

National Transportation Safety Board

Washington, D. C. 20594

Safety Recommendation

Log 2010

Date: March 30, 1989

In reply refer to: A-89-14

Mr. Robert E. Whittington Acting Administrator Federal Aviation Administration Washington, D.C. 20591

On November 15, 1987, at Akron, Colorado, a fuel-injected Piper model PA-24-260 "B" Comanche airplane, N9039P, sustained an engine stoppage while on an instrument flight rules (IFR) flight plan in instrument meteorological conditions (IMC) conducive to induction system icing. The airplane had been flying in clouds at below freezing temperature for approximately 45 minutes and was level at 9,000 feet mean sea level when the emergency occurred. Despite the use of appropriate emergency procedures, e.g., switching fuel tanks and turning the electric fuel boost pump on, the pilot was unable to restart the engine. An attempt was then made to glide to the Akron-Washington County Airport, but after breaking out of the clouds at 400 feet above ground level, the pilot landed in an open field with the landing gear retracted. The airplane was damaged substantially and the three occupants aboard the airplane sustained minor injuries.\(^1\) A similar accident involving engine stoppage in a fuel-injected PA-24-250 Comanche,\(^2\) N8185P, occurred on November 20, 1982, at Cascade Locks, Oregon. In this case, the airplane was also damaged substantially during the forced landing, but no one was injured.\(^3\)

The National Transportation Safety Board's investigation of the accidents disclosed, in both cases, that the loss of engine power occurred after snow/ice blocked the induction air filters and prevented the flow of ram air to the engines. The engine air induction box assemblies on fuel-injected PA-24-260 airplanes produced from 1965 through 1968 and on fuel-injected PA-24-250 airplanes contain a spring-loaded alternate air door downstream of the air filter at the throat of the servo regulator. The door, which is acted upon by engine vacuum/suction, is intended to open automatically and provide an alternate source of heated air in the event of blockage of the induction air filter. However, the alternate air doors in N9039P and N8185P failed to operate, apparently because they had been frozen shut by moisture which had entered this portion of the air induction box before engine stoppage. The design of the alternate air systems in these airplanes differs from those installed in older

³For more detailed information, read Field Accident Brief No. 2549 (attached).

¹For more detailed information, read Field Accident Brief No. 1523 (attached).

²Fuel-injected engines in Piper model PA-24-250 airplanes installed only under Supplemental Type Certificate (STC) No. SA811WE.

carbureted and post-1968 fuel-injected PA-24 models in two important aspects: (1) they contain no provision for manual operation of the alternate air doors, and (2) they use ambient engine compartment air rather than the much hotter air available from an exhaust heater muff typically used in connection with preventing fuel system icing.

On May 4, 1987, Piper issued Service Bulletin No. 861, Heated Alternate Air Induction System Installation. The bulletin, applicable to fuel-injected PA-24-260 airplanes with spring-loaded alternate air doors, reannounced the availability of a manually operated alternate air door installation kit. The availability of this kit had been announced previously in Piper Service Spares Letter No. SP-310, Manual Alternate Air Induction System Installation, which was issued on April 12, 1971. Piper considers compliance with the bulletin to be mandatory and recommends that it be accomplished at the next regularly scheduled maintenance event, but not to exceed 50 hours of operation. Piper explains the purpose of the bulletin as follows:

It has been determined that operation in icing conditions can cause ice to accumulate in the fuel injection system, resulting in a loss of engine power.

This service bulletin announces the continued availability of a manually operated heated alternate air induction system kit, Piper part number 760-516, which, when installed, will provide the ability to manually select heated alternate air when desired or when conditions dictate and help prevent the above described condition.

In view of the foregoing, the Safety Board believes that the Piper kits, or their equivalent, should be installed in all applicable fuel-injected PA-24 series airplanes to ensure a reliable source of heated alternate airflow to the engine during flight in IMC at altitudes at or above the freezing level.

Therefore, the National Transportation Safety Board recommends that the Federal Aviation Administration:

Issue an airworthiness directive applicable to Piper fuel-injected PA-24-250 and PA-24-260 Comanche airplanes requiring a manually operated, heated alternate air induction system kit, Piper part No. 760-516, or its equivalent, to be installed in accordance with Piper Service Bulletin No. 861 or other appropriate service instructions. (Class II, Priority Action) (A-89-14)

KOLSTAD, Acting Chairman, and BURNETT, LAUBER, NALL, and DICKINSON, Members, concurred in this recommendation.

James L. Kolstad Acting Chairman

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Brief of Accident

File No	File No 1523	11/15/87 AKRON		A/C Red. No. N9039F	—	Time (Lcl) - 1542 HST	1542 HST	
Type Operating Cert	ormation-	ificate-NONE (GENERA	1	Aircraft Damase		. Indicate	Injuries	
Type of Flight C Accident	Type of Operation Flisht Conducted Under Accident Occurred Durins	n Under -14 CFR 91 d Durins -LANDING	SURS SURS Fire NONE	SUBSTANTIAL ire Crew NONE Pass	Fætal 0	Serious 0 0	Minar 1 2	N 0 0 0
Hake/Hodel - PIPER Landing Gear - TRICYC Hax Gross Wt - 3100 No. of Seats - 6	Gear Line Constants and Consta	mation PIPER FA-24-260B - TRICYCLE-RETRACTABLE - 3100	End Make/Hodel - (Number Engines - Engine Type - R	LYCOMING IO-S40-D4AS I I RECIF-FUEL INJECTED 240 HF	ELT	ELT Installed/Activated		- YES/YES YES
Weather Data We Briefing We Briefing Rethod Completenes: Basic Weather Wind Dir/Spr Visibility Lowest Sky/ Lowest Sky/ Lowest Ceil: Obstructions Precipitation	ther Data Priefing	Weather Data Wx Briefind - FSS Wx Briefind - FSS Method - TELEFHONE Completeness - FUL Basic Weather - INC Wind Dir/Speed - 330/016 KTS Visibility - I.000 SM Lowest Sky/Clouds - N/A Lowest Ceilind - A00 FT OVERCAST Obstructions to Vision- BLOWING SNOW Precipitation - ICE PELLETS Condition of Light - DAYLIGHT	Itinerary Last Departure Foint ENGLEWOOD,CO Destination DES MOINES,IA ATC/Airspace Type of Flight Plan Type of Clearance Type of Clearance	nt n - 1FR - FORCED LANDING	Airport Proxim OFF AIRPORT/ Airport Data AKRON-WASHIN RUNWAS Ident RUNWAS Lth/W RUNWAS Surfair	ximits RT/STRIP HINGTON (ent - h/Wid - rface - atus	00 N/A N/A A/N A/A A/A	1
Fersonnel Information Filot-In-Command Certificate(s)/Rating(COMMERCIAL SE LAND	nei Informat; -In-Command tificate(s)/R COMMERCIAL SE LAND	sonnel Information lot-In-Command Certificate(s)/Rating(s) COMMERCIAL SE LAND	Age - 42 Biennial Flight Review Current - YES Months Since - 13 Alrcraft Type - FA-24	Hedical Certi Total Make/Mode Instrumen	ficate - VALID MEDI Flight Time (Hours) - 1091 1- 249 t- 254	MEDICAL-WAI Durs) Last 24 Last 30 Last 90	IVERS/LIMIT Hrs - 7 Days- UNK/NR Days- 37	NR 37

- AIRPLANE Instrument Rating(s)

BER TKOF, SNOW WAS MELTED OFF THE ACFT IN A HEATED HANGAR. THE FLT WAS BGN IN BLOWING SNOW WITH A FORECAST OF OCNL HOD ICING TO 20,000°, THE PLT SAID THE SNOW LASTED ONLY S TO 10 MIN \$ ONLY A TRACE OF ICE (LESS 1/16 INCH) ACCUMULATED DRG THE IST 15 MIN OF FLT, NO FURTHER ACCUMULATION OF ICE WAS NOTED, APRX 45 MIN AFTER TKOF, WHILE CRUISING IN INC CONDUCIVE TO INDUCTION SYS ICING, THE ENG SUDDENLY LOST PWR \$ WOULD NOT RESTART, THE PLT ISCNAD TWD AN ARFT IN IT TO APRX 400°, AGL, BUT MADE A WHEELS-UP LNDG WHEN HE COULD NOT SEE THE ARPT, HE ATTRIBUTED THE LOSS OF FWR TO BLOCKAGE OF THE AIR INDUCTION FILTER. THE INDUCTION SYS WAS EQUIPPED WITH A SPRING LOADED DOOR WHICH WAS INTENDED TO OFEN AUTOMATICALLY \$ ALLOW HEATED ALTN AIR TO FLOW TO THE ENG IN THE EVENT THE FILTER BCM BLOCKED. NONETHELESS, FIFER INDCD THAT OFN IN ICING CONDS COULD CAUSE ICE TO ACCUMULATE IN THE FUEL INJECTED AND IN THIS, PIFER SUC BULLETIN WO: 861 WAS ISSUED ON 5/4/87 FOR INSTLN OF AN IMPROVED MANUALLY OPERATED ALTN AIR SYS ON FUEL INJECTED FA-24-260 ACFT, ---Narrative---

File No 15	11/15/87 AKRON,CO A/C Res. No. N9039	P Time (Lc1) - 1542 HST
Occurrence #1 Phase of Operation	NON-MECHANICAL	
<pre>finding(s) finding(s)</pre>	CONTROL/SYSTEM -	
 MAINTENANCE, S WEATHER CONDITI 	MAINTENANCE, SERVICE BULLETINS - NOT PERFORMED - COMPANY/OPERATOR MGHT WEATHER CONDITION - ICING CONDITIONS	
 WCHICEN CONDITION - SNOW S. INDUCTION AIR CONTROL/SY 6. INDUCTION AIR CONTROL, AI 	5. INDUCTION AIR CONTROL/SYSTEM - ICE 6. INDUCTION AIR CONTROL/SYSTEM - ICE	
Occurrence #2 Phase of Operation	nce #2 FORCED LANDING f Operation DESCENT - EMERGENCY	
Occurrence #3	! !	
Phase of Operation	LANDING - FLARE/TOUCHDOWN	

Finding(s)

7. WEATHER CONDITION - LOW CEILING

8. WHEELS UP LANDING

----Probable Cause----

The National Transportation Safety Board determines that the Probable Cause(s) of this accident is/are finding(s) $5r\delta$

Factor(s) relating to this accident is/are finding(s) 2,3,4,7

National Transpo ion Safety Board Washington: D.C. 20594

Brief of Accident

Rasic Information Type Operating Certificate-NONE (GENERAL		11111111111111			
		(ND	Aircraft Damase	ur 1e	
Type of Operation Flight Conducted Under Accident Occurred During	-PERSONAL -14 CFR 91 Ins -LANDING	7. 4.1. F.	Crew	5001420	0 1 0 2 0 2 2 0 0 1 0 0 0 0 0 0 0 0 0 0
Make/Model - PIPER- Landing Gear - TRICYC Max Gross Wt - 2900 No. of Seats - 4	tion PIPER FA-24-250 - TRICYCLE-RETRACTABLE - 2900	Eng Hake/Model - LY Number Engines - 1 Engine Type - RE Rated Power -	- LYCOMING IO-540-C1BS - 1 - RECIF - FUEL INJECTED - 250 HF	ELT Installed/Activated - YES/YES Stall Warning System - YES Weather Radar - NO	ated - YES/YES
Weather Data Wx Briefins Ecompleteness Completeness FUL Basic Weather Wind Dir/Speed- 220/005 KTS Visibility Cloud Conditions(1st) - 4000 FT B Cloud Conditions(1st) - 7000 FT D Cloud Conditions(1st) - RAIN Condition of Lisht - DAYLIGHT Precipitation Certificate(s)/Rating(s) PRIVATE SE LAND	ROKEN VERCA P	Itinerary Last Departure Point HERMISTON,OR Destination NORTH BEND,OR ATC/Airspace Type of Flight Plan ST Type of Elearance Type of Clearance Type of Clearance Type of Clearance Type of Clearance Type of Flight Review Current Current AS Since - OR		Airport Proximity ON AIRPORT CASCADE LOCKS ST. STATE Runway Ident - 24 Runway Ident - 1800/ 30 Runway Lth/Wid - 1800/ 30 Runway Status - ASFHALT Runway Status - SNOW - WET VISUAL STRAIGHT-IN Medical Certificate - VALID MEDICAL-NO WAIVERS/LIMIT Flight Time (Hours) Total - 309 Last 24 Hrs - 3 Hake/Model- 224 Last 30 Days- 12 Instrument- 63 Last 90 Days- 51	STATE 24 1800/ 30 ASFHALT SNOW - WET WAIVERS/LIMIT Hrs - 3 Tays- 51

Instrument Rating(s) - AIRPLANE

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factor(s) relating to this accident is/are finding(s) 3,5,6,7	The National Transportation Safety Roand date	Finding(s) 5. LANDING GEAR - OVERLOAD 6. WEATHER CONDITION - RAIN 7. WEATHER CONDITION - SNOW	Finding(s) 4. GROUND LOOP/SWERVE - INTENTIONAL - PILOT IN COMMAND	Occurrence #2 ABRUPT HANEUVER Phase of Operation LANDING - ROLL	Finding(s) 1. FUEL SYSTEM, RAM AIR - BLOCKED(PARTIAL) 2. FUEL SYSTEM, RAM AIR - ICE 3. WEATHER CONDITION - ICING CONDITIONS	Occurrence #1 FORCED LANDING Phase of Operation CRUISE - NORMAL	File No 2549 11/20/82 CASCADE LOCKS,OR
Probable Cause(s) of this accident						,	Brief of Accident (Continued) OCKS,OR A/C Res. No. N8185P