category, equipped with any Dowty or Messier-Dowty main landing gear (MLG) listed in Table 1 of this AD.

TABLE 1.—AFFECTED MLGs.

MLG having part number (P/N)—	Which have sliding member P/N—
201072011	201072301 or 201072305.
201072012	201072301 or 201072305.
201072013	201072301 or 201072305.
201012014	201072303. 201072301 or 201072305.
201072015	201072301 or
201072016	201072305. 201072301 or 201072305.

Compliance: Required as indicated, unless accomplished previously.

To prevent fatigue cracking of the sliding member, which could result in possible separation of the MLG from the airplane and consequent reduced controllability of the airplane upon landing and possible injury to passengers, accomplish the following:

### Inspection and Replacement If Necessary

(a) Within 1,000 flight cycles or six months after the effective date of this AD, whichever occurs first, perform a magnetic inspection of the sliding members of the MLG for cracking, in accordance with the Accomplishment Instructions of Fokker Service Bulletin SBF100–32–133, dated April 1, 2002. If any crack is found during the inspection, before further flight, replace the sliding members with serviceable parts in accordance with the Accomplishment Instructions of the service bulletin.

**Note 1:** Fokker Service Bulletin SBF100–32–133, dated April 1, 2002, refers to Messier-Dowty Service Bulletin F100–32–103, dated March 11, 2002, as an additional source of service information.

#### **Parts Installation**

(b) As of the effective date of this AD, no person may install a sliding member of the MLG, P/N 201072301 or P/N 201072305, on any airplane, unless it has been inspected in accordance with the Accomplishment Instructions of Fokker Service Bulletin SBF100-32-133, dated April 1, 2002, and found to be serviceable.

#### Reporting Requirement Difference

(c) Although the service bulletin referenced in this AD specifies to submit certain information to the manufacturer, this AD does not include such a requirement.

#### **Alternative Methods of Compliance**

(d) In accordance with 14 CFR 39.19, the Manager, International Branch, FAA, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

**Note 2:** The subject of this AD is addressed in Dutch airworthiness directive 2002–060, dated April 29, 2002.

Issued in Renton, Washington, on January 9, 2004.

#### Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 04–1307 Filed 1–21–04; 8:45 am] BILLING CODE 4910–13–P

#### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. 2002-NM-232-AD] RIN 2120-AA64

### Airworthiness Directives; Dassault Model Mystere-Falcon 50 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Dassault Model Mystere-Falcon 50 series airplanes. This proposal would require one-time detailed inspections for structural discrepancies of various fuselage attachments; and corrective actions, if necessary, to restore the structure to the original design specifications. This action is necessary to prevent early fatigue, corrosion, or fretting, which could result in structural failure of major components of the airplane and reduced structural integrity of the airplane. This action is intended to address the identified unsafe condition.

**DATES:** Comments must be received by February 23, 2004.

**ADDRESSES:** Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 2002-NM-232-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anmnprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2002-NM-232-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from Dassault Falcon Jet, PO Box 2000, South Hackensack, New Jersey 07606. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1137; fax (425) 227-1149.

#### SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002–NM–232–AD." The postcard will be date stamped and returned to the commenter.

#### **Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the

FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-232-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

#### Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, notified the FAA that an unsafe condition may exist on certain Dassault Model Mystere-Falcon 50 series airplanes. The DGAC advises that nonconformities to the original design specifications of the airplane were recorded at assembly for various fuselage attachments. Those nonconformities to specifications, if not corrected, could result in early fatigue, corrosion, or fretting, which could cause structural failure of major components, and reduced structural integrity of the airplane.

# **Explanation of Relevant Service Information**

Dassault Aviation has issued Service Bulletin F50-332, dated March 13, 2002, which describes procedures for detailed inspections for structural discrepancies (e.g., missing rivets, and loose or non-reinforced rivets and screws) of various fuselage attachments. The inspection areas include points on the cabin inner structure, the stub, and the outer structure. This service bulletin also describes procedures for corrective actions, if necessary, to restore the structure to the original design specifications. The corrective actions include installing new shims, installing new reinforcement fittings, re-torquing or re-installing screws, and installing missing rivets, as applicable, at the appropriate point(s) of the fuselage. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition. The DGAC classified this service bulletin as mandatory and issued French airworthiness directive 2002-033-039(B) R1, dated May 15, 2002, to ensure the continued airworthiness of these airplanes in France.

### **FAA's Conclusions**

This airplane model is manufactured in France and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DGAC, reviewed all available

information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

# **Explanation of Requirements of Proposed Rule**

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously, except as discussed below.

# Difference Between the Service Bulletin and This Proposed AD

Although the service bulletin specifies to submit a reporting card to the manufacturer, this proposed AD would not include such a requirement.

# Cost Impact

The FAA estimates that 21 airplanes of U.S. registry would be affected by this proposed AD. The average labor rate is \$65 per work hour. The estimated work hours per inspection are between 5 hours and 123 hours, depending on the operating point(s) that would be inspected. The estimated cost per airplane is between \$325 and \$7,995.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

#### **Regulatory Impact**

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant

economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

### The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**Dassault Aviation:** Docket 2002–NM–232–AD.

Applicability: Model Mystere-Falcon 50 series airplanes, having serial numbers (S/N) 253 through 278 inclusive; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent early fatigue, corrosion, or fretting, which could result in structural failure of major components, and possible reduced structural integrity of the airplane, accomplish the following:

#### **Inspections and Corrective Actions**

(a) Within 78 months after the effective date of this AD, perform one-time detailed inspection(s) for structural discrepancies of the fuselage attachments at all applicable operating points specified in paragraph 2.B. of the Accomplishment Instructions of Dassault Service Bulletin F50–332, dated March 13, 2002. Perform the inspections in accordance with the Accomplishment Instructions of the service bulletin.

Note 1: For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

(b) If any structural discrepancy of the fuselage attachments (e.g., missing rivets, and

loose or un-reinforced rivets and screws) is found during the inspections required by paragraph (a) of this AD: Prior to further flight, accomplish all applicable corrective actions (e.g., installing new shims, installing new reinforcement fittings, re-torquing or reinstalling screws, and installing missing rivets), as applicable, at the appropriate operating point(s) of the fuselage, in accordance with the Accomplishment Instructions of Dassault Service Bulletin F50–332, dated March 13, 2002.

#### No Reporting Requirements

(c) Although the service bulletin specifies to submit a reporting card to the manufacturer, this AD does not include such a requirement.

#### **Alternative Methods of Compliance**

(d) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

**Note 2:** The subject of this AD is addressed in French airworthiness directive 2002–033–039(B) R1, dated May 15, 2002.

Issued in Renton, Washington, on January 9, 2004.

#### Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04–1306 Filed 1–21–04; 8:45 am] BILLING CODE 4910–13–P

#### DEPARTMENT OF TRANSPORTATION

## **Federal Aviation Administration**

14 CFR Part 39

[Docket No. 2002-NM-261-AD] RIN 2120-AA64

# Airworthiness Directives; Saab Model SAAB 2000 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking

(NPRM).

**SUMMARY:** This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Saab Model SAAB 2000 series airplanes. This proposal would require repetitive inspections of the installation of the spoilers of the windshield wiper assemblies for discrepancies, and replacement with new spoilers if necessary. The proposal also would require eventual replacement of the spoilers of the windshield wiper assemblies with new spoilers. This action is necessary to prevent failure of the windshield wiper assembly, which could result in loss of visibility, damage to the propeller(s) and/or engine(s), or penetration of the fuselage skin and

consequent rapid depressurization of the airplane. This action is intended to address the identified unsafe condition.

**DATES:** Comments must be received by February 23, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-261-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227–1232. Comments may also be sent via the Internet using the following address: 9-anmnprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2002–NM–261–AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from Saab Aircraft AB, SAAB Aircraft Product Support, S–581.88, Linköping, Sweden. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

#### FOR FURTHER INFORMATION CONTACT:

Rosanne Ryburn, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (425) 227–2139; fax (425) 227–1149.

#### SUPPLEMENTARY INFORMATION:

### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

• Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.

- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002–NM–261–AD." The postcard will be date stamped and returned to the commenter.

## **Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-261-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

### Discussion

The Luftfartsverket (LFV), which is the airworthiness authority for Sweden, notified the FAA that an unsafe condition may exist on certain Saab Model SAAB 2000 series airplanes. The LFV advises that it has received reports of findings of cracked or missing spoilers of the windshield wiper assemblies on certain airplanes listed in Saab Service Bulletin 2000–56–002 (Replacement of Windshield Wiper Spoiler), dated November 28, 1998, on which the actions specified in that service bulletin have not been done. Parts that separate from the airplane in this area can cause damage to the propeller(s), engine(s), and structure. One report indicated that, after excessive vibration and subsequent separation of a windshield wiper spoiler, the spoiler fell into the propeller and subsequently hit the wing leading edge and fuselage. Such conditions, if not corrected, could result in loss of visibility, damage to the propeller(s) and/or engine(s), or penetration of the fuselage skin and consequent rapid depressurization of the airplane.