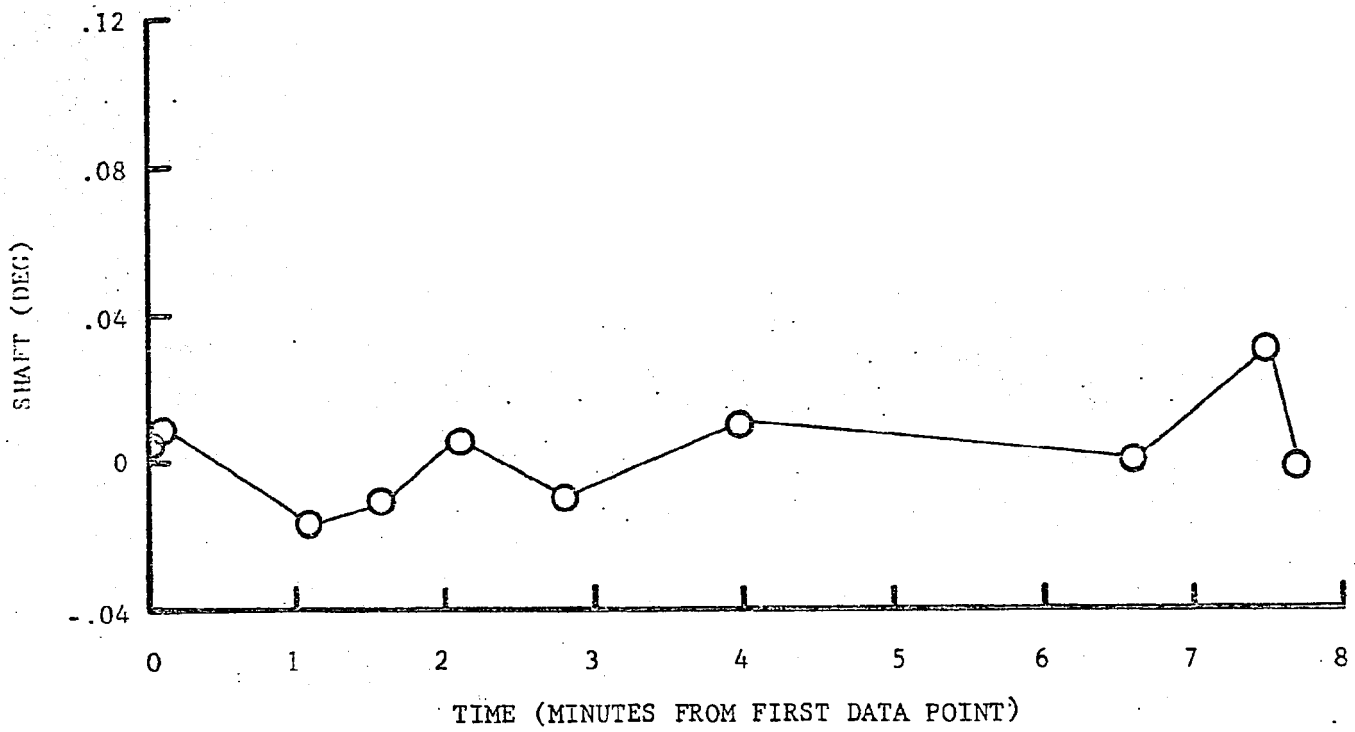


Figure 7-18 Sextant Residuals (CDH to TPI)

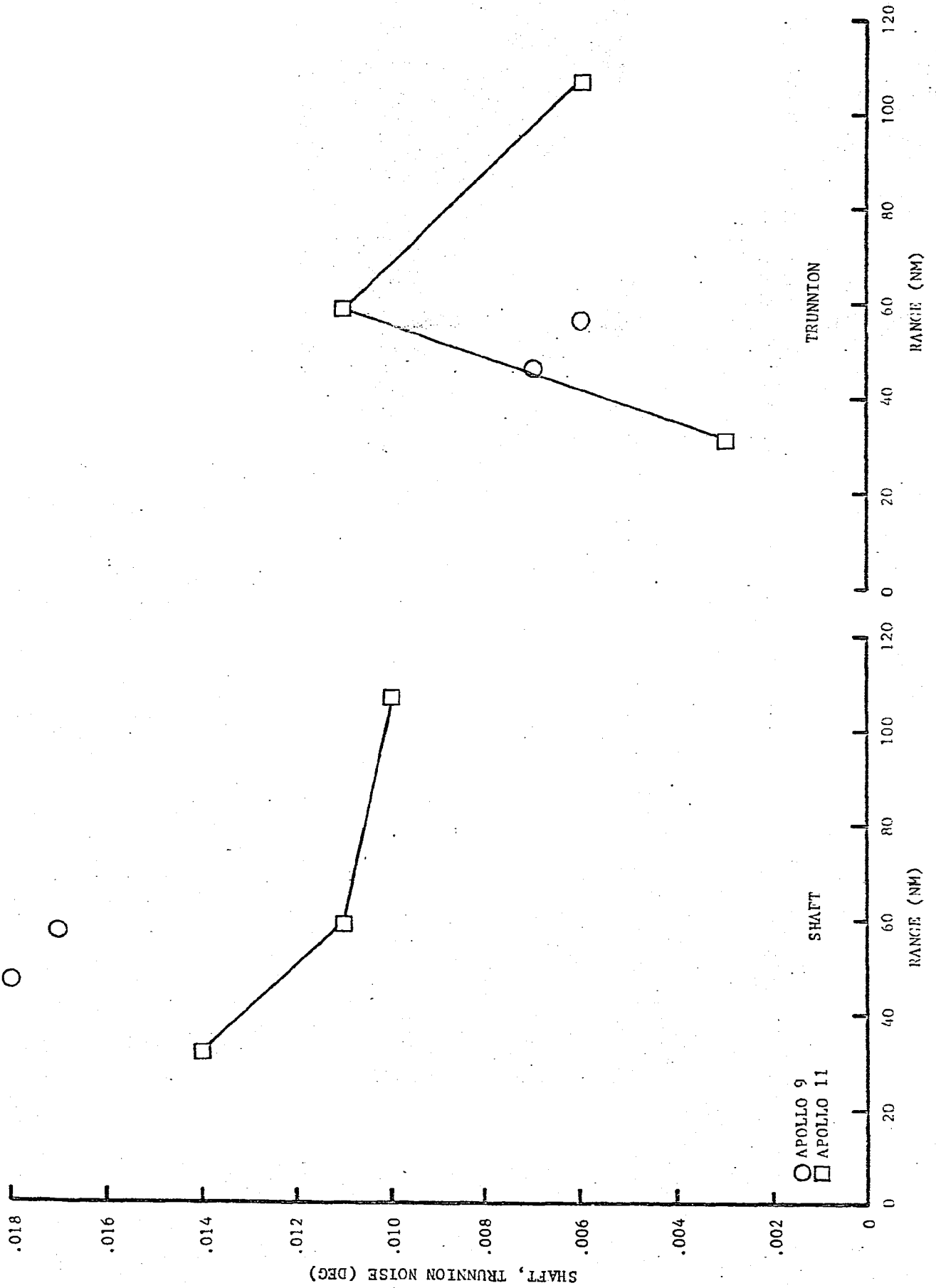


Figure 7-10 Servo-actuator Angular Random Noise as a Function of Amplitude

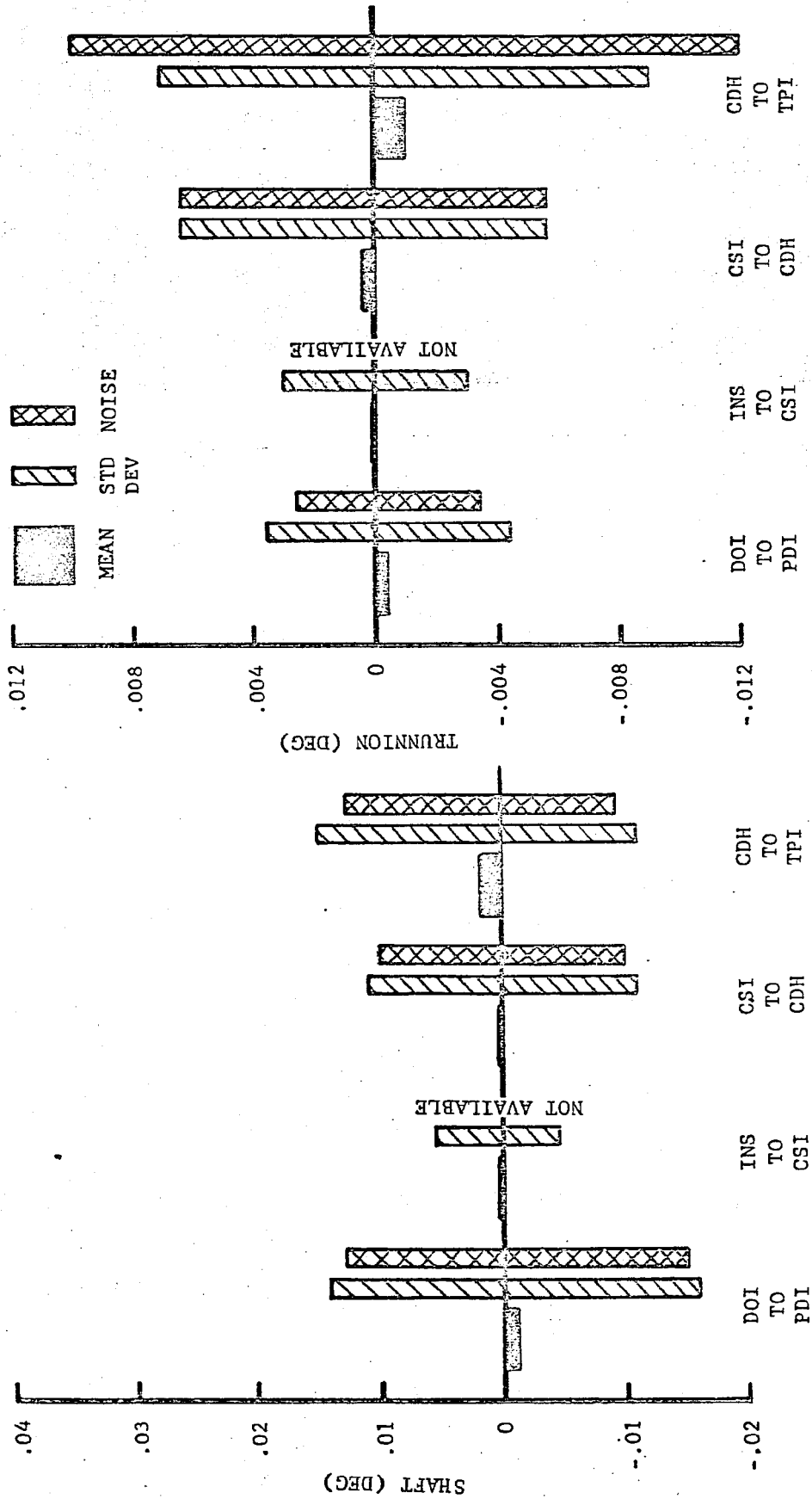


Figure 7-20 Sextant Residual Statistics

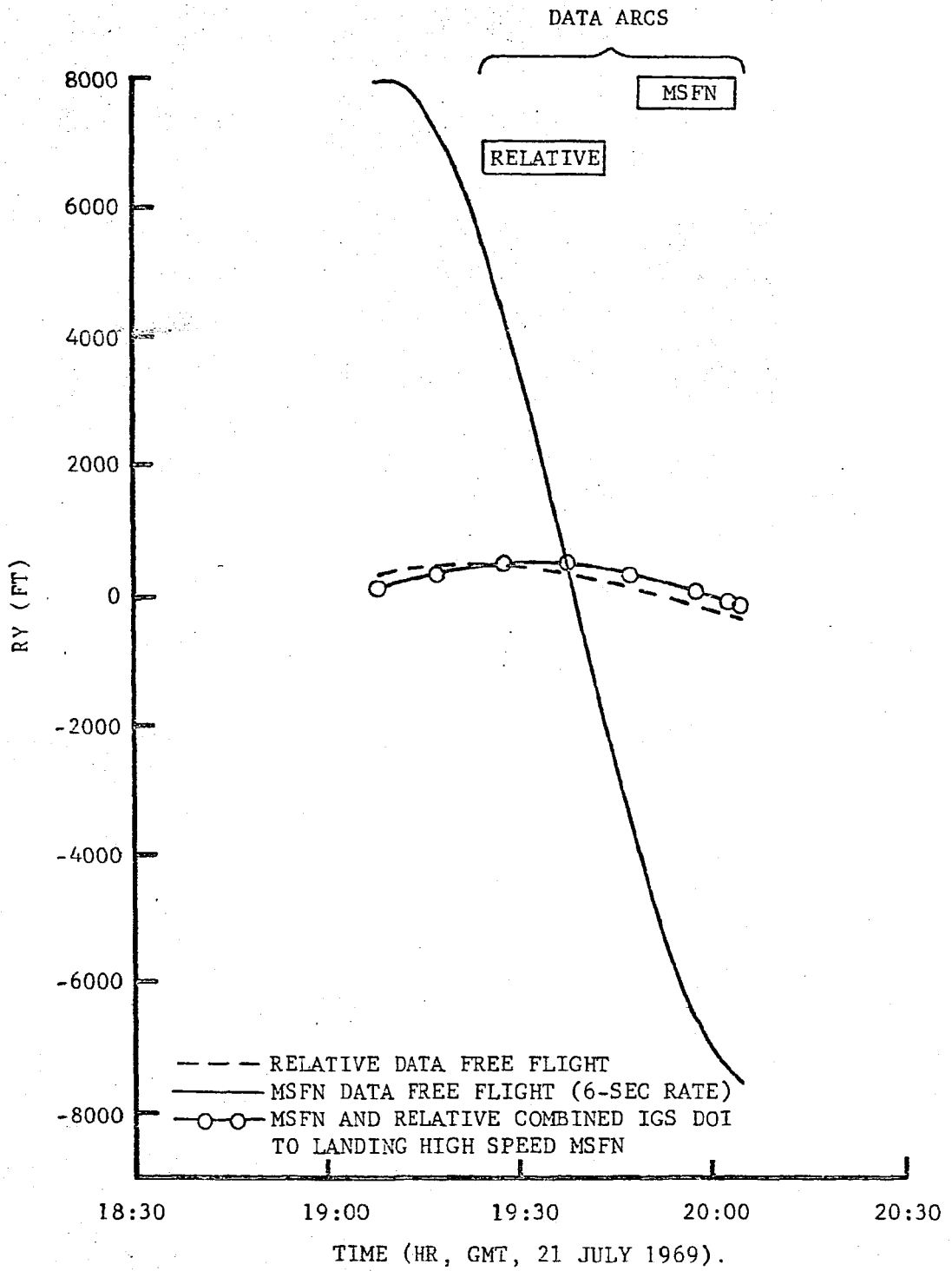


Figure 7-21 Out-of-Plane Component of LM Position Relative to CSM (DOI to PDI)

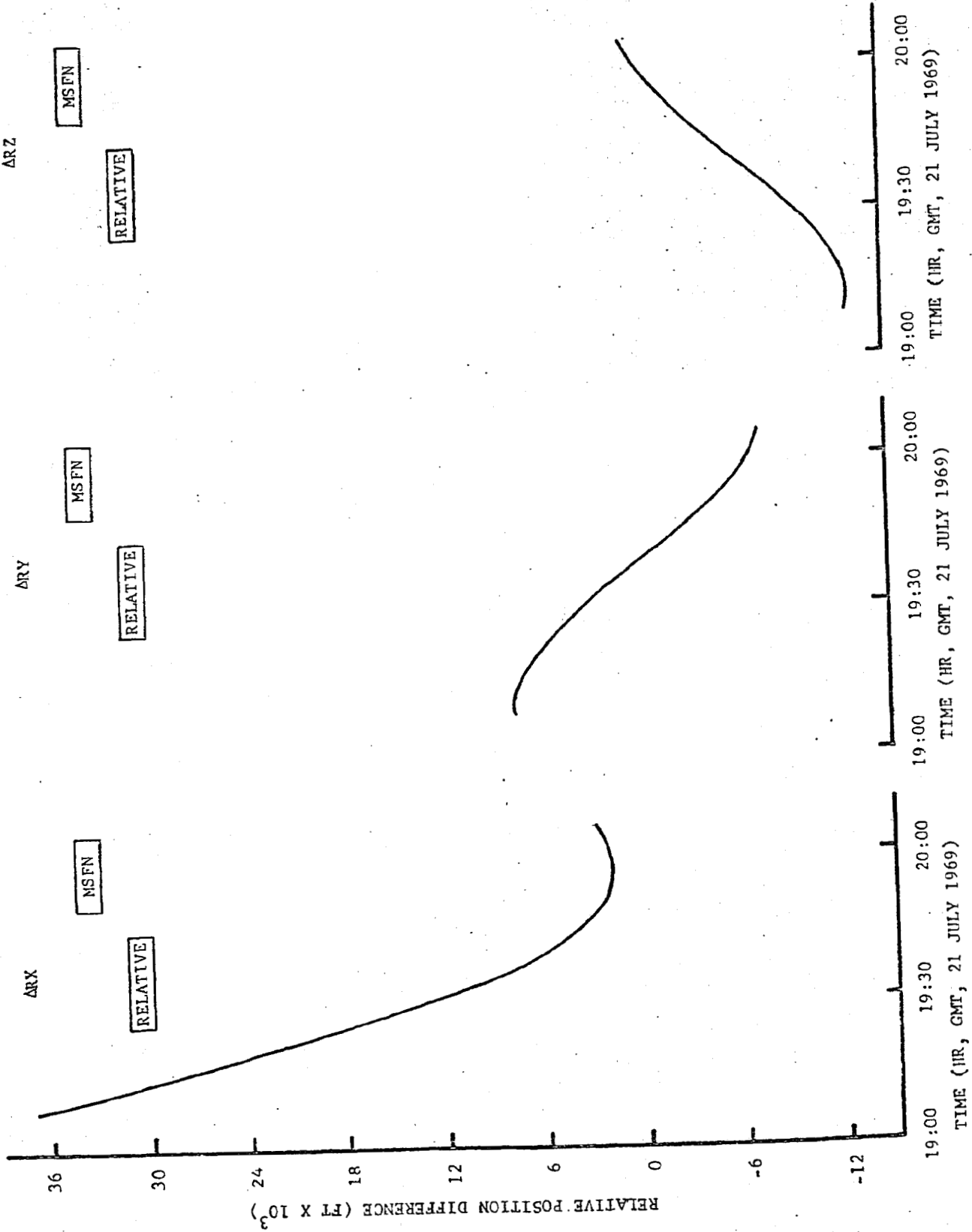
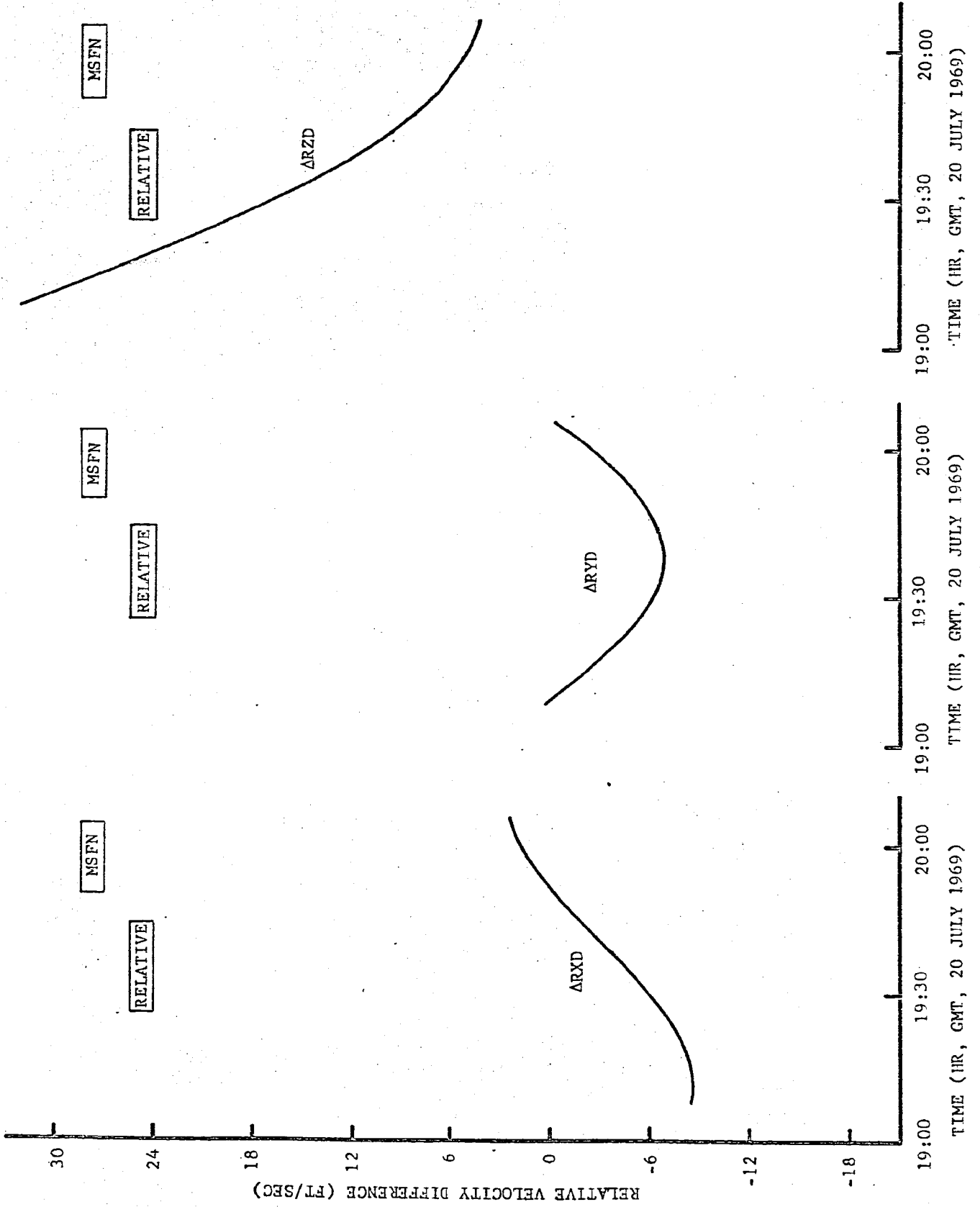


Figure 7-22 Differences Between Position Components of Relative Trajectories (DOI to PDI)



Figures 7-24 and 7-25 show the differences between position and velocity components of the two relative trajectories obtained for the Insertion to CSI period. These trajectories were obtained from a MSFN (low speed) data free flight fit and an onboard data free flight fit. Note that the differences between the RX and RZ components are nominal whereas the RY component (crossrange) is large. This characteristic is expected since onboard data fits produce a much better relative trajectory in the out-of-plane sense. Figure 7-26 illustrates the better crossrange position obtained from onboard data fits.

Figure 7-27 illustrates that the trajectory obtained from onboard tracking data eliminates three to four thousand feet of relative crossrange error which the MSFN data could not. The phase differences evident in Figures 7-26 and 7-27 result primarily from differences in the determination of the right ascension of the ascending node of the orbits. The results of this phase difference are very evident in the plot of the differences between out-of-plane position components of trajectories derived from MSFN and from onboard data (Figure 7-28 (ΔRY)).

The important feature to note in these figures is that the trajectories based on onboard tracking data eliminate a large portion of the crossrange error present in independent MSFN fits for both vehicles. It is also interesting to note that in the out-of-plane position curves shown in Figures 7-26 and 7-27, that the trajectories produced from relative data match across the CSI burn much more closely than the fits produced from MSFN data. While this agreement does depend, to some extent, on a good match between the CSM trajectories, the relative data did produce a more continuous trajectory in the out-of-plane sense from one independent fit to another.

Despite the large out-of-plane differences, it can be seen that trajectories produced from onboard tracking data are generally consistent with MSFN based fits, especially in overlapping data arcs (Figures 7-24 and 7-28). Therefore, because of better characteristics in the relative sense, trajectories produced from relative tracking data are more suitable for detailed rendezvous analysis purposes.

