

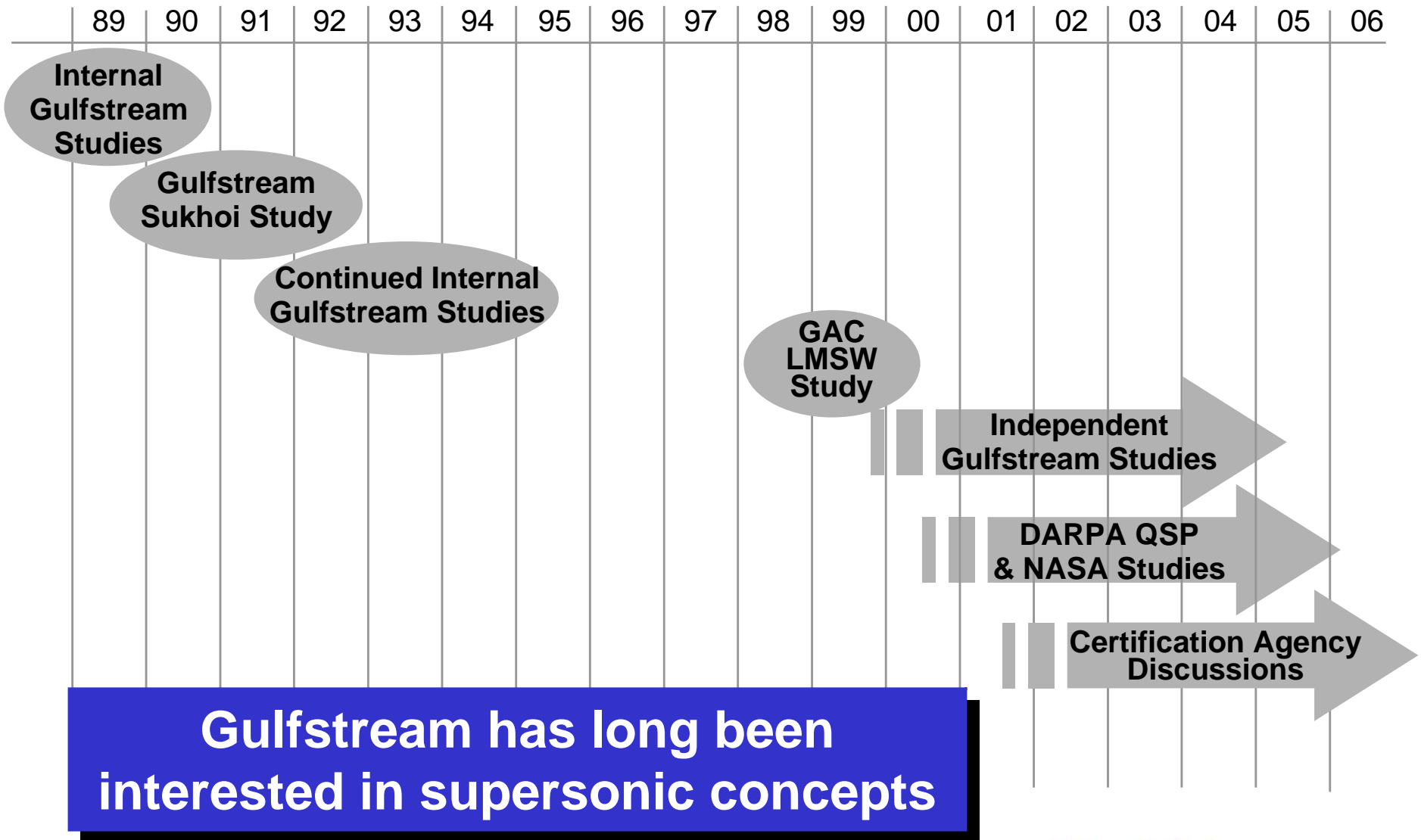


# **Recent Supersonic Vehicle Studies At Gulfstream Aerospace**

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**FAA Civil Supersonic Aircraft Workshop  
Arlington, Virginia  
13 November 2003**

# Gulfstream Supersonic Studies Timeline



# All Supersonic Studies are Not the Same

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NASA's High Speed Civil Transport (HSCT) program, initiated in the late 1980s and terminated in 1999, differs greatly from the envisioned Quiet Supersonic Jet (QSJ)

- Design Requirements
  - HSCT... a 300 pax, Mach 2.4, 600k lb airliner
  - QSJ... a relatively small 100k to 150k lb, Mach 1.8 transport
- Study Focus
  - HSCT... not intended to fly supersonic over land
  - QSJ... requires supersonic over land flight for success
- Focused efforts to design a small, low boom configuration, if successful, may pave the way forward for future high speed vehicles.

