

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

Boeing: Docket No. FAA-2007-28620; Directorate Identifier 2007-NM-090-AD.

Comments Due Date

(a) The FAA must receive comments on this AD action by August 23, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Boeing Model 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747SR, and 747SP series airplanes, certificated in any category; as identified in Boeing Alert Service Bulletin 747-53A2658, dated February 22, 2007.

Unsafe Condition

(d) This AD results from a report that an operator found a 1.65-inch crack on the station (STA) 1241 bulkhead fitting on the left side of a Boeing Model 747-200F series airplane that had accumulated 17,332 total flight cycles. We are issuing this AD to detect and correct cracking in the STA 1241 bulkhead fittings, which could result in reduced structural integrity of the airplane.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Inspections and Corrective Action

(f) At the applicable time specified in paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 747-53A2658, dated February 22, 2007: Do internal surface high-frequency eddy current and external ultrasonic inspections for cracking of the STA 1241 bulkhead fittings just above the canted pressure deck; determine the edge margin at seven fastener positions on each side of the airplane; and do all applicable related investigative/corrective actions; by doing all of the actions specified in the Accomplishment Instructions of Boeing Alert Service Bulletin 747-53A2658, dated February 22, 2007, except as provided by paragraphs (f)(1) and (f)(2) of this AD. Do all applicable related investigative/corrective actions before further flight. Repeat the inspections thereafter at the applicable interval specified in paragraph 1.E., "Compliance" of the service bulletin.

(1) Where the service bulletin specifies to contact Boeing for appropriate action, before further flight, do the action using a method approved in accordance with the procedures specified in paragraph (g) of this AD.

(2) Where the service bulletin specifies a compliance time after the date on the service bulletin, this AD requires compliance within the specified compliance time after the effective date of this AD.

Alternative Methods of Compliance (AMOCs)

(g)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

Issued in Renton, Washington, on June 25, 2007.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-13263 Filed 7-6-07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-28598; Directorate Identifier 2007-NM-036-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 757-200, -200CB, -200PF, and -300 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for all Boeing Model 757-200, -200CB, -200PF, and -300 series airplanes. This proposed AD would require installation of an automatic shutoff system for the center tank fuel boost pumps, and installation of a placard in the airplane flight deck if necessary. This proposed AD would also require revisions to the Limitations and Normal Procedures sections of the airplane flight manual to advise the flightcrew of certain operating restrictions for airplanes equipped with an automated center tank fuel pump shutoff control. This proposed AD would also require a revision to the Airworthiness Limitations (AWLs) section of the Instructions for Continued Airworthiness to incorporate AWL No. 28-AWL-20 and No. 28-AWL-26. This proposed AD would also require replacement of the fuel control panel assembly with a modified part, installation of two secondary pump control relays for the center tank fuel pumps, other specified actions, and concurrent modification of the fuel control panel assembly. This proposed AD results from fuel system reviews conducted by the manufacturer. We are proposing this AD to prevent center tank fuel pump operation with continuous low pressure, which could lead to friction sparks or overheating in the fuel pump inlet or could create a potential ignition source inside the center fuel tank; these conditions, in combination with flammable fuel vapors, could result in a center fuel tank

explosion and consequent loss of the airplane.

DATES: We must receive comments on this proposed AD by August 23, 2007.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- Government-wide rulemaking Web site: Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- Mail: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- Fax: (202) 493-2251.

- Hand Delivery: Room W12-140 on the ground floor of the West Building, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for the service information identified in this proposed AD.

FOR FURTHER INFORMATION CONTACT: Judy Coyle, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6497; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed in the **ADDRESSES** section. Include the docket number "FAA-2007-28598; Directorate Identifier 2007-NM-036-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that Web site, anyone can find and read the comments in any of our dockets, including the name of the individual

who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you may visit <http://dms.dot.gov>.