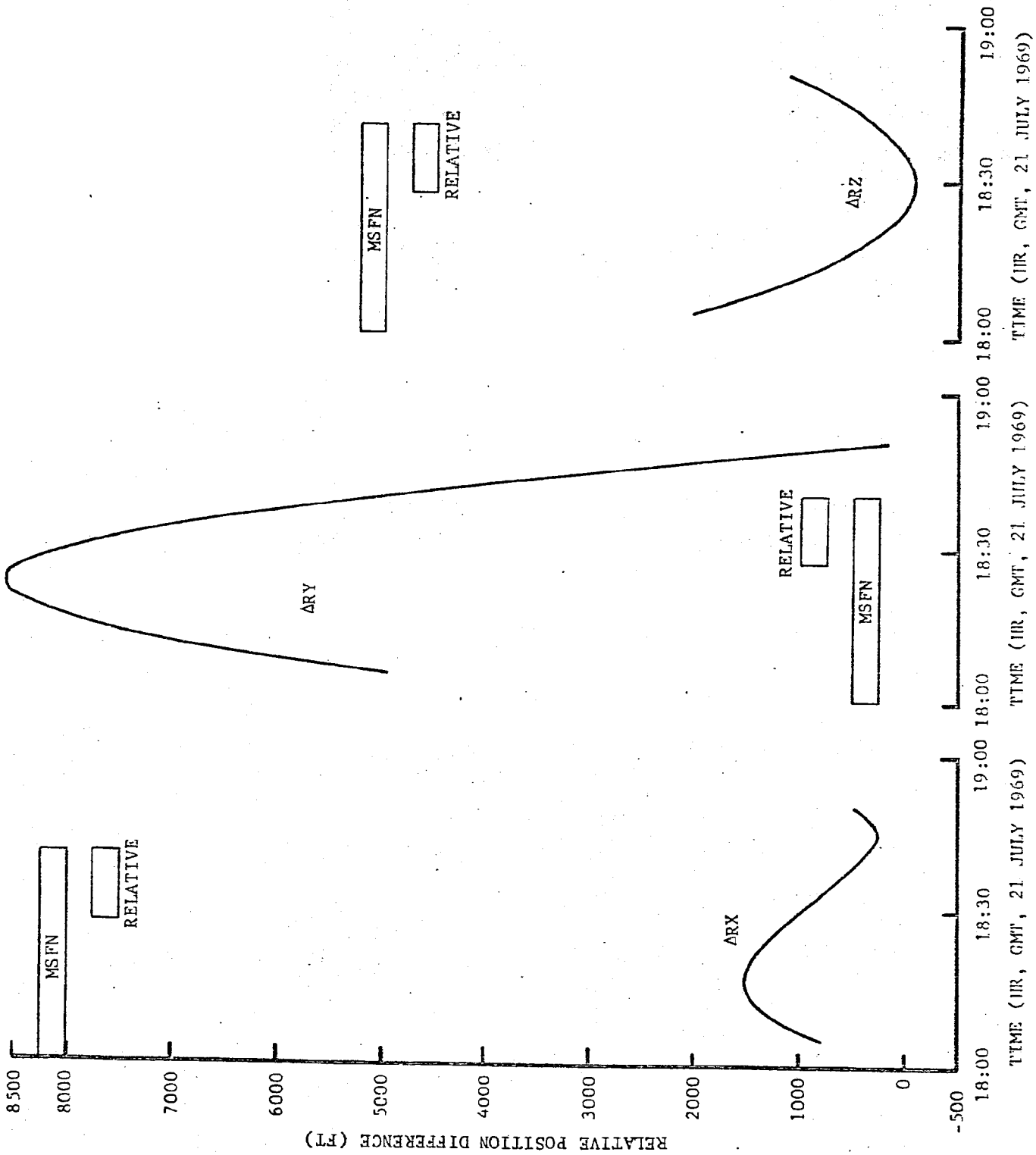


Figure 7-24 Differences Between Position Components of Relative Trajectories (Insertion to CST)

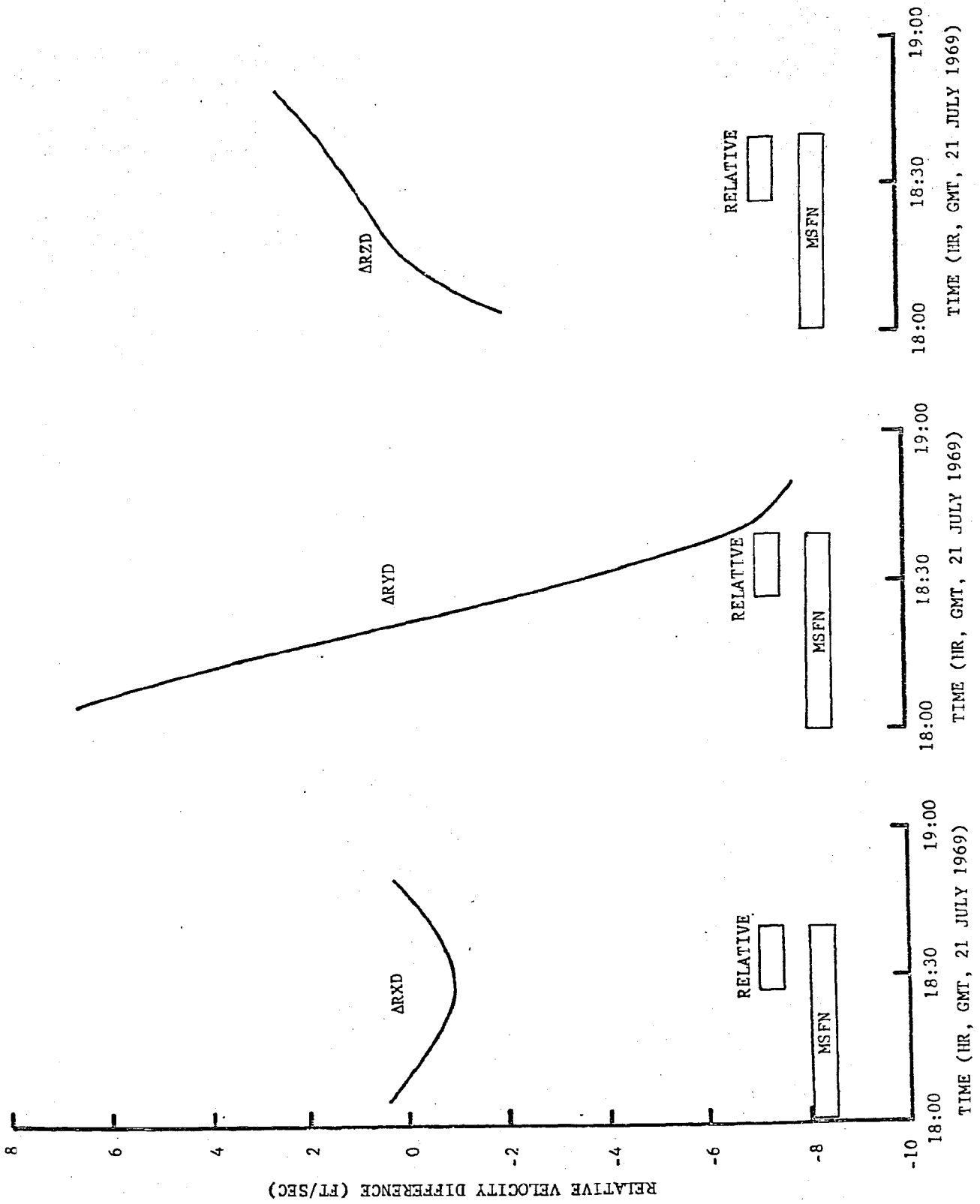


Figure 7-25 Differences Between Velocity Components of Relative Trajectories (Insertion to GSI)

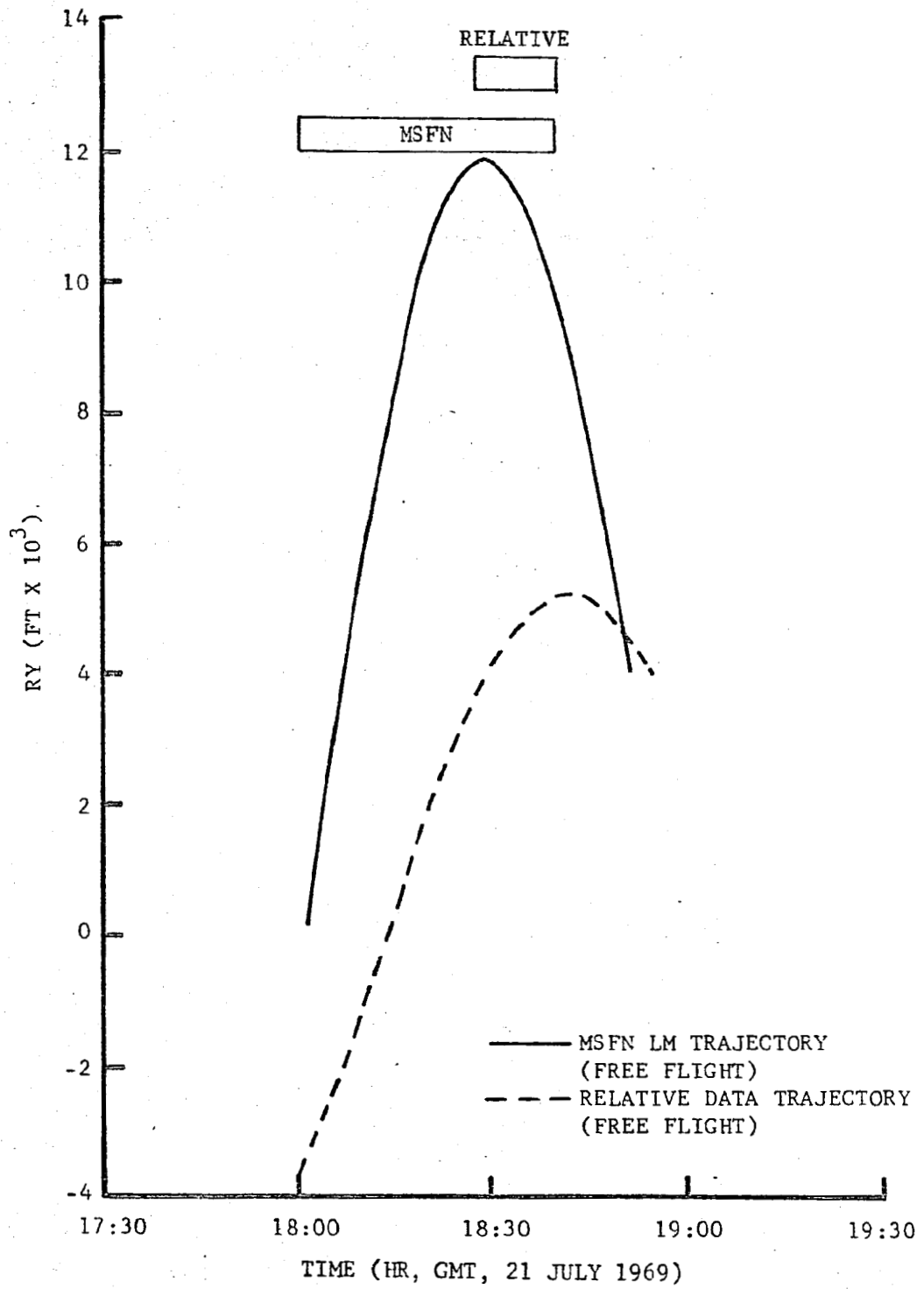


Figure 7-26 Out-of-Plane Component of LM Position Relative to CSM (Insertion to CSI)

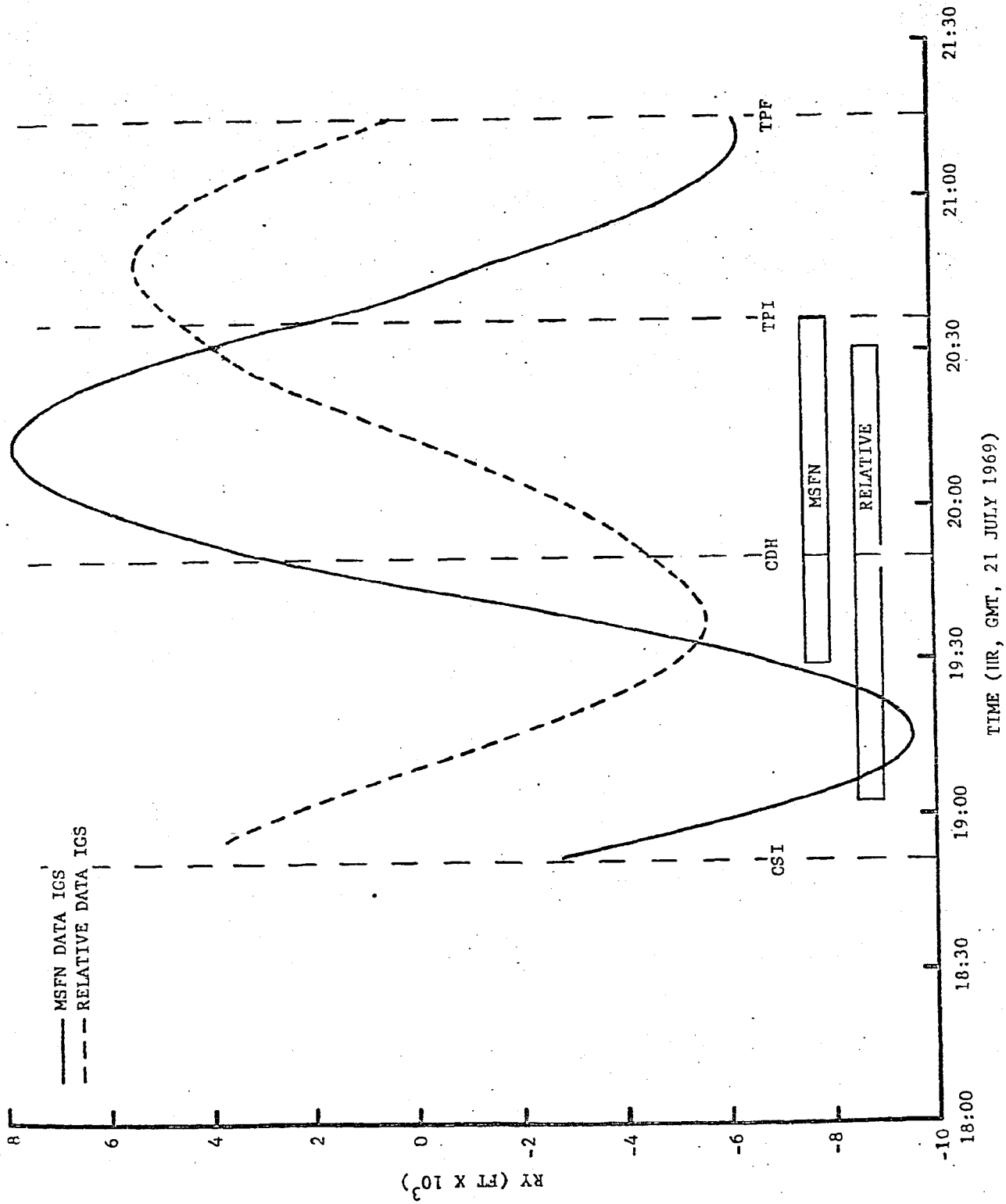


Figure 7-27 Out-of-Plane Component of LM Position Relative to CSM (CSI to TPF)

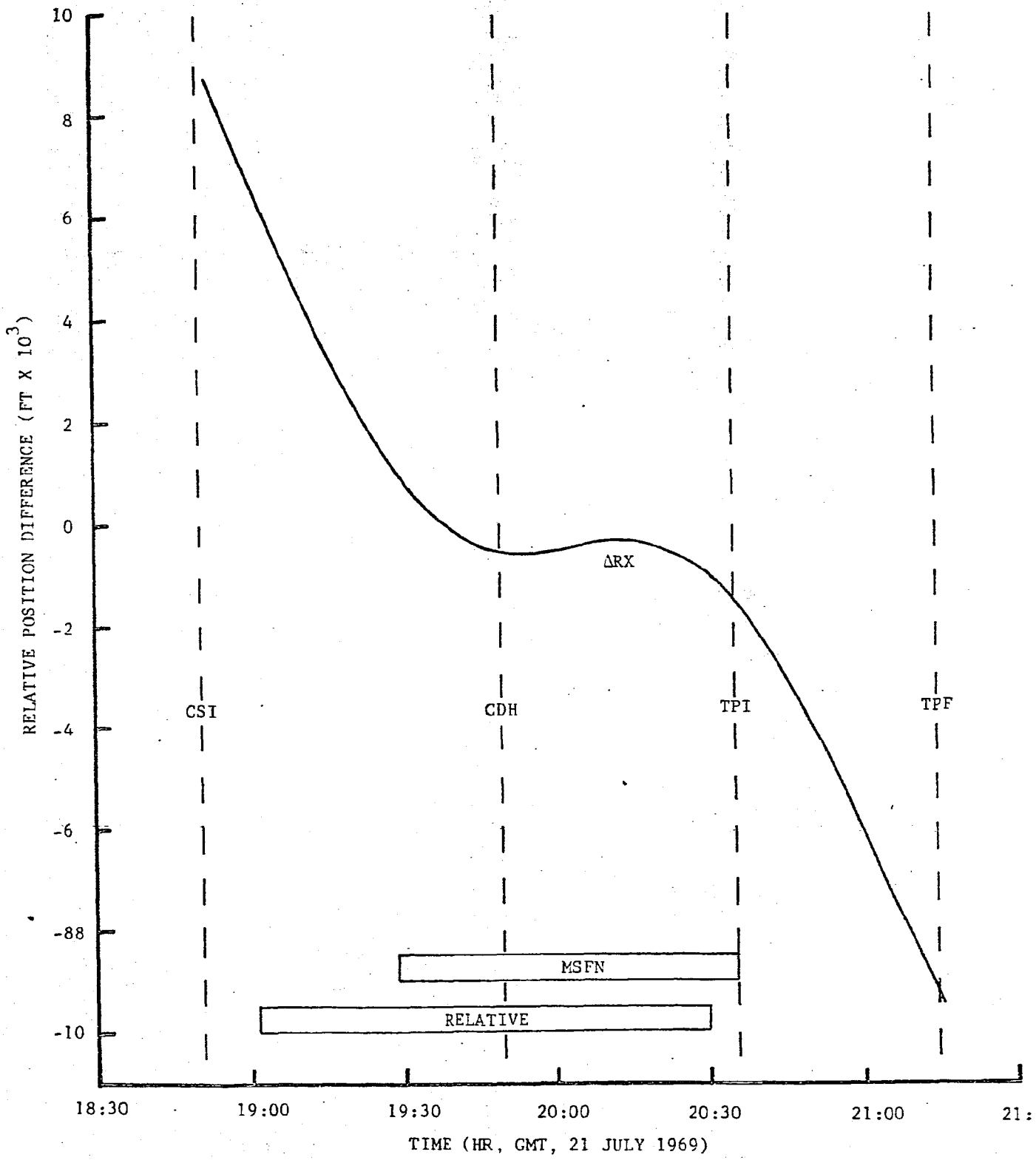


Figure 7-28 Differences Between Position Components of Relative Trajectories (CSI to TPF)

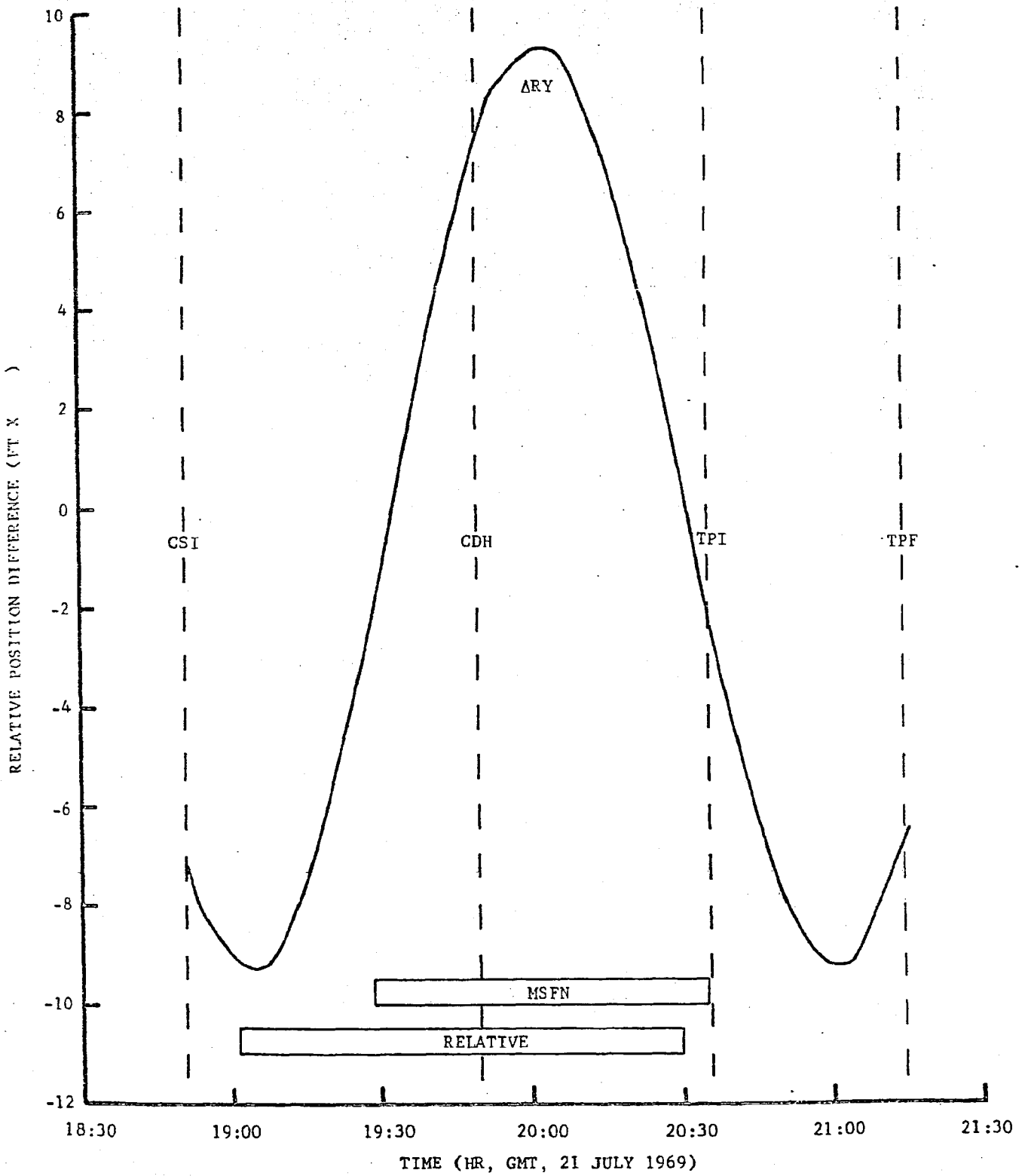


Figure 7-28 (Continued)

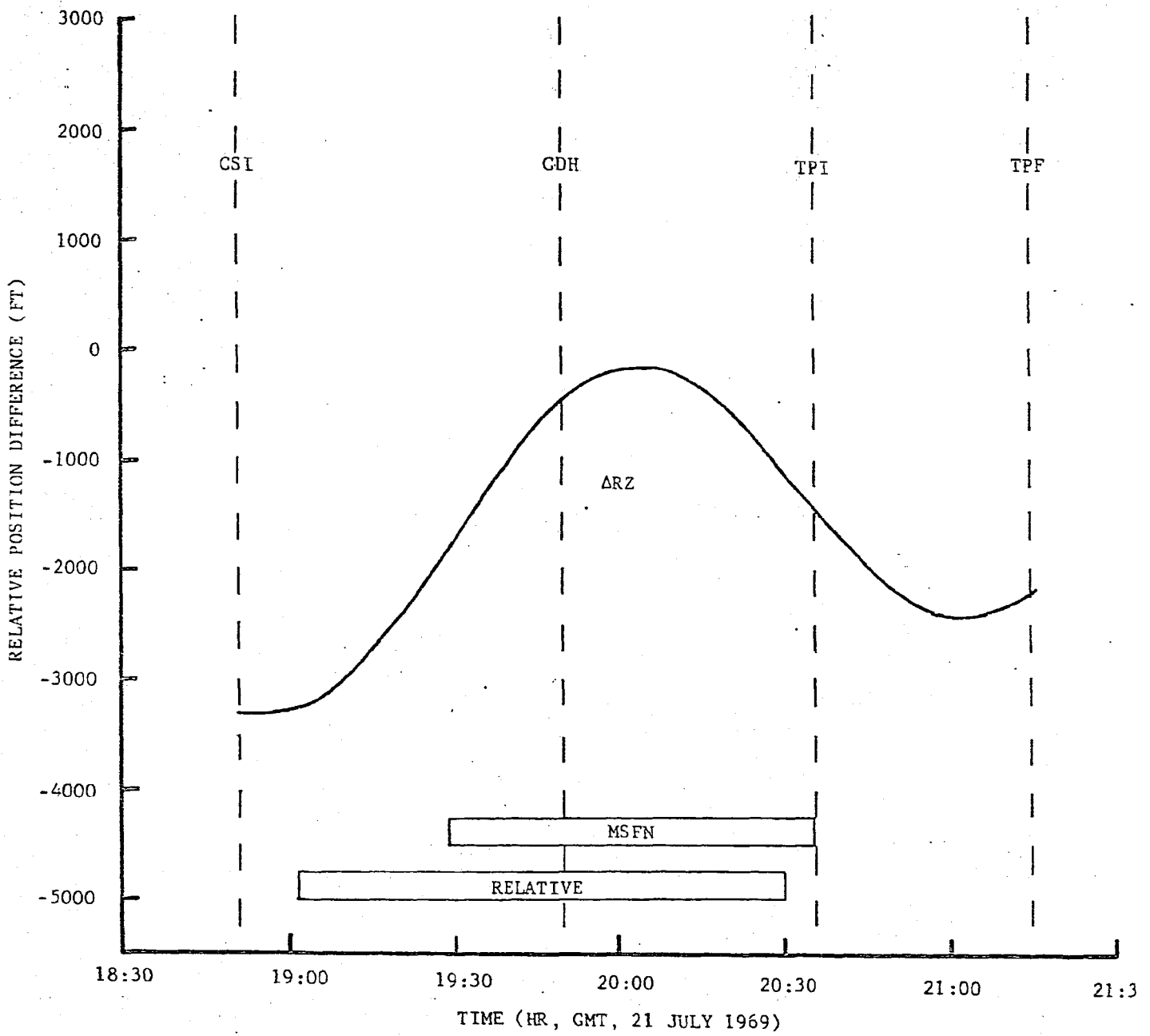


Figure 7-28 (Concluded)

