



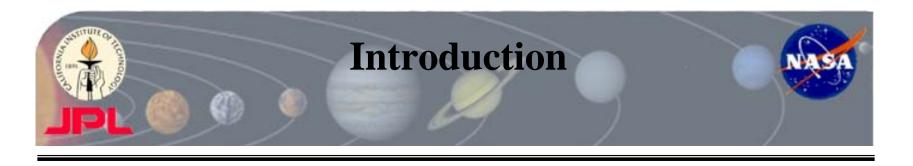
Teaming with the Terrestrial Planet Finder Formation Control Testbed

NASA Health & Safety Managers Meeting March 3, 2005

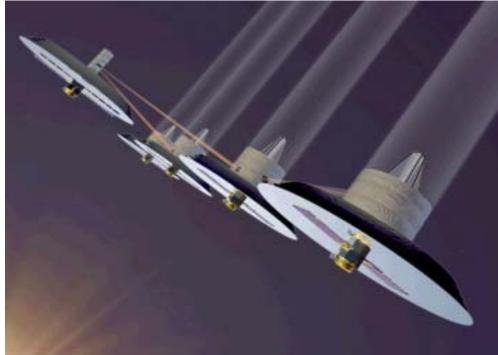
Presented by Michael Swanson JPL Occupational Safety Program Office

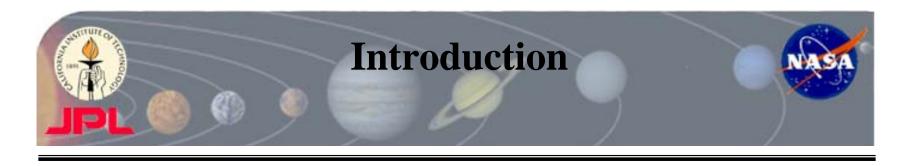


- Introduction
- Overview
- Invitation to the Environmental, Health, and Safety (EH&S) Program Office
- Health & Safety Issues
- Health & Safety Approach
- Safety as "Value Added"
- Acknowledgements



- Terrestrial Planet Finder (TPF) NASA Origins
 - Two complementary observatories that will study all aspects of planets outside our solar system, searching for earth-like planets that might harbor life:
 - A visible-light coronagraph, to launch round 2014
 - A formationflying infrared interferometer, to launch before 2020

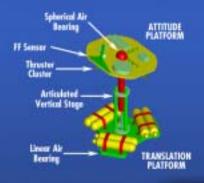




- Formation Control Testbed (FCT)
 - 6-DOF robotic testbed that will demonstrate an endto-end precision formation flying system
 - Purpose:
 - Validate formation control architecture and algorithms
 - Demonstrate autonomous TPF-like precision formation flying with three robots to a performance level of:
 - 10 m diameter operating space
 - Up to 8 m separation
 - ± 5 cm range control
 - ± 5 arc minute attitude control
 - ± 60 arc-minute bearing control

FORMATION CONTROL TESTBED (FCT) System-level validation of formation flying control in a hardware testbed

CONCEPT



- Three robot formation
- Six degrees-of-freedom using spherical/linear air-bearings
- Robot Size/Weight: ~4 ft tall x 4ft diameter, ~780 lbs each
- · Operation duration per charge of gas: ~1 hr
- Mechanical flat floor with glass top
- Operating area: ~30ft diameter
- Performance:
 - Formation Control: +/- 5cm
 - Attitude Control: +/- 5 arc-min
- · On-board Avionics (each robot):
 - Avionics Computer (PPC750), Cold gas propulsion 16 thrusters, Reaction Wheels – 3 single axis, Pseudo-Stor Tracker, Formation
 - Sensor (inter-s/c range/bearing), Wireless communication
- Flight-like commanding and telemetry

