

LM

ORIENTATION

COURSE NO. 30005-012

For Training Purposes Only

DATE: April 12, 1966  
REVISED: November 1966

Prepared by:

Product Support Department

## LESSON PLAN

### I. Introduction

#### A. Objective

To provide the student with an orientation of the LM spacecraft. Included will be a brief functional description of each LM operational subsystems to a block diagram level with emphasis on the major signal flow paths.

#### B. Motivation

In order to provide adequate operational support to the Apollo Mission one must have a basic functional knowledge of LM operational subsystems and the associated subsystem interfaces.

### II. Presentation

#### A. Film - Apollo Lunar Mission Profile

#### B. LM Basic Design Concept

##### 1. Ascent Stage

- a. Cabin area
- b. Aft equipment bay

##### 2. Descent Stage

##### 3. Summary

#### C. Crew Station

##### 1. Controls and Display Panels

#### D. Propulsion Subsystem

##### 1. General Configuration

- a. Descent
- b. Ascent
- c. Reaction Control

2. Descent Propulsion
  - a. Propellant Pressurization
  - b. Propellant Feed
  - c. Engine Control
3. Ascent Propulsion
  - a. Propellant Pressurization
  - b. Propellant Feed
  - c. Thrust Control
  - d. RCS Crossfeed
4. Reaction Control
  - a. Cluster Configuration
  - b. Attitude and Translational Control
  - c. Propellant Pressurization
  - d. Propellant Feed
5. Controls and Displays
6. Summary

E. Environmental Control Subsystem

1. Atmosphere Revitalization Section
  - a. Pressure Schedules
  - b. Closed Suit Loop Operation
  - c. Open Suit Loop Operation
2. Oxygen Supply and Cabin Pressure Control Section
  - a. Tankage
  - b. Descent Feed
  - c. Ascent Feed
3. Water Management Section
  - a. Tankage
  - b. Descent Feed
  - c. Ascent Feed

4. Heat Transport Section
    - a. Primary Loop Operation
    - b. Secondary Loop Operation
  5. Controls and Displays
  6. Summary
- F. Electrical Power Subsystem
1. General Configuration
    - a. Ascent Stage
    - b. Descent Stage
  2. System Functional Description
    - a. Batteries
    - b. Electrical Control Assembly (ECA)
    - c. Relay Junction Box (RJB)
    - d. Deadface Relay Box (DFRB)
    - e. Inverters
    - f. Electroexplosive Devices
  3. Controls and Displays
  4. Summary
- G. Guidance, Navigation, and Control Subsystem *AI Mast*
1. System Concept
  2. Functional Description
    - a. Hand Controllers
    - b. Primary Guidance and Navigation Section (PGNS)
    - c. Abort Guidance Section (AGS)
    - d. Control Electronics Section (CES)
  3. Controls and Displays
  4. Summary

H. Instrumentation

1. System Concept
2. Functional Description
  - a. Signal Conditioning Electronics Assembly (SCEA)
  - b. Caution and Warning Electronics Assembly (CWEA)
  - c. Pulse Code Modulation Timing Electronics Assembly (PCMTEA)
  - d. Data Storage Electronics Assembly (DSEA)
3. Controls and Displays

I. Communications

1. Operational Capabilities
  - a. IN FLIGHT
  - b. LUNAR STAY
2. System Concept
3. Functional Description
  - a. SIGNAL PROCESSOR ASSEMBLY
  - b. S-Band
  - c. VHF
4. Controls and Displays

J. Final Summary

## LM ORIENTATION

### Abbreviation Document

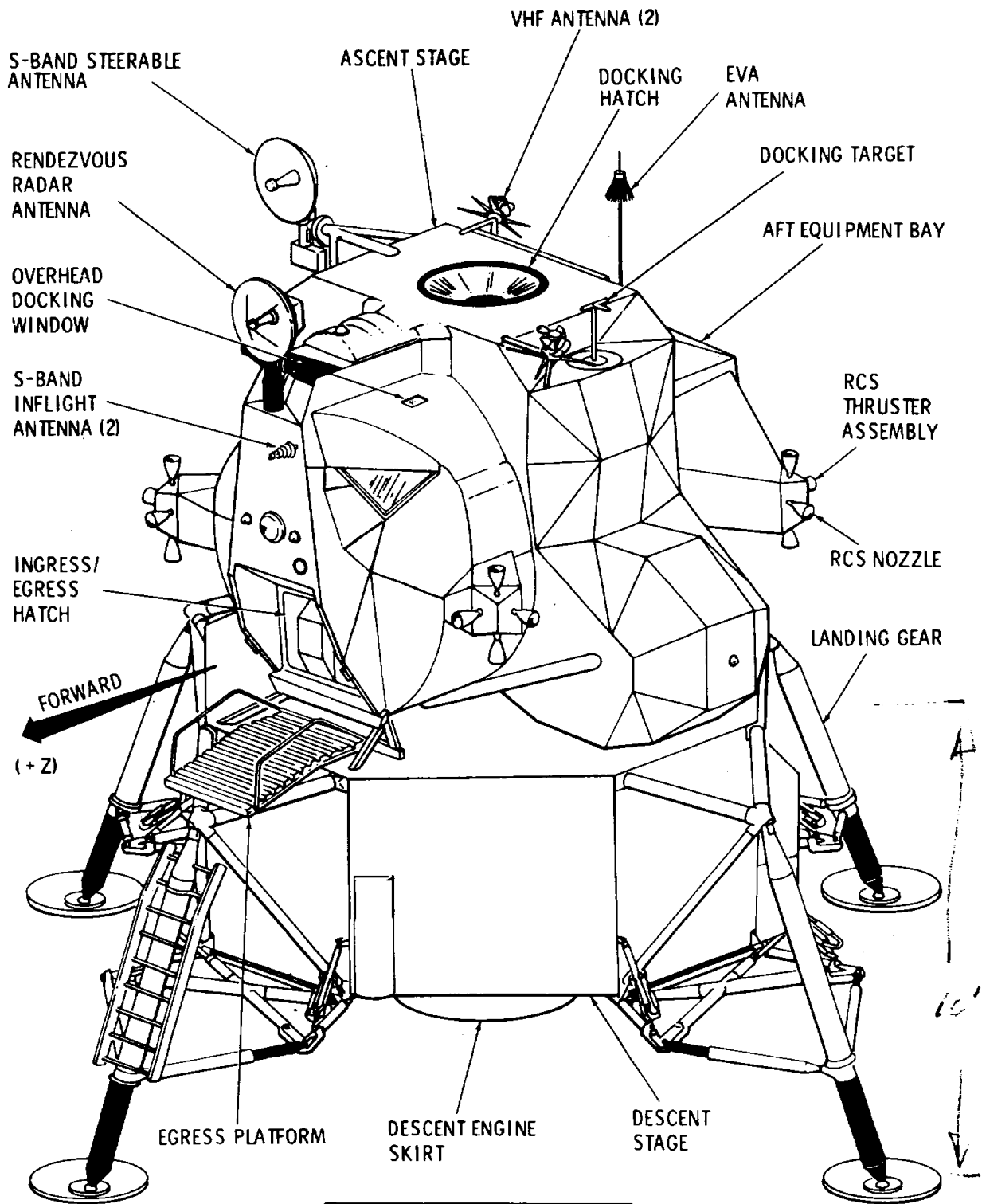
AEA	Abort Electronics Assembly
AGS	Abort Guidance Section
AOT	Alignment Optical Telescope
ARS	Atmospheric Revitalization Section
ASA	Abort Sensor Assembly
ATCA	Attitude and Translation Control Assembly
CES	Control Electronic Section
CM	Command Module
CSM	Command and Service Module
C/W	Caution/Warning
CWEA	Caution and Warning Electronics Assembly
DECA	Descent Engine Control Assembly
DEDA	Data Entry and Display Assembly
DFRB	Deadface Relay Box
DSEA	Data Storage Electronics Assembly
DSKY	Display and Keyboard
ECA	Electric Control Assembly
ECS	Environmental Control Subsystem
EPS	Electrical Power Subsystem
EVA	Extravehicular Astronaut
GN&C	Guidance, Navigation and Control Subsystem

GOX	Gaseous Oxygen
He	Helium
H <sub>2</sub> O	Water
HTS	Heat Transport Section
IMU	Inertial Measuring Unit
LM	Lunar Module
LGC	LM Guidance Computer
LR	Landing Radar
LUT	Launch Umbilical Tower
MSC	Manned Spacecraft Center (Houston)
MSFN	Manned Space Flight Network
OSCPS	Oxygen Supply and Cabin Pressure Section
Ox	Oxidizer
PCM	Pulse Code Modulation
PCMTEA	Pulse Code Modulation and Timing Electronics Assembly
PGNS	Primary Guidance and Navigation Section
PLSS	Portable Life Support System
RCS	Reaction Control Subsystem
RJB	Relay Junction Box
RR	Rendezvous Radar
S & C	Stabilization and Control
SCA	Signal Conditioner Assembly
SCEA	Signal Conditioning Electronics Assembly

SM	Service Module
TEA	Timing Electronics Assembly
TV	Television
VHF	Very High Frequency (30-300 mc)
WMS	Water Management Section

This does not constitute a complete list of IM abbreviations. Grumman document LLI-790-1 may be consulted for abbreviations not listed here.

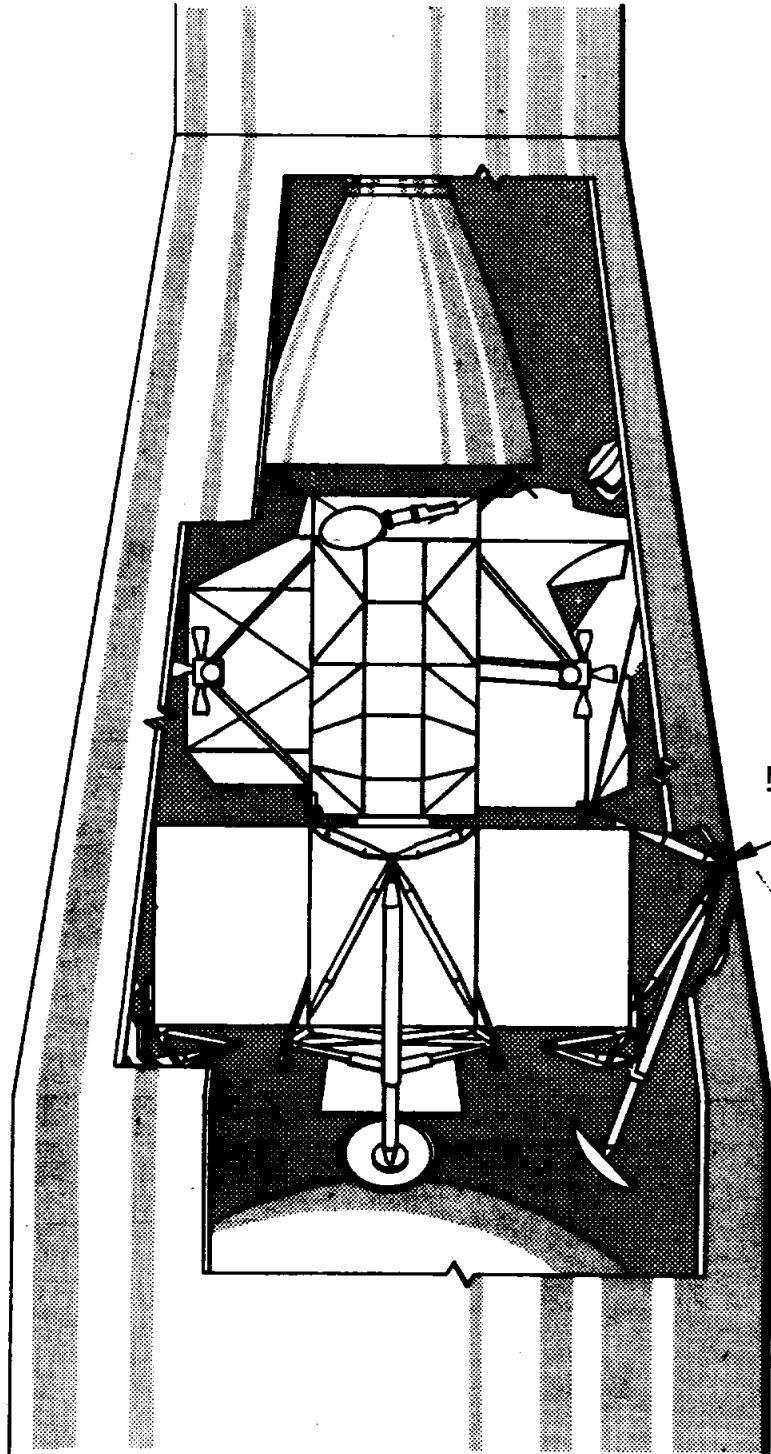




**LM VEHICLE**

T30005-15  
DEC 66

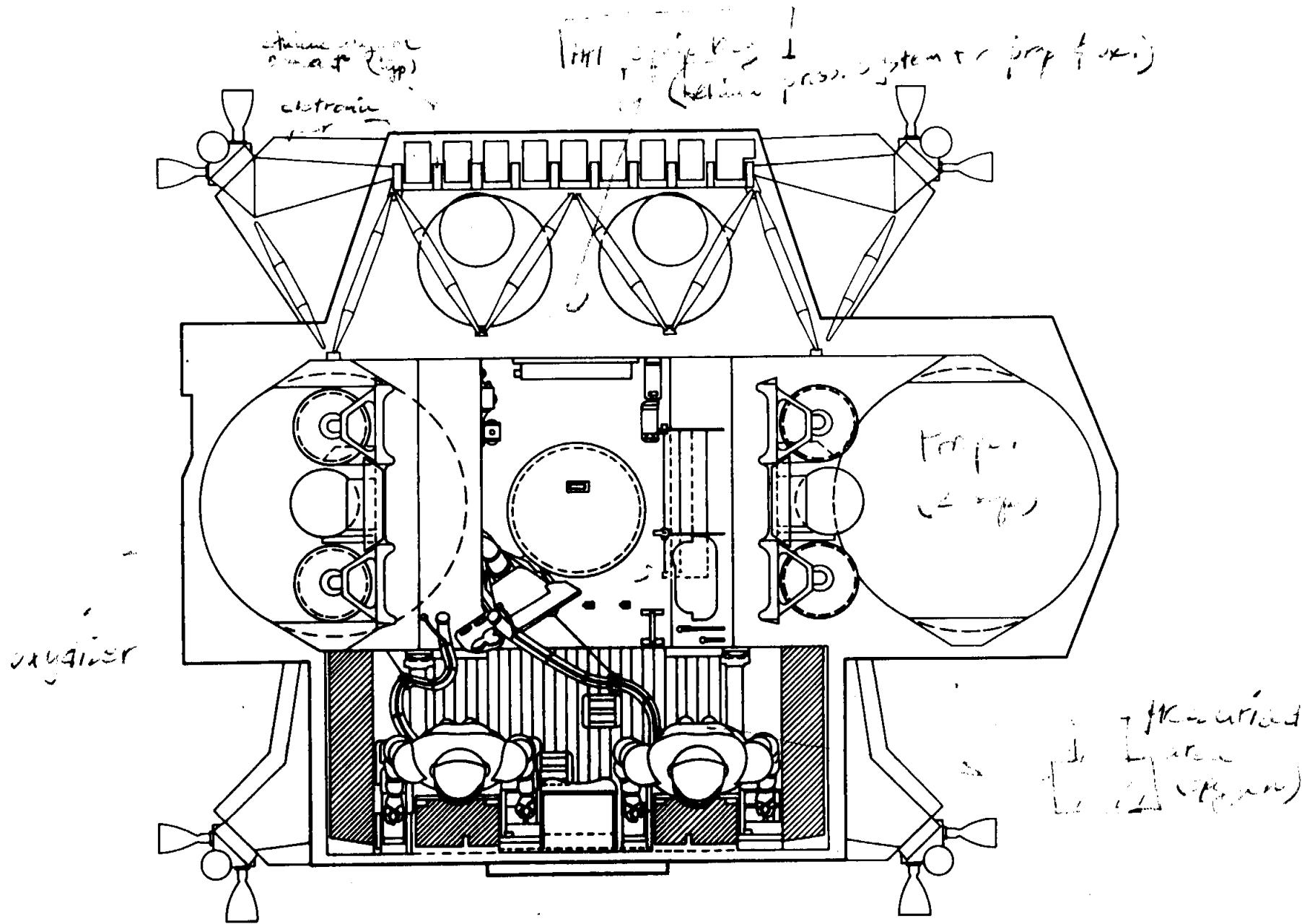
u. det



**LM SIVB BOOSTER INTERFACE**

T30005-14  
DEC 66

*How long for transposition & docking?*

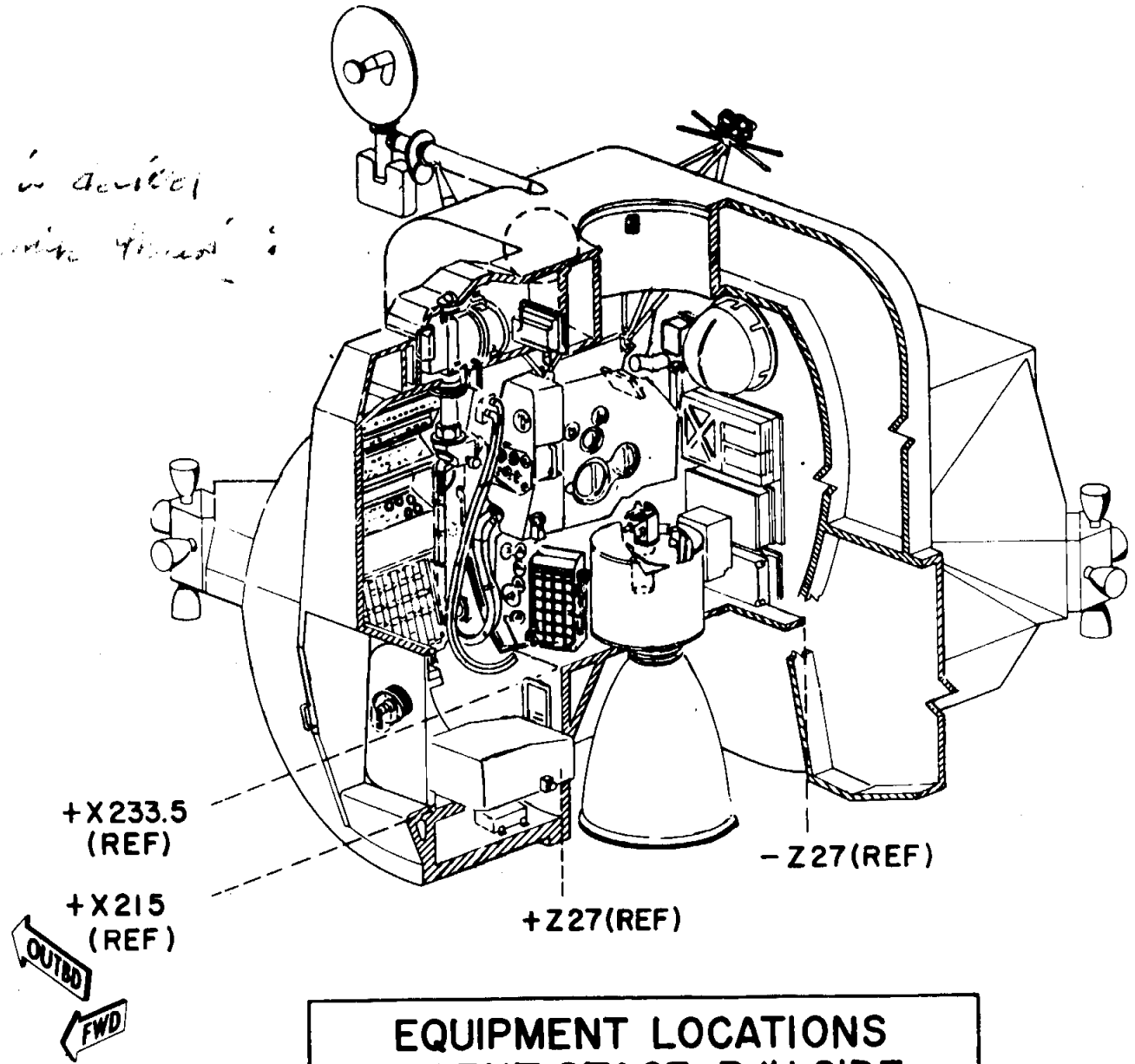


ASCENT STAGE PLAN VIEW

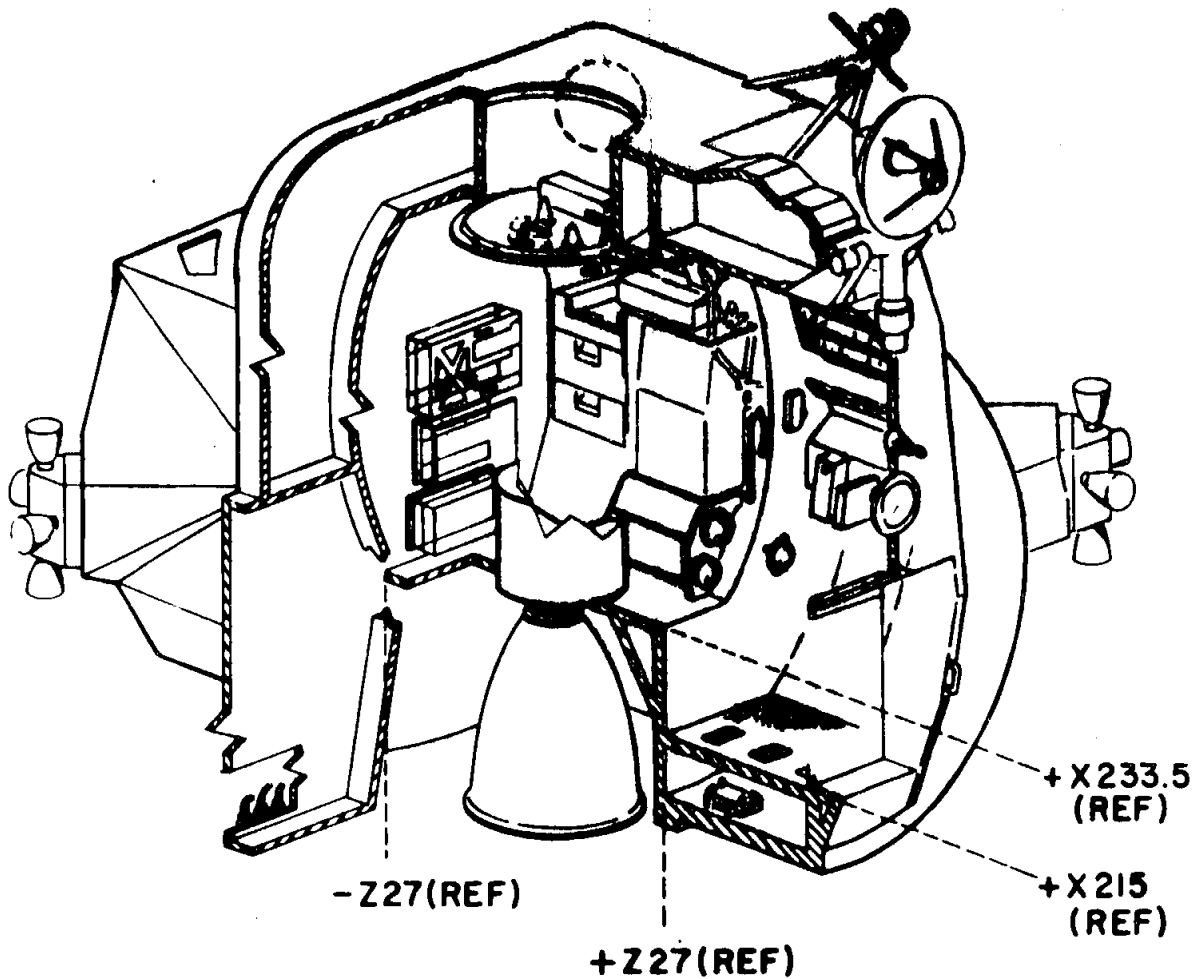
Top View

T30005-131  
MAP 66

What is device  
and what is function?



