# National Transportation Safety Board 

Washington, D.C. 20594<br>Safety Recommendation

## Date: January 20, 1987

In reply refer to: A-87-1 and -2

Honorable Donald D. Engen<br>Administrator<br>Federal Aviation Administration<br>Washington, D.C. 20591

On March 22, 1985, a Swearingen SA 226 TC airplane, operated as Sun Aire Flight 534, made an unintentional gear-up landing at Los Angeles International Airport, Los Angeles, California. Of the 14 occupants on board, only 1 person, a passenger who was seated in the plane of rotation of the right engine propeller, was seriously injured. The injury resulted when pieces of the engine propeller blades penetrated the side of the airplane cabin and struck the passenger. On January 9, 1983, Republic Airlines, Flight 927, a Convair 580, touched down on a snow-covered runway at Brainerd, Minnesota, and struck a 2 - to 3 -foot high snowbank near the right edge of the runway. Of the 33 occupants on board, 1 passenger was killed and 1 passenger was injured seriously when a broken propeller blade penetrated the cabin. Both of these passengers were seated in the plane of rotation of the engine propeller blades. More recently, on October 30, 1986, a passenger seated in the propeller plane of rotation aboard a Wings West, SA 226TC suffered injuries to the right leg when a piece of the right propeller blade went through the fuselage after the airplane made a gear up landing at Santa Barbara, California. This accident is still under investigation by the National Transportation Safety Board.

The Safety Board is aware of at least four other accidents and incidents since 1981 in which the airplane cabin interior adjacent to the engine propellers was damaged when pieces of propeller blades penetrated the cabin. Fortunately, the passenger seats adjacent to the propellers were unoccupied during each of these four occurrences, and, as a result, there were no associated injuries.

In the cases cited above, the majority of occurrences which precipitated the propeller blade penetrations were landing gear malfunctions which caused a spinning propeller blade to separate or fracture when it hit a solid object such as a runway surface. However, in one occurrence noted, a blade separation and penetration into the cabin occurred in-flight due to an engine failure.

Based on the foregoing, the Safety Board believes that attention is needed concerning the design of seating configurations or the incorporation of shielding to reduce the potential for injury to occupants seated in areas within the plane of rotation of engine propeller blades. The Board realizes that because of weight and cost penalties the amount of cabin shielding needed to prevent propeller blade penetration may be prohibitive with respect to retrofit of airplanes now in service. Further, the removal of
passenger seats adjacent to the plane of propeller rotation of airplanes currently in service may not be practical due to airplane operating limitations and associated operating cost penalties. However, if these features are considered during the initial design phases of propeller-powered airplanes, then a reduced potential for occupant injuries might be achieved without a major increase of cost or degradation of performance.

Therefore, the National Transportation Safety Board recommends that the Federal Aviation Administration:

Initiate action to evaluate the feasibility of incorporating design features in new propeller-powered airplanes, which, in the event of a propeller blade separation or fracture, would reduce the potential for injury to persons seated in areas within the plane of rotation of the engine propellers or which would provide for passenger seating arrangements totally outside of the plane of propeller rotation. (Class III-Longer Term Action) (A-87-1)

If it is determined that design features can be incorporated in new propeller-powered airplanes that will reduce the potential of injury to persons seated in areas within the plane of rotation of engine propellers or that will provide for passenger seating arrangements totally outside the plane of propeller rotation, take the necessary action to include these features into 14 CFR 23 and 14 CFR 25 design requirements. (Class III-Longer Term Action) (A-87-2)

Brief of Accident/Incident

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