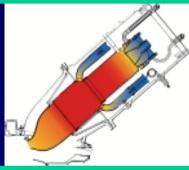
# Advanced Turbine Systems Annual Program Review









#### **Frank Macri**

ATS Program Manager Rolls-Royce Allison

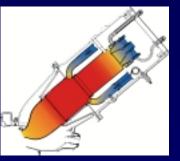
#### **November 1999 Program Update**

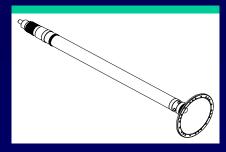
In cooperation with the US Department of Energy

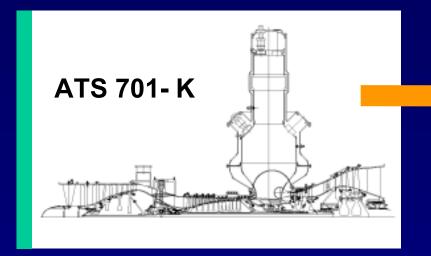




## ATS focus at Rolls-Royce Allison has changed















Allison

### ATS focus at Rolls-Royce Allison has changed

- Low / minimal risk alternative is being pursued
- No longer pursuing unique 701-K engine core
  - Market strength
  - Required investment to commercialize
- Will address market (>10 MW) with 601-K engine core
  - Core is mature and fielded
  - Achieve DOE performance goals through infusion of ATS technologies



#### **Advanced Turbine Systems program goals**

- Enhance the global competitiveness of U.S. industry
- Environmental benefits
  - Reduced emissions
  - Economic benefits
  - 15% performance improvement over state-of-the-art systems
  - Busbar cost of electricity at least 10% below current systems
- Competitive benefits
  - Successful in global markets for power generation / cogeneration and mechanical drive applications
- Commercialization of innovative technology





#### Rolls-Royce Allison ATS approach

## Technology building blocks to improve engine efficiency and reduce combustion emissions

- Combustion technology advances to achieve single digit NOx control
- Hot section material enhancements
  - Ceramic structural components
  - Advanced coating technology
- Lower bearing/sump temperatures with diffuser,
   center sump and super-critical shaft enhancements





### Rolls-Royce Allison ATS approach

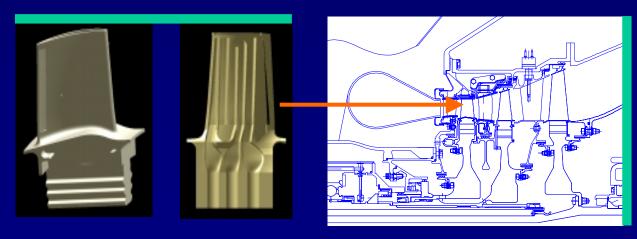
- Demonstrate/develop technologies internally and at field test sites
- Apply the demonstrated technologies to the Rolls-Royce
   Allison family of industrial engine products
  - 501-K
  - 601-K
- Commercialize these building block technologies at lower risk, lower cost
- Applicable to power generation and to mechanical drive and marine applications



### ATS advanced design concepts

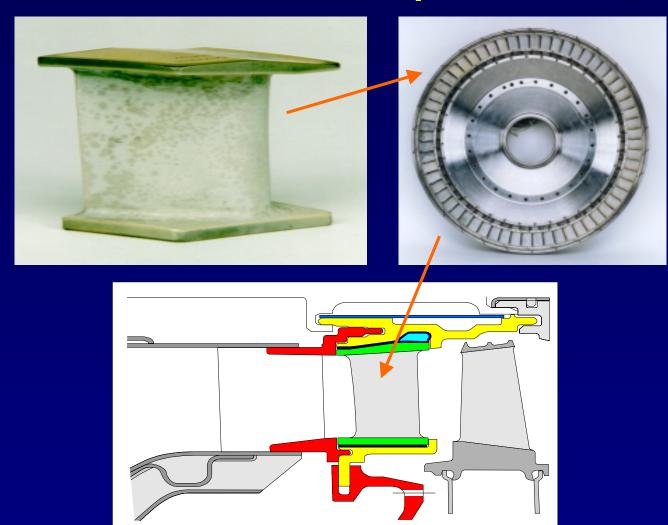
Advanced design technologies completed for insertion in future growth Model 601-K configurations