**EQUIPMENT LOCATIONS  
ASCENT STAGE-R/H SIDE**

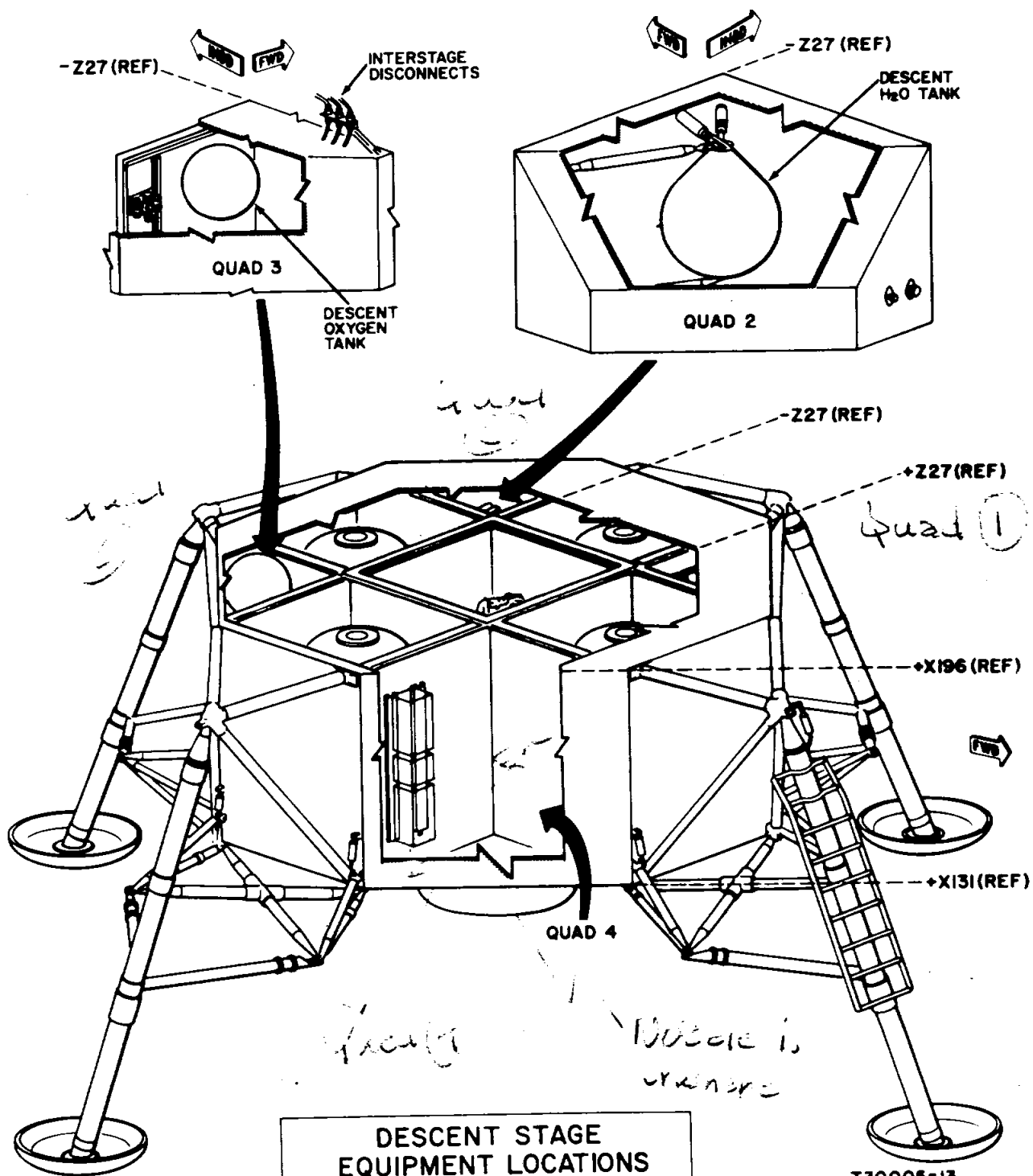


OUTBD

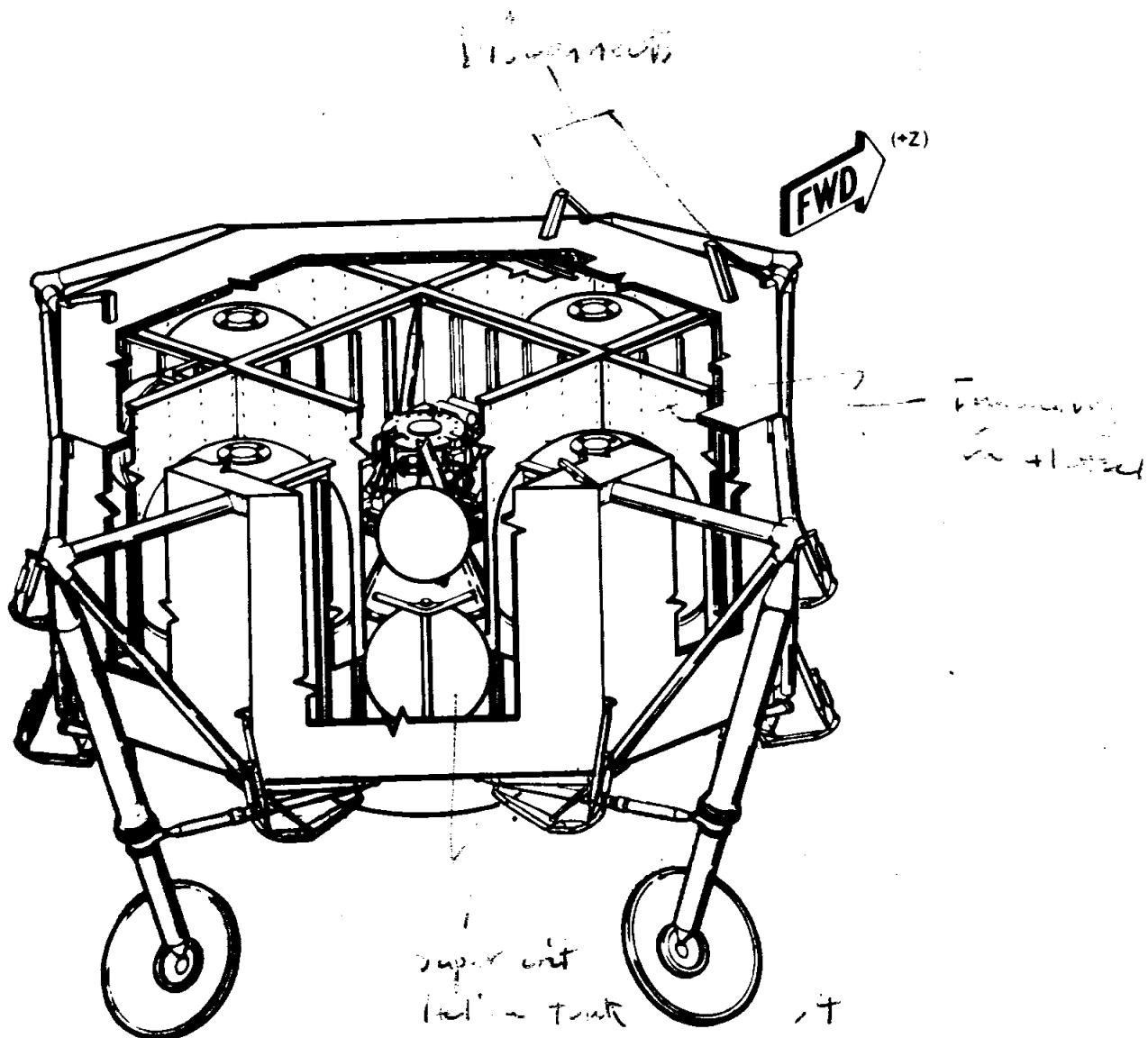
FWD

EQUIPMENT LOCATIONS  
ASCENT STAGE-L/H SIDE

T30005-31  
NOV 66



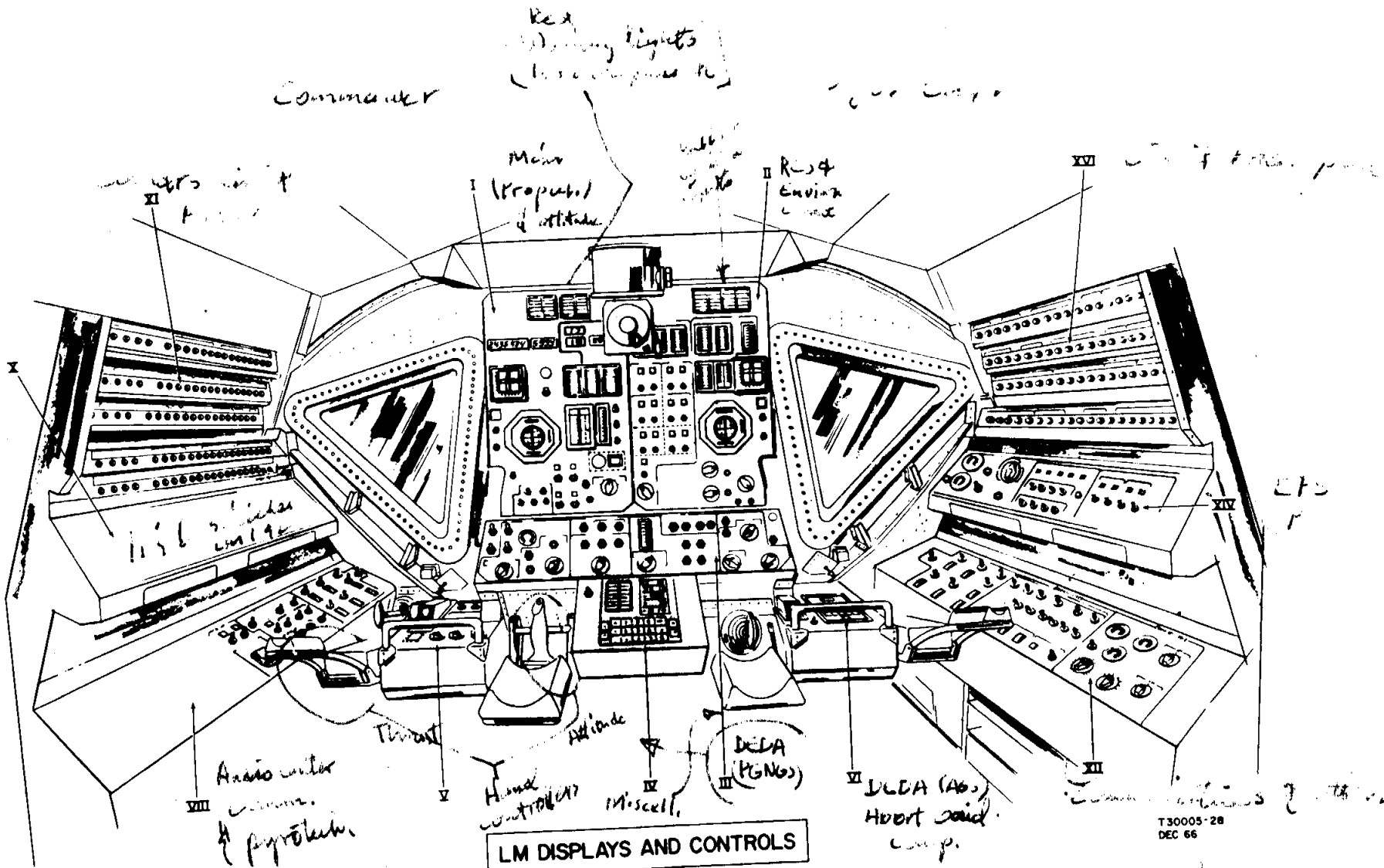
T30005-13  
NOV 66



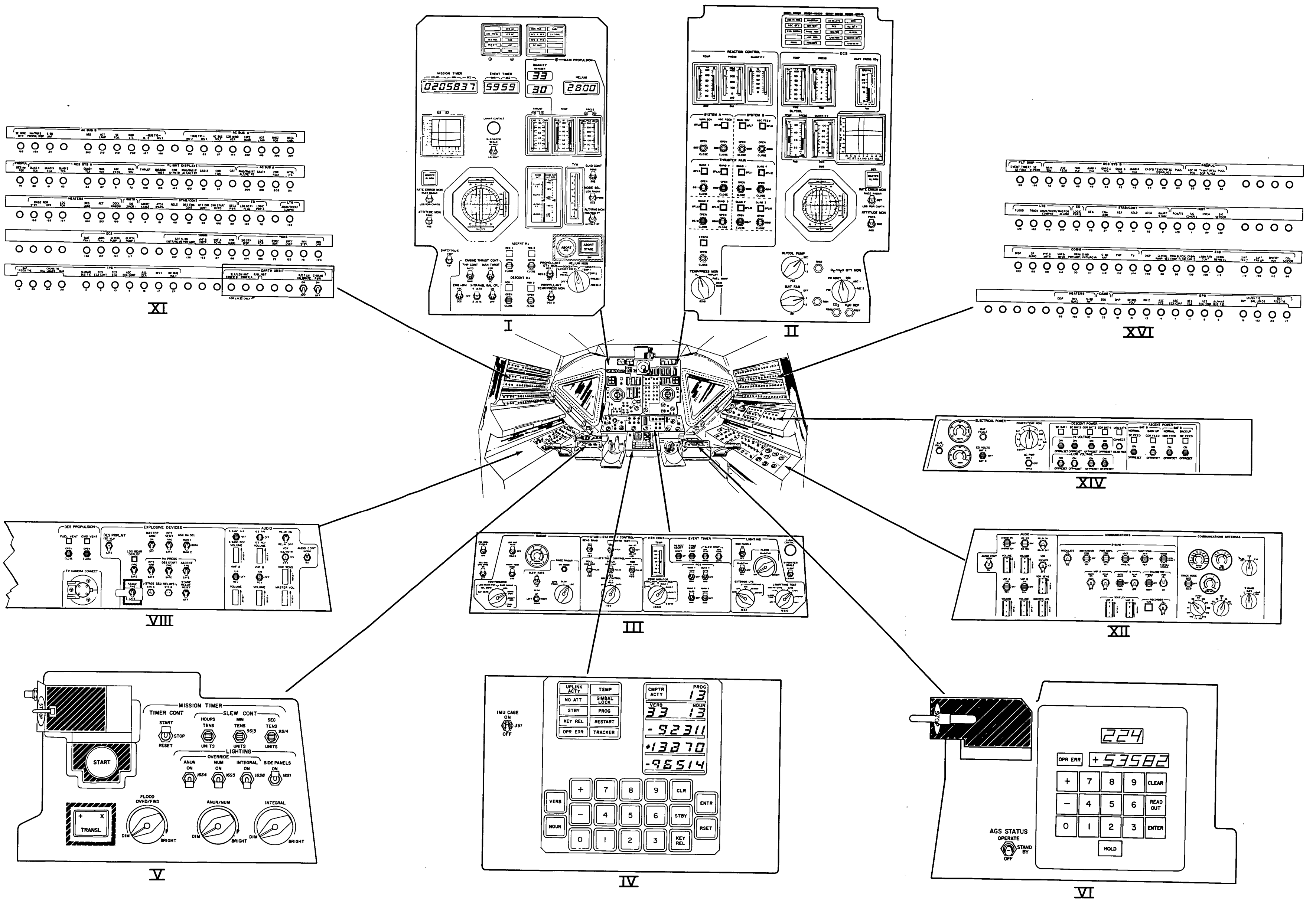
super unit  
 helium tank  
 used to press. prop tanks  
 Gasoline helium used to start  
 then switch to oxygen. Hel.

**LM DESCENT STAGE**

T30005-18  
 NOV 66

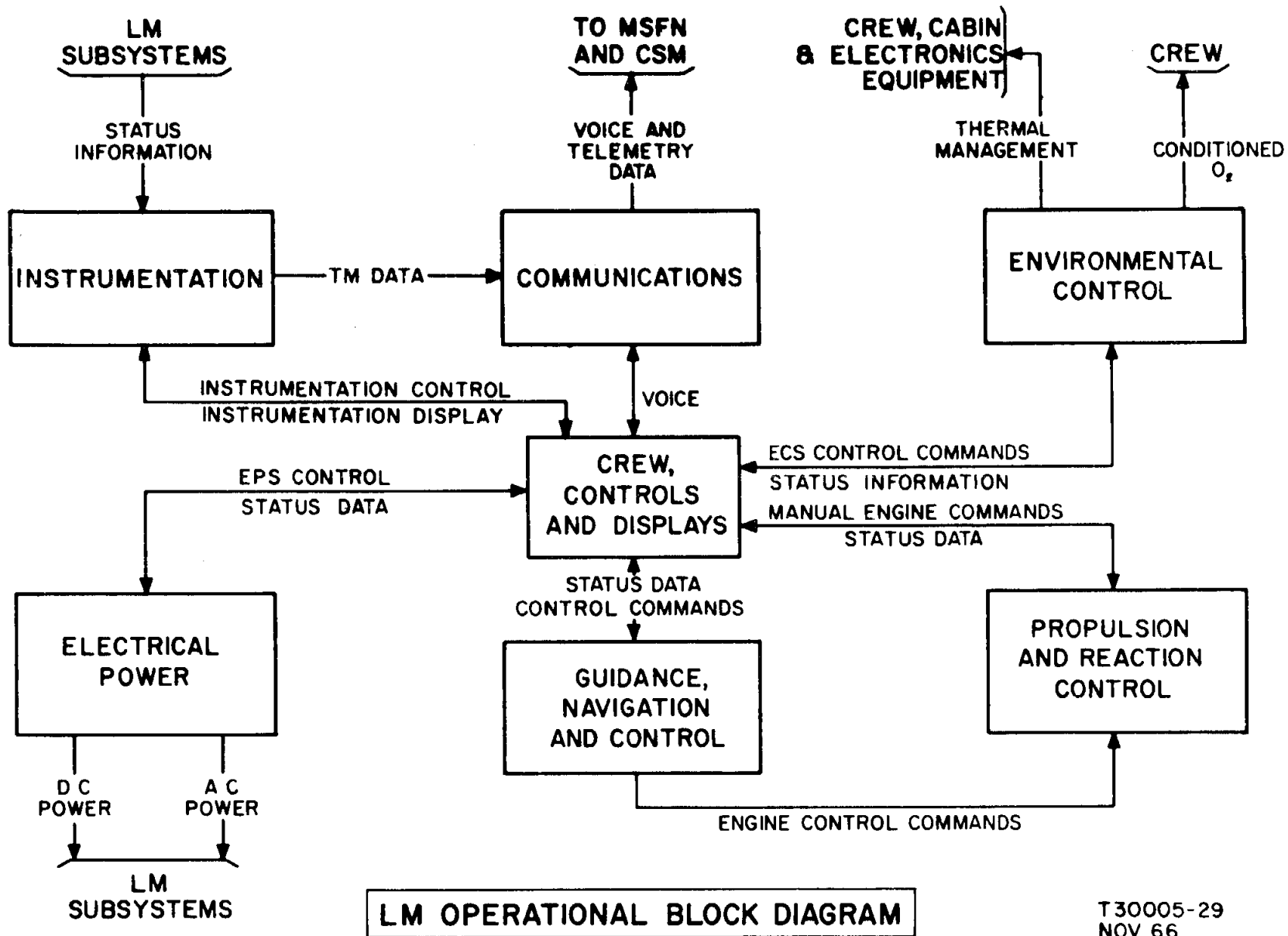




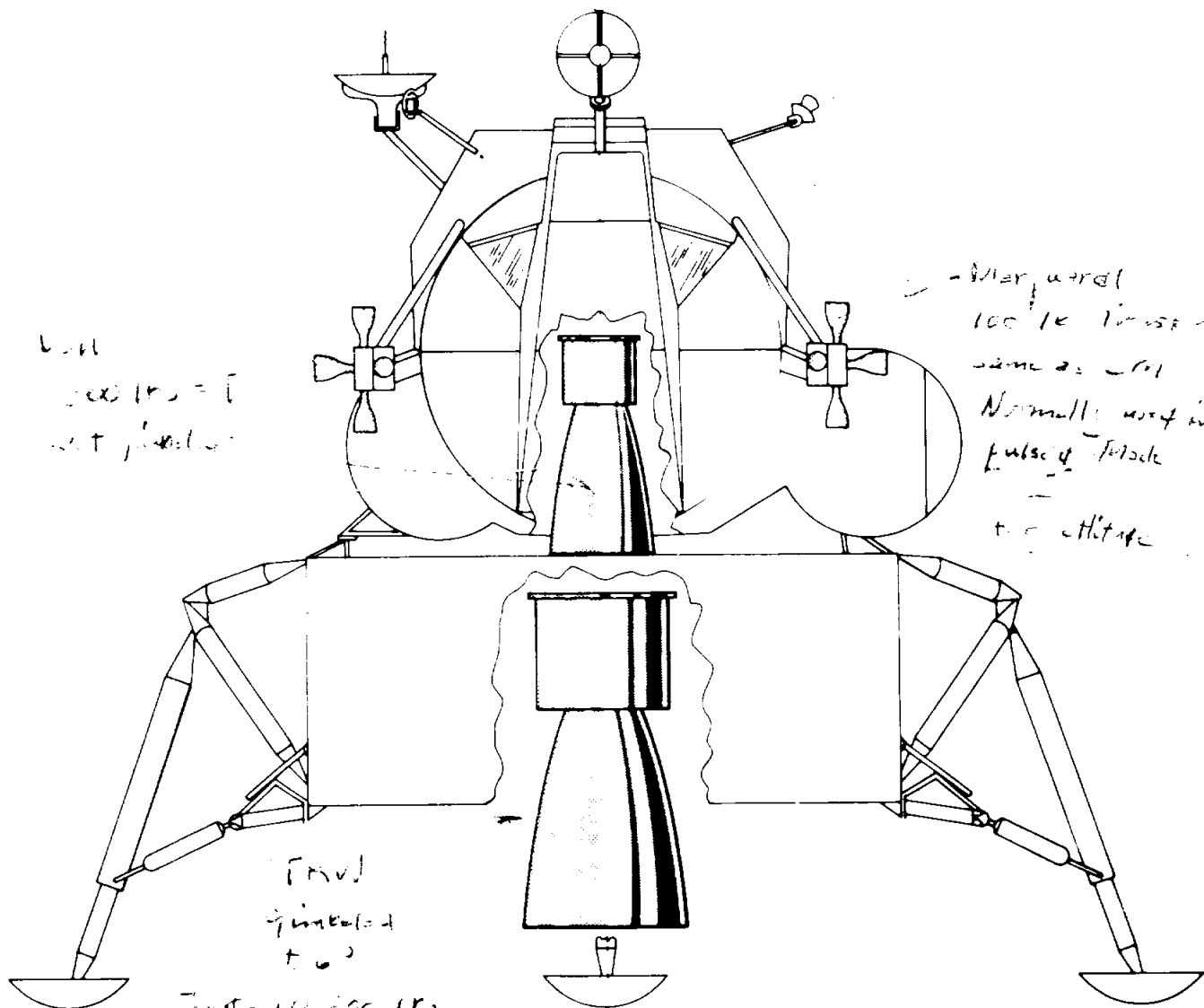


# LM DISPLAYS AND CONTROLS

T30005-28  
DEC 66



T30005-29  
NOV 66



Well  
 2000 lbs = [ ]  
 not possible

- Star used  
 100 lb thrust  
 same as [ ]  
 Normally used in  
 pulse mode  
 to attitude

Truss  
 gimbaled  
 to [ ]

Thrust = 10,000 lbs  
 Can be gimbaled 10 deg.

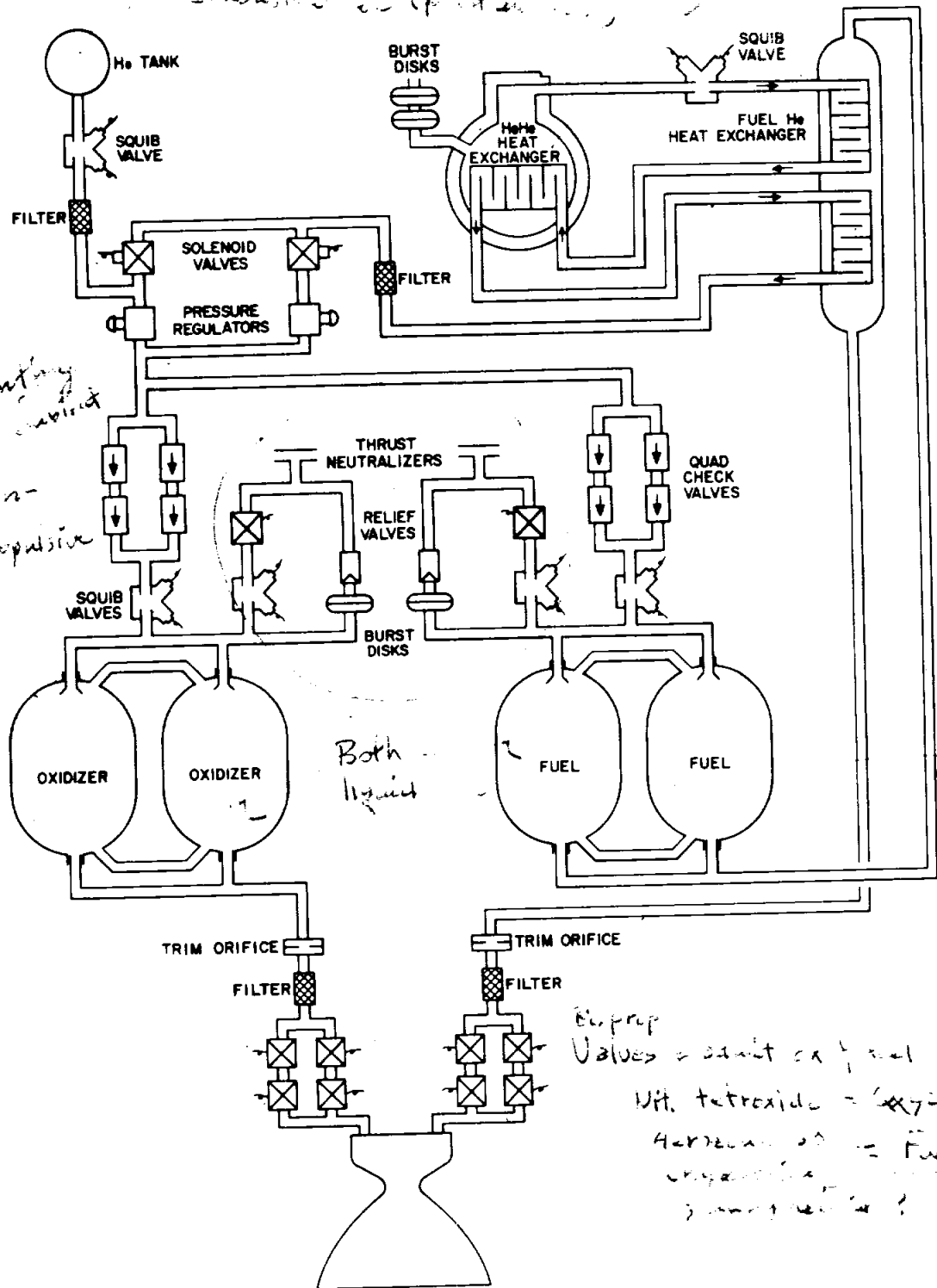
Variables  
 Continuously  
 (Not 11. write [ ])

**LEM PROPULSION**

T30005-66  
 FEB 66

*To tank prop. to  
 sensitive control system*

*2017-1-12 10:10*



*Venting to cabinet  
 Non-propulsive*

*Final pass  
 to He exchanger  
 to set up  
 the upstroke  
 relief -  
 continuous  
 pressure*

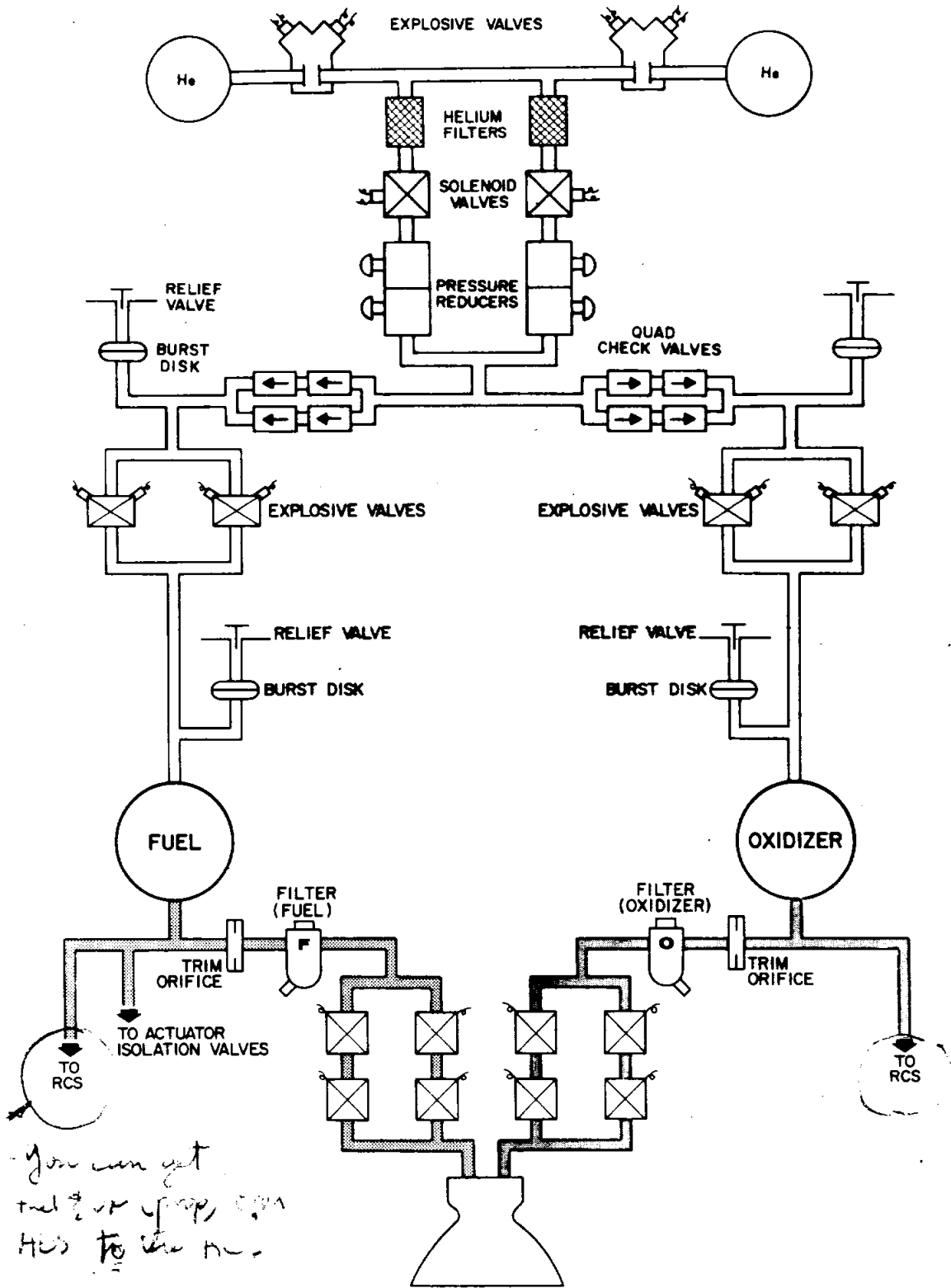
*Both liquid*

*Flow at  
 100%*

*Exp. prop  
 Values a suit on fuel  
 Nit. tetroxide = oxidizer  
 Hydrogen peroxide = fuel  
 ...*