Examining the Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Operations office (telephone (800) 647-5527) is located on the ground floor of the West Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Discussion

The FAA has examined the underlying safety issues involved in fuel tank explosions on several large transport airplanes, including the adequacy of existing regulations, the service history of airplanes subject to those regulations, and existing maintenance practices for fuel tank systems. As a result of those findings, we issued a regulation titled "Transport Airplane Fuel Tank System Design Review, Flammability Reduction and Maintenance and Inspection Requirements" (66 FR 23086, May 7, 2001). In addition to new airworthiness standards for transport airplanes and new maintenance requirements, this rule included Special Federal Aviation Regulation No. 88 ("SFAR 88," Amendment 21-78, and subsequent Amendments 21-82 and 21-83).

Among other actions, SFAR 88 requires certain type design (i.e., type certificate (TC) and supplemental type certificate (STC)) holders to substantiate that their fuel tank systems can prevent ignition sources in the fuel tanks. This requirement applies to type design holders for large turbine-powered transport airplanes and for subsequent modifications to those airplanes. It requires them to perform design reviews and to develop design changes and maintenance procedures if their designs do not meet the new fuel tank safety standards. As explained in the preamble to the rule, we intended to adopt airworthiness directives to mandate any changes found necessary to address unsafe conditions identified as a result of these reviews.

In evaluating these design reviews, we have established four criteria intended to define the unsafe conditions

associated with fuel tank systems that require corrective actions. The percentage of operating time during which fuel tanks are exposed to flammable conditions is one of these criteria. The other three criteria address the failure types under evaluation: Single failures, single failures in combination with a latent condition(s), and in-service failure experience. For all four criteria, the evaluations included consideration of previous actions taken that may mitigate the need for further action.

We have determined that the actions identified in this proposed AD are necessary to reduce the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors, could result in fuel tank explosions and consequent loss of the airplane.

Boeing has found that certain failures will result in the center tank fuel pumps continuing to run after the tank has been depleted. Depending on the failure, pump low pressure may not be annunciated, or power may not be removed from the pump when the pump has been commanded "OFF." Operation of the center tank fuel pump with continuous low pressure could lead to friction sparks or overheating in the fuel boost pump inlet. This condition, if not corrected, could result in a fuel tank explosion and consequent loss of the airplane.

Other Relevant Rulemaking

On September 24, 2002, we issued AD 2002-19-52, amendment 39-12900 (67 FR 61253, September 30, 2002), applicable to all Boeing Model 737-600, -700, -700C, -800, and -900 series airplanes, Model 747 airplanes, and Model 757 airplanes. That AD requires revising the airplane flight manual (AFM) to advise the flightcrew of certain operating restrictions for maintaining minimum fuel levels, prohibits use of the horizontal stabilizer tank on certain airplanes, and prohibits the installation of certain fuel pumps. That AD requires concurrent removal of the currently required AFM revisions and insertion of new AFM revisions, requires installation of placards to alert the flightcrew to the operating restrictions, and prohibits installation of any uninspected pumps. That AD permits the AFM revision and placard to be removed under certain conditions. Installation of a placard in accordance with paragraph (e) of AD 2002-19-52, amendment 39-12900, is acceptable for compliance with paragraph (h) of this AD.

On November 23, 2002, we issued emergency AD 2002-24-51, amendment

39–12992, applicable to all Boeing Model 737–600, –700, –700C, –800, and –900 series airplanes, Model 747 airplanes, and Model 757 airplanes. (We issued a **Federal Register** version of AD 2002–24–51 on December 23, 2002 (68 FR 10, January 2, 2003).) That AD requires revising the AFM to require the flightcrew to maintain certain minimum fuel levels in the center fuel tanks and, for certain airplanes, to prohibit the use of the horizontal stabilizer fuel tank and certain center auxiliary fuel tanks. Accomplishing the actions specified in paragraphs (g), (h), (i), and (j) of this proposed AD would terminate the AFM revision specified in paragraph (e) of AD 2002–24–51 for Model 757–200, –200CB, –200PF, and –300 series airplanes that have the automatic shutoff system installed.