# QSJ Initial Design Goals

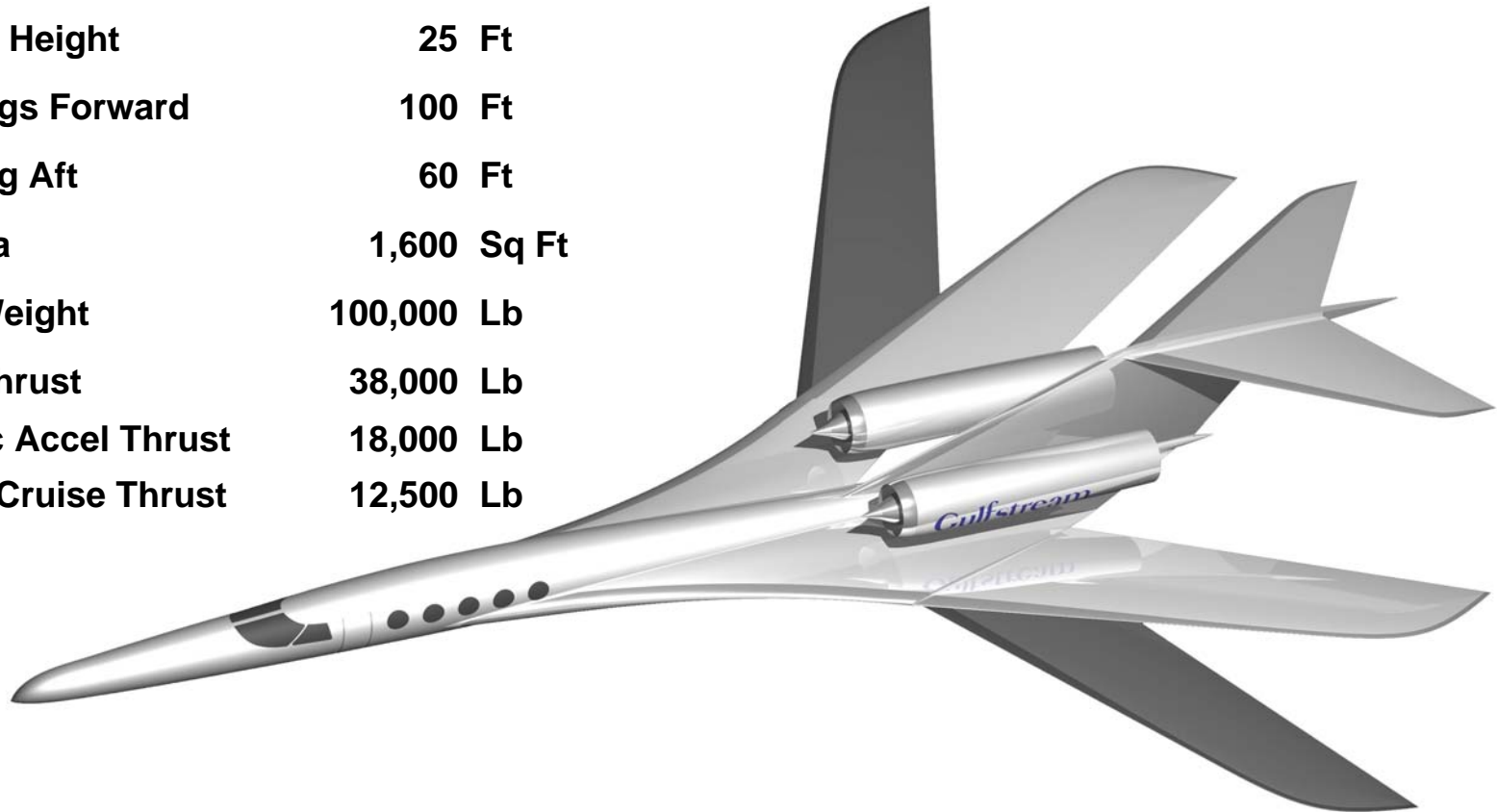
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How Far	→ NBAA IFR Range	4,800 NM
How Fast	→ Cruise Mach	1.6 - 2.0
How Much	→ Max Ramp Weight	100,000 Lb
	→ Design Payload	1,600 Lb
	→ Cabin Size	1,300 Cu Ft (GII Size)
From Where	→ Takeoff Field Length - SL;ISA+20C	6,500 Ft
	→ ACN, Approach Category, and Design Group	<30 / C / III
Safely	→ Civil Certification	FAA FAR 25 or Similar Standard
<b>Responsibly</b>	→ <b><u>Environmental Issues</u></b>	
	→ Boom Overpressure	Acceptable for Overland SS Flight
	→ Takeoff Emissions	ICAO with Margin
	→ Cruise Emissions	Minimum Impact
	→ Airport Noise	Stage 4 with 10dB Margin
Reliably	→ Mission Readiness	> 0.99
Cost	→ Engine Life (STBO)	>= 2,000 Hr
Effectively	→ Civil Market Price	\$ 70 - 100 M

# A Baseline QSJ Configuration

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Overall Length	140 Ft
Maximum Height	25 Ft
Span Wings Forward	100 Ft
Span Wing Aft	60 Ft
Wing Area	1,600 Sq Ft
Takeoff Weight	100,000 Lb
Takeoff Thrust	38,000 Lb
Transonic Accel Thrust	18,000 Lb
Mach 1.8 Cruise Thrust	12,500 Lb



**Variable geometry provides... improved airport performance, lower noise, and improved subsonic range at the expense of... increased complexity, more difficult certification, and weight.**

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# QSJ Technical Challenges

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## Propulsion Integration

- Engine Fuel Efficiency
- Engine Life
- High Inlet Performance
- Low Inlet Distortion
- Rotor Burst Protection