**DESCENT PROPULSION SCHEMATIC**

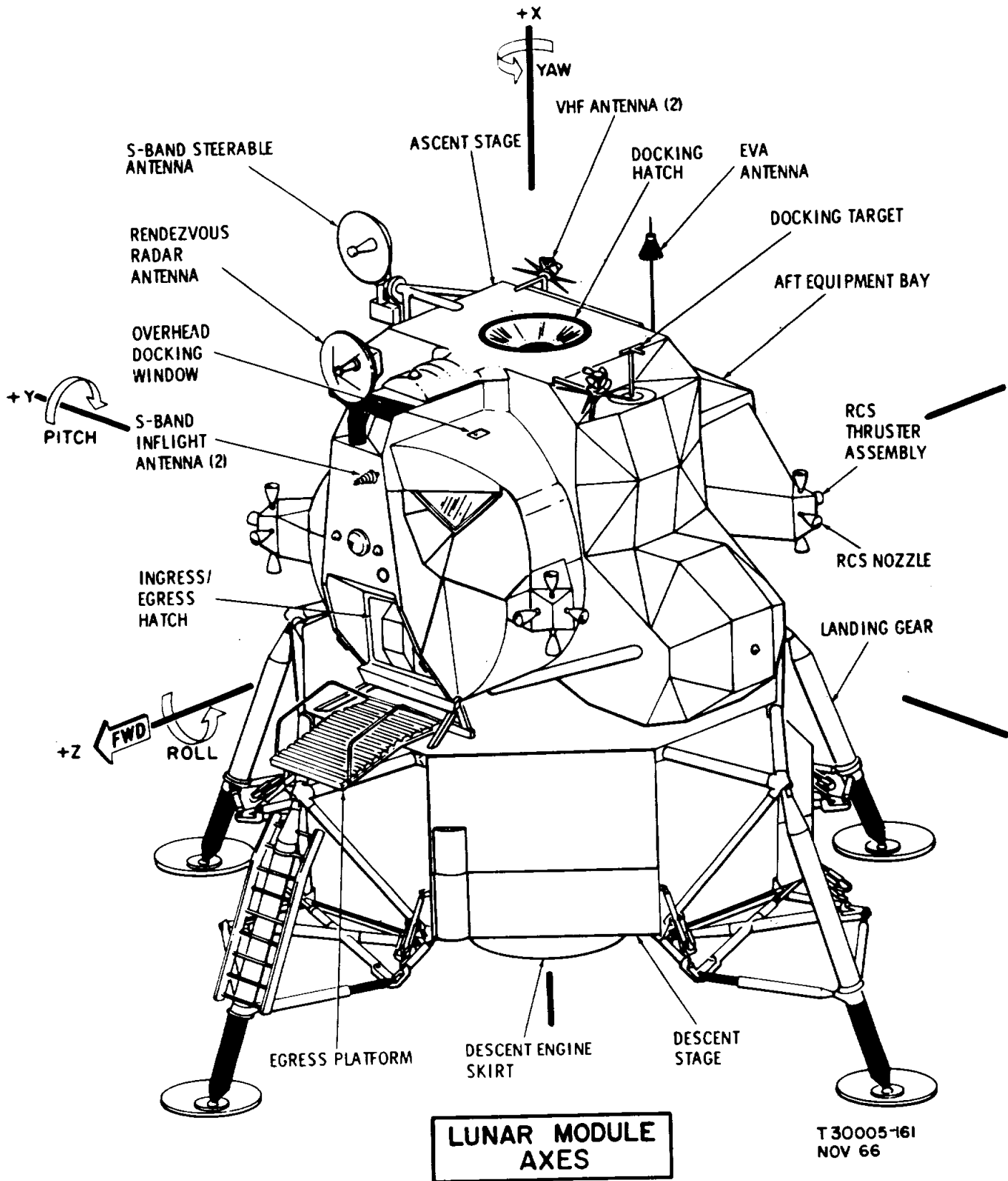
T 30005-154  
 NOV 66

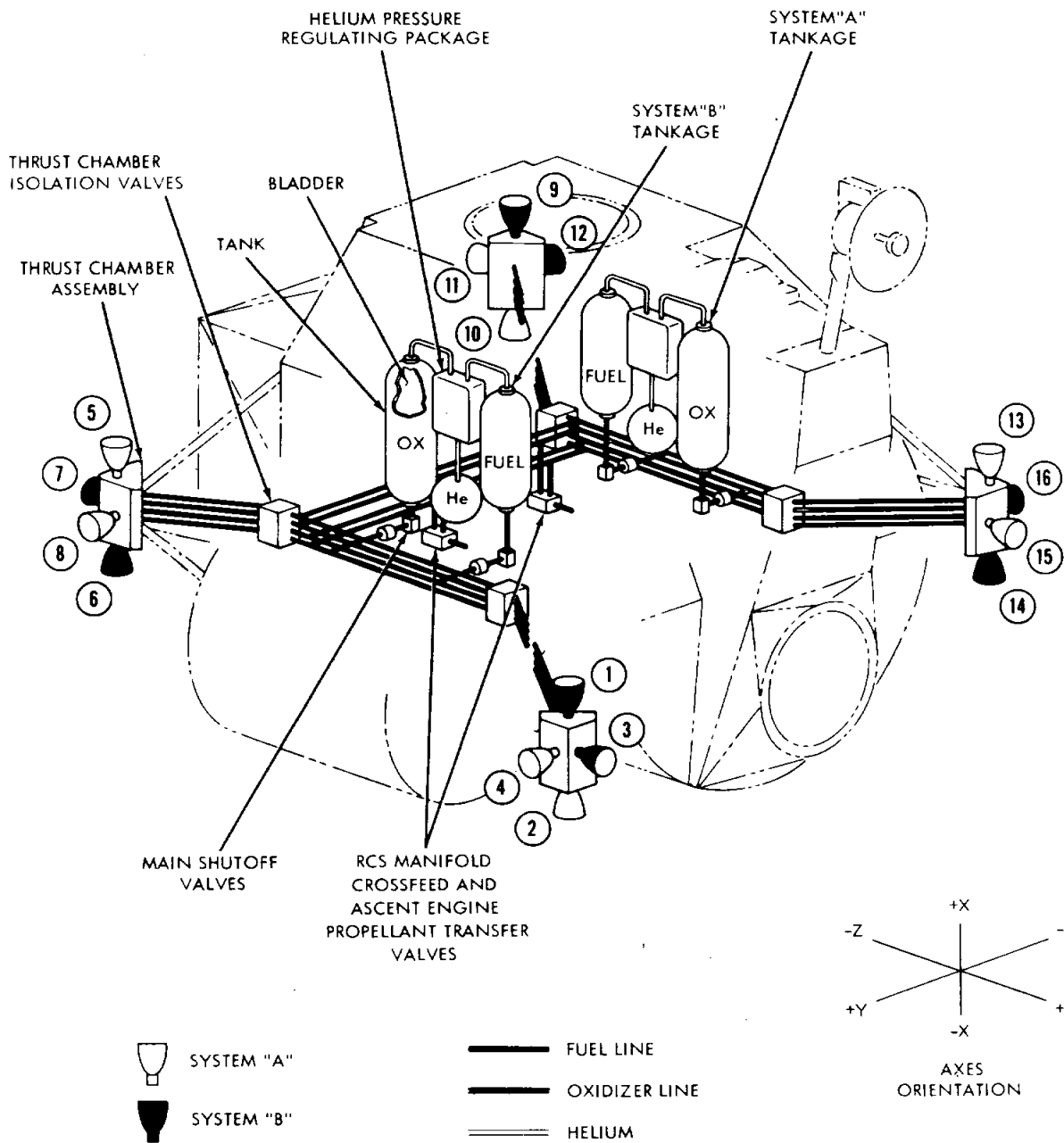


*You can get  
fuel & ox prop from  
He's to the no.*

*Not necessary*

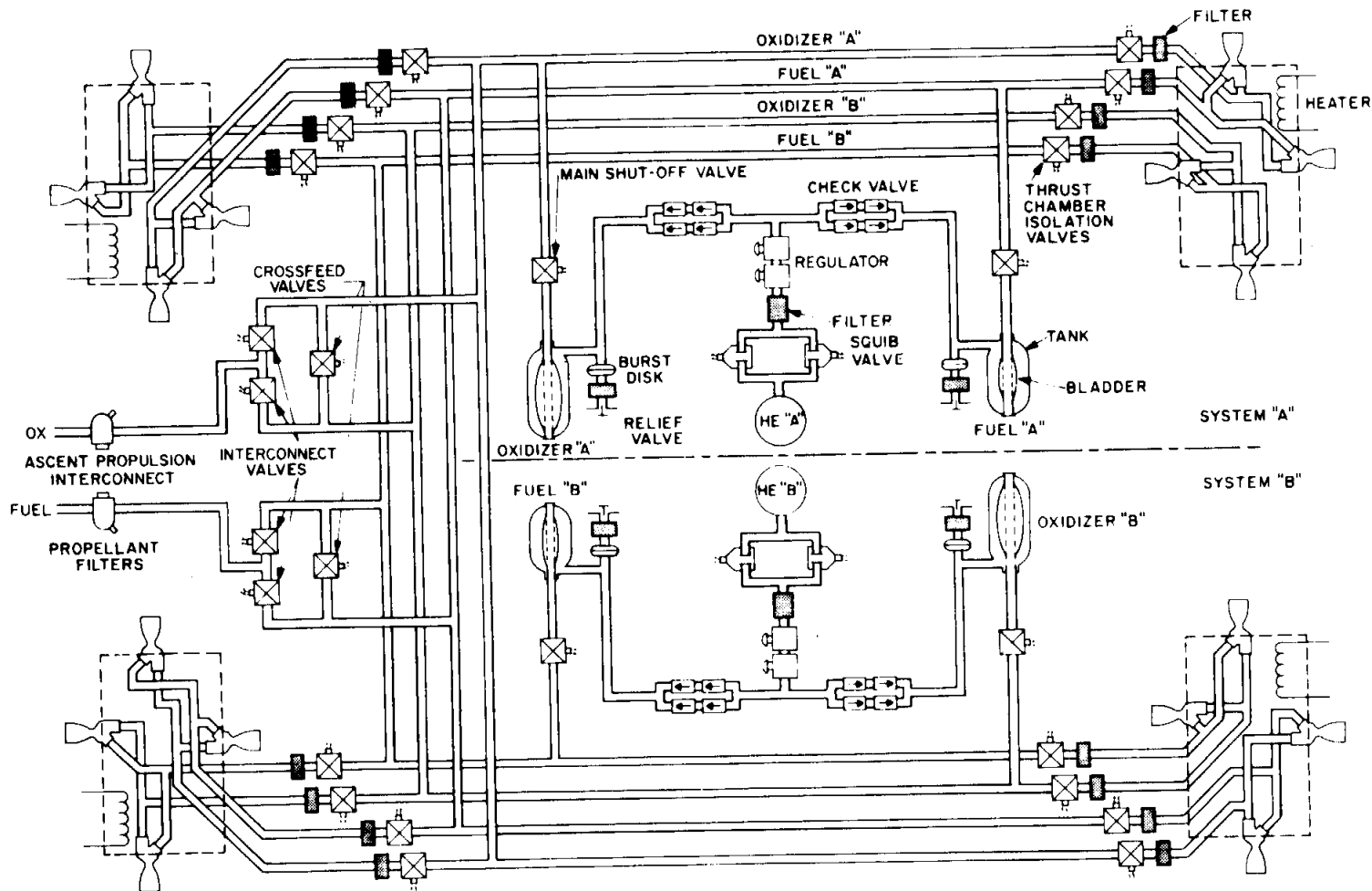
**ASCENT PROPELLANT SYSTEM**





# RCS INSTALLATION

T30005-149  
NOV 66

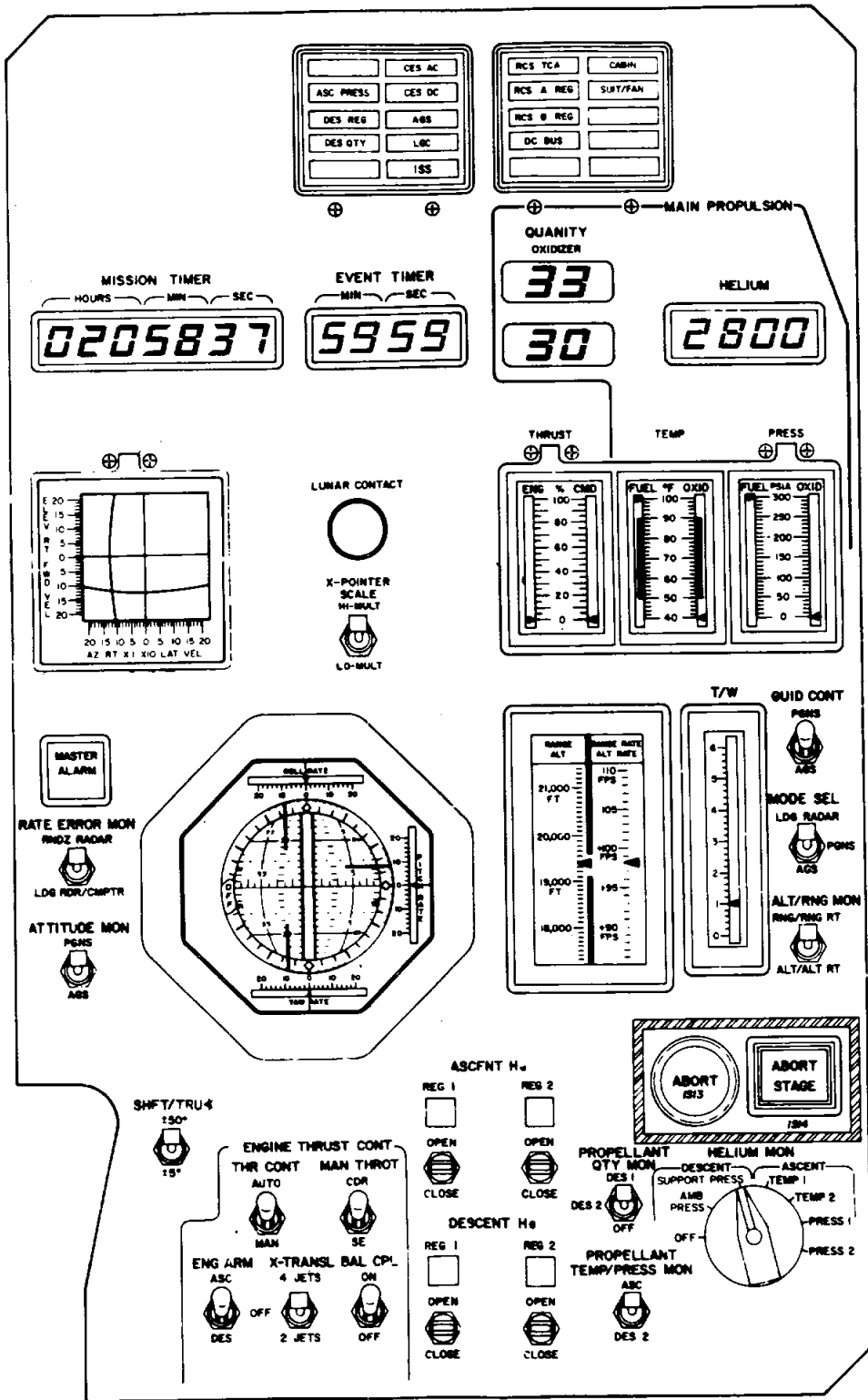


**REACTION CONTROL  
SUBSYSTEM SCHEMATIC**

T30005-151  
NOV 66

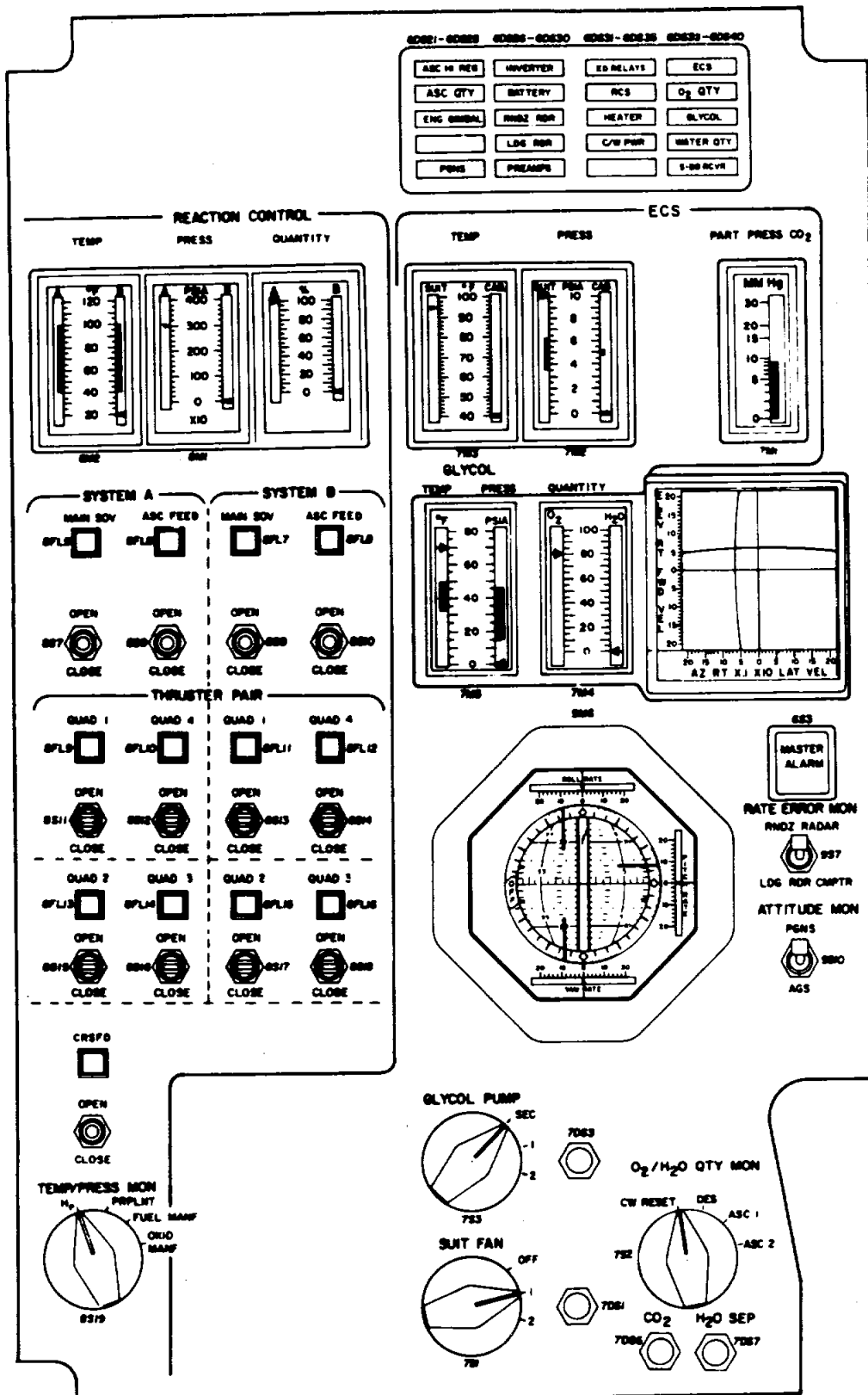


Document 11/2-16-67  
 Page 45



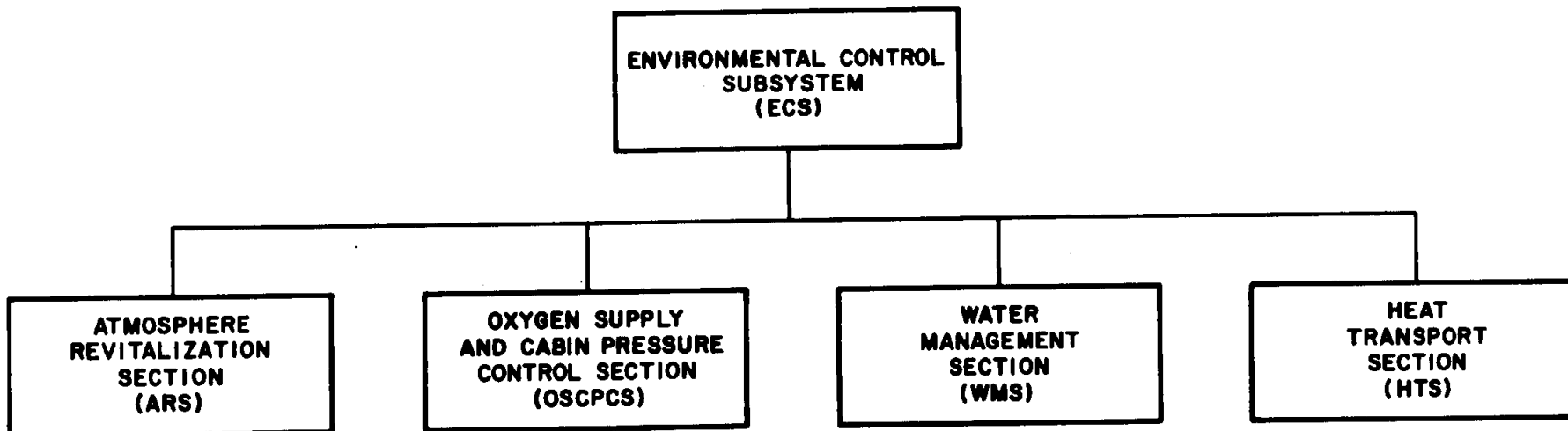
PANEL I

T30005-152  
 DEC 66



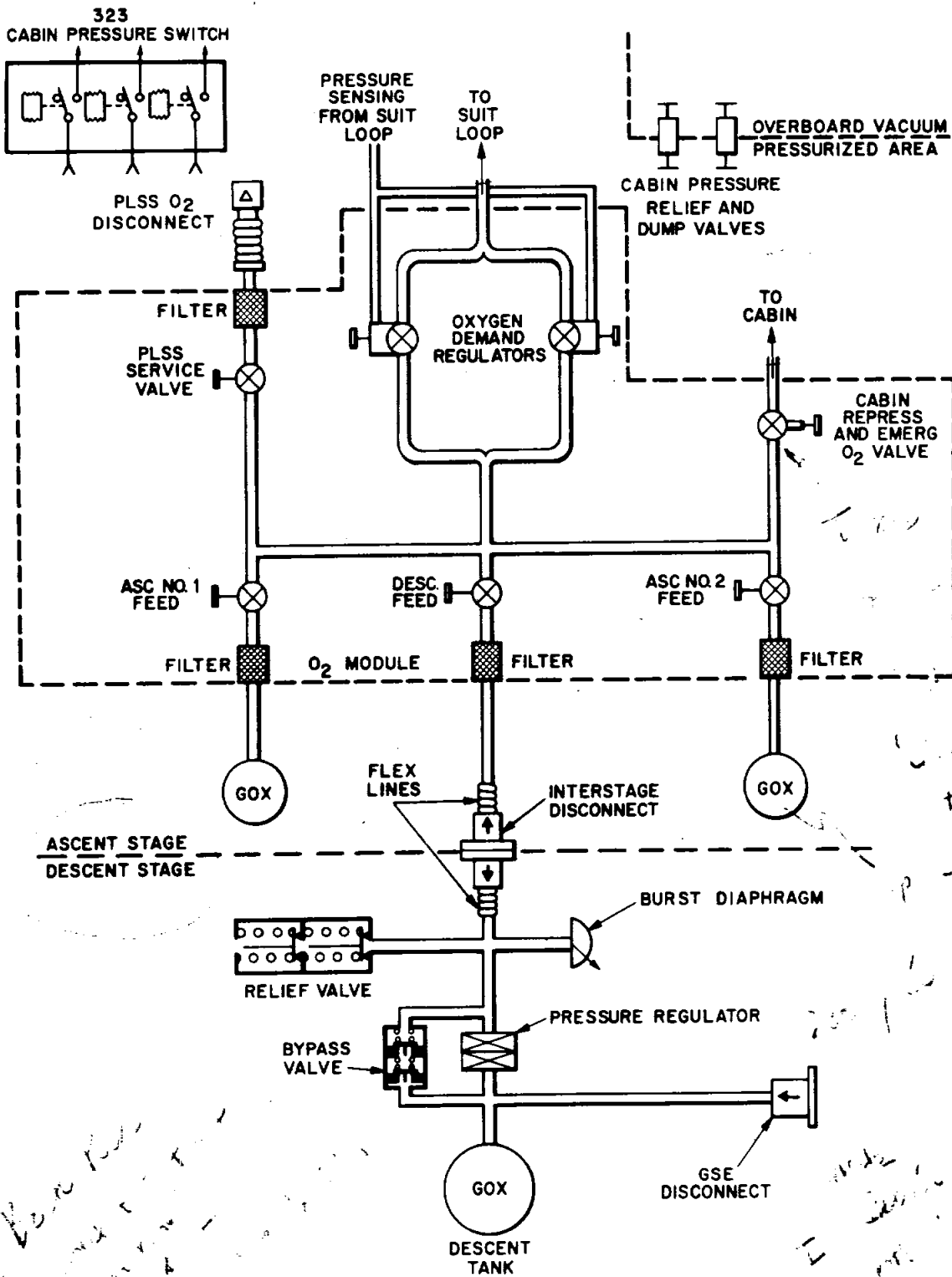
PANEL II

T30005-160  
DEC 66



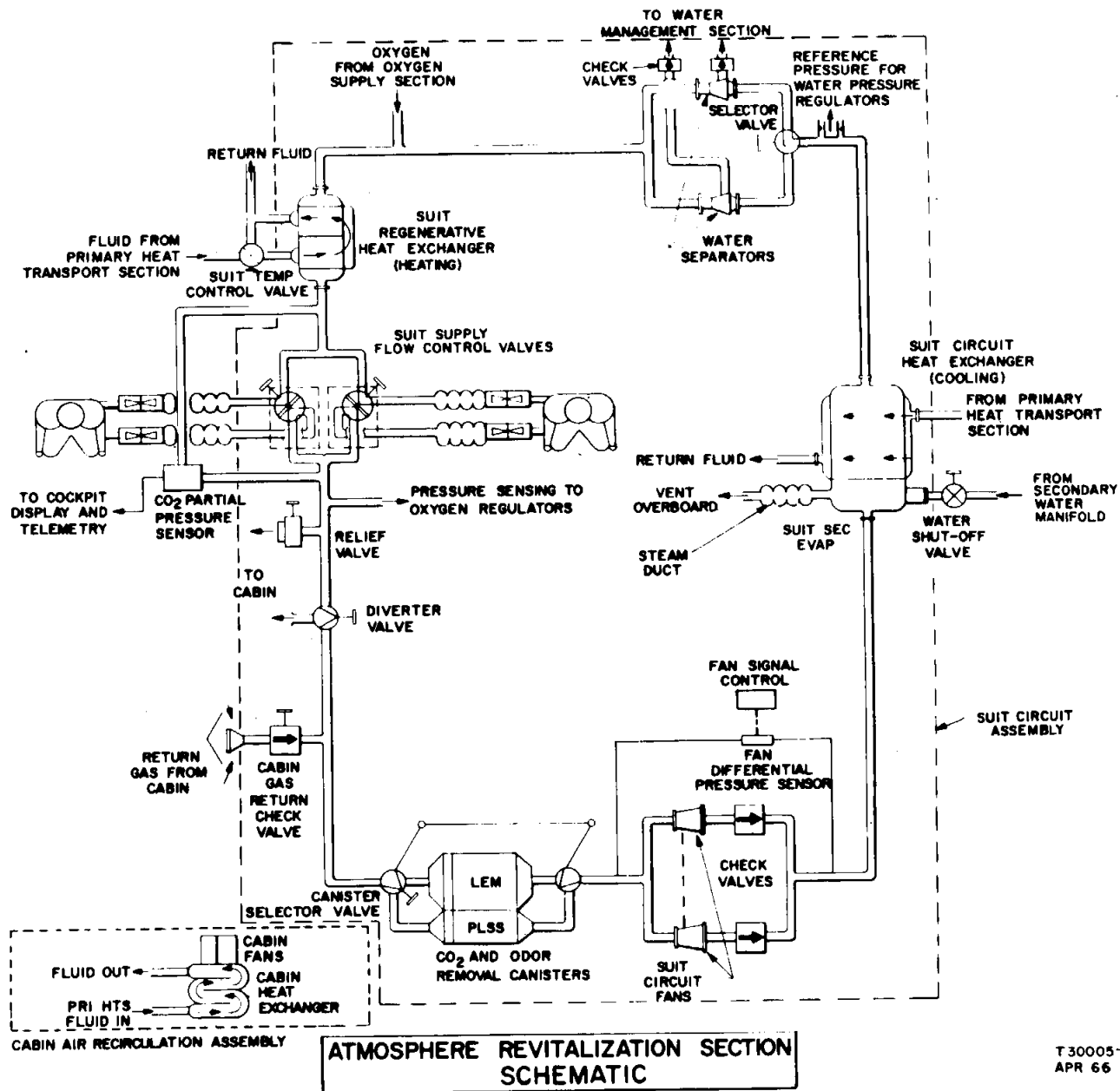
**LM ECS BASIC BLOCK DIAGRAM**

T30005-61  
FEB. 66



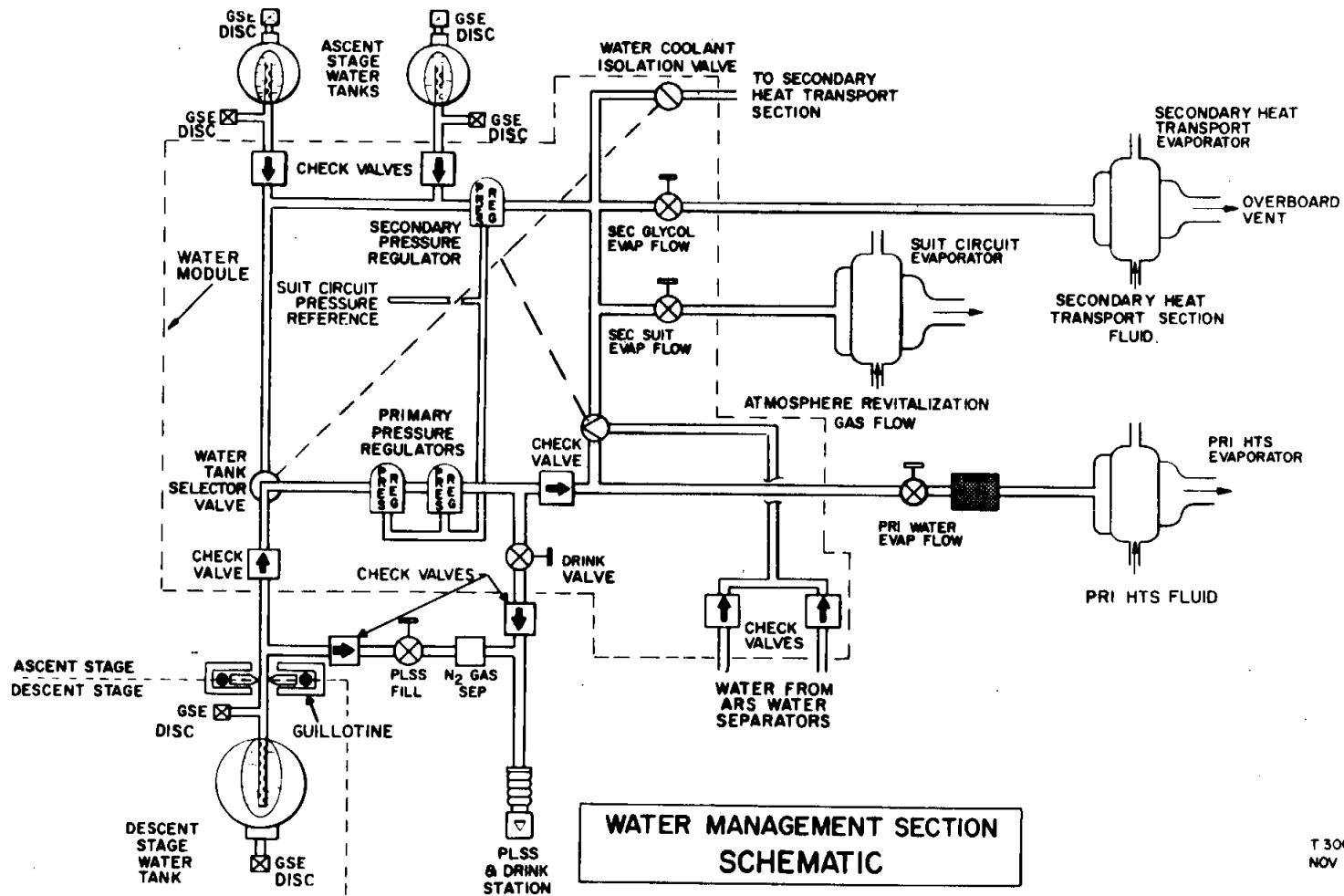
**OXYGEN SUPPLY AND CABIN PRESSURE CONTROL SECTION SCHEMATIC**

T30005-158  
APR 66



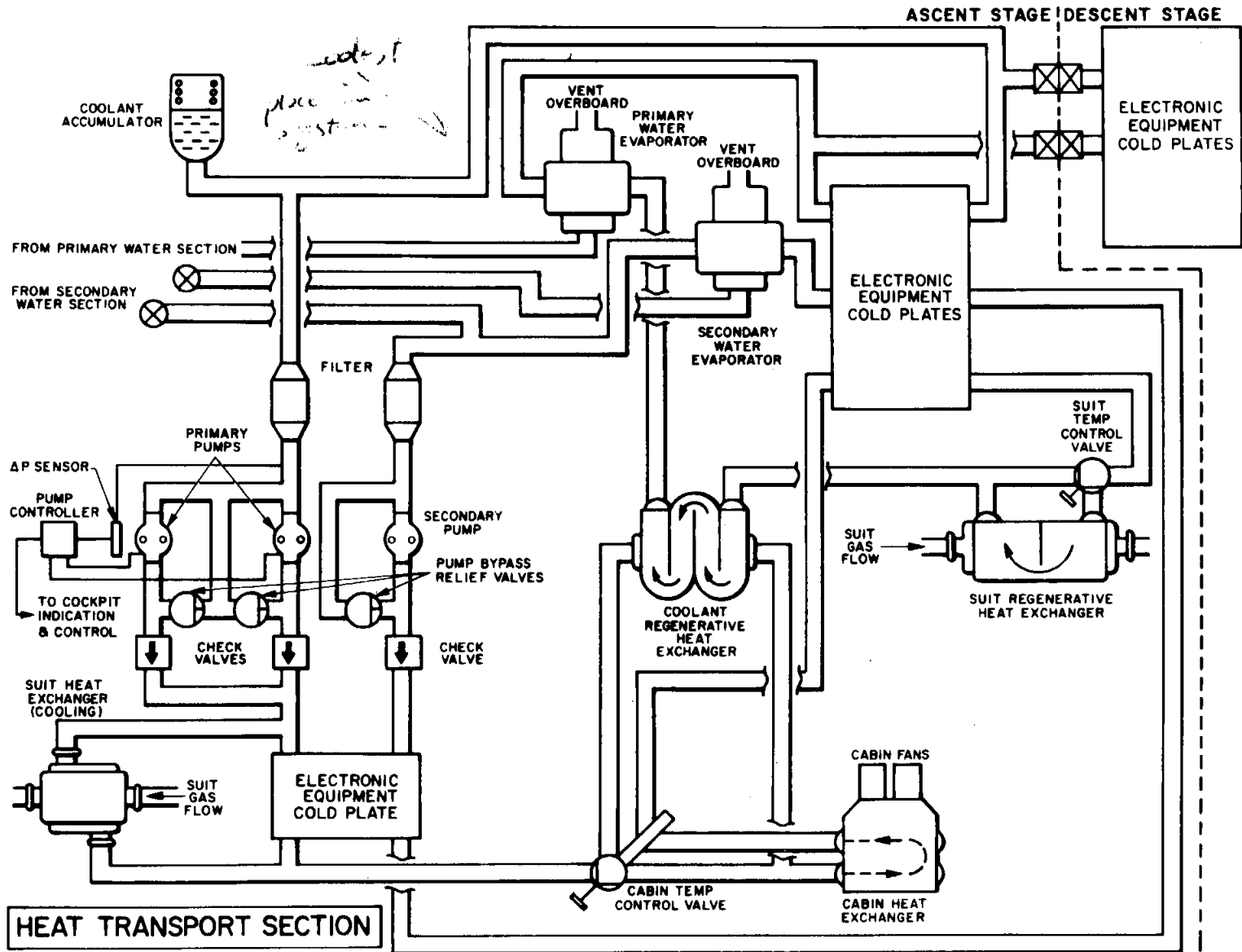
**ATMOSPHERE REVITALIZATION SECTION SCHEMATIC**

T30005-157  
APR 66



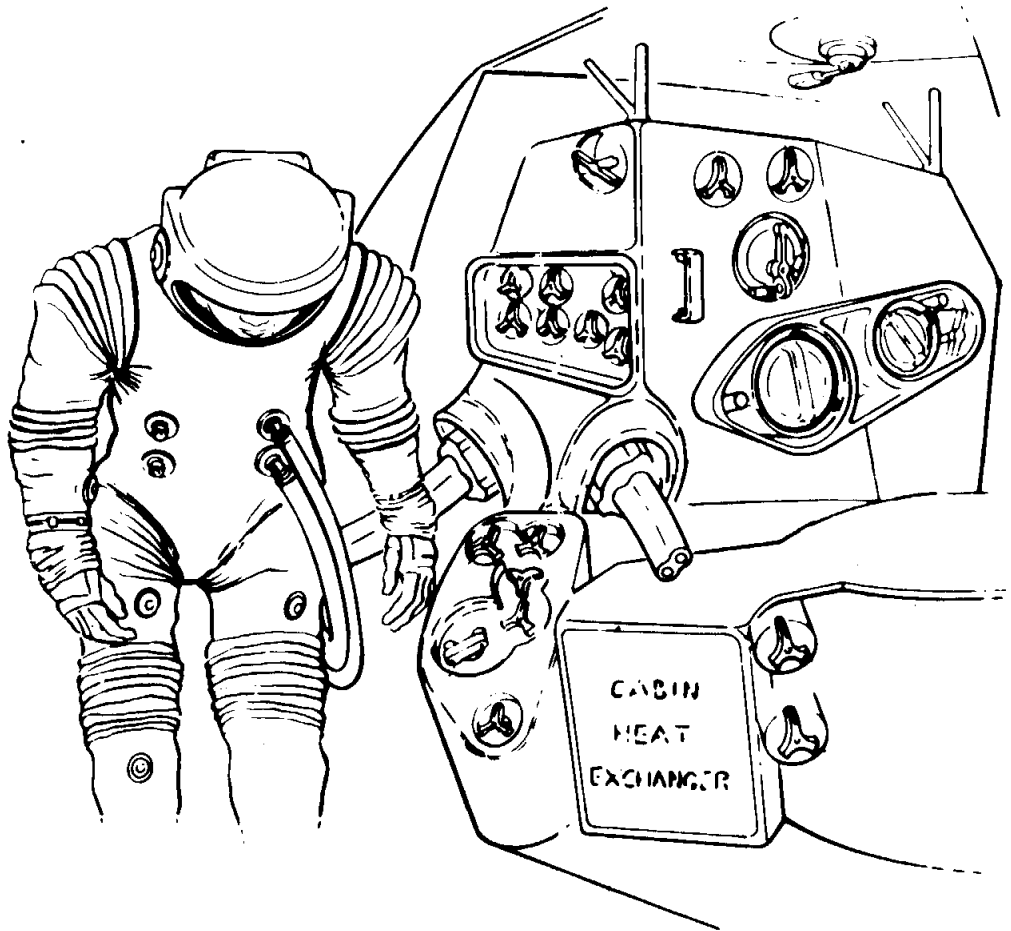
**WATER MANAGEMENT SECTION  
SCHEMATIC**