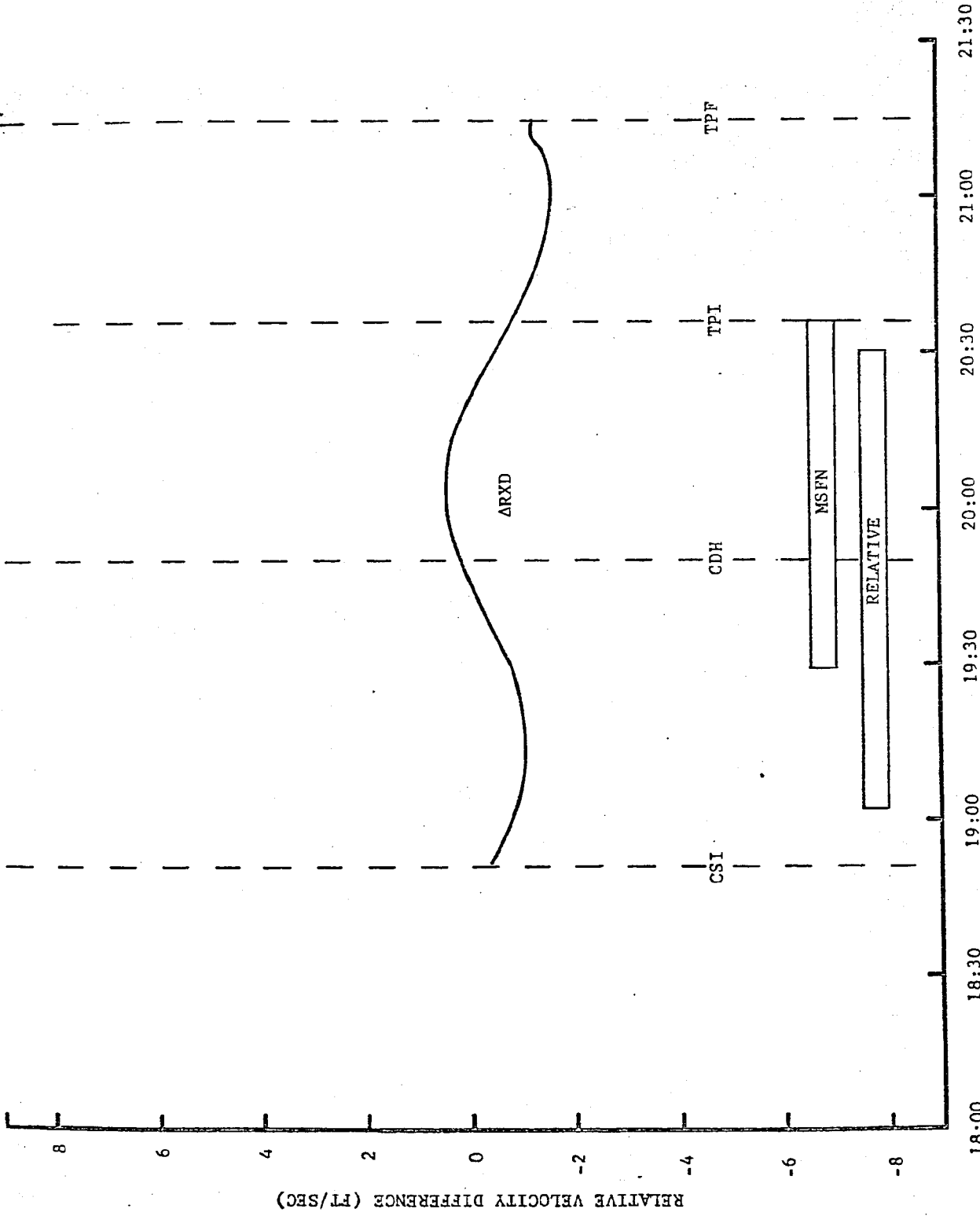
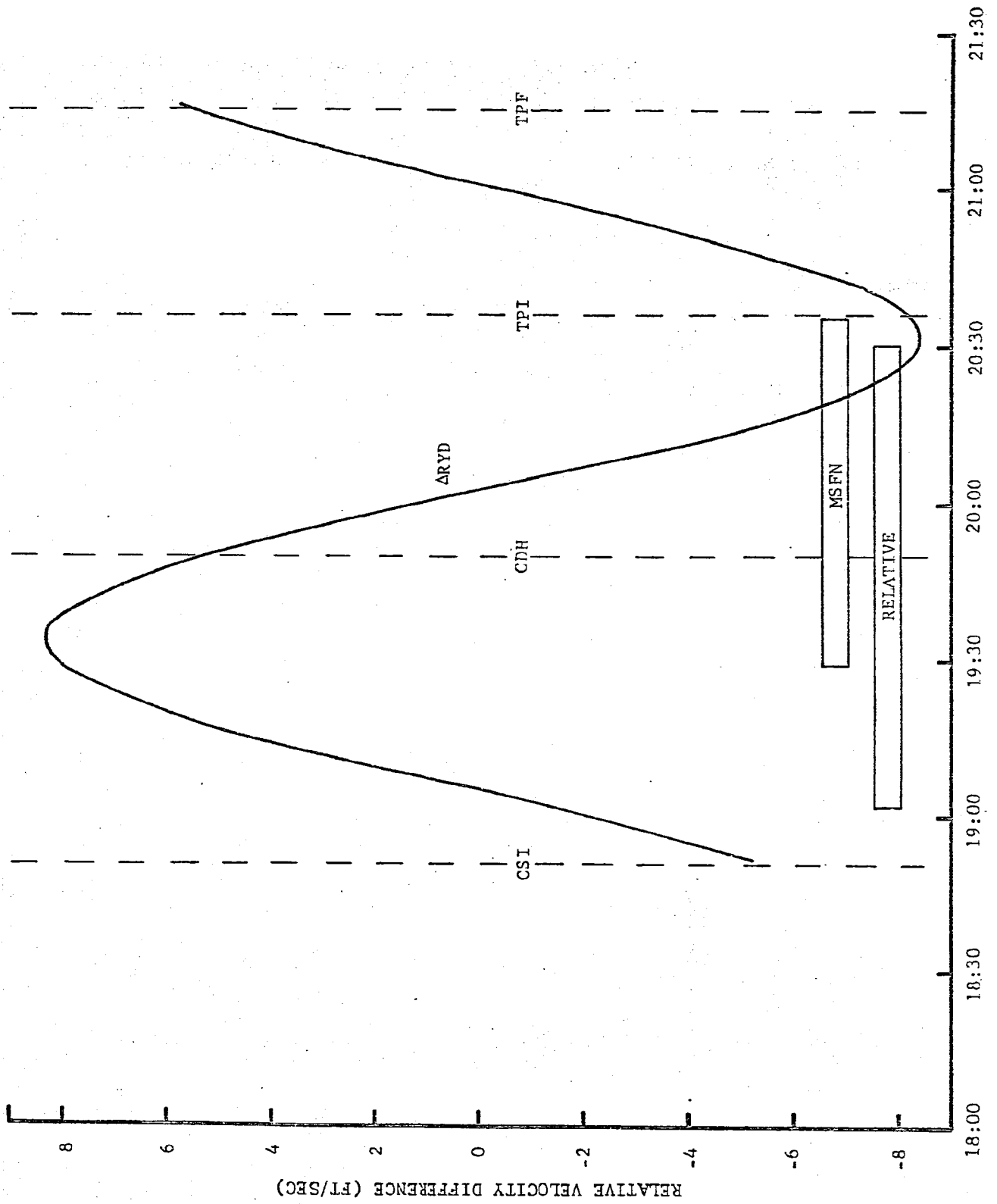


Figure 7-29 Differences Between Velocity Components of Relative Trajectories (CSI to TPF)



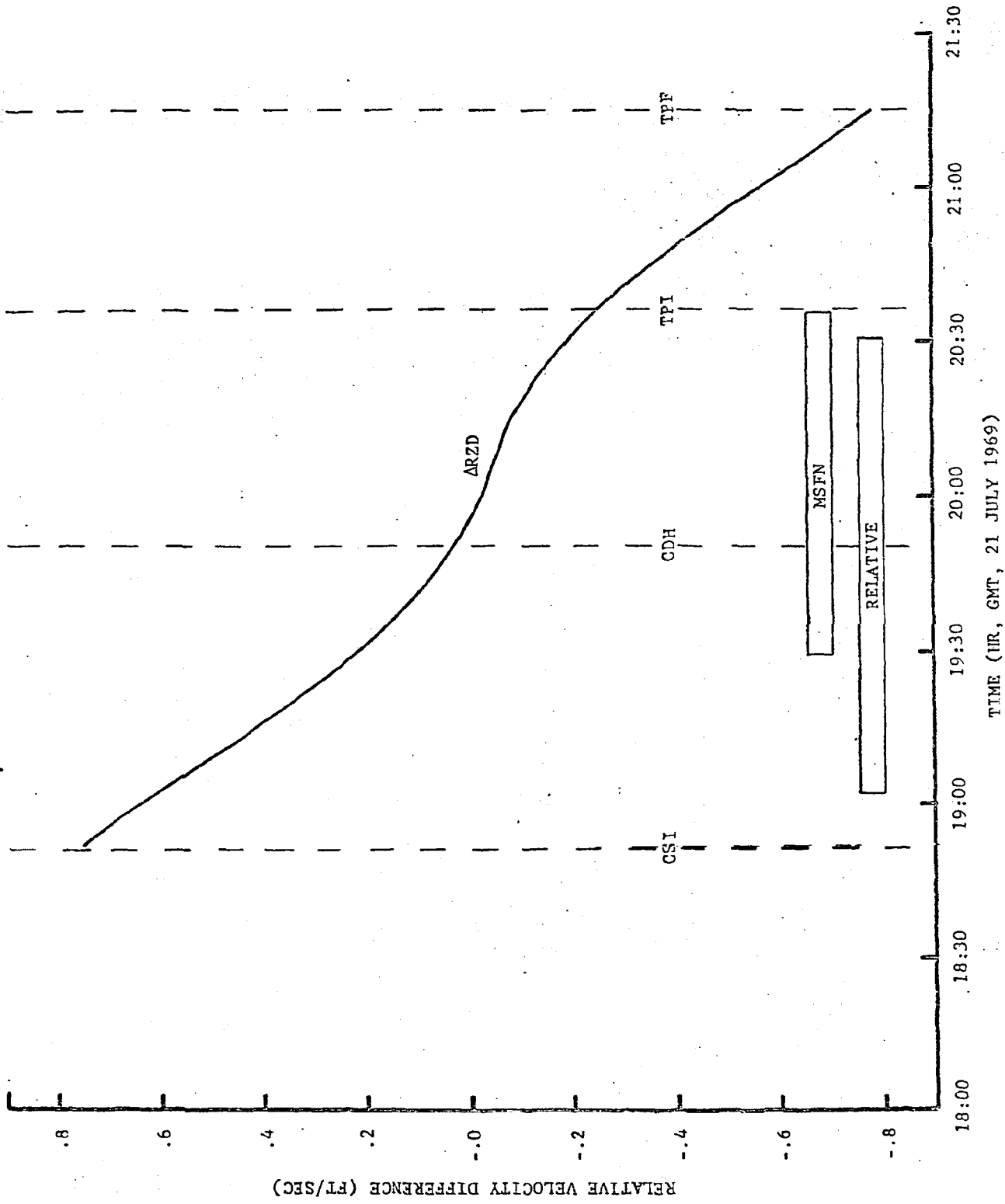


Figure 7-29 Concluded

Table 7.9 Comparison of Rendezvous Radar Noise Estimates with Specification Requirements

Free Flight Segment	Average Range (n. mi.)	Angular Noise 3σ (deg)		Range Noise 3σ (feet)	
		Est.	Spec.	Est.	Spec.
INS to CSI	140	.049	.1146	117.	2126.
CSI to CDH	107	.038	.1146	111.	1627.
CDH to TPI	59	.037	.1146	81.	890.

Table 7.10 Comparison of VHF Ranging and Sextant Noise Estimates with Specification Requirements

Free Flight Segment	Average Range (n. mi.)	Sextant Angular Noise 3σ (deg)		VHF Ranging Range Noise 3σ (feet)	
		Est.	Spec(1)	Est.	Spec.
DOI to PDI	32	.043	-	69.	180.
INS to CSI	140		NA		NA
CSI to CDH	107	.035	-	69.	180.
CDH to TPI	59	.047	-	57.	180.

(1) No specification value was available.

Specification Comparisons

Table 7.9 compares the 3σ values of noise estimated from the rendezvous radar residuals with specification requirements. It can be seen that the estimates were all well within specification limits. Noise estimates for the sextant and VHF ranging data are listed in Table 7.10. Although no specification value was found for the sextant, the values obtained (RSS of individual angle noise estimates) were all within acceptable limits. The VHF ranging noise estimates also compare well with specifications.

Conclusions

The following conclusions were drawn from the analysis.

1. The onboard data was generally of good quality. The sextant data, examined for the first time, appeared to be as accurate as the rendezvous radar angular measurements.
2. Estimates of data random noise were all within specification and expected values.
3. Trajectories produced from onboard tracking data proved to be generally consistent with those produced from MSFN data. It was found that a method used in the past to demonstrate trajectory consistency was inadequate. On Apollo 10, trajectories were compared only at selected times. Because of the significantly large phase differences found to be present in relative trajectories, the values for out-of-plane position differences obtained at selected times may be misleading. The out-of-plane position components must be plotted as a function of time in order to see the total differences in the trajectories.

7.4 LANDING RADAR DATA ANALYSIS

The landing radar data analysis consisted of generating and evaluating landing radar residuals (difference between observed measurement and computed measurement) and mapping of the lunar surface profile and ground-track with the slant range measurement.

The landing radar data were obtained by processing the downlink telemetry data with a special purpose computer program which outputs onboard observations on punched cards in a HOPE-compatible format.

The HOPE Program was used to compute simulated landing radar observables from selected LM trajectories and from auxiliary information such as REFSMAT, gimbal angles, and radar operating mode. The LM trajectories were generated by the HOPE Program utilizing telemetered acceleration data in the IGS burn option to model the descent burn. Residuals were then formed by subtracting the computed from the actual observable value. Paragraph 7.4.2 presents statistics and selected plots of residuals obtained from various LM state vectors.

Terrain mapping data were obtained from a small, special purpose computer program designed to compute terrain altitude above a mean lunar radius as a function of latitude and longitude. The results of an attempt to correlate this terrain data with lunar contour maps are presented in Paragraph 7.4.3.

7.4.1 Descent Trajectories

Six different descent trajectories were examined in the landing radar data analysis. The origins of these trajectories are summarized as follows:

- (a) RTCC - This vector was obtained in the RTCC in real time.
- (b) MSFN (LS) - This vector was obtained from an IGS fit using low speed MSFN data obtained from acquisition of signal to LM touchdown (revolution 14). The doppler data were compacted to two observations per minute.
- (c) Onboard - This vector was obtained from a free flight fit using CSM sextant and VHF ranging observations. The technique required fixing the CSM trajectory as a reference and updating the LM state from onboard observations and the CSM reference trajectory.

- (d) BET #3 - The MSFN state vector described in item (b) above, was used as the basis for this trajectory. The BET #3 was obtained by correcting the MSFN low speed state with a linear error analysis program so that the resultant powered descent trajectory would impact a desired landing site with a relative velocity of zero. The landing coordinates used as reference were the MPB photographic estimate.
- (e) Lear - High speed MSFN data (ten samples per second) obtained over a 232 second data arc just prior to PDI were fit by the Lear Powered Flight Processor producing this state vector.
- (f) Onboard/MSFN (H-S) - This trajectory was obtained with the HOPE Program and used high speed MSFN doppler data which had been compacted to 30 observations/minute and from CSM sextant and VHF ranging data using the HOPE orbit determination program. The descent burn was modeled by the HOPE IGS burn option. The HOPE weighted least squares solution vector included position and velocity at epoch (which was prior to PDI), and Y platform misalignment. The tracking data interval was from DOI to LM touchdown. Figure 7-2 shows the tracking data timeline.

In order to gauge the quality of the landing radar data, it was necessary to determine that the above trajectories did accurately represent the actual descent trajectory. This quality judgement was based largely on the landing point conditions obtained from each trajectory. These landing sites obtained from each trajectory are summarized graphically in Figure 7-30. Note that both the BET #3 and the Onboard/MSFN H-S estimates are very close to the 16mm photographic estimate (accepted as the best estimate).

Since the data type being examined is a velocity measurement, it is most important that the reference trajectory be virtually free of velocity errors in the data arc. The onboard/MSFN H-S trajectory contains a large velocity error at landing where the BET #3 was constructed in such a manner that the velocities are zero at landing. Therefore, the BET #3 was chosen as the basic reference upon which to base the analysis of landing radar velocity residuals.

7.4.2 Landing Radar Velocity Residuals

Table 7.12 lists the velocity residual statistics obtained from all the trajectories considered in the analysis. Note the small mean values obtained from the reference trajectory (BET #3). In the absence of a

real standard of comparison, the mean values obtained from BET #3 were reasonably small. Standard deviations indicate that V_{YA} and V_{ZA} are somewhat more erratic than V_{XA} . However, these values are still of reasonably good quality as shown by Figures 7-31 through 7-33. These figures show the BET #3 velocity residuals plotted versus time. In addition, specification limits have been plotted. Note that a few points fall outside specification.

It is difficult to isolate measurement errors from trajectory errors in this particular case. The descent trajectory is a particularly difficult one to reconstruct, and the landing radar velocity data are particularly sensitive to trajectory errors. Notice that the velocity residuals in Figures 7-31 through 7-33 tend toward zero at landing where the BET #3 velocities were constrained to zero. In contrast, the trajectory obtained from the Onboard/MSFN H-S fit is known to contain velocity errors at landing. The resultant total velocity at landing is 8.02 fps, with the primary contribution in the Z direction (North). The residual statistics show a mean value for V_{YA} of 6.966 fps. Since V_{YA} was directed roughly North, the large mean value reflects the -7.96 fps in the Z component of velocity at landing. The residuals obtained from the Onboard/MSFN H-S fit are plotted in Figures 7-34 through 7-36.

The residual statistics listed in Table 7.12 also indicate that the best trajectories do produce the best landing radar velocity residual statistics, that is, the BET #3 and the Onboard/MSFN H-S trajectories produce the smallest residual mean values. This fact, together with the sensitivity which the data has exhibited to trajectory velocities indicate that descent trajectory reconstruction activities will be aided considerably by the landing radar velocity data.*

* Subsequent reconstructions using landing radar data have produced a trajectory landing at acceptable coordinates (Lat. = .649 deg, Long. = 23.490 deg) with a total relative velocity of .96 fps. A report of this reconstruction will be forthcoming under a separate cover.

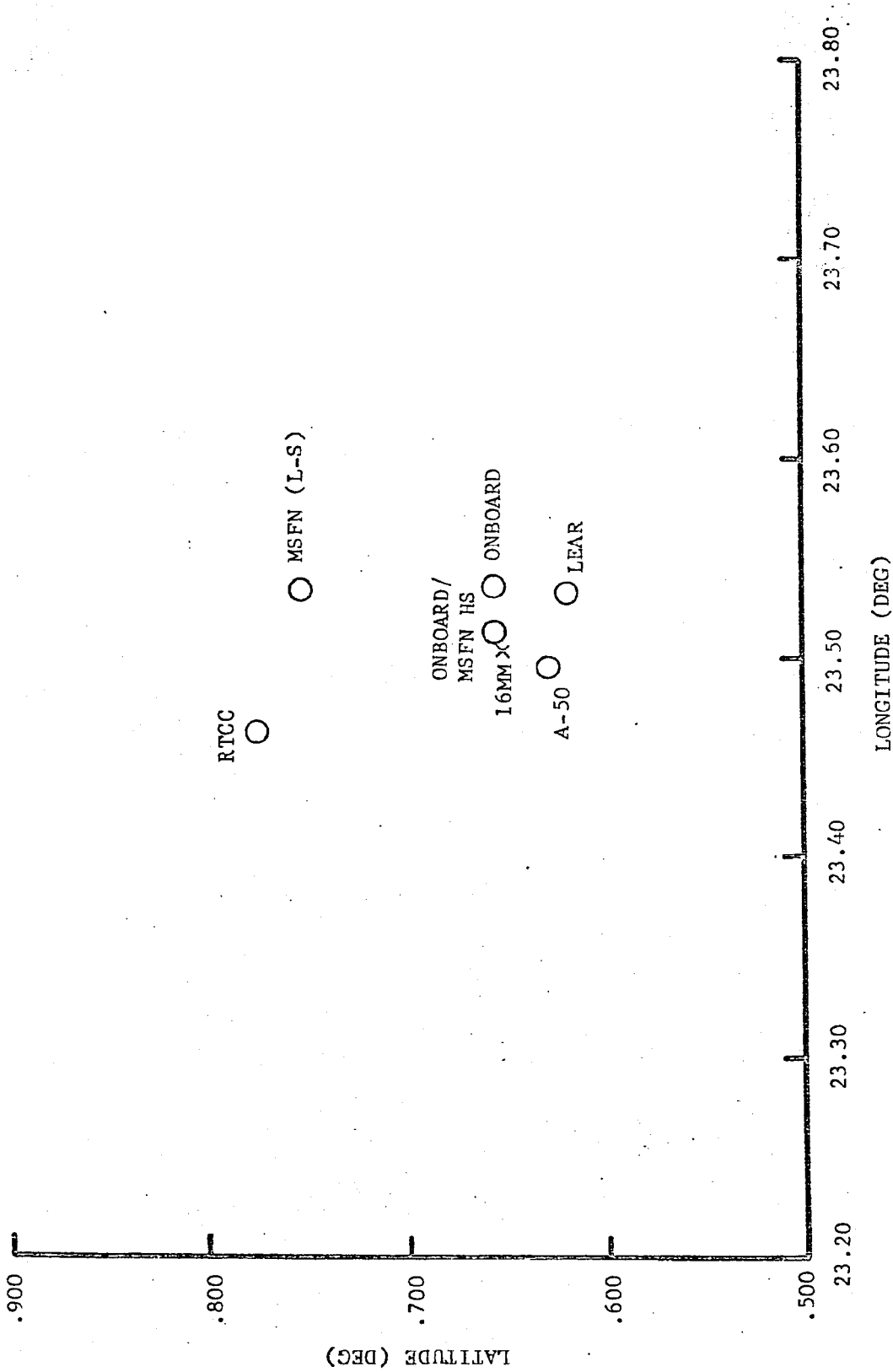


Figure 7-30 IM Landing Site Coordinates

Table 7.11 LM Landing Site Coordinates

VECTOR SOURCE	LATITUDE (deg)	LONGITUDE (deg)	RADIUS (n.mi.)
RTCC	0.777	23.461	936.59
MSFN (L-S)	0.756	23.537	937.93
ONBOARD (VHF, SXT)	0.656	23.538	936.90
BET #3	0.630	23.497	937.15
LEAR	0.620	23.532	936.66
16MM	0.647	23.505	N/A
ONBOARD/MSFN H-S	0.655	23.515	937.04

Table 7.12 Landing Radar Velocity Residual Statistics

Vector Source	V_{XA}	V_{YA}	V_{ZA}
<u>RTCC</u>			
Mean	9.543	3.909	3.022
S.Dev.	1.532	5.455	3.918
Noise	1.172	3.891	3.446
<u>MSFN (LS)</u>			
Mean	-1.997	6.501	4.533
S.Dev.	1.758	4.081	3.486
Noise	1.120	3.281	3.661
<u>ONBOARD</u>			
Mean	2.681	6.724	4.640
S.Dev.	1.475	4.209	3.430
Noise	1.316	3.948	2.203
<u>BET #3</u>			
Mean	.857	.893	-.173
S.Dev.	1.829	4.306	3.689
Noise	1.142	4.565	2.361
<u>LEAR</u>			
Mean	4.733	5.625	4.287
S.Dev.	1.018	4.189	3.723
Noise	.718	3.932	2.340
<u>ONBOARD/MSFN (H-S)</u>			
Mean	.234	6.966	1.729
S.Dev.	1.183	3.866	2.978
Noise	.575	3.336	2.349

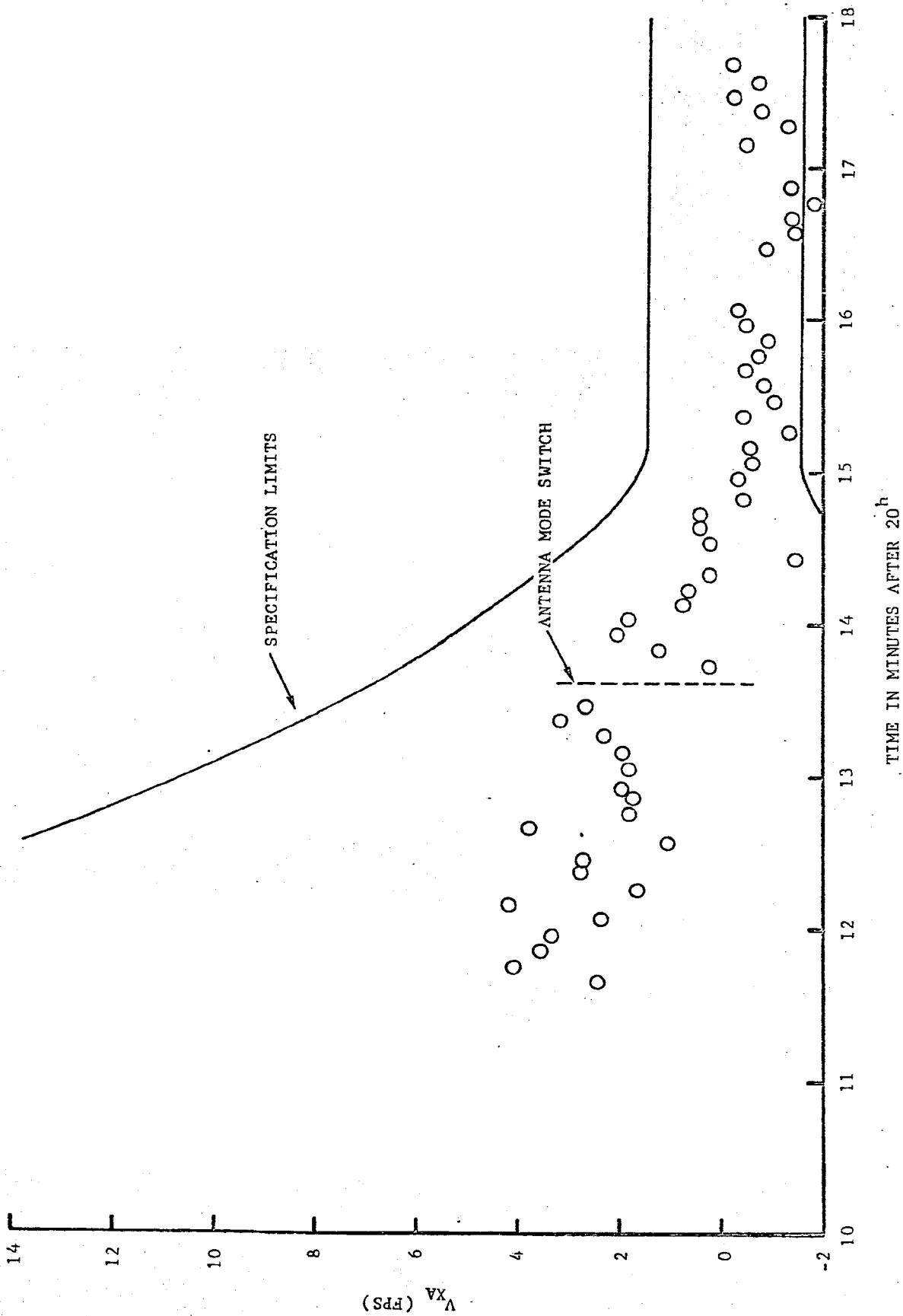


Figure 7-31. Landing Radar X-Antenna Velocity Residuals (NET#3)

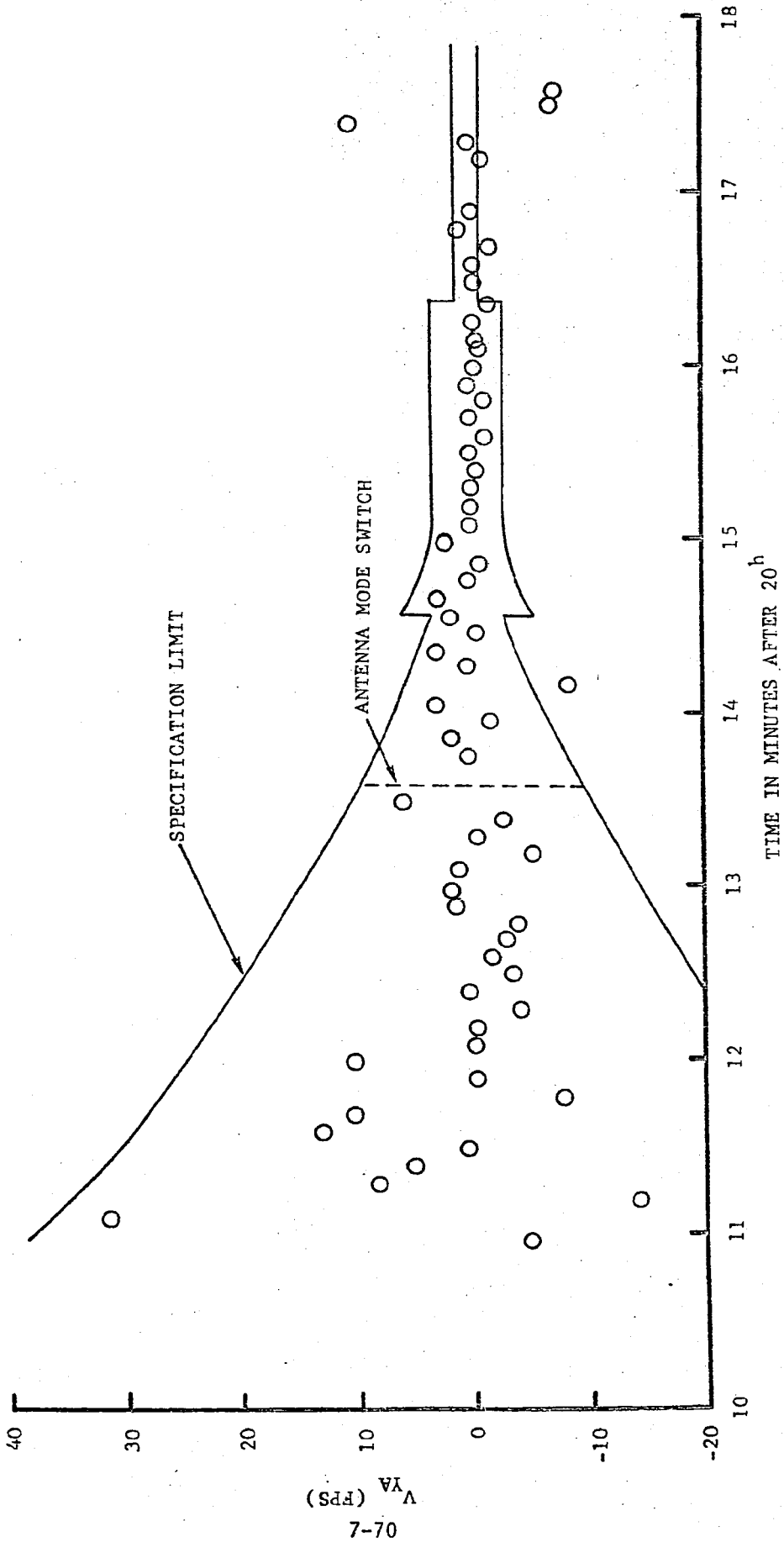


Figure 7-32 . Landing Radar Y-Antenna Velocity Residuals (BET#3)