## Structural Arrangement

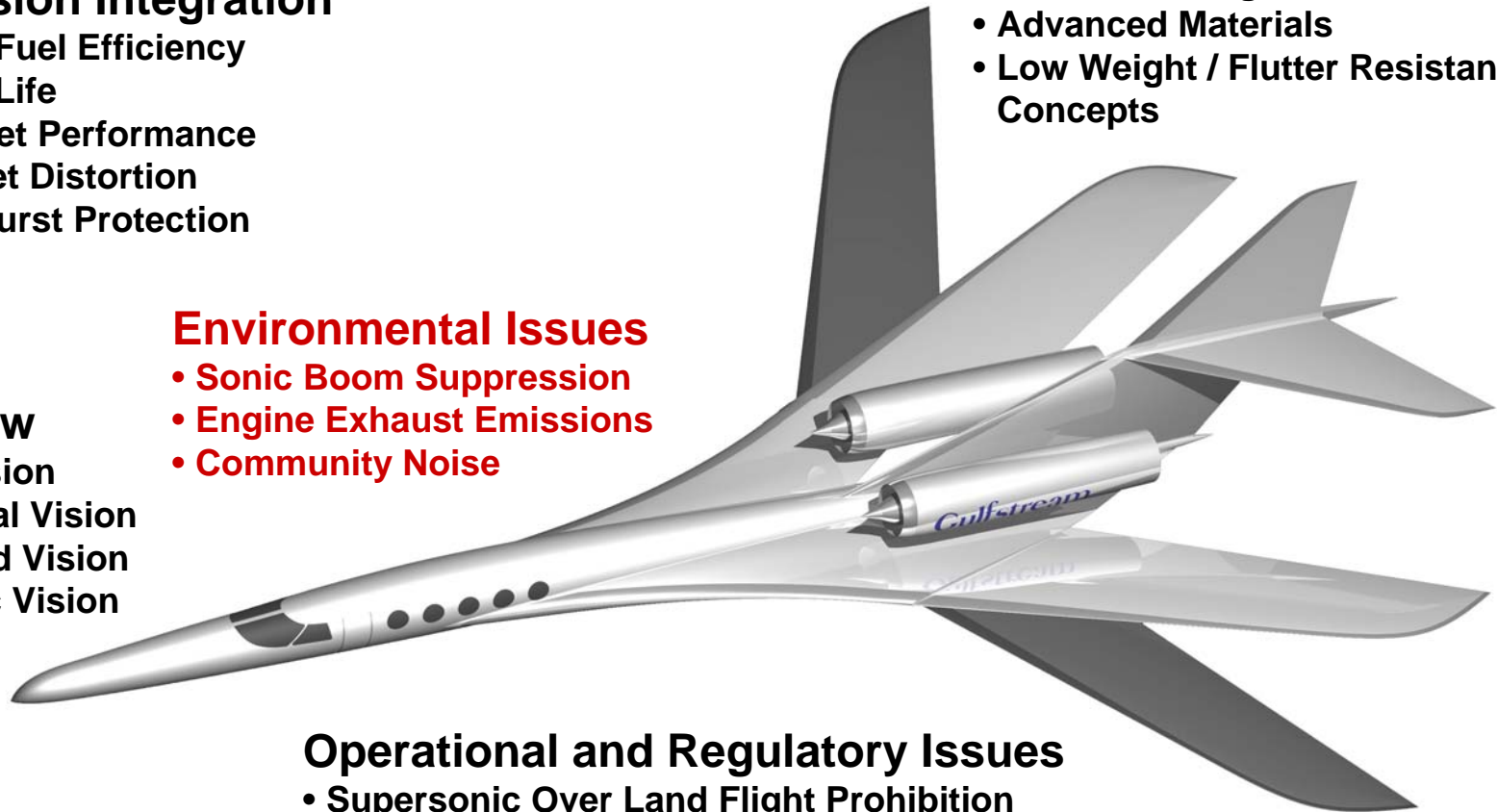
- Structural Stiffness
- Thermal Management
- Advanced Materials
- Low Weight / Flutter Resistant Concepts

## Environmental Issues

- Sonic Boom Suppression
- Engine Exhaust Emissions
- Community Noise

## Pilot View

- Video Vision
- Conformal Vision
- Enhanced Vision
- Synthetic Vision



## Advanced Systems

- FBW / FBL / PBW
- Variable Geometry Systems
- CG Management

## Operational and Regulatory Issues

- Supersonic Over Land Flight Prohibition
- Certification and Safety Standards
- ATC Integration
- High Altitude Operations

