

# **Dual Fuel Issues Related To Performance, Emissions and Combustion Instability In Gas Turbine Systems**

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# Motivation

- **Gas turbine operation on dual fuels has significant operational payoffs.**
- **The requirement to operate on both gaseous and liquid fuels has potential impacts on:**
  - **Emissions (NO<sub>x</sub>, CO and UHC)**
  - **Stability**
  - **Autoignition, flashback and lean blowout.**
- **There is a need for systematic studies directly comparable to natural gas operation.**

# Goals and Objectives

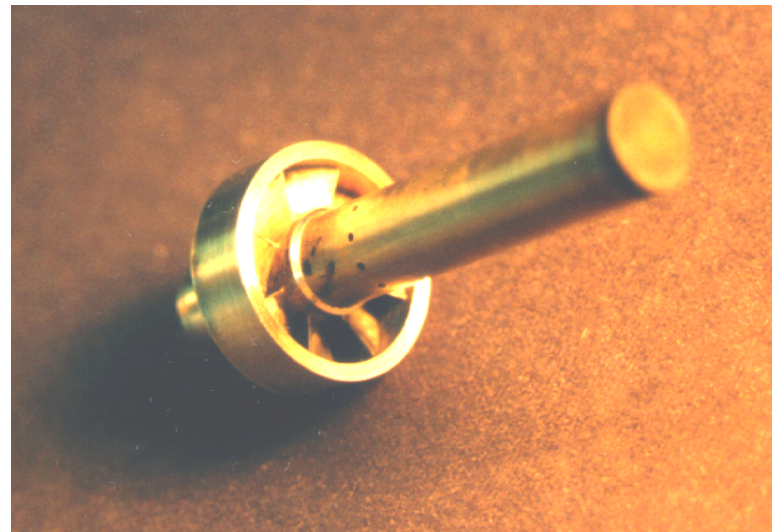
- **Address dual fuel issues related to performance, emissions and combustion stability by:**
  - **Sequentially studying effects of premixing, fuel chemistry, atomization, drop vaporization and multi-point injection**
  - **Investigating liquid fuel effects in a model combustor previously studied for natural gas-air mixtures for comparison**

# Approach

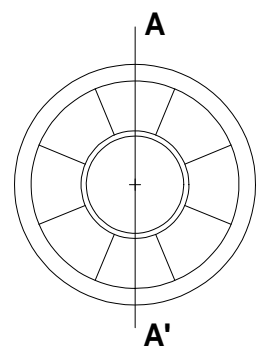
- **Three phase research program involving:**
  - **Prevaporized liquid fuel studies**
  - **Uni-element liquid fuel spray studies**
  - **Multi-point injection studies**
- **Utilize previously developed optically accessible model gas turbine combustor.**
- **Employ simple swirl injectors**
- **Apply extensive diagnostic tools available to characterize controlling combustion processes.**



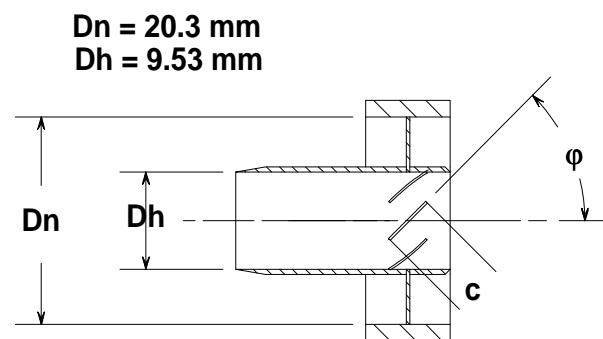
# PSU SINGLE SWIRL INJECTOR



**TOP VIEW**

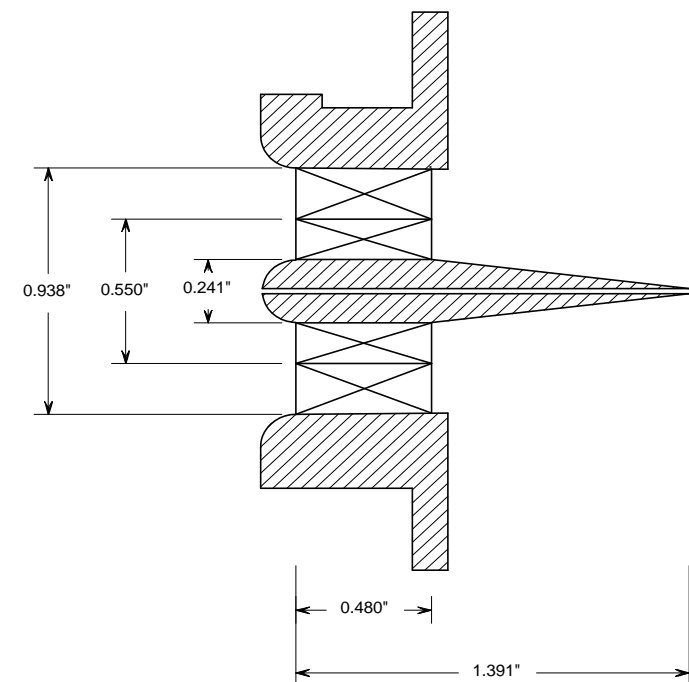
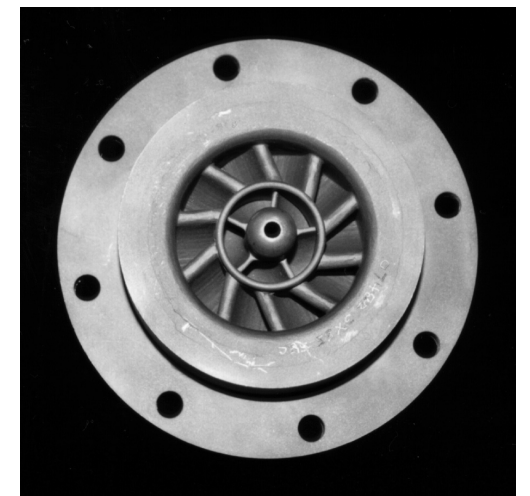
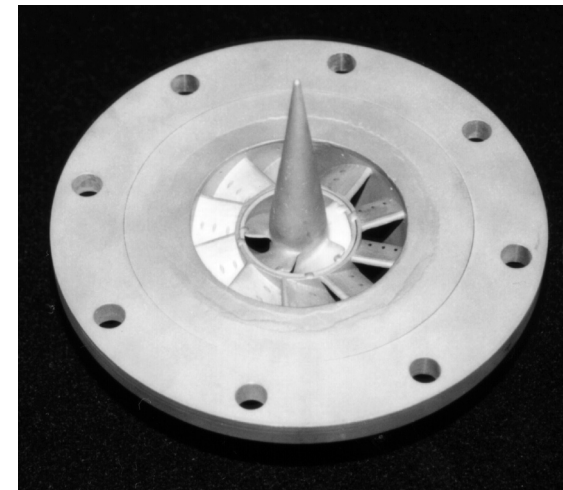


