address in the **FOR FURTHER INFORMATION CONTACT** section.

The Office of the U.S. Trade Representative has reviewed this rule and concurs with its issuance.

After consideration of all relevant material presented, including the Committee's recommendation, and other information, it is found that finalizing the interim final rule, without change, as published in the **Federal Register** (68 FR 43614, July 24, 2003) will tend to effectuate the declared policy of the Act.

List of Subjects

7 CFR Part 993

Marketing agreements, Plums, Prunes, Reporting and recordkeeping requirements.

7 CFR Part 999

Dates, Filberts, Food grades and standards, Imports, Nuts, Prunes, Raisins, Reporting and recordkeeping requirements, Walnuts.

PART 993—DRIED PRUNES PRODUCED IN CALIFORNIA

PART 999—SPECIALTY CROPS: IMPORT REGULATIONS

Accordingly, the interim final rule amending 7 CFR parts 993 and 999 which was published at 68 FR 43614 on July 24, 2003, is adopted as a final rule without change.

Dated: October 17, 2003.

A.J. Yates,

Administrator, Agricultural Marketing Service.

[FR Doc. 03–26712 Filed 10–22–03; 8:45 am] BILLING CODE 3410–02–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-164-AD; Amendment 39-13308; AD 2003-19-05]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model DC-10-10, -10F, -15, -30, -30F (KC-10A and KDC-10), -40, and -40F Airplanes; and Model MD-10-10F and -30F Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document corrects an error that appeared in airworthiness directive (AD) 2003–19–05 that was

published in the **Federal Register** on September 22, 2003 (68 FR 54992). The error resulted in an incorrect Type Certificate holder name. This AD is applicable to certain McDonnell Douglas Model DC–10–10, –10F, –15, –30, –30F (KC–10A and KDC–10), –40, and –40F airplanes; and certain Model MD–10–10F and –30F airplanes. This AD requires inspections for cracking and corrosion of the bolt assemblies and bushings on the hinge fittings of the inboard and outboard flaps of the left and right wings, and follow-on and corrective actions.

DATES: Effective October 27, 2003.

FOR FURTHER INFORMATION CONTACT: Ron Atmur, Aerospace Engineer, Airframe Branch, ANM-120L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5224; fax (562) 627-5210.

SUPPLEMENTARY INFORMATION:

Airworthiness Directive (AD) 2003–19–05, amendment 39–13308, applicable to certain McDonnell Douglas Model DC–10–10, –10F, –15, –30, –30F (KC–10A and KDC–10), –40, and –40F airplanes; and certain Model MD–10–10F and –30F airplanes; was published in the **Federal Register** on September 22, 2003 (68 FR 54992). That AD requires inspections for cracking and corrosion of the bolt assemblies and bushings on the hinge fittings of the inboard and outboard flaps of the left and right wings, and follow-on and corrective actions.

As published, the Type Certificate (TC) holder name appears as "BOEING" in the regulatory text of the AD. The correct TC holder name is McDonnell Douglas, which is correctly referenced throughout the preamble of the the AD.

Since no other part of the regulatory information has been changed, the final rule is not being republished in the **Federal Register**.

The effective date of this AD remains October 27, 2003.

§39.13 [Corrected]

On page 54993, in the second column, paragraph 2. of Part 39—Airworthiness Directives of AD 2003–19–05 is corrected to read as follows:

2. Section 39.13 is amended by adding the following new airworthiness directive:

2003-19-05 McDonnell Douglas:

Amendment 39–13308. Docket 2002–NM–164–AD.

* * * * *

Issued in Renton, Washington, on October 17, 2003.

Neil D. Schalekamp,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 03–26721 Filed 10–22–03; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-229-AD; Amendment 39-13347; AD 98-16-17 R1]

RIN 2120-AA64

Airworthiness Directives; Cessna Model 750 Citation X Series Airplanes

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule; rescission.

SUMMARY: This amendment rescinds Airworthiness Directive (AD) 98–16–17 R1, which is applicable to all Cessna Model 750 Citation X series airplanes. That AD requires repetitive in-flight functional tests to verify proper operation of the secondary horizontal stabilizer pitch trim system, and repair if necessary. The requirements of that AD were intended to detect and correct contamination and damage in the system actuator, which could result in simultaneous failure of both primary and secondary pitch trim systems, and consequent reduced controllability of the airplane. Since the issuance of that AD, an improved part has been developed, which, if installed, would terminate the repetitive tests; that improved part has been installed on all affected airplanes or is being installed in production. Therefore, the identified unsafe condition no longer exists. EFFECTIVE DATE: October 23, 2003.

FOR FURTHER INFORMATION CONTACT: Joel M. Ligon, Aerospace Engineer, Systems and Propulsion Branch, ACE–116W, FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; telephone (316) 946–4138; fax (316) 946–4407.

SUPPLEMENTARY INFORMATION: A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Cessna Model 750 Citation X series airplanes was published in the **Federal Register** on January 7, 2000 (65 FR 1075). That action proposed to rescind AD 98–16–17, amendment 39–10693 (63 FR 42206,