

109EP-27 in the "Supplementary" section of the final rule, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that this AD will affect 18 helicopters of U.S. registry. It will take approximately 3 work hours to disable the windshield wipers and modify the electrical system of the windshield wipers and 4 work hours per helicopter if the timed relays must be replaced by modifying the electrical system of the windshield wipers. The average labor rate is \$65 per work hour. Required parts will cost approximately \$367 per helicopter. Based on these figures, we estimate the total cost impact of the AD on U.S. operators is \$14,796, assuming the relays are replaced on the entire fleet. However, the manufacturer states in its ABT that it will reimburse owners for 3 or 4 work hours at a fixed rate of \$40 per work hour and will provide the parts for free. Assuming a warranty credit of 4 work hours (\$2,880) and free parts (\$6,606), the estimated total cost impact of this AD is \$5,310.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

2004-11-06 Agusta S.p.A: Amendment 39-13652. Docket No. 2003-SW-32-AD.

Applicability: Model A109E helicopters, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent the incompatibility of certain relays with the windshield wiper electrical system, overheating of the resistor due to system overload, and an electrical fire, accomplish the following:

(a) For helicopters, serial number (S/N) 11502 through 11504, and 11122 through 11130, except 11123, 11127, and 11129:

(1) Within 5 hours time-in-service, do the following:

(i) Disable the windshield wipers by following the Compliance Instructions, Part I, paragraphs 2.1 through 2.5, of Agusta Alert Bollettino Tecnico No. 109EP-27, Revision A, dated February 7, 2003 (ABT).

(ii) Install a placard stating that the windshield wipers are inoperative by following the Compliance Instructions, Part I, paragraph 2.6, of the ABT.

(2) Within 6 months, modify the electrical system of the windshield wipers using the Compliance Instructions, Part II, paragraphs 1. through 15., of the ABT, and remove the placard that was installed as required by paragraph (a)(1)(ii) of this AD.

(b) For helicopters, S/Ns 11151, 11501, and 11001 through 11133, except 11122, 11124 through 11128, and 11130, with timed relay, part number (P/N) T412-DJ1001-C installed, on or before June 6, 2005, or when you replace a timed relay, P/N T412-DJ1001-C, with either relay, P/N TDH-8070-1001P or P/N T412-2006, whichever occurs first:

(1) If windshield wiper kit, P/N 109-0811-44-105 or -106 is installed, modify the windshield wiper electrical system and replace the timed relay, P/N T412-DJ1001-C, with a timed relay, P/N TDH-8070-1001P or P/N T412-2006, by following the Compliance Instructions, Part III, paragraphs 1. through 1.16, of the ABT.

(2) If windshield wiper kit, P/N 109-0811-44-101 or -102 is installed, modify the windshield wiper electrical system and replace the timed relay, P/N T412-DJ1001-C, with a timed relay, P/N TDH-8070-1001P or P/N T412-2006, by following the Compliance Instructions, Part III, paragraphs 2. through 2.19, of the ABT.

(c) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Contact the Safety Management Group, FAA, for information about previously approved alternative methods of compliance.

(d) Modifying the windshield wiper electrical system shall be done following the

Agusta Alert Bollettino Tecnico No. 109EP-27, Revision A, dated February 7, 2003. The Director of the Federal Register approved this incorporation by reference in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Agusta, 21017 Cascina Costa di Samarate (VA) Italy, Via Giovanni Agusta 520, telephone 39 (0331) 229111, fax 39 (0331) 229605-222595. Copies may be inspected at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

(e) This amendment becomes effective on July 9, 2004.

Note: The subject of this AD is addressed in Ente Nazionale per l'Aviazione Civile (Italy), AD No. 2003-032, dated February 10, 2003.

Issued in Fort Worth, Texas, on May 21, 2004.

David A. Downey,

Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 04-12440 Filed 6-3-04; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-SW-29-AD; Amendment 39-13650; AD 2004-11-05]

RIN 2120-AA64

Airworthiness Directives; Eurocopter France Model EC 130 B4 and AS 350 B3 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for the specified Eurocopter France (Eurocopter) model helicopters that requires inspecting the fuel transfer line and air exhaust duct for chafing, inspecting the air exhaust duct for a hole, and if necessary, repositioning the air exhaust duct to achieve the minimum clearances. This amendment is prompted by a report of damage to the fuel transfer line due to wear associated with vibrations and chafing of the fuel transfer line and the air exhaust duct. The actions specified by this AD are intended to detect chafing wear of the air exhaust duct and the fuel transfer line, which could result in a hole in the fuel transfer line, fuel leaking into the

engine compartment and creating a fire hazard that could lead to a fire and a subsequent forced landing.

DATES: Effective July 9, 2004.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 9, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005, telephone (972) 641-3460, fax (972) 641-3527. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT: Ed Cuevas, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Safety Management Group, Fort Worth, Texas 76193-0111, telephone (817) 222-5355, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION: A proposal to amend 14 CFR part 39 to include an AD for the specified model helicopters was published in the **Federal Register** on January 8, 2004 (69 FR 1275). That action proposed to require inspecting the fuel transfer line and air exhaust duct for chafing, and if necessary, repositioning the air exhaust duct to achieve at least 20 mm (0.8 in) of clearance in interference Area A and 12 mm (0.5 in) of clearance in interference Area B as depicted in Figure 1 of Eurocopter Alert Service Bulletin (ASB) No. 71A001 for Model EC 130 B4 helicopters and ASB No. 71.00.16 for Model AS 350 B3 helicopters, both dated May 12, 2003. These are one-time inspections.

The Direction Generale De L'Aviation Civile (DGAC), the airworthiness authority for France, notified the FAA that an unsafe condition may exist on Eurocopter Model EC 130 B4 and AS 350 B3 helicopters. The DGAC advises of receiving a report of damage to the fuel transfer line due to interference associated with vibrations and chafing of the bleed valve air exhaust duct.

Eurocopter has issued ASB No. 71A001 for Model EC 130 B4 helicopters and ASB No. 71.00.16 for Model AS 350 B3 helicopters, both dated May 12, 2003, which specify checks for interference between the bleed valve air exhaust duct and the

engine fuel line. The DGAC classified these ASBs as mandatory and issued ADs No. 2003-208(A) and 2003-209(A), both dated May 28, 2003, to ensure the continued airworthiness of these helicopters in France.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that this AD will affect 100 helicopters of U.S. registry, and the required actions will take approximately 0.5 work hour per helicopter to accomplish and 1 work hour to replace either the fuel transfer line or the air exhaust duct at an average labor rate of \$65 per work hour. Required parts will cost approximately \$817 for the fuel transfer line and \$522 for the air exhaust duct. Based on these figures, we estimate the total cost impact of the AD on U.S. operators to be \$6,188, assuming 2 fuel transfer lines and 2 air exhaust ducts are replaced.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

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Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

2004-11-05 Eurocopter France:

Amendment 39-13650. Docket No. 2003-SW-29-AD.

Applicability: Model EC 130 B4 helicopters with an optional engine flushing system installed, and AS 350 B3 helicopters with an optional engine flushing system installed and modified in accordance with MOD 073098, certificated in any category.

Compliance: Required within 10 hours time-in-service, unless accomplished previously.

To detect chafing wear of the air exhaust duct and the fuel transfer line, which could result in a hole in the fuel transfer line, fuel leaking into the engine compartment and creating a fire hazard that could lead to a fire and a subsequent forced landing, accomplish the following:

(a) Inspect the fuel transfer line located between the bleed valve of the engine starting system and the engine fuel filter for chafing in the interference areas in accordance with the Operational Procedure, paragraph 2.B.1., of Eurocopter Alert Service Bulletin (ASB) No. 71A001, dated May 12, 2003, for Model EC 130 B4 helicopters, or Eurocopter ASB No. 71.00.16, dated May 12, 2003, for Model AS 350 B3 helicopters.

(1) If the depth of the deepest wear mark is less than or equal to 0.05 mm (0.002 in), apply the maintenance procedure stated in the Engine Maintenance Manual.

(2) If the depth of the deepest wear mark is more than 0.05 mm (0.002 in) and less than or equal to 0.2 mm (0.008 in), replace the fuel transfer line within the next 50 hours TIS or within one month, whichever occurs first.

(3) If the depth of the deepest wear mark is more than 0.2 mm (0.008 in), replace the fuel transfer line before further flight.

(b) Inspect the air exhaust duct located between the bleed valve of the engine starting system and the engine fuel filter for a hole in the interference areas in accordance with the Operational Procedure, paragraph 2.B.1., of Eurocopter ASB No. 71A001, dated May 12, 2003, for Model EC 130 B4 helicopters, or Eurocopter ASB No. 71.00.16, dated May 12, 2003, for Model AS 350 B3 helicopters. If there is a hole in the air exhaust duct, replace the air exhaust duct within one month or before performing any engine flushing operation, whichever occurs first.

(c) Measure the clearances between the fuel transfer line and the air exhaust duct located between the bleed valve of the engine starting system and the engine fuel filter in the interference areas in accordance with the Operational Procedure, paragraph 2.B.1., of Eurocopter ASB No. 71A001, dated May 12, 2003, for Model EC 130 B4 helicopters, or Eurocopter ASB No. 71.00.16, dated May 12,

2003, for Model AS 350 B3 helicopters. If the clearance is less than 20 mm (0.8 in) in interference Area A or less than 12 mm (0.5 in) in interference Area B, reposition the air exhaust duct in accordance with the Operational Procedure, paragraph 2.B.2., of Eurocopter ASB No. 71A001, dated May 12, 2003, for Model EC 130 B4 helicopters, or Eurocopter ASB No. 71.00.16, dated May 12, 2003, for Model AS 350 B3 helicopters.

(d) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Contact Manager, Safety Management Office, Rotorcraft Directorate, FAA, for information about previously approved alternative methods of compliance.

(e) The inspections, measuring, and repositioning, if necessary, shall be done in accordance with Eurocopter ASB No. 71A001 for Model EC 130 B4 helicopters and ASB No. 71.00.16 for Model AS 350 B3 helicopters, both dated May 12, 2003. The Director of the Federal Register approved this incorporation by reference in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005. Copies may be inspected at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

(f) This amendment becomes effective on July 9, 2004.

Note: The subject of this AD is addressed in Direction Generale De L'Aviation Civile (France) AD 2003-208(A) and AD 2003-209(A), both dated May 28, 2003.

Issued in Fort Worth, Texas, on May 21, 2004.

David A. Downey,

Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 04-12441 Filed 6-3-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Parts 121 and 139

[Docket No. FAA-2000-7479; Amendment Nos. 121-304, 139-26]

RIN 2120-AG96

Certification of Airports; Correction

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; correction.

SUMMARY: The Federal Aviation Administration (FAA) is making minor technical changes to a final rule

published in the **Federal Register** on February 10, 2004 (69 FR 6380). That final rule revises the airport certification regulations and establishes certification requirements for certain airports.

FOR FURTHER INFORMATION CONTACT:

Linda Bruce, Airport Safety and Operations Division, Office of Airport Safety and Standards, 800 Independence Avenue SW., Washington, DC 20591; telephone (202) 267-8553.

DATES: Effective Date: This correction is effective on June 9, 2004.

SUPPLEMENTARY INFORMATION: The FAA published in the **Federal Register** of February 10, 2004 (69 FR 6380), a final rule revising the airport certification regulation and establishing certification requirements for airports serving scheduled air carrier operations in aircraft designed for more than 9 passenger seats, but less than 31 passenger seats. The final rule also amends the air carrier operation regulations to conform with changes to airport certification requirements. The final rule is necessary to ensure safety in air transportation at all certificated airports and becomes effective June 9, 2004.

The final rule preamble states that air carriers can continue to operate aircraft with more than 9 seats, but less than 31 seats, into airports that are not obligated to obtain the appropriate airport operating certificate until December 9, 2005. However, the rule language is causing the regulated community some uncertainty in interpreting this provision. Therefore, the FAA is clarifying this rule language. This clarification is consistent with the intent of the preamble for the final rule and will remove uncertainty in the regulated community. In addition, there are several minor technical edits to the rule language.

We intend no substantive changes to any of the requirements established by the final rule. These corrections do not impose any additional requirements on operators affected by these regulations.

Justification for Expedited Rulemaking

Section 553 of the Administrative Procedure Act, 5 U.S.C. 553(b)(B), provides that, when an agency for good cause finds that notice and public procedure are impracticable, unnecessary, or contrary to the public interest, the agency may issue a rule without providing notice and an opportunity for public comment. We have determined there is good cause for making today's action final without prior proposal and opportunity for comment because the changes to the

rule are minor technical corrections and do not change the requirements of the rule. Thus, notice and public procedure are unnecessary.

Corrections

■ In final rule FR Doc. 04-2255, published on February 10, 2004 (69 FR 6380), make the following corrections:

■ 1. On page 6380, in column 1 in the heading section, beginning on line four, correct "Amendment Nos. 121-304, 135-94" to read "Amendment Nos. 121-304, 139-26".

§ 121.590 [Corrected]

■ 2. On page 6424, in column 1, § 121.590(b), correctly designate paragraph (b) as (b)(1).

■ 3. On page 6424, in column 1, § 121.590, add paragraph (b)(2) to read as follows:

* * * * *

(2) Until December 9, 2005, an air carrier and a pilot being used by the air carrier in the conduct of domestic type operations and flag type operations, may operate an airplane designed for more than 9 but less than 31 passenger seats, at a land airport, in any State of the United States, the District of Columbia, or any territory or possession of the United States, that does not hold an airport operating certificate issued under part 139 of this chapter, and that serves small air carrier aircraft (as defined under "Air carrier aircraft" and "Class III airport" in § 139.5 of this Chapter).

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§ 139.203 [Corrected]

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■ 4. On page 6428, § 139.203(b), in item 23 of the table, in the fifth column, add an "X" under Class IV.

§ 139.303 [Corrected]

■ 5. On page 6429, in columns 1 and 2, § 139.303(e), correctly designate subparagraphs (i) through (vi) as (1) through (6).

§ 139.305 [Corrected]

■ 6. On page 6429, in column 2, § 139.305, correct the text of paragraph(a)(3) to read as follows:

* * * * *

(3) The pavement must be free of cracks and surface variations that could impair directional control of air carrier aircraft, including any pavement crack or surface deterioration that produces loose aggregate or other contaminants.

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§ 139.315 [Corrected]

■ 7. On page 6431, in column 1, § 139.315(e), correctly designate