**CROSS SECTION VIEW A-A'**

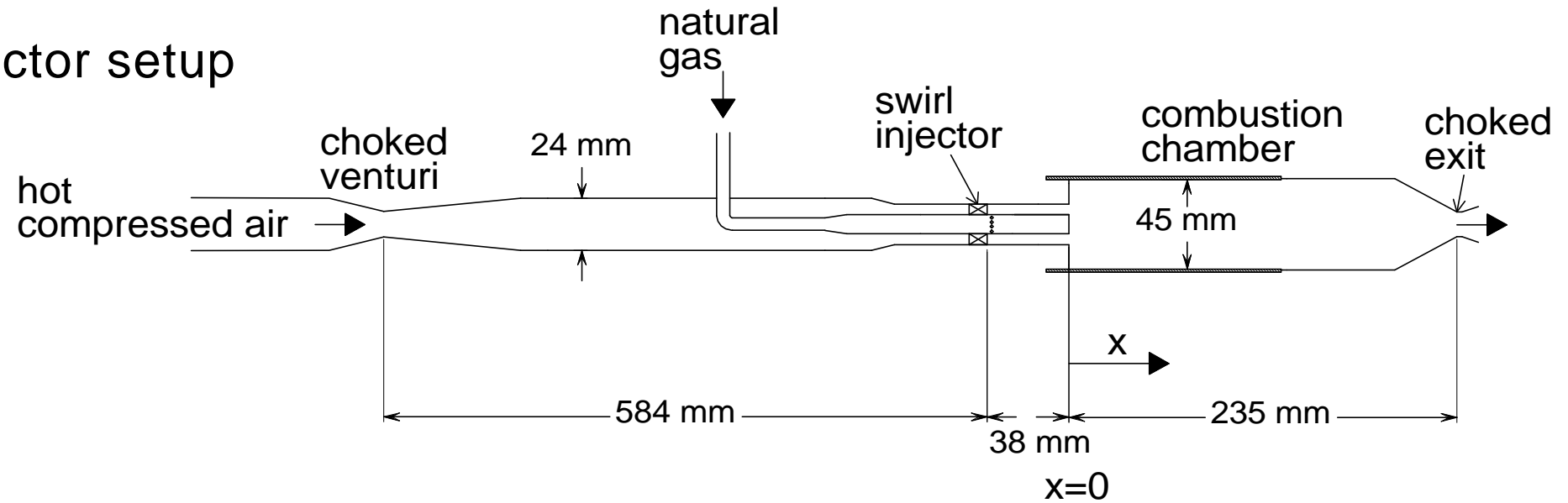


Dn = 20.3 mm  
Dh = 9.53 mm

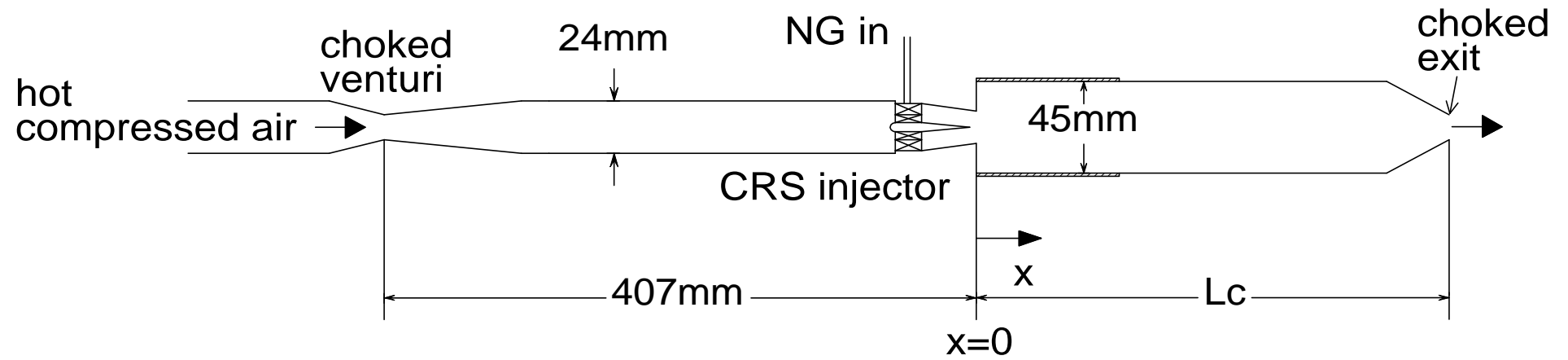
# GE CRS INJECTORS



### PSU injector setup



### CRS injector setup



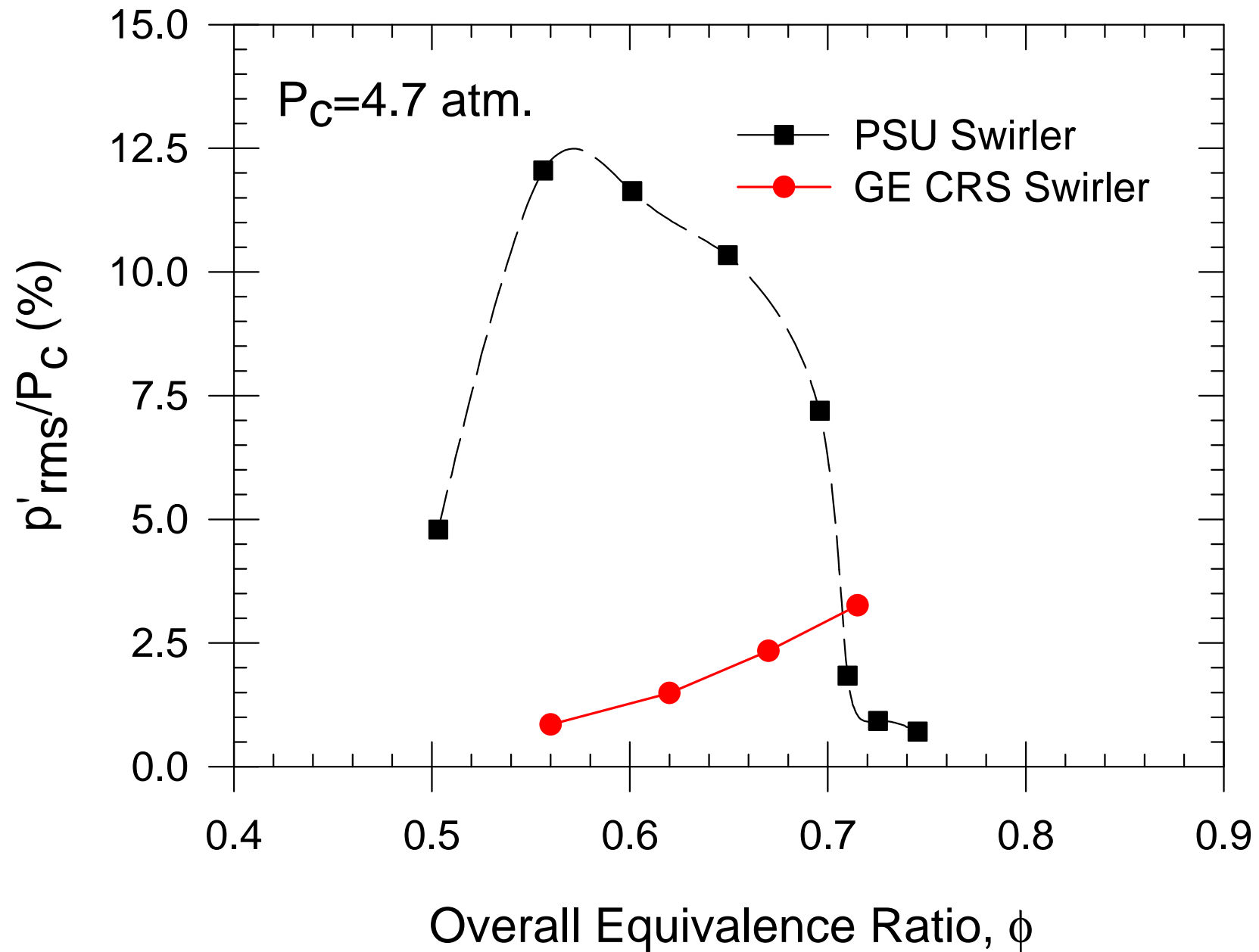
$L_c = 235\text{mm or } 350\text{mm}$

# General Results

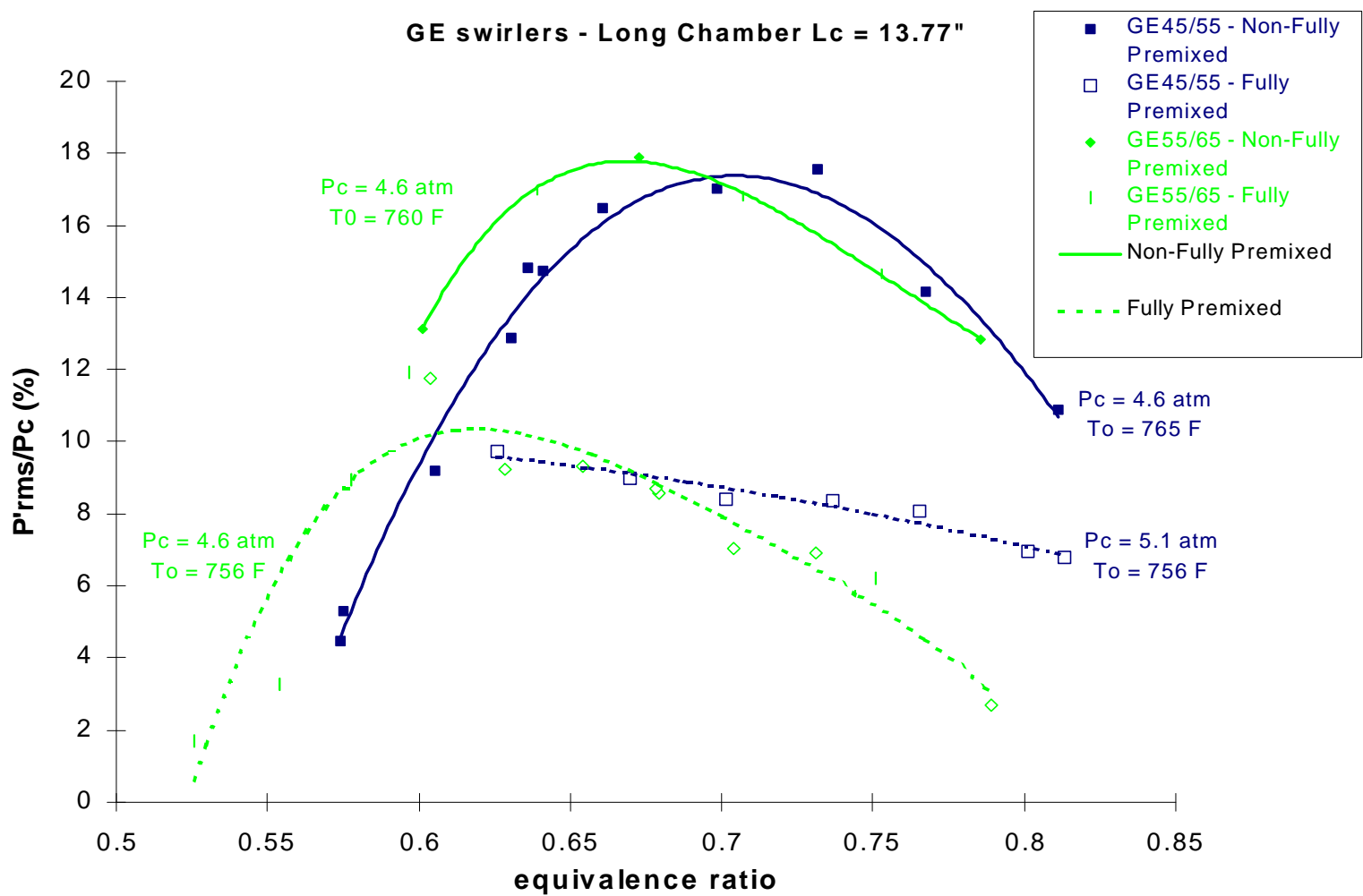
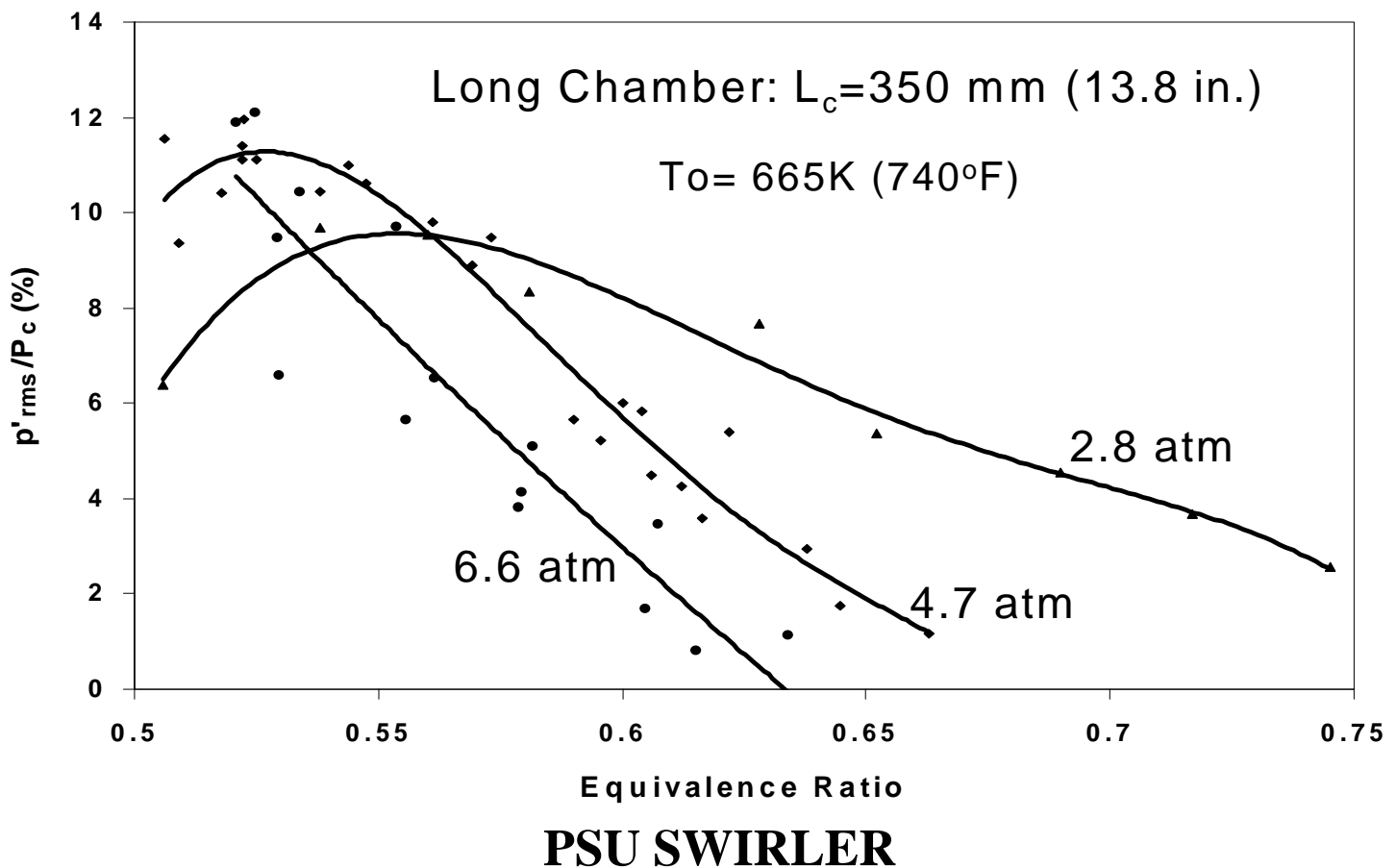
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- Modes of Instability Identified
  - \* Strong 1L mode of combustor:
    - $A_{\text{rms}} = 10\text{-}15\%$  of chamber pressure
    - $F = 1700 - 1900$  Hz for short chamber
    - $F = 1100 - 1300$  Hz for long chamber
  - \* Weak 2L mode of combustor
- Operating Parameters with a Strong Influence
  - \* Injector type (GE CRS versus PSU swirler)
  - \* Inlet air temperature  $T_o$
  - \* Equivalence ratio  $\phi$
  - \* Level of premixing (fuel injection location  $X_{\text{inj}}$ )
  - \* Swirl Angle
  - \* Chamber length  $L_c$
  - \* Inlet air velocity  $V$