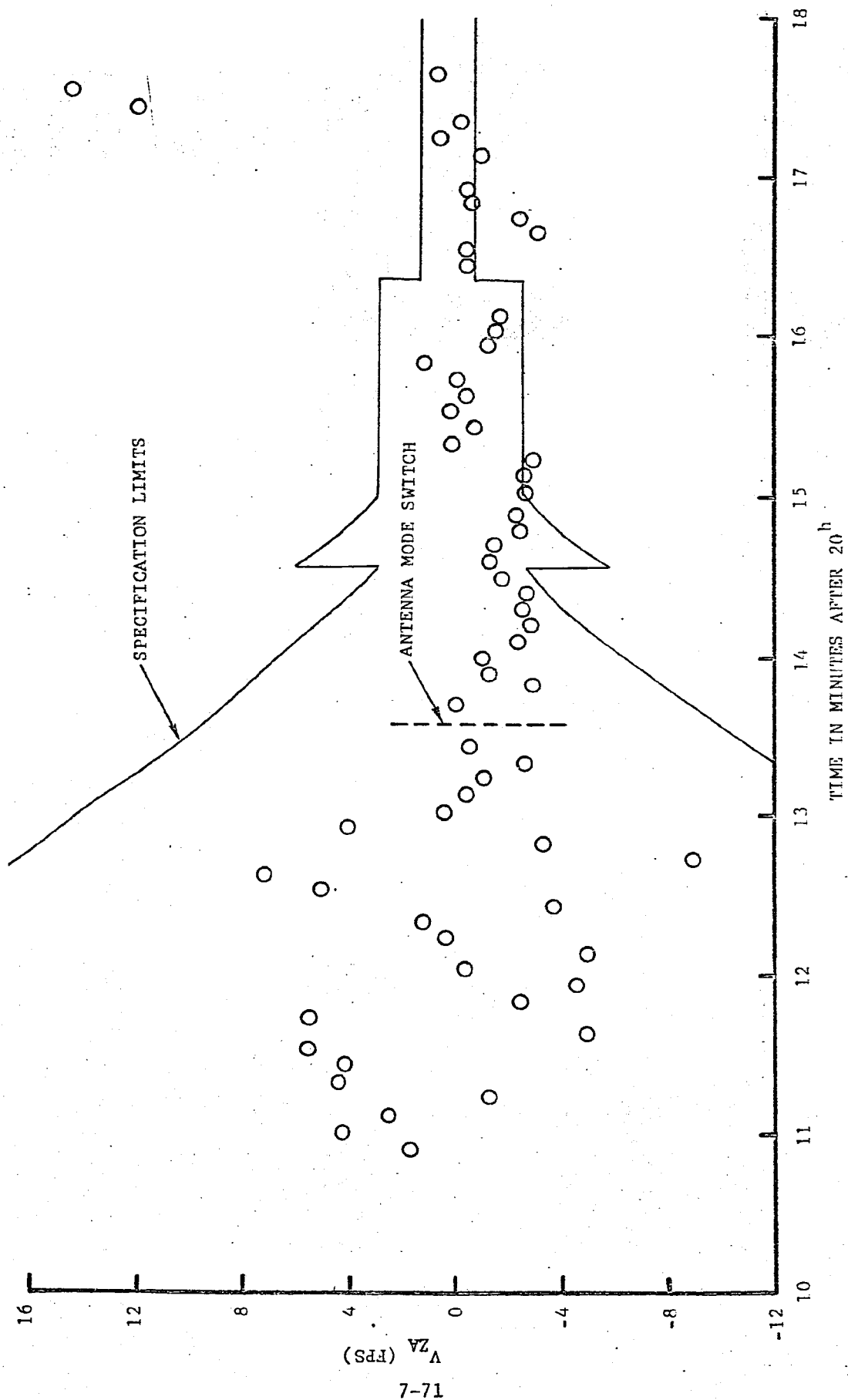


Figure 7-29 Landing Radar 2-Antenna Voltage Distribution

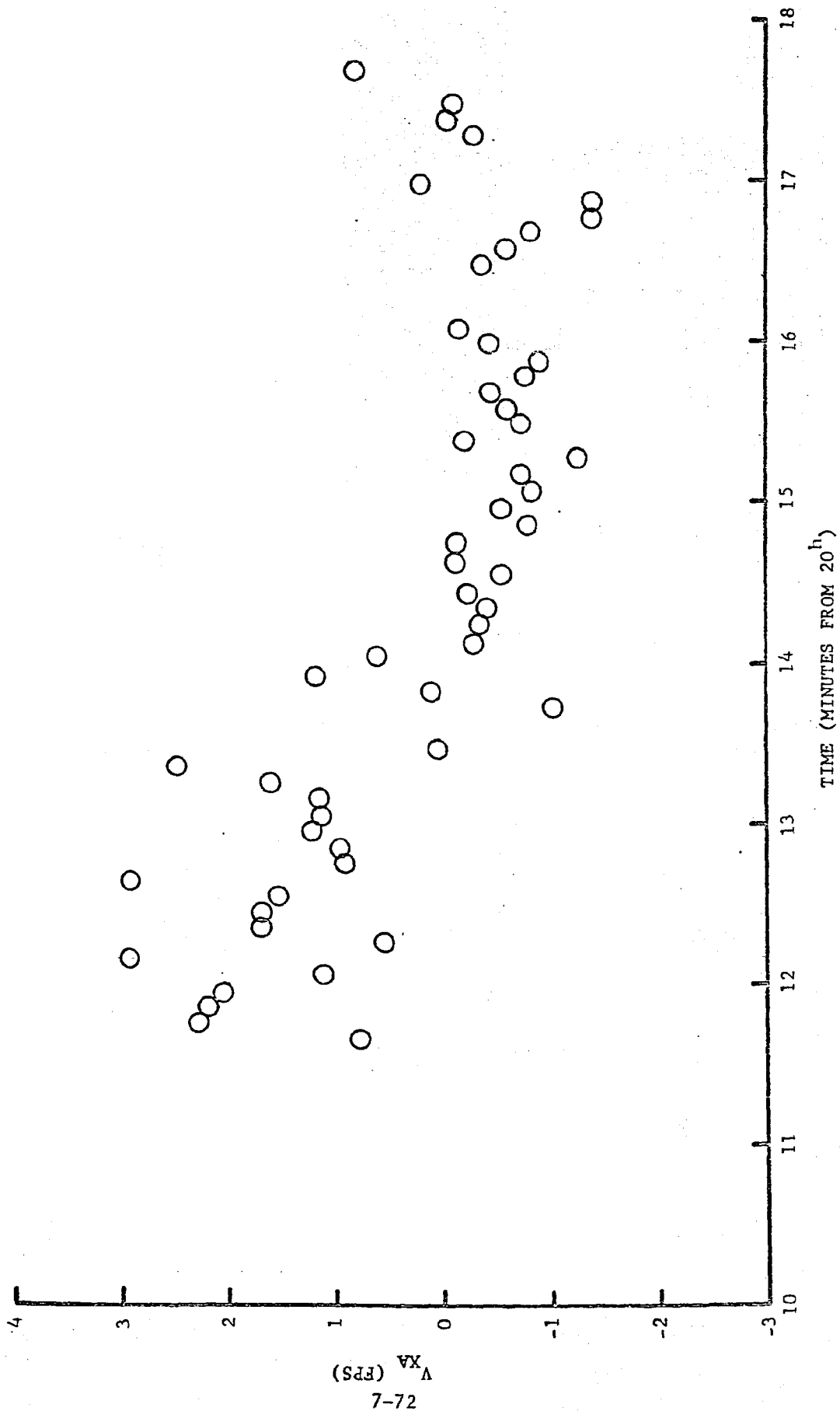


Figure 7-34 . Landing Radar X-Antenna Velocity Residuals (Onboard/MSN H-S)

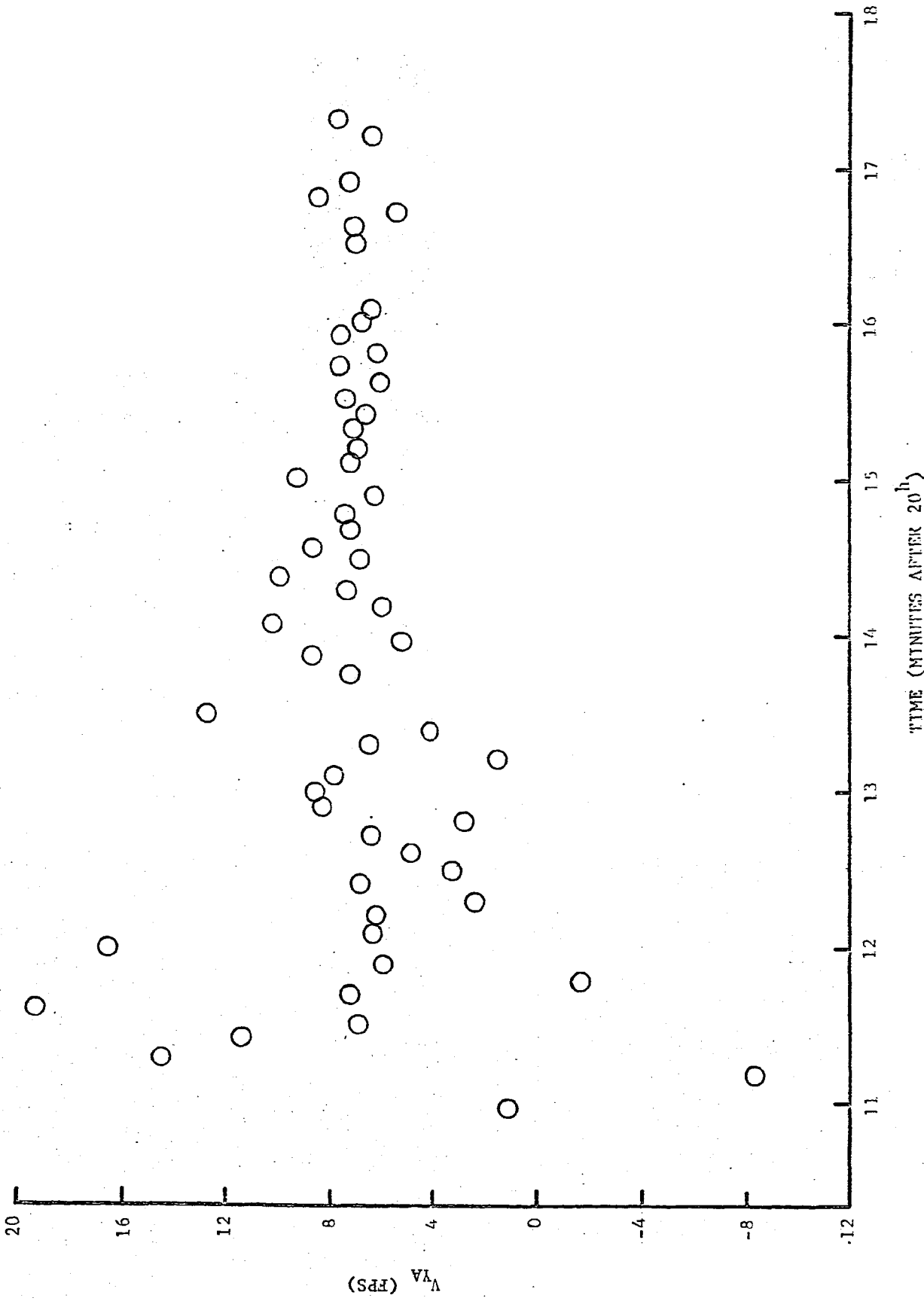


Figure 7-35. Landing Radar Y-Antenna Velocity Residuals (Onboard/MSPN II-S)

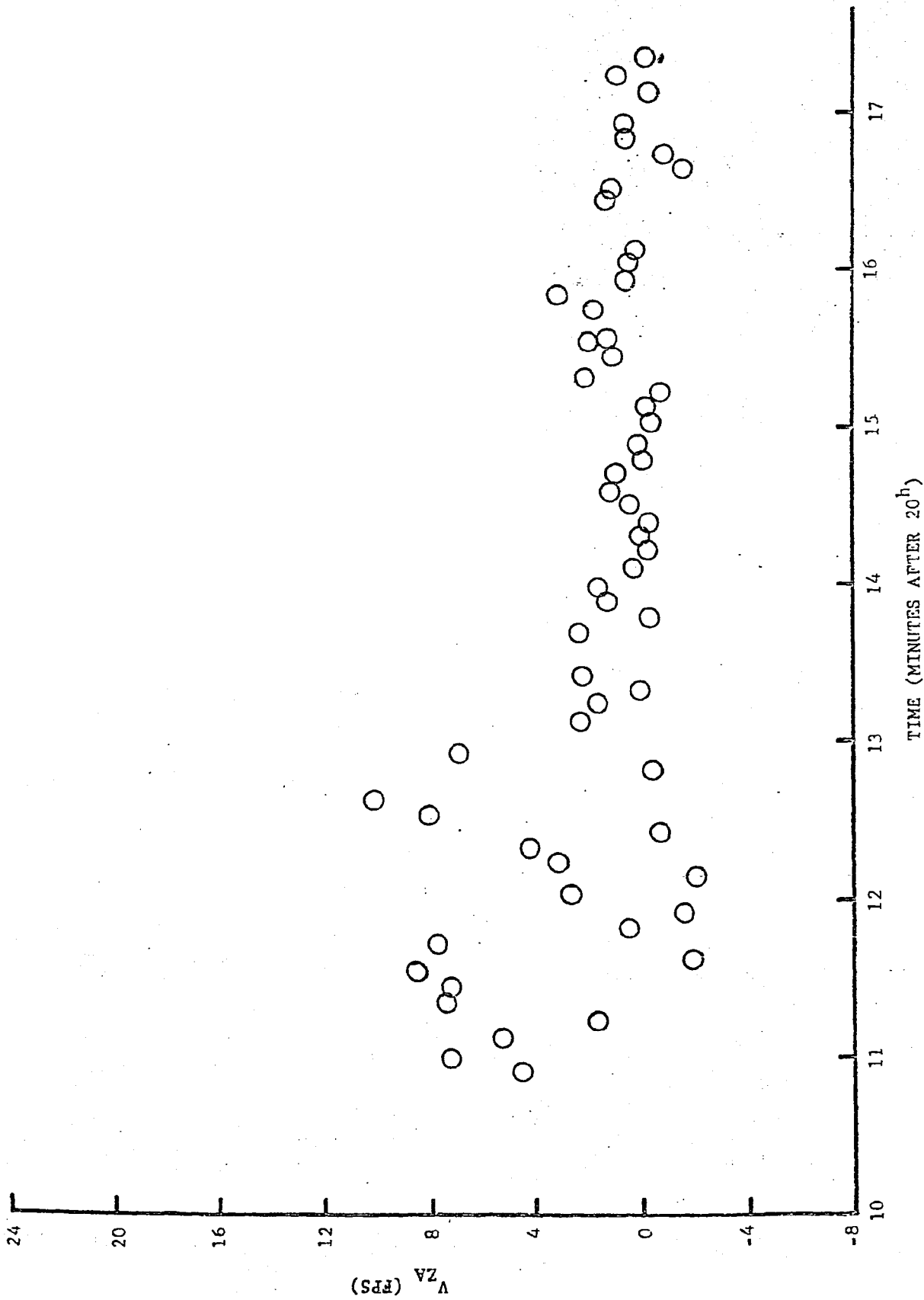


Figure 7-36. Landing Radar Z-Antenna Velocity Residuals (Onboard/MSFN II-S)

7.4.3 Lunar Surface Altitude Along Groundtrack

Landing Radar range residuals are used to compute an estimate of the relative altitude of the surface along the groundtrack of the piercepoint.

Surface altitude relative to the landing site is plotted versus angular range in Figure 7-37. Time ticks are indicated at the LR range read times (2-second intervals).

The ground track of the range beam piercepoint is shown in Figure 7-38. The plot is made on Lunar Maps ORB-II-6 and ORB-I-3 (scale 1: 100,000)*. The latitude does not agree with postflight estimates of Tranquility Base coordinates. Time ticks are at LR range read times and correspond to those on the surface altitude plot. The size of the range beam on the surface is indicated by the small ellipses drawn periodically along the groundtrack.

Little quantitative information can be obtained from Figure 7-38.1 except to note that the gentle upward slope of the terrain on the approach to the landing site is in general agreement with the surface altitude plot.

On Figure 7-38.2, surface altitude variations can be correlated to several prominent features:

The 170 ft drop in altitude between the readings at 102:39:37.19 and 102:39:39.19 correspond to range beam centers at the top and bottom of a cliff.

The point at 102:39:51.19 is centered in a fairly large crater. A depression of approximately 300 ft is clearly outlined in the surface altitude plot.

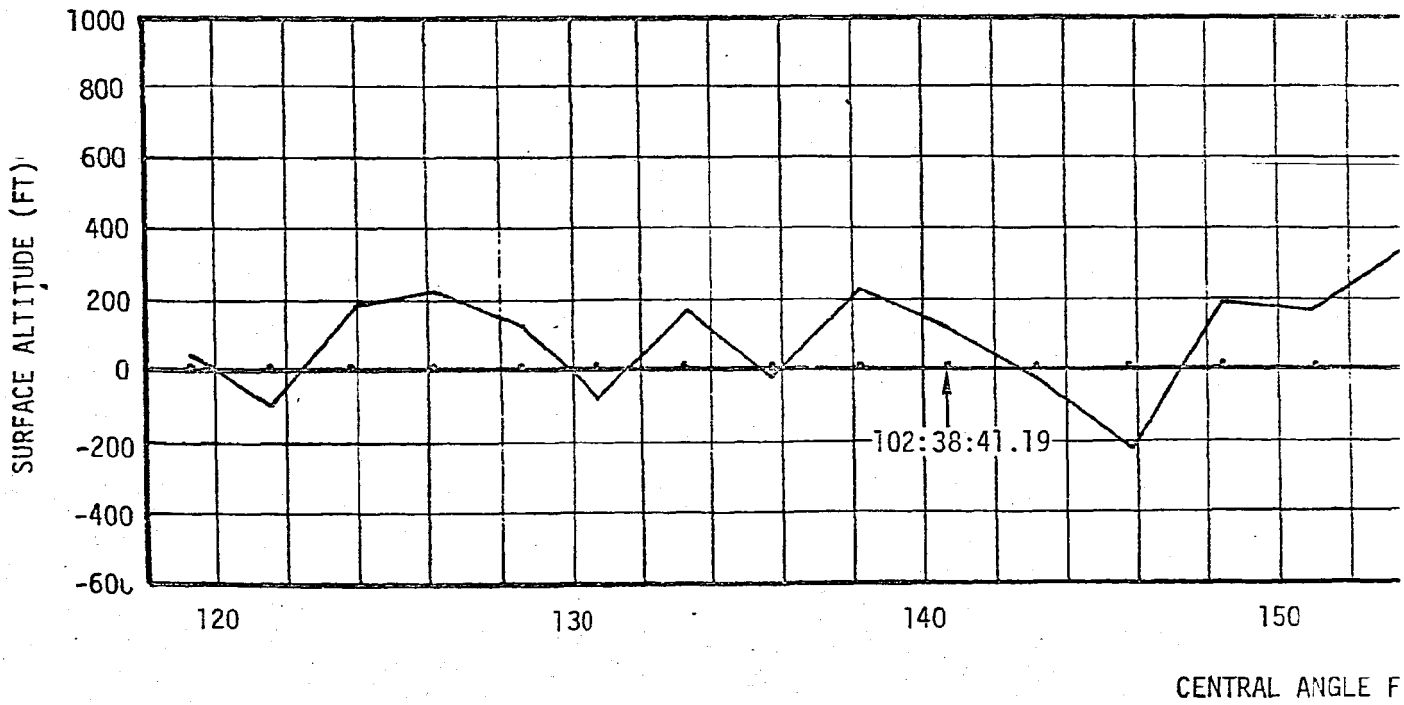
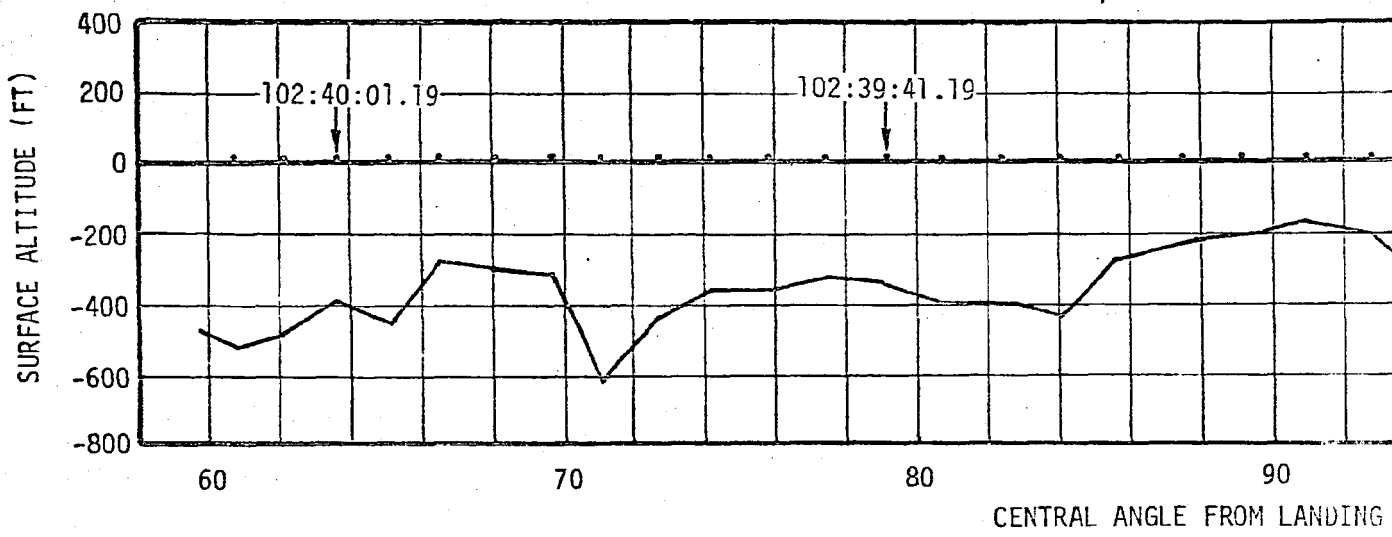
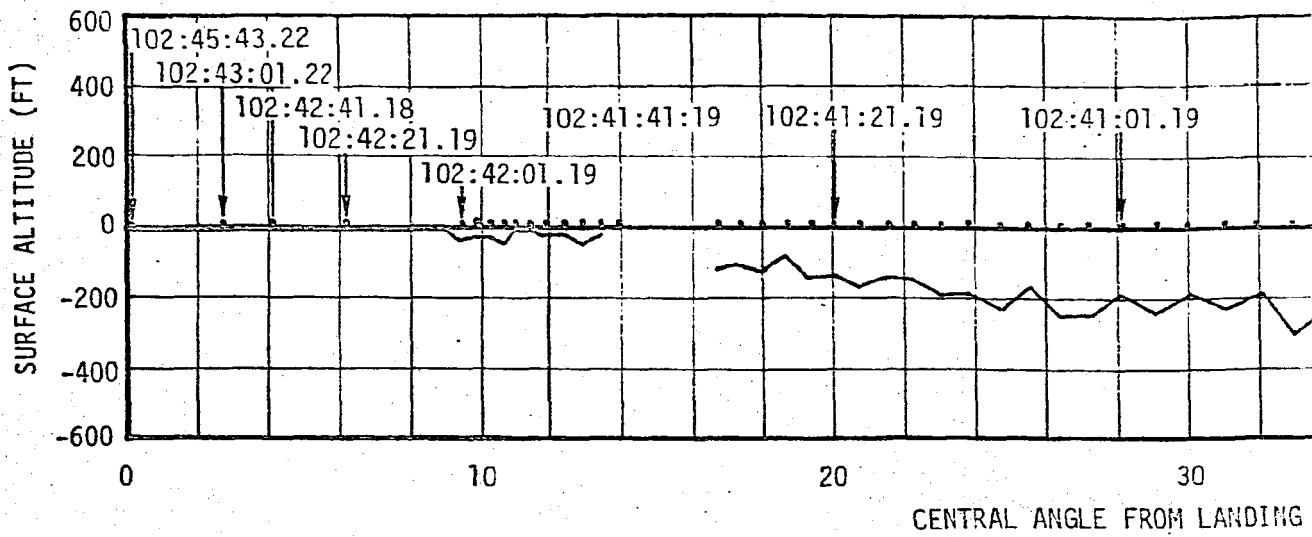
* Map legend: Contour lines (at 50 meter intervals) are indexed by an estimate of the radius in meters with the first three digits omitted. Crater markings such as 45R (110) indicate - Height of rim above terrain = 45 meters, Crater depth (floor to rim) = 110 meters.

The point at 102:39:23.19 falls inside a crater, and a depression of approximately 200 ft is indicated.

As the range beam intersection grows in size with increasing LM altitude, surface details become increasingly difficult to resolve. The overall downward terrain slope along the ground-track in Figure 7-38.3 is in general agreement with the surface altitude plot.

The altitude of the LM above the LLS radius during LR range data coverage is shown in Figure 7-39.

The data presented in this section results from a HOPE program orbit determination which includes LR velocity in the DC fit. This option has only recently become available and the results presented here are among the first obtained using Apollo 11 data. The principal effect of including LR velocity in the fit is to produce a more accurate relative velocity profile. Surface altitude plots, derived from earlier versions of the descent trajectory, show unrealistic terrain slopes due to small inplane velocity errors.



