

Actions	Compliance	Procedures
(1) Perform a visual inspection of the vertical stabilizer attach fitting (P/N 40301-7), the horizontal stabilizer attach fitting (P/N 40303-1/-4/-7 or 95267-1), attachment bolt (P/N NAS1105-68), and the vertical fin aft spar (P/N 40261-24 or P/N 95253-1) for corrosion and cracks.	Within the next 50 hours time-in-service (TIS) after January 22, 2008 (the effective date of this AD) and repetitively thereafter at intervals not to exceed 100 hours TIS.	Follow Thrush Aircraft, Inc. Service Bulletin No. SB-AG-45, Revision B, dated June 1, 2007.
(2) If corrosion or cracks are found in P/N 40301-7, 40303-1/-4/-7, 95267-1, NAS1105-68, 40261-24, or 95253-1 during any inspection required in paragraph (e)(1) of this AD: (i) Replace the vertical stabilizer attach fitting with new P/N 95266-3; the horizontal stabilizer attach fitting with new P/N 95267-5; and the attachment bolt with NAS6207-68; and (ii) If corrosion or cracks are found in the P/N 40261-24 or P/N 95253-1 vertical fin aft spar, repair in accordance with Thrush SB-AG-45, Revision B, or replace with a new P/N 40261-24 or new P/N 95253-1.	Before further flight after any inspection where corrosion or cracks are found. This action terminates the repetitive inspections required in paragraph (e)(1) of this AD.	Follow Thrush Aircraft, Inc. Service Bulletin No. SB-AG-45, Revision B, dated June 1, 2007.
(3) If no corrosion or cracks are found in P/N 40301-7, 40303-1/-4/-7, 95267-1, NAS1105-68, 40261-24, or 95253-1 during any inspection required in paragraph (e)(1) of this AD: (i) Replace the vertical stabilizer attach fitting with new P/N 95266-3; the horizontal stabilizer attach fitting with new P/N 95267-5; and the attachment bolt with NAS6207-68; and (ii) Perform a visual inspection of the vertical fin aft spar (P/N 40261-24 or P/N 95253-1) for corrosion and cracks, and. (iii) If corrosion or cracks are found in the P/N 40261-24 or P/N 95253-1 vertical fin aft spar, repair in accordance with Thrush SB-AG-45, Revision B, or replace with a new P/N 40261-24 or new P/N 95253-1.	Within the next 2,000 hours TIS after January 22, 2008 (the effective date of this AD) or within 2 years after January 22, 2008 (the effective date of this AD), whichever occurs first. This action terminates the repetitive inspections required in paragraph (e)(1) of this AD.	Follow Thrush Aircraft, Inc. Service Bulletin No. SB-AG-45, Revision B, dated June 1, 2007.

### Special Flight Permit

(f) Under 14 CFR part 39.23, we are limiting the special flight permits authorized for this AD to ferry aircraft to a maintenance facility for inspection by the following conditions:

- (1) Hopper must be empty;
- (2) Vne reduced to 126 m.p.h. (109 knots); and
- (3) No flight into known turbulence.

### Alternative Methods of Compliance (AMOCs)

(g) The Manager, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Cindy Lorenzen, Aerospace Engineer, ACE-115A, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Blvd., Suite 450, Atlanta, GA 30349; telephone: (770) 703-6078; facsimile: (770) 703-6097; e-mail: [cindy.lorenzen@faa.gov](mailto:cindy.lorenzen@faa.gov); or Mike Cann, Aerospace Engineer, ACE-117A, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Blvd., Suite 450, Atlanta, Georgia 30349; telephone: (770) 703-6038; facsimile: (770) 703-6097; e-mail: [michael.cann@faa.gov](mailto:michael.cann@faa.gov). Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

### Material Incorporated by Reference

(h) You must use Thrush Aircraft, Inc. Service Bulletin No. SB-AG-45, Revision B, dated June 1, 2007, to do the actions required

by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Thrush Aircraft, Inc., P.O. Box 3149, 300 Old Pretoria Road, Albany, Georgia 31706-3149; telephone: 229-883-1440; facsimile: 229-436-4856; or on the Internet at: <http://www.thrushaircraft.com>.