## Aerodynamic Performance

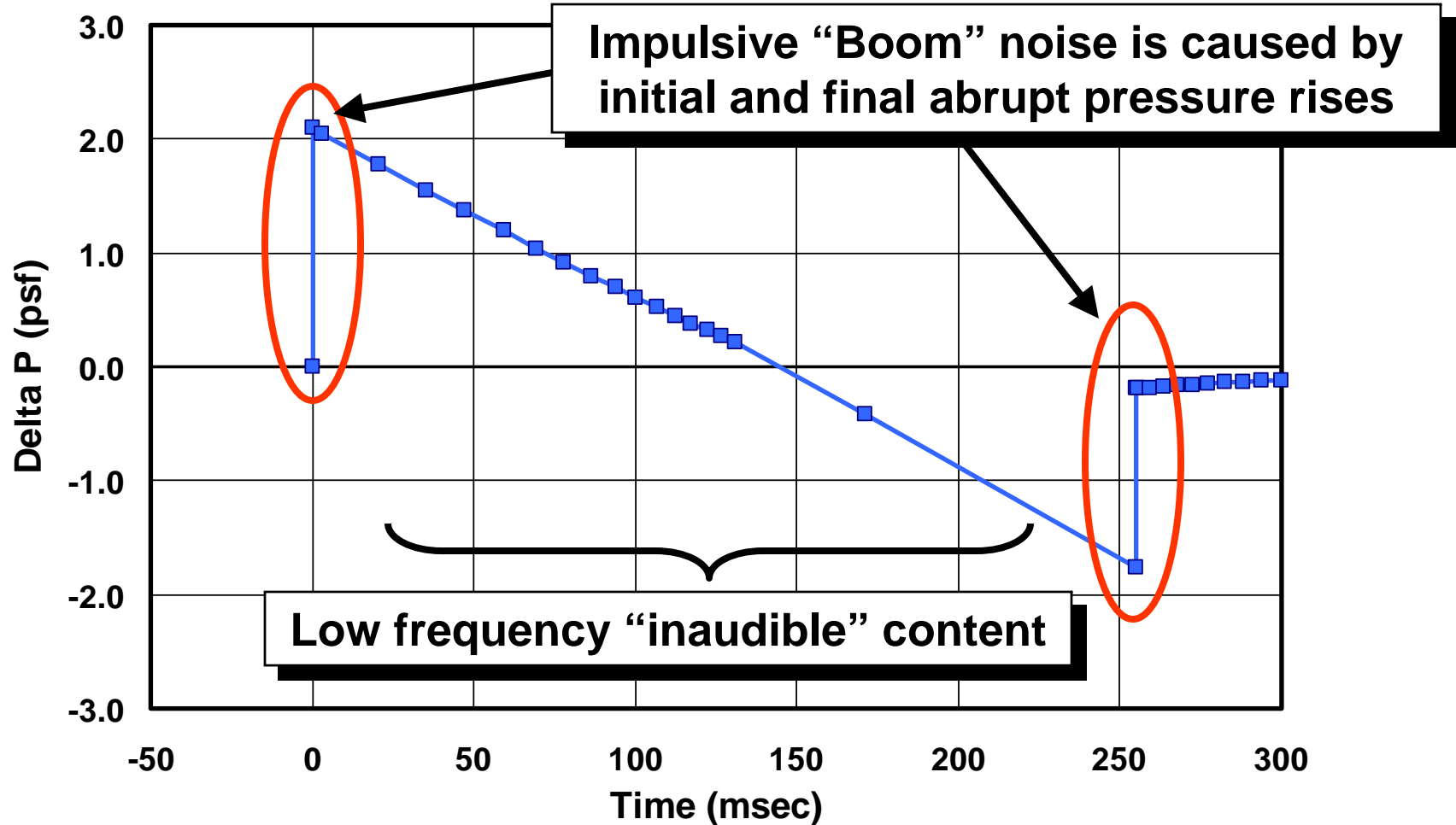
- High Supersonic L/D
- High CLmax
- Handling Qualities

