

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2004-09-16 Fairchild Dornier GmbH (Formerly Dornier Luftfahrt GmbH): Amendment 39-13605. Docket 200-NM-263-AD.

Applicability: All Model 328-100 and -300 series airplanes, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the rudder flight control system due to cracking of the bearing lugs of the rudder spring tab lever assembly, which could result in loss of rudder control and consequent reduced controllability of the airplane, accomplish the following:

Repetitive Inspections

(a) Within 400 flight hours or 2 months after the effective date of this AD, whichever is first: Do detailed and eddy current inspections for cracking of the bearing lugs of the rudder spring tab lever assembly by doing all the actions per Paragraphs 2.A., 2.B., and 2.D. of the Accomplishment Instructions of Dornier Alert Service Bulletin ASB-328-27-036 (for Model 328-100 series airplanes); or ASB-328J-27-013 (for Model 328-300 series airplanes); both dated February 12, 2003; as applicable. If no cracking is found, repeat the inspections thereafter at intervals not to exceed 24 months.

Note 1: For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

Corrective Action/Repetitive Inspections

(b) If any cracking is found during any inspection required by paragraph (a) of this AD: Before further flight, replace the spring tab lever assembly with a new assembly by doing all the actions per Paragraph 2.C. of the Accomplishment Instructions of Dornier Alert Service Bulletin ASB-328-27-036; or ASB-328J-27-013, both dated February 12, 2003; as applicable. Repeat the inspections required by paragraph (a) of this AD thereafter at intervals not to exceed 24 months.

(c) Dornier Alert Service Bulletins ASB-328-27-036 and ASB-328J-27-013, both dated February 12, 2003, recommend reporting crack findings and returning damaged lever assemblies to the manufacturer, but this AD does not contain such requirements.

Note 2: There is no terminating action available at this time for the repetitive inspections required by this AD.

Alternative Methods of Compliance

(d) In accordance with 14 CFR 39.19, the Manager, International Branch, FAA, ANM-116, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

Incorporation by Reference

(e) The actions shall be done in accordance with Dornier Alert Service Bulletin ASB-328-27-036, dated February 12, 2003; or Dornier Alert Service Bulletin ASB-328J-27-013, dated February 12, 2003; as applicable. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from AvCraft Aerospace GmbH, P.O. Box 1103, D-82230 Wessling, Germany. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Note 3: The subject of this AD is addressed in German airworthiness directives 2003-383 and 2003-384, both dated November 13, 2003.

Effective Date

(f) This amendment becomes effective on June 9, 2004.

Issued in Renton, Washington, on April 26, 2004.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-10022 Filed 5-4-04; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-253-AD; Amendment 39-13613; AD 2004-09-23]

RIN 2120-AA64

Airworthiness Directives; Fokker Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to all Fokker Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 series airplanes, that requires repetitive inspections of the control panel of the direct current (DC) generator for discrepancies, and replacement of any discrepant part. This action is necessary to prevent loss of both DC generator systems and loss of several other airplane systems, which could lead to the pilot's inability to maintain controlled flight. This action is intended to address the identified unsafe condition.

DATES: Effective June 9, 2004.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of June 9, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA,

Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1137; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Fokker Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 series airplanes was published in the **Federal Register** on July 29, 2003 (68 FR 44493). That action proposed to require repetitive inspections of the control panel of the direct current (DC) generator for discrepancies, and replacement of any discrepant part.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comment received.

The commenter asks that the wording specified in paragraph (b) of the proposed AD that states, “* * * prior to further flight, replace any discrepant part with a new part having the same part number * * *” be changed due to limited availability of new parts. The commenter suggests the following wording: “* * * prior to further flight, replace any discrepant part with a serviceable part having the same part number. * * *” The commenter states that finding new internal parts for this unit may cause unnecessary delays in returning the airplane to service, when a serviceable part is acceptable for the replacement.

The FAA agrees with the intent of the commenter's request. We have changed paragraph (b) of this final rule to allow for installation of either new or serviceable parts.

Conclusion

After careful review of the available data, including the comment noted above, we have determined that air safety and the public interest require the adoption of the rule with the change described previously. We have determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

Cost Impact

The FAA estimates that 39 airplanes of U.S. registry will be affected by this AD, that it will take about 4 work hours per airplane to accomplish the inspection, and that the average labor rate is \$65 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$10,140, or \$260 per inspection cycle.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

2004-09-23 Fokker Services B.V.: Amendment 39-13613. Docket 2002-NM-253-AD.

Applicability: All Model F27 Mark 100, 200, 300, 400, 500, 600, and 700 series airplanes; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent the loss of both direct current (DC) generator systems and loss of several other airplane systems, which could lead to the pilot's inability to maintain controlled flight, accomplish the following:

Initial and Repetitive Inspections

(a) Within 60 days after the effective date of this AD, do a detailed inspection of the control panel of the DC generator for discrepancies, per the Accomplishment Instructions of Fokker Service Bulletin F27/24-79, dated April 28, 1999. Repeat the inspection thereafter at intervals not to exceed 4,000 flight hours.