(3) You may review copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Kansas City, Missouri, on December 10, 2007.

**John R. Colomy,**

*Acting Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-24218 Filed 12-14-07; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2005-21470; Directorate Identifier 2003-NM-45-AD; Amendment 39-15302; AD 2007-25-20]

RIN 2120-AA64

**Airworthiness Directives; McDonnell Douglas Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30 and DC-10-30F (KC-10A and KDC-10) Airplanes; Model DC-10-40 and DC-10-40F Airplanes; and Model MD-11 and MD-11F Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain McDonnell Douglas Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30 and DC-10-30F (KC-10A and KDC-10) airplanes; Model DC-10-40 and DC-10-40F airplanes; and Model MD-11 and MD-11F airplanes. This AD requires, for certain airplanes, modifying the thrust reverser command wiring of the number 2 engine. For certain other airplanes, this AD requires modifying the thrust

reverser system wiring from the flight compartment to engines 1, 2, and 3 thrust reversers. This AD also requires installing thrust reverser locking systems on certain airplanes. This AD results from a determination that the thrust reverser systems on these McDonnell Douglas airplanes do not adequately preclude unwanted deployment of a thrust reverser. We are issuing this AD to prevent an unwanted deployment of a thrust reverser during flight, which could result in reduced controllability of the airplane.

**DATES:** This AD becomes effective January 22, 2008.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of January 22, 2008.

On October 1, 2001 (66 FR 44950, August 27, 2001), the Director of the Federal Register approved the incorporation by reference of McDonnell Douglas Service Bulletin DC10-78-060, dated December 17, 1999.

On April 25, 2001 (66 FR 15785, March 21, 2001), the Director of the Federal Register approved the incorporation by reference of McDonnell Douglas Alert Service Bulletin DC10-78A057, Revision 01, dated February 18, 1999.

**ADDRESSES:** For Boeing and McDonnell Douglas service information identified in this AD, contact Boeing Commercial Airplanes, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). For Rohr service information identified in this AD, contact Rohr, Inc., 850 Lagoon Drive, Chula Vista, California 91910-2098.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD

docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Philip C. Kush, Aerospace Engineer, Propulsion Branch, ANM-140L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5263; fax (562) 627-5210.

**SUPPLEMENTARY INFORMATION:**

**Discussion**

The FAA issued a supplemental notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain McDonnell Douglas Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30 and DC-10-30F (KC-10A and KDC-10) airplanes; Model DC-10-40 and DC-10-40F airplanes; and Model MD-11 and MD-11F airplanes. That supplemental NPRM was published in the **Federal Register** on July 23, 2007 (72 FR 40090). That supplemental NPRM proposed to require, for certain airplanes, modifying the thrust reverser command wiring of the number 2 engine. For certain other airplanes, the supplemental NPRM proposed to require modifying the thrust reverser system wiring from the flight compartment to engines 1, 2, and 3 thrust reversers. The supplemental NPRM also proposed to require installing thrust reverser locking systems on certain airplanes. The supplemental NPRM also proposed to revise the original NPRM by revising, for certain airplanes, the requirements for the modification of the thrust reverser system wiring from the flight compartment to engines 1, 2, and 3 thrust reversers.

**Comments**

We provided the public the opportunity to participate in the development of this AD. We have considered the comment received.

**Request for Notification of Service Bulletin/Rulemaking**

FedEx requests that we and/or Boeing notify operators of any service bulletin or rulemaking that will cover Model MD-11 and MD-11F airplanes that are not specified in Boeing Alert Service Bulletin MD11-78A007, Revision 4, dated February 22, 2007 (which is referred to as a source of service information for doing a modification specified in the supplemental NPRM). FedEx states that it has no comments on the proposed requirements of the supplemental NPRM.