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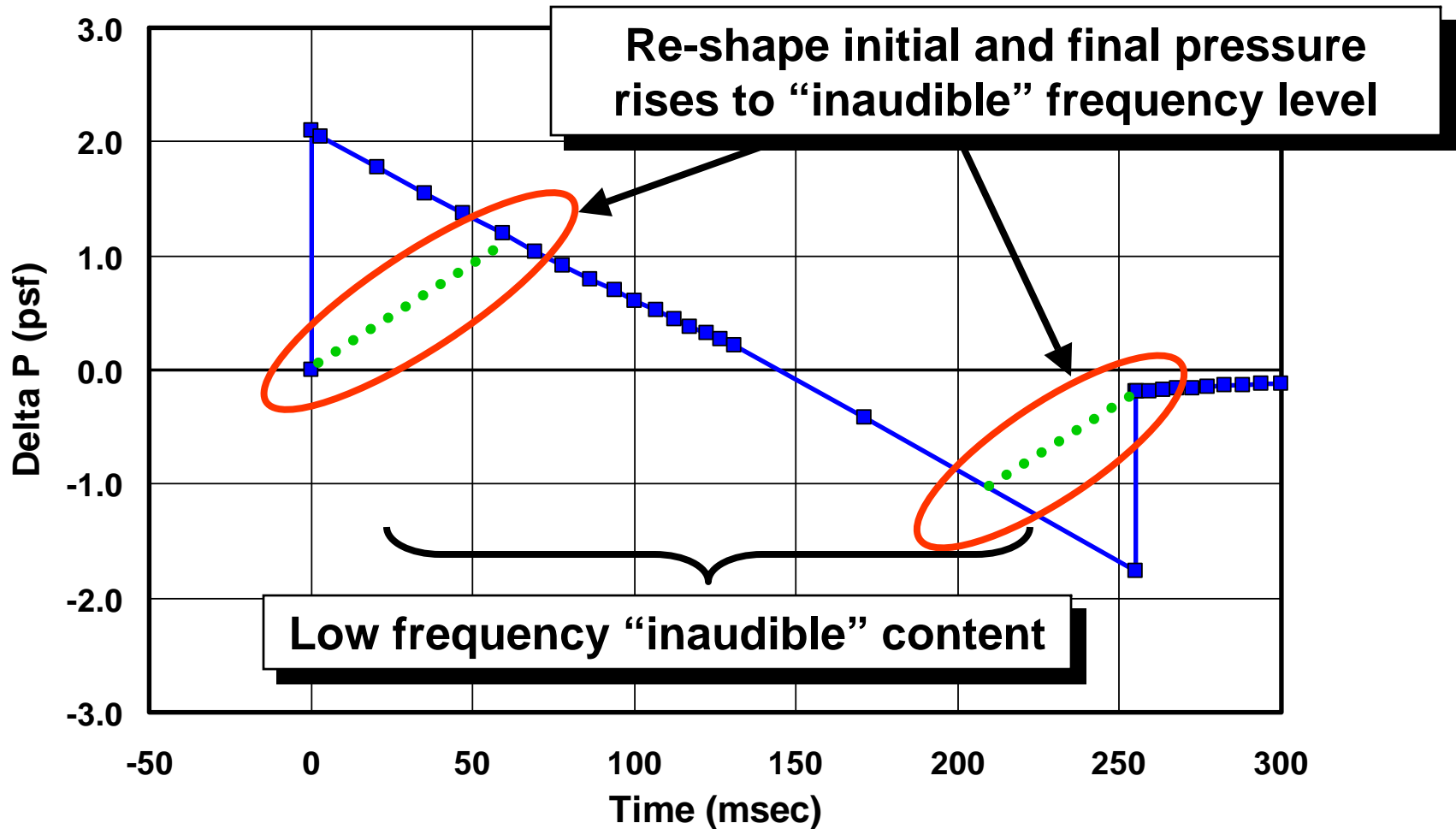
# Sonic Boom Noise Source