# COMPARISON OF STABILITY MAPS FOR TWO TYPES OF SWIRLERS (SHORT CHAMBER CONFIGURATION)

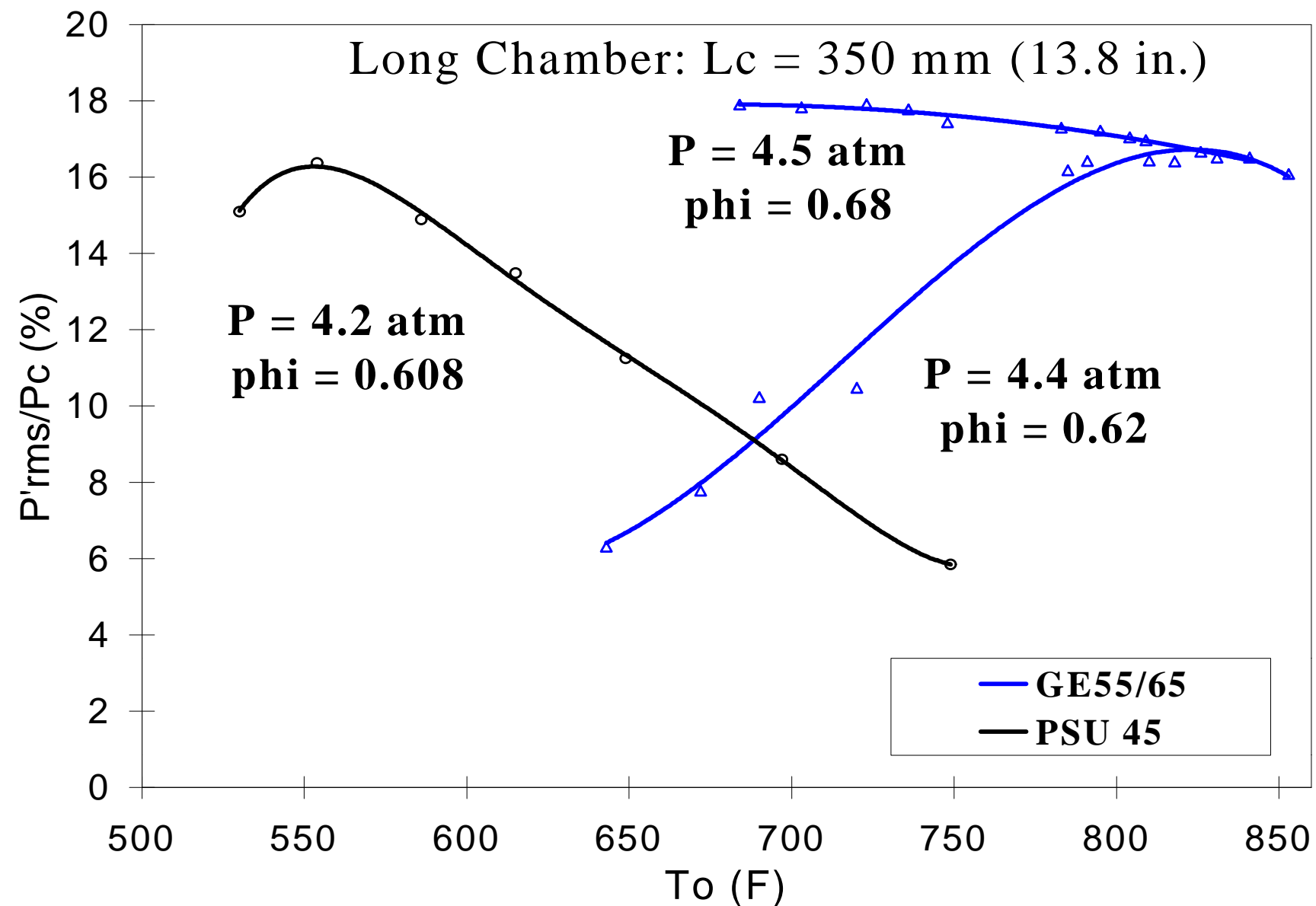


# COMPARISON OF STABILITY MAPS FOR TWO TYPES OF SWIRLERS (LONG CHAMBER CONFIGURATION)

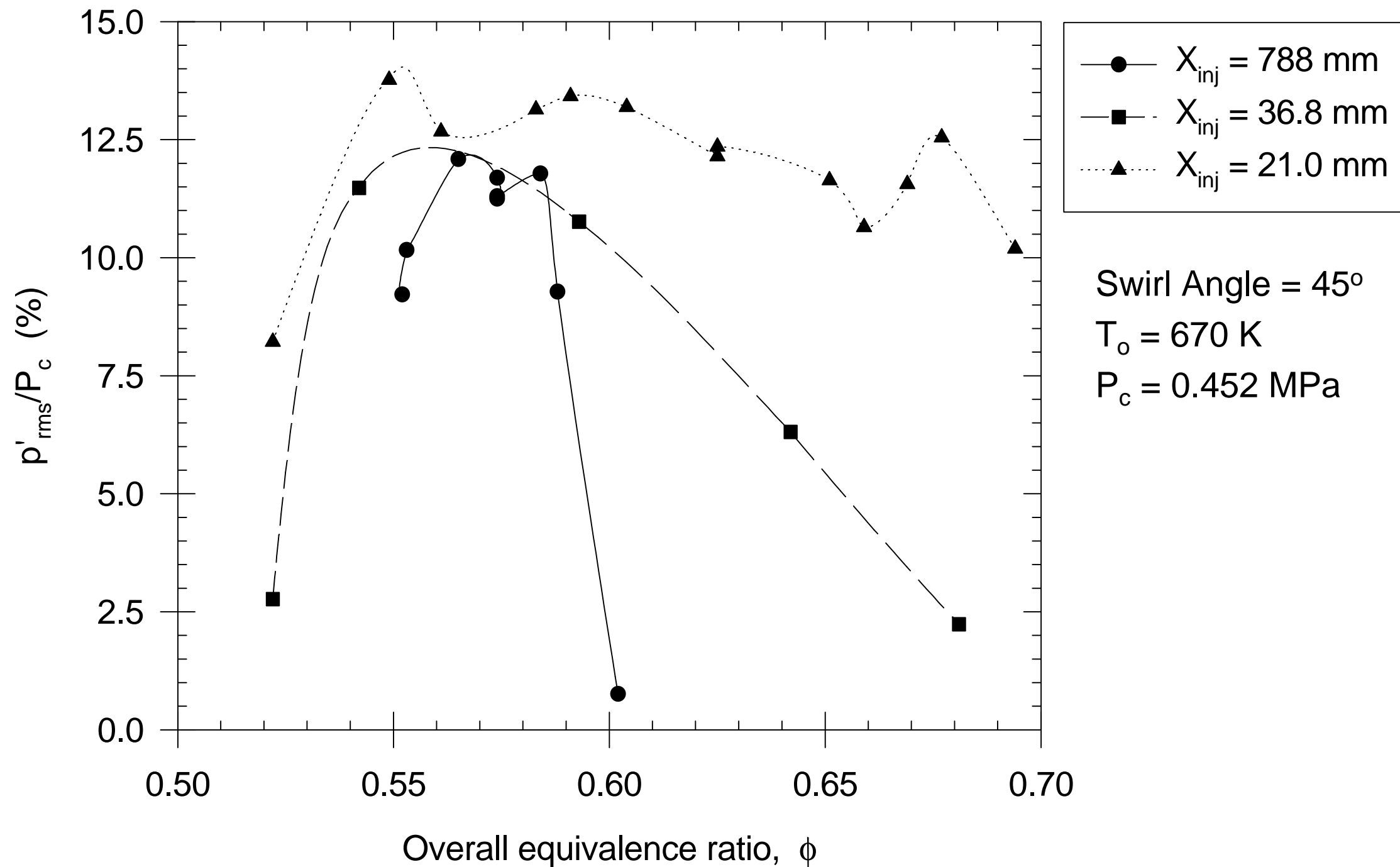


