

Issued in Renton, Washington, on April 27, 2004.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2004-NM-17-AD; Amendment 39-13505; AD 2004-05-10]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 767 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document corrects a typographical error that appeared in airworthiness directive (AD) 2004-05-10 that was published in the **Federal Register** on March 5, 2004 (69 FR 10321). The typographical error resulted in an incorrect reference to a previous AD. This AD is applicable to certain Boeing Model 767 series airplanes. This AD requires repetitive detailed visual inspections of the aft pressure bulkhead for damage and cracking, and repair if necessary. This AD also requires eddy current inspections prior to the airplane accumulating 25,000 flight cycles.

DATES: Effective March 22, 2004.

FOR FURTHER INFORMATION CONTACT: Suzanne Masterson, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6441; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION: Airworthiness Directive (AD) 2004-05-10, amendment 39-13505, applicable to certain Boeing Model 767 series airplanes, was published in the **Federal Register** on March 5, 2004 (69 FR 10321). That AD requires repetitive detailed visual inspections of the aft pressure bulkhead for damage and cracking, and repair if necessary. That AD also requires eddy current inspections prior to the airplane accumulating 25,000 flight cycles.

As published, the restatement heading on page 10323 specified that certain paragraphs were a "restatement of AD 88-09-03 R1." In paragraph (a) the compliance time was specified as, "Prior to the accumulation of 6,000 flight cycles or within the next 1,000

flight cycles after September 26, 1988 (effective date of AD 88-09-03 R1, amendment 39-6001). * * *'' However, the preamble to that AD discusses and specifies in several places the correct referenced AD number as AD 88-19-03 R1.

Since no other part of the regulatory information has been changed, the final rule is not being republished in the **Federal Register**.

The effective date of this AD remains March 22, 2004.

§ 39.13 [Corrected]

■ On page 10323, in the first column, the restatement header and paragraph (a) of AD 2004-05-10 is corrected to read as follows:

* * * * *

Restatement of AD 88-19-03 R1

(a) Prior to the accumulation of 6,000 flight cycles or within the next 1,000 flight cycles after September 26, 1988 (effective date of AD 88-19-03 R1, amendment 39-6001), whichever occurs later, unless accomplished within the last 5,000 flight cycles, and thereafter at intervals not to exceed 6,000 flight cycles, perform a detailed inspection of the aft side of the entire body station 1582 pressure bulkhead for damage (as defined in the Structural Repair Manual) and cracking, in accordance with Boeing Service Bulletin 767-53-0026, dated November 19, 1987; or Revision 1, dated March 16, 1989.

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Issued in Renton, Washington, on April 26, 2004.

Kevin M. Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-278-AD; Amendment 39-13608; AD 2004-09-19]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A319 and A320 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Airbus Model A319 and A320 series airplanes, that requires modifying the electrical bonding of the fuel return line in each wing between ribs 7 and 8. This action

is necessary to reduce the potential for electrical arcing within the fuel tank due to insufficient electrical bonding, which could result in a fire or explosion in the fuel tank. This action is intended to address the identified unsafe condition.

DATES: Effective June 9, 2004.

The incorporation by reference of a certain publication listed in the regulations is approved by the Director of the Federal Register as of June 9, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Airbus Model A319 and A320 series airplanes was published in the **Federal Register** on February 6, 2004 (69 FR 5794). That action proposed to require modifying the electrical bonding of the fuel return line in each wing between ribs 7 and 8.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. The FAA has duly considered the single comment received.

The commenter supports the proposed rule.

Explanation of Change to Final Rule

The proposed AD states that the subject of the proposed AD is addressed in French airworthiness directive 2002-476(B), dated September 18, 2002. Since the preparation of the proposed AD, the Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, has issued French airworthiness directive F-2002-476 R1,