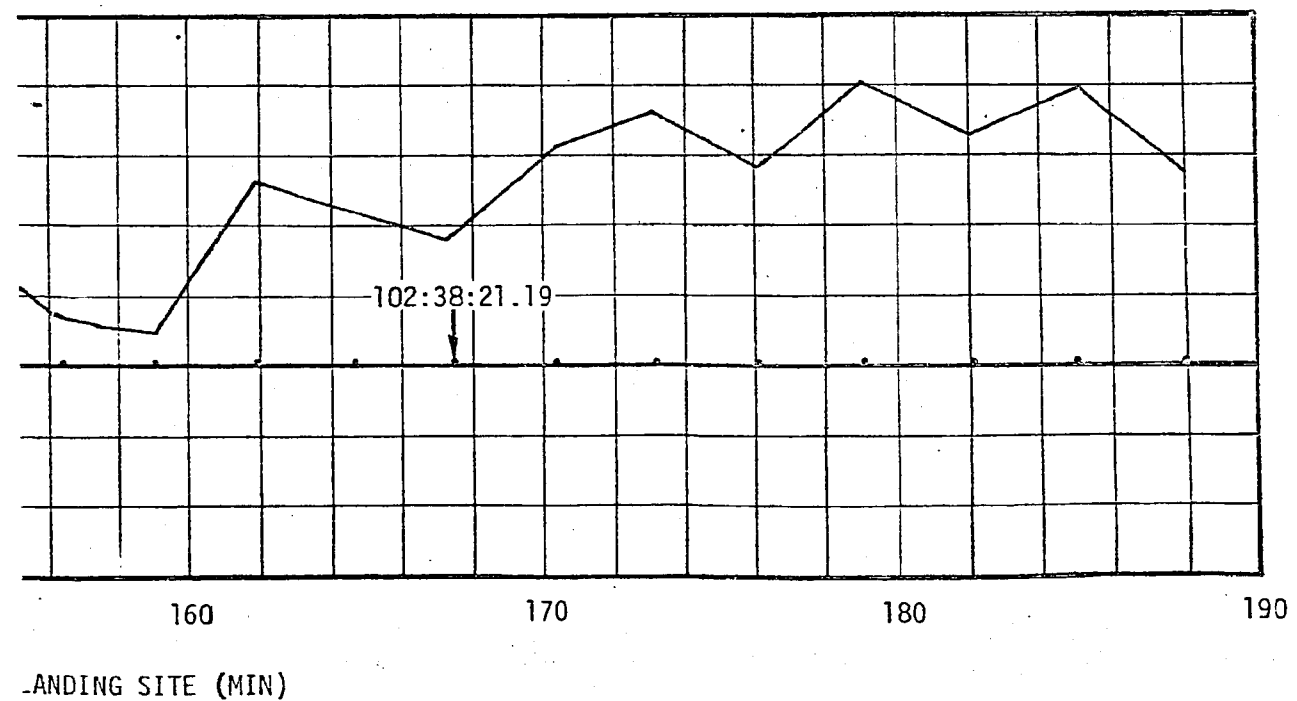
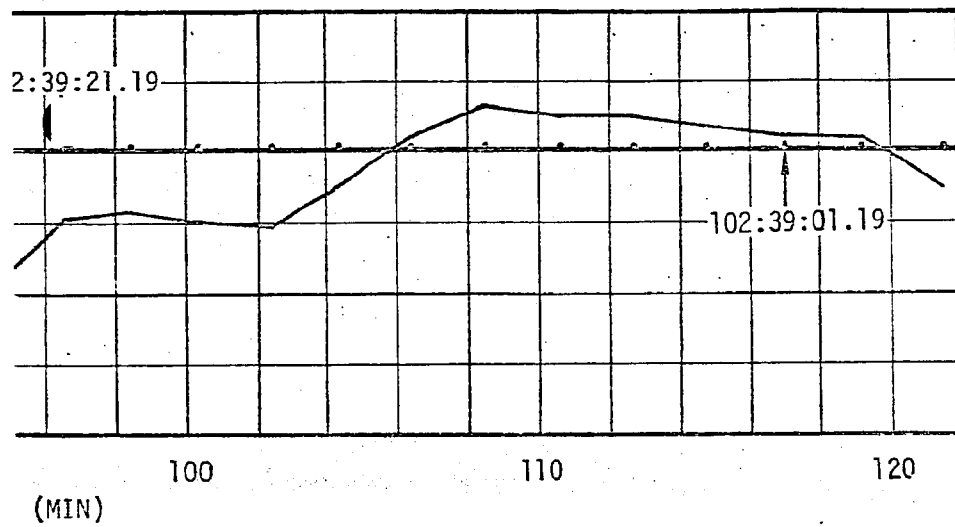
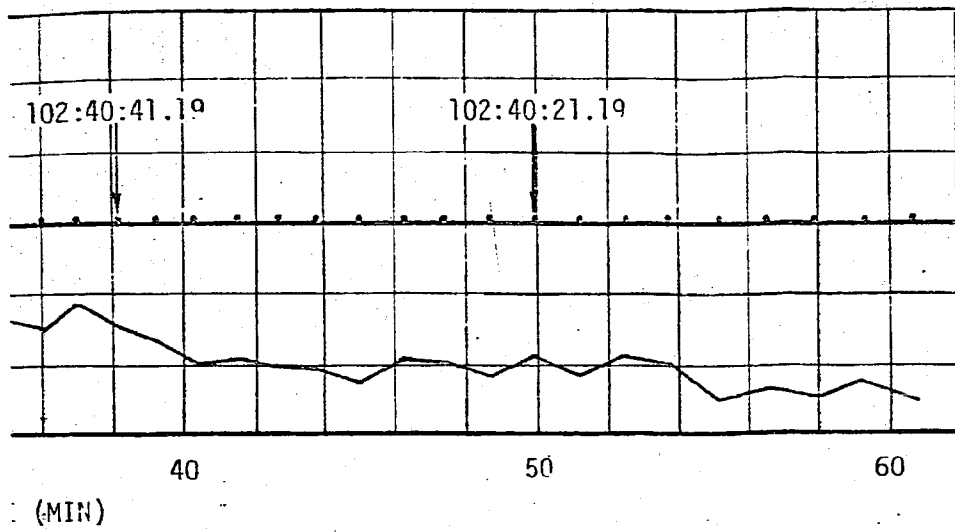
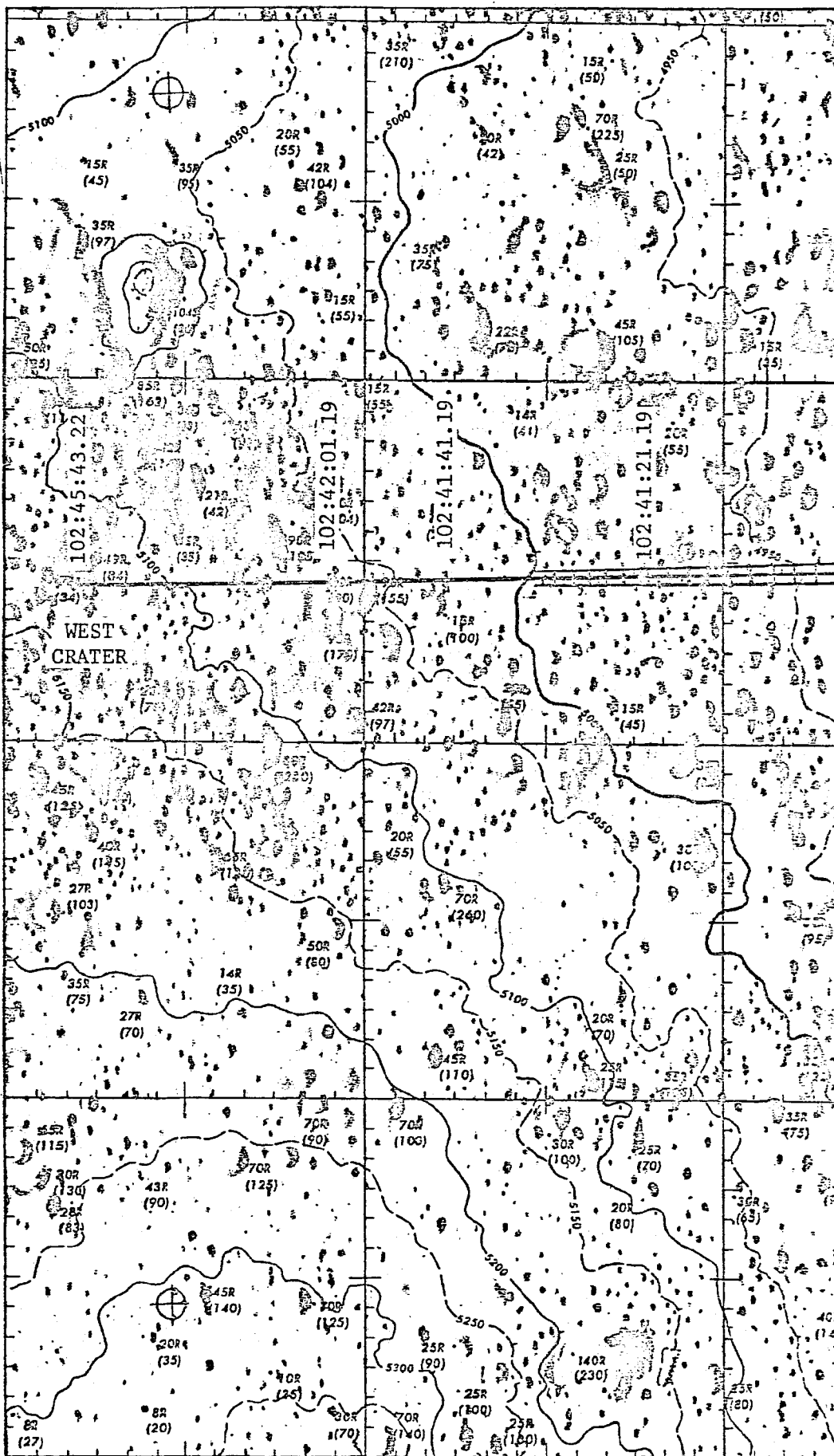


Figure 7-37 Surface Altitude Along Groundtrack.

FOLDOUT FRAME

LATITUDE (DEG:MIN)

+0:60
+0:48
+0:36
+0:24
+0:12

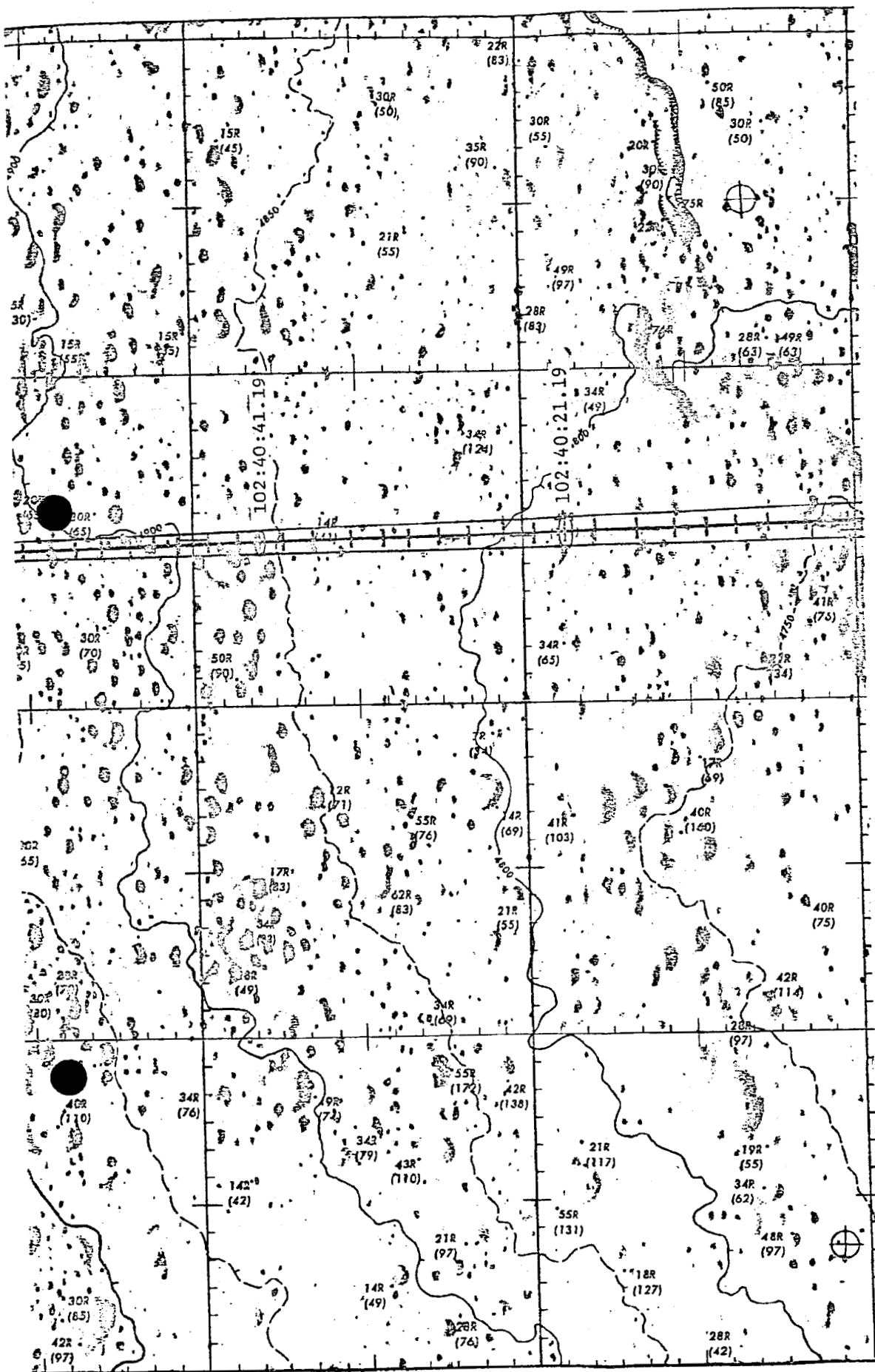


23:24

23:36

23:48

LONG



24:0

24:12

24:24

DE (DEG:MIN)

Figure 7-38.1 Groundtrack of LR Range Beam Piercepoint

α

+0:60

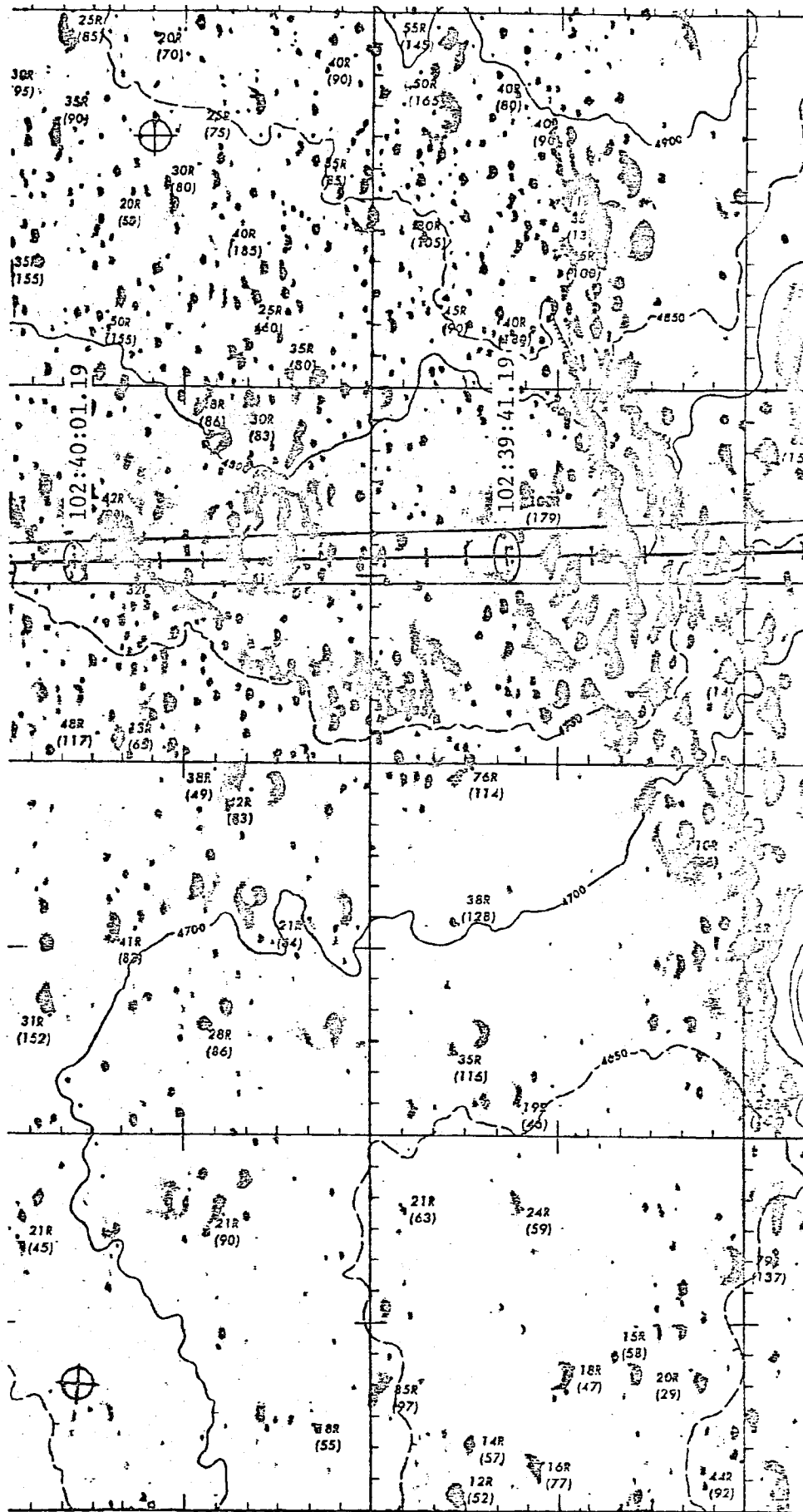
+0:48

+0:36

+0:24

+0:12

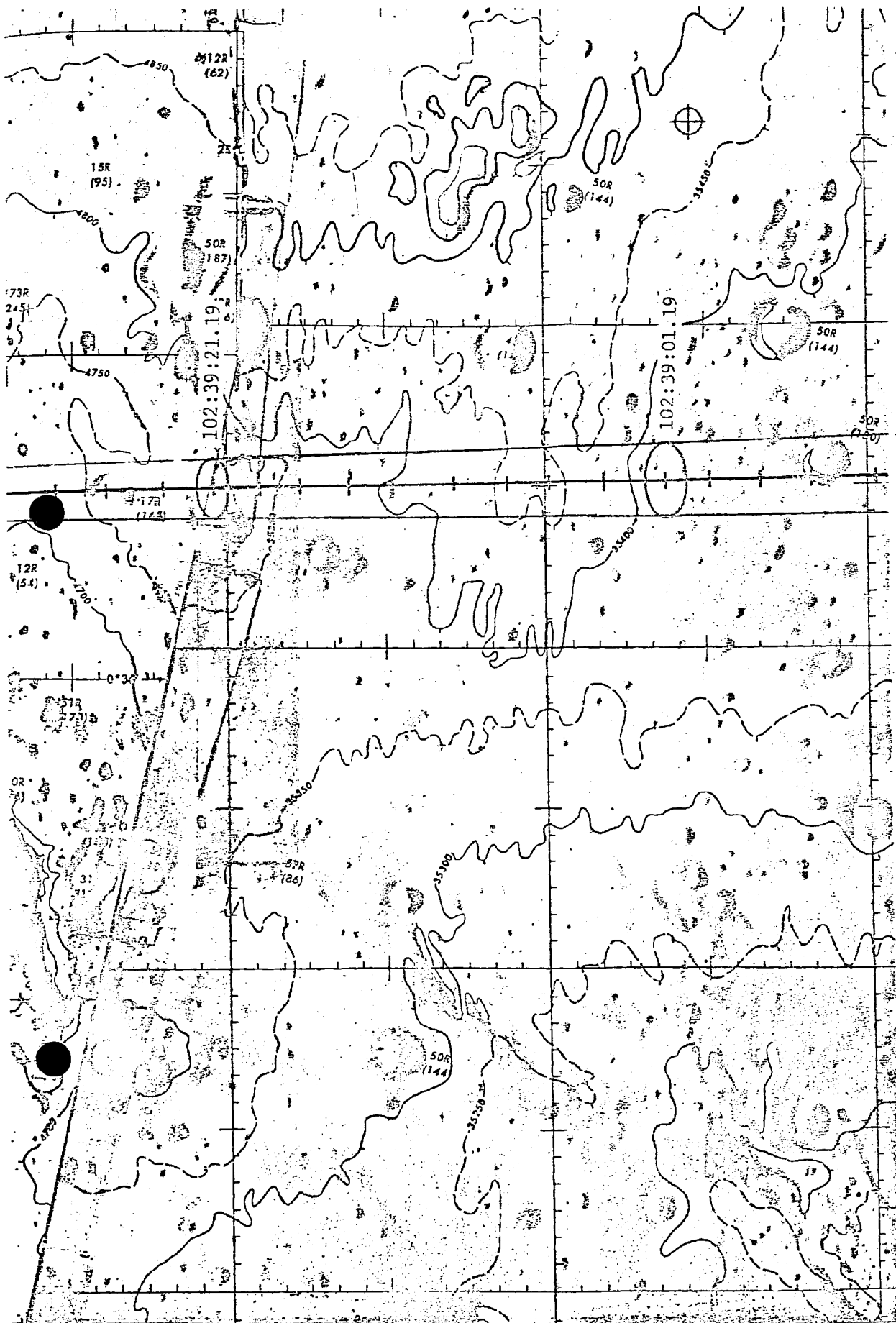
LATITUDE (DEG:MIN)



24:36

24:48

FOLDOUT FRAME



25:0
LONGITUDE (DEG:MIN)

25:12

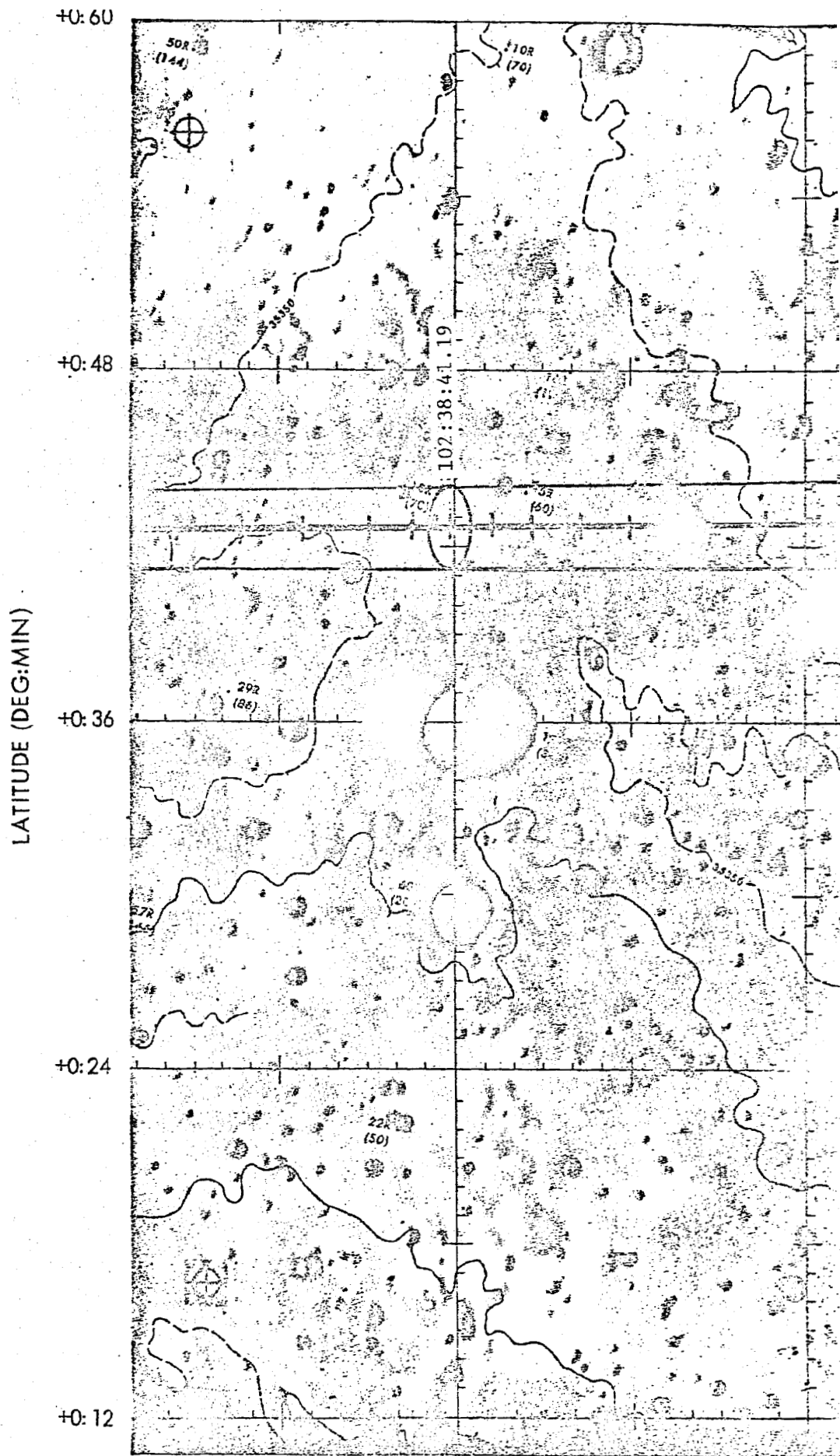
25:24

Figure 7-38.2

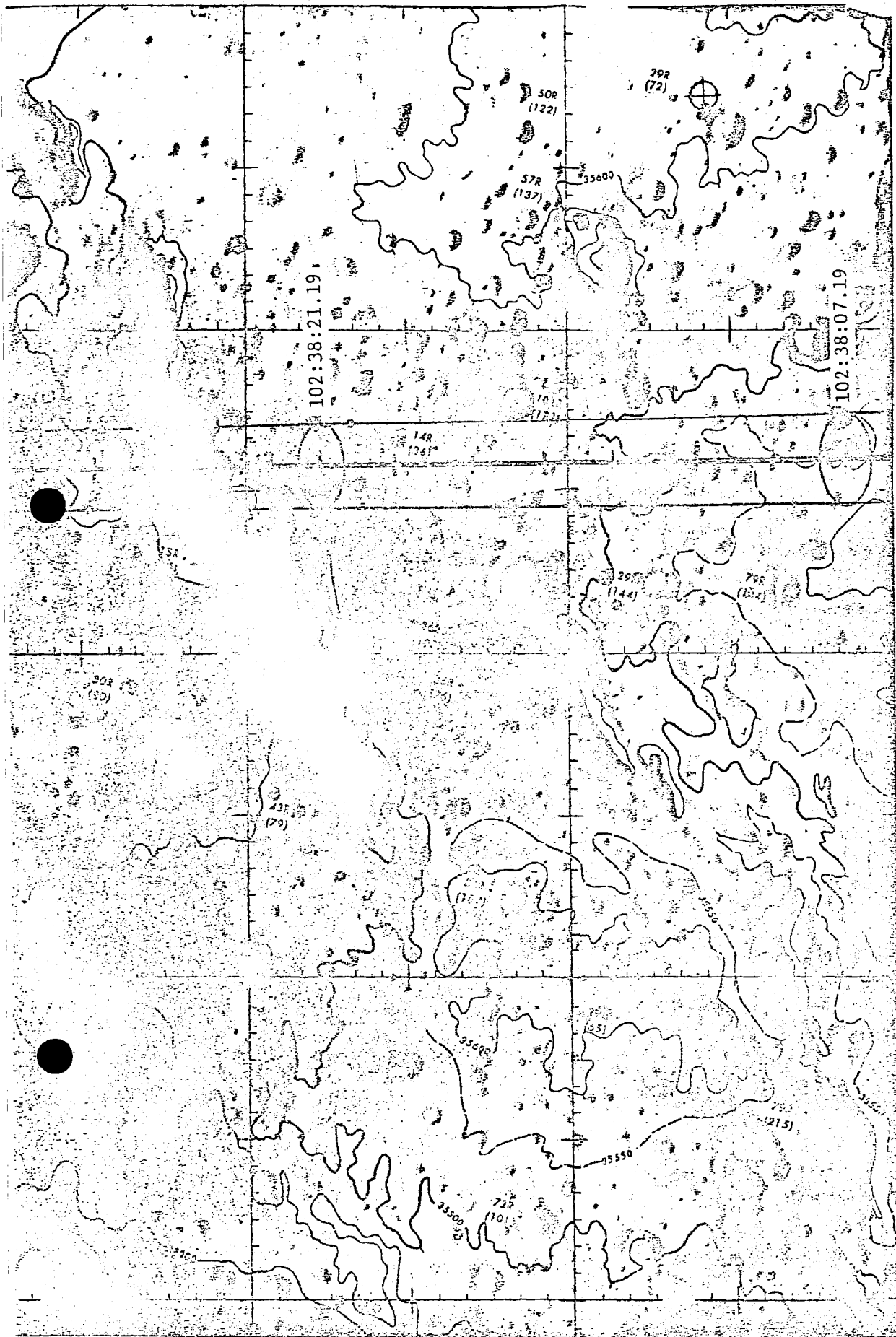
7-81

FOLDOUT FRAME

2



FOLDOUT FRAME



26:0

26:12

26:24

LONGITUDE (DEG:MIN)

