Service Bulletin A310–53–2125, including Appendix 01, dated January 9, 2007, at the applicable compliance times listed in Table 1 (threshold) or Table 2 (grace period) of this AD, whichever occurs later.

Airbus model	Whichever occurs first after the effective date of this AD	
	Accumulated time since first flight (in flight cycles)	Accumulated time since first flight (in flight hours)
A310–200 A310–300 with an average flight time (AFT) \leq to 4 hours A310–300 with an AFT >4 hours	41,500 33,000 20,500	83,500 93,500 102,000

TABLE 2.—GRACE PERIODS

Airbus model	Whichever occurs first after the effective date of this AD	
	Flight cycles	Flight hours
A310–200 A310–300 with an average flight time (AFT) \leq 4 hours A310–300 with an AFT >4 hours	1,500 1,200 740	3,000 3,400 3,600

FAA AD Differences

Note: This AD differs from the MCAI and/ or service information as follows:

No differences.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Tom Stafford, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1622; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

Related Information

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2007– 0111, dated April 25, 2007; and Airbus Service Bulletin A310–53–2125, dated January 9, 2007; for related information. Issued in Renton, Washington, on August 28, 2007.

Stephen P. Boyd,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. E7–17686 Filed 9–6–07; 8:45 am] BILLING CODE 4910-13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-28413; Directorate Identifier 2007-NE-25-AD]

RIN 2120-AA64

Airworthiness Directives; General Electric Company (GE) CF6–80C2 and CF6–80E1 Series Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT). **ACTION:** Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for GE CF6–80C2 and CF6–80E1 series turbofan engines with fuel manifold part numbers (P/Ns) 1303M31G12 and 1303M32G12 installed. This proposed AD would require removing and discarding the loop clamps that assemble the fuel manifold to the compressor rear frame (CRF) friction damper brackets, visually inspecting the fuel manifold for wear at each clamp location, and replacing the clamps with new, zero-time parts. This proposed AD results from fuel manifold vibration

during engine operation that causes the loop clamps that assemble the manifold to the CRF to deteriorate. Fourteen fuel leak events occurred over the past several years. We are proposing this AD to prevent fuel leaks during engine operation that could result in an undercowl fire.

DATES: We must receive any comments on this proposed AD by November 6, 2007.

ADDRESSES: Use one of the following addresses to comment on this proposed AD.

• DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending your comments electronically.

• Government-wide rulemaking Web site: Go to http://www.regulations.gov and follow the instructions for sending your comments electronically.

• *Mail:* U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

• *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

• Fax: (202) 493-2251.

You may examine the comments on this proposed AD in the AD docket on the Internet at *http://dms.dot.gov.*, or in Room W12–140 on the ground floor of the West Building, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Robert Green, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; telephone (781) 238–7754; fax (781) 238–7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send us any written relevant data, views, or arguments regarding this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA– 2007–28413; Directorate Identifier 2007–NE–25–AD" in the subject line of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to http:// dms.dot.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of the DOT Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the Federal Register published on April 11, 2000 (65 FR 19477–78) or you may visit http:// dms.dot.gov.

Examining the AD Docket

You may examine the AD docket on the Internet at *http://dms.dot.gov*; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is the same as the Mail address provided in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

Discussion

During a pre-flight inspection in November 2002, a manifold fuel leak was found on a CF6–80C2B1F turbofan engine. This fuel leak resulted from fuel manifold vibration during engine operation, which caused the loop clamps that assemble the manifold to the CRF to deteriorate. The cushion on the loop clamp wore away resulting in the metal strip of the clamp rubbing into the fuel manifold. A fuel leak resulted. The high surface temperature of the CRF during engine operation could have ignited the leaking fuel resulting in an under-cowl fire.

In 2003, GE issued service bulletins for the CF6-80C2 and CF6-80E1series turbofan engines recommending repetitive visual inspection of the fuel manifolds at intervals of 4,500 flighthours (FH). Despite these recommendations, five fuel leak events occurred in 2006; one resulting in an under-cowl fire on a CF6-80C2B6F turbofan engine. There have been three additional fuel leak events to date in 2007. A total of thirteen CF6-80C2 series turbofan engine fuel leaks (November 2002 through May 2007) and one CF6-80E1 fuel leak (January 2007) have been reported. Five of the recent events occurred within the 4,500 FH since the last inspection.