- ATS/701-K low pressure compressor to achieve significant efficiency improvement and power increase
- ATS/701-K advanced HP turbine for additional power increase
  - 2400°F turbine design
  - Blade cooling technology validated through rig test













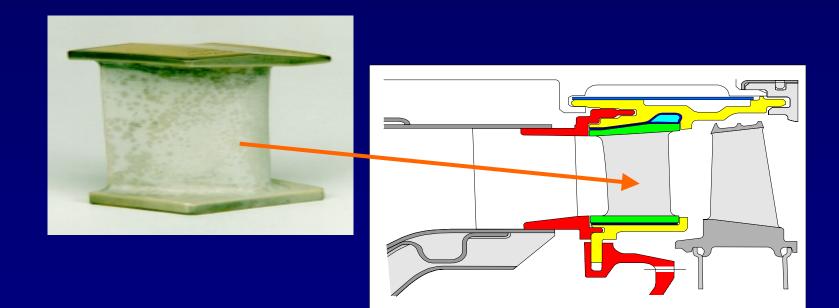
#### **Approach**

- Design and fabricate ceramic first stage turbine vane and associated mounting hardware using Model 501-K turbine architecture
- Evaluate ceramic first stage vane integrity through design analysis and thermal shock testing
- Conduct initial engine proof test
- Evaluate through 4000 hours field test EXXON facility



#### **Benefits of ceramic turbine airfoils**

- Reduce turbine module cooling flows
- Higher power, increased efficiency









EXXON Natural Gas
Processing Plant - Engine Skid

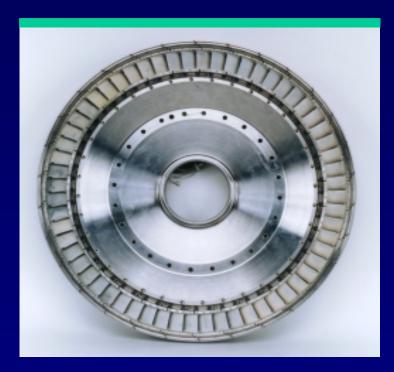


Model 501- KB5 Engine



#### Accomplishments

- Ceramic vane and mount assembly successfully designed and fabricated
- 22 hours proof testing completed at Standard AERO
- 793 hours of field test operation completed without incident over full range of power



501- K Ceramic First Stage Vane Assembly



Phase I test results support need for improved environmental barrier coating (EBC) technology to increase oxidation resistance



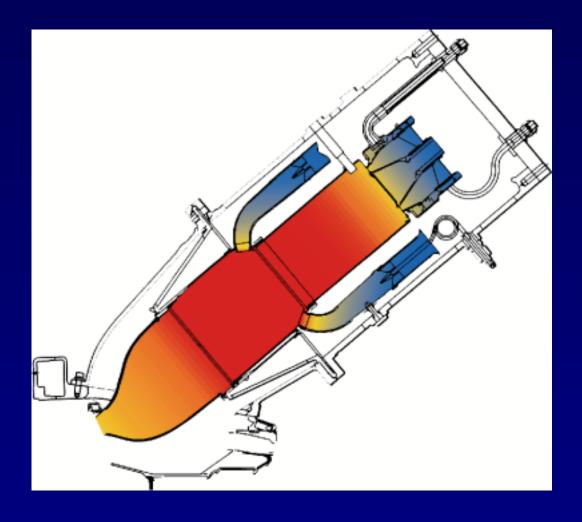
Ceramic Vane After Field Test Operation



#### **Phase II plan developed** (future activity)

- Collaborative effort to develop EBC system
  - Solar Turbines
  - Pratt & Whitney
  - Oak Ridge National Labs
  - NASA Glenn Research Center
- Develop test matrix using AS-800 and SN-282 ceramic materials for 501-K design first stage vanes with several EBC candidates
- Field test evaluation at EXXON 4000 hours









#### **Approach**

- Dry low emissions (DLE) effort is focused in two areas
  - Lean premix (LPM) technology
  - Catalytic combustion technology
- Program goal is single digit NOx emissions with stable operation
- LPM approach involves primary and secondary staging (premix) development
  - Primary staging design based on LE4 system currently used on model 501- K industrial fleet



- Design and development approach
  - Advanced CFD simulation
  - Component bench testing
  - Atmospheric diagnostics
  - System tests at simulated engine conditions
  - Engine development test
- LPM module development rig testing
  - Optimization of swirler design configurations
  - Atmospheric bench test evaluation