Relevant Service Information

We have reviewed Boeing Alert Service Bulletin 757–28A0081, dated February 16, 2006, for Model 757–200, –200CB, and –200PF series airplanes; and Boeing Alert Service Bulletin 757–28A0082, dated February 16, 2006, for Model 757–300 series airplanes. These service bulletins describe procedures for installing an automatic shutoff system for the center tank fuel boost pumps. Installation of the automatic shutoff system includes the following actions:

- In the main flight compartment, installing time delay relays in the P33 and P37 relay panels and installing automatic shutoff fuel test switches at the E2–1 rack stanchion.
- At the P11–3 and P11–4 circuit breaker panels in the flight compartment, adding new circuit breakers and replacing the light plate with a new improved light plate.
- Changing certain wire bundles in the P11–3 and P11–4 circuit breaker panels, in the P33 and P37 relay panels, between the P33 relay panel and the P5 overhead panel, between the P37 relay panel and the P5 overhead panel, and between the P11–3 circuit breaker panel and the P33 relay panel.
- Installing new wire bundles between the P33 relay panel and the test switch at the E2–1 rack stanchion and between the P37 relay panel and the test switch at the E2–1 rack stanchion.

We have also reviewed Section 9 of the Boeing 757 Maintenance Planning Data (MPD) Document, D622N001–9, Revision January 2006 (hereafter referred to as “Revision January 2006 of the MPD”). Subsection G, “AIRWORTHINESS LIMITATIONS—FUEL SYSTEMS AWLs,” of Revision January 2006 of the MPD describes new airworthiness limitations (AWLs) for fuel tank systems. The AWLs include:

- AWL inspections, which are periodic inspections of certain features for latent failures that could contribute to an ignition source.
- Critical design configuration control limitations (CDCCLs), which are limitation requirements to preserve a critical ignition source prevention feature of the fuel tank system design that is necessary to prevent the occurrence of an unsafe condition. The purpose of a CDCCL is to provide instruction to retain the critical ignition source prevention feature during configuration change that may be caused by alterations, repairs, or maintenance actions. A CDCCL is not a periodic inspection. Revision January 2006 of the MPD adds new fuel system AWL No. 28–AWL–20, which is a repetitive inspection of the automatic shutoff system for the center tank fuel boost pumps to verify functional integrity.

We have also reviewed Boeing Alert Service Bulletin 757–28A0105, Revision 1, dated April 2, 2007, for Model 757–200, –200CB, –200PF, and –300 series airplanes. This service bulletin describes procedures for replacing fuel control panel assembly part number (P/N) 233N3206–() (equipment number M10055) with a modified fuel control panel assembly, installing two secondary override pump control relays for the center tank fuel pumps in the P33 and P37 relays panels, and doing other specified actions. The other specified actions include the following:

- Changing the W2066 wire bundle located between the P5 overhead panel in the flight compartment and the P33 relay panel located in the main equipment center.
- Changing the W2070 wire bundle located between the P5 overhead panel in the flight compartment and P37 relay panel located in the main equipment center.
- Changing the W1230 wire bundle at the P33 relay panel.
- Changing the W1270 wire bundle at the P37 relay panel.
- Doing a functional test of the left and right primary and secondary override pump control relays.
- Doing a pump reversal test of the left and right override fuel pumps.

We have also reviewed Boeing Temporary Revision (TR) 09–006, dated January 2007. Boeing TR 09–006 is published as Section 9 of the Boeing 757 MPD Document, D622N001–9, Revision January 2007 (hereafter referred to as “Revision January 2007 of the MPD”). Subsection G of Revision January 2007 of the MPD adds new fuel system AWL No. 28–AWL–26, which is a repetitive inspection of the power failed on

protection system for the center tank fuel boost pumps to verify functional integrity.

Boeing Alert Service Bulletin 757–28A0105 specifies prior or concurrent accomplishment of BAE Systems Service Bulletin 233N3206–28–03, dated October 4, 2006. The BAE Systems service bulletin describes procedures for modifying the M10055 fuel control panel assembly, P/N 233N3206–(), to provide protection from “uncommanded pump ON” situations. The modification includes rerouting the J2 connector wire bundles from the forward left main pump switch S2 to the left center pump switch S6, rerouting the J3 connector wire bundles from the forward right main pump switch S5 to the right center pump switch S7, and installing new tie clips to secure the rerouted wire bundles.

Accomplishing the actions specified in the service information is intended to adequately address the unsafe condition.

