

PGNS-41

P63-P68

13 If Gyro Angles >5°
F 06 22 ICDU ANGLES 06,IG,MG (.01°)
PRO
NO ATT Lt - On Then Off
If Gyro Angles <5°

13a Monitor Gyro Torquing
V16N93E (.001°)
KEY REL
(If A/T 00002 OR 00003 & First Pass Through
Step 13) To 6
(If A/T 00001 & First Pass Through
Step 13) Go To 11, (If A/T 00000, Go To 14)

14 F 50 25 R1 00014 RECHECK or EXIT FINE ALIGN
(RECHECK, A/T 00002 or 00003 Only) PRO To 6
(TERM) V34E To 16

Note: If Present A/T Is 00002 & A
Previous P57 Used A/T 00001 or 00003,
ENTER To Readout Present LM
Lunar Position (Step 15)

15 F 06 89 LAT, LONG/2, ALT (.001°, .01nm)
(TERM) V34E
(ACCEPT) PRO

16 F 37

P63 BRAKING PHASE

1 V37E 63E
*PROG Lt-On *
*V05N09E 01412 IGN *
* ALGORITHM NOT *
* CONVERGING *
* (TERM) V37E00E *
* *

Basic Date October 6, 1969

Changed

PGNS-42

2 F 06 61 T6,TFI (min-sec)
 R3, CROSSRANGE (-NORTH) (.1mm)
 SET EVNT TMR TO 60-TFI
 N33E
 F 06 33 TIG (hrs,min,.01sec)
 KEY REL
 PRO

3 F 50 25 R1 00014 PERFORM IMU FINE ALIGN
 (ACCEPT) PRO - See P52/6
 (BYPASS) ENTR

4 F 50 18 REQUEST MNVR TO FDAI RPY ANGLES (.01°)
 (AUTO or TRIM) GUID CONT: PGNS
 MODE CONT: PGNS - AUTO
 PRO
 (MAN) MODE CONT: PGNS - ATT HOLD
 MNVR
 PRO To 4
 (BYPASS) ENTR To 6

5 06 18 AUTO MNVR TO FDAI RPY ANGLES (.01°)
 Mon Auto Mnr To 4

*F 50 25 00500 LR *
 * TO DESCENT POS *
 LDG ANT-DES, 10 sec, AUTO
 *PRO *
 *F 50 25 00203 *
 * GUID CONT - PGNS *
 * MODE CONT(PGNS)-AUTO*
 * THR CONT - AUTO *
 * PRO *
 * *
 *PROG Lt - On *
 *V05 N09E 01703 TIG *
 * SLIPPED *
 * V37E 00E EXIT P63 *

Basic Date | October 6, 1969 |
 Changed | |

PGNS-43

6 06 62 VI,TFI, VM (.1fps,min-sec,.1fps)

59:25 DSKY BLANKS
ENG ARM - DES

59:30 06 62 AVE G ON

59:45 Verify ΔVM (R3) <00005

59:55 F 99 62 ENG ON ENABLE
VERIFY +X ULLAGE
PRO
(NO ULLAGE) V34E Exit P63

IGN 06 63 VI (.1fps)
H DOT(-DESCENT) (.1fps)
H(+ABOVE RLS) (ft)
+:05 DES ENG CMD OVRD - ON

*(DPS ABORT) ABORT - PUSH *
(APS ABORT) ABORT STAGE - PUSH

40,000 ft V57E
F 06 68 SLANT RANGE, TG, ΔH (LR-LGC)
(.1nm, min-sec, ft)
(UPDATE) PRO
(EXIT V57) V34E (To 06 63)

F 50 68 SLANT RANGE, TG, ΔH (LR-LGC)
(.1nm, min-sec, ft)
Verify ΔH Decreasing
(STOP UPDATE) ENTR (To 06 68)
(CONTINUE UPDATE) PRO (To 06 63)

(MAN) MODE CONT:PGNS-ATT HOLD

October 6, 1969

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PGNS-44

ALT. & VEL Lt - On
 * RANGE/VELOCITY *
 * NOT GOOD *
 *PROGRAM Lt - On *
 * V05N09E *
 * 00511 LR Not in *
 * Pos 1 *
 *LDG ANT-DES,Wait *
 * 10sec, Then AUTO *

+08:30 P64 DISPLAYED

P64 APPROACH PHASE

1 P64 DISPLAYED

2 F 06 64 R1,TR/LPD, (sec-deg)
 R2 H DOT(-For Descent) (.1fps)
 R3 H(+ Landing Site Radius) (ft)

F 05 09 00523 LR DID NOT
 * ACHIEVE POS 2 *
 *(CONTINUE) PRO *
 * * *
 *(RECHECK) V32E *
 * * *
 *(TERM R12) V34E *

Monitor Attitude Change
 To Enable Landing Site
 Visibility.

(MAN) MODE CONT:PGNS-ATT HOLD
 (TO USE LPD) PRO
 (Nominal Landing Site) To 5 When TR=0

3 06 64 Observe Nominal Landing Site
 Using LPD And N64 LPD Display.

Basic Date October 6, 1969
 Changed

PGNS-45

4 Redesignate Landing Site
As Desired (+Pitch Redesignates
Landing Site Toward LM)

5 P65 DISPLAYED

P65 LANDING PHASE (AUTO)

1 P65 Displayed

2 F 06 60 V (HOR) (.1fps)
H DOT (-Descent) (.1fps)
H (+ >LANDING SITE RADIUS) (ft)

Monitor Attitude Maneuver To
Local Vertical Attitude
(+Z Downrange)

H DOT (R2)= -00030
MODE CONT: PGNS-ATT HOLD
or AUTO

(ROD LANDING) MODE CONT: PGNS-ATT HOLD
ROD - Activate
To P66

(MAN LANDING) TTCA - Advance Until:
THRUST: CMD=10%
THR CONT - MAN
To P67

(AUTO LANDING) To 3

3 H(actual)= 5.6 ft LUNAR CONTACT Lt - On
MODE CONTROL - ATT HOLD
ENGINE STOP - Push
PRO

TD+3:00

V37E 68E To P68

P66 LANDING PHASE (ROD)

1 P66 Displayed
(From P67)TTCA-Retard Gradually
THRUST IND(Left Side)-Monitor
(Maintain Constant Thrust/
Chamber Pressure)

Basic Date— October 6, 1969
Changed—

LM-f

PGNS-46

2 F 06 60 V(HOR) (.1fps)
H DOT(-Descent) (.1fps)
H(+ >Landing Site Radius) (ft)
ROD - Input ROD As Desired

(MAN LANDING) TTCA - Advance Until:
THRUST IND: CMD=10%
THR CONT - MAN
To P67

3 H(Actual)=5.6 ft LUNAR CONTACT Lt - On
ENGINE STOP - Push
TD+3:00 PRO
V37E 68E To P68

P67 LANDING PHASE (MANUAL)

1 P67 Displayed

2 F 06 60 V(HOR) (.1fps)
H DOT (-Descent) (.1fps)
H(+ >Landing Site Radius) (ft)
(ROD LANDING) THR CONT - AUTO
To P66

3 H(actual)=5.6 ft LUNAR CONTACT Lt - ON
ENGINE STOP - Push
TD+3:00 PRO
V37E 68E To P68

P68 LANDING CONFIRMATION

1 V37E 68E

2 F 06 43 LAT(+NORTH),LONG(+EAST),ALT (.01°, .1nm)
RECORD LAT _____ °
LONG _____ °
ALT _____ nm (Nominal zero)
PRO

3 F 37

Basic Date October 6, 1969
Changed _____

P70-P71

LM-6

PGNS-47

P70 DPS ABORT

1 ABORT-PUSH (From P63,64,65,66,67)
 *F 50 25 R1 00203 *
 * GUID CONT - PGNS *
 * MODE CONT: PGNS - AUTO*
 * THR CONT - AUTO *
 * PRO *

2 06 63 VI,H DOT,H (.1fps,ft)
 VI Increasing
 H DOT Remains Positive
 H Increasing

H<25000-Monitor Attitude Mnr To
 Local Vertical With Windows
 Downrange. X-axis Override
 Inhibited.

H>25000-or H DOT >00400-Monitor Atti-
 tude Mnr To Abort Attitude
 With Windows Downrange.
 X-axis Override Restored.

(To Monitor Time To Go And Crossrange Velocity)
 V16 N77E

16 77 TG,V(Y) (min-sec,.1fps)
 N85E

3 16 85 VG XYZ (LM) (.1fps)
 (If Burn >400 sec,
 PDI+6:20
 DES REG (2) - CLOSE)
 VGX = 100 fps
 DES ENG CMD OVRD - OFF

DPS NULL COMPONENTS
 OFF KEY REL

4 F 16 63 VI,H DOT, H (.1fps,ft)
 ENG STOP - Push
 ENG ARM - OFF
 ABORT - Reset
 PRO

Basic Date - October 6, 1969
 Changed - October 31, 1969

LM-6

P70-P71

PGNS-48

5 F 16 85 VG XYZ (LM) (.1fps)
 (DISPLAY ORB PARAM) V82E
 (TERM) PRO To 7

9 F 16 44 APO ALT,PER ALT,TFF (.1nm,min-sec)
 RECORD APO ALT _____ nm,
 PER ALT _____ nm,
 TFF _____ min-sec
 PRO To 5

7 F 37

P71 APS ABORT

1 ABORT STAGE -Push (From P63,64,65,
 66,67,70)

*F 50 25 R1 00203 *
 * GUID CONT - PGNS *
 * MODE CONT: PGNS - AUTO *
 * PRO *

APS IGN 06 63 VI,H DOT,H (.1fps,ft)

ENG START - Push
 ENG ARM - ASC
 If ENG STOP Lt - On
 ENG STOP - Reset
 BAL CPL-ON
 SYS A&B ASC FEED 2 (2) - OPEN
 MAIN SOV(2)-CLOSE
 CRSFD - OPEN

VI Increasing
 H DOT Remains Positive
 H Increasing

Basic Date October 6, 1969
 Changed _____

PGNS-49

H<25000-Monitor Attitude Mnr To Local Vertical With Windows Downrange. X-axis Override Inhibited.
H>25000 or H DOT>00400-Monitor Attitude Mnr To Abort Attitude With Windows Downrange. X-axis Override Restored.

(To Monitor Time To Go And Crossrange Velocity)
V16 N77E

16 77 TG,V(Y) (min-sec,.1fps)
N85E

2 16 85 VG XYZ (LM) (.1fps)
VGX = 200 fps,
MAIN SOV (2) - OPEN
SYS A&B ASC FEED 2(2)-CLOSE
CRSFD-CLOSE
VGX - 100 fps,ENG ARM - OFF

***NO Cutoff ***
*** ABORT STAGE - Reset***

APS OFF NULL COMPONENTS KEY REL

3 F 16 63 VI,H DOT,H (.1fps,ft)
ENG STOP - Push Then Reset
PRO

4 F 16 85 VG XYZ (LM) (.1fps)
(DISPLAY ORB PARAM) V82E
(TERM) PRO To 6

5 F 16 44 APO ALT,PER ALT,TFF (.1nm,min-sec)
RECORD APO ALT _____ nm,
PER ALT _____ nm,
TFF _____ min-sec
PRO To 4

6 F 37

October 6, 1969
October 31, 1969
Basic Date Changed

M-6