# EFFECT OF INLET AIR TEMPERATURE ON INSTABILITIES

## Comparison for Two Types of Swirl Injectors - Natural Gas



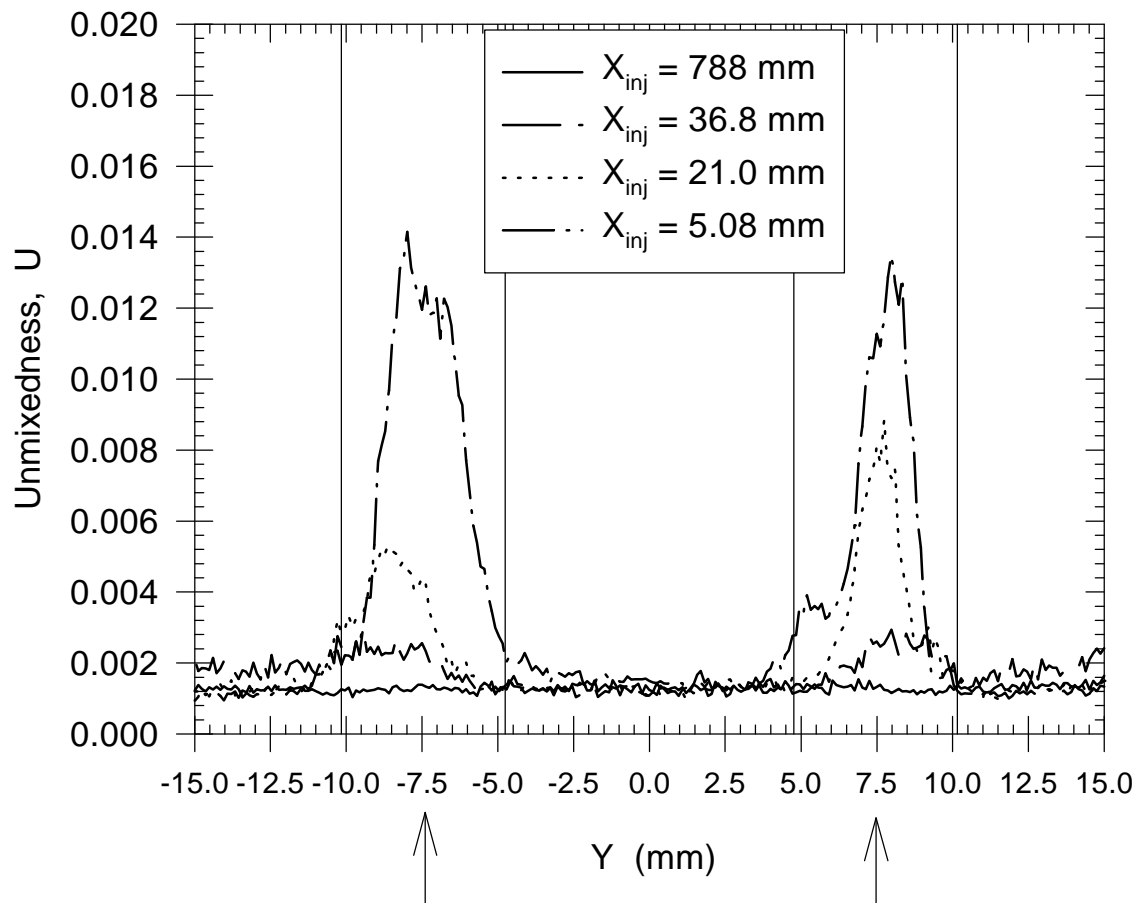
# Stability map of PSU swirler as a function of level of premixing