Figure 7-38.3

7-83

FOLDOUT FRAME

2

Reproduced from
best available copy

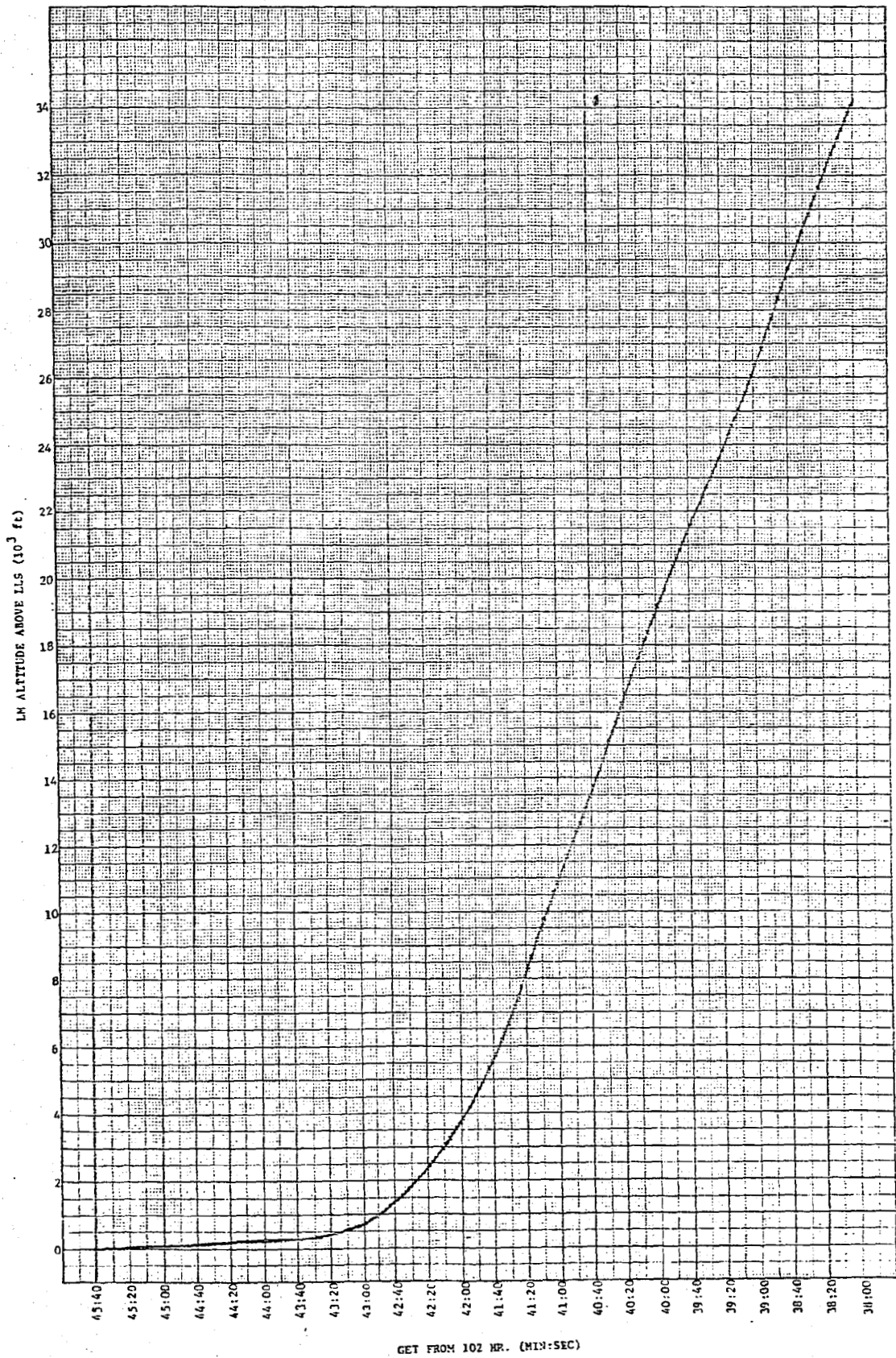


Figure 7-39 Altitude of LM During LR Range Sampling

Page Intentionally Left Blank

REFERENCES

1. Barnett, E. L. and Moore, H. L., "Analysis of Apollo 10 LM Rendezvous Radar Data and CSM VHF Ranging Data", TRW IOC 5522.8-67, 25 July 1969.
2. Barnett, E. L., "Method to Estimate Radar Data Noise", TRW IOC 3422.6-110, 17 May 1967.
3. Barnett, E. L., "Postflight Reconstruction of the Apollo 11 Descent Trajectory Using the HOPM Program", TRW IOC 5522.8-106, 5 November 1969.
4. "CSM/LM Spacecraft Operational Data Book - Part I", North American Rockwell Corporation, SNA 8-D-027, 1 November 1968.
5. Friedlander, M. M., "Analysis of LM Rendezvous Radar Data for Apollo 9", TRW IOC 5522.8-72, 9 May 1969.
6. Friedlander, M. M., "Preliminary Analysis of Apollo 11 Landing Radar Data", TRW IOC 5522.8-76, 15 August 1969.
7. Hinely, A. O. and Norris, P., "Lunar Model Analysis For Apollo 12", TRW IOC 5522.8-84, 31 August 1969.
8. Jankowski, S. C., "Summary of the CSM, LM Vectors Used to Generate the Preliminary NAT for Apollo 11", TRW IOC 5522.8-77, 19 August 1969.
9. "Master End Item Spec for Lunar Module", Grumman AED, LSP-470-2D.
10. Schiesser, E. L., "Apollo 11 Landing Site and LM Landing Position Determination", NASA/MSC Memorandum 69-FM41-349, 30 October 1969.

Page Intentionally Left Blank

APPENDIX A

Summary of CSM, LM Vectors Used to Generate the Preliminary NAT for Apollo 11

Appendix A documents the vectors used to generate the NAT trajectory in order that the user may know the quality of the trajectory. Since most of the vectors were not based on postflight fits but rather on RTCC vectors which were then propagated, propagation errors arise which can degrade the trajectory.

In order to reduce the error, the lunar orbit propagation times were kept to a minimum. Also, the total difference in position and velocity (which is a measure of the quality of the trajectory) were calculated at a common time point for adjacent trajectory intervals and tabulated in Table A.1 (CSM) and Table A.2 (LM) for user convenience. Whenever two intervals were separated by a maneuver, the ΔV as exhibited in Tables A.1 and A.2 represents the difference between the total velocity difference and the measured velocity of the maneuver.

Each table lists the vector ID and RTCC batch number, the source of the vector, the initial time of the vector, the propagation interval, the total differences in position and velocity of adjacent intervals, and comments relevant to a particular propagation interval. Maneuvers are listed between the appropriate free flight intervals for easy reference.

Most lunar trajectories were generated using RTCC SS2 (inclination constrained) solution vectors as opposed to SS1 (no a priori) solution vectors. Unlike the Apollo 10 SS2 vectors which were constrained to the pre-LOI1, rev 18, and rev 29 planes, the Apollo 11 SS2 vectors were constrained on a rev-to-rev basis. Each SS2 vector contained two revs of data and was constrained to the SS1 solution plane of one of these two revs (exceptions existed at maneuvers).

By using the new SS2 scheme, the Apollo 11 out-of-plane error was not allowed to accumulate as it did during the Apollo 10 mission.

It should be noted that the vectors used to generate the trajectory from insertion to TPI were based on free flight solutions utilizing MSFN data and not RTCC vectors. The quality of the vector from DOI to PDI was

questionable, but was included because no better vector was available at that time.

In general, the quality of the CSM trajectory was better than the quality of the LM trajectory during the rendezvous period.

Table A.1 Apollo 11 CSM NAT Trajectories

Vector ID	Source	Vector Time (d:h:m:s) GMT	Trajectory Interval (d:h:m:s)	ΔR (ft)	ΔV (fps)	Comment
CHUSS01	RTCC	16:17:33:48	16:16:22:03	346	0.6	T11 to extraction
MFLX101	RTCC	16:17:49:24	16:17:49:13	6066	0.8	Extraction to evasive
HRSK151	RTCC	17:03:33:18	16:18:12:04.39	---	0.2	Evasive to MCC2
PFEX034	RTCC	17:16:21:06	17:16:17:01.47	15438	0.6	MCC2 to 19°04'00"m
ANGX389	RTCC	19:07:45:24	19:04:00:00	---	---	19°04'00"m to L011
PIEX407	RTCC	19:17:47:36	19:17:27:52.38	2138	8.91	Rev 1
RIDK420	RTCC	19:17:47:36	19:17:27:52.38	6201	3.8	Rev 2 to L012
GNHX455	RTCC	19:22:07:24	19:19:28:00	16789	5.76	Revs 3 and 4
GNHX472	RTCC	20:02:02:12	19:21:43:36	5438	3.50	Rev 5
HANX479	RTCC	20:04:02:48	20:01:37:00	5092	3.56	Rev 6
HANX485	RTCC	20:05:58:48	20:03:37:00	3963	3.17	Rev 7
CROX490	RTCC	20:07:57:18	20:05:37:00	4629	6.03	Rev 8
CROX496	RTCC	20:10:01:42	20:07:37:00	4093	2.10	Rev 9
CROX501	RTCC	20:11:53:36	20:09:32:00	10114	8.10	Rev 10
MFLX507	RTCC	20:13:52:00	20:11:32:00	7597	6.39	Revs 11 and 12
ANGX067	RTCC	20:15:50:18	20:13:27:00	5166	6.36	Rev 13 pre sep
ACNX327	RTCC	20:18:12:36	20:17:22:00	3866	2.36	Post sep and rev 14
HANX535	RTCC	20:19:46:54	20:18:11:58.24	5126	3.48	Rev 15
CDSX322	RTCC	20:21:47:06	20:21:17:00	8472	5.12	Rev 16
GNHX550	RTCC	20:23:43:12	20:23:17:00	3921	3.15	Rev 17
GNHX555	RTCC	21:01:41:42	21:01:17:00	4023	3.07	Rev 18
CROX564	RTCC	21:03:40:48	21:03:17:00	3915	3.47	Rev 19
HANX569	RTCC	21:05:38:06	21:05:12:00	3754	3.44	Rev 20
HANX573	RTCC	21:07:36:18	21:07:12:00	4342	4.21	Rev 21
HANX579	RTCC	21:09:34:36	21:09:12:00	3814	3.05	Rev 22
CROX584	RTCC	21:11:33:06	21:11:07:00	6678	5.94	Rev 23
ACHX595	RTCC	21:15:29:12	21:15:02:00	5070	3.38	Revs 24 and 25
CHEX600	RTCC	21:17:27:30	21:15:02:00	5813	2.99	Rev 26
HANX625	RTCC	21:21:30:00	21:20:57:00	4052	4.46	Locking and sep revs 27 and 28
HANX651	RTCC	22:00:03:06	22:00:02:07.1	3065	2.85	Post sep and rev 29
CDSX657	RTCC	22:01:20:12	22:00:02:07.1	7604	5.00	Rev 30
ANGX695	RTCC	22:07:37:36	22:02:57:00	---	---	FEI
PAO	RTACF ^{AW}	22:20:01:54.53	22:14:00:00	---	---	MCC5
PAO	RTACF	23:05:31:59.99	22:20:02:05.29	19159	0.1	---
PAO	RTACF	23:15:31:59.99	23:15:30:00	20419	0.3	---
PAO	RTACF	24:01:32:00	24:01:30:00	21098	4.7	CSH/LH sep
PAO	RTACF	24:16:35:06	24:16:21:18	---	---	Entry interface

*Real Time computer center (RTCC) on-line listing.
 **Real Time auxiliary computer facility (RTACF) checkout monitor for the Postflight Analysis Office (PAO).

Table A.2 Apollo 11 LN NAT Trajectories

Vector	Source	Vector Time (d:h:m:s)GMT	Trajectory Interval (d:h:m:s)	Interval (d:h:m:s)GMT	AR (ft)	ΔV (fps)	Comment
MSFN FIT	TRW	20:19:08:06.6	20:17:45:38.	20:19:08:14.1	---	---	Undock to DOI
REAL TIME	MPAD	20:20:04:35.17	20:19:08:43.9	20:20:05:04.4	---	---	DOI to PDI
MSFN FIT	TRW	21:18:01:14.88	21:18:01:14.88	21:18:51:34.7	8400	0.7	INS to CSI
MSFN FIT	TRW	21:19:49:46	21:18:52:21.7	21:19:49:46	8558	1.4	CSI to CDH
MSFN FIT	TRW	21:19:50:04.08	21:19:50:04.08	21:20:36:30.8	---	---	CDH to TPI

APPENDIX B

Supplementary Data

Appendix B contains supplementary information which is too detailed for the main body of the report. This information includes a summary of the radar data used in each BET fit segment, a summary of ground and on-board data weights used in HOPE, a summary of the USBS station locations, and a summary of the components used in the R2 and L1 lunar potential models.

Tables B.1 and B.2 list by vehicle statistics computed from the data used in each BET fit, the type and number of observables, and the mean and standard deviations obtained from the residuals calculated in the final iteration of the fit. The range statistics are in feet, doppler units are cycles per second, range rate units are in feet per second, and angular units are degrees.

Table B.3 lists the data weights used in the HOPE Program for ground based radar data and Table B.4 lists the data weights used in the HOPE Program for onboard data by type of observable.

Table B.5 lists the terms of the R2 lunar potential model.

Table B.6 lists the terms of the Langley 1 lunar potential mode, a modification of the R2 model.

Table B.7 lists the S-band tracking stations and their locations as used in the Apollo 11 postflight analysis. All locations are referenced to the Fischer Ellipsoid of 1960. The mean surface refractivity numbers for each station for the month of July are also listed.

Table B.1 CSM BET Ground Based Tracking Data Statistics

<u>Station</u>	<u>Data Type</u>	<u>Number of OBS</u>	<u>Mean</u>	<u>σ</u>
Rev 13 Segment				
MAD	2-way doppler	122	-.013	.236
MIL	3-way doppler	103	-.009	.246
ACN	3-way doppler	101	-.018	.235
Rev 14 Segment				
MAD	2-way doppler	142	.002	.186
GDS	3-way doppler	138	.009	.182
ACN	3-way doppler	63	-.004	.171
Rev 25 Segment				
MAD	2-way doppler	136	.002	.160
MIL	3-way doppler	52	.001	.165
ACN	3-way doppler	47	.007	.157
Rev 26 Segment				
MAD	2-way doppler	128	.001	.223
MIL	3-way doppler	114	.0005	.199
ACN	3-way doppler	114	.005	.188
GDS	3-way doppler	70	.006	.182

Table B.2 LM BET Ground Based and Onboard Tracking Data Statistics

<u>Station</u>	<u>Data Type</u>	<u>Number of OBS</u>	<u>Mean</u>	<u>σ</u>
Undock to DOI Segment				
RID	2-way doppler	141	-.023	.193
CYI	3-way doppler	140	-.049	.202
ACN	3-way doppler	112	-.014	.193
ANG	3-way doppler	137	.030	.219
MIL	3-way doppler	114	-.012	.189
DOI to PDI Segment				
	Sextant shaft	13	-.0097	.015
	Sextant trunnion	13	-.0004	.004
	VHF ranging	18	-26.000	74.000
Insertion to CSI Segment				
RID	2-way doppler	74	-.022	.315
BDA	3-way doppler	69	-.019	.319
ANG	3-way doppler	72	.059	.315
ACN	3-way doppler	60	.011	.320
MIL	3-way doppler	63	.014	.316
CDH to Post-TPI Segment				
	Sextant shaft	31	.030	.026
	Sextant trunnion	31	.011	.023
	VHF ranging	29	-394.000	222.000
	Rend. radar shaft	65	-.012	.107
	Rend. radar trunnion	65	-.084	.056
	Rend. radar range	65	142.000	271.000
	Rend. radar range rate	65	-.115	.543

Table B.3 Ground Based Radar Data Weighting

Data Type	Radar	Weighting
Range	USB: 30-ft. antenna 85-ft. antenna	600 ft.
Doppler (2-way)	USB: 30-ft. antenna 85-ft. antenna	0.1 cycle/sec.
Doppler (3-way)	USB: 30-ft. antenna 85-ft. antenna	0.1 cycle/sec.

Table B.4 Onboard Data Weighting

Data Type	Shaft	Trunnion	Range	Range Rate
Rendezvous radar	.01	.01	30.	1.
Sextant	.001	.001		
VHF ranging			30.	

Table B.5 R2 Lunar Potential Model

Term	Value
J2	2.07108×10^{-4}
J3	-2.1×10^{-5}
C22	2.0716×10^{-5}
C31	3.4×10^{-5}
All other harmonics are zero	

Table B.6 L1 Lunar Potential Model

Term	Value
J2	2.07180×10^{-4}
J3	-2.1×10^{-5}
C22	2.0716×10^{-5}
C31	3.4×10^{-5}
C33	2.583×10^{-6}
All other harmonics are zero	

Table B.7 USBS Station Locations

Station	Antenna	Identification	Latitude* (deg)	Longitude* (deg)	Altitude* (ft)	Surface Refractivity
Antigua	30'	ANG	17.01692	298.24715	141.08	378
Ascension	30'	ACN	-7.95510	345.67330	1843.83	353
Bermuda	30'	BDA	32.35195	295.34287	68.90	377
Canary Island	30'	CYI	27.76454	344.36519	567.69	343
Honeysuckle Creek	85'	HSK	-35.58361	148.97805	3757.55	296
Carnarvon	30'	CRO	-24.90705	133.72620	82.00	325
Goldstone	85'	GDS	35.34154	243.12655	2975.066	279
Grand Bahama	30'	GMB	26.63286	281.76234	16.40	386
Guam	30'	GWM	13.31062	144.73747	416.67	373
Guaymas	30'	GYM	27.96382	249.27943	62.34	368
Hawaii	30'	HAW	22.12666	200.33528	3772.97	308
Madrid	85'	MAD	40.45514	355.83183	2551.18	299
Merritt Island	30'	MIL	28.50866	279.30738	32.81	385
Texas	30'	TEX	27.65428	262.62220	32.81	395
Honeysuckle Creek Wing	85'	NBE	-35.40111	148.98153	2199.15	296
Goldstone Wing	85'	PIR	35.38952	243.14078	3186.02	279
Madrid Wing	85'	RID	40.42843	355.75128	2524.93	299

*All quantities are referenced to the Fischer Ellipsoid of 1960.

APPENDIX C

LM Rendezvous Radar Data, CSM VHF Ranging Data and CSM Sextant (Apollo 11)

The LM rendezvous radar data that was used in the analysis are listed in the two card format of the HOPE orbit determination program. The first card specifies the vehicle taking the observation, the vehicle that is being observed, the time of the observation (year (mod 1900), month, day, hour, minute, and second (GMT)), three code numbers, shaft observable, trunnion observable, range observable, and range rate observable. The second card specifies the inner, middle, and outer gimbal angles. The units are feet, degrees, and seconds.

The CSM VHF ranging data are also listed in the same format. The card format differences are the following: 1) vehicle ID's are reversed, 2) code numbers are different, 3) range is the only observable, and 4) gimbal angles are not needed to process the ranging data.

The CSM sextant data are also listed. The card format is also similar to the rendezvous radar cards.

DOI TO PDI

VEH1	VEH2	YMMDDHHMMSS	SSS	XFSHAFT(DEG)	INNER(DEG)	MIDDLE(DEG)	TRUN(DEG)	RANGE(FT)	RRATE(FPS)
CSM	LEM	69	720184617	695	62	88.2641602	29.1824434		1
CSM	LEM	69	7201925	3.547114		54.6350098	4.6472168	338.5437012	2
CSM	LEM	69	7201925	34.824	62	.5163574	23.3789155	84093.437	1
CSM	LEM	69	7201925	40.676	62	327.7551270	1.8347168	2.5158691	2
CSM	LEM	69	7201926	3.816114		.6372070	23.3349702		1
CSM	LEM	69	7201926	3.816114		327.9528809	1.8457031	2.6037598	2
CSM	LEM	69	7201927	29.605114				89136.613	1
CSM	LEM	69	7201927	29.605114				97035.563	2
CSM	LEM	69	7201926	10.746	62	1.2524414	22.9257295	3.0322266	1
CSM	LEM	69	7201928	39.305114		329.1064453	1.8676758	104205.380	2
CSM	LEM	69	7201927	46.867	62	.4174805	22.0852754	2.5927734	1
CSM	LEM	69	7201930	19.016114		332.3254395	2.0214844	115871.521	2
CSM	LEM	69	7201928	56.437	62	359.9121094	22.3022554		1
CSM	LEM	69	7201930	26.348	62	333.7866211	2.1643066	2.5048828	2
CSM	LEM	69	7201931	33.504114		359.7253418	22.1319673		1
CSM	LEM	69	7201931	33.504114		336.0058594	2.2631836	2.5158691	2
CSM	LEM	69	7201931	12.766	62	359.5056152	22.1786592	125897.112	1
CSM	LEM	69	7201932	34.348114		336.9177246	2.3510742	2.5268555	2
CSM	LEM	69	7201932	12.098	62	.5493164	22.4670503	134950.523	1
CSM	LEM	69	7201932	12.098	62	337.7526855	2.5158691		2
CSM	LEM	69	7201932	12.098	62			3.5156250	2

CSM	LEM	69	720193335	516114				144915.354	1
									2
CSM	LEM	69	7201933	1.707 62	359.3627930	23.0026338			1
					338.0273437	2.6696777		3.0212402	2
CSM	LEM	69	720193436	426114				155791.600	1
									2
CSM	LEM	69	720193538	477114				167700.785	1
									2
CSM	LEM	69	7201934	1.316 62	.2197266	22.7362154			1
					339.1369629	2.8125000		3.8012695	2
CSM	LEM	69	720193638	824114				180278.346	1
									2
CSM	LEM	69	720193612	625 62	1.1206055	22.1814058			1
					340.9167480	3.1311035		4.8120117	2
CSM	LEM	69	720193741	965114				194435.693	1
									2
CSM	LEM	69	720193718	957 62	.2307129	22.0221040			1
					341.3562012	3.2958984		4.4714355	2
CSM	LEM	69	720193850	777114				211023.488	1
									2
CSM	LEM	69	720193829	965 62	.1098633	22.7060030			1
					340.7299805	3.5156250		4.8559570	2
CSM	LEM	69	720193953	195114				227125.195	1
									2
CSM	LEM	69	720194053	867114				243712.990	1
									2
CSM	LEM	69	720194154	324114				261212.203	1
									2
CSM	LEM	69	720194254	785114				279562.070	1
									2
CSM	LEM	69	720194356	004114				299127.164	1
									2

INS TO CSI

VEH1	VEH2	YMMDDHHMSS.SSS	XFSHAFT(DEG)	INNER(DEG)	TRUN(DEG)	RANGE(FT)	RRATE(FPS)
				MIDDLE(DEG)	OUTER(DEG)		
CSM	LEM	69 721182722.246 62	359.3078613	22.7444551			
			36.1230469	1.6699219	1.5380859		
LEM	CSM	69 721183052.766 11	359.9890137	.1208496	888923.836		-193.9901
			259.4860840	1.0437012	.1098633		
LEM	CSM	69 721183158.105 11	359.5605469	.0769043	876692.312		-179.5511
			256.1901855	1.0437012	.1428223		
LEM	CSM	69 7211833 7.746 11	359.9670410	.0439453	864835.992		-164.4841
			251.7846680	1.1096191	.2087402		
LEM	CSM	69 721183415.707 11	357.8906250	359.3188477	854105.273		-150.6721
			249.9938965	.9448242	.8789063		
LEM	CSM	69 721183523.215 11	354.1003418	355.5175781	844350.078		-136.8601
			249.8620605	4.0539551	4.3395996		
LEM	CSM	69 721183641.754 11	354.1113281	2.9113770	834294.719		-121.7931
			245.4125977	1.8786621	357.1325684		
LEM	CSM	69 7211837 .504 11	356.2646484	4.5483398	832043.516		-118.0261
			242.1936035	3.3398438	355.4736328		
LEM	CSM	69 721183719.246 11	358.7145996	2.5708008	829942.398		-114.2601
			238.6779785	3.8562012	357.5939941		
CSM	LEM	69 721183749.316 62	.8459473	21.9534395			
			.8569336	5.7238770	8.0639648		
LEM	CSM	69 721183753.215 11	.9667969	358.7475586	826115.359		-107.9821
			234.4592285	5.5480957	1.6040039		
CSM	LEM	69 721183753.574 62	.8789063	21.9479463			
			.6262207	5.7458496	8.1188965		
CSM	LEM	69 721183842.324 62	358.9782715	21.8518159	7.3608398		
			357.9016113	6.0974121	818270.469		
CSM	LEM	69 7211839 4.004114					
					812133.586		