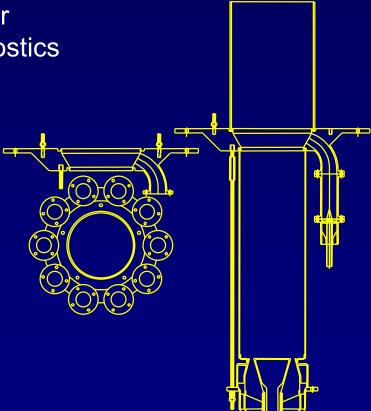


**Atmospheric Test Rig** 



#### **Atmospheric Bench Testing**

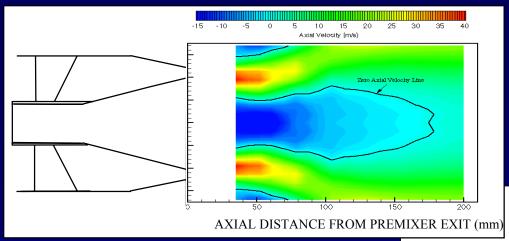
2-stage combustor with optical access for advanced diagnostics





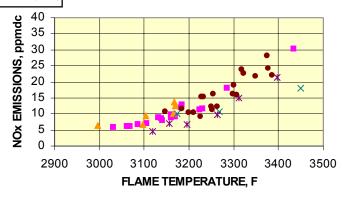


#### **Primary premixer performance characteristics**



Axial velocity field

NOx emission characteristics

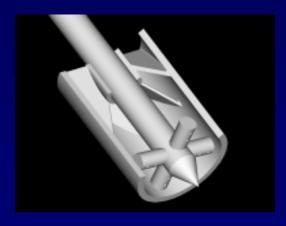


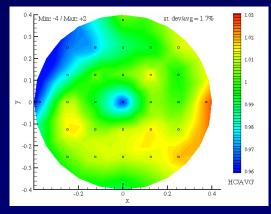




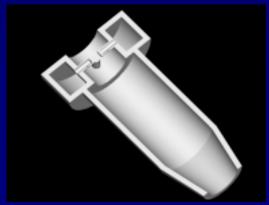
Development of secondary premixer designs have produced configurations that premix reactants to within +3 / -4 % uniformity

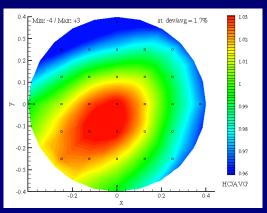
Swirler W/4
Fuel Bar Inlet





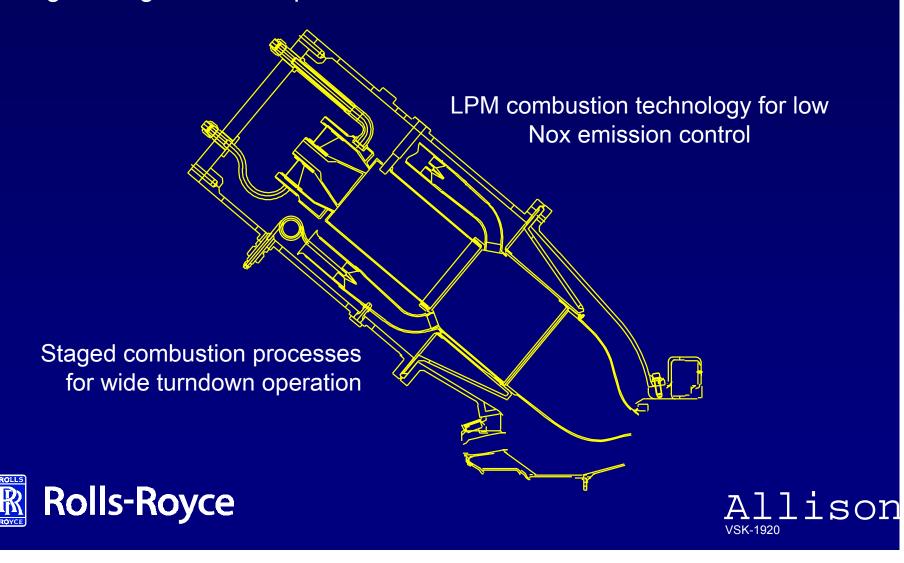
Downstream Vortex Premixer



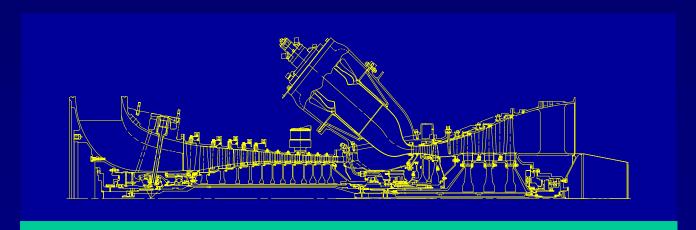




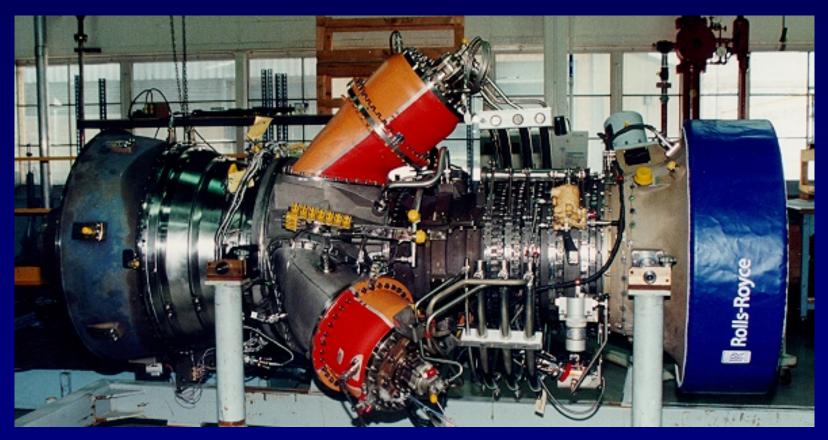
Full scale LPM combustion system preliminary design configuration completed



- Established plan to demonstrate and commercialize single digit NOx DLE technology on the Rolls-Royce Allison Model 601-K9 DLE platforms
  - 3 Can Silo-Combustion system architecture selected
  - Full scale sector rig test (1st Qtr 2000)
  - ATS engine demonstration (4th Qtr 2000)







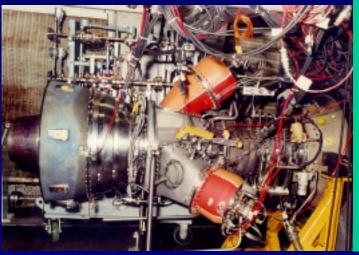
Model 601- K9 DLE





- K9 DLE engine test development currently in progress
- Features RB211 combustion platform
- 25 ppm NOx emissions guarantee







- Initiated a cooperative program among Solar, Catalytica, and Rolls-Royce Allison to test catalyst durability
  - Mechanical system evaluation
  - Durability at high pressure, high temperature conditions
  - Susceptibility to contaminants
    - Fuel
    - Lubricants
    - Salt (marine environments)
    - Dust and other inlet contaminants







#### **Objective**

Demonstrate enhanced durability thermal barrier coatings in industrial engine applications

#### **Approach**

- Procure 501-K first stage turbine vanes in an improved material (Mar-M247)
- Conduct cyclic thermal exposure tests of coated coupons to select coating system
- Coat vanes with selected coating system
- Subject coated vane to cyclic testing
- Install and run coated vanes in field engines





#### **Benefits**

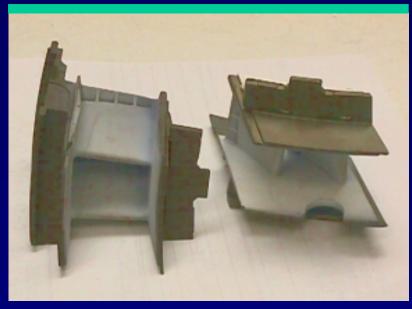
- Enhanced durability of first stage vanes when compared with existing vane alloy/coating system
- Reduced operating expense to end user based upon increased time between overhaul
- Improved performance resulting from higher temperature operation



#### **Accomplishments**

- Coupons have been coated using three candidate thermal barrier coating systems
- Coated coupons are undergoing cyclic thermal exposure testing
- MAR-M247 vanes have been cast and are awaiting machining
- Tooling for coating vanes designed and fabricated