Note 1: For the purposes of this AD, a detailed inspection is defined as: “An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required.”

(b) If any discrepancy is found during any inspection required by paragraph (a) of this AD, prior to further flight, replace any discrepant part with a new or serviceable part having the same part number, per the Accomplishment Instructions of Fokker Service Bulletin F27/24-79, dated April 28, 1999.

Note 2: The service bulletin references Bendix (Allied Signal) publication R766-28, Technical Manual, Maintenance Instructions with Illustrated Parts Catalog for Generator Control Panel type no. 1539-11-B and 1539-12-B, paragraphs 2-12 through 2-15, as an additional source of service information for accomplishing the inspections and any parts replacement required by paragraphs (a) and (b) of this AD.

Alternative Methods of Compliance

(c) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, is authorized to approve alternative methods of compliance for this AD.

Incorporation by Reference

(d) The actions shall be done in accordance with Fokker Service Bulletin F27/24-79, dated April 28, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on

the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Note 3: The subject of this AD is addressed in Dutch airworthiness directive 1999-093, dated June 30, 1999.

Effective Date

(e) This amendment becomes effective on June 9, 2004.

Issued in Renton, Washington, on April 22, 2004.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-10138 Filed 5-4-04; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 95

[Docket No. 30412; Amdt. No. 448]

IFR Altitudes; Miscellaneous Amendments

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts miscellaneous amendments to the required IFR (instrument flight rules) altitudes and changeover points for certain Federal airways, jet routes, or direct routes for which a minimum or maximum en route authorized IFR altitude is prescribed. This regulatory action is needed because of changes occurring in the National Airspace System. These changes are designed to provide for the safe and efficient use of the navigable airspace under instrument conditions in the affected areas.

EFFECTIVE DATE: 0901 UTC, June 10, 2004.

FOR FURTHER INFORMATION CONTACT:

Donald P. Pate, Flight Procedure Standards Branch (AMCAFS-420), Flight Technologies and Programs Division, Flight Standards Service, Federal Aviation Administration, Mike Monroney Aeronautical Center, 6500 South MacArthur Blvd. Oklahoma City, OK 73169 (Mail Address: P.O. Box 25082 Oklahoma City, OK 73125) telephone: (405) 954-4164.

SUPPLEMENTARY INFORMATION: This amendment to part 95 of the Federal Aviation Regulations (14 CFR part 95) amends, suspends, or revokes IFR altitudes governing the operation of all aircraft in flight over a specified route or any portion of that route, as well as the changeover points (COPs) for Federal airways, jet routes, or direct routes as prescribed in part 95.

The Rule

The specified IFR altitudes, when used in conjunction with the prescribed changeover points for those routes, ensure navigation aid coverage that is adequate for safe flight operations and free of frequency interference. The reasons and circumstances that create the need for this amendment involve matters of flight safety and operational efficiency in the National Airspace System, are related to published aeronautical charts that are essential to the user, and provide for the safe and efficient use of the navigable airspace. In addition, those various reasons or circumstances require making this amendment effective before the next scheduled charting and publication date of the flight information to assure its timely availability to the user. The effective date of this amendment reflects those considerations. In view of the close and immediate relationship between these regulatory changes and safety in air commerce, I find that notice and public procedure before adopting this amendment are impracticable and

contrary to the public interest and that good cause exists for making the amendment effective in less than 30 days.

Conclusion

The FAA has determined that this regulation only involves an established body of technical regulations for which frequent and routine amendments are necessary to keep them operationally current. It, therefore—(1) Is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034; February 26, 1979); and (3) does not warrant preparation of a regulatory evaluation as the anticipated impact is so minimal. For the same reason, the FAA certifies that this amendment will not have a significant economic impact on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 95

Airspace, Navigation (air).

Issued in Washington, DC on April 30, 2004.

James J. Ballough,

Director, Flight Standards Service.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, part 95 of the Federal Aviation Regulations (14 CFR part 95) is amended as follows effective at 0901 UTC, June 10, 2004.

■ 1. The authority citation for part 95 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40103, 40106, 40113, 40114, 40120, 44502, 44514, 44719, 44721.

■ 2. Part 95 is amended to read as follows:

REVISIONS TO IFR ALTITUDES & CHANGEOVER POINTS

[Amendment 448—Final Effective Date June 10, 2004]

From	To	MEA
§ 95.6001 Victor Routes—U.S.		
§ 95.6003 VOR Federal Airway 3 Is Amended To Read in Part		
Savannah, GA VORTAC *1,500—MOCA	Owens, SC FIX	*3,000
Owens, SC FIX	Vance, SC VORTAC	2,000
§ 95.6016 VOR Federal Airway 16 is Amended To Read in Part		
Damas, TN FIX *7,500—MCA Stove FIX SW BND	*Stove, VA FIX	7,500
Stove, VA FIX	Speel, VA FIX	6,000
Speel, VA FIX	Pulaski, VA VORTAC	5,400