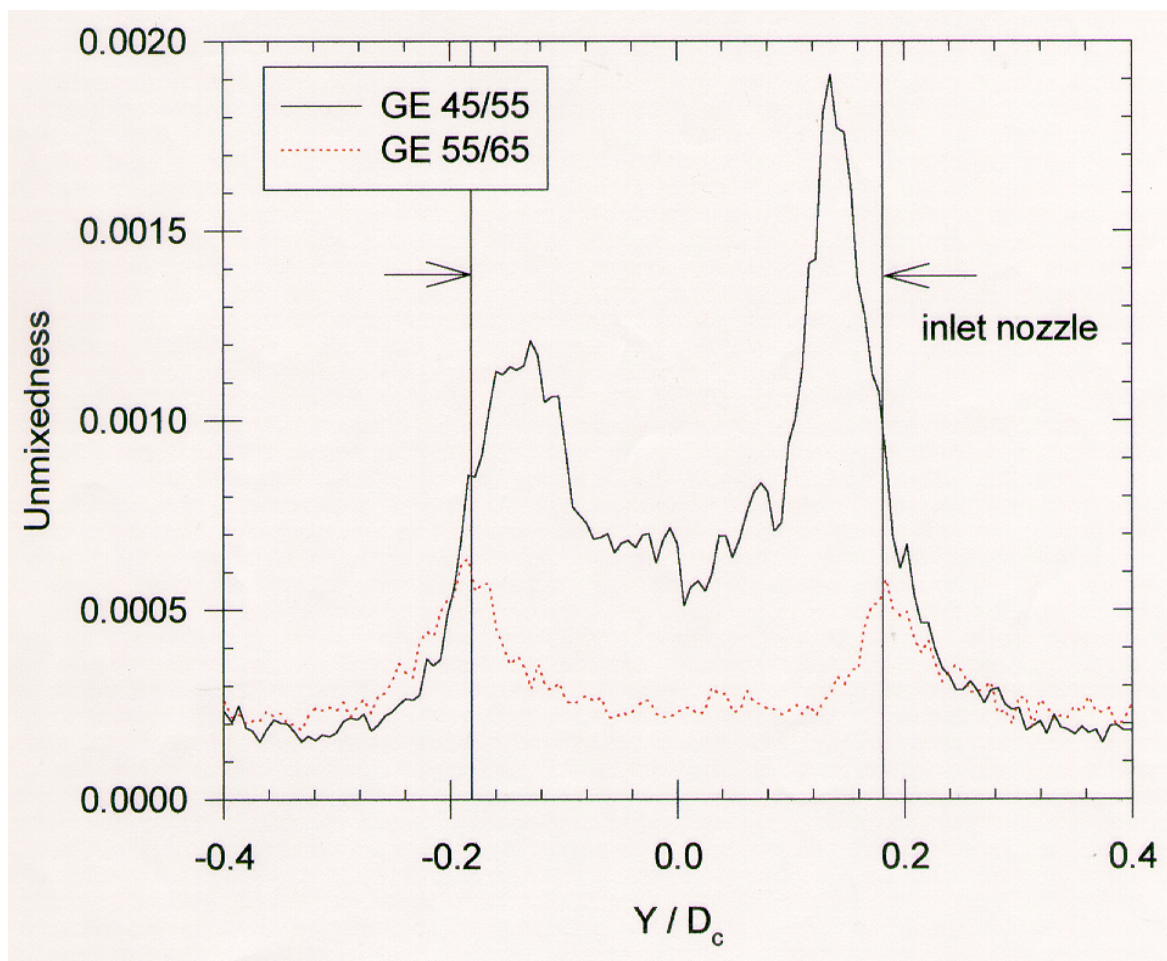


# Mixedness Profiles for Both Swirl Injector Types

PSU SWIRLER

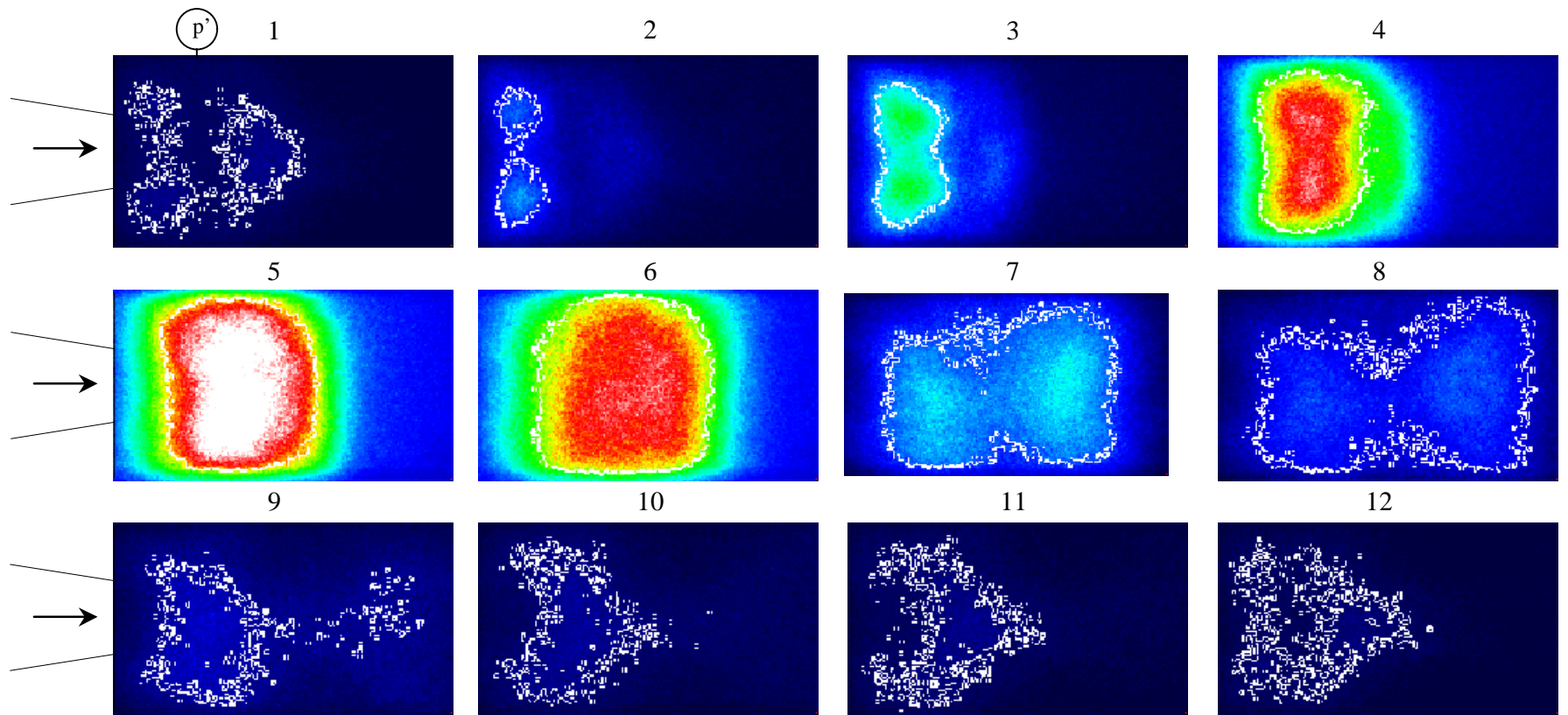


GE CRS INJECTORS

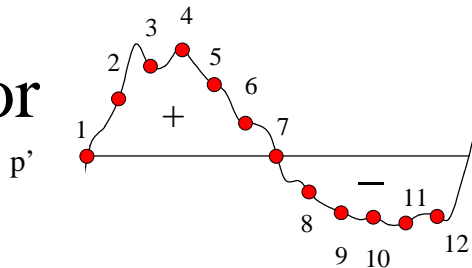




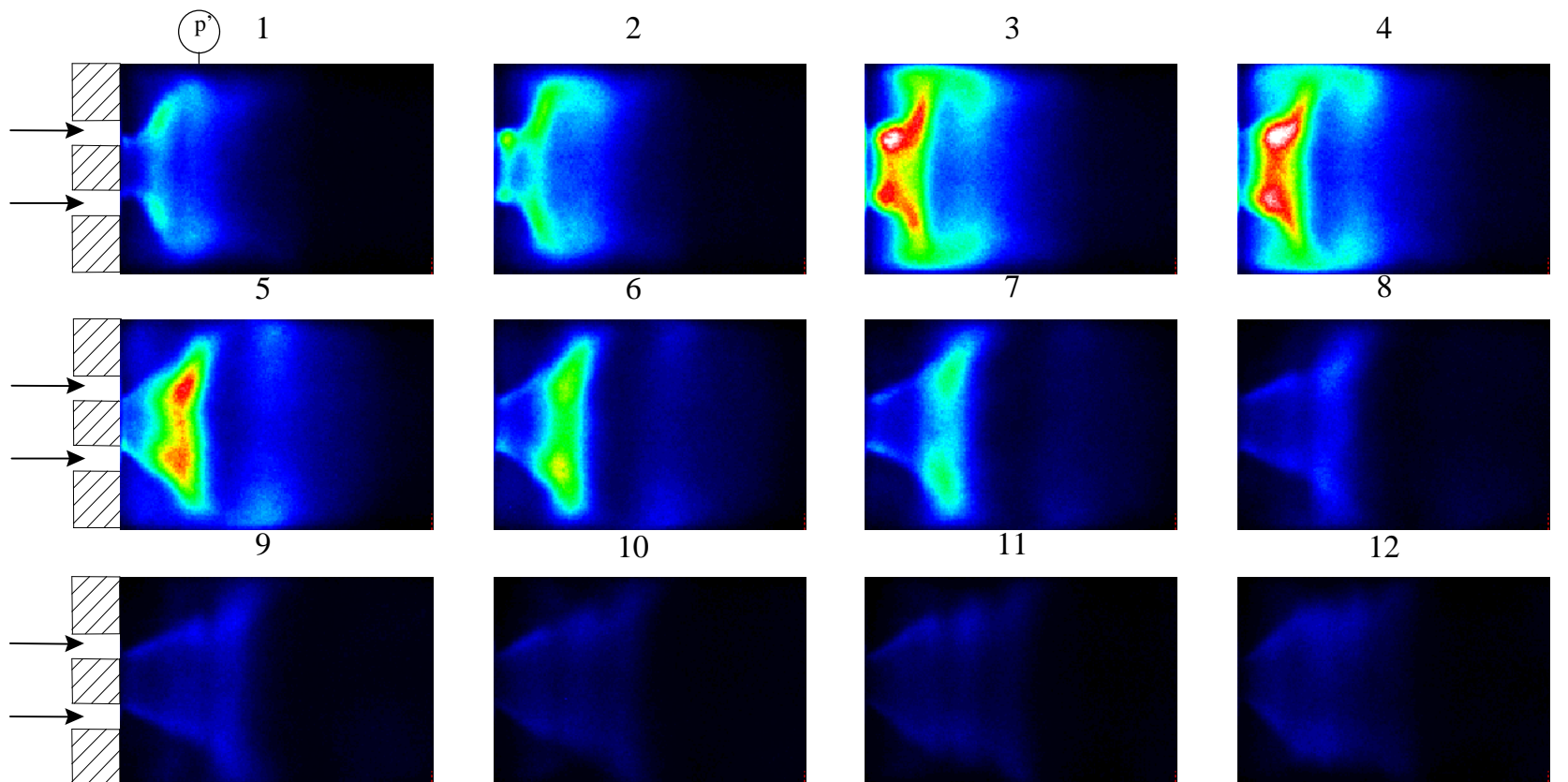
# Phase-resolved CH\* chemiluminescence images



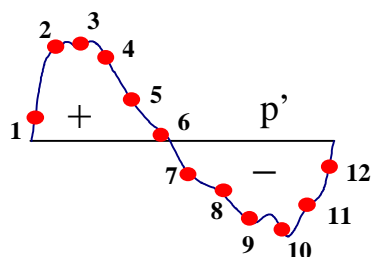
CRS Injector



$T_o = 709\text{K}$ ,  $P_c = 0.271\text{MPa}$ ,  $\phi = 0.61$ ,  
 $\varphi = 45^\circ/55^\circ$  and  $L_c = 350\text{mm}$



