

the existing coats just beyond the abraded area. Let the area dry for 30 minutes to 1 hour. Then, apply one sealer coat of Polyurethane MILC85285 TYI CL2, color Number 27925 (semi-gloss white) per Fed. Std. 595. Reinstall the blade.

Note 3: BHT-ALL-SPM, chapter 4, pertains to painting.

(d) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Contact the Safety Management Group, FAA, for information about previously approved alternative methods of compliance.

(e) Special flight permits may be issued by following 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the requirements of this AD can be accomplished provided you do not find a crack in the paint during a check or inspection.

(f) This amendment becomes effective January 18, 2005.

Note 4: The subject of this AD is addressed in Transport Canada (Canada) Airworthiness Directive CF-2004-21, dated October 28, 2004.

Issued in Fort Worth, Texas, on December 23, 2004.

Kim Smith,

*Acting Manager, Rotorcraft Directorate,
Aircraft Certification Service.*

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-182-AD; Amendment 39-13882; AD 2004-24-06]

RIN 2120-AA64

Airworthiness Directives; Saab Model SAAB SF340A and SAAB 340B Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document corrects a typographical error that appeared in airworthiness directive (AD) 2004-24-06 that was published in the **Federal Register** on November 30, 2004 (69 FR 69505). The typographical error resulted in incorrect reference to certain main landing gear (MLG) part numbers as retract actuator bracket attachment bolt (RABAB) part numbers. This AD is applicable to certain Saab Model SAAB SF340A and SAAB 340B series airplanes. This AD requires replacement of the RABAB of the MLG with a new RABAB; reidentification of the MLG shock strut; an inspection for corrosion, fretting, or other damage of certain RABABs; and applicable corrective actions.

DATES: Effective January 4, 2005.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer; International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: Airworthiness Directive (AD) 2004-24-06, amendment 39-13882, applicable to certain Saab Model SAAB SF340A and SAAB 340B series airplanes, was published in the **Federal Register** on November 30, 2004 (69 FR 69505). That AD requires replacement of the retract actuator bracket attachment bolt (RABAB) of the main landing gear (MLG) with a new RABAB; reidentification of the MLG shock strut; an inspection for corrosion, fretting, or other damage of certain RABABs; and applicable corrective actions.

As published, the AD contains an incorrect reference to the old RABAB part number. Instead of the RABAB part number, certain MLG assembly part numbers were listed as RABAB part numbers.

Since no other part of the regulatory information has been changed, the final rule is not being republished in the **Federal Register**.

The effective date of this AD remains January 4, 2005.

§ 39.13 [Corrected]

■ In the **Federal Register** of November 30, 2004, on page 69506, make the following corrections:

■ 1. In the first column, following instruction 2, the airworthiness directive number "2004-24-067" is corrected to read "2004-24-06";

■ 2. In the second column, paragraph (c) of this AD 2004-24-06 is corrected to read as follows:

* * * * *

(c) As of the effective date of this AD, no person may install a MLG shock strut, part number (P/N) AIR83022-5 through -18 inclusive, or P/N AIR83064-1 through -5 inclusive, on any airplane; and no person may install a RABAB, P/N AIR124792, on any MLG shock strut.

* * * * *

Issued in Renton, Washington, on December 21, 2004.

Kevin M. Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-28653 Filed 12-28-04; 1:42 pm]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-409-AD; Amendment 39-13853; AD 2004-22-25]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 767-200, -300, and -300F Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document corrects an error that appeared in airworthiness directive (AD) 2004-22-25, which was published in the **Federal Register** on November 9, 2004 (69 FR 64839). The error resulted in the incorrect reference to cable spacers. This AD is applicable to certain Boeing Model 767-200, -300, and -300F series airplanes. This AD requires a one-time inspection for discrepancies of all wire bundles, including certain power feeder cables, of the electrical system in the forward cargo compartment ceiling at certain stations; and corrective actions if necessary.

DATES: Effective December 14, 2004.

FOR FURTHER INFORMATION CONTACT: Elias Natsiopoulou, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6478; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION: Airworthiness Directive (AD) 2004-22-25, amendment 39-13853, applicable to certain Boeing Model 767-200, -300, and -300F series airplanes, was published in the **Federal Register** on November 9, 2004 (69 FR 64839). That AD requires a one-time inspection for discrepancies of all wire bundles, including certain power feeder cables, of the electrical system in the forward cargo compartment ceiling at certain stations; and corrective actions if necessary.

As published, the third cell of paragraph (a)(2)(ii) of Table 1 of AD 2004-22-25 states, "* * * install sleeving, lacing tape, cable spacers, and straps," in accordance with Boeing Alert Service Bulletin 767-24A0128, Revision 3, dated June 24, 2004 (cited as the appropriate service information for accomplishing the required actions). We incorrectly specified "cable spacers" as part of the installation requirements if the clearance between the power feeder cables and cargo liner standoffs is less