We acknowledge FedEx's request. We have been advised that when a service bulletin is released Boeing does notify the customers affected by the service bulletin. If service information is developed for other Model MD-11 and MD-11F airplanes and an unsafe condition is identified, we might consider further rulemaking. Interested persons can review the **Federal Register** to become aware of such rulemaking actions.

**Conclusion**

We have carefully reviewed the available data, including the comment received, and determined that air safety and the public interest require adopting the AD as proposed in the supplemental NPRM.

**Costs of Compliance**

There are about 612 airplanes of the affected designs in the worldwide fleet. This AD affects about 245 airplanes of U.S. registry. The following tables provide the estimated costs for U.S. operators to comply with this AD, for the applicable actions, at an average hourly labor rate of \$80 per work hour.

**COST FOR WIRING MODIFICATION/THRUST REVERSER LOCKING SYSTEM INSTALLATION**

Action	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Modify wiring (Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30 and DC-10-30F (KC-10A and KDC-10) airplanes).	34 .....	\$1,562 .....	\$4,282 .....	40	\$171,280.
Modify wiring (Model DC-10-40 and DC-10-40F airplanes).	34 .....	\$5,238 .....	\$7,958 .....	45	\$358,110.
Modify wiring (Model MD-11 and -11F airplanes).	Between 124 and 192.	Between \$11,912 and \$17,672.	Between \$21,832 and \$33,032.	160	Between \$3,493,120 and \$5,285,120.

COST FOR WIRING MODIFICATION/THRUST REVERSER LOCKING SYSTEM INSTALLATION—Continued

Action	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Install thrust reverser locking system (Model DC-10-40 and DC-10-40F airplanes).	218 .....	Between \$165,535 and \$207,792.	Between \$182,975 and \$225,232.	45	Between \$8,233,875 and \$10,135,440.

COST OF CONCURRENT ACTIONS FOR MODEL MD-11 AND MD-11F AIRPLANES

Action	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Update program software, as applicable .....	2 .....	None .....	\$160 .....	Up to 160	Up to \$25,600.
Modify wing pylon harnesses, as applicable	100 .....	\$5,268 .....	\$13,268 .....	Up to 160	Up to \$2,122,880.
Modify pylon thrust reverser harnesses and J-box, as applicable.	Between 82 and 192.	Between \$10,472 and \$15,999.	Between \$17,032 and \$31,359.	Up to 160	Up to \$5,017,440.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on

the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the ADDRESSES section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

- Accordingly, under the authority delegated to me by the Administrator,

the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

- 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

**2007-25-20 McDonnell Douglas:**  
Amendment 39-15302. Docket No. FAA-2005-21470; Directorate Identifier 2003-NM-45-AD.

**Effective Date**

- (a) This AD becomes effective January 22, 2008.

**Affected ADs**

- (b) None.

**Applicability**

- (c) This AD applies to airplanes, certificated in any category, as listed in Table 1 of this AD.

TABLE 1.—APPLICABILITY

McDonnell Douglas airplane—	As identified in—
(1) Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30 and DC-10-30F (KC-10A and KDC-10) airplanes.	Boeing Service Bulletin DC10-78-066, Revision 01, dated November 30, 2001.
(2) Model DC-10-40 and DC-10-40F airplanes .....	Boeing Service Bulletin DC10-78-067, dated October 30, 2002.
(3) Model MD-11 and MD-11F airplanes .....	Boeing Alert Service Bulletin MD11-78A007, Revision 4, dated February 22, 2007.

**Unsafe Condition**

(d) This AD was prompted by a determination that the thrust reverser systems on these McDonnell Douglas

airplanes do not adequately preclude unwanted deployment of a thrust reverser. We are issuing this AD to prevent an unwanted deployment of a thrust reverser

during flight, which could result in reduced controllability of the airplane.