Model 501- K Vane Configuration

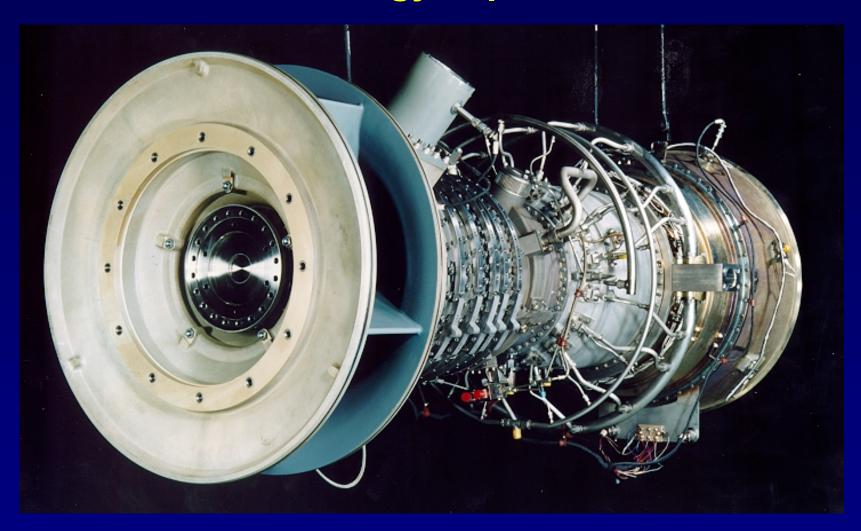


**Coated Specimens** 

#### **Future Activity**

- Finish machine vanes and apply optimized coating system to 501-K first stage vanes for engine installation
- Field test evaluation of coated vanes at specific 501-K user sites
- Apply coating technology to 601-K first stage vanes and demonstrate during 4th Qtr 2000 ATS engine test









#### **Approach**

- Mechanical technology improvements for 601-K focused on the following areas:
  - Improved diffuser design
  - Super-critical shaft integration
  - Center sump bearing simplification

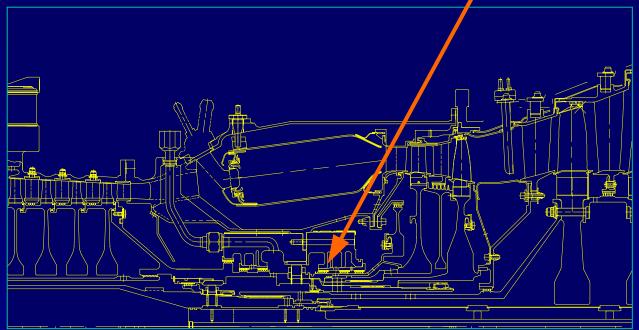
#### **Benefits**

 Eliminates need for inter-shaft bearing and sump arrangement - reduces engine cost



Original shafting and center sump configuration

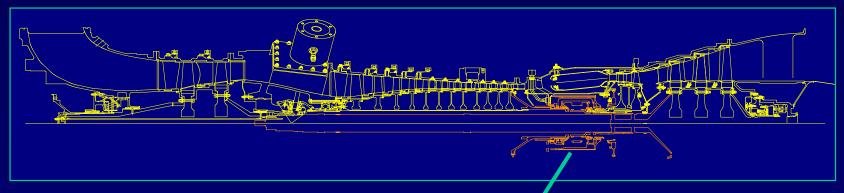


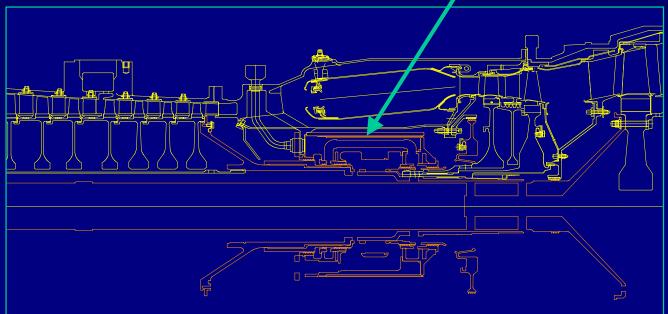






Super-critical shaft with simplified sump arrangement









#### **Future Activity**

- Complete diffuser, shafting and sump design
- Develop technology under 601-K11 program
- Demonstrate technology enhancement in ATS 601-K9 engine configuration (4th Qtr 2000)



## **ATS technologies**