FAA’s Determination and Requirements of the Proposed AD

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other airplanes of this same type design. For this reason, we are proposing this AD, which would require the following actions:

- Installation of an automatic shutoff system for the center tank fuel boost pumps.
- Installation of a placard in the airplane flight deck, if necessary. (Placards are necessary only for “mixed fleet operation,” which means that some airplanes in an operator’s fleet are equipped with automatic shutoff systems while other airplanes are not.)
- Revisions to the Limitations and Normal Procedures sections of the AFM to advise the flightcrew of certain operating restrictions for airplanes equipped with an automated center tank fuel pump shutoff control.
- Revision to the AWLs section of the Instructions for Continued Airworthiness to incorporate AWL No. 28–AWL–20, which would require repetitive inspections of the automatic shutoff system for the center tank fuel boost pumps to verify functional integrity.

• Replacement of fuel control panel assembly P/N 233N3206–() with a modified fuel control panel assembly, installation of two secondary override pump control relays for the center tank fuel pumps in the P33 and P37 relays panels, and other specified actions.

- Concurrent modification of the M10055 fuel control panel assembly, P/N 233N3206-().
- Revision to the AWLs section of the Instructions for Continued Airworthiness to incorporate AWL No. 28-AWL-26, which would require repetitive inspections of the power failed on protection system for the center tank fuel boost pumps to verify functional integrity.

This proposed AD would also allow accomplishing the revision to the AWLs section of the Instructions for Continued Airworthiness in accordance with later revisions of the MPD as an acceptable method of compliance if they are approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA.

Costs of Compliance

There are about 1,094 Model 757-200, -200CB, -200PF, and -300 series airplanes of the affected design in the worldwide fleet. The following table provides the estimated costs, at an average labor rate of \$80 per hour, for U.S. operators to comply with this proposed AD. The estimated cost of parts in the following table depends on the configuration of an airplane.

ESTIMATED COSTS

Model	Action	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
757-200, -200CB, and -200PF, series airplanes.	Installation of the automatic shutoff system.	91	\$8,309 to \$9,194.	\$15,309 to \$16,474.	631	\$9,836,659 to \$10,395,094.
757-300 series airplanes	Installation of the automatic shutoff system.	51	\$8,598 to \$8,654.	\$12,678 to \$12,734.	75	\$950,850 to \$955,050.
757-200, -200CB, -200PF, and -300 series airplanes.	Placard installation, if necessary	1	\$10	\$90	706	\$63,540.
	AFM revision	1	None	\$80	706	\$56,480.
	Maintenance program revision ..	1	None	\$80	706	\$56,480.
	Installation of secondary pump control relays.	29	\$2,097	\$4,417	706	\$3,118,402.
	Concurrent modification of the fuel control panel assembly.	2	\$40	\$200	706	\$141,200.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket. See the ADDRESSES section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13

by adding the following new airworthiness directive (AD):

Boeing: Docket No. FAA-2007-28598; Directorate Identifier 2007-NM-036-AD.

Comments Due Date

(a) The FAA must receive comments on this AD action by August 23, 2007.

Affected ADs

(b) Accomplishing certain paragraphs of this AD terminates certain requirements of AD 2002-24-51, amendment 39-12992.

Applicability

(c) This AD applies to all Boeing Model 757-200, -200CB, -200PF, and -300 series airplanes, certificated in any category.

Note 1: This AD requires revisions to certain operator maintenance documents to include new inspections of the automatic shutoff system for the center tank fuel boost pumps. Compliance with these inspections is required by 14 CFR 43.16 and 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these inspections, the operator may not be able to accomplish the inspections described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance according to paragraph (p) of this AD. The request should include a description of changes to the required inspections that will ensure acceptable maintenance of the automatic shutoff system.

Unsafe Condition

(d) This AD results from fuel system reviews conducted by the manufacturer. We are issuing this AD to prevent center tank fuel pump operation with continuous low pressure, which could lead to friction sparks or overheating in the fuel pump inlet or could create a potential ignition source

inside the center fuel tank; these conditions, in combination with flammable fuel vapors, could result in a center fuel tank explosion and consequent loss of the airplane.

Compliance

(e) You are responsible for having the actions required by this AD performed within

the compliance times specified, unless the actions have already been done.

