

opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked.”

Fuel System Modifications, Initial Inspections, and Repair if Necessary

(h) Within 24 months after the effective date of this AD, do the applicable actions

specified in Table 1 of this AD, and repair any discrepancy before further flight, in accordance with the service bulletin.

TABLE 1.—MODIFICATIONS AND INITIAL INSPECTIONS

Action	Additional source of service information for accomplishing the action
For airplanes having any serial number prior to 4962: Install new, improved fuel dump masts in accordance with paragraph 2.C.(1)(d) of the service bulletin.	Lockheed Service Bulletin 382–28–9, dated May 13, 1983.
Mark the fuel quantity indicating system (FQIS) wires in accordance with paragraph 2.C.(1)(a)2, 2.C.(4)(b), and 2.C.(4)(c) of the service bulletin.	Lockheed Service Bulletin 382–28–19, Revision 3, dated November 30, 2006.
Do the dry bay zonal inspection and inspect the static ground terminals of the fuel system plumbing in accordance with paragraph 2.C.(1)(a) of the service bulletin.	Lockheed Service Bulletin 382–28–19, Revision 3, dated November 30, 2006.
Install ground fault interrupters (GFIs) and flame arrestors for protection of the fuel system in accordance with paragraphs 2.C.(1)(b) and 2.C.(7)(c) of the service bulletin.	Lockheed Service Bulletin 382–28–20, Revision 4, dated May 21, 2007.
Inspect the GFIs for protection of the fuel system in accordance with paragraph 2.C.(1)(b)1 of the service bulletin.	Paragraph 2.C.(2) of the service bulletin.
Install the lightning bonding jumpers (straps) in accordance with paragraphs 2.C.(1)(c) and 2.C.(6)(a) of the service bulletin.	Lockheed Service Bulletin 382–28–21, Revision 2, dated November 20, 2006.
Inspect the lightning and static bonding jumpers (straps) in accordance with paragraphs 2.C.(1)(c) of the service bulletin.	Lockheed Service Bulletin 382–28–19, Revision 3, dated November 30, 2006.
Apply a certain sealant to the interior of the main wing fuel tanks; and apply a certain sealant to the all external fuel tank nose caps, mid sections, and tail sections; as applicable; in accordance with paragraphs 2.C.(1)(e)1, 2.C.(1)(e)3, and 2.C.(7)(i)1 of the service bulletin.	Lockheed Service Bulletin 382–28–24, Revision 1, dated November 5, 2007, including the Errata Notice, dated January 7, 2008.

No Alternative Inspections, Inspection Intervals, or CDCCLs

(i) After accomplishing the actions specified in paragraphs (g) and (h) of this AD, no alternative inspections, inspection intervals, or CDCCLs may be used unless the inspections, intervals, or CDCCLs are part of a later revision of the service bulletin that is approved by the Manager, Atlanta Aircraft Certification Office (ACO), FAA; or unless the inspections, intervals, or CDCCLs are approved as an alternative method of compliance in accordance with the procedures specified in paragraph (k) of this AD.

No Reporting Requirement

(j) Although Lockheed Service Bulletin 382–28–19, Revision 3, dated November 30, 2006, specifies to notify Lockheed of any discrepancies found during inspection, this AD does not require that action.

Alternative Methods of Compliance (AMOCs)

(k)(1) The Manager, Atlanta ACO, FAA, ATTN: Robert A. Bosak, Aerospace Engineer, Propulsion and Services Branch, ACE–118A, FAA, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, Suite 450, Atlanta, Georgia 30349; telephone (770) 703–6094; fax (770) 703–6097; has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District

Office (FSDO), or lacking a PI, your local FSDO.

Issued in Renton, Washington, on June 9, 2008.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8–13322 Filed 6–12–08; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2008–0649; Directorate Identifier 2008–CE–038–AD]

RIN 2120–AA64

Airworthiness Directives; DG Flugzeugbau GmbH Model DG–500MB Powered Sailplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation

product. The MCAI describes the unsafe condition as:

A DG–500MB experienced, after the engine shutdown, an uncommanded retraction of its powerplant.

Investigations revealed that some bolts of the extension retraction mechanism had fractured because of fatigue stress due to increasing push-pull loads acting on incorrectly tightened screws.

This condition, if not corrected, could lead to damage of the propeller and the fuselage, thereby reducing the structural integrity of the sailplane.

The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

DATES: We must receive comments on this proposed AD by July 14, 2008.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* (202) 493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Gregory Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; fax: (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2008-0649; Directorate Identifier 2008-CE-038-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA AD No. 2008-0095, dated May 16, 2008 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

A DG-500MB experienced, after the engine shutdown, an uncommanded retraction of its powerplant.

Investigations revealed that some bolts of the extension retraction mechanism had fractured because of fatigue stress due to increasing push-pull loads acting on incorrectly tightened screws.

This condition, if not corrected, could lead to damage of the propeller and the fuselage, thereby reducing the structural integrity of the sailplane.

To address this unsafe condition, this Airworthiness Directive mandates the replacement of eight bolts, the four connecting the fork 5M203 to the 5M204 adapter and those connecting the adapter 5M204 to the spindle drive, by new ones of higher strength and a rework of the coupling of the 5M203 fork to the 5M204 adapter as well as the coupling of the 5M204 adapter to the spindle drive, by glueing the parts together, in addition to the pre-existing bolts.

You may obtain further information by examining the MCAI in the AD docket.

Relevant Service Information

DG Flugzeugbau GmbH has issued Technical Note No. 843/27, dated April 14, 2008. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

FAA's Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Differences Between This Proposed AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the proposed AD.

Costs of Compliance

We estimate that this proposed AD would affect 4 products of U.S. registry. We also estimate that it would take about 3 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$80 per work-hour. Required parts would cost about \$63 per product.

Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$1,212, or \$303 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

DG Flugzeugbau GmbH: Docket No. FAA–2008–0649; Directorate Identifier 2008–CE–038–AD.

Comments Due Date

(a) We must receive comments by July 14, 2008.

Affected ADs

(b) None.

Applicability

(c) This AD applies to DG–500MB powered sailplanes, all serial numbers, certificated in any category.

Subject

(d) Air Transport Association of America (ATA) Code 71: Power Plant.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states: A DG–500MB experienced, after the engine shutdown, an uncommanded retraction of its powerplant.

Investigations revealed that some bolts of the extension retraction mechanism had fractured because of fatigue stress due to increasing push-pull loads acting on incorrectly tightened screws.

This condition, if not corrected, could lead to damage of the propeller and the fuselage, thereby reducing the structural integrity of the sailplane.

To address this unsafe condition, this Airworthiness Directive mandates the replacement of eight bolts, the four connecting the fork 5M203 to the 5M204 adapter and those connecting the adapter 5M204 to the spindle drive, by new ones of higher strength and, a rework of the coupling of the 5M203 fork to the 5M204 adapter as well as the coupling of the 5M204 adapter to the spindle drive, by glueing the parts together, in addition to the pre-existing bolts.

Actions and Compliance

(f) Unless already done, within the next 30 days after the effective date of this AD, modify the spindle drive assembly in accordance with DG Flugzeugbau GmbH Technical Note No. 843/27, dated April 14, 2008.

FAA AD Differences

Note: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office,

FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4130; fax: (816) 329–4090. Before using any approved AMOC on any powered sailplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

Related Information

(h) Refer to MCAI European Aviation Safety Agency (EASA) AD No. 2008–0095, dated May 16, 2008; and DG Flugzeugbau GmbH Technical Note No. 843/27, dated April 14, 2008, for related information.

Issued in Kansas City, Missouri, on June 6, 2008.

David R. Showers,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8–13324 Filed 6–12–08; 8:45 am]

BILLING CODE 4910–13–P

SOCIAL SECURITY ADMINISTRATION

20 CFR Parts 404, 408, 416, and 422

[Docket No. SSA–2008–0005]

RIN 0960–AG75

Clarification of Evidentiary Standard for Determinations and Decisions

AGENCY: Social Security Administration.
ACTION: Notice of Proposed Rulemaking.

SUMMARY: These proposed rules clarify that we apply the preponderance of the evidence standard when we make determinations and decisions at all levels of our administrative review processes. These proposed rules would not change our policy that the Appeals Council applies the substantial evidence standard when it reviews an administrative law judge's decision to determine whether to grant a request for review. We also propose to explicitly define substantial evidence and

preponderance of the evidence in applying these rules.

DATES: To be sure that your comments are considered, we must receive them no later than August 12, 2008.

ADDRESSES: You may submit comments by any one of four methods—Internet, facsimile, regular mail, or hand-delivery. Commenters should not submit the same comments multiple times or by more than one method. Regardless of which of the following methods you choose, please state that your comments refer to Docket No. SSA–2008–0005 to ensure that we can associate your comments with the correct regulation:

1. Federal eRulemaking portal at <http://www.regulations.gov>. (This is the most expedient method for submitting your comments, and we strongly urge you to use it.) In the “Comment or Submission” section of the Web page, type “SSA–2008–0005”, select “Go,” and then click “Send a Comment or Submission.” The Federal eRulemaking portal issues you a tracking number when you submit a comment.

2. Telefax to (410) 966–2830.

3. Letter to the Commissioner of Social Security, P.O. Box 17703, Baltimore, MD 21235–7703.

4. Deliver your comments to the Office of Regulations, Social Security Administration, 922 Altmeyer Building, 6401 Security Boulevard, Baltimore, Maryland 21235–6401, between 8 a.m. and 4:30 p.m. on regular business days.

All comments are posted on the Federal eRulemaking portal, although they may not appear for several days after receipt of the comment. You may also inspect the comments on regular business days by making arrangements with the contact person shown in this preamble.

Caution: All comments we receive from members of the public are available for public viewing on the Federal eRulemaking portal at <http://www.regulations.gov>. Therefore, you should be careful to include in your comments only information that you wish to make publicly available on the Internet. We strongly urge you not to include any personal information, such as your Social Security number or medical information, in your comments.

FOR FURTHER INFORMATION CONTACT: Joshua Silverman, Office of Regulations, Social Security Administration, 6401 Security Boulevard, Baltimore, MD 21235–6401, (410) 594–2128, for information about these rules. For information on eligibility or filing for benefits, call our national toll-free number, 1–800–772–1213 or TTY 1–800–325–0778, or visit our Internet site,