**Compliance**

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

**Wiring Modification**

(f) For Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30, and DC-10-30F (KC-10A and KDC-10) airplanes: Within 60 months after the effective date of this AD, modify the thrust reverser command wiring of the number 2 engine by doing all the actions specified in the Accomplishment Instructions of Boeing Service Bulletin DC10-78-066, Revision 01, dated November 30, 2001.

(g) For Model MD-11 and MD-11F airplanes: Within 60 months after the effective date of this AD, modify the thrust reverser system wiring from the flight compartment to engines 1, 2, and 3 thrust reversers by doing all the actions specified in the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-78A007, Revision 4, dated February 22, 2007.

**Wiring Modification/Installation of Thrust Reverser Locking System**

(h) For Model DC-10-40 and DC-10-40F airplanes: Within 60 months after the effective date of this AD, modify the thrust reverser command wiring of the number 2 engine by doing all the actions specified in

the Accomplishment Instructions of Boeing Service Bulletin DC10-78-067, dated October 30, 2002, and install thrust reverser locking systems by doing all the applicable actions specified in the Accomplishment Instructions of McDonnell Douglas Service Bulletin DC10-78-064, dated June 24, 2003.

**Prior or Concurrent Actions**

(i) For Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30, and DC-10-30F (KC-10A and KDC-10) airplanes: Prior to or concurrently with the actions required by paragraph (f) of this AD, do the actions specified in Table 2 of this AD.

**TABLE 2.—PRIOR OR CONCURRENT ACTIONS FOR MODEL DC-10-10, DC-10-10F, DC-10-15, DC-10-30, AND DC-10-30F (KC-10A AND KDC-10) AIRPLANES**

Do—	Required by—	In accordance with—
Repetitive detailed visual inspections, functional checks, and torque checks of the thrust reverser systems, and applicable corrective actions.	Paragraphs (c) and (i) of AD 2001-05-10, amendment 39-12147.	McDonnell Douglas Alert Service Bulletin DC10-78A057, Revision 01, dated February 18, 1999.
A modification of the indication light system for the thrust reversers.	Paragraph (a) of AD 2001-17-19, amendment 39-12410.	McDonnell Douglas Service Bulletin DC10-78-060, dated December 17, 1999; or McDonnell Douglas Service Bulletin DC10-78-060, Revision 01, dated June 30, 2003.

(j) For Model MD-11 and MD-11F airplanes: Prior to or concurrently with the

actions required by paragraph (g) of this AD, do the actions specified in Table 3 of this AD.

**TABLE 3.—PRIOR OR CONCURRENT ACTIONS FOR MODEL MD-11 AND MD-11F AIRPLANES**

Do—	In accordance with—
An update of the program software of display electronic units .....	McDonnell Douglas Service Bulletin MD11-31-091, dated November 5, 1998.
A modification of the wing pylon harnesses .....	Rohr Service Bulletin MD-11 54-200, Revision 1, dated May 14, 2001.
A modification of the pylon thrust reverser harnesses and J-box .....	Rohr Service Bulletin MD-11 54-201, Revision 2, dated December 12, 2005.

**Actions Accomplished According to Previous Issues of Service Bulletins**

(k) Actions accomplished before the effective date of this AD according to Boeing Service Bulletin DC10-78-066, dated March 6, 2001; Rohr Service Bulletin MD-11 54-201, dated November 30, 1999; or Rohr Service Bulletin MD-11 54-201, Revision 1, dated November 23, 2005; are considered acceptable for compliance with the applicable corresponding actions specified in this AD.

**Alternative Methods of Compliance (AMOCs)**

(1)(1) The Manager, Los Angeles Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies,

notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

**Material Incorporated by Reference**

(m) You must use the service bulletins listed in Table 4 of this AD to perform the actions that are required by this AD, unless the AD specifies otherwise.