T 30005-156  
NOV 66



T30005-159  
NOV 66

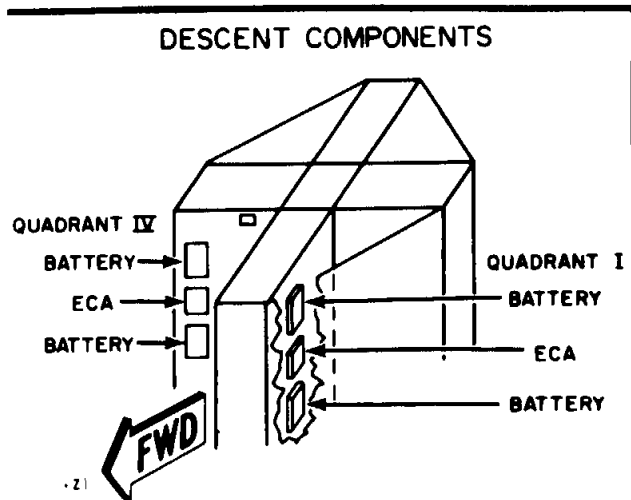
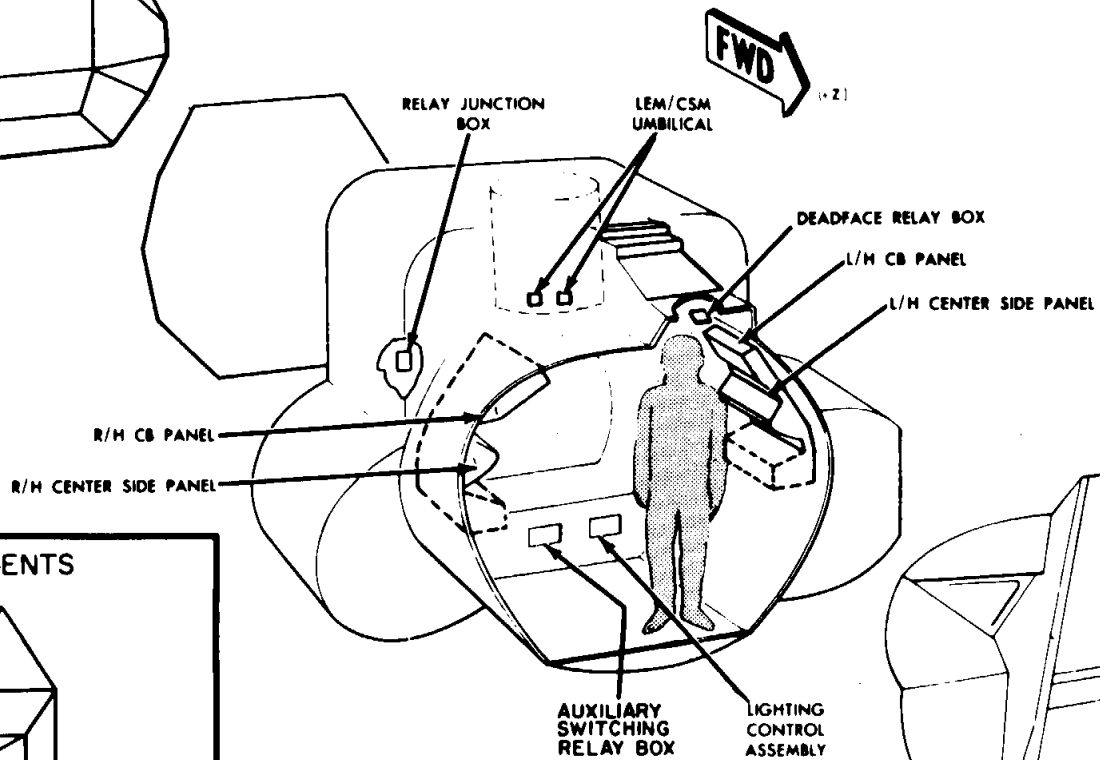
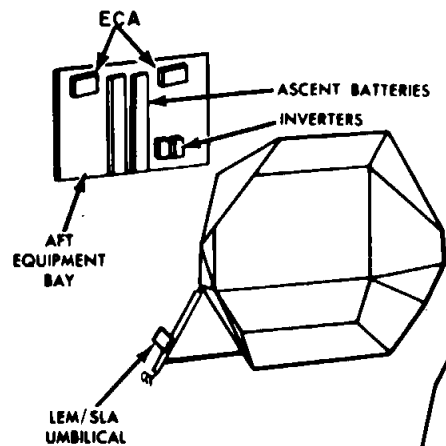
Loss of primary coolant loop - - - is about 10%  
because the pump assembly is not at the top of the loop



ECS - SPACE SUIT INTERFACE

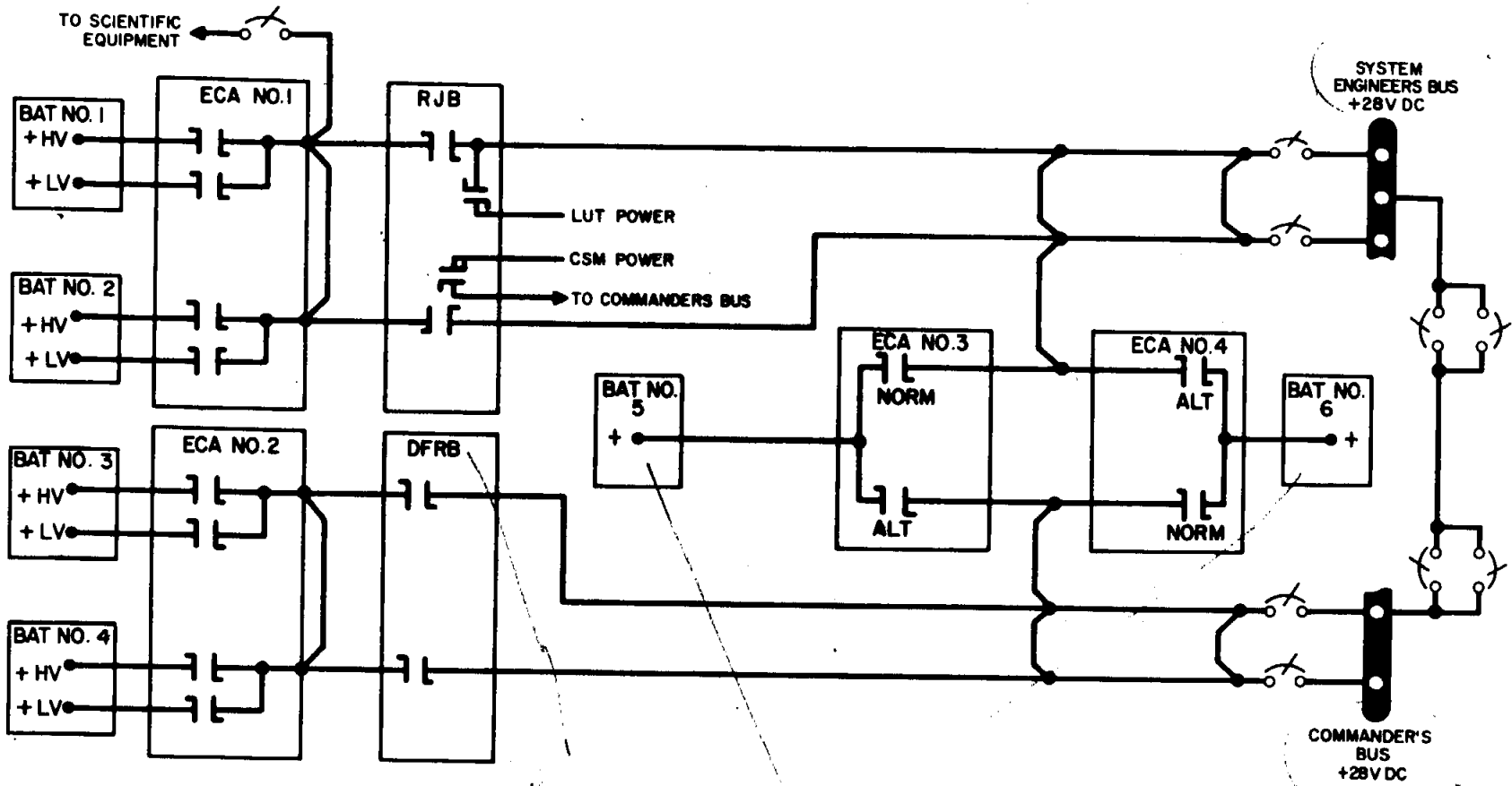
T30005-33  
NOV 66





**EPS COMPONENT LOCATION**

T30005-19  
DEC 66



*Don't put*

*Don't put*

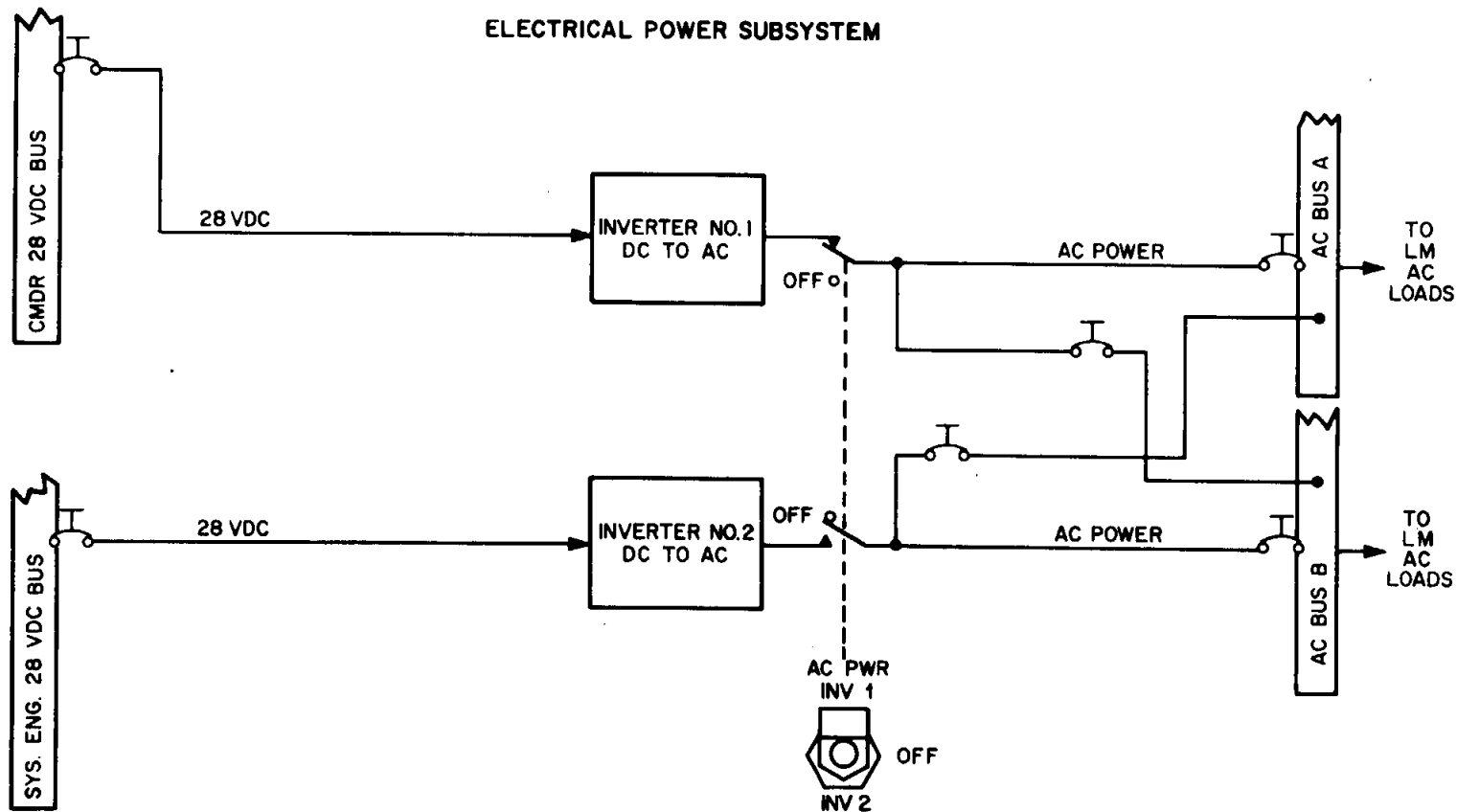
*Mount Ball*

*Don't put  
to scientific*

**EPS FEEDER SYSTEM**

T30005-126  
NOV 66

# ELECTRICAL POWER SUBSYSTEM

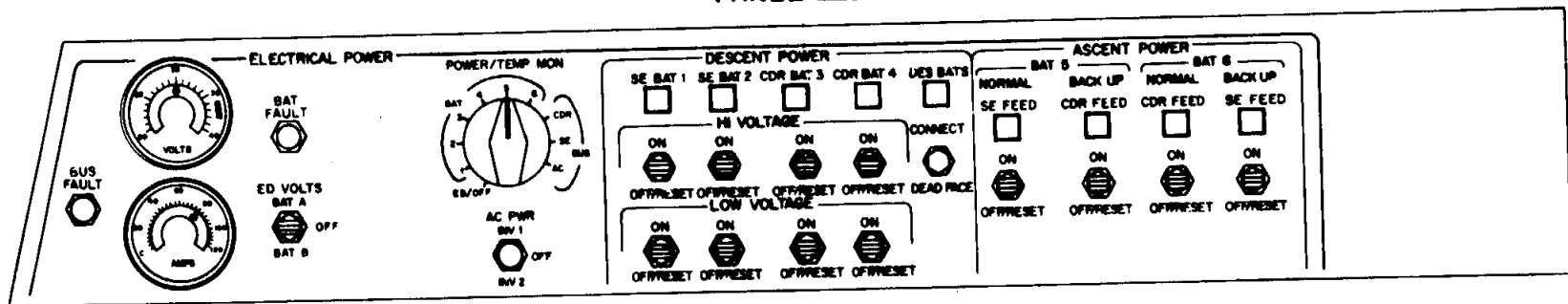


AC DISTRIBUTION BLOCK DIAGRAM

T30005-21  
NOV 66

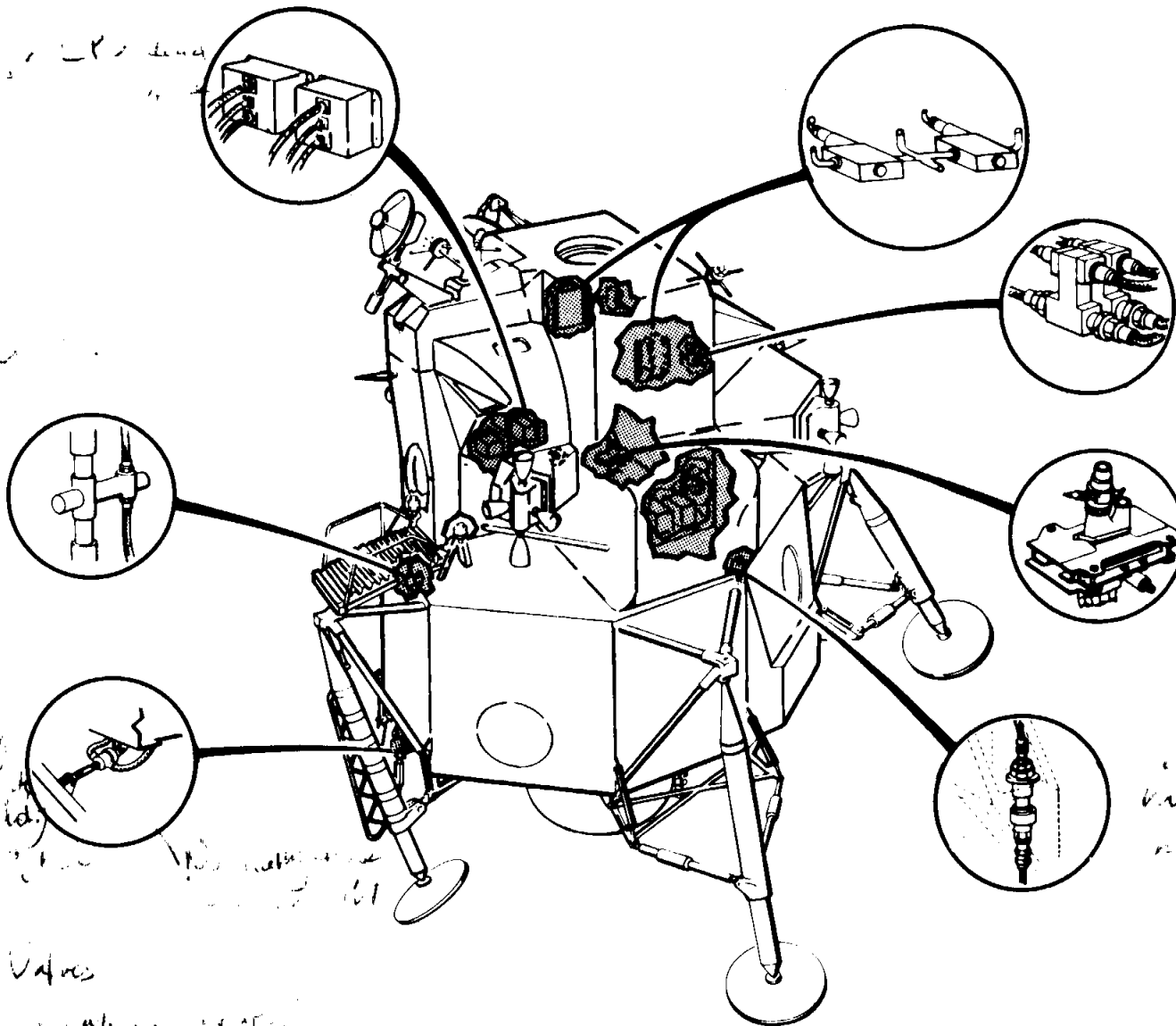
# ELECTRICAL POWER SUBSYSTEM

## PANEL XIV



CONTROLS AND DISPLAYS

T30005-128  
DEC 66



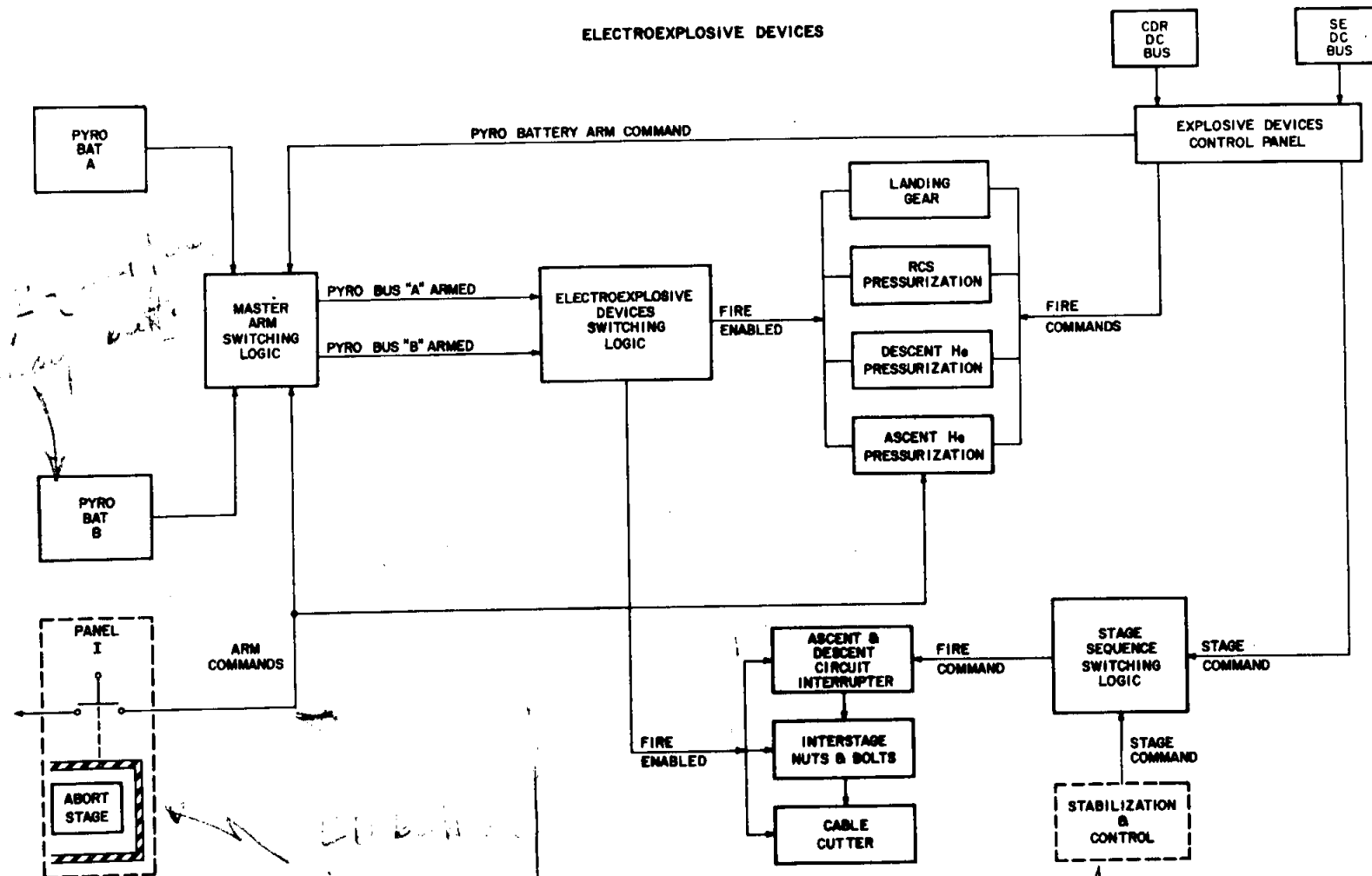
Squib Valves  
 Max. in prop. ...  
 ...

interster  
 nut of bolt (+)

**EXPLOSIVE DEVICES-LOCATIONS**

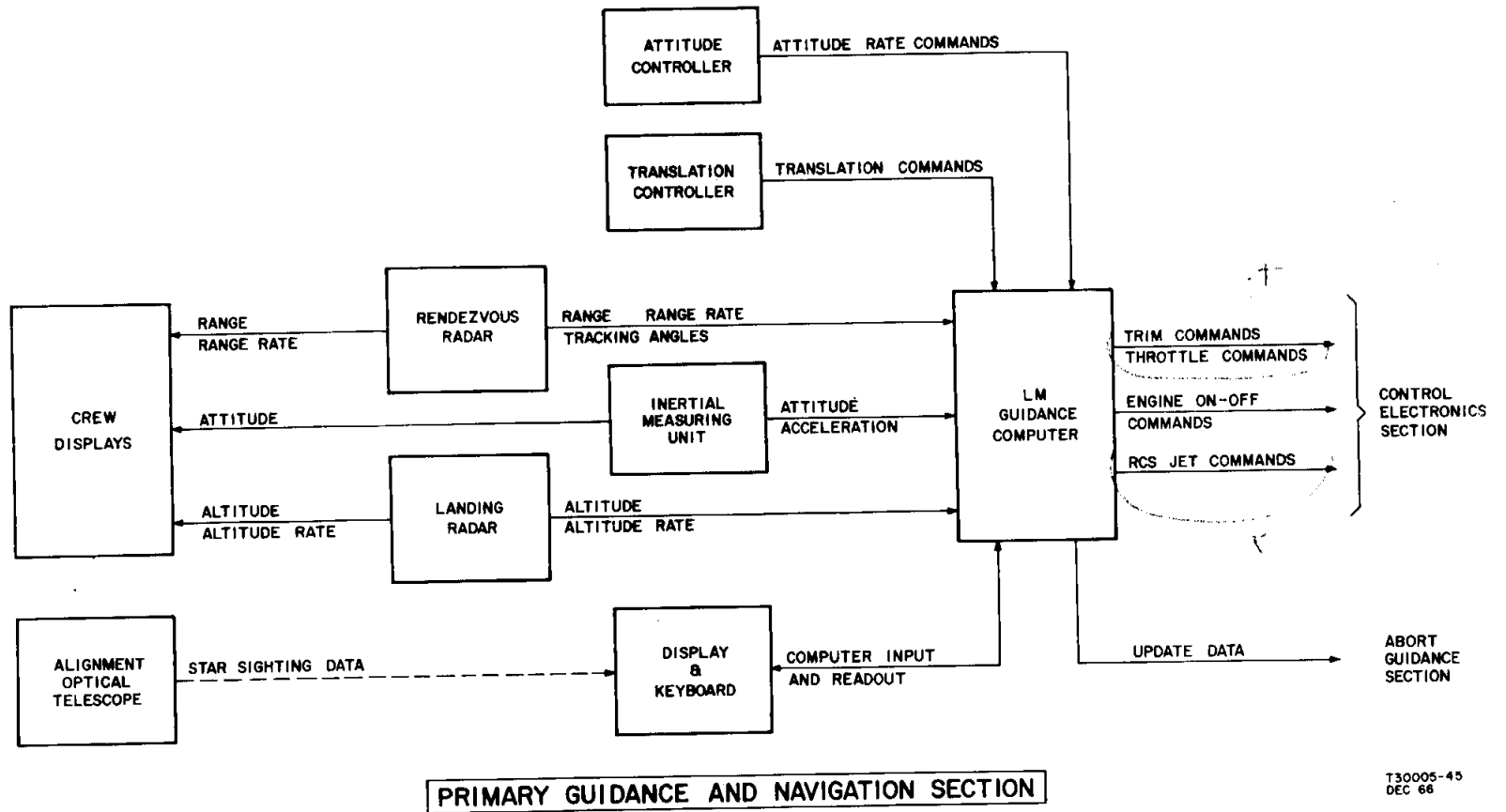
T30005-26  
 JUNE 66

ELECTROEXPLOSIVE DEVICES

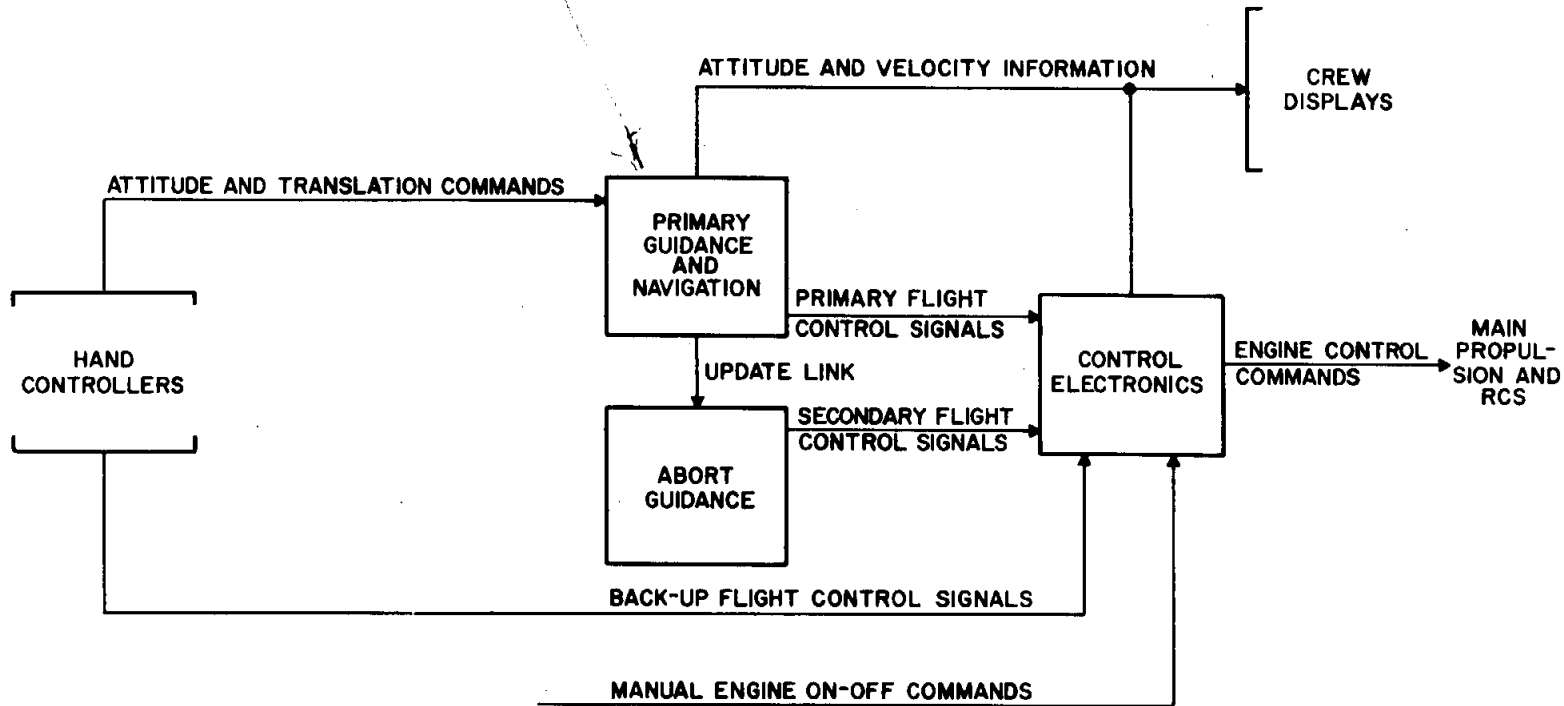


*Switch connected*

FUNCTIONAL BLOCK DIAGRAM  
ELECTROEXPLOSIVE SYSTEM



*can be manual control  
of rendezvous*

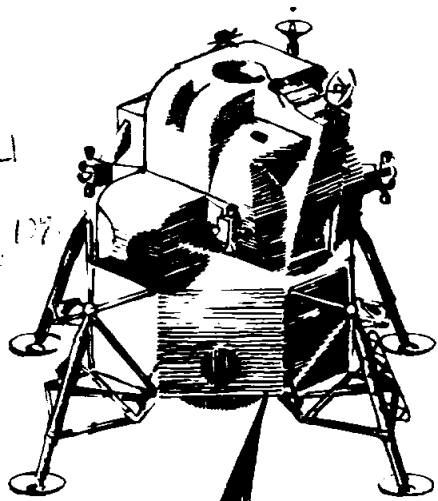


*Hand controllers  
provide manual control  
of rendezvous*

**LM GUIDANCE, NAVIGATION  
AND CONTROL SUBSYSTEM**



2000 ft. above  
 10 sec. by 10 sec. interval  
 8 radar beams at 100 ft. interval  
 How radar beam  
 swept out  
 10000 ft. 1 hr



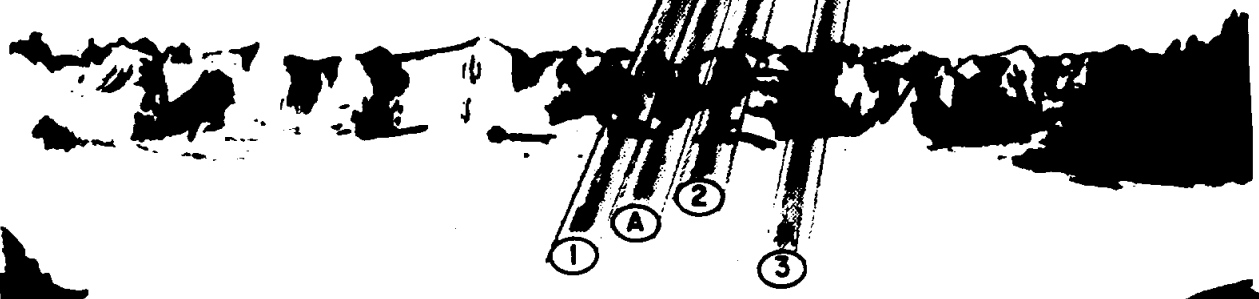
using radar  
 to determine  
 position  
 & velocity

ALTITUDE  $\rightarrow$  A  
 ALTITUDE RATE  
 (VERTICAL VELOCITY)

FORWARD (Z-AXIS) AND  
 LATERAL (Y-AXIS) VELOCITY

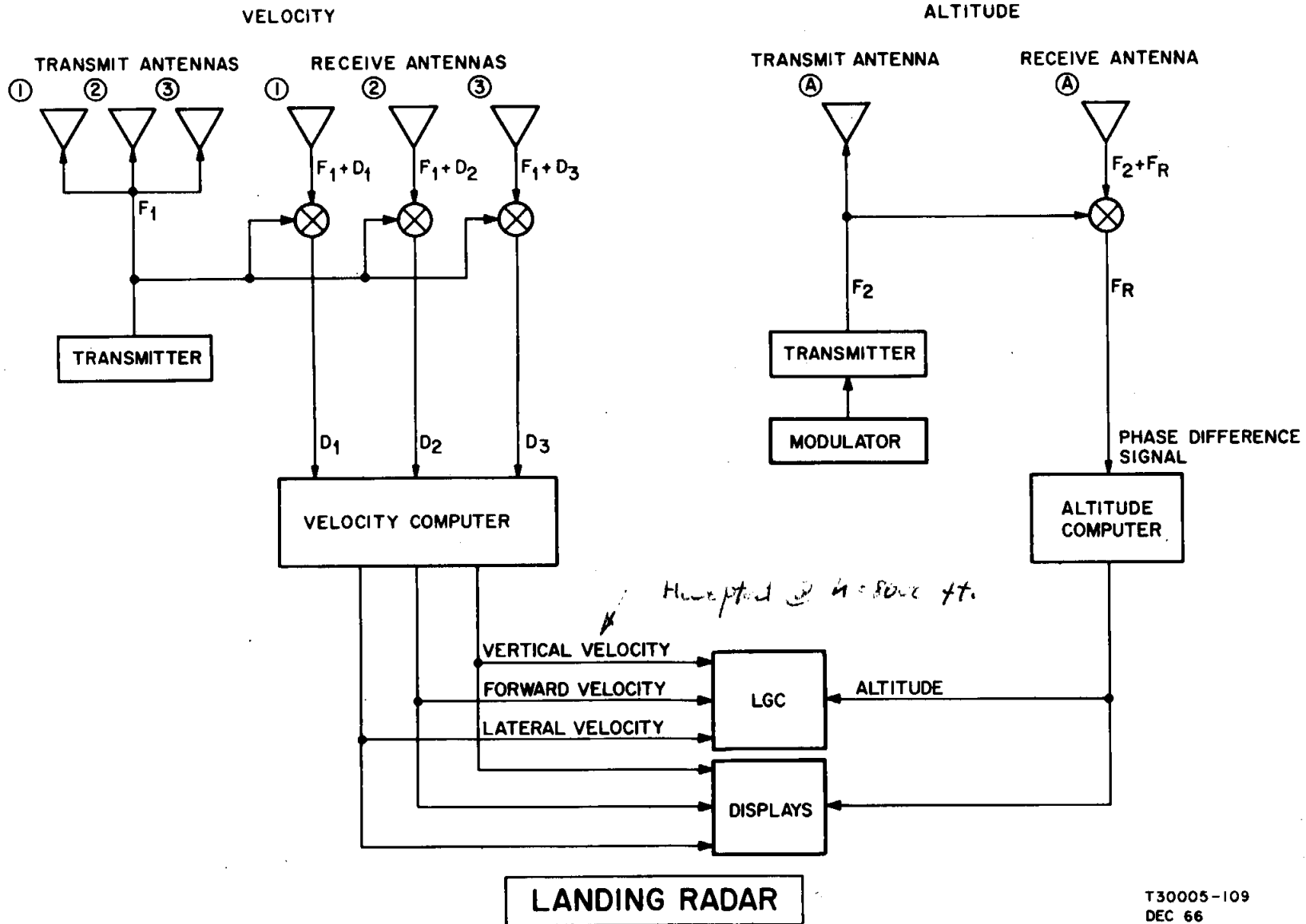
Uses Doppler  
 shift  
 returns  
 2 beams

200  $\rightarrow$  lateral  
 100  $\rightarrow$  vertical



**LANDING RADAR OPERATION**

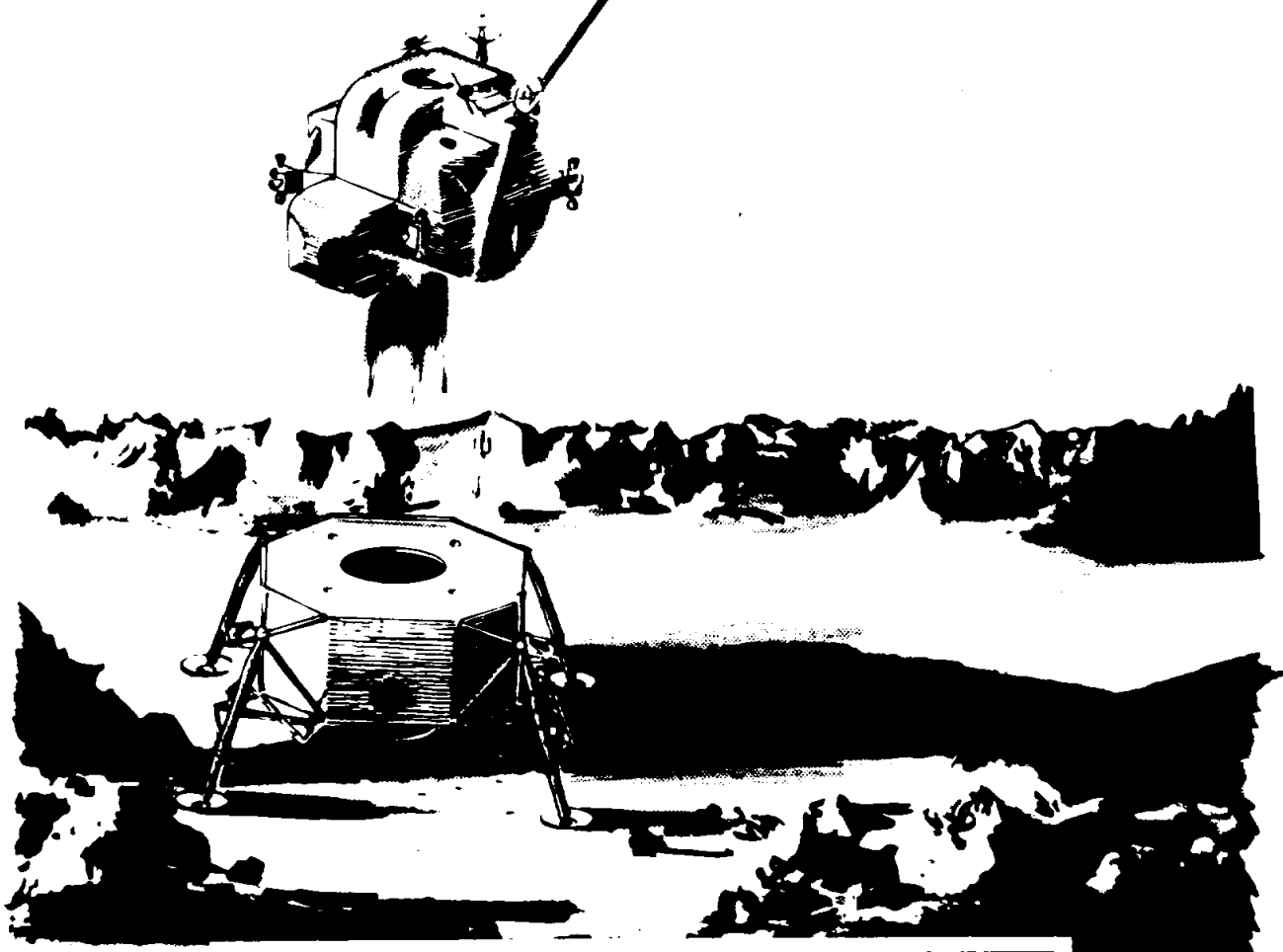
T 30005-104  
 APR 66



CMC - transponder sends  
back constant reply  
to the signal sent out  
by CMC RA

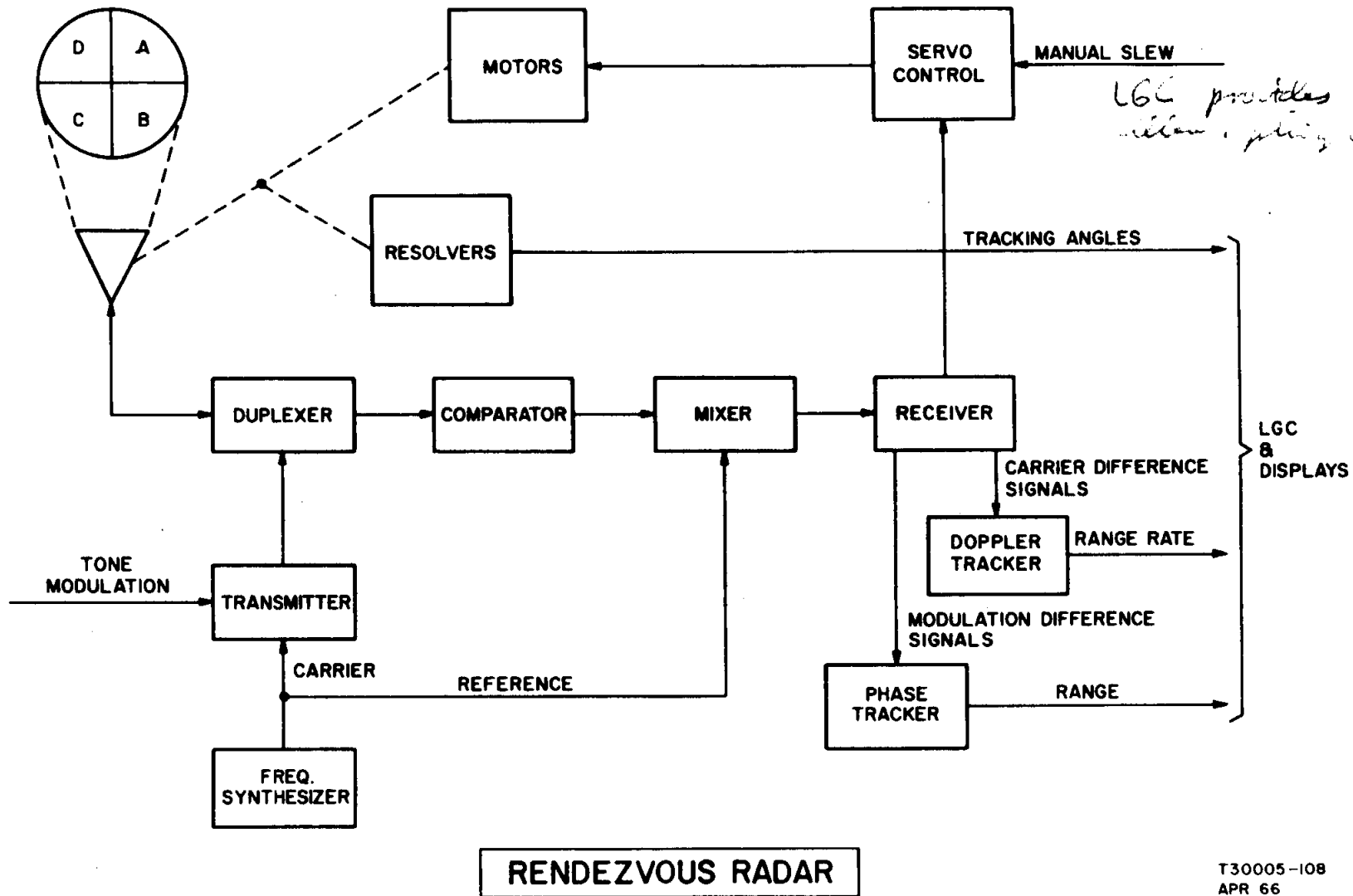
Range & Range  
rate up to 400 miles

RANGE  
RANGE RATE  
TRACKING ANGLES



**RENDEZVOUS RADAR OPERATION**

T 30005-103  
APR 66

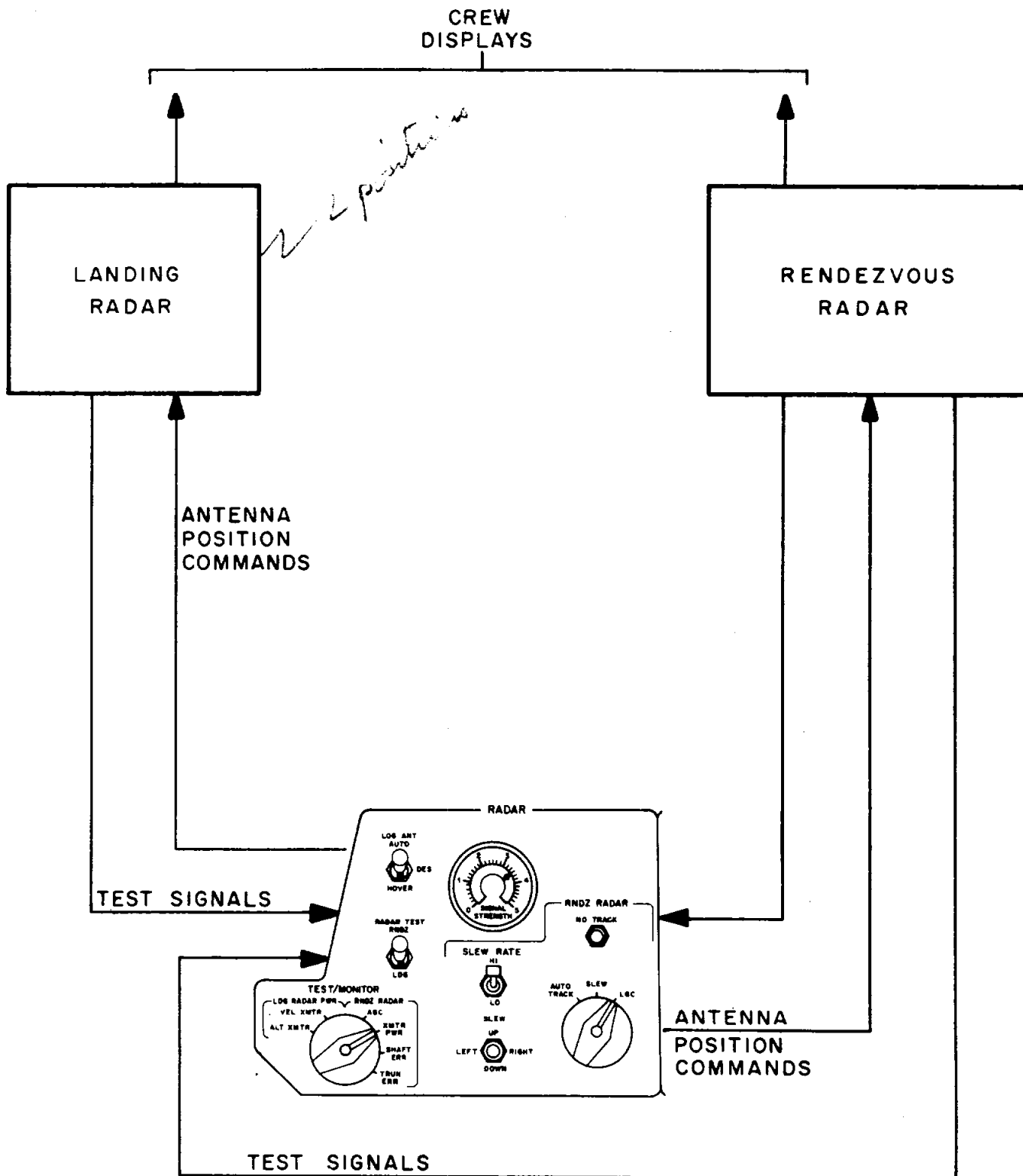


*LGC provides initial  
align. giving some time*

LGC  
&  
DISPLAYS

T30005-108  
APR 66

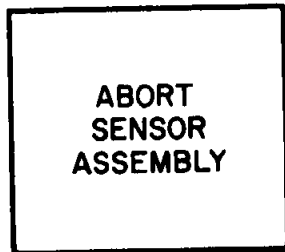
*And also  
HTL in p.h  
If want to investigate on this, all + reference  
in the + the no will be good*



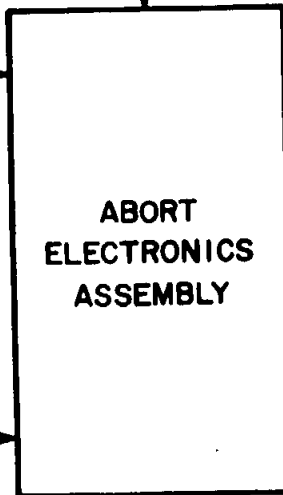
**PANEL III  
RADAR CONTROL**

PRIMARY  
GUIDANCE &  
NAVIGATION  
SECTION

UPDATE DATA



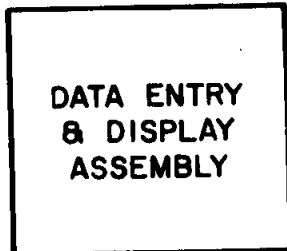
GYRO DATA  
ACCELERATION



ATTITUDE ERROR  
SIGNALS

CONTROL  
ELECTRONICS  
SECTION

ENGINE ON-OFF  
COMMANDS

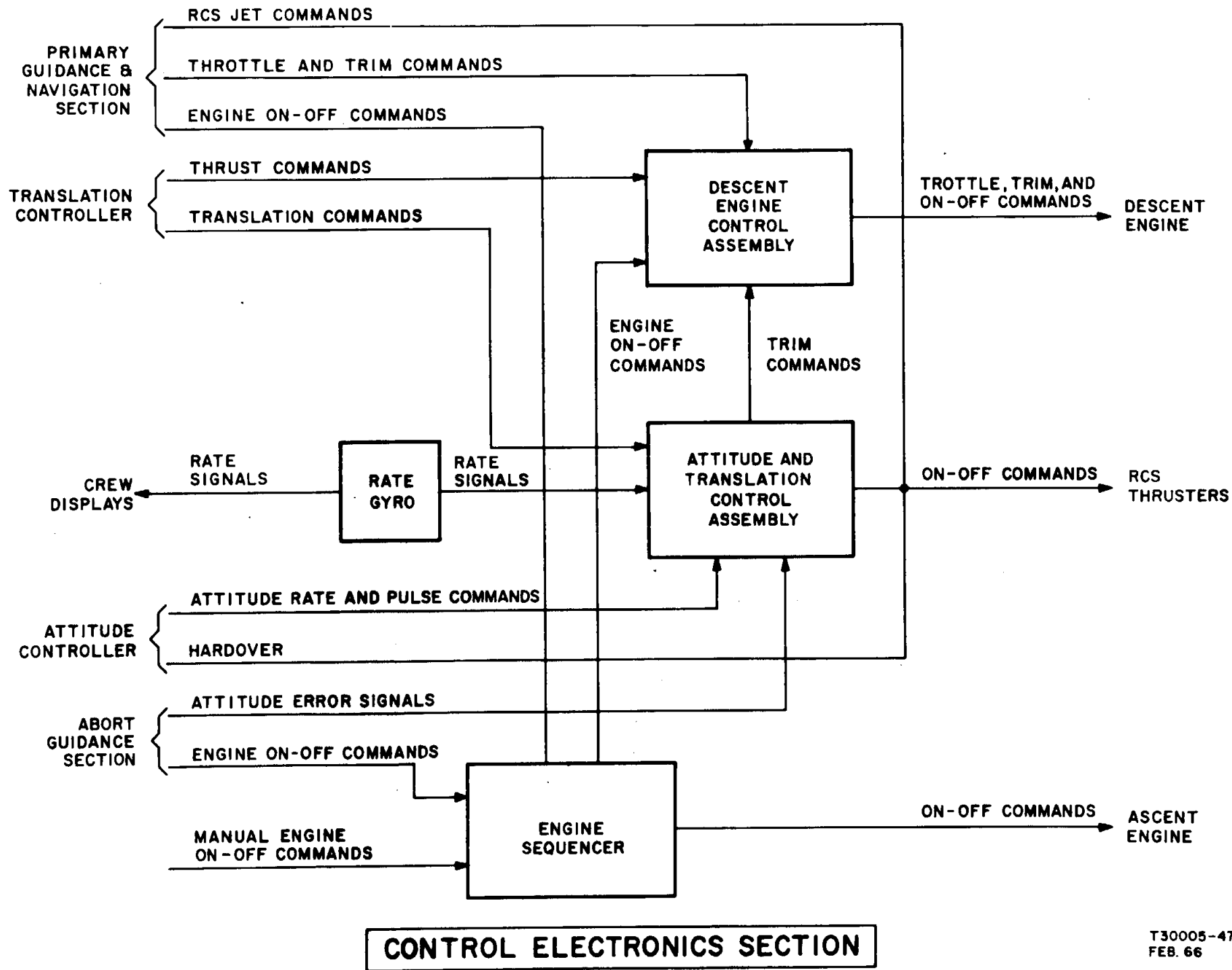


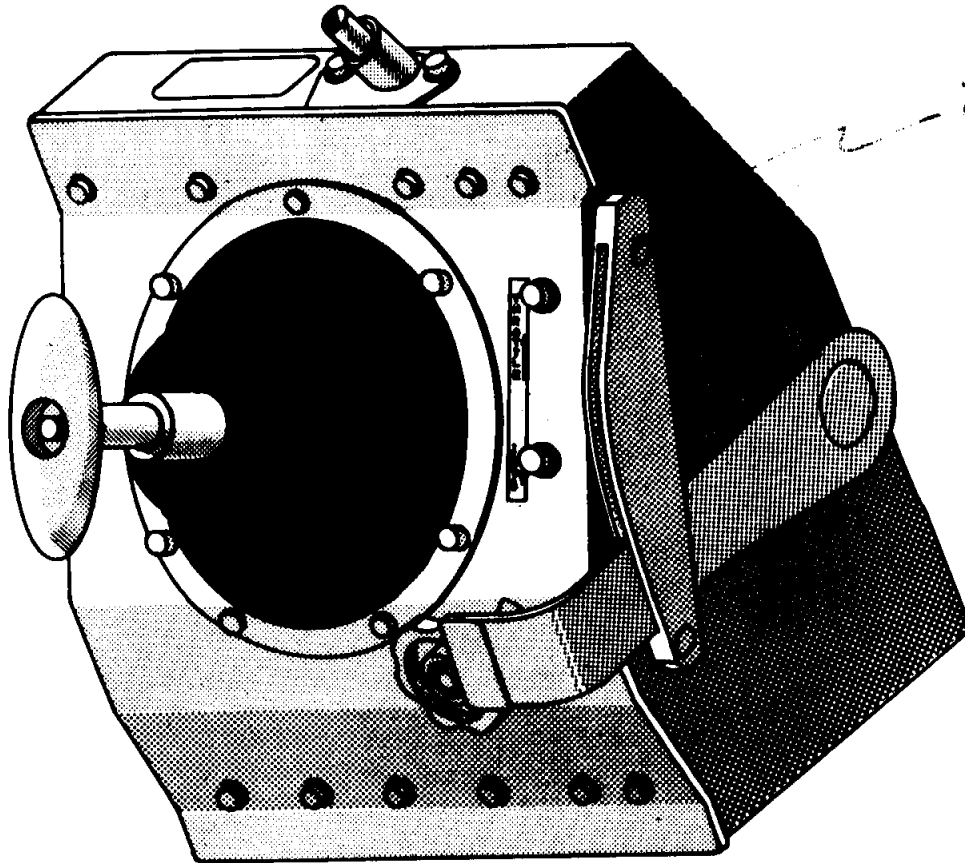
COMPUTER INPUT  
& READOUT

**ABORT GUIDANCE SECTION**

T30005-46  
FEB 66

Descent engine may be used for an abort starting descent.  
In this case the engine fuel system





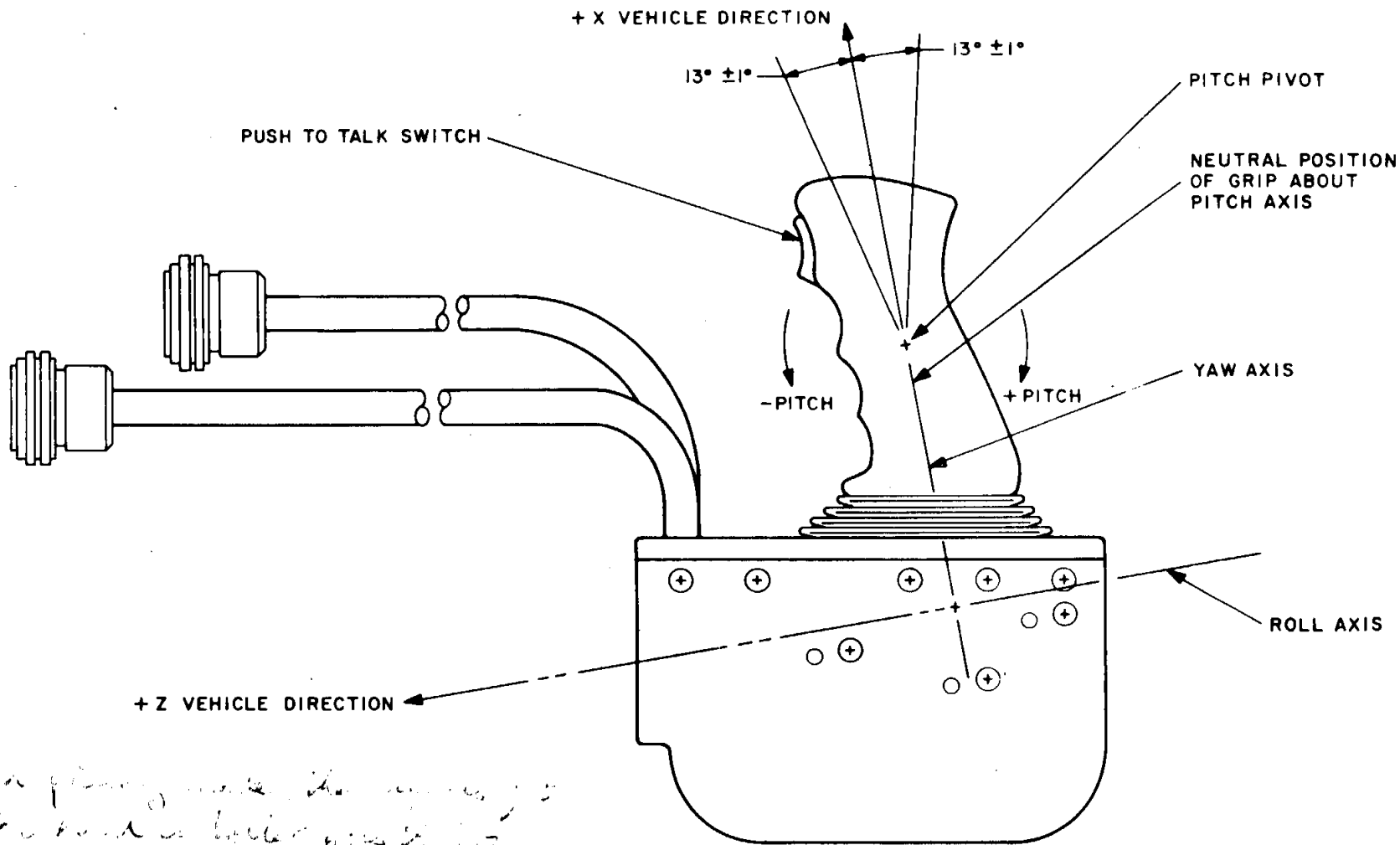
in this position  
 of  $\gamma$  axis  $\uparrow$   
 in the direction  
 of  $\gamma$  axis

THRUST TRANSLATION CONTROLLER ASSEMBLY

T30005-51  
 DEC 66

$\uparrow$  X  $\uparrow$   
 $\uparrow$  Z  $\uparrow$   
 $\uparrow$  Y  $\uparrow$



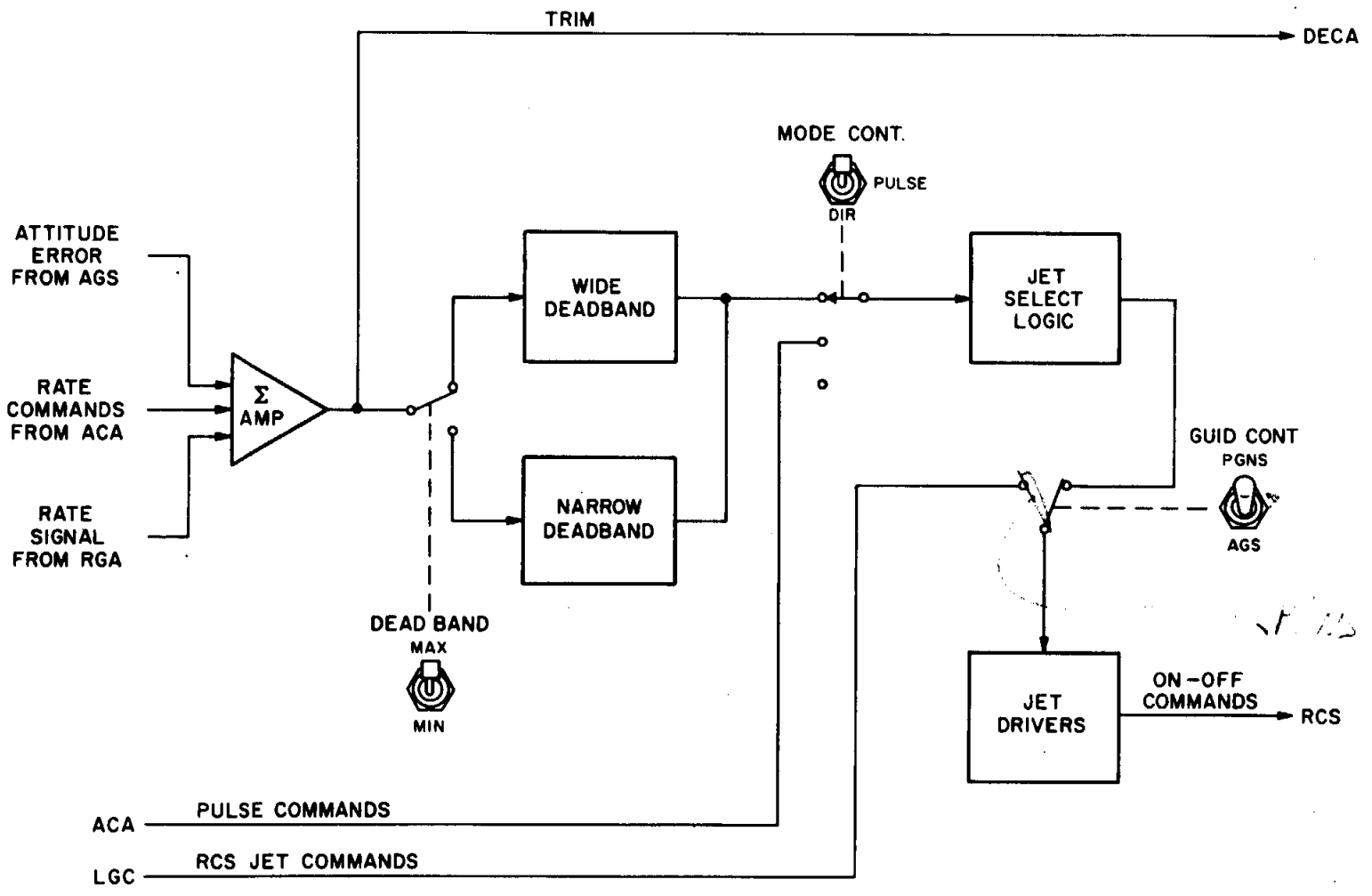


*In planning make the...  
 the hand is held...  
 1.5 ft. H2O...  
 low...  
 better...*

**ACA MANIPULATION**

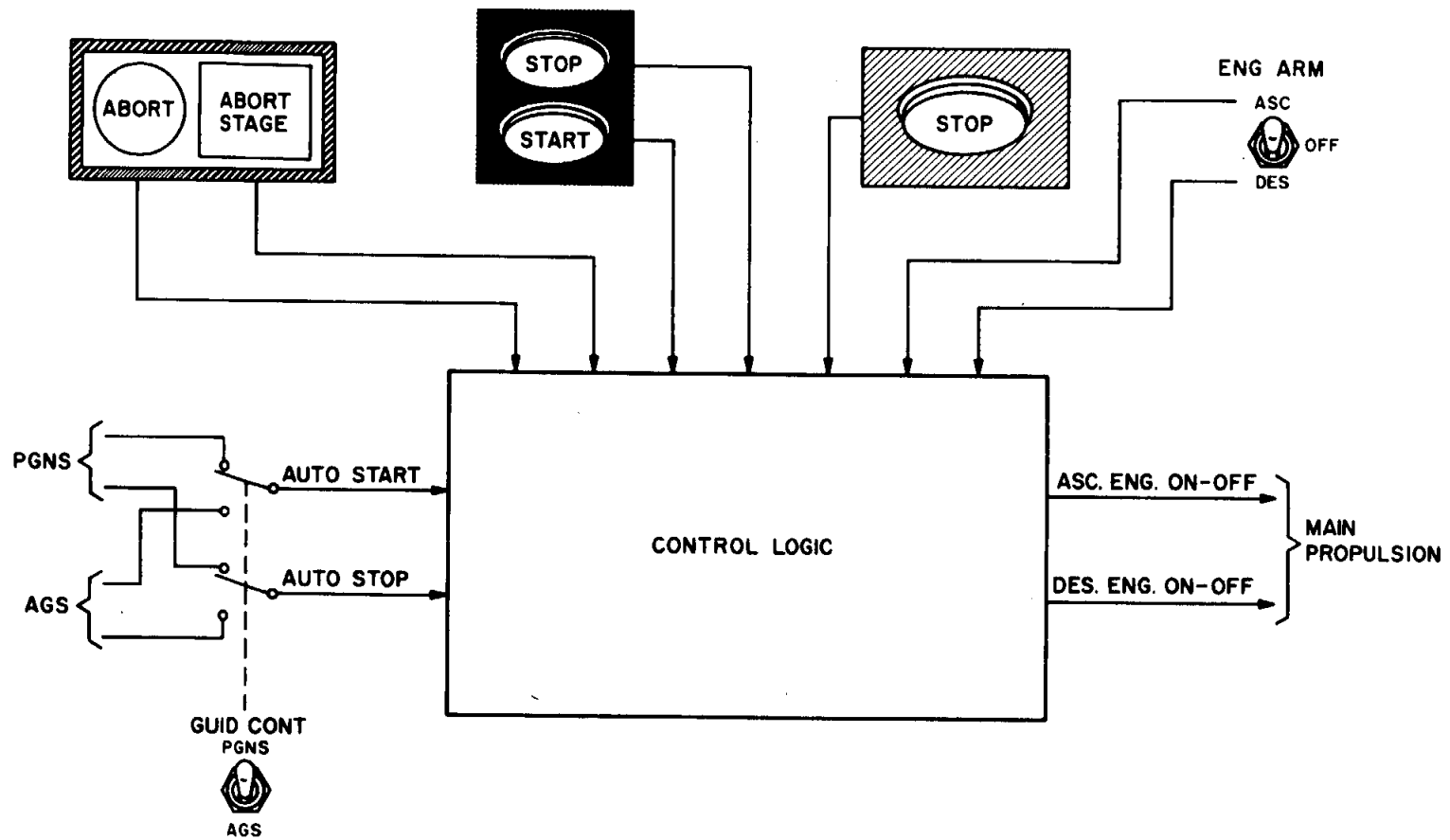
*Att. to New Home*

T30005-52  
 DEC 66



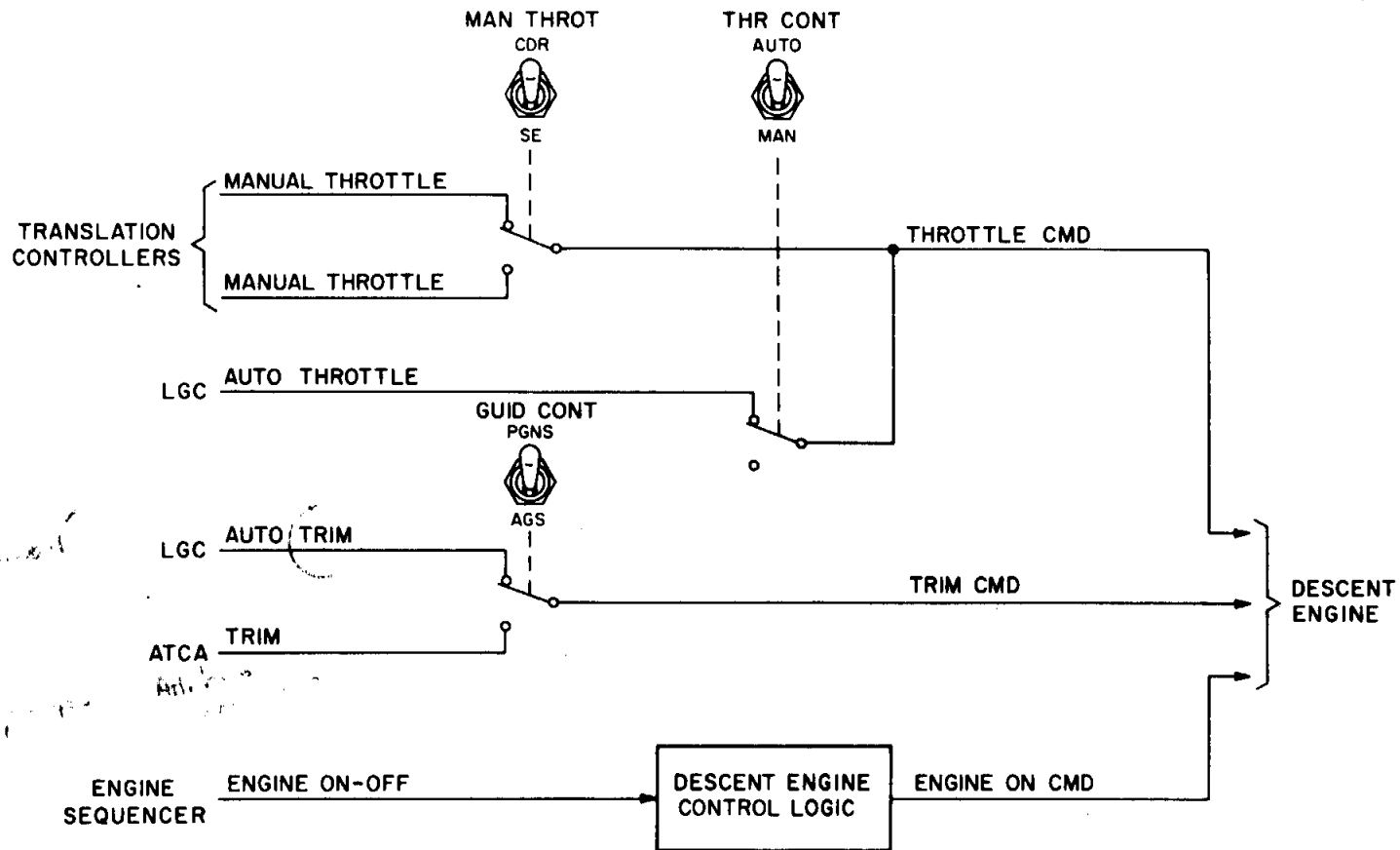
**ATTITUDE AND TRANSLATION CONTROL ASSEMBLY**

T30005-107  
APR 66



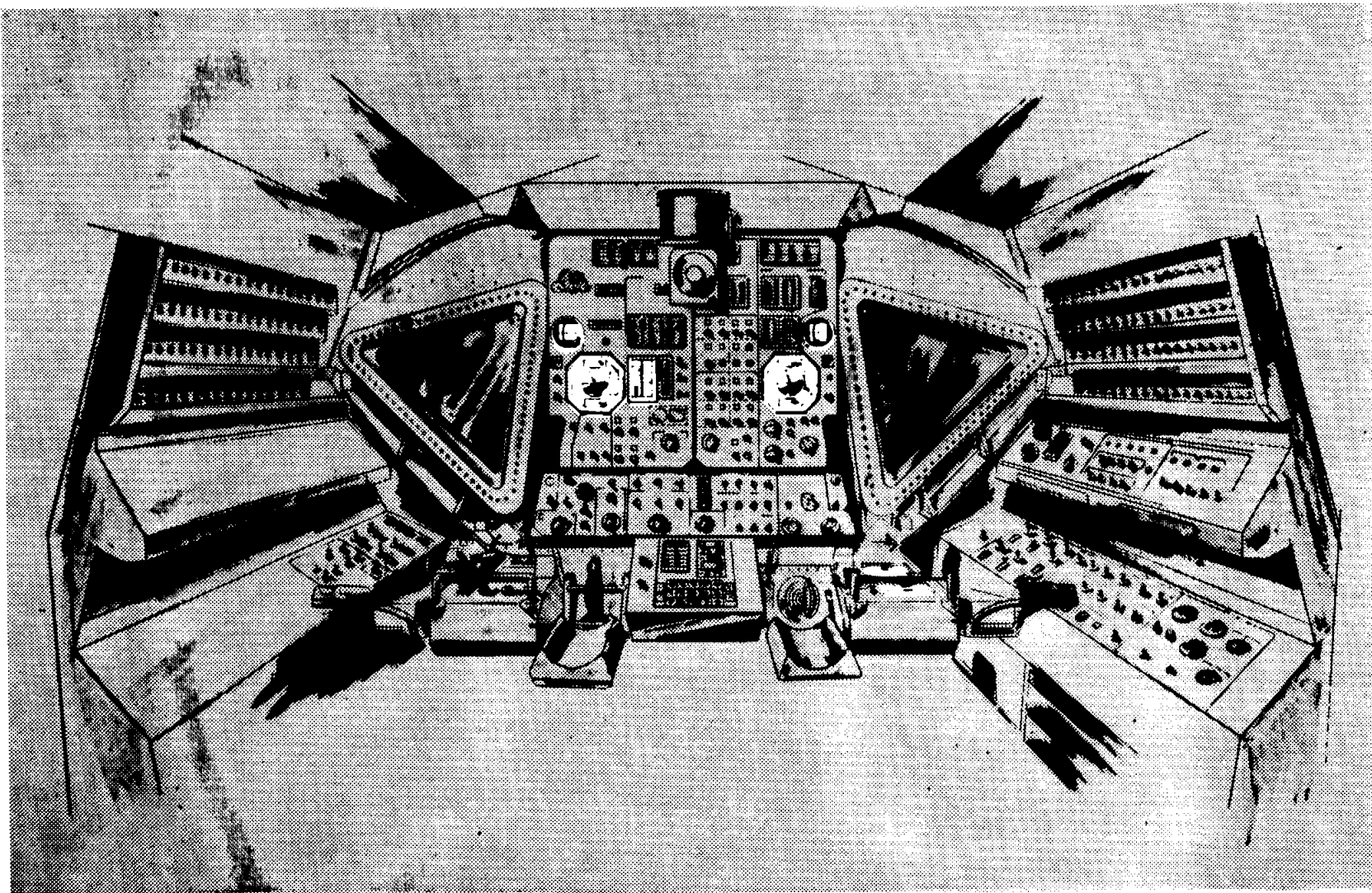
**ENGINE SEQUENCER**

T30005-105  
APR 66



DESCENT ENGINE CONTROL ASSEMBLY

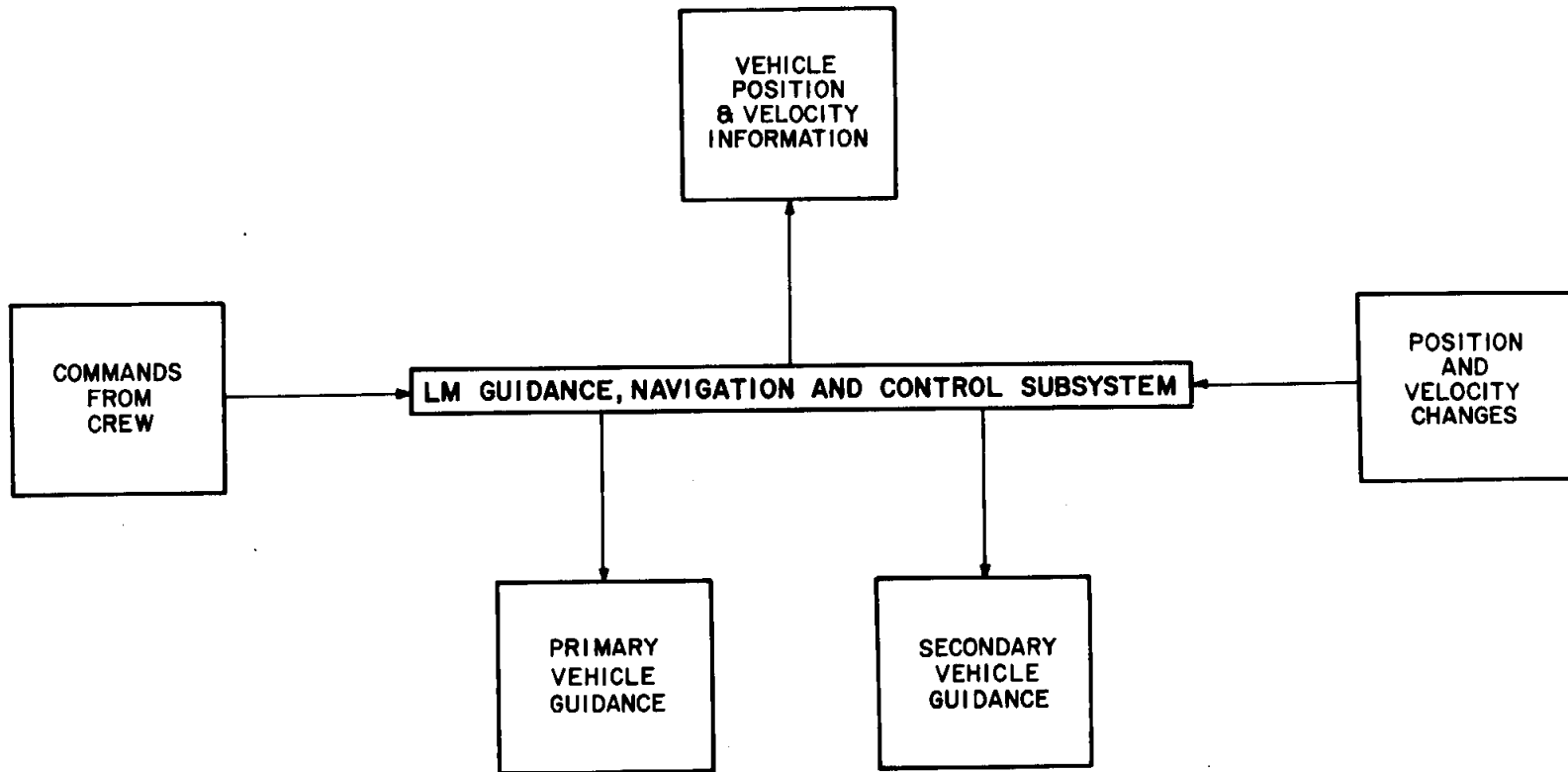
T30005-106  
APR 66



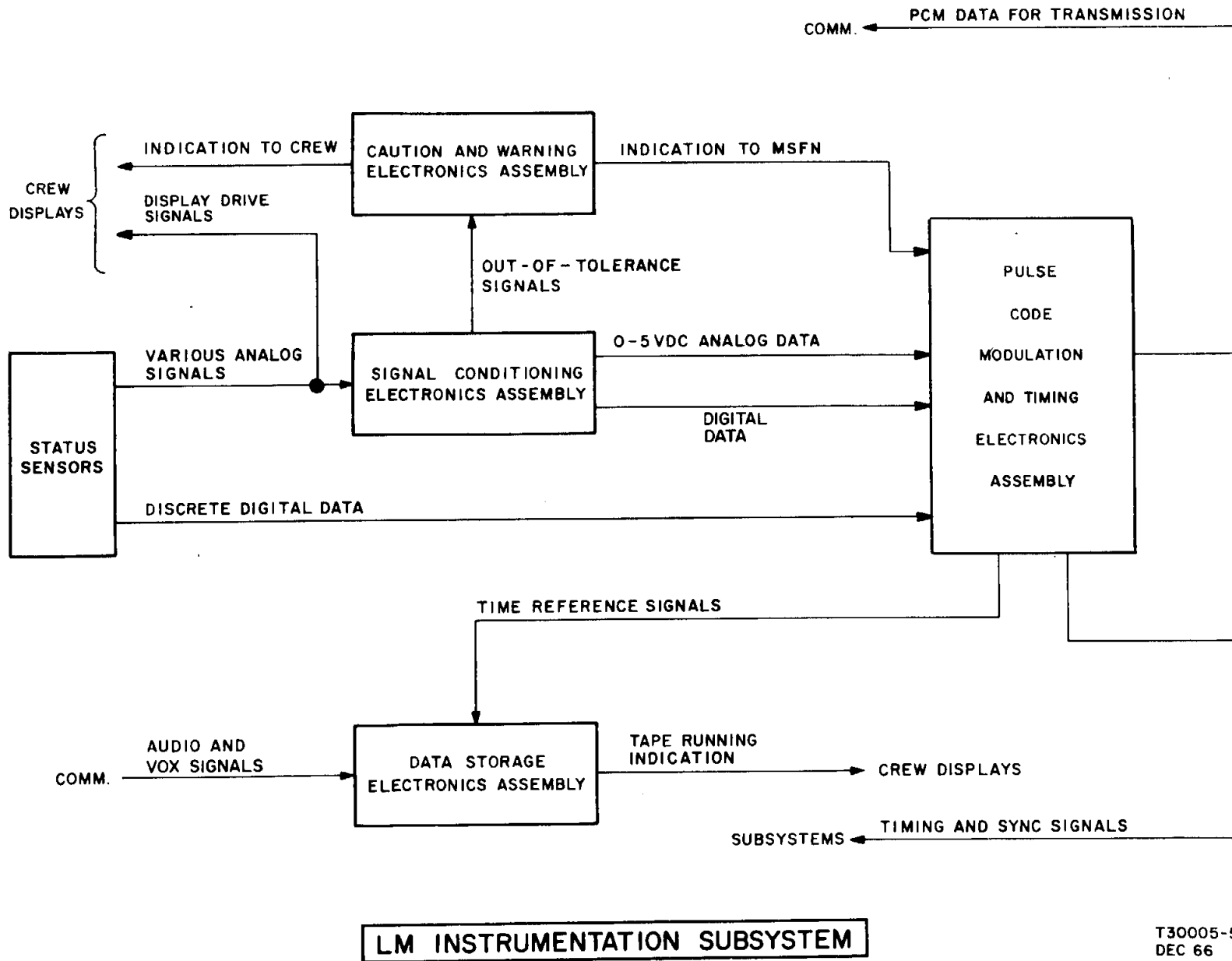
*White noise*

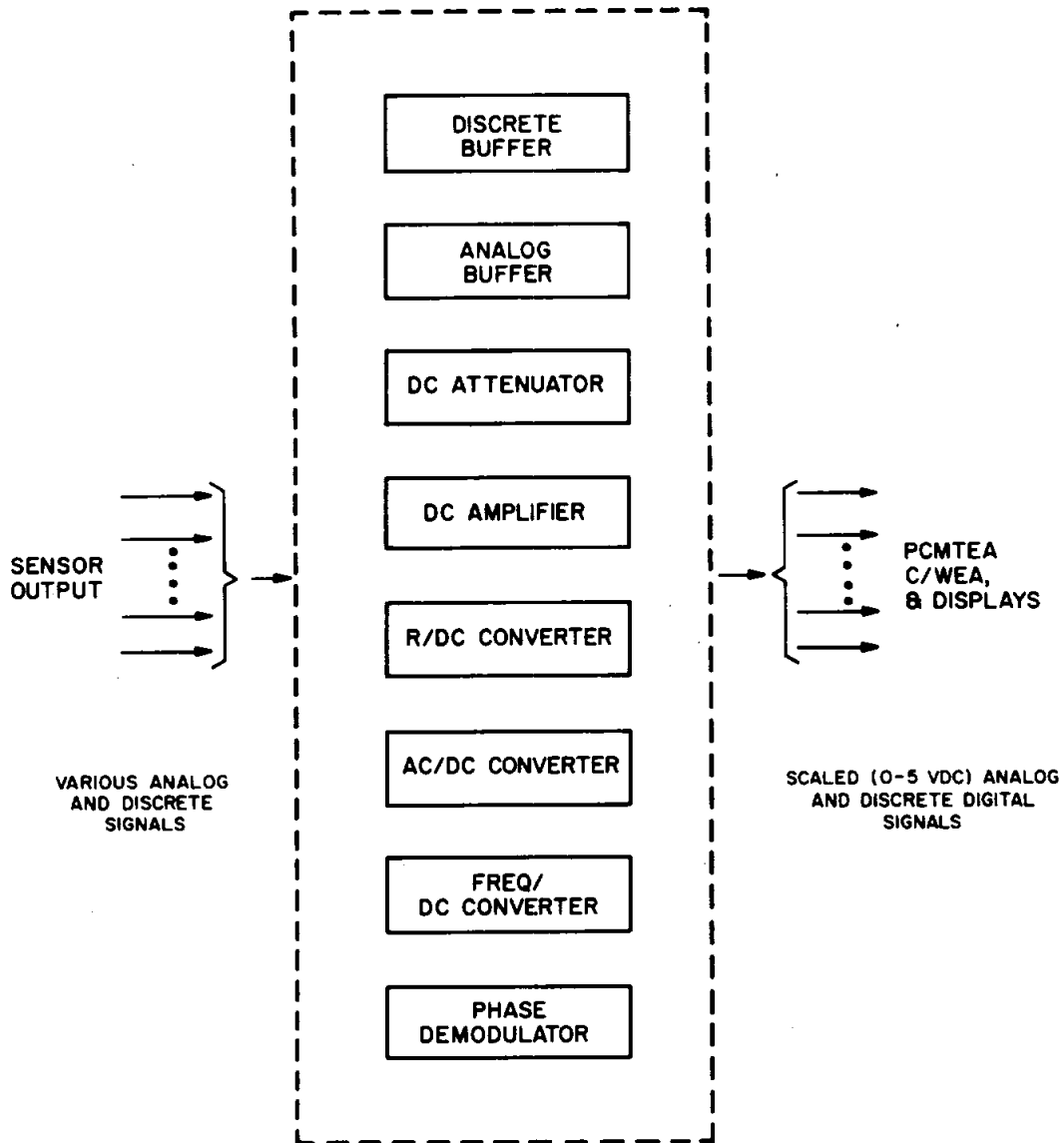
**G.N. AND C. ASSOCIATED DISPLAYS**

T 30005-50  
APR 66



T30005-43  
DEC 66

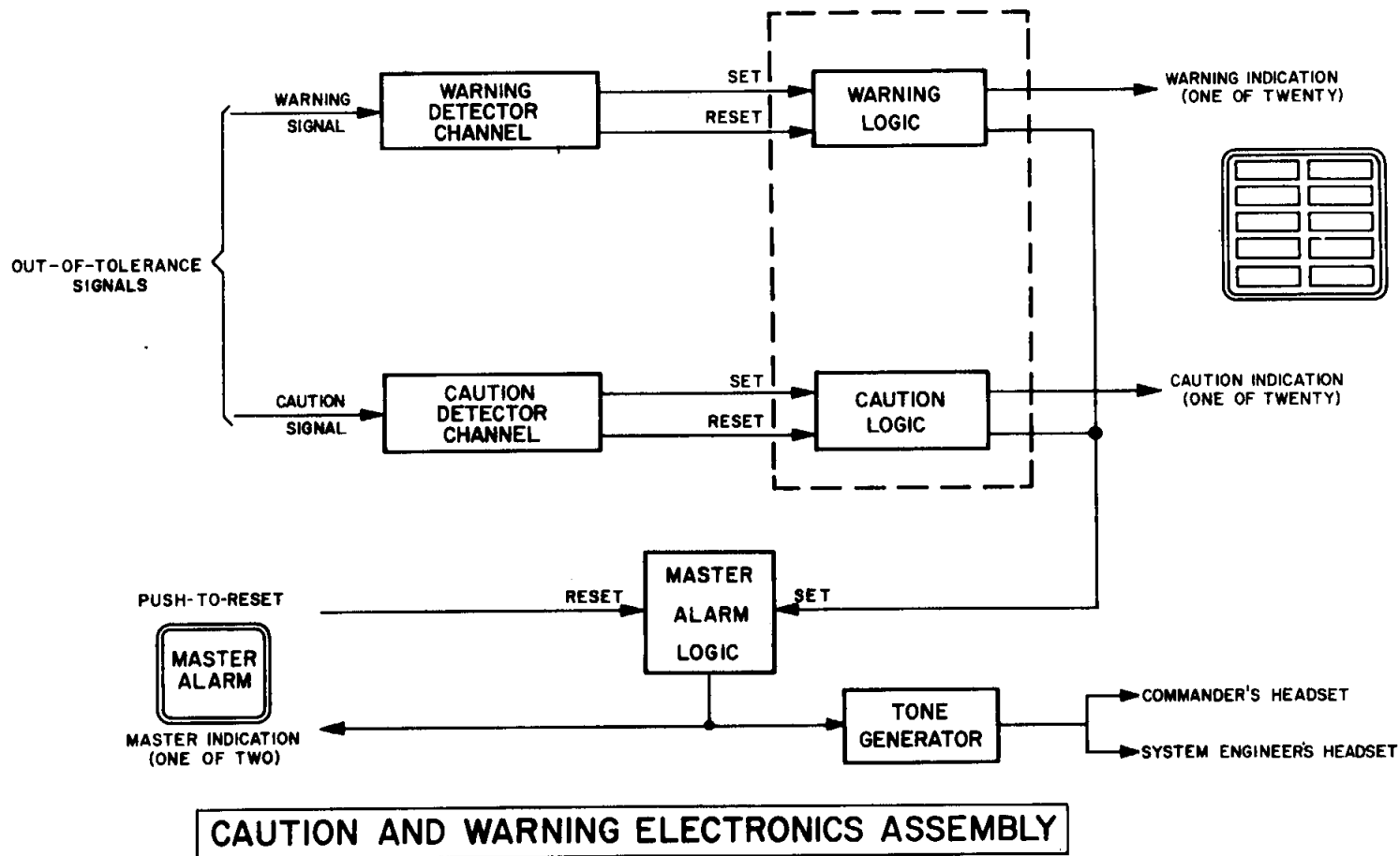


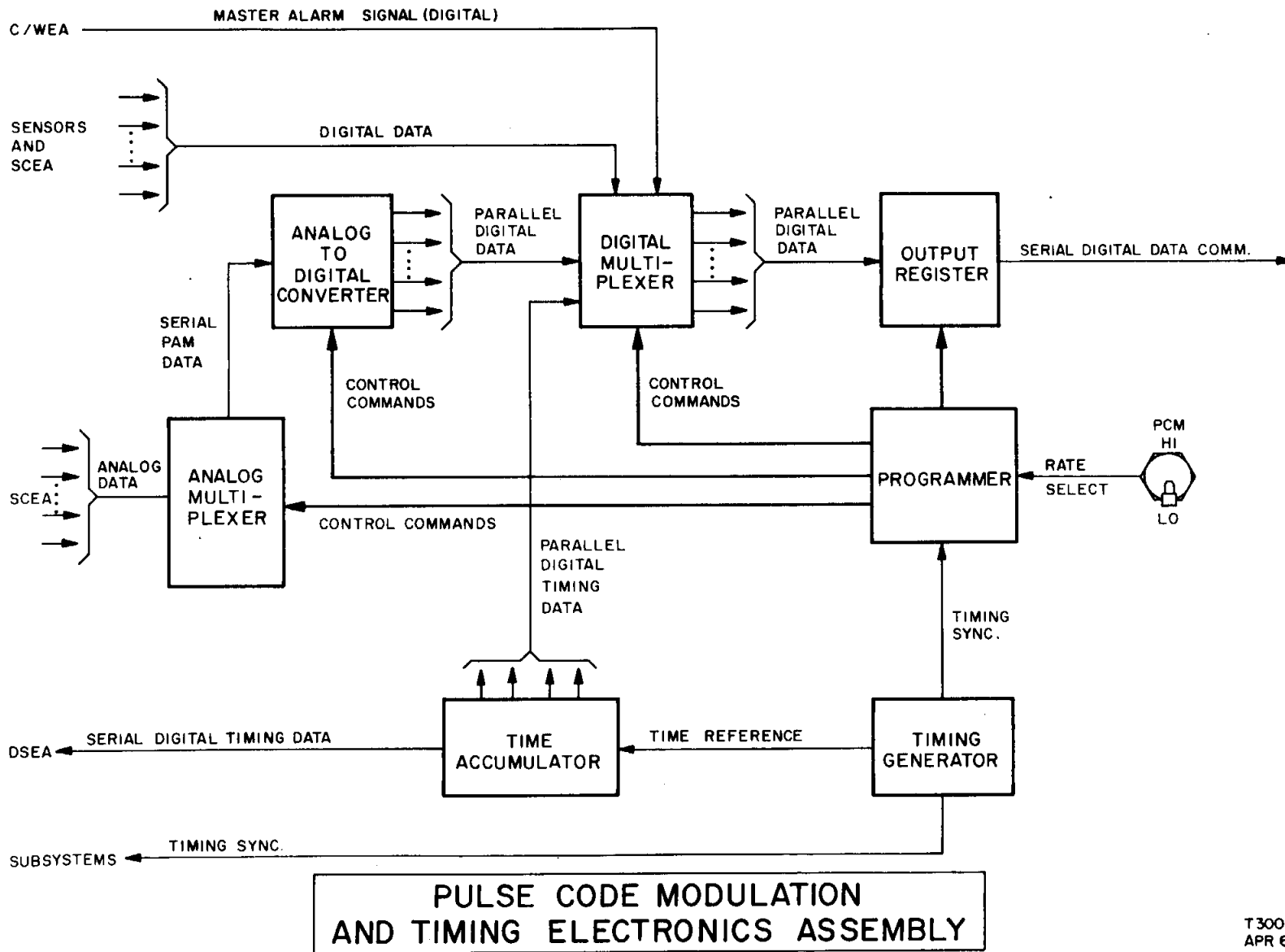


**SIGNAL CONDITIONING ELECTRONICS ASSEMBLY**

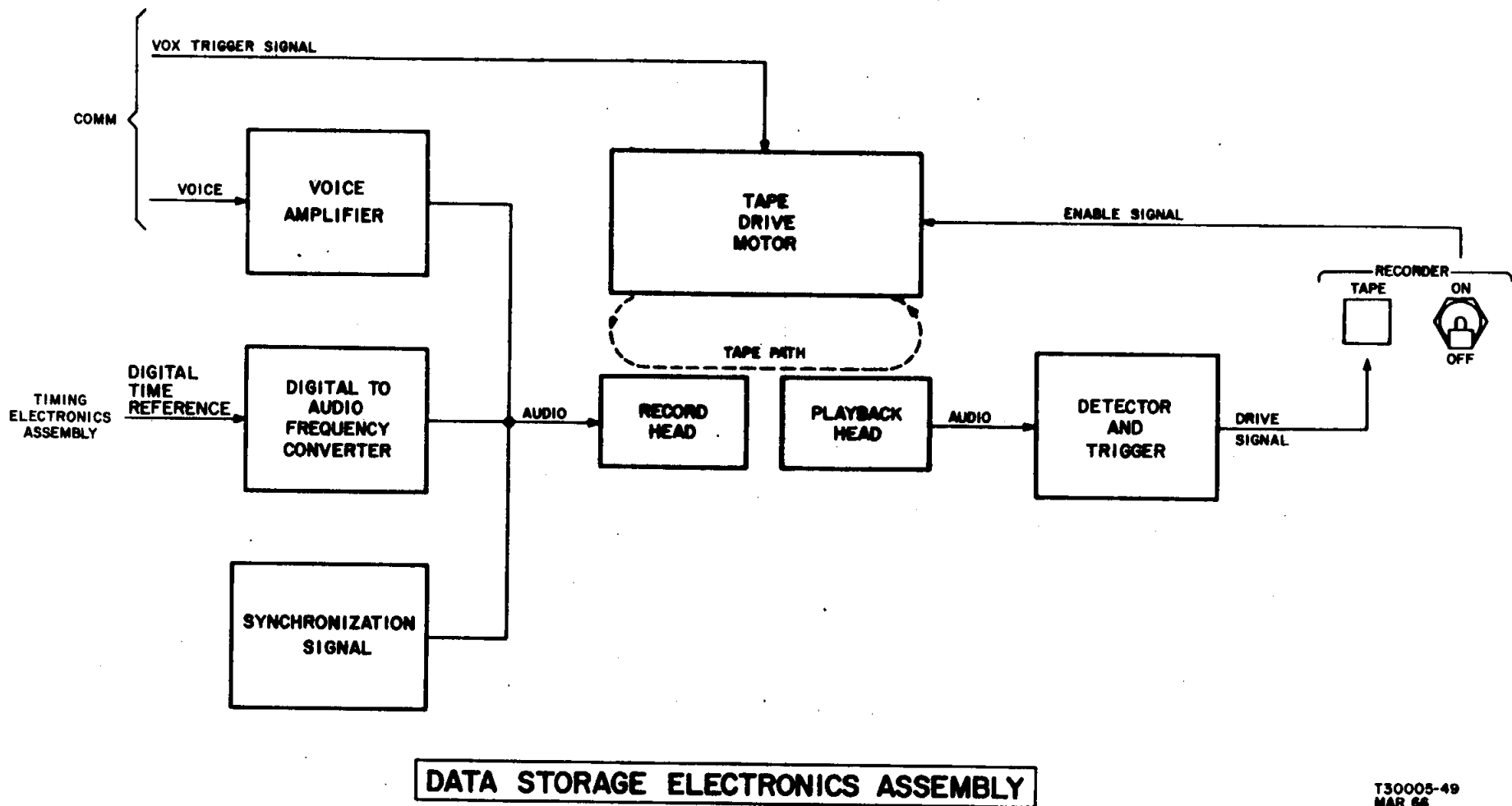
T30005-59  
MAR 66



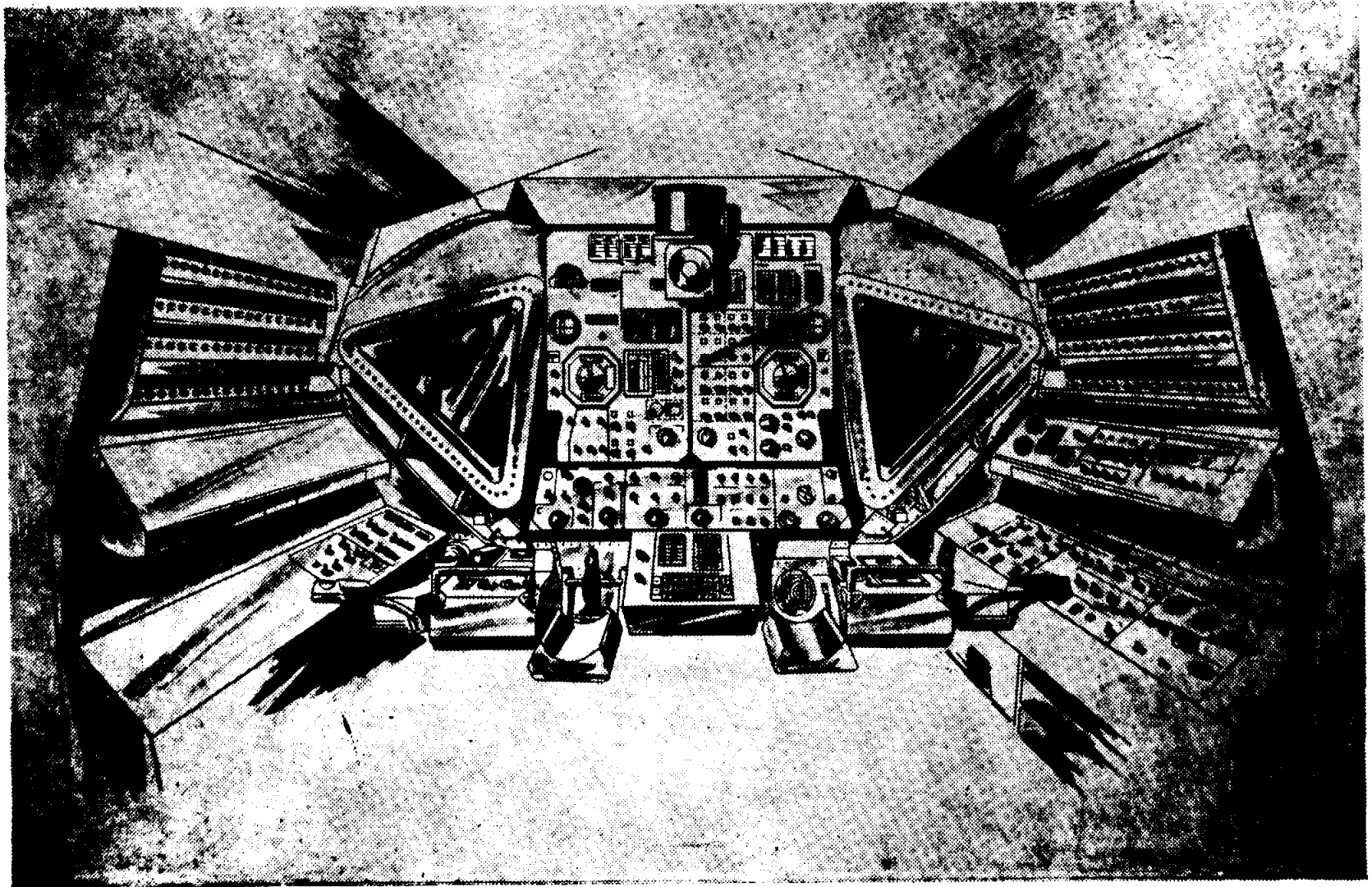




T 30005-58  
APR 66

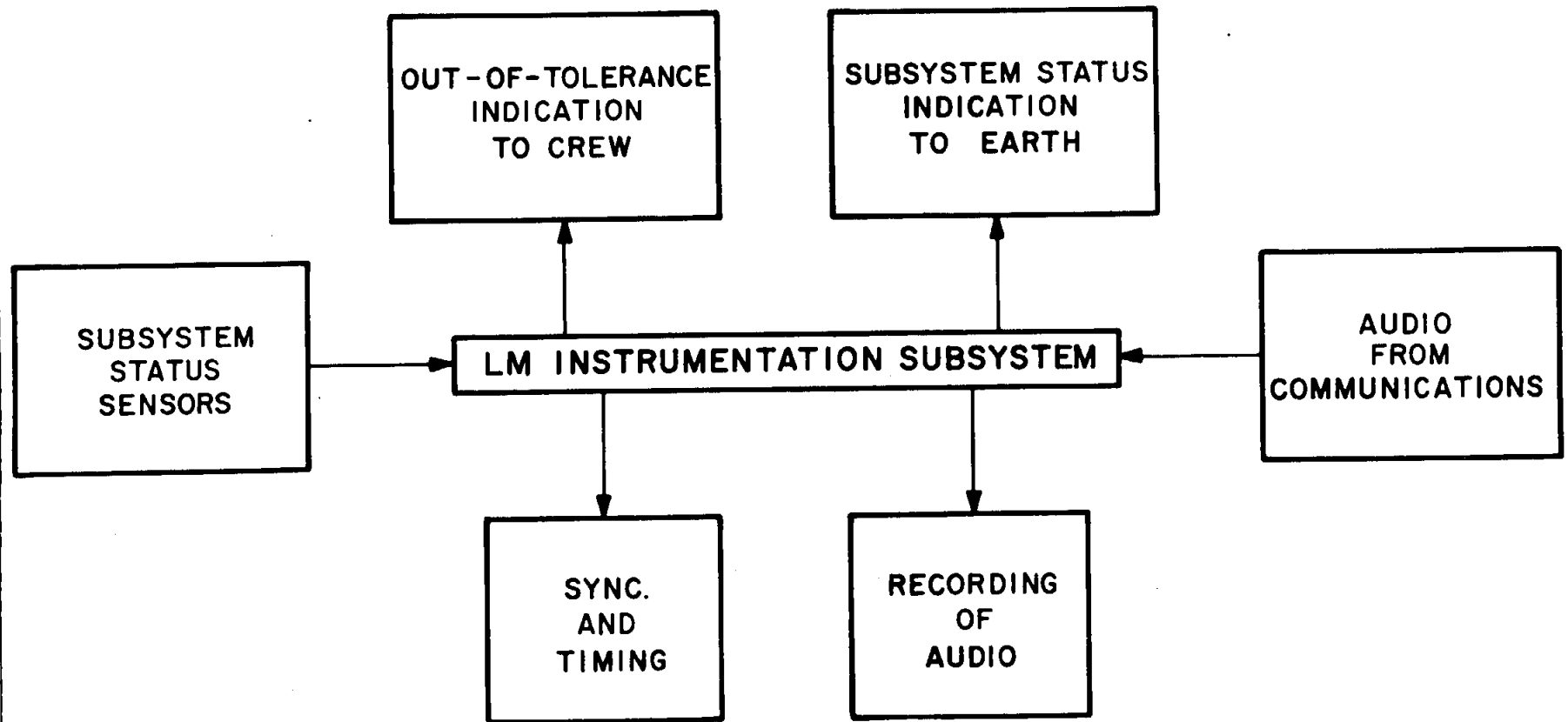


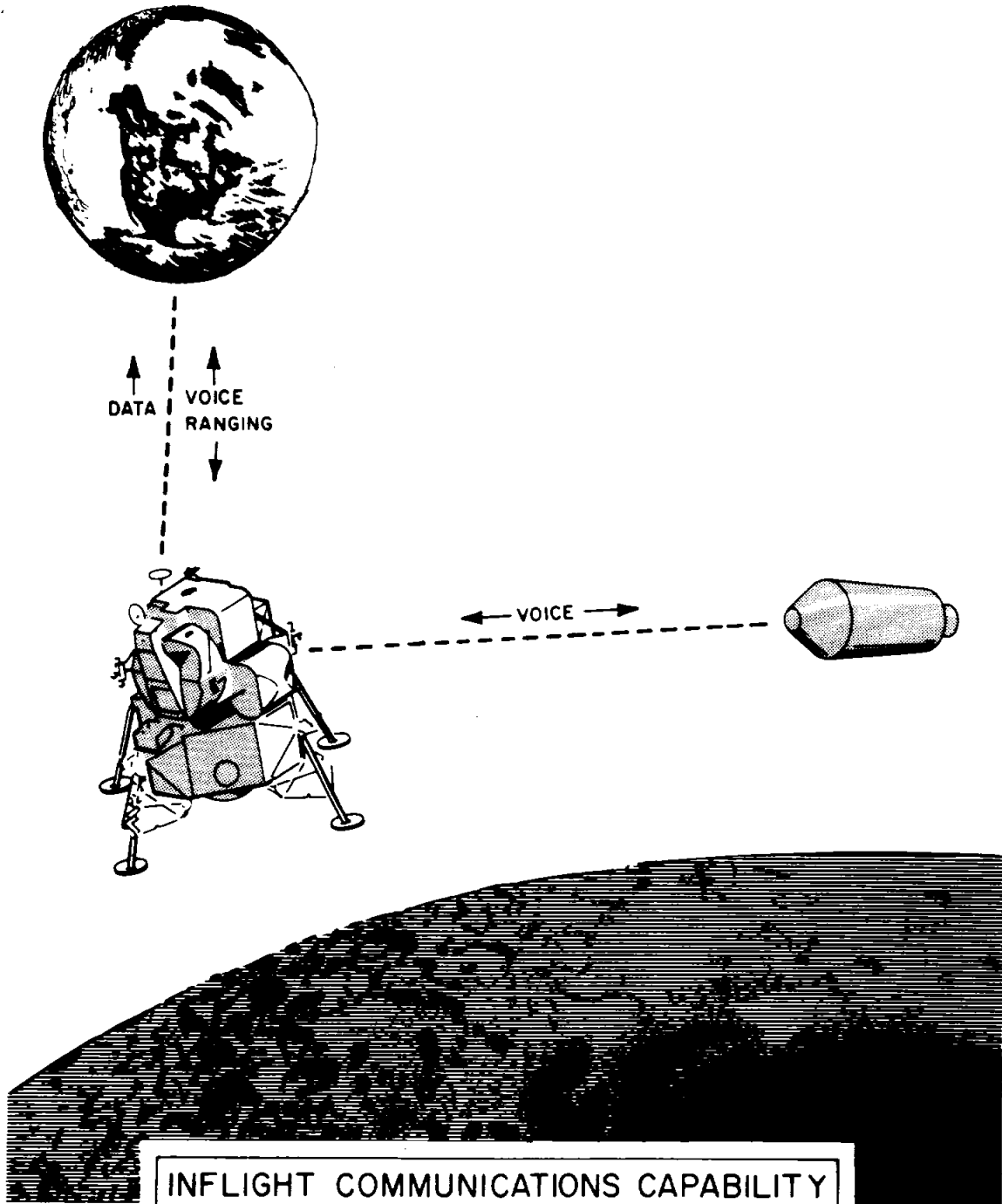
T30005-49  
MAR 66



**INSTRUMENTATION ASSOCIATED DISPLAYS**

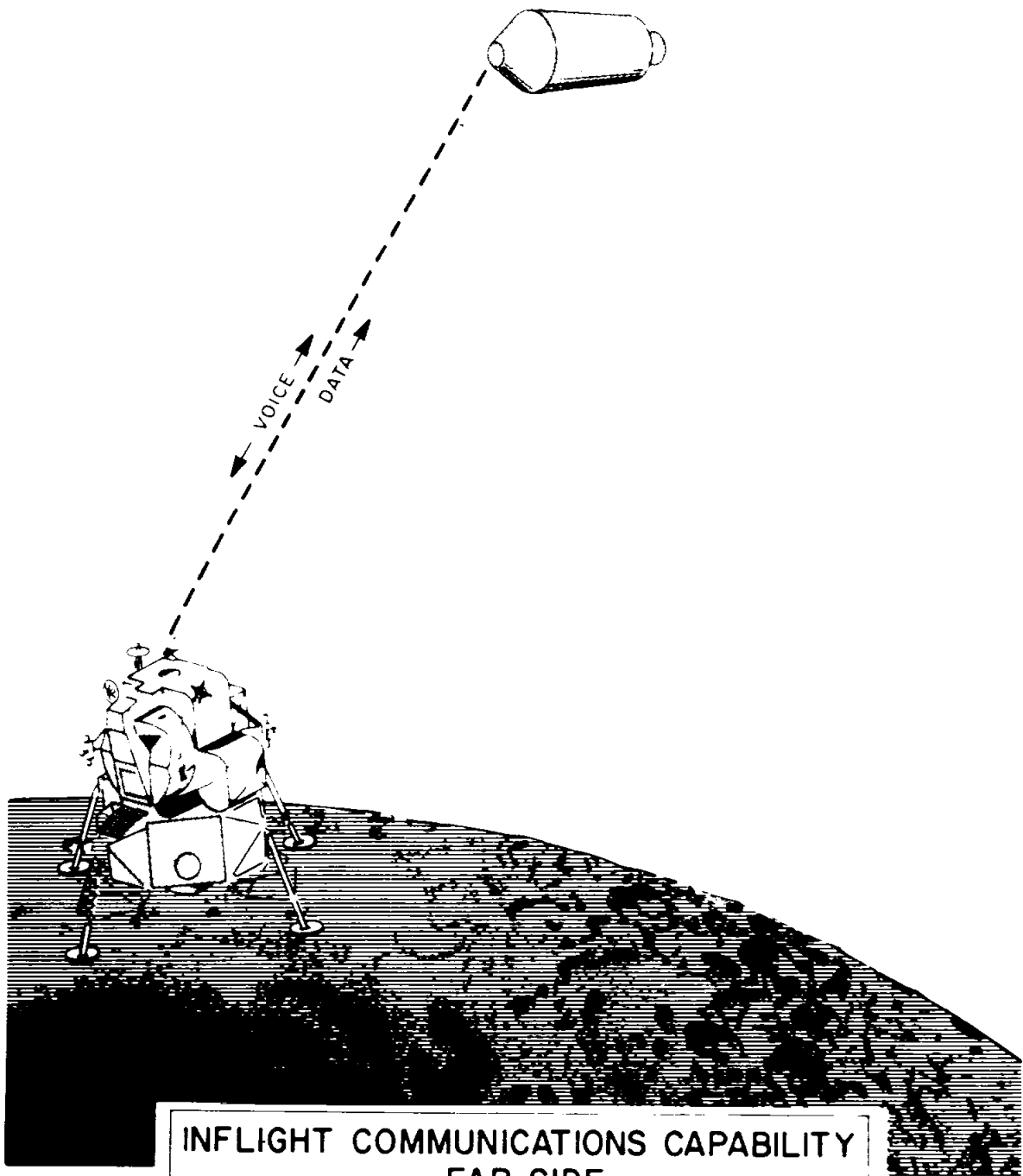
T30005-112  
APR 66





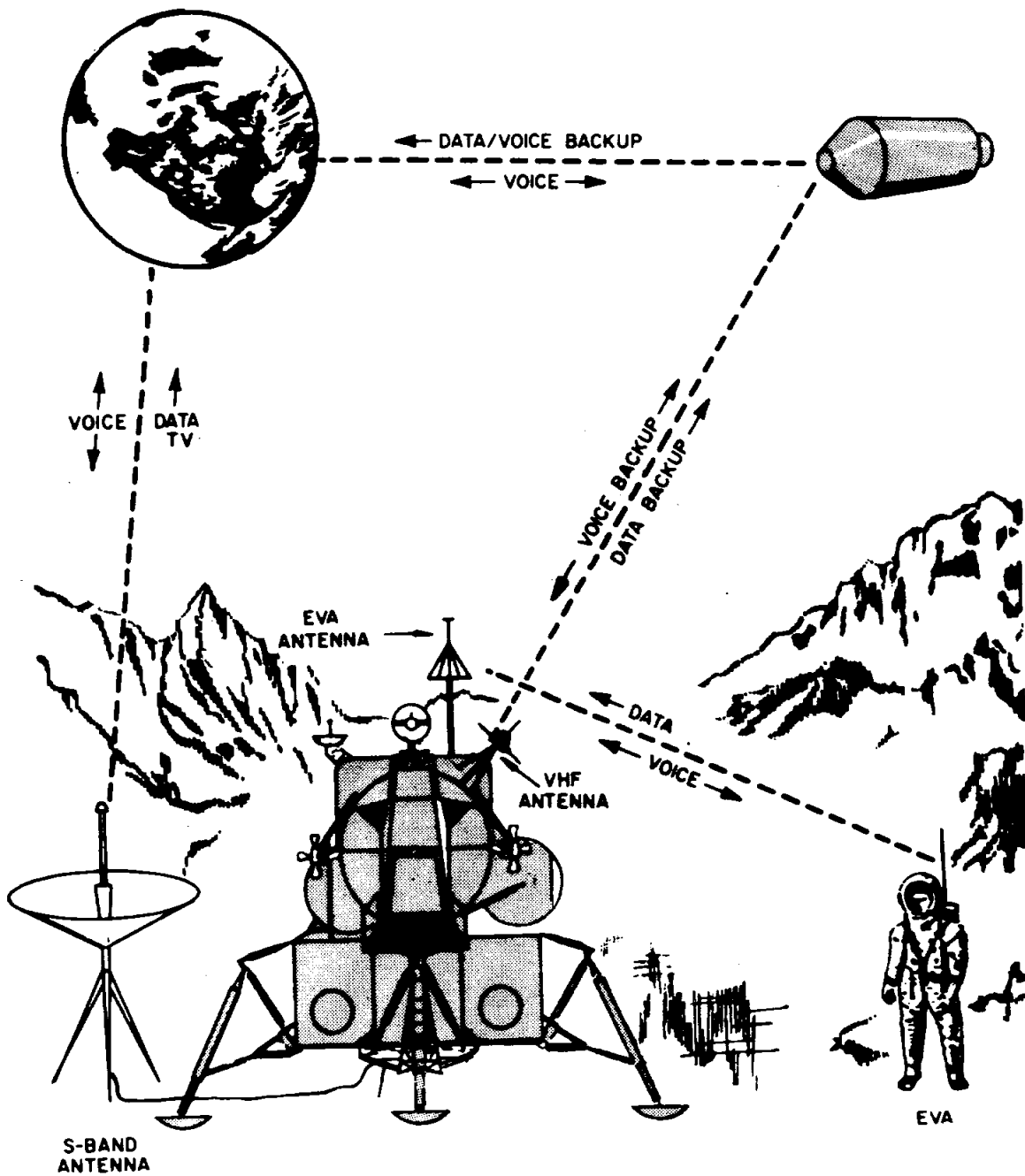
INFLIGHT COMMUNICATIONS CAPABILITY  
EARTH SIDE

T30005-38  
FEB 66



INFLIGHT COMMUNICATIONS CAPABILITY  
FAR SIDE

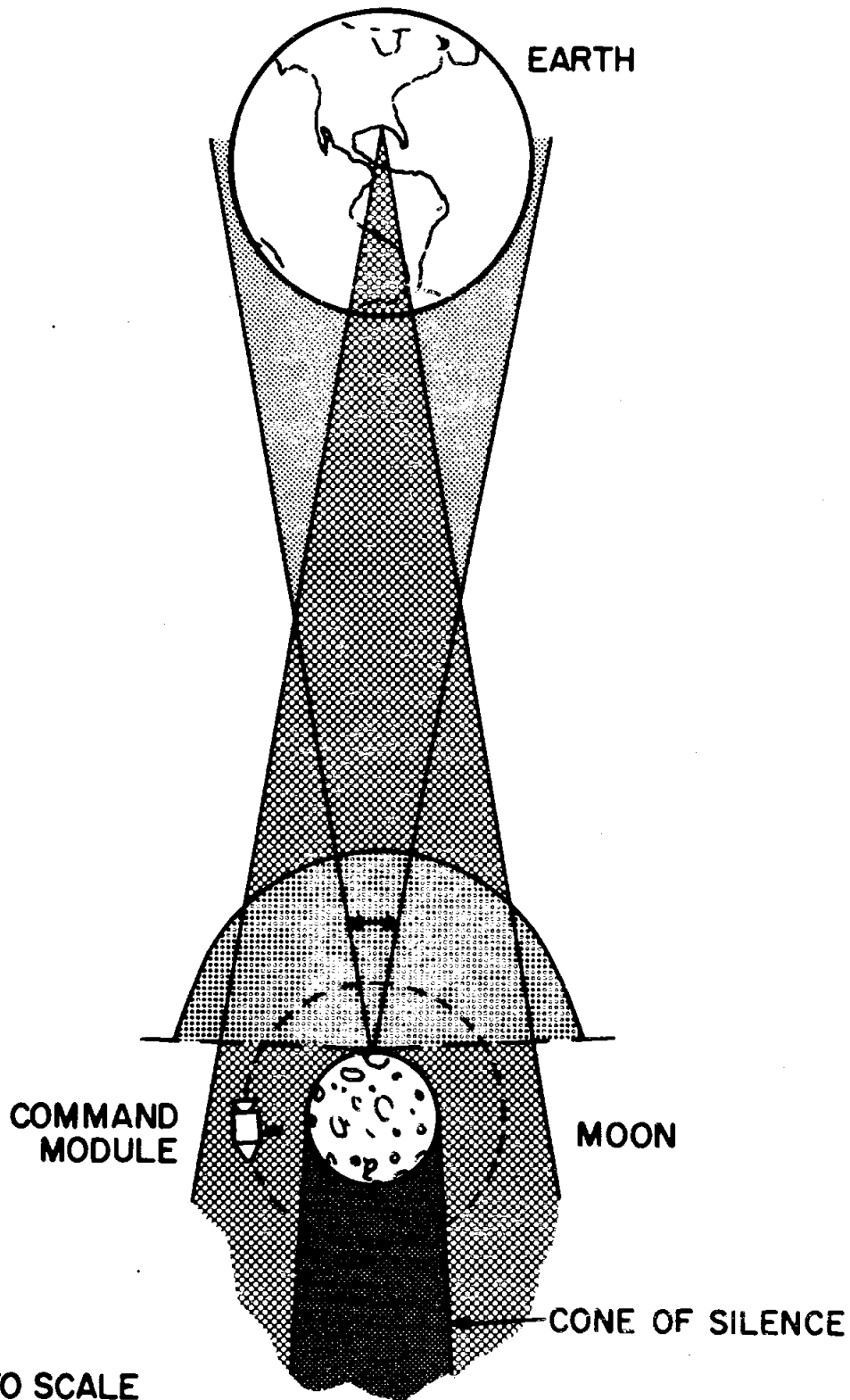
T30005-39  
JULY 66



**LUNAR SURFACE COMMUNICATIONS  
CAPABILITY**

T30005-40  
SEPT 66

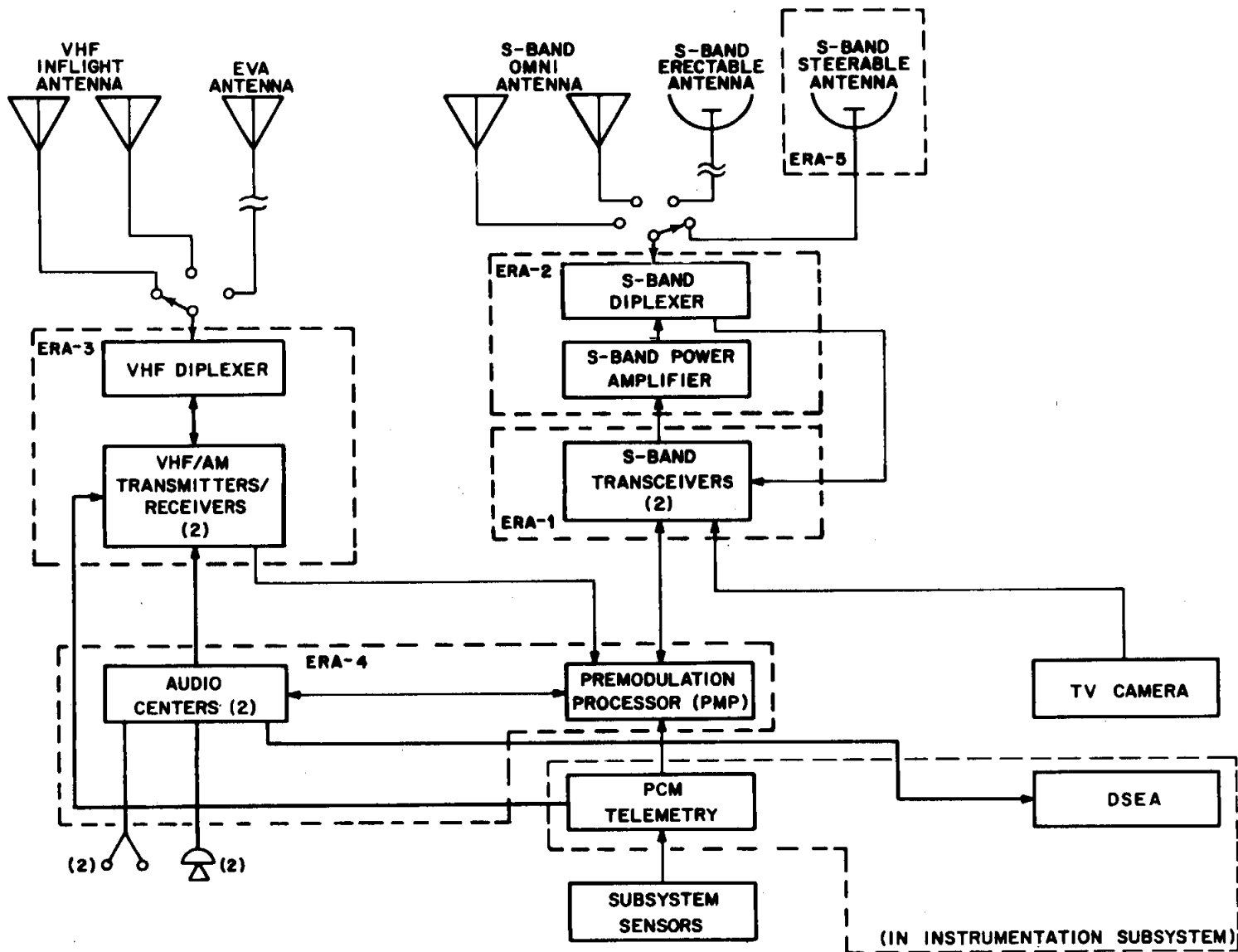




NOT DRAWN TO SCALE

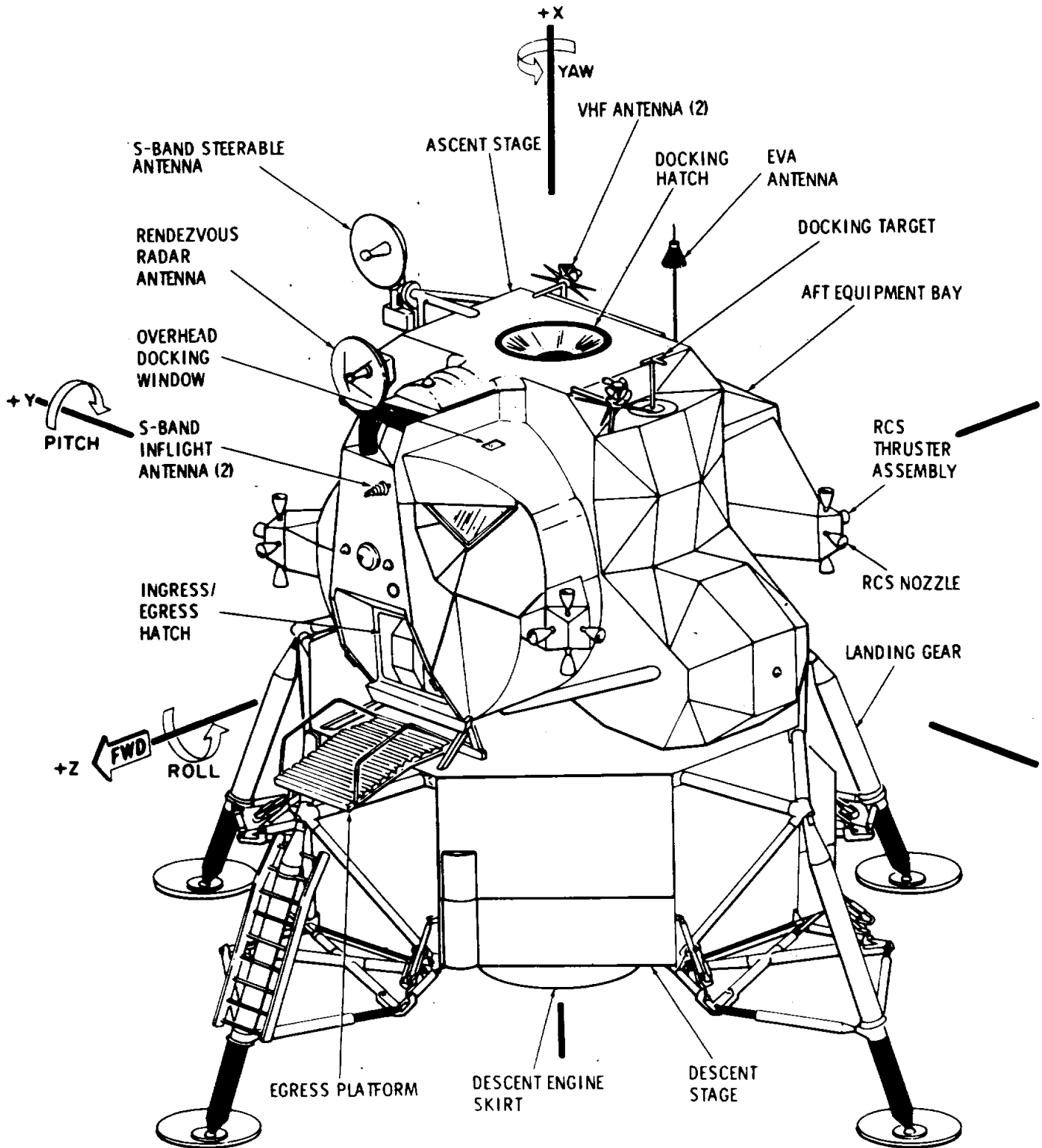
**COMMUNICATIONS DURING LUNAR STAY**

T30005-101  
MAR 66



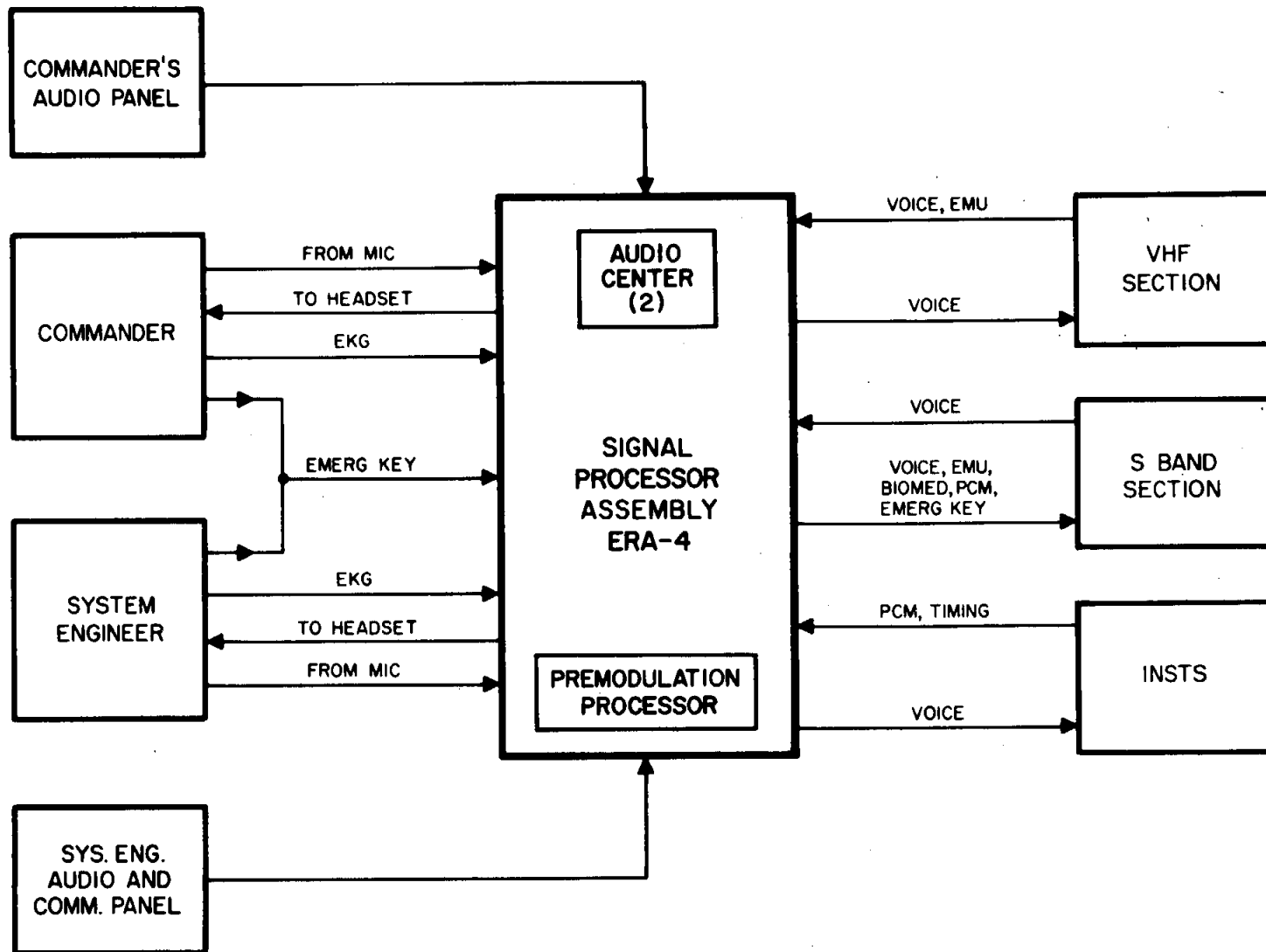
**LM COMMUNICATIONS SUBSYSTEM**

T30005-37  
DEC 66

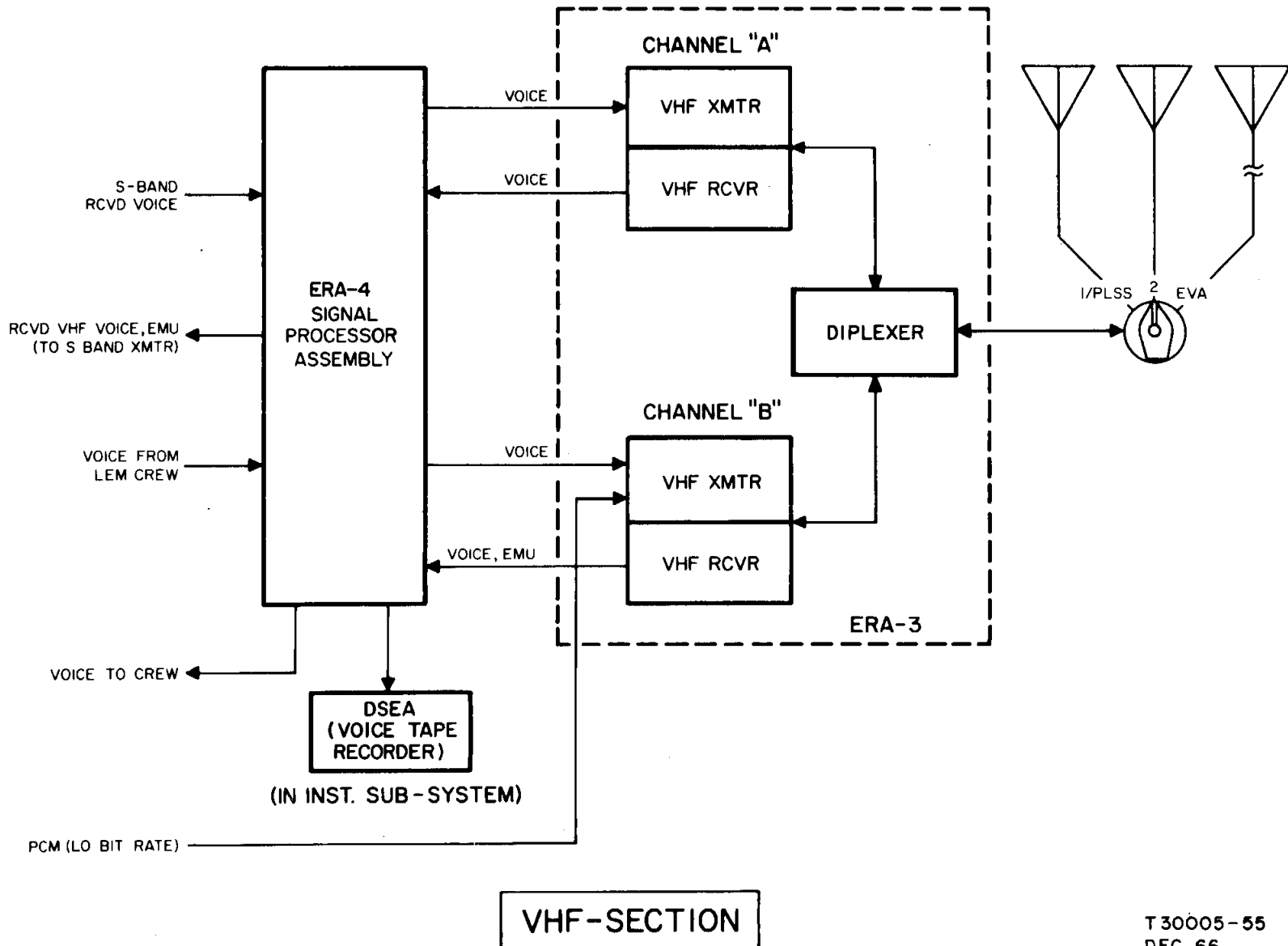


**LUNAR MODULE AXES**

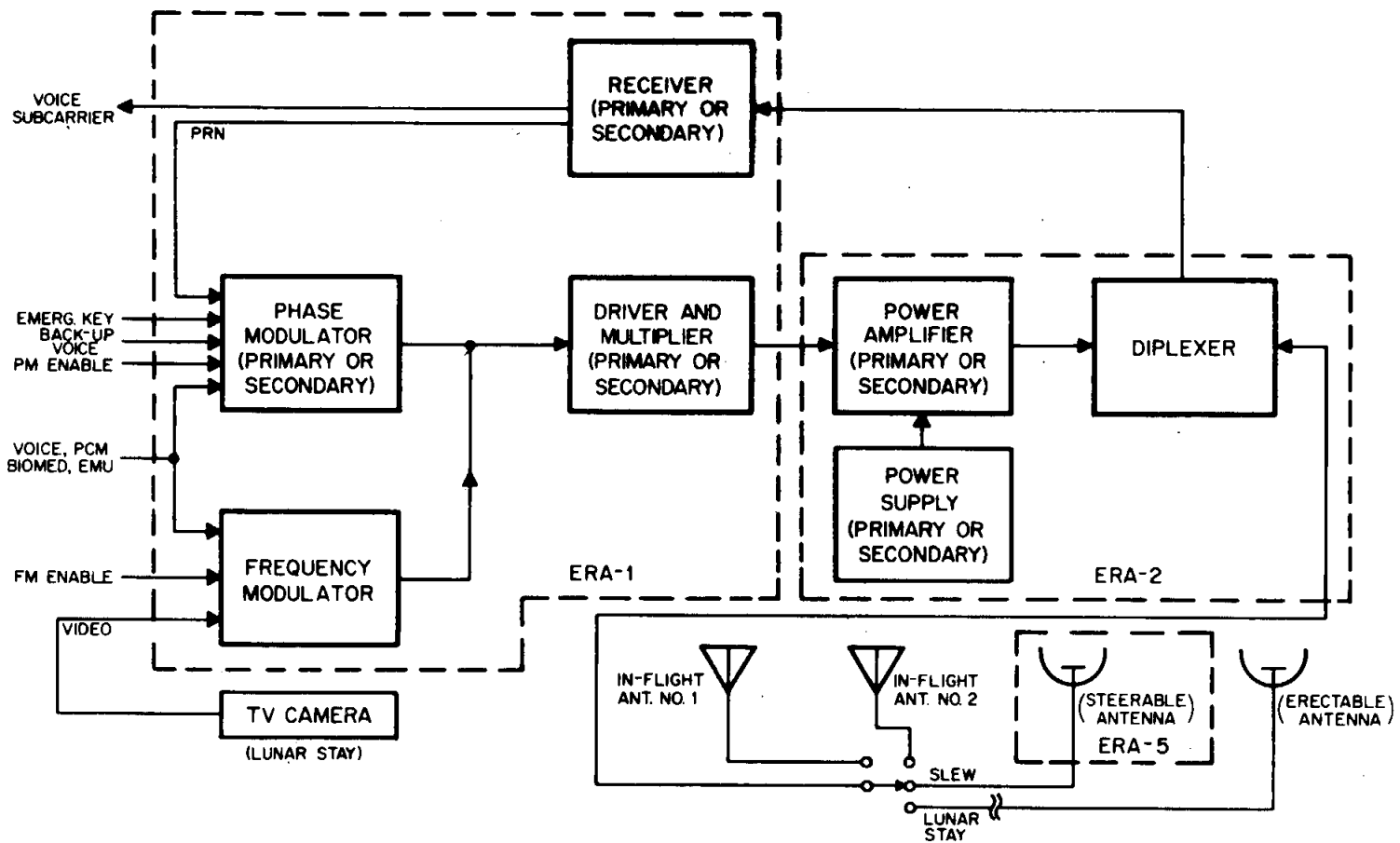
T 30005-161  
NOV 66



**SIGNAL PROCESSOR ASSEMBLY  
INTERFACE**

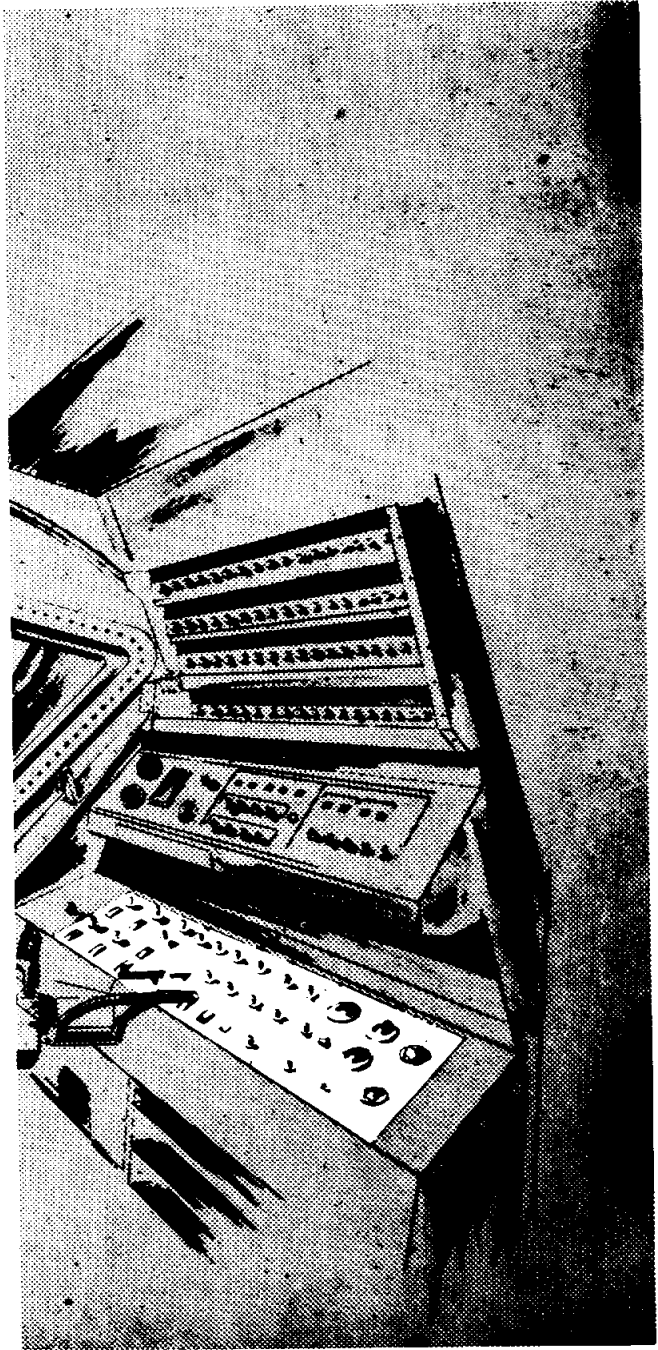


T 30005-55  
DEC 66



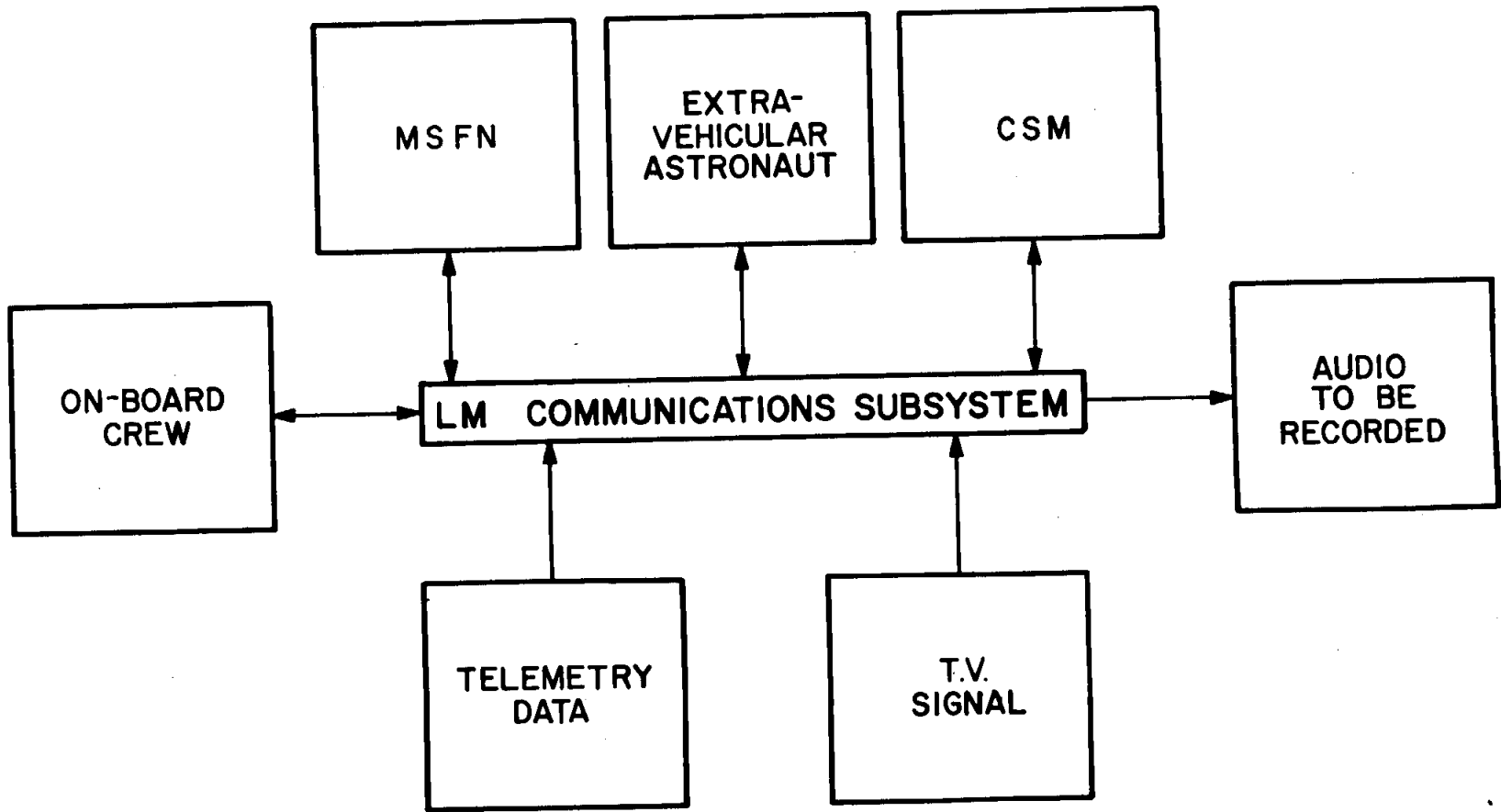
**S-BAND SECTION**

T30005-56  
DEC 66



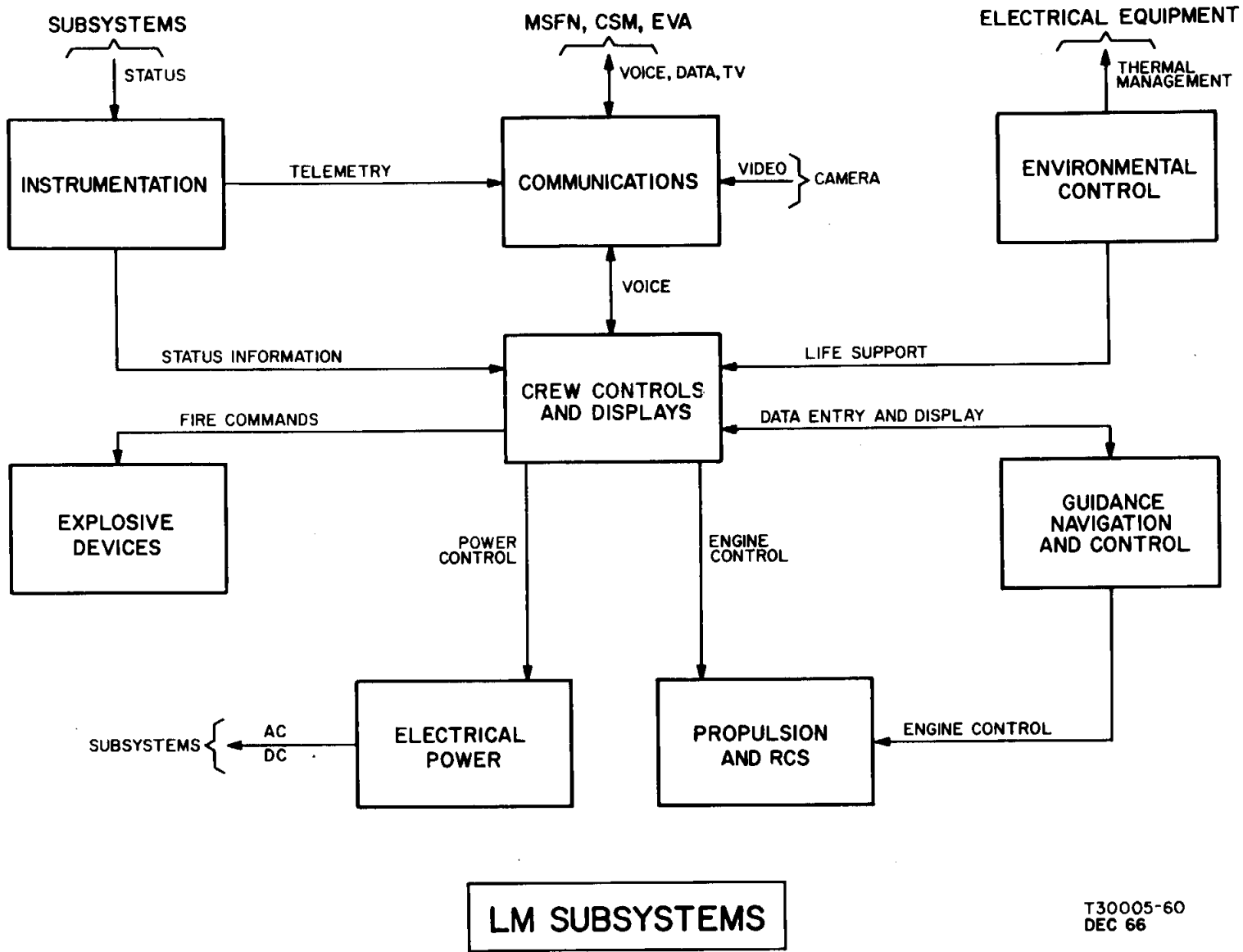
**CONTROLS**

T30005-113  
APR 66



T30005-41  
DEC 66





T30005-60  
DEC 66