CSI TO CDH

VEH1	VEH2	YMMDDHHMMSS	SS	XF	SHAFT(DEG)	INNER(DEG)	MIDDLE(DEG)	TRUN(DEG)	RANGE(FT)	RRATE(FPS)
CSM	LEM	69	72119	217	414114				746693.828	1
CSM	LEM	69	72119	7	2.098114				731503.539	2
CSM	LEM	69	72119	8	2.566114				728283.195	1
CSM	LEM	69	72119	9	4.617114				724941.336	2
CSM	LEM	69	72119	943	887 62	1.1096191	22.5576875			1
CSM	LEM	69	72119	946	906 62	259.3762207	1.5930176	3.2080078		2
CSM	LEM	69	72119	946	906 62	1.0656738	22.6428316			1
CSM	LEM	69	72119	5.555	62	259.1455078	1.5930176	3.1970215		2
CSM	LEM	69	72119	5.555	62	.8898926	22.8378389			1
CSM	LEM	69	72119	9.316114		258.0139160	1.6259766	3.1640625		2
CSM	LEM	69	72119	1029.414	62	1.4282227	21.2393281	721417.187		1
CSM	LEM	69	72119	1113.117114		258.4094238	1.6040039	3.2629395		2
CSM	LEM	69	72119	1215.125114				717953.797		1
CSM	LEM	69	72119	1258.777	62	.2856445	22.6373384	714490.414		2
CSM	LEM	69	72119	1258.777	62	249.4555664	2.5158691	4.0979004		1
CSM	LEM	69	72119	13.547	62	.2416992	22.6675508			2
CSM	LEM	69	72119	1320.387114		249.3347168	2.5268555	4.0869141		1
CSM	LEM	69	72119	1352.465	62	359.3298340	22.5933931	710844.742		2
CSM	LEM	69	72119	1357.098	62	246.7749023	3.0102539			1
CSM	LEM	69	72119	1357.098	62	359.3847656	22.7691743	4.1857910		2
CSM	LEM	69	72119	1357.098	62	246.3684082	3.0651855	4.3505859		1

LEM	CSM	69	721193149.117	11	359.0551758	359.9011230	632287.039	-90.4031
CSM	LEM	69	7211932	1.266	62	70.4113770	7.6574707	359.4946289
CSM	LEM	69	7211932	2.215114		358.6816406	22.0578096	
CSM	LEM	69	7211932	2.215114		194.2053223	.0219727	359.9340820
CSM	LEM	69	7211932	2.215114				630700.781
CSM	LEM	69	721193224.527	62	358.4619141	22.0468233		
LEM	CSM	69	721193256.945	11	193.1066895	.0549316	359.8352051	
CSM	LEM	69	7211933	5.035114		358.5717773	357.1875000	-93.5421
CSM	LEM	69	7211933	5.035114		67.5878906	1.3293457	
LEM	CSM	69	7211934	4.824	11	1.0986328	359.3518066	-96.0531
LEM	CSM	69	721193512.375	11	61.8640137	7.3498535	.2966309	
LEM	CSM	69	721193617.215	11	357.2753906	2.5378418	613076.797	-98.5651
LEM	CSM	69	721193736.617	11	62.3693848	355.5834961	357.1545410	
LEM	CSM	69	721193842.035	11	355.4956055	4.6252441	606623.359	-101.0761
LEM	CSM	69	721194047.637	11	61.1718750	5.3063965	354.4299316	
LEM	CSM	69	7211941	6.355	11	356.5393066	354.8803711	-104.8431
LEM	CSM	69	721194125.125	11	56.3269043	4.1308594	4.3615723	
LEM	CSM	69	721194144.145	11	.2307129	354.8474121	591540.312	-106.7261
LEM	CSM	69	721194145	11	49.5922852	10.7226562	4.6691895	
LEM	CSM	69	7211941699		353.4411621	357.3193359	577807.992	-111.7481
LEM	CSM	69	721194125.125	11	50.3833008	3.9111328	1.7138672	
LEM	CSM	69	721194125.125	11	1.7578125	356.2536621	575706.875	-112.3761
LEM	CSM	69	721194125.125	11	41.2426758	9.5581055	3.5156250	
LEM	CSM	69	721194144.145	11	.4284668	.2526855	573680.797	-113.6321
LEM	CSM	69	721194144.145	11	41.6491699	4.9438477	359.2749023	
LEM	CSM	69	7211942	4.105	11	4.3505859	571429.594	-114.2601
LEM	CSM	69	721194227.074	11	42.6379395	.7910156	355.1220703	
LEM	CSM	69	721194246.027	11	356.6601562	4.6472168	569103.359	-114.2601
LEM	CSM	69	721194246.027	11	43.5168457	359.3627930	354.8583984	
LEM	CSM	69	7211943	5.457	11	3.7792969	566551.992	-115.5151
LEM	CSM	69	721194324.848	11	43.9343262	1.7578125	355.5285645	
LEM	CSM	69	721194324.848	11	44.2199707	3.2409668	564375.836	-116.1431
LEM	CSM	69	721194324.848	11	44.0222168	2.8564453	355.9130859	
LEM	CSM	69	721194324.848	11	353.3203125	2.4938965	562124.633	-116.7711
LEM	CSM	69	721194324.848	11	44.0222168	2.5927734	356.6601562	
LEM	CSM	69	721194324.848	11	357.0336914	1.5380859	559873.437	-118.0261
LEM	CSM	69	721194324.848	11	39.4189453	2.8454590	357.7917480	

LEM	CSM	69	721194347.367	11	2.8564453	•4614258	557171.992	-118.0261
					32.5195313	•0988770	359.0332031	2
LEM	CSM	69	7211944	7.426	2.4279785	359.2199707	554770.719	-118.6541
					32.0141602	4.4384766	•4504395	2

CDH TO TPI

VEH1	VEH2	YMMDDHHMMSS	SSS	XF	SHAFT(DEG)	INNER(DEG)	MIDDLE(DEG)	TRUN(DEG)	RANGE(FT)	RRATE(FPS)
LEM	CSM	69	721195	358	5058594	353	4082031	481156	477	-124.3041
					7.8552246	353	7487793	6.3281250		
LEM	CSM	69	721195	3.687	1.5380859	3.2629395	473127	199		-123.6771
					1.6259766	348	9807129	356.0339355		
CSM	LEM	69	721195	535.055	114			468954	590	
LEM	CSM	69	721195	9.316	11	355.4626465	355.8471680	465097	918	-123.0491
					4.6472168	356	1328125	4.0649414		
CSM	LEM	69	721195	635.664	114			461420	207	
LEM	CSM	69	721195	717.098	11	359.6923828	355.6384277	456768	477	-123.0491
					357.1545410	356	3635254	3.9770508		
CSM	LEM	69	721195	738.605	114			453764	301	
LEM	CSM	69	721195	824.957	11	354.8254395	3.9111328	448439	039	-122.4211
					358.7475586	358	4838867	355.8361816		
LEM	CSM	69	721195	926.266	11	.3076172	.5932617	441010	078	-122.4211
					350.3430176	357	9345703	359.0441895		
CSM	LEM	69	721195	945.348	114			438209	445	
LEM	CSM	69	72120	031.117	11	359.9121094		433055	636	-121.7931
					347.7282715	15	1281738	359.6154785		
LEM	CSM	69	72120	236.715	11	357.7368164	357.8356934	417822	719	-121.1651
					343.8171387	357	7807617	1.9665527		
LEM	CSM	69	72120	318.117	11	359.7143555	.0329590	412870	078	-121.1651
					339.9060059	7.4157715	359.6484375			
LEM	CSM	69	72120	336.746	11	1.6589355	358.7915039	410543	836	-120.5381
					337.0605469	5.4272461	1.0766602			
CSM	LEM	69	72120	348.926	114			408740	285	
LEM	CSM	69	72120	412.254	11	1.7138672	355.8691406	406266	559	-120.5381
					335.3356934	4.0209961	3.9880371			

CSM	LEM	69	72120	450.437114				401327.426		1
LEM	CSM	69	72120	520.215	11	355.5834961	2.8564453	398087.199	-120.5381	2
LEM	CSM	69	72120	628.125	11	338.2470703	5.0207520	356.5173340	-119.9101	2
CSM	LEM	69	72120	655.426114		357.2424316	358.3850098	389982.879		2
CSM	LEM	69	72120	6.906	62	333.4460449	9.8327637	386380.180		1
CSM	LEM	69	72120	7.6.906	62	359.4506836	22.5467012			2
CSM	LEM	69	72120	713.957	62	93.6364746	1.0876465	1.2304688		1
LEM	CSM	69	72120	735.625	11	359.3408203	22.5851533			2
CSM	LEM	69	72120	756.324114		93.2739258	1.1755371	1.2744141		1
CSM	LEM	69	72120	813.984	62	327.3706055	359.3737793	381878.559	-119.2821	2
CSM	LEM	69	72120	842.707	11		.7690430	4724121		1
LEM	CSM	69	72120	844.977	62	359.7692871	22.8350923	379149.602		2
CSM	LEM	69	72120	858.887		90.1977539	1.3732910			1
CSM	LEM	69	72120	917.125	62	358.5498047	2.4499512	1.8237305		2
LEM	CSM	69	72120	951.086	11	325.7336426	.7800293	373999.359	-118.6541	2
CSM	LEM	69	72120	954.586	62	359.2858887	22.3791597	357.4072266		2
CSM	LEM	69	72120	963.7114		89.1979980	1.4721680	1.5820313		1
LEM	CSM	69	72120	986.758	11	358.7145996	21.8023775	370825.324		2
CSM	LEM	69	72120	1059.355	11	88.2751465	1.5930176			1
LEM	CSM	69	72120	1086.11		355.8581543	3.9550781	1.3183594		2
CSM	LEM	69	72120	1128.637114		325.2392578	1.0546875	365895.039	-118.6541	2
LEM	CSM	69	72120	1159.355	11	358.5607910	22.0330904	355.8691406		2
CSM	LEM	69	72120	1183.596	62	86.2866211	1.5930176	1.1975098		1
LEM	CSM	69	72120	1210.047	11	358.5827637	4.0759277	362379.523	-118.0261	2
CSM	LEM	69	72120	1251.125	62	319.3395996	1.4392090	357865.758		2
CSM	LEM	69	72120	1286.637114		1318359	22.8433321	355.8471680		1
LEM	CSM	69	72120	1310.047	11	82.2106934	1.0546875	1.4172363		2
CSM	LEM	69	72120	1369.125	62	357.1655273	358.9453125	354055.246	-116.7711	1
LEM	CSM	69	72120	1414.114		314.7033691	359.3627930	342482.559		2
								1.1206055		2

LEM	CSM	69	721201328.746	11	353.8366699	357.7258301	340306.398	-116.7711	2
CSM	LEM	69	721201339.566	62	317.1752930	359.6923828	2.3620605		1
LEM	CSM	69	721201347.437	11	359.7802734	22.7746675	1.1425781	-117.3991	2
CSM	LEM	69	721201434.098	62	75.1464844	1.1425781	338130.238		1
CSM	LEM	69	721201436.617114		354.5178223	356.2097168	3.8342285		2
CSM	LEM	69	721201447.098	62	315.6152344	.3845215	1.95555664		2
CSM	LEM	69	721201452.555	11	.1318359	22.5192354	332059.707		1
LEM	CSM	69	7212016	.547	72.8723145	1.6149902			2
LEM	CSM	69	7212017	8.656	359.6594238	22.7499483	1.8786621		1
LEM	CSM	69	721201816.816	11	72.0703125	1.7687988	330551.199	-116.7711	2
LEM	CSM	69	7212020	9.406	357.4951172	2.7136230	357.5280762		2
LEM	CSM	69	721202038.305	11	309.6826172	357.1325684	322596.957	-115.5151	2
LEM	CSM	69	721202145.957	11	356.8249512	358.4838867	1.5600586		2
LEM	CSM	69	721202251.035	11	307.2436523	1.9885254	314792.797	-115.5151	2
LEM	CSM	69	72120238.305	11	2.0764160	359.4177246	.9228516		2
LEM	CSM	69	721202424.887	11	298.8830566	3.8122559	306876.078	-114.8871	2
LEM	CSM	69	721202454.926	11	354.5068359	3.9770508	294006.719	-114.2601	2
LEM	CSM	69	721202513.324	11	303.3984375	357.4401855	1.3732910	-113.6321	2
LEM	CSM	69	721202532.207	11	.3186035	358.9892578	.1538086	-113.6321	2
LEM	CSM	69	721202550.605	11	292.5878906	358.3410645	283032.117	-113.0041	2
LEM	CSM	69	7212026	9.004	356.2426758	.5383301	.2636719	-112.3761	2
LEM	CSM	69	7212026	9.004	295.4223633	355.3527832	275678.199	-111.7481	2
LEM	CSM	69	7212026	9.004	358.0554199	.1867676	.4504395	-112.3761	2
LEM	CSM	69	7212026	9.004	290.5883789	359.6154785	265125.699	-111.7481	2
LEM	CSM	69	7212026	9.004	357.6928711	359.9670410	261748.898	-112.3761	2
LEM	CSM	69	7212026	9.004	288.0944824	1.6259766	.8239746	-111.7481	2
LEM	CSM	69	7212026	9.004	359.3078613	.0549316	259694.680	-112.3761	2
LEM	CSM	69	7212026	9.004	282.3706055	6.7016602	1.1315918	-111.7481	2
LEM	CSM	69	7212026	9.004	359.8681641	359.7692871	257593.559	-111.7481	2
LEM	CSM	69	7212026	9.004	280.5578613	3.4167480	1.9775391	-111.7481	2
LEM	CSM	69	7212026	9.004	359.7473145	359.5056152	255558.100	-111.7481	2
LEM	CSM	69	7212026	9.004	279.9755859	358.0883789	3.0432129	-111.7481	2
LEM	CSM	69	7212026	9.004	358.6486816	358.7255859	253513.260	-111.7481	2
LEM	CSM	69	7212026	9.004	280.2832031	357.3303223	4.1638184		2
LEM	CSM	69	7212026	9.004	356.8029785	357.6379395			2
LEM	CSM	69	7212026	9.004	281.3049316	359.8242187			2
LEM	CSM	69	7212026	9.004	354.1333008	356.2536621			2
LEM	CSM	69	7212026	9.004	283.1616211	2.8674316			2

LEM	CSM	69	721202627.586	11	350.5078125	354.3969727	251440.279	-111.1211	2
LEM	CSM	69	721202646.504	11	285.9741211	6.4270020	5.3063965	-111.1211	2
LEM	CSM	69	72120275.555	11	346.2451172	351.9360352	249329.779	-111.1211	2
LEM	CSM	69	721202724.695	11	289.4567871	10.7666016	6.4050293	-111.1211	2
LEM	CSM	69	721202743.746	11	343.1799316	354.0563965	247209.898	-111.1211	2
LEM	CSM	69	721202821.785	11	291.6979980	6.6467285	4.9768066	-111.1211	2
LEM	CSM	69	721202840.785	11	350.3430176	.4724121	245099.398	-111.1211	2
LEM	CSM	69	721202859.797	11	283.8208008	1.8237305	359.9890137	-110.4931	2
LEM	CSM	69	721202918.816	11	359.8352051	.0109863	242998.279	-110.4931	2
LEM	CSM	69	721202937.816	11	273.5266113	359.9890137	.7470703	-110.4931	2
LEM	CSM	69	721202918.816	11	.3186035	357.4731445	240906.539	-109.8651	2
LEM	CSM	69	721202937.816	11	272.2961426	358.0773926	3.2958984	-110.4931	2
LEM	CSM	69	721202918.816	11	359.7912598	354.8913574	238814.799	-109.8651	2
LEM	CSM	69	721202937.816	11	272.0874023	356.7041016	5.9216309	-110.4931	2
LEM	CSM	69	721202918.816	11	359.3188477	356.2316895	236732.439	-110.4931	2
LEM	CSM	69	721202937.816	11	271.7578125	359.0332031	4.6032715	-109.8651	2
LEM	CSM	69	721202918.816	11	358.8574219	357.6818848	234640.699	-109.8651	2
LEM	CSM	69	721202937.816	11	271.4172363	.1757813	3.1530762	-109.8651	2
LEM	CSM	69	721202918.816	11	.1428223	358.9892578	232548.959	-109.8651	2
LEM	CSM	69	721202937.816	11	269.3408203	1.0546875	1.8786621	-109.2371	2
LEM	CSM	69	721202918.816	11	358.6816406	.2526855	230466.600	-109.2371	2
LEM	CSM	69	721202937.816	11	270.0439453	.9777832	.6042480	-109.2371	2

APPENDIX D

Apollo 11 Landing Radar Data

The LM landing radar data that was used in the analysis is listed in the two card format of the HOPE orbit determination program. The first card specifies the vehicle, the time of the observation (year (mod 1900), month, day, hour, minute, and second), three code numbers, V_{XA} measurement, V_{YA} measurement, V_{ZA} measurement, and the slant range measurement (ρ). The second card specifies the inner, middle, and the outer gimbal angles. The units are feet and feet per second.

LANDING RADAR OBSERVATIONS

VEHI	YYMMDDHHMMSS	SSS	XFVX(FPS)	INNER(DEG)	VY(FPS)	MIDDLE(DEG)	VZ(FPS)	OUTER(DEG)	RANGE(FT)
LEM	69	72020	953.164125	79.1564941	.2746582	28.0261230			43753.4501
LEM	69	72020	955.164125	78.9147949	359.7583008	20.3796387			41897.5701
LEM	69	72020	957.164125	78.3654785	359.9450684	12.7001953			41144.1951
LEM	69	72020	959.164125	77.9599844	.6701660	4.9329613			39329.5501
LEM	69	72020	1.164125	77.8491211	1.2744141	1.2524414			39005.8501
LEM	69	72020	3.164125	78.0468750	1.1865234	3.0541992			39671.3601
LEM	69	72020	5.164125	78.1457520	.1867676	3.6694336			38520.3001
LEM	69	72020	7.164125	77.9919434	358.9782715	3.9221191			38444.7701
LEM	69	72020	9.164125	77.7502441	353.8793945	4.5593262			37781.1851
LEM	69	72020	1011.164125	77.2558594	359.9352051	5.2294922			37473.6691
LEM	69	72020	1013.164125	77.9040527	.9569336	4.7680664			37295.6351
LEM	69	72020	1015.164125	77.8491211	1.2634277	3.8891602			37284.8451
LEM	69	72020	1017.164125	77.2448730	.6921387	4.2187500			36594.2851
LEM	69	72020	1019.164125	77.4645996	.5603027	5.0207520			36480.9901
LEM	69	72020	1021.164125	77.3657227	1.3732910	4.9987793			36480.9901
LEM	69	72020	1025.164125	77.1780551	359.5058594	3.1201172			35666.3451
LEM	69	72020	1027.164125						35785.0351

LEM	69	720201025.164125	76.8933105	359.34.8203	2.9663086	35191.5841	2
LEM	69	720201031.164125	76.1682129	.8789063	3.186 352	34262.4901	2
LEM	69	720201033.164125	76.7724609	.7910156	3.2849121	34867.8851	2
LEM	69	720201035.164125	77.1130371	359.9890137	3.2629395	34511.9141	2
LEM	69	720201037.164125	77.0031738	359.9340820	3.1091309	34290.6201	2
LEM	69	720201039.164125	75.8056641	.5493164	3.0212402	33195.4351	2
LEM	69	720201041.164125	74.2895508	.6811523	2.9223633	33119.9151	2
LEM	69	720201043.164125	75.3222656	.2307129	2.8894043	33087.5351	2
LEM	69	720201045.164125	76.5197754	.0878906	2.9443359	32245.9151	2
LEM	69	720201047.164125	74.0039062	359.6484375	2.9553223	31733.3001	2
LEM	69	720201049.164125	73.9160156	1.4721690	2.9553223	31560.7501	2
LEM	69	720201051.164125	73.4436035	.7031250	2.8344727	30800.0551	2
LEM	69	720201053.164125	72.6635742	359.6704102	2.7355957	30514.1201	2
LEM	69	720201054.086125	72.9492188	359.8022461	1.1206055		1
LEM	69	720201055.164125	72.9931641	359.8352051	1204.332		2
LEM	69	720201057.164125	73.2128906	359.8022461	359.2968750	30449.3301	2
LEM	69	720201058.176125	72.0373535	359.6044922	359.0551758	30109.4951	2
LEM	69	720201059.164125	71.7077637	-94.0511999			1
LEM	69	7202011.246125		.1318359	359.3408203		2
LEM	69	7202011.164125	70.6640625	.5273438	359.4726562	29311.0351	2
LEM	69	7202011.164125	69.8510742	.9098789	1232.243		1
LEM	69	7202011.164125	70.1636777	1.1206055	359.6813965	28884.8301	2
LEM	69	7202011.164125			359.8352051		2

LEM	69	7202011	3.164125	71.1584473	.8349609	.1098633	27573.8451
LEM	69	7202011	5.164125	70.0277344	.3076172	.2746582	28388.4001
LEM	69	7202011	6.277125		-46.0559998		
LEM	69	7202011	7.164125	70.0048828	.2416992	.3186035	27676.3501
LEM	69	7202011	7.934125	69.0270996	359.9450684	.3735352	27325.6751
LEM	69	7202011	9.164125	68.9941406	.3076172	1176.941	27309.4901
LEM	69	7202011	11.164125	68.7963867	.5712891	.4394531	26921.0501
LEM	69	7202011	11.957125	69.5324707	.0997559	.4394531	26726.8301
LEM	69	7202011	13.164125	68.8233457	-04.7783995	.4394531	26273.6501
LEM	69	7202011	13.957125	68.3459473	359.4177246	.3186035	25836.6551
LEM	69	7202011	15.164125	67.9174805	359.4936426	1138.975	25707.1751
LEM	69	7202011	17.164125	67.9284668	.4284668	.2526855	25758.8901
LEM	69	7202011	17.957125	66.8627930	.9008789	.1318359	25307.9451
LEM	69	7202011	19.164125	66.4562988	-59.6303952	.0549316	
LEM	69	7202011	19.957125	65.9729004	.6811523	359.9340820	25836.6551
LEM	69	7202011	21.164125	65.6872559	.5383301	1136.895	25707.1751
LEM	69	7202011	23.164125	66.4233398	1.0107422	359.6374512	25758.8901
LEM	69	7202011	24.086125	67.1044922	.8239746	359.2419434	
LEM	69	7202011	25.164125	67.1923828	-81.4463987		
LEM	69	7202011	26.164125	66.7639160	.2966309	358.9892578	25307.9451
LEM	69	7202011	26.086125	67.7636719	359.5275879	358.9123535	25307.9451
LEM	69	7202011	27.164125		.2526855	1038.426	
LEM	69	7202011	27.957125		358.9782715		25361.8951

LEM	69	720201129.164125	68.4228516	.9887695	358.9672852	25016.6151	2
LFM	69	720201130.955125	67.5439453	.4064941	358.9233398		2
LEM	69	720201131.164125	66.6870117	-95.5056700	358.9892578	24628.1751	1
LEM	69	720201131.926125	66.5441895	359.7253418	358.9782715		2
LEM	69	720201133.164125	66.1816406	359.5275879	359.0332031		1
LEM	69	720201135.164125	66.7095844	359.6594238	359.0032441	24487.9051	2
LEM	69	720201135.887125	67.3022461	359.9670410	359.0112305	24547.2501	2
LEM	69	720201137.164125	66.9177246	-73.6896000	358.9782715		1
LEM	69	720201137.895125	67.2363281	.7250977	359.0112305	24282.8951	2
LEM	69	720201139.164125	66.4562988	.2526955	358.9343262		1
LEM	69	720201139.887125	66.1486816	359.9340820	359.0441895	23856.6901	2
LEM	69	720201141.164125	65.6652832	359.5635469	359.0332031		1
LEM	69	720201141.887125	65.7202148	359.9340820	359.1101074	23479.0401	2
LEM	69	720201143.164125	65.5554190	-74.4167995	359.1540527		1
LEM	69	720201143.965125	65.3796387	1.3513184	359.1870117	23203.8951	2
LEM	69	720201145.164125	64.7534180	1.3732910	359.2529297		1
LEM	69	720201145.875125	63.5888672	.6591797	359.1979980	22723.7401	2
LEM	69	720201147.164125	63.2702637	.5053711	359.2309570		1
LEM	69	720201147.895125	62.9077148	.1423223	359.1760254	22393.8551	2
LEM	69	720201149.164125	63.5559082	-89.2343988	359.1760254		1
LEM	69	720201149.887125	64.0293203	1.1755371	359.1650391	22383.8551	2

LEM	69	720201149	926125	64.3469238	1.4282227	359.0991211	920.542	22254.3751	1
LEM	69	720201151	164125	63.509766	.6811523	359.0771484			2
LEM	69	720201151	955125	891.4247894					2
LEM	69	720201153	164125	63.3471680	.4064941	359.0441895		21542.2351	1
LEM	69	720201153	897125	62.5451660	359.7473145	359.2529207			2
LEM	69	720201153	897125	-93.3239994					1
LEM	69	720201155	164125	62.4572754	359.5935059	359.3627930		21202.3501	2
LEM	69	720201155	887125	62.1057129	359.7692871	359.5715332			2
LEM	69	720201157	164125	62.0617676	359.9121094	359.7143555	911.189		1
LEM	69	720201157	895125	62.0577812	.6262207	359.9011230		27921.81	2
LEM	69	720201157	895125	845.0567856					1
LEM	69	720201159	164125	61.9519043	.8459473	.0549316		20835.4901	2
LEM	69	720201159	895125	61.9628906	1.5930176	.1757813			2
LEM	69	720201159	895125	-40.7231994					1
LEM	69	7202012	1.164125	61.2817383	1.2854004	.2856445		20376.9151	2
LEM	69	7202012	1.887125	60.8312989	1.3942773	.3845215			2
LEM	69	7202012	3.164125	60.2050781	.9997559	.4284668	905.633		1
LEM	69	7202012	3.895125	60.2929687	.9997559	.5383311		20139.5351	2
LEM	69	7202012	5.164125	59.9633789	.6701660	.5383301			1
LEM	69	7202012	5.887125	60.5346680	.6152344	.6262207		19923.7351	2
LEM	69	7202012	5.887125	60.5017090	-62.5391994	.6262207			1
LEM	69	7202012	7.164125	61.2268066	.4614258	.6262207			2
LEM	69	7202012	7.887125	61.2507656	.3735352	.6481934		19643.1951	2
LEM	69	7202012	9.164125	61.4355469	.4724121	.846.517			1
LEM	69	7202012	9.897125	61.4355469	.4284668	.6591797		10448.0751	2
LEM	69	7202012	9.897125	787.2255859		.6152344			1

LEM	69	720201211.164125	61.3476562	.8230746	.6481934	19058.3071	2
LEM	69	720201211.895125	60.9092031	.8789063	.5932617		2
LEM	69	720201213.164125	60.7653809	1.4941406	.5822754	18774.6071	2
LFM	69	720201213.895125	59.8205566	1.7468262	.5493164		2
LFM	69	720201215.164125	59.7436523	2.3510742	835.422		1
LEM	69	720201215.957125-734.4175873	58.9196777	1.3073730	.4943848	18375.2351	2
LEM	69	720201217.164125	59.1833496	1.0217285	.2856445		2
LEM	69	720201217.937125	58.7878418	.0549316	.1538086	17981.5351	2
LEM	69	720201219.164125	59.6118164	-.68.5991993	.0219727		1
LEM	69	720201219.895125	59.9743652	.8020020	359.8681641	17857.4501	2
LEM	69	720201221.164125	60.6445313	1.2634277	783.587		1
LEM	69	720201221.895125-713.8095856	59.6777344	1.0107422	359.4946289	17522.9601	2
LEM	69	720201223.164125	59.4140625	.5822754	359.4067383		1
LEM	69	720201223.895125	58.5681152	.3625488	359.3078613	17103.8651	2
LEM	69	720201225.164125	58.4033203	359.9670410	359.3847656		1
LEM	69	720201225.164125	58.2495117	.3405762	359.4396973	16859.3751	2
LEM	69	720201225.926125	58.2385254	.2197266	766.945		1
LEM	69	720201227.164125	58.6560059	1.0766602	359.5385742	16600.4151	2
LEM	69	720201227.895125-679.8063812	58.7219238	1.1425781	359.5385742		1
LEM	69	720201229.164125	59.1833496	1.8896484	359.5605469	16422.3901	2
LEM	69	720201229.895125	58.4912109	-.48.9647999	359.5495605		1
LEM	69	720201229.895125	58.4912109	1.4611816	359.5495605		2