PSU Swirler



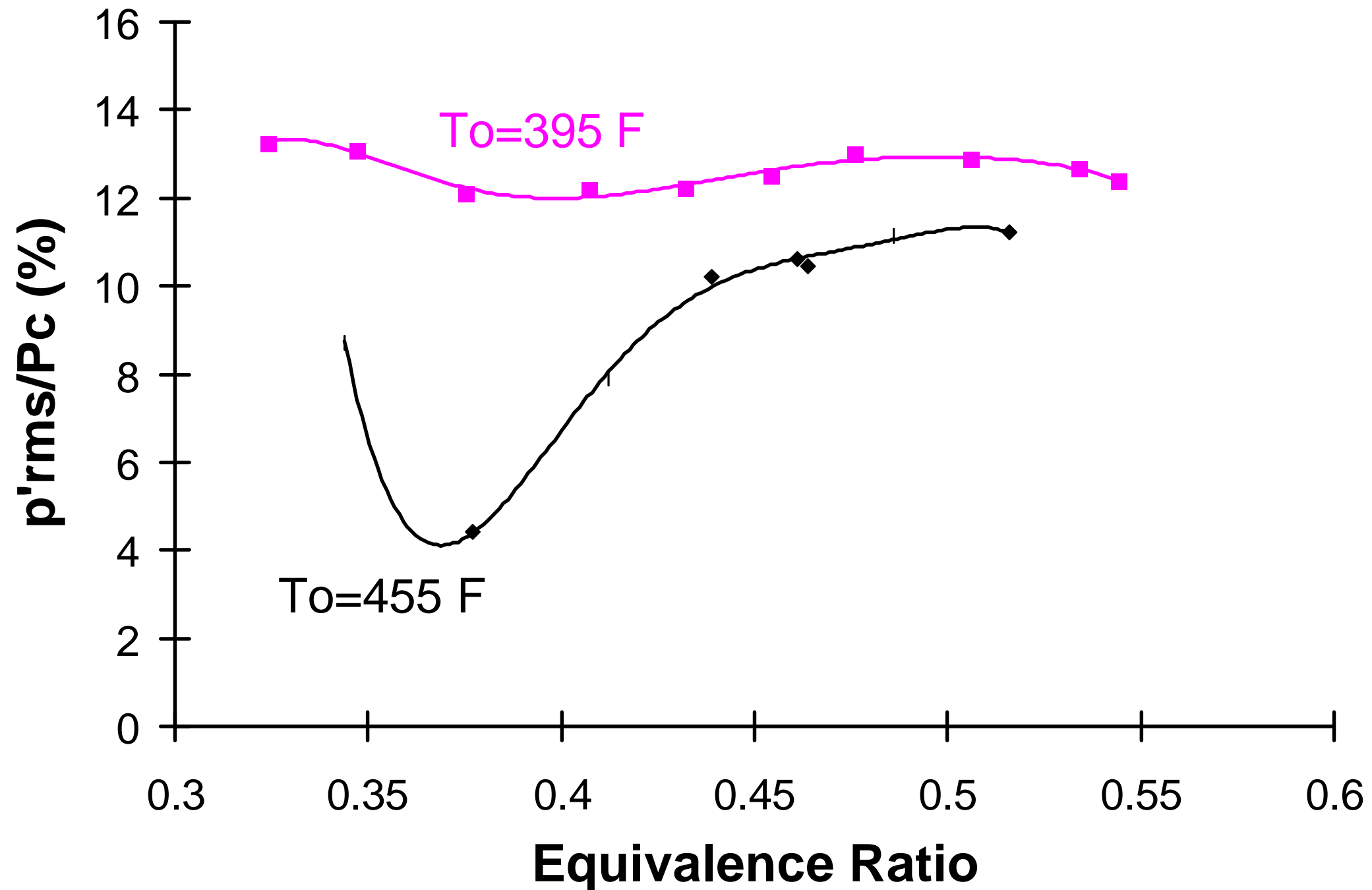
$X_{inj} = 36.8\text{ mm}$ ,  $P_c = 0.448\text{ Mpa}$ , Swirl Angle =  $45^\circ$   
 $T_o = 662\text{ K}$  and  $\phi = 0.58$  (1L mode)

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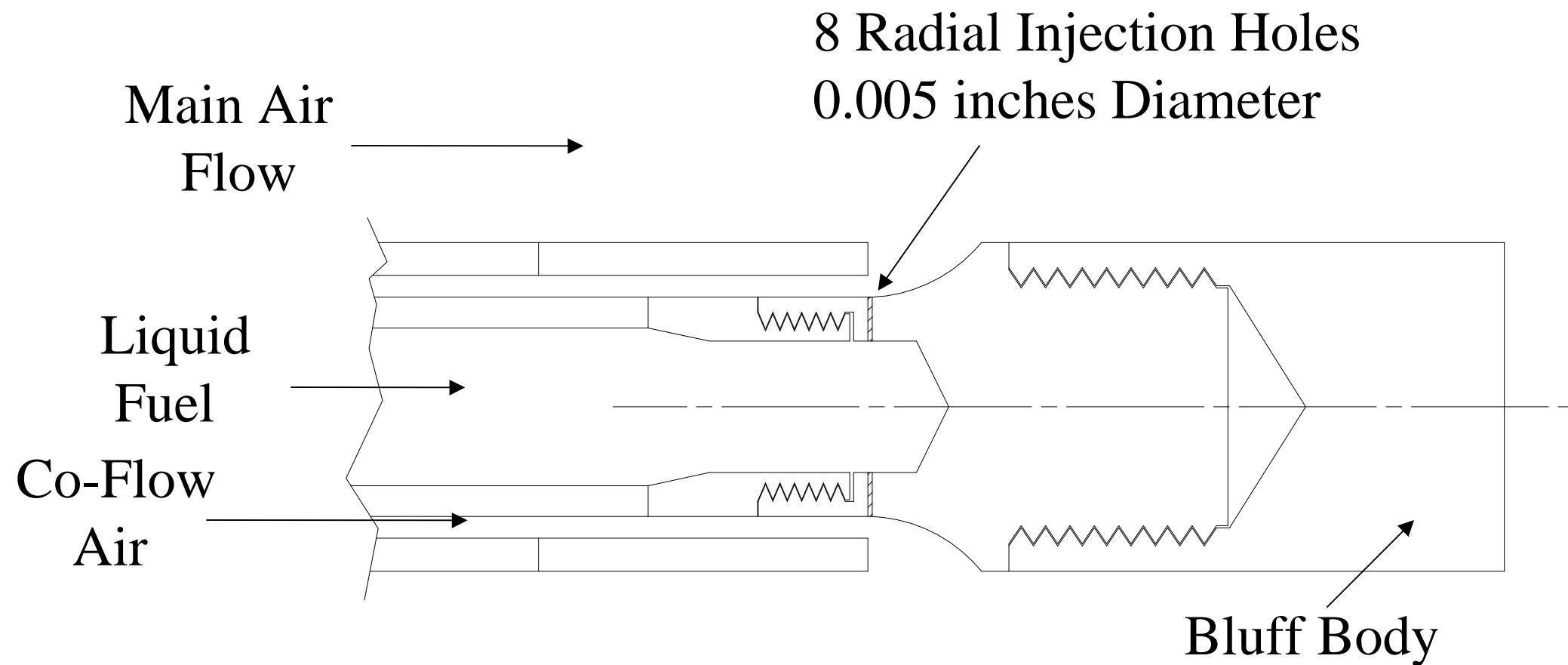


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# Stability Map for Ethylene/Air Combustion Long Chamber ( $L_c = 13.8$ in.), PSU 45 Swirl Injector



# Liquid Fuel Injector



# CONCLUSIONS AND PERSPECTIVES

- **Both types of swirl injectors produce very different stability behaviors:**  
*Short combustion chamber (235 mm long):* GE CRS Injectors produce no instabilities, PSU single swirl injector produces significant instabilities;  
*Long combustion chamber (350 mm long):* both injector types produce instabilities, but inlet temperature has opposite effects;
- **Equivalence ratio modulation is not the only mechanism** responsible for the instability behavior;
- **Controlling mechanism is complex** (equivalence ratio modulation, unmixedness, fluid mechanics effects);
- **Comprehensive results database** for natural gas will allow comparison with future data for liquid fuels

# Acknowledgements

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