Pt. 125, App. D

Parameters	Range	Accuracy sensor input to DFDR readout	Sampling inter- val (per second)	Resolution ⁴ read out 1 sec.
rime (GMT or Frame Counter) (range 0 to 4095, sampled 1	24 Hrs	±0.125% Per Hour	0.25 (1 per 4 seconds).	
per frame). Iltitude	 1,000 ft to max certificated altitude of aircraft. 	±100 to ±700 ft (See Table 1, TSO-C51a).	1	5' to 35' 1
irspeed	50 KIAS to V _{so} , and V _{so} to 1.2 V _D .	±5%, ±3%	1	1 kt.
leading Iormal Acceleration (Vertical)	360° -3g to +6g	±2° ±1% of max range excluding datum error of ±5%.	1	0.5° 0.01g.
Pitch Attitude Roll Attitude Radio Transmitter Keying	±75° ±180° On-Off (Discrete)	±2°	1	0.5°. 0.5°.
hrust/Power on Each Engine railing Edge Flap or Cockpit	Full range forward Full range or each dis-	±2% ±3° or as pilot's Indicator	1	0.2% ² 0.5% ²
Control Selection. eading Edge Flap or Cockpit Control Selection.	crete position. Full range or each discrete position.	±3° or as pilot's indicator	0.5	0.5% 2
hrust Reverser Position	Stowed, in transit, and reverse (Discrete).		1 (per 4 sec- onds per en- gine).	
Brake Selection.	Full range or each discrete position. Discrete	±2% unless higher accuracy uniquely required.	1	0.2% 2.
Marker Beacon Passage Autopilot Engagement congitudinal Acceleration	Discrete	±1.5% max range excluding	1 4	0.01g
Pilot Input and/or Surface Posi-	Full range	datum error of ±5%. ±2° unless higher accuracy	1	0.2% 2.
tion-Primary Controls (Pitch, Roll, Yaw) ³ .		uniquely required.	4	0.01-
ateral Acceleration	±1g Full range	±1.5% max range excluding datum error of ±5%. ±3% unless higher accuracy	1	0.01g. 0.3% ²
Glideslope Deviation	±400 Microamps	uniquely required.	1	0.3%2
ocalizer DeviationAFCS Mode and Engagement	±400 Microamps Discrete	±3%	1	0.3% 2.
Status. Radio Altitude	-20 ft to 2,500 ft	±2 Ft or ±3% Whichever is Greater Below 500 Ft and ±5% Above 500 Ft.		1 ft + 5% ² above 500'.
Master Warning	Discrete		1	
Main Gear Squat Switch Status angle of Attack (if recorded di-	As installed	As installed	1	0.3% ² .
rectly). Dutside Air Temperature or Total Air Temperature.	-50° C to +90° C	±2° C	0.5	0.3° C
Hydraulics, Each System Low Pressure.	Discrete		0.5	or 0.5% ² .
Groundspeed	As Installed	Most Accurate Systems Installed (IMS Equipped Aircraft Only).	1	0.2% 2.
If additional recording capacity is		e following parameters is recomr	nended. The param	eters are listed
Orift Angle	When available. As installed.	As installed	4	
Vind Speed and Direction	When available. As installed.	As installed	4	
atitude and Longitude	When available. As installed.	As installed	4	
Brake pressure/Brake pedal position. Additional engine parameters:	As installed	As installed	1	
EPR	As installed	As installed	1 (per engine)	
N ¹	As installed	As installed	1 (per engine)	
N ²	As installed	As installed	1 (per engine)	
EGT	As installed	As installed	1 (per engine)	
hrottle Lever Position	As installed	As installed	1 (per engine)	

Federal Aviation Administration, DOT

Parameters	Parameters Range Accuracy sensor input to DFDR readout		Sampling interval (per second)	Resolution 4 read out
Fuel FlowTCAS:	As installed	As installed	1 (per engine)	
TA	As installed	As installed	1	
RA	As installed	As installed	1	
Sensitivity level (as se- lected by crew).	As installed	As installed	2	
GPWS (ground proximity warning system).	Discrete		1	
Landing gear or gear selector position.	Discrete		0.25 (1 per 4 seconds).	
DME 1 and 2 Distance	0–200 NM;	As installed	0.25	1 mi.
Nav 1 and 2 Frequency Selection.	Full range	As installed	0.25	

[Doc. No. 25530, 53 FR 26150, July 11, 1988; 53 FR 30906, Aug. 16, 1988]

APPENDIX E TO PART 125—AIRPLANE FLIGHT RECORDER SPECIFICATIONS

The recorded values must meet the designated range, resolution, and accuracy requirements during dynamic and static conditions. All data recorded must be correlated in time to within one second.

Parameters	Range	Accuracy (sensor input)	Seconds per sampling interval	Resolution	Remarks
1. Time or Relative Times Counts.1.	24 Hrs, 0 to 4095.	±0.125% Per Hour.	4	1 sec	UTC time preferred when available. Count increments each 4 seconds of system operation.
Pressure Altitude.	- 1000 ft to max certificated alti- tude of aircraft. +5000 ft.	±100 to ±700 ft (see table, TSO C124a or TSO C51a).	1	5' to 35'	Data should be obtained from the air data computer when practicable.
 Indicated air- speed or Cali- brated airspeed. 	50 KIAS or minimum value to Max V _{so} , to 1.2 V _{.D} .	±5% and ±3%	1	1 kt	Data should be obtained from the air data computer when practicable.
 Heading (Primary flight crew reference). 	0–360° and Discrete "true" or "mag".	±2°	1	0.5°	When true or magnetic head- ing can be selected as the primary heading reference, a discrete indicating selec- tion must be recorded.
 Normal Acceleration (Vertical)⁹. 	-3g to +6g	±1% of max range exclud- ing datum error of ±5%.	0.125	0.004g.	
6. Pitch Attitude	±75°	±2°	1 or 0.25 for air- planes oper- ated under § 125.226(f).	0.5°	A sampling rate of 0.25 is recommended.
7. Roll Attitude ²	±180°	±2°	1 or 0.5 for air- planes oper- ated under § 121.344(f).	0.5°	A sampling rate of 0.5 is recommended.
Manual Radio Transmitter Keying or CVR/ DFDR synchro- nization reference	On-Off (Discrete) None.		1		Preferably each crew mem- ber but one discrete ac- ceptable for all trans- mission provided the CVR/ FDR system complies with TSO C124a CVR synchro- nization requirements (paragraph 4.2.1 ED-55).

¹ When altitude rate is recorded. Altitude rate must have sufficient resolution and sampling to permit the derivation of altitude to 5 feet.

2 Percent of full range.

3 For airplanes that can demonstrate the capability of deriving either the control input on control movement (one from the other) for all modes of operation and flight regimes, the "or" applies. For airplanes with non-mechanical control systems (fly-by-wire) the "and" applies. In airplanes with split surfaces, suitable combination of inputs is acceptable in lieu of recording each surface separately.

4 This column applies to aircraft manufactured after October 11, 1991.