**TABLE 4.—ALL MATERIAL INCORPORATED BY REFERENCE**

Service Bulletin	Revision level	Date
Boeing Alert Service Bulletin MD11-78A007 .....	4 .....	February 22, 2007.
Boeing Service Bulletin DC10-78-066 .....	01 .....	November 30, 2001.
Boeing Service Bulletin DC10-78-067 .....	Original .....	October 30, 2002.
McDonnell Douglas Alert Service Bulletin DC10-78A057, including Attachment A .....	01 .....	February 18, 1999.
McDonnell Douglas Service Bulletin DC10-78-060 .....	Original .....	December 17, 1999.
McDonnell Douglas Service Bulletin DC10-78-060 .....	01 .....	June 30, 2003.
McDonnell Douglas Service Bulletin DC10-78-064 .....	Original .....	June 24, 2003.
McDonnell Douglas Service Bulletin MD11-31-091 .....	Original .....	November 5, 1998.
Rohr Service Bulletin MD-11 54-200 .....	1 .....	May 14, 2001.

TABLE 4.—ALL MATERIAL INCORPORATED BY REFERENCE—Continued

Service Bulletin	Revision level	Date
Rohr Service Bulletin MD-11 54-201 .....	2 .....	December 12, 2005.

(1) The Director of the Federal Register approved the incorporation by reference of the service bulletins listed in Table 5 of this AD in accordance with 5 U.S.C. 552(a) and 1 CFR part 51.

TABLE 5.—NEW MATERIAL INCORPORATED BY REFERENCE

Service Bulletin	Revision level	Date
Boeing Alert Service Bulletin MD11-78A007 .....	4	February 22, 2007.
Boeing Service Bulletin DC10-78-066 .....	01	November 30, 2001.
Boeing Service Bulletin DC10-78-067 .....	Original	October 30, 2002.
McDonnell Douglas Service Bulletin DC10-78-060 .....	01	June 30, 2003.
McDonnell Douglas Service Bulletin DC10-78-064 .....	Original	June 24, 2003.
McDonnell Douglas Service Bulletin MD11-31-091 .....	Original	November 5, 1998.
Rohr Service Bulletin MD-11 54-200 .....	1	May 14, 2001.
Rohr Service Bulletin MD-11 54-201 .....	2	December 12, 2005.

(2) On October 1, 2001 (66 FR 44950, August 27, 2001), the Director of the Federal Register approved the incorporation by reference of McDonnell Douglas Service Bulletin DC10-78-060, dated December 17, 1999.

(3) On April 25, 2001 (66 FR 15785, March 21, 2001), the Director of the Federal Register approved the incorporation by reference of McDonnell Douglas Alert Service Bulletin DC10-78A057, Revision 01, including Attachment A, dated February 18, 1999.

(4) Contact Boeing Commercial Airplanes, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024), for a copy of Boeing and McDonnell Douglas service information. Contact Rohr, Inc., 850 Lagoon Drive, Chula Vista, California 91910-2098, for a copy of Rohr service information. You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on November 29, 2007.

**Stephen P. Boyd,**

*Assistant Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-23934 Filed 12-14-07; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

**[Docket No. FAA-2007-0336; Directorate Identifier 2007-NM-201-AD; Amendment 39-15308; AD 2007-26-06]**

**RIN 2120-AA64**

**Airworthiness Directives; Boeing Model 747-200B, 747-300, and 747-400 Series Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Boeing Model 747-200B, 747-300, and 747-400 series airplanes. This AD requires an inspection to determine the manufacturer and manufacture date of the oxygen masks in the passenger service units of the outboard and center main deck, the flight attendant service units, flightcrew rest, upper and lower module of the door 5 overhead crew rest, lavatory modules, and miscellaneous ceiling panels, as applicable, and related investigative/corrective actions if necessary. This AD results from a report that several passenger masks with broken in-line flow indicators were found following a mask deployment. We are issuing this AD to prevent the in-line flow indicators of the passenger oxygen masks from fracturing and separating, which could inhibit oxygen flow to the

masks and consequently result in exposure of the passengers and cabin attendants to hypoxia following a depressurization event.

**DATES:** This AD becomes effective January 2, 2008.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of January 2, 2008.

We must receive comments on this AD by February 15, 2008.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory