

**Federal Aviation Administration, DOT**

**Pt. 135, App. F**

<sup>3</sup> For airplanes that can demonstrate the capability of deriving either the control input or control movement (one from the other) for all modes of operation and flight regimes, the "or" applies. For airplanes with non-mechanical control systems (fly-by-wire) the "and" applies. In airplanes with split surfaces, suitable combination of inputs is acceptable in lieu of recording each surface separately.

<sup>4</sup> This column applies to aircraft manufactured after October 11, 1991.

[Doc. No. 25530, 53 FR 26153, July 11, 1988; 53 FR 30906, Aug. 16, 1988]

**APPENDIX E TO PART 135—HELICOPTER FLIGHT RECORDER SPECIFICATIONS**

Parameters	Range	Accuracy sensor input to DFDR readout	Sampling interval (per second)	Resolution <sup>2</sup> read out
Time (GMT) .....	24 Hrs .....	±0.125% Per Hour .....	0.25 (1 per 4 seconds).	1 sec
Altitude .....	–1,000 ft to max certificated altitude of aircraft.	±100 to ±700 ft (See Table 1, TSO-C51a).	1 .....	5' to 30'.
Airspeed .....	As the installed measuring system.	±3% .....	1 .....	1 kt
Heading .....	360° .....	±2° .....	1 .....	0.5°.
Normal Acceleration (Vertical)	–3g to +6g .....	±1% of max range excluding datum error of ±5%.	8 .....	0.01g
Pitch Attitude .....	±75° .....	±2° .....	2 .....	0.5°
Roll Attitude .....	±180° .....	±2° .....	2 .....	0.5°.
Radio Transmitter Keying .....	On-Off (Discrete) .....	.....	1 .....	0.25 sec
Power in Each Engine: Free Power Turbine Speed and Engine Torque.	0–130% (power Turbine Speed) Full range (Torque).	±2% .....	1 speed 1 torque (per engine).	0.2% <sup>1</sup> to 0.4% <sup>1</sup>
Main Rotor Speed .....	0–130% .....	±2% .....	2 .....	0.3% <sup>1</sup>
Altitude Rate .....	±6,000 ft/min .....	As installed .....	2 .....	0.2% <sup>1</sup>
Pilot Input—Primary Controls (Collective, Longitudinal Cyclic, Lateral Cyclic, Pedal).	Full range .....	±3% .....	2 .....	0.5% <sup>1</sup>
Flight Control Hydraulic Pressure Low.	Discrete, each circuit .....	.....	1 .....	
Flight Control Hydraulic Pressure Selector Switch Position, 1st and 2nd stage.	Discrete .....	.....	1 .....	
AFCS Mode and Engagement Status.	Discrete (5 bits necessary).	.....	1 .....	
Stability Augmentation System Engage.	Discrete .....	.....	1 .....	
SAS Fault Status .....	Discrete .....	.....	0.25 .....	
Main Gearbox Temperature Low.	As installed .....	As installed .....	0.25 .....	0.5% <sup>1</sup>
Main Gearbox Temperature High.	As installed .....	As installed .....	0.5 .....	0.5% <sup>1</sup>
Controllable Stabilator Position.	Full Range .....	±3% .....	2 .....	0.4% <sup>1</sup> .
Longitudinal Acceleration .....	±1g .....	±1.5% max range excluding datum error of ±5%.	4 .....	0.01g.
Lateral Acceleration .....	±1g .....	±1.5% max range excluding datum of ±5%.	4 .....	0.01g.
Master Warning .....	Discrete .....	.....	1 .....	
Nav 1 and 2 Frequency Selection.	Full range .....	As installed .....	0.25 .....	
Outside Air Temperature .....	–50° C to +90° C .....	±2° C .....	0.5 .....	0.3° C

<sup>1</sup> Per cent of full range.

<sup>2</sup> This column applies to aircraft manufactured after October 11, 1991.

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**APPENDIX F TO PART 135—AIRPLANE FLIGHT RECORDER SPECIFICATION**

The recorded values must meet the designated range, resolution, and accuracy requirements during dynamic and static conditions. All data recorded must be correlated in time to within one second.

Parameters	Range	Accuracy (sensor input)	Seconds per sampling interval	Resolution	Remarks
1. Time or Relative Time Counts <sup>1</sup> .	24 Hrs, 0 to 4095.	±0.125% Per Hour.	4 .....	1 sec .....	UTC time preferred when available. Counter increments each 4 seconds of system operation.