## N-Wave Sonic Boom Signature



# Sonic Boom Shaping Strategy

## N-Wave Sonic Boom Signature



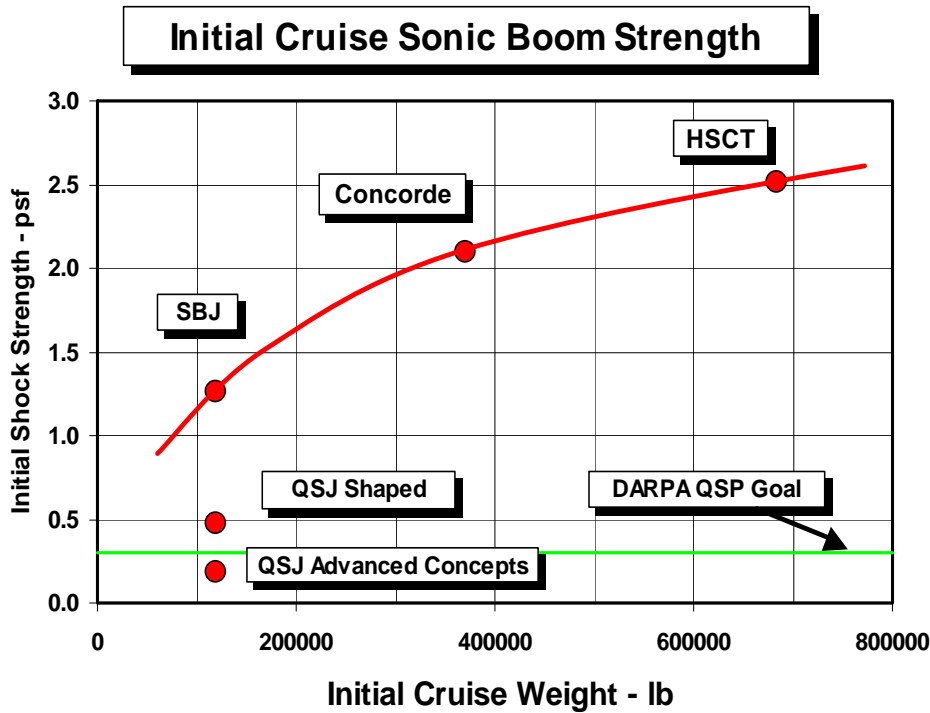


# Sonic Boom Suppression - Approach

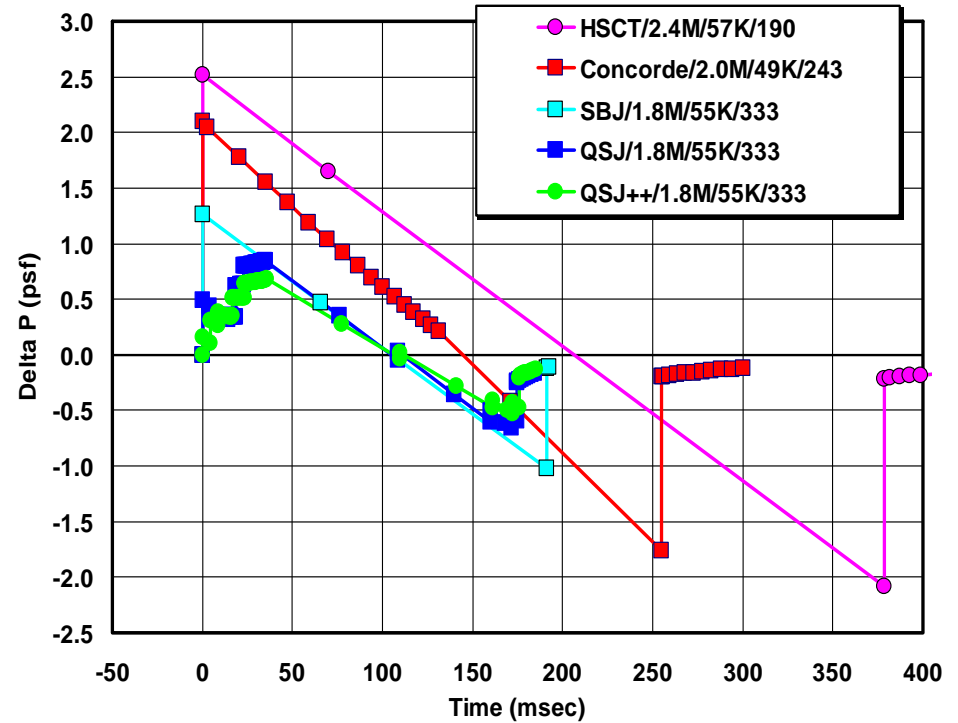
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- **Boom Strength ~ Vehicle Weight**
  - QSJ Size An Advantage
  - Minimize Weight
- **Aerodynamic Shaping**
  - Vehicle Configuration
  - Engine Placement
  - Boom Reduction Concept Development
  - Wind Tunnel Testing (Concept Validation)
- **Define Acceptable Boom Characteristics/Signature**
  - Human Ear Response
  - Building Structures Response
  - Sonic Boom Simulation
    - Gulfstream Boom Lab
    - NASA LaRC Boom Lab
- **Develop/Test Technology Demonstrator Vehicle**

# Progress on Sonic Boom Reduction



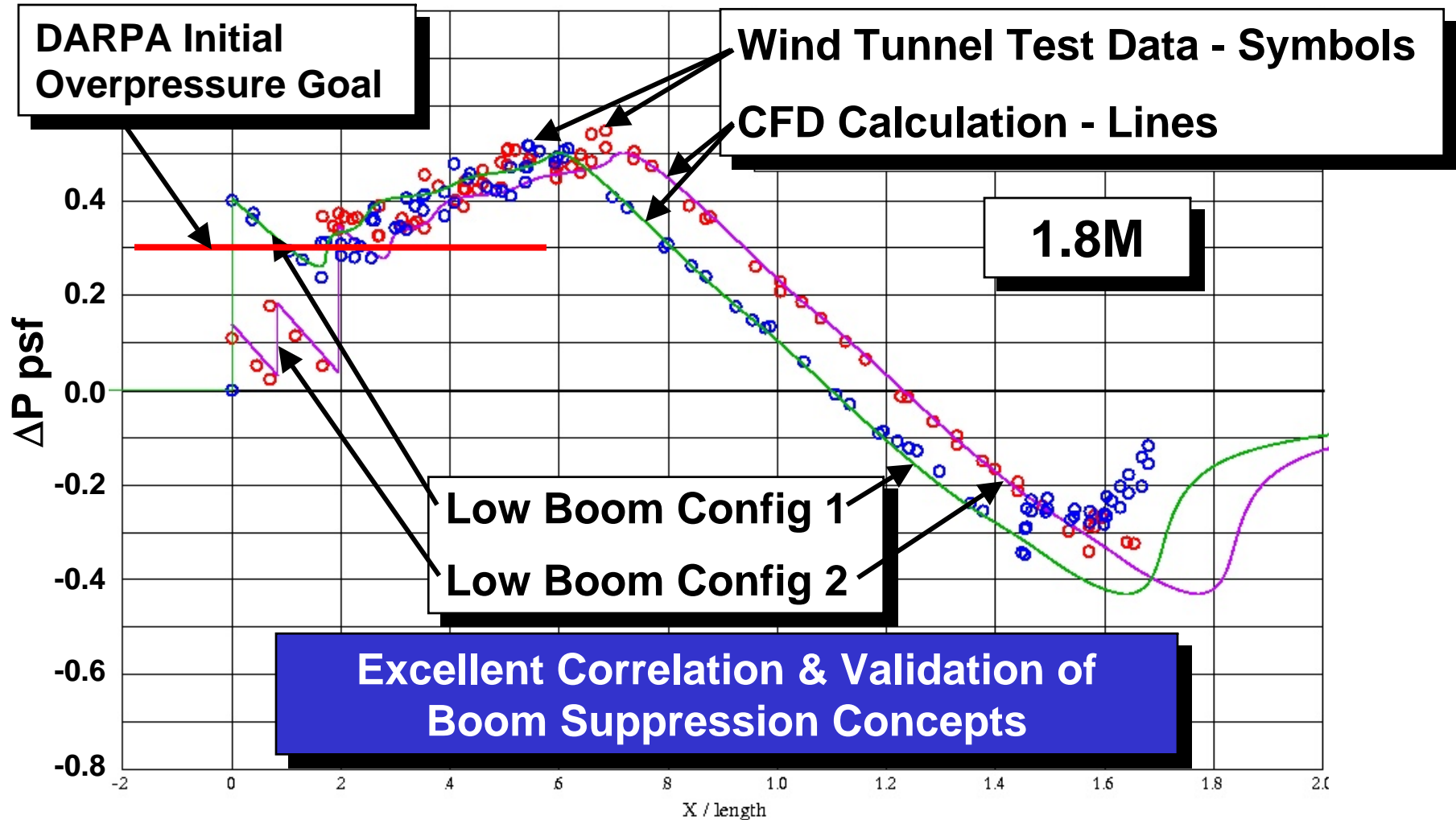
Lighter weight vehicles have weaker sonic boom signatures.



Special shaping further reduces the sonic boom signature.

**Vehicle size, shaping, and advanced concepts dramatically reduce the sonic boom signature.**

# Advanced Boom Reduction Concept Testing

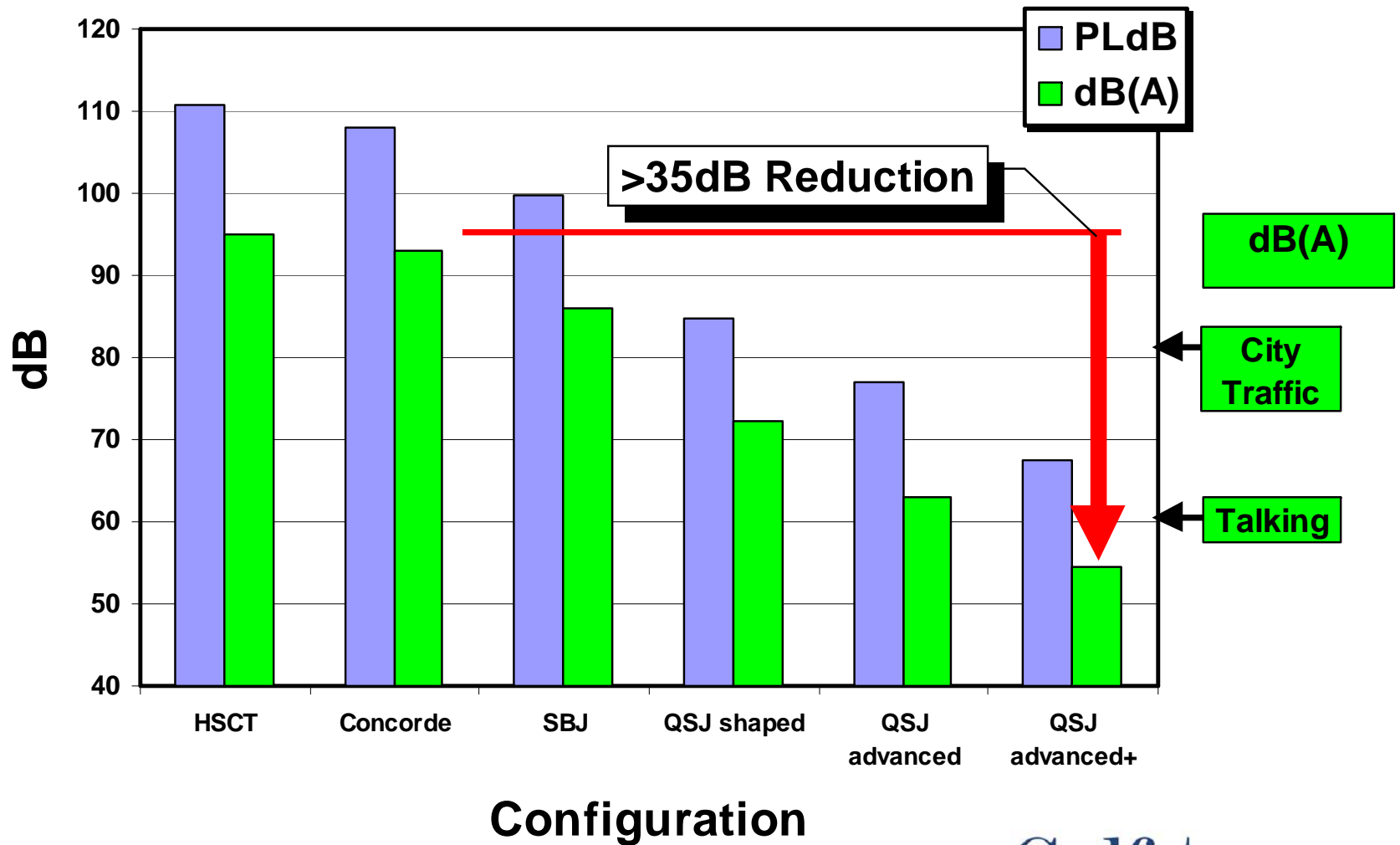


Gulfstream Wind Tunnel Models in NASA Langley UPWT

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# Cruise Acoustic Signature Levels

QSJ Advanced+ >35dB Quieter Than Concorde



# Engine Exhaust Emissions

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## Airport Environment, Requirements Based on ICAO Regulations and Demonstrated Engine Performance

- Oxides of Nitrogen <60 g/kN
- Unburned Hydrocarbons <4 g/kN
- Carbon Monoxide <40 g/kN
- Characteristic Smoke Number <10

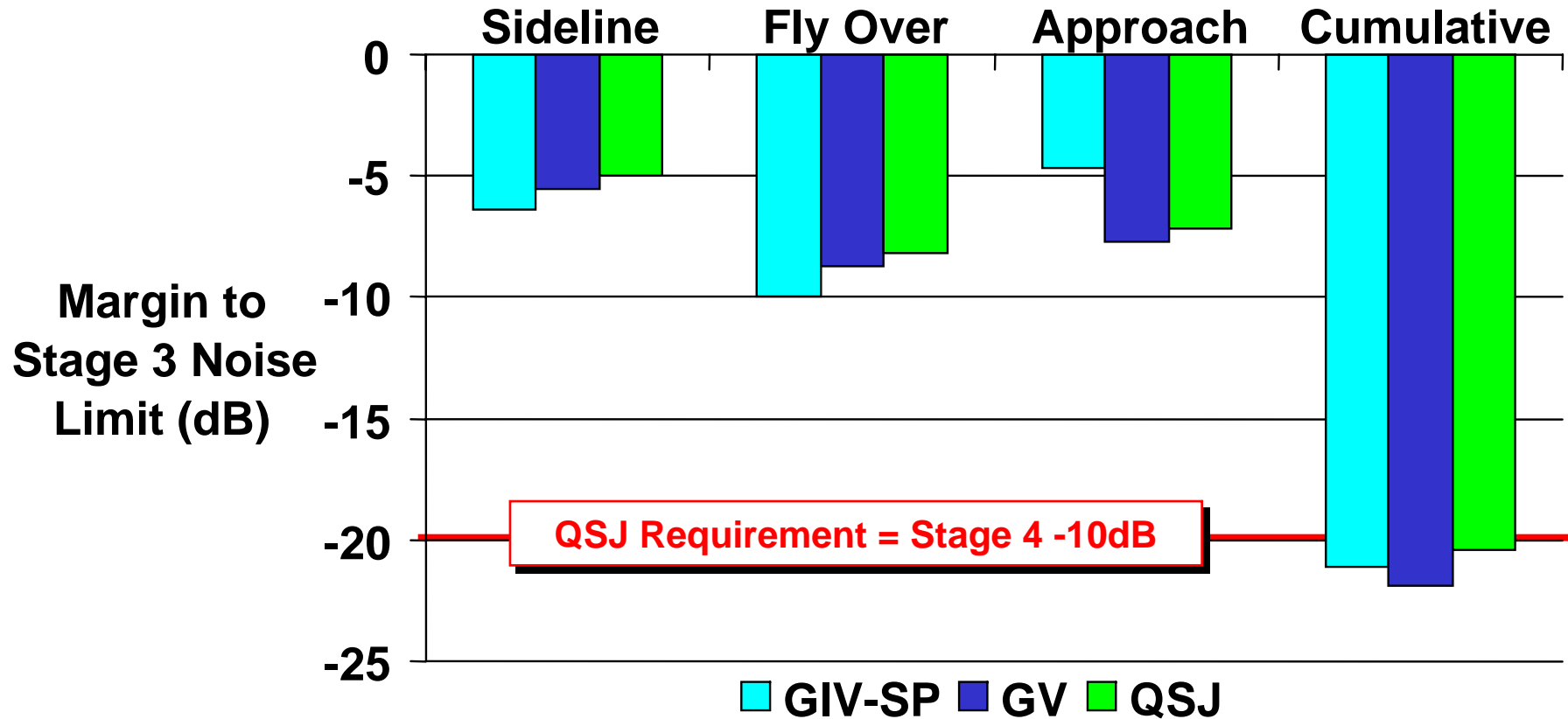
**Considered  
Technically  
Feasible**

## Cruise Emissions -- No Regulations for NO<sub>x</sub>, Ozone Depletion, H<sub>2</sub>O , CO<sub>2</sub>

- Gulfstream Atmospheric Modeling Study -- Favorable Results for QSJ NOX/Ozone Impact
- H<sub>2</sub>O, CO<sub>2</sub> -- Design for Min Fuel Burn -- Minimize SFC and Zero Fuel Weight, and Maximize L/D

**“Green” aircraft solutions are driven by aeronautical fundamentals.**

# Estimated Certification Noise Levels



- Future designs should not be any noisier than today's product standard.
- Initial estimates indicate Stage 4 -10dB is achievable.

# Progress Being Made

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- **Smaller supersonic configurations designed to cruise below Mach 2.0 are viewed as a logical first step toward next generation supersonic transports.**
- **Configurations with shaped, low boom signatures are considered feasible.**
- **Boom suppression technology demonstrator is needed.**
- **Certification standards and criteria for acceptable supersonic over land flight need to be developed.**

**Progress toward a viable supersonic business jet requires elimination of the prohibition of supersonic over land flight.**