Because of the increased number of fuel leak events and the low FH intervals since the last inspection, this proposed AD would reduce the initial re-inspection FH interval, and would require replacing the fuel manifold loop clamps with new, zero-time parts at each inspection. This proposed AD is necessary to prevent fuel leaks during engine operation that could result in an under-cowl fire.

Differences Between the Proposed AD and the Manufacturer's Service Information

In the manufacturer's original service bulletins, GE recommends inspecting the fuel manifold for wear at 4,500 FH. However, the service bulletins do not recommend replacing the loop clamps at each inspection.

Because recent field events involving fuel leaks indicate a reduced inspection interval is required, this AD would require removing loop clamps and inspecting the fuel manifold according to the following schedule:

• For engines previously inspected using GEAE service bulletins, SB 73– 0226, dated March 5, 2003, for CF6– 80C2 engines; or SB 73–0061, dated April 14, 2003, for CF6–80E1 engines, within 1,750 FH since-last-inspection.

• If used loop clamps or clamps of unknown heritage were installed at last shop visit, within 1,750 FH since-lastshop-visit.

• For first-run engines and engines with zero-time, new loop clamps installed at last shop visit, within 7,500 FH.

• For affected engines that have already exceeded the 1,750 FH initial inspection threshold, within 4,500 FH of the last inspection or 4 months after the effective date of the AD, whichever comes first.

FAA's Determination and Requirements of the Proposed AD

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other products of this same type design. We are proposing this AD, which would require removing and discarding the loop clamps that assemble the fuel manifold to the compressor rear frame damper (CRF) friction damper brackets, visually inspecting the fuel manifold for wear at each loop clamp location, and replacing the clamps with new, zero-time parts.

Costs of Compliance

We estimate that this proposed AD would affect 350 CF6–80C2 series turbofan engines installed on airplanes of U.S. registry. We also estimate that it would take about 4 work-hours per engine to perform the proposed actions, and that the average labor rate is \$80 per work-hour. Required parts would cost about \$162 per engine. Based on these figures, we estimate the total cost of the proposed AD to U.S. operators for one manifold visual inspection and loop clamp replacement to be \$168,700.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed AD:

1. Is not a "significant regulatory action" under Executive Order 12866;

2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

3. Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. You may get a copy of this summary at the address listed under **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

General Electric Company: Docket No. FAA– 2007–28413; Directorate Identifier 2007– NE–25–AD.

Comments Due Date

(a) The Federal Aviation Administration (FAA) must receive comments on this airworthiness directive (AD) action by November 6, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to:

(1) GE CF6-80C2A1, -80C2A2, -80C2A3, -80C2A5, -80C2A8, -80C2A5F, -80C2B1, -80C2B2, -80C2B4, -80C2B6, -80C2B1F, -80C2B1F1, -80C2B1F2, -80C2B2F, -80C2B3F, -80C2B4F, -80C2B5F, -80C2B6FA, -80C2B6FA, -80C2B7F, -80C2B8F, -80C2D1F, -80C2L1F, -80C2K1F turbofan engine models configured with fuel manifold part numbers (P/Ns) 1303M31G12 and 1303M32G12. These engines are installed on, but not limited to Boeing 747, 767, MD11, and Airbus A300 and A310 airplanes. (2) This AD also applies to GE CF6-

80E1A1, -80E1A2, -80E1A3, -80E1A4, -80E1A4/B turbofan engine models with fuel

TABLE 1.—INSPECTION SCHEDULE FOR INITIAL INSPECTIONS

manifold P/N's 1303M31G12 and 1303M32G12. These engines are installed on Airbus A330 airplanes.

