located on the outboard flaps, or within 25 flight cycles after August 11, 1999, whichever occurs later: Measure the screw/nut play of the jackscrews having P/N 1–5319–1 or 1–5319–1 Amdt A (on the left wing) and P/N 2–5319–1 or 2–5319–1 Amdt A (on the right wing) on the outboard flaps, in accordance with the procedures specified in Dassault Falcon 2000 AMM Temporary Revision (TR) 27–504, dated October 1998.

Note 2: Jackscrews having P/N 1–5319–1 or 2–5319–1 may be reconditioned in accordance with Dassault Service Bulletin AVIAC 5319–27–01, dated September 16, 1999. These jackscrews may be reconditioned and reused more than one time.

(1) If the initial measurement is equal to or less than 0.014 inch: Repeat the measurement thereafter at intervals not to exceed 330 flight hours or 7 months, whichever occurs first. If any repetitive measurement detects a nut/screw play greater than 0.014 inch, perform the actions required by paragraph (f)(2) of this AD.

(2) If the initial measurement is greater than 0.014 inch: Perform the actions required by paragraphs (f)(2)(i) and (f)(2)(ii) of this AD.

- (i) Prior to further flight, replace the jackscrew with a new or reconditioned jackscrew, in accordance with Dassault Falcon 2000 AMM 27–510, dated November 1995.
- (ii) Prior to the accumulation of 1,000 total flight cycles on the new or reconditioned jackscrew, perform a follow-on measurement of the screw/nut play in accordance with the procedures specified in Dassault Falcon 2000 AMM TR 27–504, dated October 1998.
- (iii) If any follow-on measurement required by paragraph (f)(2)(ii) of this AD detects a nut/screw play equal to or less than 0.014 inch, perform the actions required by paragraph (f)(1) of this AD. If any follow-on measurement required by paragraph (f)(2)(ii) of this AD detects a nut/screw play greater than 0.014 inch, perform the actions required by paragraphs (f)(2)(i) and (f)(2)(ii) of this AD.
- (g) Prior to the accumulation of 750 total flight cycles on the jackscrew located on the inboard flap in the inboard position, or within 25 flight cycles after the effective date of this AD, whichever occurs later: Measure the screw/nut play of the jackscrew having P/ N 5318-1 or 1-5318-1 Amdt A, which is located on the inboard flap in the inboard position, to detect discrepancies, in accordance with the procedures specified in Dassault Falcon 2000 AMM TR 27-504. dated October 1998. If the measurement is greater than 0.014 inch, prior to further flight, replace the discrepant jackscrew with a new or reconditioned jackscrew, in accordance with Dassault Falcon 2000 AMM 27-510. dated November 1995.
- (h) Prior to the accumulation of 1,000 total flight cycles on the jackscrew located on the inboard flap in the outboard position, or within 25 flight cycles after the effective date of this AD, whichever occurs later: Measure the screw/nut play of the jackscrew having P/N 5318–1 or 5318–1 Amdt A, which is located on the inboard flap in the outboard position, in accordance with the procedures specified in Dassault Falcon 2000 AMM TR 27–504, dated October 1998.
- (1) If the initial measurement is equal to or less than 0.014 inch: Repeat the

measurements thereafter at intervals not to exceed 330 flight hours or 7 months, whichever occurs first. If repetitive measurement detects a nut/screw play greater than 0.014 inch, perform the actions required by paragraph (h)(2) of this AD.

(2) If the initial measurement is greater than 0.014 inch: Perform the actions required by paragraphs (h)(2)(i) and (h)(2)(ii) of this AD

- (i) Prior to further flight, replace the jackscrew with a new or reconditioned jackscrew, in accordance with Dassault Falcon 2000 AMM 27–510, dated November 1995.
- (ii) Prior to the accumulation of 1,000 total flight cycles on the new or reconditioned jackscrew, perform a follow-on measurement of the screw/nut play in accordance with the procedures specified in Dassault Falcon 2000 AMM TR 27–504, dated October 1998.
- (iii) If any follow-on measurement required by paragraph (h)(2)(ii) of this AD detects a nut/screw play equal to or less than 0.014 inch, perform the actions required by paragraph (h)(1) of this AD. If any follow-on measurement required by paragraph (h)(2)(ii) of this AD detects a nut/screw play greater than 0.014 inch, perform the actions required by paragraphs (h)(2)(i) and (h)(2)(ii) of this AD.