Service Bulletin References

(f) The term “service bulletin,” as used in this AD, means the Accomplishment Instructions of the following service bulletins, as applicable.

TABLE 1.—SERVICE BULLETIN REFERENCES

Airplanes	Action	Service Bulletin
Model 757–200, –200CB, and –200PF series airplanes.	Installation specified in paragraph (g) of this AD.	Boeing Alert Service Bulletin 757–28A0081, dated February 16, 2006.
Model 757–300 series airplanes	Installation specified in paragraph (g) of this AD.	Boeing Alert Service Bulletin 757–28A0082, dated February 16, 2006.
For Model 757–200, –200CB, –200PF, and –300 series airplanes.	Installation specified in paragraph (k) of this AD.	Boeing Alert Service Bulletin 757–28A0105, Revision 1, dated April 2, 2007.

Installation of Automatic Shutoff System for the Center Tank Fuel Boost Pumps

(g) Within 36 months after the effective date of this AD: Install an automatic shutoff system for the center tank fuel boost pumps, by accomplishing all of the actions specified in the applicable service bulletin. If a placard has been previously installed on the airplane in accordance with paragraph (h) of this AD, the placard may be removed from the flight deck of only that airplane after the automatic shutoff system has been installed. Installing automatic shutoff systems on all airplanes in an operator’s fleet, in accordance with this paragraph, terminates the placard installation required by paragraph (h) of this AD, for all airplanes in an operator’s fleet.

Placard Installation for Mixed Fleet Operation

(h) Concurrently with installing an automatic shutoff system on any airplane in an operator’s fleet, as required by paragraph (g) of this AD: Install a placard adjacent to the pilot’s primary flight display on all airplanes in the operator’s fleet not equipped with an automatic shutoff system for the center tank fuel boost pumps. The placard reads as follows (alternative placard wording may be used if approved by an appropriate FAA Principal Operations Inspector):

“AD 2002–24–51 fuel usage restrictions required.”

Installation of a placard in accordance with paragraph (e) of AD 2002–19–52, amendment 39–12900, is acceptable for compliance with the requirements of this paragraph. Installing an automatic shutoff system on an airplane, in accordance with paragraph (g) of this AD, terminates the placard installation required by this paragraph, for only that airplane. Installing automatic shutoff systems on all airplanes in an operator’s fleet, in accordance with paragraph (g) of this AD, terminates the placard installation required by this paragraph, for all airplanes in an operator’s fleet. If automatic shutoff systems are installed concurrently on all airplanes in an operator’s fleet in accordance with paragraph (g) of this AD, or if operation according to the fuel usage restrictions of AD 2002–24–51 is maintained until automatic shutoff systems are installed on all airplanes in an operator’s

fleet, the placard installation specified in this paragraph is not required.

Airplane Flight Manual (AFM) Revision

(i) Concurrently with accomplishing the actions required by paragraph (g) of this AD: Do the actions specified in paragraphs (i)(1) and (i)(2) of this AD.

(1) Revise Section 1 of the Limitations section of the Boeing 757 AFM to include the following statement. This may be done by inserting a copy of this AD in the AFM.

“Intentional dry running of a center tank fuel pump (CTR L FUEL PUMP or CTR R FUEL PUMP message displayed on EICAS) is prohibited.”

Note 2: When a statement identical to that in paragraph (i)(1) of this AD has been included in the general revisions of the AFM, the general revisions may be inserted into the AFM, and the copy of this AD may be removed from the AFM.

(2) Revise Section 3.1 of the Normal Procedures section of the Boeing 757 AFM to include the following statements. This may be done by inserting a copy of this AD in the AFM.

“Procedures contained on this page are applicable to airplanes equipped with the automatic center tank fuel pump power removal system per Boeing Service Bulletin 757–28A0081 (757–200 Series) or 757–28A0082 (757–300 Series).

CENTER TANK FUEL PUMPS

Center tank fuel pump switches must not be “ON” unless personnel are available in the flight deck to monitor low PRESS lights.

For ground operations prior to engine start: The center tank fuel pump switches must not be positioned ON unless the center tank contains usable fuel. With center tank fuel pump switches ON, verify both center tank fuel pump low PRESS lights are illuminated and EICAS CTR L FUEL PUMP and CTR R FUEL PUMP messages are displayed.

For ground operations after engine start and flight operations: The center tank fuel pump switch must be selected OFF when the respective CTR L FUEL PUMP or CTR R FUEL PUMP message displays. Both center tank fuel pump switches must be selected OFF when either the CTR L FUEL PUMP or CTR R FUEL PUMP message displays if the

center tank is empty. During cruise flight, both center tank pump switches may be reselected ON whenever center tank usable fuel is indicated.

DE-FUELING AND FUEL TRANSFER

When transferring fuel or de-fueling center or main wing tanks, the center fuel pump low PRESS must be monitored and the fuel pump switches positioned to “OFF” at the first indication of low pressure. Prior to transferring fuel or de-fueling, conduct a lamp test of the respective fuel pump low PRESS lights.

De-fueling main wing tanks with passengers onboard is prohibited if main tank fuel pumps are powered. De-fueling center wing tank with passengers onboard is prohibited if the center wing tank fuel pumps are powered with the automatic center tank fuel pump power removal system inhibited. Fuel may be transferred from tank to tank, or the aircraft may be de-fueled with passengers onboard, provided fuel quantity in the tank from which fuel is being transferred from is maintained above 2,000 pounds (900 kilograms).”

Note 3: When statements identical to those in paragraph (i)(2) of this AD have been included in the general revisions of the AFM, the general revisions may be inserted into the AFM, and the copy of this AD may be removed from the AFM.

Airworthiness Limitations (AWLs) Revision for AWL No. 28–AWL–20

(j) Concurrently with accomplishing the actions required by paragraph (g) of this AD: Revise the AWLs section of the Instructions for Continued Airworthiness by incorporating AWL No. 28–AWL–20 of Subsection G of the Boeing 757 Maintenance Planning Data (MPD) Document, D622N001–9, Section 9, Revision January 2006, into the MPD. Accomplishing the revision in accordance with a later revision of the MPD is an acceptable method of compliance if the revision is approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA.

Installation of Secondary Override Pump Control Relays

(k) Within 60 months after the effective date of this AD: Replace fuel control panel assembly part number 233N3206–()

(equipment number M10055) with a modified fuel control assembly, install the secondary override pump control relays for the center tank fuel pumps in the P33 and P37 relays panels, and do all other specified actions as applicable, by accomplishing all of the applicable actions specified in the applicable service bulletin. The other specified actions must be accomplished before further flight after installing the secondary override pump control relays.

Concurrent Modification of the M10055 Fuel Control Panel Assembly

(l) For airplanes identified in paragraph 1.A.1. of Boeing Alert Service Bulletin 757-28A0105, Revision 1, dated April 2, 2007, equipped with any fuel control panel assembly identified in paragraph 1.A. of BAE Systems Service Bulletin 233N3206-28-03, dated October 4, 2006: Before or concurrently with accomplishing the actions required by paragraph (k) of this AD, modify the fuel control panel assembly, in accordance with BAE Systems Service Bulletin 233N3206-28-03, dated October 4, 2006.

AWLs Revision for AWL No. 28-AWL-26

(m) Before or concurrently with accomplishing the actions required by paragraph (k) of this AD: Revise the AWLs section of the Instructions for Continued Airworthiness by incorporating AWL No. 28-AWL-26 of Boeing Temporary Revision (TR) 09-006, dated January 2007, into the MPD. Boeing TR 09-006 is published as Section 9 of the Boeing 757 MPD Document, D622N001-9, Revision January 2007. Accomplishing the revision in accordance with a later revision of the MPD is an acceptable method of compliance if the revision is approved by the Manager, Seattle ACO.

Terminating Action for AD 2002-24-51

(n) Accomplishing the actions required by paragraphs (g), (h), (i), and (j) of this AD terminates the AFM limitations required by paragraph (e) of AD 2002-24-51 for Model 757-200, -200CB, -200PF, and -300 series airplanes that have the automatic shutoff system installed, except for the following limitation:

“Warning—Do not reset a tripped fuel pump circuit breaker.”

Except for this limitation, all other AFM limitations required by paragraph (e) of AD 2002-24-51 for Model 757-200, -200CB, -200PF, and -300 series airplanes may be removed from the AFM after accomplishing the actions required by paragraphs (g), (h), (i), and (j) of this AD.

Credit for Actions Done According to Previous Issue of Service Bulletin

(o) Actions accomplished before the effective date of this AD in accordance with Boeing Alert Service Bulletin 757-28A0105, dated January 31, 2007, are considered acceptable for compliance with the corresponding actions specified in paragraph (k) of this AD.

Alternative Methods of Compliance (AMOCs)

(p)(1) The Manager, Seattle ACO, has the authority to approve AMOCs for this AD, if

requested in accordance with the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(3) Installation of TDG Aerospace, Inc. Universal Fault Interrupter (UFI), installed and maintained in accordance with Supplemental Type Certificate (STC) ST01950LA, is approved as an AMOC with paragraphs (a) through (m) of this AD.

Note 4: Information concerning the existence of approved AMOCs with this AD, if any, may be obtained from the Seattle ACO.

Issued in Renton, Washington, on June 25, 2007.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-13265 Filed 7-6-07; 8:45 am]

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DEPARTMENT OF HEALTH AND HUMAN SERVICES

Food and Drug Administration

21 CFR Part 2

[Docket No. 2006N-0454]

RIN 0910-AF93

Use of Ozone-Depleting Substances; Removal of Essential-Use Designations; Public Meeting

AGENCY: Food and Drug Administration, HHS.

ACTION: Proposed rule; notice of public meeting.

SUMMARY: The Food and Drug Administration (FDA) is announcing a public meeting to solicit comments on a proposed rule that would amend FDA's regulation on the use of ozone-depleting substances (ODSs) in self-pressurized containers to remove essential-use designations for certain oral pressurized metered-dose inhalers (MDIs). In the **Federal Register** of June 11, 2007 (72 FR 32030), the agency proposed to remove the essential use designation for MDIs containing flunisolide, triamcinolone, metaproterenol, pirbuterol, albuterol and ipratropium in combination, cromolyn, and nedocromil. Information from the public meeting, which is required by agency regulations, will be considered in finalizing the rulemaking.

DATES: The public meeting will be held on August 2, 2007, from 9 a.m. to 3:30 p.m. Submit written or electronic comments for consideration at the meeting and requests to speak at the meeting by July 25, 2007. Register to attend the meeting by July 25, 2007. Submit written or electronic comments on the proposed rule and this notice by August 10, 2007.

ADDRESSES: The public meeting will be held at FDA, Center for Drug Evaluation and Research Advisory Committee Conference Room, 5630 Fishers Lane, rm. 1066, Rockville, MD 20852. You may submit comments, identified by Docket No. 2006N-0454 and RIN number 0910-AF93, by any of the following methods:

Electronic Submissions

Submit electronic comments in the following ways:

- Federal eRulemaking Portal: <http://www.regulations.gov>. Follow the instructions for submitting comments.

- Agency Web site: <http://www.fda.gov/dockets/ecomments>. Follow the instructions for submitting comments on the agency Web site.

Written Submissions

Submit written submissions in the following ways:

- FAX: 301-827-6870.
- Mail/Hand delivery/Courier [For paper, disk, or CD-ROM submissions]: Division of Dockets Management (HFA-305), Food and Drug Administration, 5630 Fishers Lane, rm. 1061, Rockville, MD 20852.

To ensure more timely processing of comments, FDA is no longer accepting comments submitted directly to the agency by e-mail. FDA encourages you to continue to submit electronic comments by using the Federal eRulemaking Portal or the agency Web site, as described previously in the **ADDRESSES** portion of this document under the *Electronic Submissions* portion of this paragraph.

Instructions: All submissions received must include the agency name and Docket No(s), and Regulatory Information Number (RIN) (if a RIN number has been assigned) for this rulemaking. All comments received may be posted without change to <http://www.fda.gov/ohrms/dockets/default.htm>, including any personal information provided. For additional information on submitting comments, see the “Comments” heading of the **SUPPLEMENTARY INFORMATION** section of this document.

Docket: For access to the docket to read the proposed rule, background documents, or comments received, go to <http://www.fda.gov/ohrms/dockets/>