Unsafe Condition

(d) This AD results from fuel manifold vibration during engine operation that causes the clamps that assemble the manifold to the compressor rear frame (CRF) to deteriorate. We are issuing this AD to prevent fuel leaks. The high surface temperature of the CRF could ignite the leaking fuel resulting in an under-cowl fire.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Removal and Replacement of Loop Clamps and Fuel Manifold Inspection Compliance Times

(f) Remove and discard the loop clamps, P/N J1220G10, that assemble the fuel manifold to the Compressor Rear Frame friction damper brackets, inspect the fuel manifold for wear at each clamp location as specified in paragraph (g) of this AD, and replace the loop clamps with new, zero-time parts at each inspection. Use the following schedule in Table 1 of this AD:

lf:	Then replace clamps and inspect within:
(1) The engine was previously inspected using GEAE service bulletins, SB 73–0226, dated March 5, 2003, for CF6–80C2 engines; or SB 73–0061, dated April 14, 2003, for CF6–80E1 engines.	1,750 flight hours (FH) time-since-last-inspection (TSLI).
(2) Used loop clamps or clamps of unknown heritage were installed at last shop visit.	1,750 FH time-since-last-shop-visit.
(3) The engine is a first-run engine or is an engine with zero-time, new loop clamps previously installed.	7,500 FH since zero-time, new loop clamps were installed.
(4) The engine has already exceeded the 1,750 FH initial inspection threshold.	4,500 FH TSLI, or 4 months after the effective date of this AD, which- ever occurs first.

Inspection of Fuel Manifold P/Ns 1303M31G12 and 1303M32G12

(g) Remove any tape applied to clamp locations. Visually inspect the full circumference of the manifold at each clamp location.

(h) If the fuel manifold shows any signs of wear, determine the depth of the wear as follows:

(1) Measure the outside diameter of the tube using a pinpoint micrometer adjacent to the worn area.

(2) Measure the worn area at the smallest diameter.

(3) Subtract the measurement of the worn tube diameter from the unworn diameter measurement. Allowable wear is 0.0035 inch.

(i) Before further flight, replace fuel

manifolds with wear greater than 0.010 inch. (ii) Replace fuel manifolds with wear

greater than 0.0035 inch, but less than 0.010 inch, within 50 flight cycles.

Revise Air Carrier's Continuous Airworthiness Maintenance Program (CAMP) and Airworthiness Limitation Section (ALS)

(i) Within 30 days of the effective date of this AD, revise the air carrier's approved CAMP and ALS of Chapter 5 in the CF6– 80C2 and CF6–80E1 Instructions for Continued Airworthiness to add:

(1) Repetitive inspections of fuel manifolds, P/Ns 1303M31G12 and 1303M32G12, as detailed in paragraphs (g) and (h) of this AD, at 7,500 FH intervals.

(2) Mandatory removal of loop clamps, P/ N J1220G10, that assemble the fuel manifold, P/Ns 1303M31G12 and 1303M32G12, to the Compressor Rear Frame friction damper brackets, at each inspection.

(3) Replacement of loop clamps, P/N J1220G10, with new, zero-time parts, at each inspection.

Alternative Methods of Compliance

(j) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Related Information

(k) GE service bulletins GEAE CF6–80C2 S/ B 73–0226, dated March 5, 2003, and GEAE CF6–80E1 S/B 73–0061, dated April 14, 2003; and the following General Electric engine manuals pertain to the subject of this AD: CF6–80C2 Engine Manual GEK 92451; CF6–80C2L1F Engine Manual GEK 112213; CF6–80C2K1F Engine Manual GEK 112721; and CF6–80E1 Engine Manual GEK 99376.

(l) Contact Robert Green, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA, 01803; e-mail: *robert.green@faa.gov;* telephone (781) 238–7754; fax (781) 238– 7199, for more information about this AD. Issued in Burlington, Massachusetts, on August 30, 2007.

Thomas A. Boudreau,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service. [FR Doc. E7–17682 Filed 9–6–07; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2006-28778; Airspace Docket No. 07-AGL-6]

Proposed Establishment of Class E5 Airspace; Prairie Du Sac, WI

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Notice of proposed rulemaking.

SUMMARY: This action proposes to establish Class E airspace at Prairie Du Sac, WI. Additional controlled airspace is necessary to accommodate aircraft using a new Area Navigation (RNAV) Global Positioning System (GPS) Instrument Approach Procedure (IAP) at Sauk-Prairie Airport. The FAA is proposing this action to enhance the safety and management of aircraft operations at Sauk-Prairie Airport, Prairie Du Sac, WI.

DATES: Comments must be received on or before October 1, 2007.

ADDRESSES: Send comments on this proposal to the Docket Management System, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590. You must identify the docket number FAA-2007-28778/Airspace Docket No. 07-AGL-6, at the beginning of your comments. You may also submit comments on the Internet at *http://dms.dot.gov* You may review the public docket containing the proposal, any comments received, and any final disposition in person in the Dockets Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone 1-800-647-5527) is on the ground floor of the building at the above address.

FOR FURTHER INFORMATION CONTACT:

Grant Nichols, System Support, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; at telephone number (816) 329–2522.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested parties are invited to participate in this proposal rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall regulatory, aeronautical, economic, environmental, and energy-related aspects of the proposal. Communications should identify both docket umbers and be submitted in triplicate to the address listed above. Commenters wishing the FAA to acknowledge receipt of their comments on this notice must submit with those comments a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. FAA-2007-28778/Airspace Docket No. 07-AGL-6." The postcard will be date/time stamped and returned to the commenter.

Availability of NPRM's

An electronic copy of this document may be downloaded through the internet at *http://dms.dot.gov*. Recently published rulemaking documents can also be accessed through the FAA's Web page at *http://www.faa.gov* or the Superintendent of Document's Web page at *http://www.access.gpo.gov/nara*.

Additionally, any person may obtain a copy of this notice by submitting a request to the Federal Aviation Administration (FAA), Office of Air Traffic Airspace Management, ATA-400, 800 Independence Avenue, SW., Washington, DC 20591, or by calling (202) 267-8783. Communications must identify both docket numbers for this notice. Persons interested in being placed on a mailing list for future copies of NPRMs should contact the FAA's Office of Rulemaking (202) 267-9677, to request a copy of Advisory Circular No. 11-2A, Notice of Proposed Rulemaking Distribution System, which describes the application procedure.

The Proposal

This action proposes to amend Title 14, Code of Federal Regulations (14 CFR) part 71, by establishing Class E airspace extending upward from 700 feet above the surface at Sauk-Prairie Airport, Prairie Du Sac, WI. aAnew RNAV (GPS) IAP has made this action necessary for the safety of IFR operations at Sauk-Prairie Airport, Prairie Du Sac, WI. This area would be depicted on appropriate aeronautical charts. Class E airspace areas extending upward from 700 feet or more above the surface of the earth are published in Paragraph 6005 of FAA Order 7400.9P, dated September 1, 2006, and effective September 15, 2006, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designations listed in this document would be published subsequently in the Order.

The FAA has determined that this proposed regulation only involves an established body of technical regulations for which frequent and routine amendments are necessary to keep them operationally current. It, therefore, (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034; February 26, 1979); and (3) does not warrant preparation of a Regulatory Evaluation as the anticipated impact is so minimal. Since this is a routine matter that will only affect air traffic procedures and air navigation, it is certified that this rule, when promulgated, will not have a significant economic impact on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 71

Airspace, Incorporation by reference, Navigation (Air).

The Proposed Amendment

In consideration of the foregoing, the Federal Aviation Administration proposes to amend 14 CFR part 71 as follows:

PART 71—DESIGNATION OF CLASS A, CLASS B, CLASS C, CLASS D, AND CLASS E AIRSPACE AREAS; AIRWAYS; ROUTES; AND REPORTING POINTS

1. The authority citation for part 71 continues to read as follows:

Authority: 49 U.S.C. 106(g); 40103, 40113, 40120; E.O. 10854, 24 FR 9565, 3 CFR 1959– 1963 Comp., p. 389.

§71.1 [Amended]

2. The incorporation by reference in 14 CFR 71.1 of Federal Aviation Administration Order 7400.9P, Airspace Designations and Reporting Points, dated September 1, 2006, and effective September 15, 2006, is amended as follows:

Paragraph 6005 Class E Airspace Areas Extending Upward From 700 Feet or More Above the Surface of the Earth.

AGL E5 Prairie Du Sac, WI [New]

Sauk-Prairie Airport, Prairie Du Sac, WI