LEM	69	720201231.164125	58.1945801	1.1755371	359.5166016	16023.1501
LEM	69	720201231.895125	57.4914551	.9118652	359.4726562	2
LEM	69	720201233.164125	57.5463967	.7800293	359.3957520	1
LEM	69	720201233.887125	57.0849609	.9228516	359.4506836	2
LEM	69	720201235.164125	57.5024414	1.0327148	359.5166016	1
LEM	69	720201235.887125	57.2827148	-46.5407996		2
LEM	69	720201237.164125	57.7990723	1.4501953	359.5825195	15122.1851
LEM	69	720201237.895125	57.8979492	1.5161133	359.6264648	2
LEM	69	720201239.164125	58.6120605	1.9445801	359.6264649	14825.4601
LEM	69	720201239.937125	628.5439911	1.6040039	359.6594238	2
LEM	69	720201241.164125	58.7438965	1.6149902	359.6154785	14474.7851
LEM	69	720201242.055125	58.5351562	.4064941	359.6154785	2
LEM	69	720201243.164125	57.9748535	-51.8736000		1
LEM	69	720201243.895125	56.9860840	.6591797	359.5495605	14145.6901
LEM	69	720201245.164125	56.8762207	.5603027	359.5605469	2
LEM	69	720201245.957125	56.6674805	.9887695	359.5385742	13916.5951
LEM	69	720201247.164125	57.9199219	.8459473	359.4616699	2
LEM	69	720201247.895125	58.3593750	.7141113	359.4177246	13563.0301
LEM	69	720201249.164125	56.9531250	.2746582	359.4067383	2
LEM	69	720201249.895125	56.5026855	-56.4791994		1
LEM	69	720201251.164125	56.5026855	.3295898	359.5275879	13190.7751
LEM	69	720201251.164125	56.5026855	1.2963867	359.7253418	2
LEM	69	720201251.164125	56.5026855	1.5930176	359.8681641	12910.2351
LEM	69	720201251.164125	56.5026855	1.5930176	359.8681641	2

LEM	69	720201251	895125-549	55.6457520	1.7248535	.0000000	2
LEM	69	720201251	895125-549	9455948	1.0107422	.0769043	1
LEM	69	720201253	164125	55.9643555			2
LEM	69	720201253	937125	56.0961914	.5053711	.1428223	2
LEM	69	720201255	164125	56.6674905	-48.9647999		1
LEM	69	720201255	887125	56.7553711	.0988770	.2087402	2
LEM	69	720201257	164125	56.6784668	1.4062500	.3186035	2
LEM	69	720201257	895125-526	56.4477539	1.7358398	599.306	1
LEM	69	720201259	895125	56.927916	2.6916504	.4943848	2
LEM	69	720201259	164125	56.3598633	1.7578125	.5053711	1
LEM	69	720201259	895125	56.5686035	.7360840	.5163574	2
LEM	69	720201259	895125	56.5026855	-45.3287997		1
LEM	69	7202013	1.164125	56.9750977	359.9450684	.5053711	2
LEM	69	7202013	1.887125	56.7224121	.6701660	.5383301	2
LEM	69	7202013	3.164125	56.8212891	1.1096191	555.272	1
LEM	69	7202013	3.895125-501	8047905	2.0324707	.5712991	2
LEM	69	7202013	5.164125	56.4697266	2.2631836	.5932617	2
LEM	69	7202013	5.895125	57.1179199	1.7138672	.6042480	1
LEM	69	7202013	7.164125	57.0959473	-25.4519999	.5493164	2
LEM	69	7202013	7.895125	57.3815918	1.3073730	.5053711	1
LEM	69	7202013	9.164125	56.7663574	.4174805	.4284668	2
LEM	69	7202013	9.926125-469	6047897	.5603027	513.666	1
LEM	69	7202013	9.164125	56.2280273	.1977539	.3845215	2
LEM	69	7202013	1.164125	55.7446289	.4064941		2
LEM	69	7202013	1.164125	55.6677246	.0109863	.2526855	2
LEM	69	7202013	1.164125			.1647949	2

LEM	69	720201311.895125	55.7446289	-44.8439999	.0769043	9743.3701	1
LEM	69	720201313.164125	56.1071777	359.9450684	359.9670410		2
LEM	69	720201314.055125	56.5466309	.2307129	359.8242187	473.793	2
LEM	69	720201315.164125	56.9421387	.1757813	359.7253418	9473.6201	2
LEM	69	720201315.926125	54.4063911				1
LEM	69	720201317.164125	57.1728516	.7690430	359.6154785	9171.5001	2
LEM	69	720201317.957125	56.6125488	1.9116211	359.7253418		2
LEM	69	720201319.164125	56.6125488	-12.8471999			1
LEM	69	720201319.895125	55.8984375	2.5598145	.0878906	8928.7251	2
LEM	69	720201319.895125	56.1730957	2.0104980	.1977539		2
LEM	69	720201321.164125	55.9753418	1.5490723	.2966309		2
LEM	69	720201321.895125	418.7287903	1.0437012	.4174805	8610.4201	2
LEM	69	720201323.164125	56.2390137				1
LEM	69	720201323.164125	56.0522461	1.0986328	.4833984	8351.4601	2
LEM	69	720201323.895125	56.1950684	-26.6639998	.5932617		1
LEM	69	720201325.164125	55.9863281	1.1315918	.6701660	8022.3651	2
LEM	69	720201325.895125	55.8544922	.9777832	.7250977		1
LEM	69	720201327.164125	55.6347656	1.1315918	.8239746	7795.7751	2
LEM	69	720201327.895125	386.7863922				2
LEM	69	720201329.164125	55.2502441	1.0656738	.8349609	7504.4451	2
LEM	69	720201329.895125	55.0964355	.9448242	.9118652		2
LEM	69	720201329.895125	54.6240234	-15.9287999	.9118652	7250.8801	1
LEM	69	720201331.164125	55.0305176	.5493164	.9448242		2
LEM	69	720201341.164125				6538.7401	2

LEM	69	720201341.754125	46.3623047	1.7028809	.8898926	2
LEM	69	720201343.164125	46.3952637	2.2302246	217.567	1
LEM	69	720201343.895125	45.3955078	2.0544434	.8569336	2
LEM	69	720201345.164125	45.7660430	1.5270996	.7910156	2
LEM	69	720201345.937125	45.3186035	.7360840	.5383301	1
LEM	69	720201347.164125	45.6481934	-25.9368000	.4614258	2
LEM	69	720201347.965125	44.3957520	1.1975098	.3186035	2
LEM	69	720201349.164125	44.0112305	1.5930176	206.992	1
LEM	69	720201349.937125	43.3959961	2.4389648	.2746582	2
LEM	69	720201351.164125	43.4399414	2.0874023	.0219727	2
LEM	69	720201351.926125	44.1101074	1.8566895	359.7912598	2
LEM	69	720201353.164125	43.9782715	-18.6647999		1
LEM	69	720201353.887125	44.3078613	1.1096191	359.7692871	2
LEM	69	720201355.164125	43.3959961	1.1865234	359.9560547	2
LEM	69	720201355.164125	42.6269531	1.7797852	.0988770	2
LEM	69	720201355.957125	42.6263962	3.2629395	.2526855	1
LEM	69	720201357.164125	42.4291992	3.3619164	.3076172	2
LEM	69	720201357.957125	43.2531738	-15.5136000		1
LEM	69	720201359.164125	43.0773926	1.9676758	.2636719	2
LEM	69	720201359.945125	42.7697754	359.7692871	.2856445	2
LEM	69	720201359.945125	42.7697754	359.3518066	173.013	1
LEM	69	7202014	39.7595215	359.8352051	.2856445	2
LEM	69	7202014	1.164125		4094.0151	2

LEM	69	7202014	2.27125-259.2743950	39.6057123	359.5495605	.4594395	3857.0301	1
LEM	69	7202014	3.164125	38.2245605	359.7253418	.4394531		2
LEM	69	7202014	4.026125	39.0563965	-21.0888000	.5603027		1
LEM	69	7202014	5.164125	38.5621117	359.3298340	.5163574	3690.1801	2
LEM	69	7202014	6.137125	38.2993398	359.4177246	172.493		1
LEM	69	7202014	7.164125	37.8369141	359.3408203	.5493164		2
LEM	69	7202014	7.984125-226.0430968	36.7712402	359.9670410	.6262207	3544.5151	2
LEM	69	7202014	9.164125	36.1018742	.3845215	.5603027		1
LEM	69	7202014	10.145125	35.1123047	-19.6343999	.6921387	3269.3701	2
LEM	69	7202014	11.164125	35.1123047	.9129883	.6811523		1
LEM	69	7202014	12.027125	35.4090355	.7690430	.6481934	3145.2851	2
LEM	69	7202014	13.164125	35.8044434	.3295898	164.519		1
LEM	69	7202014	13.164125	36.0900979	.3295898	.6921387	3048.1751	2
LEM	69	7202014	14.164125-202.4735985	36.8811035	.0549316	.5712891		1
LEM	69	7202014	15.164125	36.2548828	.6042480	.5712891	2864.7451	2
LEM	69	7202014	16.047125	36.0131836	-14.7864000	.5053711		1
LEM	69	7202014	17.164125	35.4968262	1.3293457	.6701660	2719.0801	2
LEM	69	7202014	18.055125	34.7937012	1.6699219	150.823		1
LEM	69	7202014	19.164125	34.7937012	2.1423340	.7250977		2
LEM	69	7202014	20.066125-173.6223965	34.5300293	2.0764160	.7580566	2594.9951	2
LEM	69	7202014	21.875125	33.7689863	1.9017578	.8459473		1
LEM	69	7202014	21.875125	33.3084375	-12.1199999	.9008789		2
LEM	69	7202014	23.164125	33.3084375	.5493164		2316.6131	2

LEM	69	720201423	.984125	32.5085449	359.3188477	1.5051270	2
LEM	69	720201425	.164125	32.5085449	358.5058594	140.248	1
LEM	69	720201426	.164125-1	32.2998047	358.6706543	1.8566805	2
LEM	69	720201427	.164125	32.4865723	359.5385742	2177.4221	2
LEM	69	720201428	.027125	32.8820801	359.9230957	2.5158691	2
LEM	69	720201429	.164125	-9.9383998		2085.7071	1
LEM	69	720201430	.176125	32.4316406	.1208496	3.0102539	2
LEM	69	720201431	.164125	32.1020508	.1428223	3.3618164	2
LEM	69	720201432	.277125-1	30.9045410	359.4726562	1960.5431	2
LEM	69	720201433	.164125	29.4104004	359.2529297	3.4616934	2
LEM	69	720201435	.957125-1	125.5799980		128.806	1
LEM	69	720201437	.176125	28.9940430	359.3298340	3.2189941	2
LEM	69	720201438	.176125	28.0371094	359.4836426	1853.7221	2
LEM	69	720201439	.176125	-10.1807998		1749.0591	2
LEM	69	720201440	.055125	27.7954102	.0439453	2.5598145	1
LEM	69	720201441	.176125	27.8723145	.3295898	2.4609375	2
LEM	69	720201442	.156125	27.4323613	.4174805	1664.8971	2
LEM	69	720201443	.176125	27.6416016	.7470703	1573.1821	2
LEM	69	720201444	.055125	108.5783987		2.1862703	1
LEM	69	720201444	.176125	27.7075195	.5932617	122.739	1
LEM	69	720201447	.055125	27.4548340	.6921387	1.9775391	2
LEM	69	720201441	.176125	-11.3927999		1.7578125	1
LEM	69	720201442	.156125	27.9272461	.8020020	1.5380859	2
LEM	69	720201444	.387125	28.8171387	359.9890137	1469.5991	2
LEM	69	720201444	.387125	30.7177734	.2307129	1.1645508	1
LEM	69	720201444	.387125	-93.2511078		1.2084961	2
LEM	69	720201444	.387125	27.0153909	.2307129	1.1975098	2
LEM	69	720201444	.387125			101.589	1
LEM	69	720201444	.387125			1.1975098	2
LEM	69	720201444	.387125			1401.6211	1
LEM	69	720201444	.387125			1.3403320	2

LEM	69	720201445	207125	25.403906	.2956445	1.3623047	1229.0911
LEM	69	720201446	437125	-10.9079999	.1099633	1.5397859	2
LEM	69	720201447	207125	24.5214844	.3955078	1.4831543	1162.0831
LEM	69	720201448	047125	24.6972656	.8459473	96.562	2
LEM	69	720201449	207125	25.1477051	1.0217285	1.4721680	1093.0271
LEM	69	720201450	066125	25.8947754	1.1535645	1.6479492	2
LEM	69	720201451	215125	-81.4015989	.8459473	1.5930176	1
LEM	69	720201452	098125	26.2353516	1.0217285	1.5161133	2
LEM	69	720201452	098125	26.0925293	-.9383998	1.6259766	978.6531
LEM	69	720201453	215125	25.5981445	.4833984	1.5380859	2
LEM	69	720201453	977125	24.7302246	.7250977	85.120	1
LEM	69	720201455	215125	24.2248535	.9777832	1.4611816	2
LEM	69	720201455	215125	23.1811523	.9997559	1.5710449	930.0981
LEM	69	720201453	977125	24.2248535	.9777832	85.120	2
LEM	69	720201457	215125	21.8078613	1.2084961	1.3952637	590.2131
LEM	69	720201457	824125	-60.5359993	1.0217285	1.4282227	801.6971
LEM	69	720201459	215125	21.4892578	.8789063	1.3183594	2
LEM	69	720201459	977125	21.0270320	-6.0599999	1.1975098	1
LEM	69	7202015	1.215125	20.8630371	.5712891	1.1865234	758.5371
LEM	69	7202015	1.977125	20.2038574	.1209496	74.545	2
LEM	69	7202015	3.215125	19.8083496	.2197266	1.1206055	2
LEM	69	7202015	3.215125	19.3139648	.2197266	.9229516	705.6661
LEM	69	7202015	3.977125	-48.0423994	.0878906	1.0986329	2
LEM	69	7202015	5.215125	18.9624023			662.5161

LEM	69	7202015	5.977125	18.6987305	.0549316	1.3293457	2
LEM	69	7202015	7.215125	18.5668945	-7.7568000	1.3952637	1
LEM	69	7202015	7.977125	18.1054687	359.9121094	625.8201	2
LEM	69	7202015	9.215125	17.9626465	359.6484375	581.5811	2
LEM	69	7202015	9.906125	17.6110840	359.8022461	66.570	1
LEM	69	7202015	11.215125	17.3144531	359.8461914	1.8237305	2
LEM	69	7202015	11.945125	17.1606445	.3515625	1.9555664	2
LEM	69	7202015	13.215125	17.0507812	-5.8176000	2.0654297	1
LEM	69	7202015	14.016125	16.6223145	.5603027	2.3291016	2
LEM	69	7202015	15.215125	16.5344238	.7250977	2.3840332	2
LEM	69	7202015	15.984125	15.4687500	.9667969	2.5598145	1
LEM	69	7202015	17.215125	12.0739746	.9997559	58.596	2
LEM	69	7202015	18.004125	9.3383789	.6811523	2.8125000	1
LEM	69	7202015	19.215125	6.3061523	.7031250	461.8121	2
LEM	69	7202015	19.937125	4.9768066	-4.3631999	3.2080078	1
LEM	69	7202015	21.215125	5.1106289	.4394531	3.3398438	2
LEM	69	7202015	21.937125	5.1086426	.1647949	3.5156250	2
LEM	69	7202015	21.937125	15.3271998	.2087402	61.369	1
LEM	69	7202015	23.215125	5.5590820	.0219727	3.6804199	2
LEM	69	7202015	24.277125	5.0436035	.4504395	3.9990234	2
LEM	69	7202015	25.215125	5.4931641	-3.6360000	4.2187500	1
LEM	69	7202015	25.215125	5.7238770	.3955078	4.4165039	2
LEM	69	7202015	25.215125		.3295899	4.5493309	2

LEM	69	720201526.437125	5.7348633	.4294668	57.729	367.9391	1
LEM	69	720201527.215125	5.5371794	.3845215	4.7460938		2
LEM	69	720201528.445125	-12.1071998		4.8669434		2
LEM	69	720201529.215125	5.9545858	.2956445	5.0207520		1
LEM	69	720201530.437125	6.0424905	.3405762	5.1196289	364.7021	2
LEM	69	720201531.215125	5.7897940	-1.9392000	5.2075195		1
LEM	69	720201532.437125	6.1093984	.2087402	5.1745605	343.1221	2
LEM	69	720201533.215125	6.3391113	.1867676	55.822		1
LEM	69	720201534.437125	6.1523438	.1098633	5.0537109		2
LEM	69	720201535.215125	-9.2735999		5.0097656	335.5691	2
LEM	69	720201536.437125	6.4919336	359.8791504	4.9108887		1
LEM	69	720201537.215125	6.8334961	359.7912598	4.8229980	332.3321	2
LEM	69	720201538.437125	6.8334961	-3.8784000	4.6911621		1
LEM	69	720201539.215125	7.0642090	359.2639160	4.6362305	321.5421	2
LEM	69	720201540.496125	7.8222656	358.8903809	51.661		1
LEM	69	720201541.215125	7.6464844	358.6816406	4.5043945		2
LEM	69	720201542.445125	-9.1447998		4.3835449	321.5421	2
LEM	69	720201543.215125	7.2619629	358.5278320	4.2626953		1
LEM	69	720201544.477125	7.2839355	358.3520508	4.1967773	316.1471	2
LEM	69	720201545.215125	6.9114004	-3.6360000	4.0100098		1
LEM	69	720201546.477125	8.3935547	358.6047363	3.9660645	312.9101	2
LEM	69	720201547.215125	11.0632324	358.7585449	47.154		1
LEM	69	720201548.215125	11.3488770	358.8244629	3.9660645		2
LEM	69	720201549.477125	-11.9783998		3.8891602	302.1201	2
LEM	69	720201550.477125					1

LEM	69	720201547.215125	11.7333984	359.0112305	3.7573242	290.2511	2
LEM	69	720201548.477125	12.3266602	350.0332031	3.7243652		2
LEM	69	720201549.215125		-5.8176000			1
LEM	69	720201550.437125	13.6669922	359.2419434	3.6254883	290.2511	2
LEM	69	720201551.215125	14.8425293	359.4726562	3.5705566		2
LEM	69	720201552.445125	14.9344199	359.5605469	41.433		1
LEM	69	720201553.215125	15.1611328	359.6813965	3.4387207	282.6981	2
LEM	69	720201554.426125	-11.3343998				1
LEM	69	720201555.215125	15.3588867	359.9780273	3.2958984	268.6711	2
LEM	69	720201556.437125	15.3608730	.0659180	3.2080078		2
LEM	69	720201557.215125	15.7104492	-4.3631999	3.1091309		1
LEM	69	720201558.437125	15.7983398	.3515625	2.9882813	267.5921	2
LEM	69	720201559.215125	15.6225586	.3735352	30.858		1
LEM	69	7202016.438125	15.6005859	.3625488	2.8125000	260.0391	2
LEM	69	7202016.437125	-9.1447998		2.7136230		2
LEM	69	7202016.437125	15.6555176	.4394531	2.5488281		1
LEM	69	7202016.437125	15.5236816	.5273438	2.4060059	255.7231	2
LEM	69	7202016.437125	15.4138184	-4.8480000	2.2192383		1
LEM	69	7202016.437125	15.4907227	.5822754	2.1093750	256.8021	2
LEM	69	7202016.437125	15.4907227	.6591797	22.537		1
LEM	69	7202016.437125	15.4907227	.8129893	1.9006349		2
LEM	69	7202016.437125	15.4687500	.8789063	1.7468262	254.6441	2
LEM	69	7202016.437125	-7.2127999				1
LEM	69	7202016.437125	15.6774002	1.0107422	1.5380859		2
LEM	69	7202016.437125	15.7653809	1.1425781	1.3952637	246.0121	2
LEM	69	7202016.437125	15.7653809	-4.3631999			1
LEM	69	7202016.437125	15.7653809	1.3623047	1.1865234		2

LEM	69	7202016	7.215125	15.4797363	1.4062500	1.1755371	244.9331
LEM	69	7202016	8.496125	10.5468750	1.2414551	15.429	1
LEM	69	7202016	9.215125	9.3493652	1.2414551	1.2524414	2
LEM	69	720201626	1.95125	5.6689453	1.0546875	1.2963867	241.6961
LEM	69	720201627	2.15125	5.5480957	.9558105	8.321	2
LEM	69	720201628	4.84125	-6.1823999	.5932617	2.2412109	169.4031
LEM	69	720201629	2.15125	.9008789	.6042480	2.2521973	2
LEM	69	720201630	4.77125	.9777832	-.7272000	2.3291016	167.2451
LEM	69	720201631	2.15125	359.2749023	.5932617	2.4499512	2
LEM	69	720201632	5.16125	359.2309570	.8459473	2.4829102	152.1391
LEM	69	720201633	2.15125	359.7583008	2.1203613	8.668	1
LEM	69	720201634	4.77125	359.4726562	2.9003906	2.6916504	2
LEM	69	720201635	2.15125	-4.6367999	2.9333496	2.7905273	143.5071
LEM	69	720201636	4.77125	358.6047363	2.9333496	2.9223633	2
LEM	69	720201637	2.15125	359.7583008	3.1311035	2.9333496	130.5591
LEM	69	720201638	4.45125	.0329590	.7272000	3.0541992	1
LEM	69	720201639	2.15125	359.7143555	3.1970215	3.1091309	2
LEM	69	720201640	4.45125	359.5605469	2.9882813	6.761	1
LEM	69	720201641	2.15125	.0219727	1.9665527	3.1420899	2
LEM	69	720201642	4.84125	-3.7352000	.8899926	3.1530762	124.0951
LEM	69	720201643	2.15125	359.7912598	.6042480	3.2739258	2
LEM	69	720201644	2.15125	.4614258	.4943848	3.3178711	117.6111
LEM	69	720201645	4.84125	3.0900234	.0000000	3.4606934	2
LEM	69	720201646	2.15125	3.0900234	.8020020	3.4606934	111.1371

LEM	69	720201644	477125	3.3178711	.7690430	3.5705566	2
LEM	69	720201645	215125	2.6477051	.5163574	6.634	1
LEM	69	720201646	484125	2.6656777	.3845215	3.7133789	2
LEM	69	720201647	215125	-3.9927999	.1538086	3.8122559	98.1891
LEM	69	720201648	484125	3.0322266	359.5495605	3.9770508	2
LEM	69	720201649	215125	2.9663086	2.6664000	4.0530551	84.1621
LEM	69	720201650	477125	3.4497070	358.2861328	4.1748047	2
LEM	69	720201651	215125	2.9223633	356.7810059	4.1748047	80.9251
LEM	69	720201652	445125	3.0871582	356.7810059	7.281	2
LEM	69	720201653	215125	5.3173828	356.8798828	4.6582031	78.7671
LEM	69	720201654	316125	-1.2880000	356.7910922	4.8339844	2
LEM	69	720201655	215125	5.0756836	356.8353375	4.9548340	78.7671
LEM	69	720201656	387125	4.7680664	.0000000	5.1525879	2
LEM	69	720201657	215125	4.6142578	356.7150979	5.2954102	76.6091
LEM	69	720201658	387125	4.4934082	356.4843750	5.548	2
LEM	69	720201659	215125	4.1748047	356.3635254	5.4162598	1
LEM	69	720201660	215125	4.5373535	356.4074707	5.5920410	2
LEM	69	720201661	215125	6.0534668	2.0983887	8.0310059	55.0291
LEM	69	720201662	215125	6.0314941	2.2631836	.000	1
LEM	69	720201663	215125	4.1528320	1.8017578	8.6132813	2
LEM	69	720201664	215125	-1.5456000	1.5490723	8.8330078	57.1871
LEM	69	720201665	215125	1.4831543	1.5490723	8.8330078	2
LEM	69	720201666	215125	354.9902344	.3405762	9.2504883	56.1081
LEM	69	720201667	215125	355.5944824	-1.9392000	9.6130371	2
LEM	69	720201668	215125		.5493164		1
LEM	69	720201669	215125				2

LEM	69	720201713.215125	355.7502773	1.5490723	9.9865723	45.3181
LEM	69	720201714.437125	356.1437988	2.6257324	10.3710937	2
LEM	69	720201715.215125	356.8469238	2.9003906	10.6567383	1
LEM	69	720201716.477125	-2.4472000	3.3068848	11.0632324	2
LEM	69	720201717.215125	357.4291992	3.5375977	11.2719727	42.0811
LEM	69	720201718.484125	358.2971191	3.7243652	11.6125488	2
LEM	69	720201719.215125	358.8354492	2.9663086	11.8103027	34.5281
LEM	69	720201720.324125	358.7915039	1.1535645	12.0849609	2
LEM	69	720201721.215125	358.6926270	1.1315918	12.3706055	28.0541
LEM	69	720201722.324125	-1.4168000	1.4721680	12.7770996	2
LEM	69	720201723.215125	1.1535645	1.7138672	13.1066895	1
LEM	69	720201724.395125	2.1093750	11.8775998	13.4912109	2
LEM	69	720201725.215125	2.6586914	358.8024902	13.7988281	22.6591
LEM	69	720201726.324125	3.1420898	357.2863770	14.1613770	2
LEM	69	720201727.215125	3.0651855	357.3413086	14.2163086	22.6591
LEM	69	720201728.316125	.9016000	2.9333496	14.2712402	1
LEM	69	720201729.215125	2.9333496	357.0886230	14.5019531	2
LEM	69	720201730.336125	3.5815430	355.5285645	14.3481445	11.8691
LEM	69	720201731.215125	4.6472168	-7.2720000	14.5788574	2
LEM	69	720201732.324125	4.8999023	355.9240723	14.7216797	24.8171
LEM	69	720201733.215125	5.3833008	356.0339355	16.296	1
LEM	69	720201733.215125	5.3833008	356.1437988	14.7216797	2
LEM	69	720201733.215125				29.1331

LEM	69	720201734	316125	5.5151367	356.2866211	14.3205566	2
				7.5991399			1
LEM	69	720201735	215125	5.2075195	356.4514160	14.8205566	2
							21.5801
LEM	69	720201736	324125	4.7680664	356.5502930	14.8095703	2
				-9.6960000			1
LEM	69	720201737	215125	4.3615723	358.1323242	14.9084473	2
							18.3431
LEM	69	720201738	348125	3.5046387	359.9230957	15.0952148	2
						1.040	1
LEM	69	720201739	215125	1.5710449	.4504395	15.6225586	2
				.3515625	2.6037598	15.6555176	2
LEM	69	720201740	387125	.5152000			1
				4.5483398	.5053711	14.0185547	2
							10.7901