Alternative Methods of Compliance

- (i)(1) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.
- (2) Alternative methods of compliance, approved previously in accordance with AD 99–14–07, amendment 39–11218, are not considered to be approved as alternative methods of compliance with this AD.

Special Flight Permits

(j) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(k) Unless otherwise specified in this AD, the actions shall be done in accordance with Dassault Falcon 2000 Airplane Maintenance Manual Temporary Revision 27–504, dated October 1998. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Dassault Falcon Jet, P.O. Box 2000, South Hackensack, New Jersey 07606. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in French airworthiness directive 1999–038–008(B) R1, dated September 20, 2000.

Effective Date

(l) This amendment becomes effective on January 6, 2003.

Issued in Renton, Washington, on November 19, 2002.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 02–30025 Filed 11–29–02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-40-AD; Amendment 39-12969; AD 2002-24-04]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B2 and B4 Series Airplanes; A300 B4–600, B4–600R, and F4–600R (Collectively Called A300–600) Series Airplanes; and Model A310 Series Airplanes

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that is applicable to all Airbus Model A300 B2 and B4 series airplanes; A300 B4-600, B4-600R, and F4-600R (collectively called A300-600) series airplanes; and Model A310 series airplanes. This amendment requires revising the Airplane Flight Manual to advise the flightcrew to don oxygen masks as a first and immediate step when the cabin altitude warning horn sounds. This action is necessary to prevent incapacitation of the flightcrew due to lack of oxygen, which could result in loss of control of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective January 6, 2003.

ADDRESSES: Information pertaining to this AD may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1175; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: \boldsymbol{A}

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Airbus Model A300 B2 and B4 series airplanes; A300 B4–600, B4–600R, and F4–600R (collectively called A300–600) series airplanes; and Model A310 series airplanes, was published in the **Federal Register** on July 17, 2002 (67 FR 46937). That action proposed to require revising the Airplane Flight Manual to advise the flightcrew to don oxygen masks as a first and immediate step when the cabin altitude warning horn sounds.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 168 Airbus Model A300 B2 and B4; A300–600; and Model A310 series airplanes of U.S. registry will be affected by this AD. It will take approximately 1 work hour per airplane to accomplish the required actions, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$10,080, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a

"significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2002–24–04 Airbus: Amendment 39–12969. Docket 2002–NM–40–AD.

Applicability: All Airbus Model A300 B2 and B4 series airplanes; A300 B4–600, B4–600R, and F4–600R (collectively called A300–600) series airplanes; and Model A310 series airplanes; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent incapacitation of the flightcrew due to lack of oxygen, which could result in loss of control of the airplane, accomplish the following:

Revision to the Airplane Flight Manual

- (a) Within 90 days after the effective date of this AD, accomplish paragraph (a)(1) or (a)(2) of this AD, as applicable, to advise the flightcrew to don oxygen masks as a first and immediate step when the cabin altitude warning horn sounds.
- (1) For Model A300 series airplanes, revise the Emergency Procedures section of the FAA-approved Airplane Flight Manual (AFM). This may be accomplished by inserting a copy of this AD in the AFM.

"EMERGENCY PROCEDURES

Cabin Depressurization:

Crew Oxygen Masks ON
Crew Communications .. established
Passenger Oxygen as required

"EMERGENCY PROCEDURES— Continued

Emergency Descent as required (see 3.02.00 page 8)"

(2) For Model A300–600 and A310 series airplanes: Revise the Procedures Following Failure section of the FAA-approved AFM. This may be accomplished by inserting a copy of this AD in the AFM.

"PROCEDURES FOLLOWING FAILURE

Cabin Press:

Removal of AD From AFM

(b) When the information included in the AFM procedures specified in paragraphs (a)(1) and (a)(2) of this AD has been incorporated into the FAA-approved general revision of the AFM, and the information contained in the general revision is identical to that specified in this AD, this AD may be removed from the AFM.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Operations Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

Note: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from International Branch, ANM–116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Effective Date

(e) This amendment becomes effective on January 6, 2003.

Issued in Renton, Washington, on November 21, 2002.

Ali Bahrami.

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 02–30341 Filed 11–29–02; 8:45 am]

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