related investigative and corrective actions, in accordance with the Accomplishment Instructions of Boeing Service Bulletin 737-57-1275, dated September 4, 2003, except as provided by paragraph (g) of this AD.

For airplanes modified by Supplemental Type Certificate (STC) ST00830SE as of the effective date of this AD: Prior to the accumulation of 25,000 total flight hours or 25,000 total flight cycles, whichever is first.

(2) For airplanes not modified by STC ST00830SE as of the effective date of this AD: Prior to the accumulation of 30,000 total flight hours or 30,000 total flight cycles, whichever is first.

(g) If accomplishing a corrective action as required by paragraph (f) of this AD, and the service bulletin specifies to contact Boeing for repair information: Before further flight, do the repair in accordance with a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA; or in accordance with data meeting the type certification basis of the airplane approved by a Boeing Company Designated Engineering Representative (DER) who has been authorized by the Manager, Seattle ACO, to make such findings. For a repair method to be approved, the approval must specifically reference this AD

#### **Alternative Methods of Compliance** (AMOCs)

(h)(1) The Manager, Seattle ACO, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

(2) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by a Boeing Company DER who has been authorized by the Manager, Seattle ACO, to make such findings. For a repair method to be approved, the approval must specifically refer to this AD.

Issued in Renton, Washington, on November 10, 2004.

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-26031 Filed 11-23-04; 8:45 am] BILLING CODE 4910-13-P

# **DEPARTMENT OF TRANSPORTATION**

# **Federal Aviation Administration**

# 14 CFR Part 39

[Docket No. FAA-2004-19676; Directorate Identifier 2004-NM-138-AD]

RIN 2120-AA64

Airworthiness Directives; Empresa Brasileira de Aeronautica S.A. (EMBRAER) Model EMB-135 and -145 Series Airplanes

**AGENCY: Federal Aviation** Administration (FAA), DOT. **ACTION:** Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for certain EMBRAER Model EMB-135 and –145 series airplanes. This proposed AD would require determining the torque values of the screws that attach the seat tracks to the airplane, and corrective action if necessary. This proposed AD is prompted by a report of undertorqued screws. We are proposing this AD to prevent improper torque of those screws, which in the case of a hard landing or a high deceleration impact condition could result in damage to the seat and possible subsequent injury to the passenger.

DATES: We must receive comments on this proposed AD by December 27, 2004.

**ADDRESSES:** Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to http://www.regulations.gov and follow the instructions for sending your comments electronically.
- Mail: Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Nassif Building, room PL-401, Washington, DC 20590.

• By fax: (202) 493–2251.

 Hand Delivery: Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Empresa Brasileira de Aeronautica S.A. (EMBRAER), P.O. Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil.

You can examine the contents of this AD docket on the Internet at http:// dms.dot.gov, or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, on the plaza level of the Nassif Building, Washington, DC.

# FOR FURTHER INFORMATION CONTACT:

Technical information: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1175; fax (425) 227-1149.

Plain language information: Marcia Walters, marcia.walters@faa.gov.

# SUPPLEMENTARY INFORMATION:

# **Docket Management System (DMS)**

The FAA has implemented new procedures for maintaining AD dockets electronically. As of May 17, 2004, new

AD actions are posted on DMS and assigned a docket number. We track each action and assign a corresponding directorate identifier. The DMS AD docket number is in the form "Docket No. FAA-2004-99999." The Transport Airplane Directorate identifier is in the form "Directorate Identifier 2004-NM-999-AD." Each DMS AD docket also lists the directorate identifier ("Old Docket Number") as a cross-reference for searching purposes.

# **Comments Invited**

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed under ADDRESSES. Include "Docket No. FAA-2004–19676; Directorate Identifier 2004–NM–138–AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments submitted by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to http:// dms.dot.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of our docket website, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You can review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477–78), or you can visit http:// dms.dot.gov.

We are reviewing the writing style we currently use in regulatory documents. We are interested in your comments on whether the style of this document is clear, and your suggestions to improve the clarity of our communications that affect you. You can get more information about plain language at http://www.faa.gov/language and http:// www.plainlanguage.gov.

# **Examining the Docket**

You can examine the AD docket on the Internet at http://dms.dot.gov, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT

street address stated in the ADDRESSES section. Comments will be available in the AD docket shortly after the DMS receives them.

#### Discussion

The Departmento de Aviacao Civil (DAC), which is the airworthiness authority for Brazil, notified us that an unsafe condition may exist on certain EMBRAER Model EMB-135 and -145 series airplanes. The DAC advises that some screws that attach the passenger seat tracks had been undertorqued during manufacture. This condition, if not corrected, could result in damage to the seat and possible subsequent injury to the passenger.

#### **Relevant Service Information**

EMBRAER has issued Service Bulletins 145LEG-53-0015 (for Model EMB-135 BJ series airplanes) and 145-

53-0049 (for Model EMB-135 and -145 series airplanes), both dated February 16, 2004. The service bulletins describe procedures for inspecting for proper torque of the screws that attach the seat tracks to the airplane, retorquing the screws if necessary, and applying torque seal. Accomplishing the actions specified in the service information is intended to adequately address the unsafe condition. The DAC mandated the service information and issued Brazilian airworthiness directive 2004-05-03, dated June 2, 2004, to ensure the continued airworthiness of these airplanes in Brazil.

# FAA's Determination and Requirements of the Proposed AD

These airplane models are manufactured in Brazil and are type certificated for operation in the United States under the provisions of § 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DAC has kept the FAA informed of the situation described above. We have examined the DAC's findings, evaluated all pertinent information, and determined that we need to issue an AD for products of this type design that are certificated for operation in the United States.

Therefore, we are proposing this AD, which would require accomplishing the actions specified in the service information described previously.

# **Costs of Compliance**

The following table provides the estimated costs for U.S. operators to comply with this proposed AD.

# **ESTIMATED COSTS**

Airplane(s)	Work hours	Average labor rate/hour	Parts	Cost/airplane
EMB-135 BJOthers	24	\$65	Minimal	\$1,560
	28	65	Minimal	1,820

The cost for the U.S. fleet of 539 airplanes would be \$840,840 to \$980,980, depending on the airplane model

# **Regulatory Findings**

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;

2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. See the ADDRESSES section for a location to examine the regulatory evaluation.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

# **The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

# § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Empresa Brasileira De Aeronautica S.A. (EMBRAER): Docket No. FAA–2004–19676; Directorate Identifier 2004–NM–138–AD.

#### **Comments Due Date**

(a) The Federal Aviation Administration must receive comments on this AD action by December 27, 2004.

# Affected ADs

(b) None.

# **Applicability**

(c) This AD applies to EMBRAER Model EMB-135 and -145 series airplanes, certificated in any category; as listed in EMBRAER Service Bulletin 145LEG-53-0015 or 145-53-0049, both dated February 16, 2004.

#### **Unsafe Condition**

(d) This AD was prompted by a report indicating that some screws that attach the passenger seat tracks were undertorqued. We are issuing this AD to prevent improper torque of those screws, which in the case of a hard landing or a high deceleration impact condition could result in damage to the seat and possible subsequent injury to the passenger.

# Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

### Inspection

(f) Within 5,000 flight hours or 36 months after the effective date of this AD, whichever occurst first, determine the torque values of the screws that attach the seat tracks to the airplane. Use EMBRAER Service Bulletin 145LEG-53-0015 (for Model EMB-135 BJ series airplanes) or 145-53-0049 (for the remaining affected airplanes), both dated February 16, 2004, to do the inspection. Before further flight, retorque any screw having improper torque and apply torque seal to all the screws, in accordance with the service bulletin.

# Alternative Methods of Compliance (AMOCs)

(g) The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

# **Related Information**

(h) Brazilian airworthiness directive 2004–05–03, dated June 2, 2004, also addresses the subject of this AD.

Issued in Renton, Washington, on November 10, 2004.

#### Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04–26030 Filed 11–23–04; 8:45 am] BILLING CODE 4910–13–P

# **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. FAA-2004-19680; Directorate Identifier 2003-NM-215-AD]

RIN 2120-AA64

# Airworthiness Directives; Boeing Model 767 Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking

(NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for certain Boeing Model 767 series airplanes. This proposed AD would require performing a test of the bonding resistance between the engine fuel feed tube fitting and the front spar, applying sealant on a hex nut inside the dry bay, and performing any applicable corrective actions. This proposed AD is prompted by a report that the engine fuel feed tubes were found not electrically bonded to the front spar. We are proposing this AD to prevent an ignition source from entering the fuel tank during a lightning strike event, which could cause a fuel tank explosion.

**DATES:** We must receive comments on this proposed AD by January 10, 2005. **ADDRESSES:** Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to <a href="http://www.regulations.gov">http://www.regulations.gov</a> and follow the instructions for sending your comments electronically.
- Mail: Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Nassif Building, room PL-401, Washington, DC 20590.
  - By fax: (202) 493–2251.
- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building,

400 Seventh Street SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207.

You can examine the contents of this AD docket on the Internet at <a href="http://dms.dot.gov">http://dms.dot.gov</a>, or at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL–401, on the plaza level of the Nassif Building, Washington, DC.

### FOR FURTHER INFORMATION CONTACT:

Technical Information: Bernie Gonzalez, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6498; fax (425) 917-6590.

Plain Language Information: Marcia Walters, marcia.walters@faa.gov.

# SUPPLEMENTARY INFORMATION:

# **Docket Management System (DMS)**

The FAA has implemented new procedures for maintaining AD dockets electronically. As of May 17, 2004, new AD actions are posted on DMS and assigned a docket number. We track each action and assign a corresponding directorate identifier. The DMS AD docket number is in the form "Docket No. FAA–2004–99999." The Transport Airplane Directorate identifier is in the form "Directorate Identifier 2004–NM–999–AD." Each DMS AD docket also lists the directorate identifier ("Old Docket Number") as a cross-reference for searching purposes.

# **Comments Invited**

We invite you to submit any written relevant data, views, or arguments regarding this proposed AD. Send your comments to an address listed under ADDRESSES. Include "Docket No. FAA–2004–19680; Directorate Identifier 2003–NM–215–AD" in the subject line of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments submitted by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to http://dms.dot.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that website, anyone can find and read the

comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You can review DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477–78), or you can visit http://dms.dot.gov.

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# **Examining the Docket**

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#### Discussion

We have received a report that, during electrical bonding and grounding tests of penetrations on the wing fuel tanks of Boeing Model 747 series airplanes, the feed tubes for the engine fuel were found not electrically bonded to the front spar. The same condition was found on Model 767 series airplanes; Model 737–100, -200, -200C, -300, -400, and -500 series airplanes; and Model 707 series airplanes. This condition, if not corrected, could result in an ignition source entering the fuel tank during a lightning strike event, which could cause a fuel tank explosion.

# **Other Relevant Rulemaking**

We previously issued AD 2004–10–06, amendment 39–13636 (69 FR 28046, May 18, 2004), applicable to Model 747 series airplanes; Model 737–100, –200, –200C, –300, –400, and –500 series airplanes; and Model 727–100, and –200 series airplanes. That AD was issued to ensure that the similar unsafe condition (hydraulic heat exchanger tube penetration fittings were found not electrically bonded to the fuel tank rear spar) was repaired. That AD requires, among other things, preparation of the electrical bonding faying surfaces for the tubing penetrations of the hydraulic