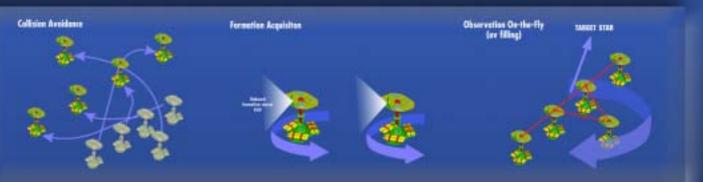


NASA

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Persetens, California

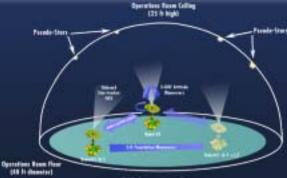
FORMATION FLYING DEMONSTRATIONS



FCT OPERATIONS



Formation Technology Lab (FTL)





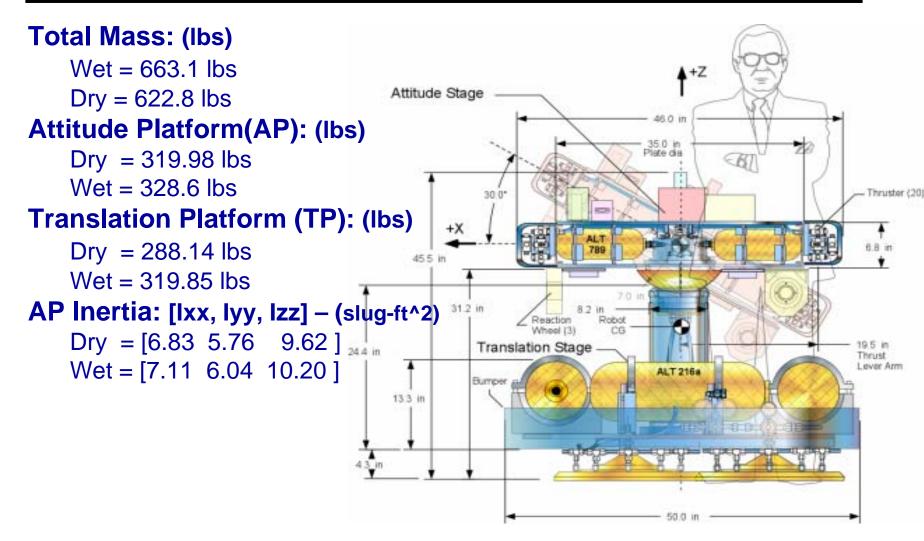


Massic Flat Floar with Glass Surface Panels (top view)



Single Flat Floor Panel (adjustable level)







- FCT Personnel Invited the Occupational Safety Program Office (OSPO) and the Systems Safety Program Office (SSPO) to participate in the Preliminary Design Phase (April 2003)
 - Initial questions related to the high pressure systems for the air bearings and cold-gas thrusters.
- Contact was Suggested by Project Management!



- Operational Safety
 - Physical hazards of three 600-pound robots under remote and computer algorithm control
 - Noise from operational thrusters
- Pressure Safety
 - High pressure air (3000 psig) stored on each robot to power linear and spherical air bearings and thrusters
- Lifting Devices Safety
 - Robot and flat floor mobilization
- Systems Safety
 - Protect critical hardware



1. Safety through "Engineering design/build"

- a. Designed in safety through selection of approved system components
- b. Designed in recommended safety margins
- c. Designed in pneumatic and electrical safety features
- d. Safety testing during Integration & Test phase (key milestones identified)

2. Safety through "Administrative Procedures"

- a. Safe Operating Procedures (SOPs)
- b. Pre-Operational Safety Review
- c. Task-Specific Health & Safety Plan
- d. Training/certification/indoctrination (institutional, site-specific)

3. Safety through "Protective Equipment"

- a. Eye glasses/face shields, gloves etc.
- b. Warning alarm (visible and/or audible)



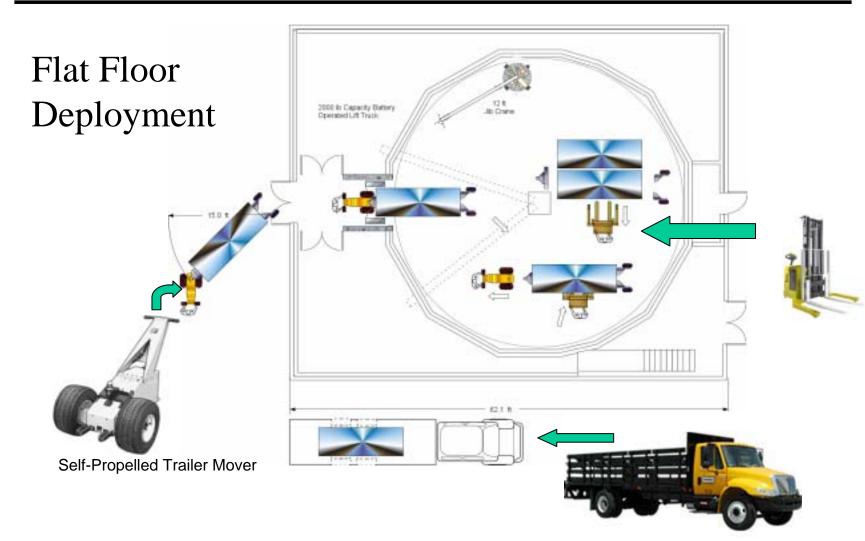
Pressure Safety Design Features

- 1. Different gender fill ports for the high and low pressure fill.
- 2. Extensive use of pneumatic safety devices (relief devices, check valves, vent valves etc.)
- 3. Extensive instrumentation of the system with pressure transducers, and gauges to provide enhanced visibility into the operational health of the system.
- 4. Color coded pneumatic lines for different pressure levels.
- 5. Refill System regulator to prevent overfilling the Robot on-board pressure vessels.
- 6. Positive hold-switch with a key at the Robot Refill Station, which requires the Refill Operator to hold the key in the on-position for the refill operation. Releasing the key will terminate the refill operation and vents the line via the three-way valve outside of the building.
- 7. Use of elastomeric gaskets under the band-clamp mounts of the pressure vessels, allowing the pressure vessels to expand during fill operational while still providing needed mechanical clamping force to keep the pressure vessels secure.
- 8. Use of mechanical stops to limit the radial adjustment of pressure vessels in the Attitude Platform to prevent inadvertent de-mating of a pressurized pressure vessel during wet/dry center of gravity balancing.
- 9. An independent low pressure pressurization of the Robot(s) for manual floatation and/or powered operations without requiring filling the on-board pressure vessels. Enhanced safety during integration, test, and maintenance activities.



- 1. Bumper ring around the FCT Robot Translation Platform using shock absorbers, to prevent direct collision and absorb impact forces in case of a collision.
- 2. Railing around the flat floor (panels) edges at the same height as the Robot bumper rings to prevent the robot to fall off the operating flat floor surface.
- 3. Protective cages around each of the four thruster clusters to protect against physical damage from impact.
- 4. Use of different size electric umbilical connectors for the Translation Platform and the Attitude Platform of the Robot external and recharge power.
- 5. Use of smart battery charger to recharge the on-board batteries, which adjusts the charging current based on the sensed level of battery charge.
- 6. Incorporation of a large easily accessible/visible panic button to shutdown the on-board avionics.
- 7. Use of work stand/support legs for the Attitude and Translation Platforms. These support legs allows for Attitude and Translation Platforms to kept off the work area surface for ease of maintenance as well as to minimize the possibility of damage to critical air-bearing components.







Master Schedule Line No.	WBS No.	Task/Milestone Description	Task/MS	Start Date	Completion Date
54	1.1.1.3.4.1	Translation Stage Pneumatics Proof & Leak Tests	т	5/25/2004 8:00	5/25/2004 17:00
55	1.1.1.3.4.2	Translation Stage Floatation Tests	T	5/26/2004 8:00	
66	1.1.1.2.2.3	Switch Box Control Console	T	6/22/2004 10:40	
114	1.1.2.2.1	Integrated Attitude Stage Tests	T	6/3/2004 8:00	
119	1.1.2.2.5.1	Proof & Leak Test Attitude Stage Pneumatics	T	5/24/2004 8:00	
123	1.1.2.2.6.1	Thrusters	T	5/12/2004 8:00	
124	1.1.2.2.6.2	Reaction Wheels	T	6/9/2004 8:00	
148	1.1.3.1	Floor/Translation Integration at Di-Tec	T	5/26/2004 8:00	
149	1.1.3.2	Tranportation & Handling Check Out at Di-Tec	T	6/29/2004 8:00	
150	1.1.3.3	Final Float Tests at Di-Tec	T	7/22/2004 10:40	
152	1.1.3.4.1	Electronics/SW at ACEi	Ť	7/8/2004 10:40	
153	1.1.3.4.2	Full Robot ACEi	T	7/15/2004 10:40	
155	1.2	Robot & Floor Transportation & Installation	T	7/27/2004 10:40	
158	1.3.1	Installation & Checkout of Bauer Compressor	Ť	7/8/2004 8:00	
159	1.3.2	Installation & Checkout of Fill Stations	Ť	7/13/2004 8:00	
160	1.3.3	Proof and Leak Testing of Fill Pressurization System	T	7/15/2004 8:00	
161	1.3.4	Installation and Proof Loading of Jib Crane & Hoist	Ť	6/23/2004 8:00	
162	1.3.5	Installation of Floor Platforms	T	7/30/2004 10:40	
164	1.3.7	Installation & Emplacement of Robot 1	T	8/4/2004 10:40	
177	1.4.1.9	Transportation & Handling Design Details	T	4/19/2004 8:00	
211	1.5.3	Robot Functional Demonstration Software	T	7/8/2004 10:40	
216	1.6.1.2	COTS Parts	T	4/15/2004 8:00	
268	1.6.2.4	Laboratory Processor Functional Demo Software (1 ea	T	7/15/2004 10:40	
273	1.7.1.1	Develop Master Schedule	Ť	3/17/2004 8:00	
273	1.7.1.2	Develop Naster Schedule Develop Safety Data Package	T	5/3/2004 8:00	
274	1.7.1.2.1	Proof, Leak & Relief Pressure Test Procedure - Attitud	T	5/3/2004 8:00	
276	1.7.1.2.2	Proof, Leak & Relief Pressure Test Procedure - Transla	T	5/4/2004 8:00	
270	1.7.1.2.3	Proof, Leak & Relief Pressure Test Procedure - Robot	T	6/9/2004 8:00	
278	1.7.1.2.4	Jib Crane & Hoist Operation Procedure	T	6/9/2004 8:00	
278	1.7.1.2.5	Motorized Tow Operation Procedure	T	6/9/2004 8:00	
275	1.7.1.2.6	Self Propelled Lift Operation Procedure	T	6/9/2004 8:00	
282	1.7.2	System Engineering	T	3/17/2004 8:00	
283	1.7.2.1	Update Robot Mass Properties	T	3/17/2004 8:00	
284	1.7.2.2	Robot Operation and Maintenance Procedures	Ť	7/22/2004 10:40	
285	1.7.2.3	Robot Transportation & Handling Procedures	T	6/29/2004 8:00	
285	1.7.2.4	Floor Operation & Maintenance Procedures	T	5/26/2004 8:00	
287	1.7.2.5	Floor Transportation & Handling Procedures	T	6/29/2004 8:00	
288	1.7.2.6	Robot Fill System Operation and Maintenance Procedu	T	7/1/2004 10:40	
289	1.7.2.7	Robot Battery Charging Operation and Maintenance Proceed	T	6/4/2004 8:00	
203	1.7.2.8	Laboratory Processor Operation and Maintenance Pro	T	7/20/2004 10:40	
290	1.7.2.9	As Built Drawing & Data Package	Ť	7/27/2004 10:40	
296	1.7.3.1.5	Pre-Operational Safety Readiness (OSR) Review	MS	7/9/2004 8:00	
296	1.7.3.1.7	Monthly Progress Reviews	T	4/12/2004 8:00	
299	1.7.3.1.8	Demonstration & Delivery Review	MS	7/21/2004 8:00	
301	1.7.3.1.2	Implementation Review	MS	3/31/2004 8:00	
301	1.7.3.1.2	Provide Project Schedule	MS	4/14/2004 8:00	
302	1.7.4.2.1	Attitude Stage Floatation Test	MS	6/23/2004 8:00	
323	1.7.4.2.1	Translation Stage Floatation Test	MS	5/31/2004 8:00	
325	1.7.4.3.2	Power system design complete	MS	6/2/2004 10:40	
333	1.7.4.3.2	Internal Pre Operational Safety Review	MS	6/30/2004 8:00	
555	1.7.4.4.1	Internal File Operational Safety Review	IVIO DIVI	0/30/2004 0:00	0/00/2004 17:00



First Robot Deployment August 24, 2004

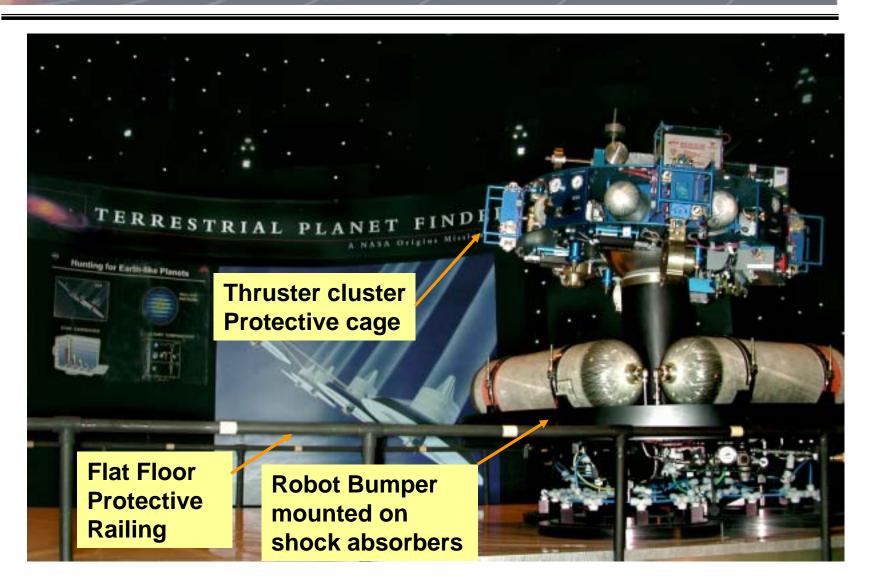
Flight Technician Team Logistics and Materials Team Protective Services Team







First Robot Operational September 2004





- 1. Carbon Dioxide initially considered and rejected due to concerns with building ventilation and fill logistics air compressor and storage system used instead.
- 2. Initial pressure system design did not have pressure relief to protect against regulator failure additional relief devices added.
- 3. Initial cylinders proposed for Translation Stage limited to 100 cycles by the DOT exemption replaced with cylinders with unlimited cycles.
- 4. Review of proposed pressure component list revealed high pressure gauge with no pressure relief replaced with gauge with built-in pressure relief.
- 5. Testing of as-delivered relief devices on the storage pressure vessels revealed lower than expected relief pressure setting on a number of them replaced/reset with the proper relief setting and certified prior to installation.
- 6. On-site check during installation revealed thinner than required tubing for Refill System highpressure (5000 psig) lines – upgraded to thicker wall thickness in compliance with ASME standards.
- 7. Eliminated the use of Jib crane for critical lift early in the design phase to stay compliant with NASA Standard an alternate approach developed in a timely manner.
- 8. Facility flooding risk identified, documented and communicated to management months prior to FCT hardware deployment interim steps taken during rainy periods by deploying roof tarp and sandbags to safeguard against leaks and flooding, while detailed design was completed for roof rework and re-grading with new concrete for proper drainage.



- Director for Astronomy and Physics: Larry Simmons
- TPF Project Manager: Dan Coulter
- FCT Team:

